

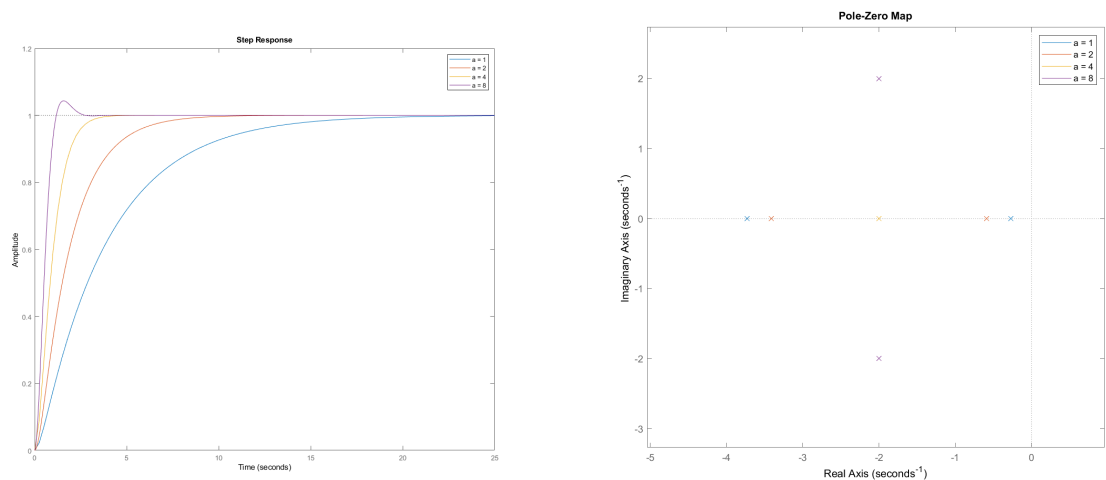
# ECEN315 : Modelling in MATLAB

## Assignment 2 : Submission

Daniel Eisen : 300447549

May 15, 2020

### Question 1



(a) Step Response for varying  $a$  values

(b) Corrospending Pole/Zero Map

Figure 1:  $\frac{O}{I} = \frac{a}{s^2+4s+a}$

For A = 1:  
T1 = 3.732051 T2 = 0.267949  
T = 1.000000  
Ts = 14.878894

For A = 4:  
T1 = 0.500000 T2 = 0.500000  
T = 0.500000  
Ts = 2.916978

For A = 2:  
T1 = 1.707107 T2 = 0.292893  
T = 0.707107  
Ts = 6.999617

For A = 8:  
T1 = 0.250000 T2 = 0.250000  
T = 0.250000  
Ts = 2.108152

MATLAB output

Time constants were calculated via the values given by the damp function ( $\omega_N$ ,  $\zeta$ ).

$$\tau_n = \frac{1}{\omega_N \cdot \zeta}$$

$$\tau = \sqrt{\tau_1 \cdot \tau_2}$$

Question 2

a)

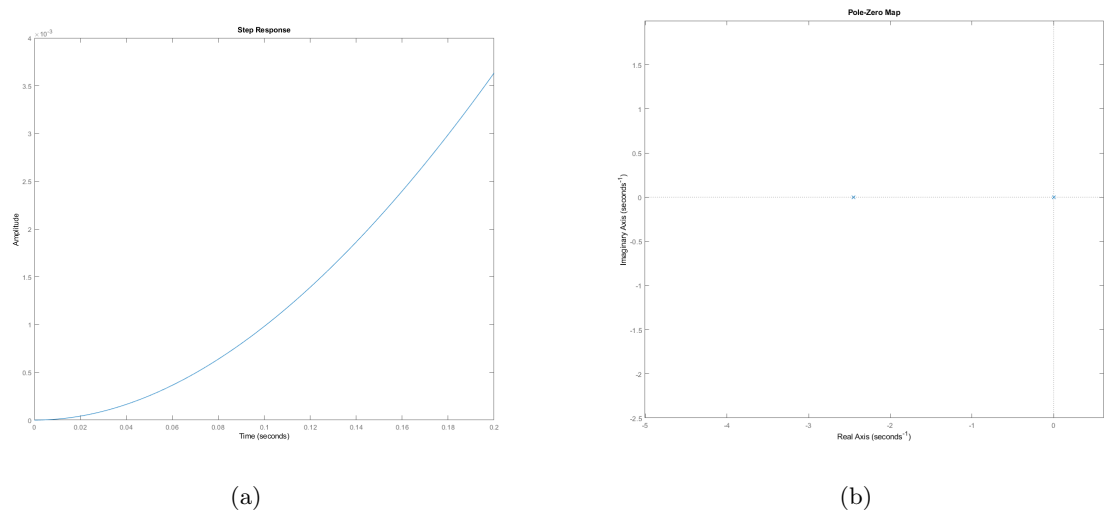


Figure 2:

Figure 2 shows that step response increases without bounds. This is however to be expected as its the angular displacement of the motor’s shaft. Its initial curve shows change but is evened out as it meets max angular velocity.

b)

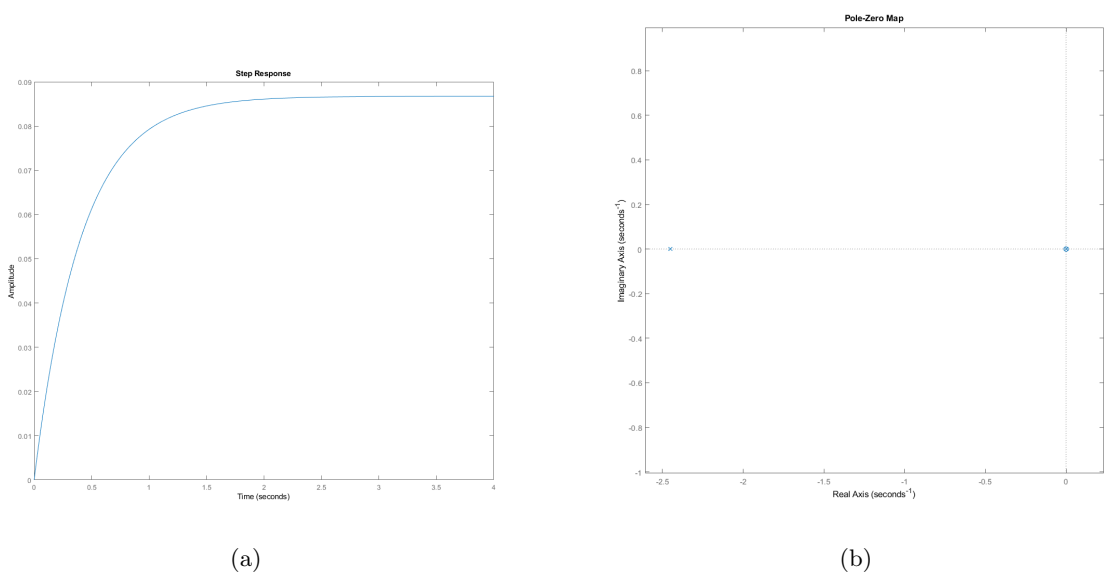


Figure 3:

Figure 3 shows the step response of the angular velocity, which is seen to reach a max value (steady state).

### Question 3

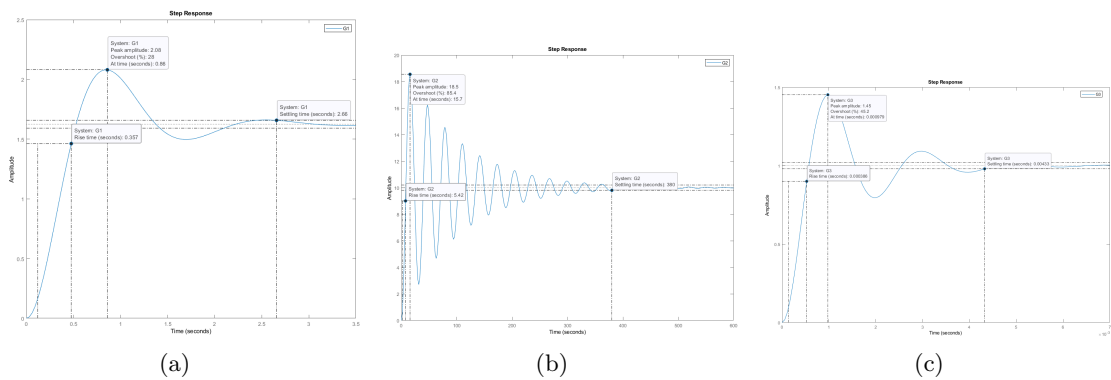


Figure 4:

$$G_1(s): Z=0.375000, Wn=4.000000, Ts=2.657041, Tp=0.859632, Tr=0.356904, OS=28.025548$$

$$G_2(s): Z=0.050000, Wn=0.200000, Ts=380.016288, Tp=15.707963, Tr=5.416048, OS=85.446128$$

$$G_3(s): Z=0.244567, Wn=3271.085447, Ts=0.004325, Tp=0.000979, Tr=0.000386, OS=45.241508$$

Figure 4 shows each of the of three systems:

$$G_1(s) = \frac{26}{s^2 + 3s + 16}$$

$$G_2(s) = \frac{0.4}{s^2 + 0.02 + 0.04}$$

$$G_3(s) = \frac{1.07 \times 10^7}{s^2 + 1.6 \times 10^3 s + 1.07 \times 10^7}$$

With the rise, settling and peak/overshoot data marked at their occurrence. The verbatim text shows the output of the MATLAB script’s calculated values; ( $\zeta$ ,  $\omega_n$ ,  $\tau_s$ ,  $\tau_p$ ,  $\tau_r$ , %OS)

### Question 4

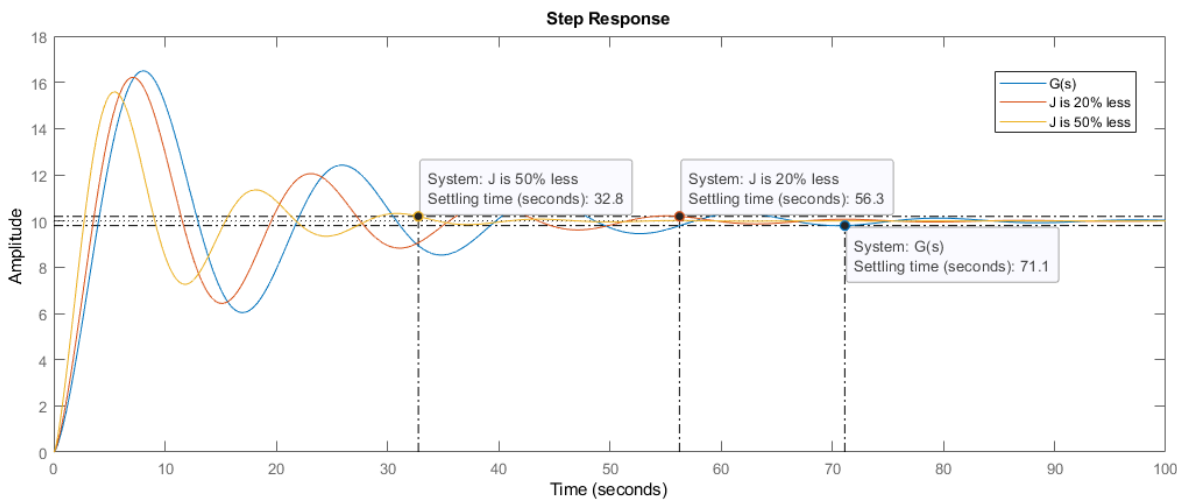


Figure 5:  $G(s) = \frac{\Theta(s)}{\Theta_d(s)}$

Figure 5 plots the output of a  $10^\circ$  step input to  $G(s)$ , compared with the reduction of  $J$  by 20% and 50% and the effect on the settling time displayed.

$$\text{MATLAB computed the TF to be equal to, } \frac{\Theta(s)}{\Theta_d(s)} = \frac{1.08e09s+1.08e09}{1.08e09s^3+8.64e09s^2+1.08e09s+1.08e09}$$

Question 5

a)

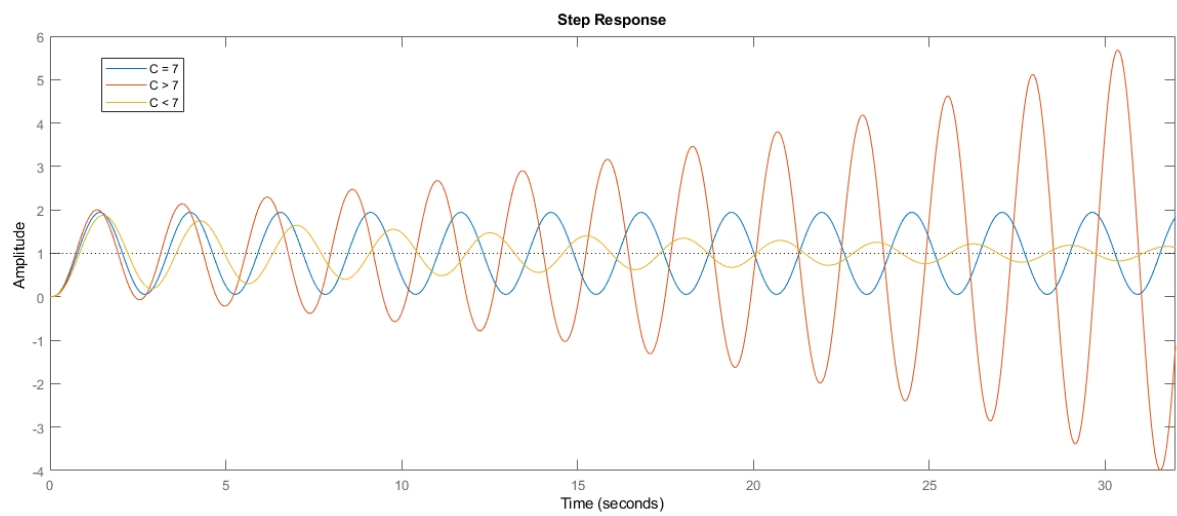


Figure 6:

MatLab Output: Minimum series gain > 7.000000

Using a search loop iterating the value of C, the gain value that gives a marginally stable system was found to be 7.

∴ the system is unstable at  $C > 7$

Figure 6 shows step response for the foundry case  $C=7$  and either side.

b)

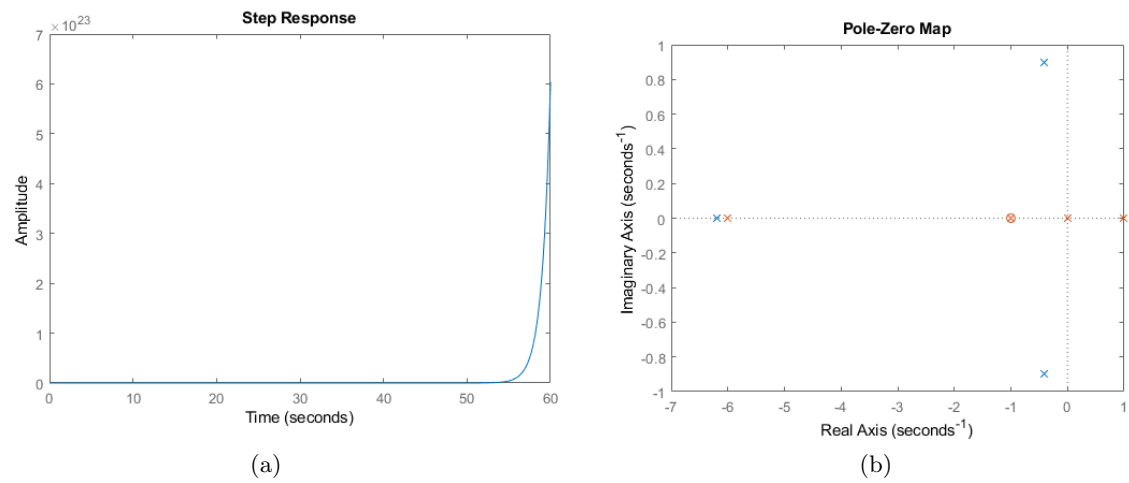


Figure 7:

Using a lead-lag controller  $C(\frac{s-z}{s-p}) = 0.01(\frac{s+1}{s-1})$  to place a pole at positive 1 with a lower steady state gain than (a): 7 compared to -0.01).

# Appendices

## Task 1

```
clear;
clc;

s = tf('s');

figure(1);
hold on
for a = [1 2 4 8]
    sys = a / (s^2 + 4*s + a);
    figure(1);
    step(sys,25);
    [Wn, Z] = damp(sys); %To get T1 and T2
    T1 = 1/Wn(1)*Z(1);
    T2 = 1/Wn(2)*Z(2);
    T = sqrt(T1*T2);
    fprintf("\nFor A = %d:\nT1 = %f T2 = %f\nT = %f\nTs = %f\n",...
        a, T1, T2, T,...
        stepinfo(sys).SettlingTime)
end
legend('a = 1', 'a = 2', 'a = 4', 'a = 8');
hold off;

figure(2);
hold on
for a = [1 2 4 8]
    sys = a / (s^2 + 4*s + a);
    [x,t] = step(sys,25);
    pzmap(sys)
end
legend('a = 1', 'a = 2', 'a = 4', 'a = 8');
hold off;
```

## Task 2

```
clear;
clc;
s = tf('s');

%a
motor_pos = 0.0425/(s*(s+2.45));
opt = stepDataOptions('StepAmplitude', 5);
figure("name","Motor Position Step Reponse");
step(motor_pos,opt,0.2)
figure("name","Motor Position Poles");
pzmap(motor_pos)

%b
motor_vel = motor_pos * s; %differentiate
opt = stepDataOptions('StepAmplitude', 5);
figure("name","Motor Velocity Step Reponse");
step(motor_vel,opt)
figure("name","Motor Velocity Poles");
pzmap(motor_vel)
```

### Task 3

```
clear;
clc;
s = tf('s');

G1 = 26 / (s^2 + 3*s + 16);
linearSystemAnalyzer(G1);
[Wn1,Z1] = damp(G1);
fprintf("G-1(s): Z=%f, Wn=%f, Ts=%f, Tp=%f, Tr=%f, OS=%f \n\n",...
        Z1(1), Wn1(1),...
        stepinfo(G1).SettlingTime,...
        stepinfo(G1).PeakTime,...
        stepinfo(G1).RiseTime,...
        stepinfo(G1).Overshoot)

G2 = 0.4 / (s^2 + 0.02*s + 0.04);
linearSystemAnalyzer(G2);
[Wn2,Z2] = damp(G2);
fprintf("G-2(s): Z=%f, Wn=%f, Ts=%f, Tp=%f, Tr=%f, OS=%f \n\n",...
        Z2(1), Wn2(1),...
        stepinfo(G2).SettlingTime,...
        stepinfo(G2).PeakTime,...
        stepinfo(G2).RiseTime,...
        stepinfo(G2).Overshoot)

G3 = 1.07E7 / (s^2 + 1.6E3*s + 1.07E7);
linearSystemAnalyzer(G3);
[Wn3,Z3] = damp(G3);
fprintf("G-3(s): Z=%f, Wn=%f, Ts=%f, Tp=%f, Tr=%f, OS=%f \n",...
        Z3(1), Wn3(1),...
        stepinfo(G3).SettlingTime,...
        stepinfo(G3).PeakTime,...
        stepinfo(G3).RiseTime,...
        stepinfo(G3).Overshoot)
```

### Task 4

```
clear;
clc;
s = tf('s');

% Constants
a=1;
b=8;
k=10.8E8;
J=10.8E8;

%a
controller = k*(s+a)/(s+b);
spacecraft = 1/(J*s^2);
G = feedback(controller*spacecraft,1) %closed loop TF

%Percent reduction
spacecraft1 = 1/((J*0.8)*s^2);
spacecraft2 = 1/((J*0.5)*s^2);

G1 = feedback(controller*spacecraft1,1);
G2 = feedback(controller*spacecraft2,1);

% degree input
G=G*10;
G1=G1*10;
G2=G2*10;

%plotting b and c
linearSystemAnalyzer(G,G1,G2)
```

# Task 5

```
%% a
clear;
clc;
s = tf('s');

G = 6 / (s*(s+6)*(s+1));

for c = (5:0.01:10)
    if stepinfo(feedback(c*G,1)).PeakTime >= Inf
        fprintf("Minimum series gain > %f\n", c)
        linearSystemAnalyzer(feedback(c*G,1), feedback((c+1)*G,1), feedback((c-1)*G,1))
        break;
    end
end

%% b
clear;
clc;
s = tf('s');

G = 6 / (s*(s+6)*(s+1));
%Lead Lag: C((s-z)/(s-p))
LL = 0.01*((s+1)/(s-1));
linearSystemAnalyzer(feedback(LL*G,1))
figure(1)
hold on
pzmap(feedback(G,1))
pzmap(feedback(LL*G,1))
hold off
```