

# A lifting line solver

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## 1 Reference

Lifting-line theory - Wikipedia

## 2 Lifting line theory summary

This section summarizes the main ideas and equations of lifting line theory:

### 2.1 Lift of an airfoil

The lift produced by an infinitesimal span airfoil  $L_y$  in steady state is given by the Kutta–Joukowski theorem:

$$\frac{dL}{dy} = L_y = \rho V \Gamma(y, \alpha) \quad (1)$$

Where  $\Gamma$  represents the circulation around the airfoil. If the airfoil is at an angle  $\alpha$ , the lift is given by:

$$\Gamma = VC(y, \alpha) = V2\pi c(y) \sin(\alpha(y)) \quad (2)$$

Where  $c(y)$  is the local chord length and  $\alpha(y)$  is the local angle of attack considering any wing twist.

The lift can be integrated over the span to obtain the total lift.

### 2.2 Lifting line

Because the circulation is not constant along the wingspan, the variation in circulation needs to be compensated by the shedding of a vortex. The shed vortices instead induce a vertical velocity over the wingspan.

This induced velocity can be computed by:

$$w(y) = \int_{-b/2}^{b/2} \frac{d\Gamma(\tilde{y})}{4\pi(y - \tilde{y})} \quad (3)$$

Note the integrand is singular at  $y = \tilde{y}$ , so it is understood in terms of the Cauchy principal value.

The circulation at a point can be extended to account for the effect of the induced velocity:

$$\Gamma(y) = V2\pi c(y) \left( \alpha(y) + \frac{w(y)}{V} \right) \quad (4)$$

$$= 2\pi c(y) \left( V\alpha(y) + \int_{-b/2}^{b/2} \frac{d\Gamma(\tilde{y})}{4\pi(y - \tilde{y})} \right) \quad (5)$$

And the total lift and drag can be computed as:

$$L = \rho V \int_{-b/2}^{b/2} \Gamma(y, \alpha) dy \quad (6)$$

$$D = \rho \int_{-b/2}^{b/2} \Gamma(y, \alpha) w(y) dy \quad (7)$$

Note: wikipedia states the induced drag includes  $\alpha(y)$  instead of  $w(y)$ , however this could lead to induced drag simply due to the angle of attack in the case of an infinite wing.

### 3 Solution

The solution process consists of find the circulation profile for a given wing geometry, i.e.  $c(y)$  and  $\alpha(y)$ .

In order to compute the circulation profile, we can discretize the problem and solve it numerically.

#### 3.1 Discretization scheme

Since we are considering an integral in the Cauchy singular value sense, we will discretize the integral using a rectangle rule.

The rectangle rule value is done as follows:

$$F(x) = \int_a^b f(x) dx \approx \sum_{i=0}^N f(x_i) \Delta_i x_i \quad (8)$$

Where  $\Delta_i$  is the difference operator which can be expressed as:

$$\Delta_i f(x_i) = \frac{f(x_{i+1}) - f(x_{i-1}))}{2} \quad (9)$$

Considering that at the limits we will instead use:

$$\Delta_0 f(x_0) = \frac{f(x_1) - f(x_0)}{2} \quad (10)$$

$$\Delta_N f(x_N) = \frac{f(x_N) - f(x_{N-1}))}{2} \quad (11)$$

This rule let us compute the Cauchy value by simply excluding the integrand at the singularity point.

### 3.1.1 Example of the integration of a Cauchy singular value

We could check the accuracy of this scheme by integrating:

$$F(x) = \int_{-1}^2 \frac{dx}{x} \quad (12)$$

The function is odd about the singularity at  $x = 0$ , which is the use case for the Cauchy singular value, so we can split the integration into two parts and play with the limits. Whose analytic solution is:

$$F(x) = \int_{-1}^{-\epsilon} \frac{dx}{x} + \int_{\epsilon}^2 \frac{dx}{x} = \quad (13)$$

$$= \int_1^{\epsilon} \frac{dx}{x} + \int_{\epsilon}^2 \frac{dx}{x} = \quad (14)$$

$$= \ln(x)|_1^{\epsilon} + \ln(x)|_{\epsilon}^2 = \quad (15)$$

$$= \ln(2) - \ln(1) = \ln(2) \quad (16)$$

TODO: check the discretization scheme against this problem.

## 3.2 Discretization of the circulation

Instead of solving the equation directly, we can discretize the problem and solve it numerically:

$$\Gamma(y_i) = 2\pi c(y_i) \left( V\alpha(y_i) + \frac{1}{4\pi} \sum_{j=1}^N \frac{\Delta_j \Gamma(y_j)}{(y_i - y_j)} \right) \quad (17)$$

Where  $\Delta_j$  is the difference operator defined above.

## 3.3 Solver

The will implement an iterative solver using considering the error and a step size  $\lambda$ , such that:

$$\tilde{\Gamma}(y_i) = 2\pi c(y_i) \left( V\alpha(y_i) + \frac{1}{4\pi} \sum_{j=1}^N \frac{\Delta_j \Gamma(y_j)}{(y_i - y_j)} \right) \quad (18)$$

And the next value in the iteration will be:

$$\Gamma(y_i) = \Gamma(y_i) + \lambda \left( \tilde{\Gamma}(y_i) - \Gamma(y_i) \right) \quad (19)$$

In the hope that it converges :)