Problem Set 3 - Solutions

Problem 1

$$\vec{F}_{g} = -mg\hat{y} = -mg\left(\sin\phi\hat{r} + \cos\phi\hat{o}\right)$$

$$= -mg\sin\phi\hat{r} - mg\cos\phi\hat{o}$$

$$\vec{F}_{n} = -\vec{T}_{n}\hat{r}$$

$$= \frac{1}{7} = \frac{1}{7} + \frac{1}{7} = (-mg \sin \theta - \frac{1}{7})\hat{\gamma}$$

$$-mg \cos \theta \hat{\theta}$$

===
$$T_{\gamma} = -mg \sin \phi - \overline{T}_{n}$$
, $\overline{T}_{\sigma} = -mg \cos \phi$ (1)

b)
$$F_{q}$$
. (1.48) with $r = R = conit$, $\dot{r} = \dot{r} = 0$:

 $F_{r} = -mR\dot{o}^{2}$, $\overline{T}_{0} = mR\dot{o}$ (2)

() component
$$\overline{T}_n$$
 of (11, (2))
=) - mg sin σ - \overline{T}_n = - m $R \dot{\phi}^2$
centripetal force

If
$$T_n = 0$$
 at $top of loop $(o = 90^\circ)$:

$$= : - mg \sin(90^\circ) - T_n = - mRo^2$$

$$= : - mg = - mR(\frac{v}{R})^2$$

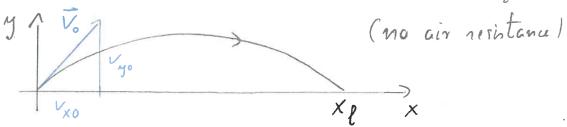
$$= : - mg = - mR(\frac{v}{R})^2$$

$$= : - wg = - mR(\frac{v}{R})^2$$$

This is the minimum speed of the car
to stay on the track b/c $\overline{T}_n = 0$ means
that the car is losing contact with the track

Problem 2 Range of projectile

 $2.15 \star$ Since the only force is the projectile's weight mg, Newton's second law implies that $\ddot{\mathbf{r}} = \mathbf{g}$ and its two components can be integrated twice to give the well-known results $x = v_{xo}t$ and $y = v_{yo}t - \frac{1}{2}gt^2$ (if we take $x_o = y_o = 0$). At landing, y = 0, which gives the time of flight as $t = 2v_{yo}/g$. The range is just the value of x at this time, namely $x_p = v_{xo}t = 2v_{xo}v_{yo}/g$.



Problem 3 ĒTÜ Ŷſ $\hat{z} \stackrel{\cdot}{\bigcirc} - \hat{\chi}$ = out of popular plane $\vec{F} = q\vec{E} + q\vec{v} \times \vec{r} \quad ; \quad \vec{E} = E\hat{g}$ $\vec{v} \times \vec{B} = (v\hat{x}) \times (B\hat{z}) = v\hat{B} \hat{x} \times \hat{z} = -v\hat{B}\hat{y}$ $= -\hat{g}, \text{ su below}$ $= \hat{f} = g(E - \nu B)\hat{g} = 0$ => $V = \frac{E}{B}$ Application: velocity selector

(Wien filter), google it!

products of unit vectors: $\hat{x} \cdot \hat{x} = 1$, $\hat{x} \cdot \hat{y} = 0$, etc in short: $\hat{e}_i \cdot \hat{e}_j = \delta_{ij} := \begin{cases} 1, i=j \\ 0, i\neq j \end{cases}$ in short: $\begin{cases} \hat{e}_i \cdot \hat{e}_j = \delta_{ij} := \begin{cases} 0, i\neq j \\ 0, i\neq j \end{cases}$ in short: $\begin{cases} \hat{e}_i \times \hat{e}_j = \epsilon_{ij} \\ \hat{e}_i \times \hat{e}_j = \epsilon_{ij} \\$

Problem 4

$$\otimes \vec{B}$$

= into plane

 \vec{a}
 \vec{b}
 \vec{b}
 \vec{b}
 \vec{c}
 \vec

a)
$$\hat{\gamma} = \cos(\varphi) \hat{x} + \sin(\varphi) \hat{y}$$

with $\varphi = \varphi(t) = \omega t$

$$= -\frac{1}{2} \int_{\tau} \hat{\gamma} = [-\sin(\varphi) \hat{x} + \cos(\varphi) \hat{y}] \int_{\omega}^{\pi} d\varphi$$

$$\frac{d}{dt} \hat{Q} = \left[-\cos(Q)\hat{x} - \sin(Q)\hat{y}\right] \frac{dQ}{dt}$$

$$= -\omega \hat{\gamma} \qquad (2)$$

b)
$$\vec{r} = R\hat{r}$$

$$\vec{v} = \frac{d}{dt}\vec{r} = R\frac{d}{dt}\hat{r} = R\omega\hat{o}$$

$$\vec{a} = \frac{d}{dt}\vec{v} = R\omega\frac{d}{dt}\hat{o} = -R\omega^2\hat{r}$$
(3)

c)
$$\hat{o} \times \hat{z} = [-\sin(0)\hat{x} + \cos(0)\hat{y}] \times \hat{z}$$

$$= -\sin(0)\hat{x} \times \hat{z} + \cos(0)\hat{y} \times \hat{z}$$

$$-\hat{y}$$

$$= \cos(0)\hat{x} + \sin(0)\hat{y} = \hat{r}$$
 (4)

$$= \frac{1}{7} = \frac{9}{7} \cdot x \cdot \vec{B} = \frac{9}{7} \cdot x \cdot \vec{B} = -\frac{9}{7} \cdot$$

1

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==
$$-gR\omega B\hat{r} = -mR\omega^2\hat{r}$$
 using (3), (5)

===
$$\omega = \frac{gB}{m}$$
 cyclotron frequency, independent of R!