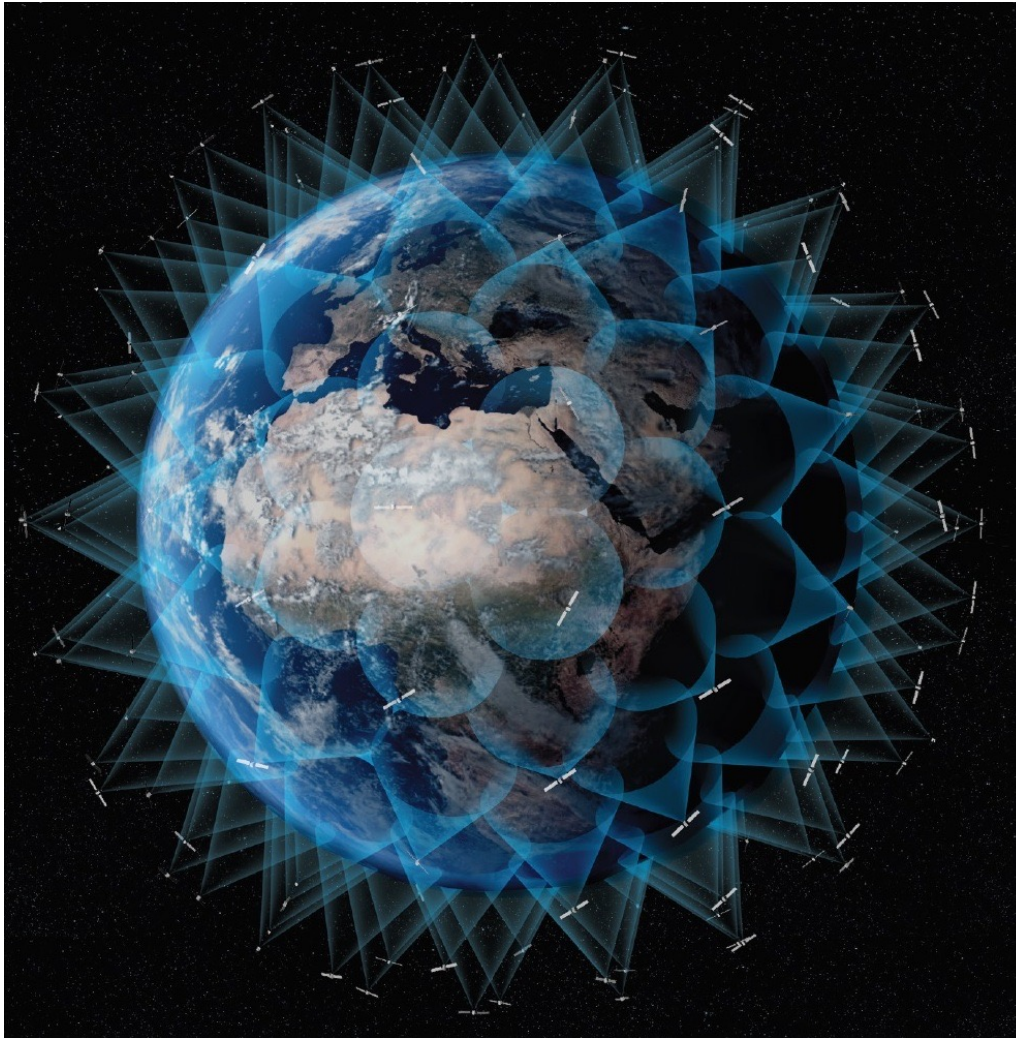


Developing inventories of by-products from satellite megaconstellation launches and disposal to determine the influence on stratospheric ozone and climate



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Eloise Marais, J. McDowell, S. Eastham



SpaceX Starlink



↑ 6166
↓ 401



Eutelsat OneWeb



↑ 640
↓ 6

~ 540,000 extra SMC satellites planned for Low Earth Orbit. New sustainability and debris guidelines will contribute to rapidly increasing launch rates and re-entry mass.

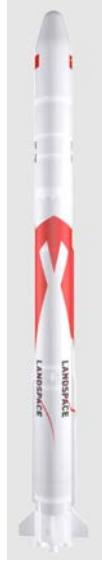
Launches (all atmospheric layers)



Hydrogen
Delta IV Heavy
LOX / LH₂
H₂O
Thermal NO_x



Kerosene
Falcon 9
LOX / RP1
H₂O
CO
BC
Thermal NO_x



Methane
Zhuque-2
LOX / CH₄
H₂O
CO
BC
Thermal NO_x



Hypergolic
Proton-M
N₂O₄ / UDMH
H₂O
CO
BC
Thermal NO_x
Fuel NO_x



Solid
Long March 11
Al / NH₄ClO₄ / HTPB
H₂O
CO
BC
Thermal NO_x
Fuel NO_x
Chlorine
Al₂O₃



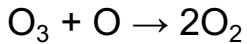
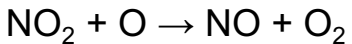
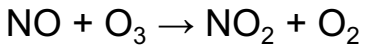
Reentries (upper atmosphere)

Payload/Rocket
Thermal NO_x
Al₂O₃

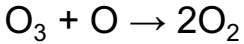
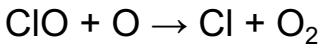
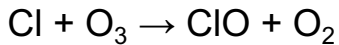
Pollutants released in all atmospheric layers.

Stratospheric Ozone Depletion

NO_x



Cl_y



Al_2O_3

Surface for
chlorine
activation

TOA Radiative Forcing

BC



Absorbs solar

Al_2O_3



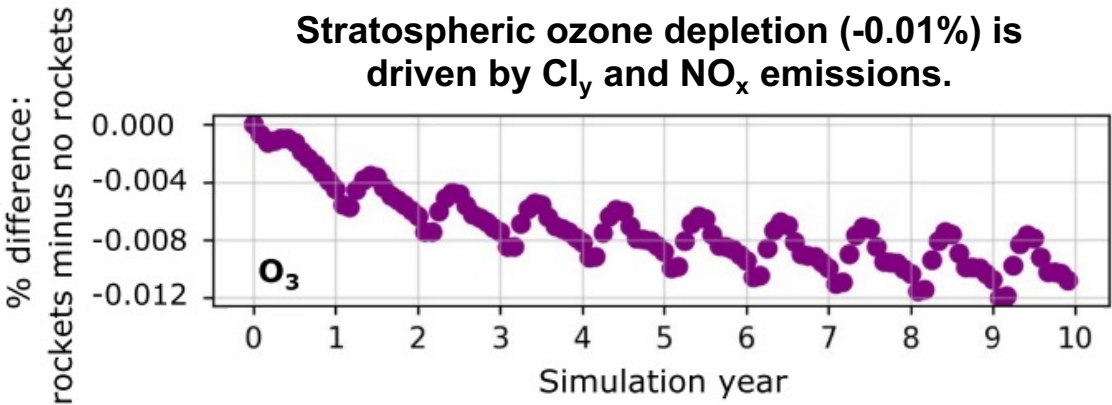
Absorbs terrestrial



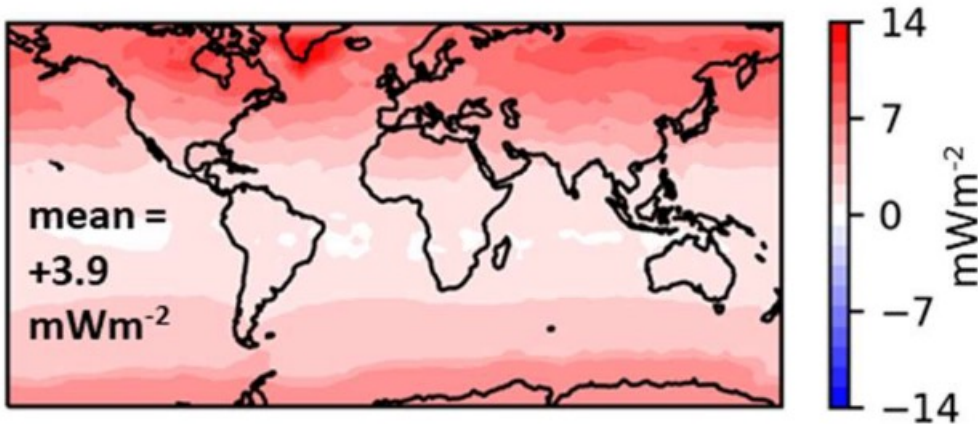
Reflects solar

Impact of a decade of increasing 2019 rocket launch and re-entry emissions

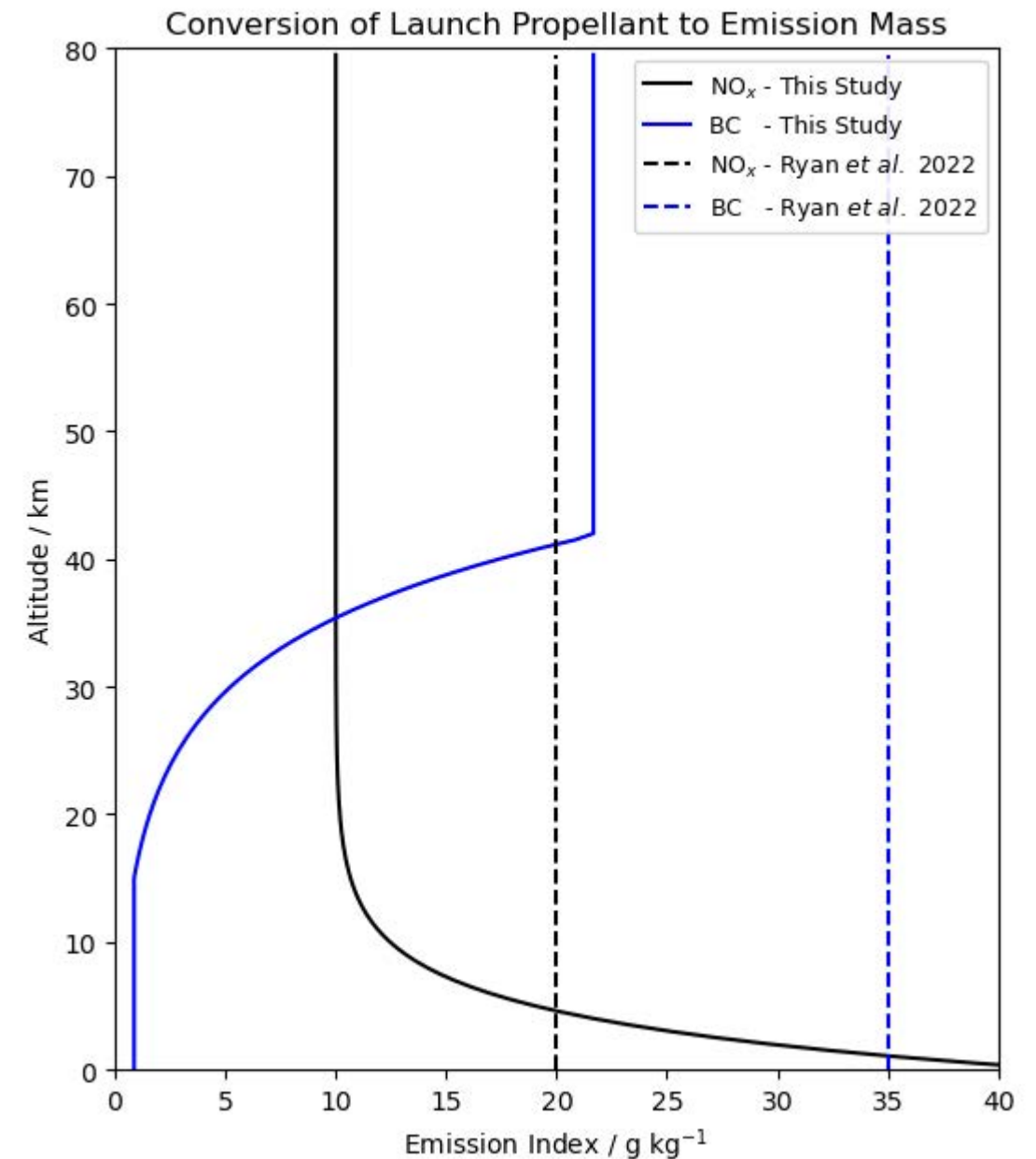
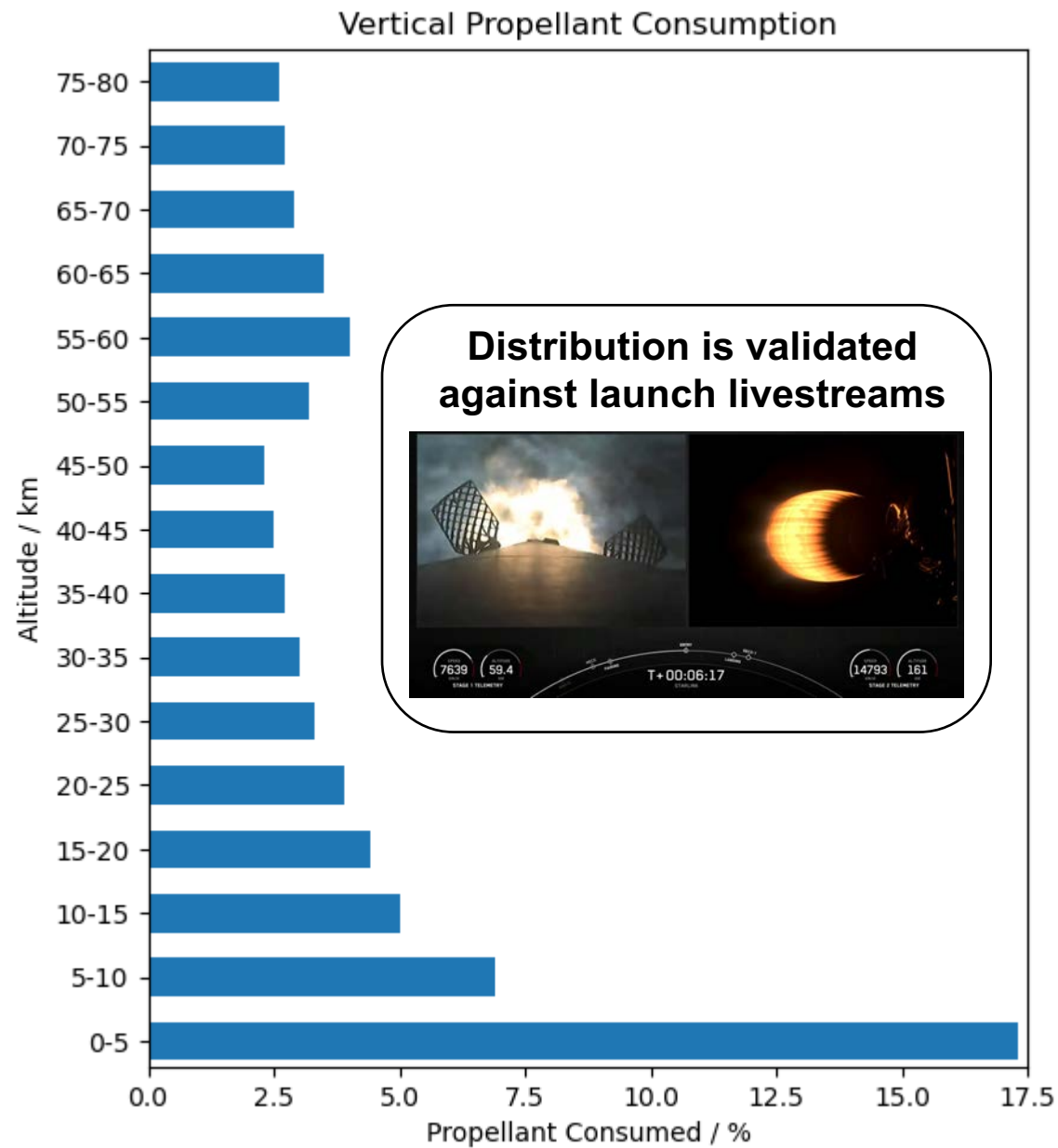
Stratospheric ozone depletion (-0.01%) is driven by Cl_y and NO_x emissions.



Global climate forcing at TOA is driven by BC emissions.



Creating a vertical distribution of launch emissions



Reusable Objects



17.5%
↓
NO_x

0%
↓
Al₂O₃

Expendable Objects

Composition:
Rocket Stages
70% Al



Survivability:
70% Core
35% Upper

Payloads
40% Al

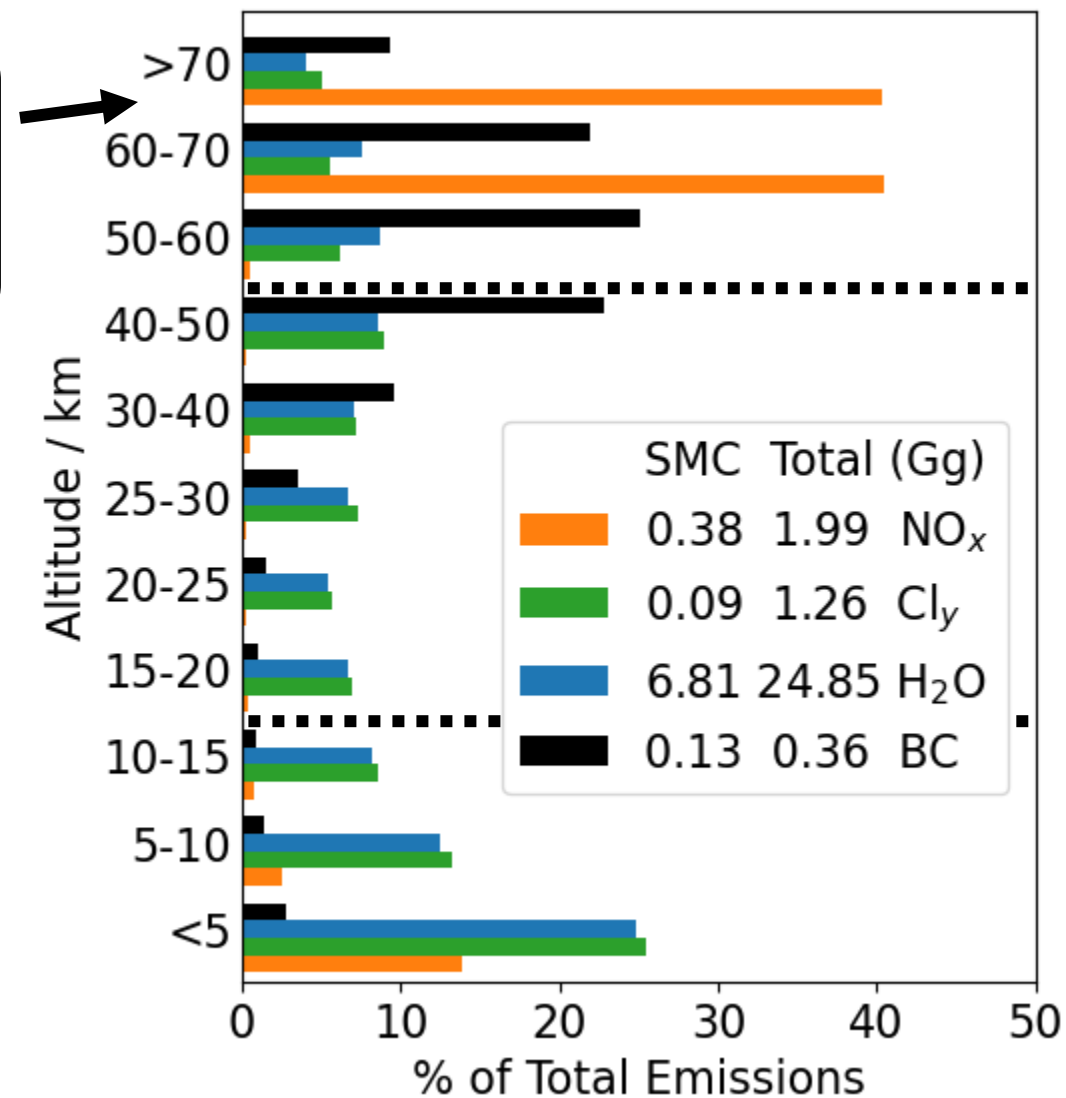
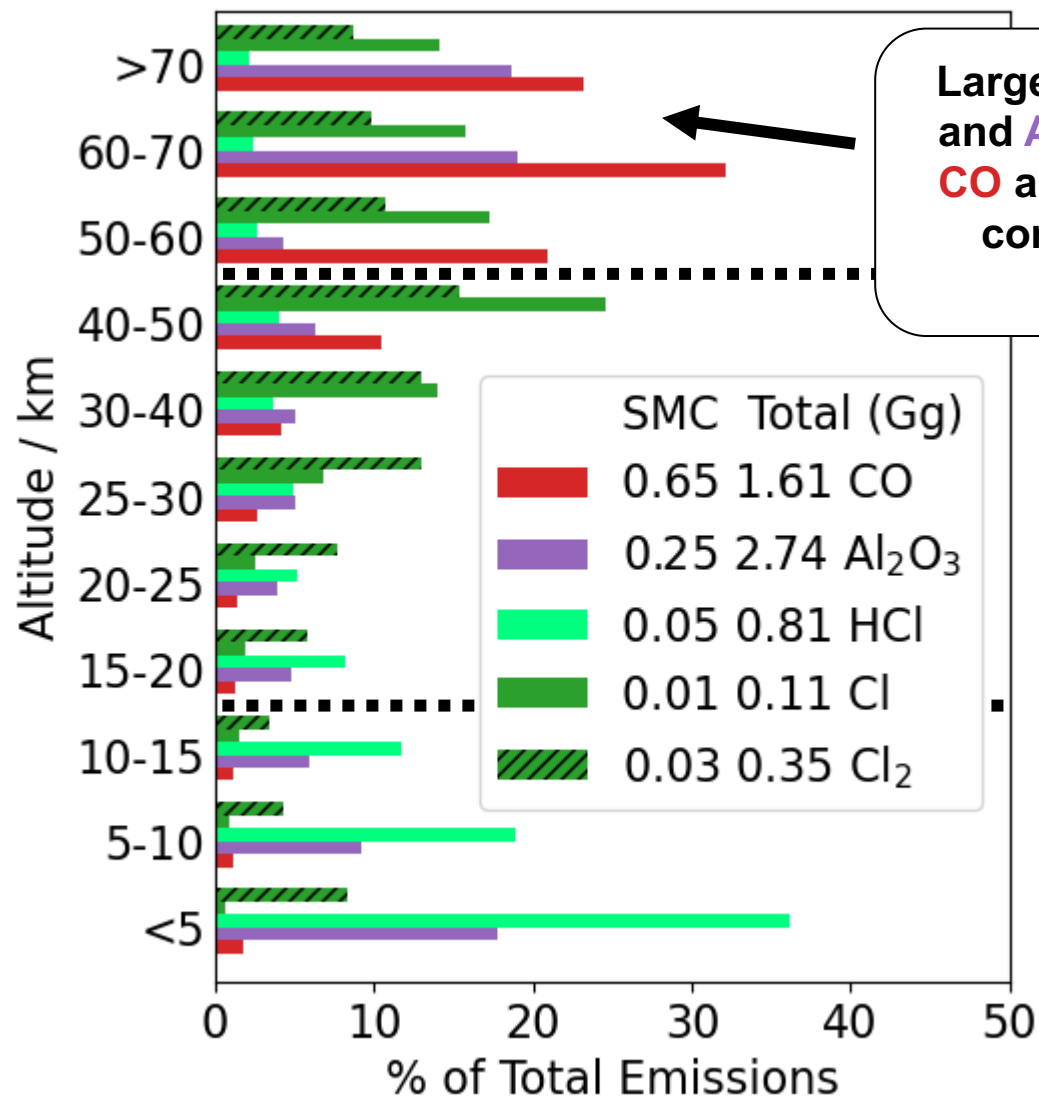


0% SMC
20% Non-SMC

40%
↓
NO_x

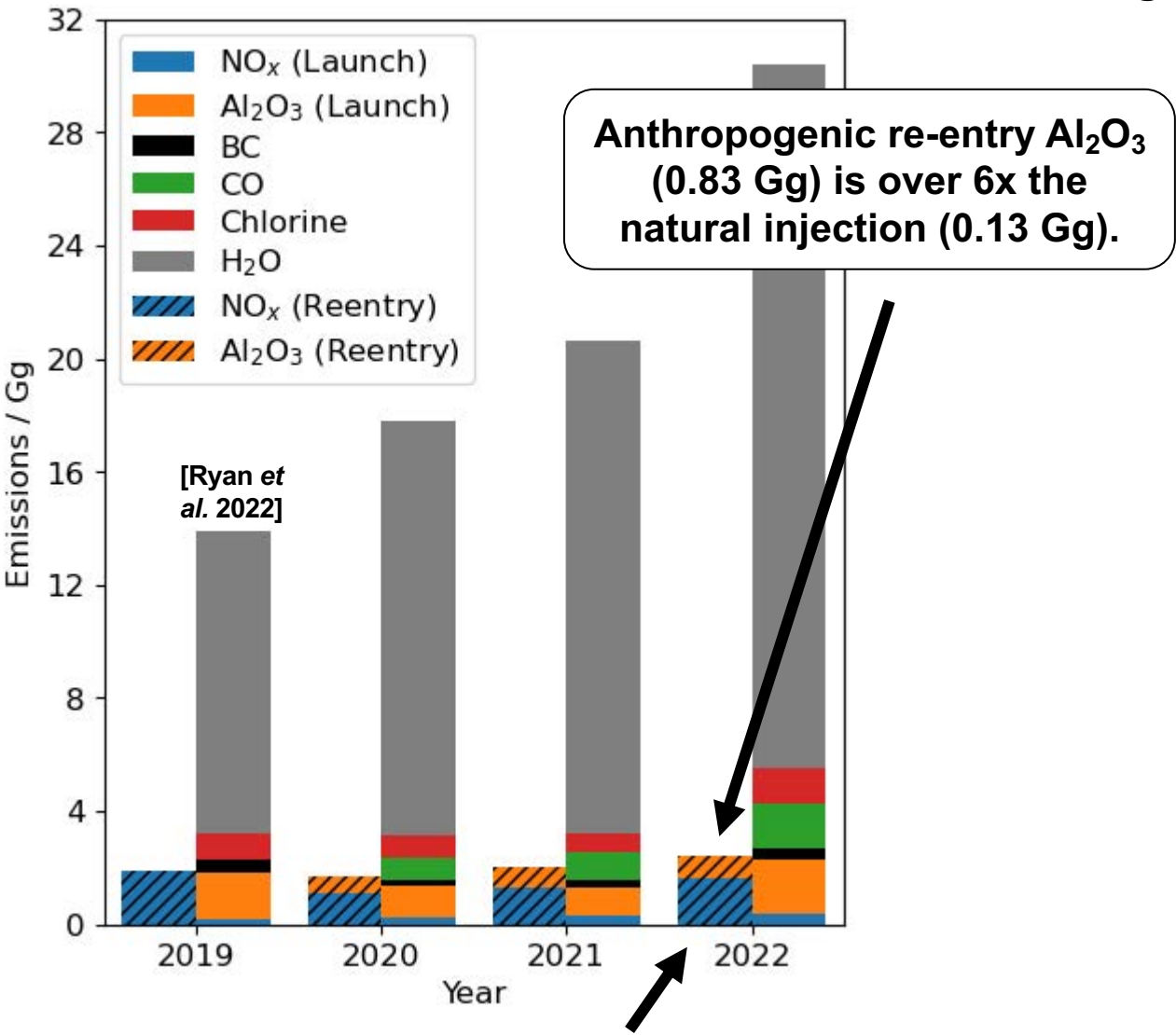
Variable %
↓
Al₂O₃

Re-entry mass has increased since 2020 (3.27-5.59 Gg, 878-1650 objects), partly driven by satellite megaconstellations (17-23%). Conversion to emissions requires broad assumptions on ablation and chemical composition.



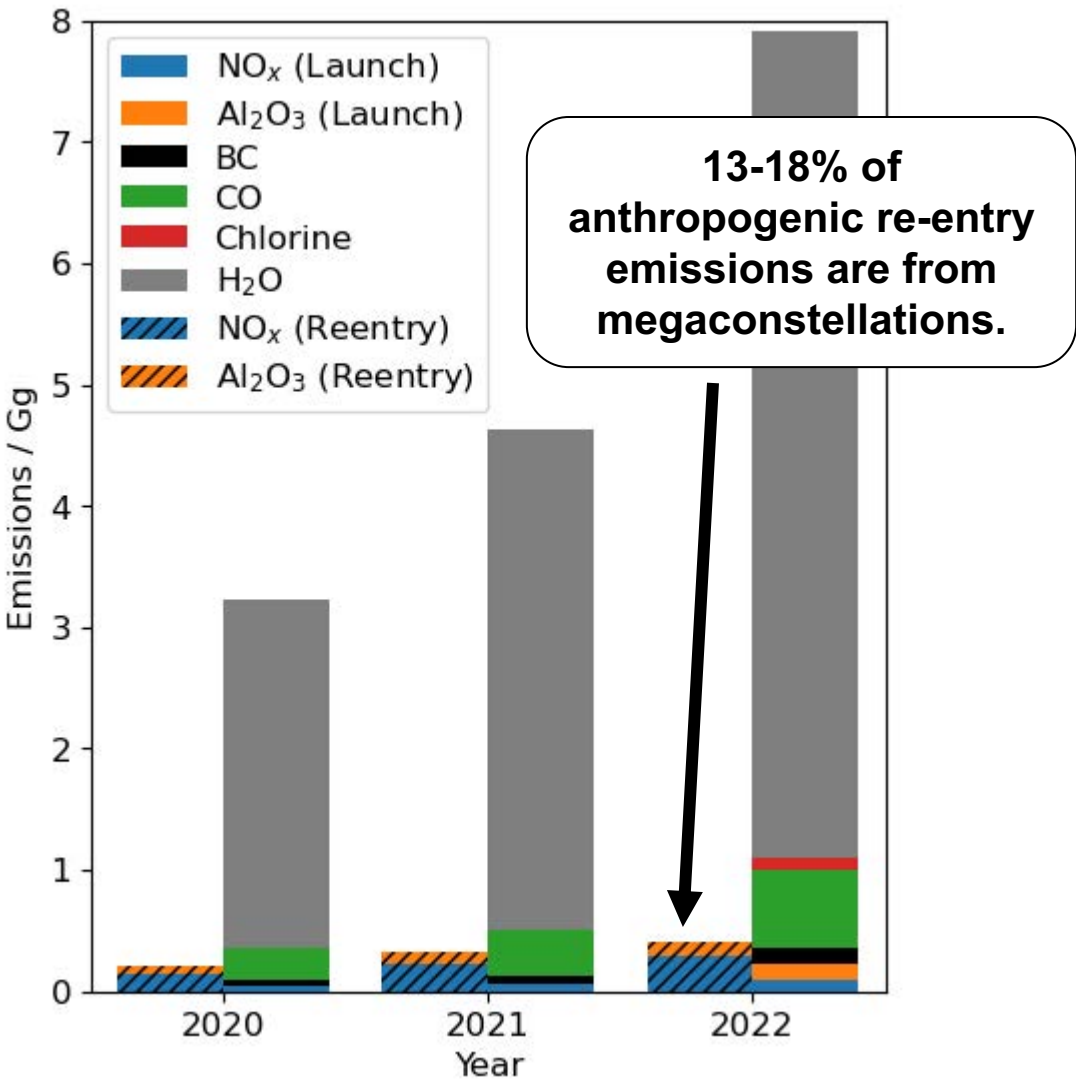
Most BC, NO_x , H_2O , CO, Cl_y , and Al_2O_3 emissions are injected above the tropopause.

All rocket launches and re-entries



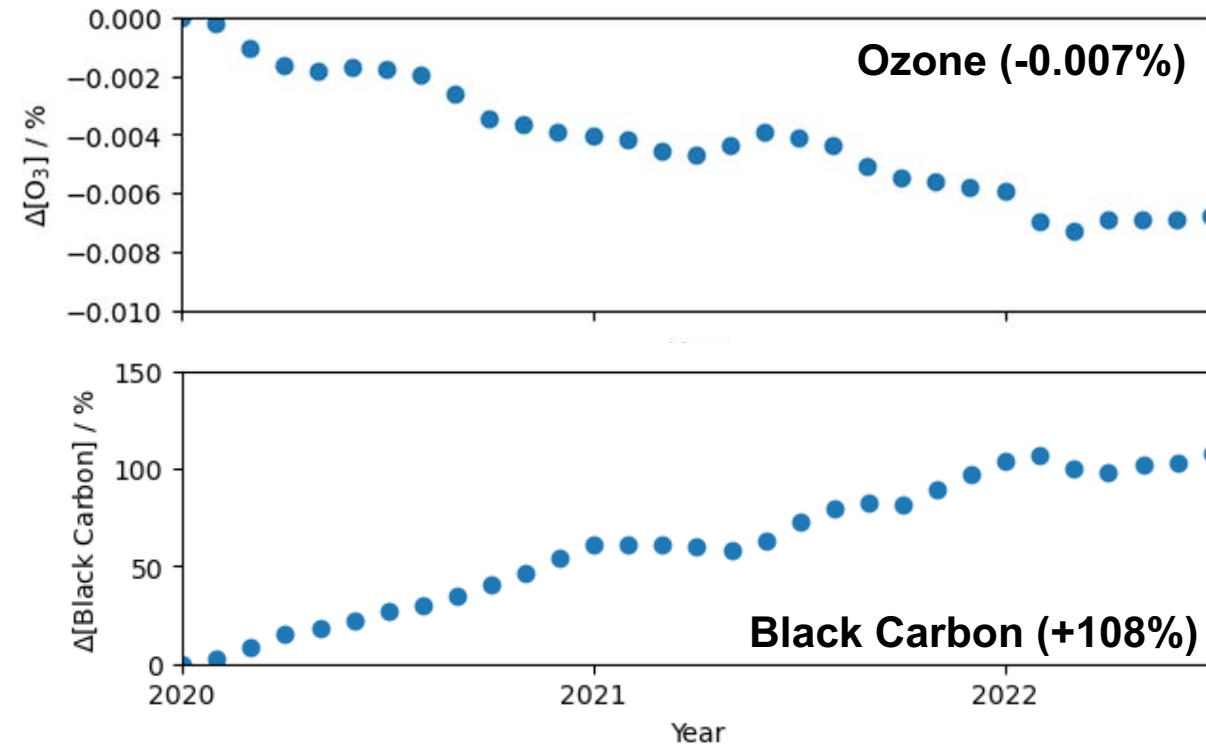
Anthropogenic re-entry NO_x (1.60 Gg) is approaching the natural injection (2-40 Gg).

Megaconstellation rocket launches and re-entries

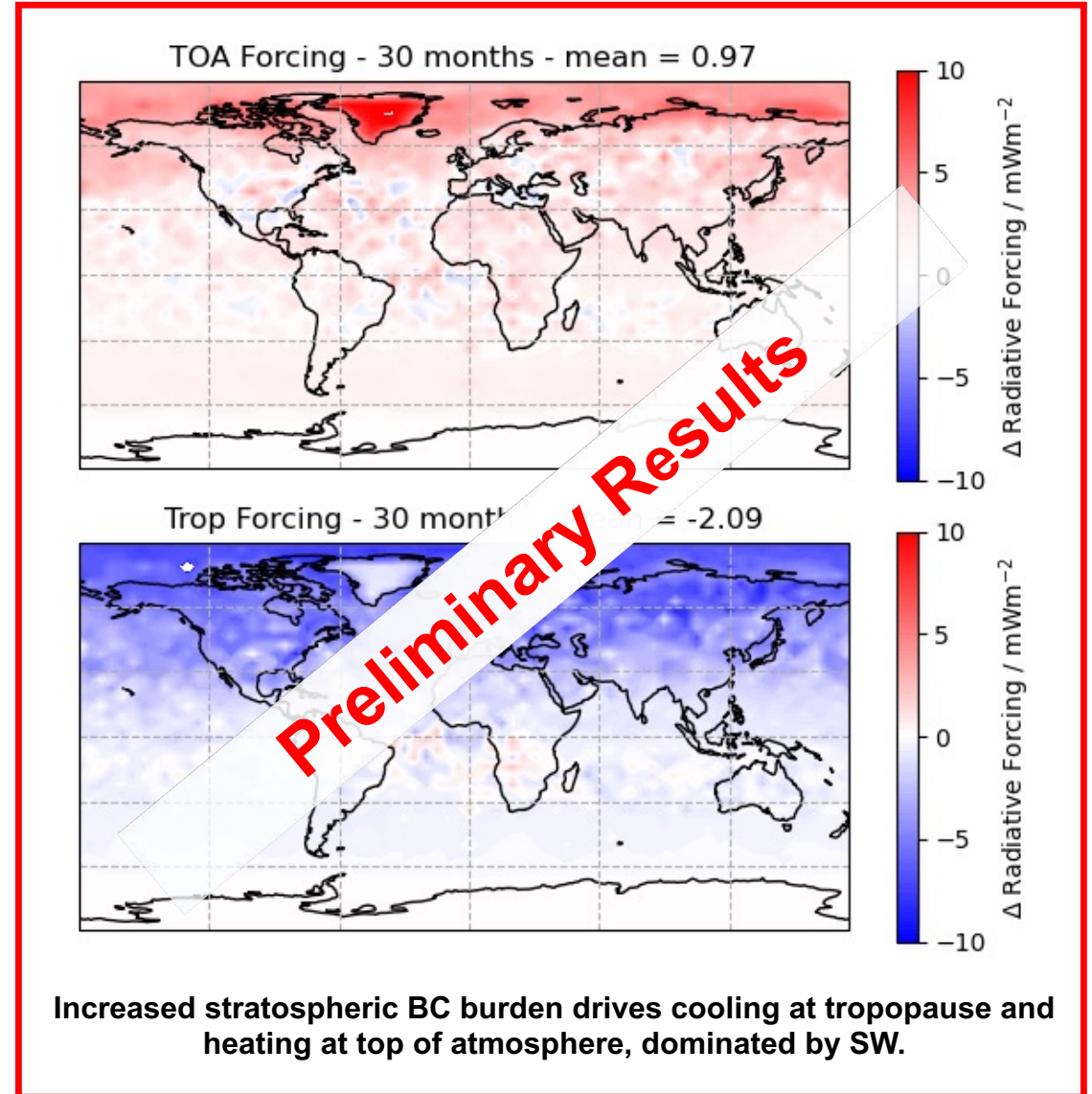


Kerosene fuel dominates megaconstellation launches, reducing harmful Cl_y and Al₂O₃ emissions.

Impact of improved 2020-2022 emissions on stratospheric concentration.



Global stratospheric ozone depletion after 2.5 years is approaching 10-year loss in Ryan *et al.* 2022 (-0.01%).



- **Compiled emission inventories for 2020-2022 SMC and non-SMC emissions.**
 - Launch and re-entry have risen from 2020-2022 for megaconstellation and non-megaconstellation sources.
 - Anthropogenic alumina re-entry emissions have exceeded the natural meteoritic injection.
- **Preliminary results demonstrate immediate environmental impacts.**
 - 2.5-years of increasing rocket launch and re-entry emissions result in global stratospheric ozone depletion of -0.007%.
 - Large increase in stratospheric black carbon burden (+108%).
 - Increasing rocket launch and re-entry emissions cause cooling at tropopause and heating at top of atmosphere.
- **More research/data is needed to address uncertainties:**
 - Experimental data for launch and re-entry emission indices at varying altitudes.
 - % survivability and chemical composition for each re-entering object.
 - Increased data availability from rocket manufacturers to aid research.
 - Particle size, mass distribution and optical properties of BC/Al₂O₃ aerosol emissions.
- **Next steps:**
 - Finish simulating the impacts of a decade of all launch and re-entry emissions on stratospheric ozone and climate.
 - Simulate the megaconstellation emissions only to see the individual environmental impact of SMCs.