Appendix A Problem 1 Python Code

```
# Compressible Flow
2 # AEE 553
3 # Homework 4 - Problem 2
4 # Evan Burke
6 from traceback import print_tb
7 import numpy as np
8 from matplotlib import pyplot as plt
9 from matplotlib.ticker import MaxNLocator
10 from matplotlib import cm
from mpl_toolkits.mplot3d import Axes3D
12 import shocks as ns
13 import oblique as os
14 import isentropic as isen
15 from scipy.optimize import fsolve
17 \text{ gamma} = 1.4
  delta = 1 # weak shock solution
  def find_beta(M=None,gamma=1.4,theta=None,delta=1):
      theta = np.deg2rad(theta)
21
      lamb = ((M**2-1)**2 - 3*(1 + (gamma-1)/2*M**2) * (1 + (gamma+1)/2*M**2)
22
     **2) * np.tan(theta)**2)**0.5
      chi = ((M**2-1)**3 - 9 * (1 + (gamma-1)/2 * M**2) * (1 + (gamma-1)/2 *
23
      M**2 + (gamma+1)/4*M**4)*np.tan(theta)**2)/lamb**3
      tan_beta = (M**2 - 1 + 2*lamb*np.cos((4*np.pi*delta+np.arccos(chi))/3)
     ) / (3 * (1 + (gamma-1)/2*M**2)*np.tan(theta))
      beta = np.arctan(tan_beta)
25
      beta = np.rad2deg(beta)
26
      print(f'Shock angle = {beta}')
27
      return beta
  def find_theta(M=None, beta=None, gamma=1.4):
30
      beta = np.deg2rad(beta)
31
      tanth = 2 / np.tan(beta) * (M**2 * np.sin(beta)**2 - 1) / (M**2 * (
32
     gamma + np.cos(2*beta)) + 2)
      theta = np.arctan(tanth)
33
      theta = np.rad2deg(theta)
34
      print(theta)
35
      return theta
37
38 t = find_theta(M=2,beta= 53.4229405)
40 betas = np.linspace(0.5,60,num=120,endpoint=True)
machs = np.linspace(1.1,15,num=120,endpoint=True)
42 print (betas)
43 print (machs)
```

```
45 data = []
46
47 for M in machs:
      for beta in betas:
          M1n = os.get_m1_normal(M1=M,beta=beta)
49
50
          M2n = os.get_m2_normal(M1n=M1n)
          p2_p1 = ns.get_static_pressure_ratio_normal_shock(M1=M1n)
51
          t = find_theta(M=M, beta=beta)
          data.append((M,beta,p2_p1,t))
54
55 print (data)
56
57 Xs = [point[0] for point in data]
58 Ys = [point[1] for point in data]
59 Zs = [point[2] for point in data]
60 ts = [point[3] for point in data]
61
62 '', fig = plt.figure()
63 ax = fig.add_subplot(111, projection='3d')
64 surf = ax.plot_trisurf(Xs, Ys, Zs, cmap=cm.jet, linewidth=0)
65 fig.colorbar(surf)
66 fig.tight_layout()
67 plt.show()','
68
69 import plotly.graph_objects as go
  '', marker_data = go.Scatter3d(
71
      x = Xs,
72
      y=Ys,
73
      z=Zs,
74
      marker=go.scatter3d.Marker(size=3),
75
      opacity=0.8,
      mode='markers'
77
  ),,,
79
80 data=[go.Scatter3d(x=Xs, y=Ys, z=Zs, mode='markers', marker_color=Zs,
     marker_colorscale='Viridis')]
#fig = go.Figure(go.Surface(x=Xs,y=Ys,z=Zs,colorscale="Plotly3",cmin=-5,
     cmax=5))
82 fig = go.Figure(data)
84 fig.update_layout(
      title='something',
      autosize=False,
86
      width=1000,
87
      height=1000,
88
          scene=dict(
          xaxis_title='X Axis Title',
90
        yaxis_title='Y Axis Title',
```

```
zaxis_title='Z Axis Title',
      ),
93
94 )
95
  '''fig = go.Figure(data=[go.Scatter3d(x=Xs, y=Ys, z=Zs, mode='markers',
                                      marker_color=ts, marker_colorscale='
98
     Viridis')])'''
99 fig.show()
'','fig = go.Figure(data=[go.Surface(z=Zs, x=Xs, y=Ys)])
102 fig.show()''
#fig=go.Figure(data=marker_data)
105 #fig.show()
'', 'fig, ax = plt.subplots(subplot_kw={"projection": "3d"})
surf = ax.plot_surface(M,beta,
plt.show()'''
```

Appendix B Problem 3 MATLAB Code

```
1 %% Compressible Flow - AEE 553
2 % Homework 5 - Problem 3
3 % Evan Burke
4 % 28 October 2022
6 clear; close; clc;
8 % Givens
9 \text{ th2} = 20; \text{ th3} = -15; \% \text{ deg}
10 M1 = 3; p1 = 1;
gamma = 1.4;
13 % Region 2 and 3 oblique shock solution
beta_solver(gamma,M1,th2);
b3 = beta_solver(gamma, M1, th3);
17 \text{ M1n2} = \text{M1} * \text{sind(b2)};
18 M1n3 = M1 * sind(b3);
20 \text{ M2n} = ((M1n2^2+2/(gamma-1))) / (2*gamma/(gamma-1)*M1n2^2-1))^0.5;
21 \text{ M3n} = ((M1n3^2+2/(gamma-1))) / (2*gamma/(gamma-1)*M1n3^2-1))^0.5;
M2 = M2n/sind(b2-th2);
M3 = M3n/sind(b3-th3);
p2 = p1*(1 + 2*gamma/(gamma+1)*(M1n2^2-1));
p3 = p1*(1 + 2*gamma/(gamma+1)*(M1n3^2-1));
29 % Check for sign change in function
30 for i=1:20 % know that p4 is greater than 1, 20 seems high enough to find
      one sign change
      [b4,b4p,th4,th4p,diff] = shock_interaction(i,gamma,M2,M3,p2,p3,th2*pi
31
      /180, th3*pi/180);
      x0(i) = i;
32
      b4s(i) = b4;
33
      b4ps(i) = b4p;
34
      th4s(i) = th4;
35
      th4ps(i) = th4p;
36
      diffs(i) = diff;
37
      if diffs(i) < 0</pre>
39
           polarity(i) = -1;
40
      else
41
           polarity(i) = 1;
      end
43
44
      if i>1
```

```
if polarity(i-1) ~= polarity(i)
                                                   break
47
                                    end
48
                      end
49
50 end
51
52 figure
53 plot(x0, diffs, [1,20], [0,0], 'r')
54 grid
ss xlabel('P4 guess')
56 ylabel('Objective Function Value')
57 legend('Objective Function', 'Zero Value')
59 % Solver Initial Conditions
60 i = 2;
_{61} x(1) = x0(end-1); % last value before sign change
62 x(2) = x0(end); % final value of polarity check, opposite sign as x(end-1)
y(1) = diffs(end-1); % value associated with x(end-1)
y(2) = diffs(end); % value associated with x(2)
65 m(1) = 1; % initial slope needed for solution, arbitrary
67 % Bisection Method
68 while abs(diff) > 1e-5
                      [b4,b4p,th4,th4p,diff] = shock_interaction(x(i),gamma,M2,M3,p2,p3,th2*
                  pi/180, th3*pi/180);
                     y(i) = diff;
70
                     m = (y(i)-y(i-1)) / (x(i)-x(i-1));
71
                     x(i+1) = -y(i)/m + x(i);
72
                      i = i + 1;
74 end
76 fprintf('p4 = %f\n',x(i))
77 fprintf('b4 = %f\n',b4*180/pi)
78 fprintf('b4p = \%f\n',b4p*180/pi)
79 fprintf('th4 = %f\n',th4*180/pi)
so fprintf('th4p = %f\n', th4p*180/pi)
81
function [b4,b4p,th4,th4p,diff] = shock_interaction(p0,gamma,M2,M3,p2,p3,
                  th2,th3)
                      syms b4 b4p th4 th4p p4 % declare symbolic vars
83
                      % currently accepts and outputs radians instead of degrees
84
                     eq_1 = p4/p3 == 1 + 2*gamma/(gamma+1) * ((M3*sin(b4))^2-1); % Oblique
                   shock eqn p2/p1 across OS
                      eq_2 = p4/p2 == 1 + 2*gamma/(gamma+1) * ((M2*sin(b4p))^2-1); % Oblique
                      shock eqn, p2/p1 across OS
                      eq_3 = tan(th4) == 2*cot(b4) * (M3^2*sin(b4)^2-1) / (M3^2 *(gamma + base)) / (M3^2 *(gamma + b
87
                   cos(2*b4)) + 2); % theta-beta-Mach relation
                      eq_4 = tan(th4p) == 2*cot(b4p) * (M2^2*sin(b4p)^2-1) / (M2^2 *(gamma + a)^2 +
                      cos(2*b4p)) + 2); % theta-beta-Mach relation
                      eq_5 = solve(eq_1,b4); % solve for b4
```

```
eq_6 = solve(eq_2,b4p); % solve for b4'
      eq_7 = solve(eq_3, th4); % solve for th4
91
      eq_8 = solve(eq_4,th4p); % solve for th4'
92
93
      b4i = subs(eq_5,p4,p0); % Placeholder value for beta4, extracting from
94
       syms
      b4 = double(abs(b4i(1))); % Converting beta4 val to double, taking
95
      positive, forming array
      b4pi = subs(eq_6,p4,p0); % Placeholder value for beta4'
96
      b4p = double(-abs(b4pi(1))); % Converting beta4' val to double, taking
      negative, forming array
      th4i = subs(eq_7,b4); % Placeholder value, theta4
98
      th4 = double(th4i); % Val to double, into array
99
      th4pi = subs(eq_8,b4p); % Placeholder value, theta4;
      th4p = double(th4pi); % Val to double, into array
      diff = th4 - th4p + th3 - th2; % 'Objective function', want 0 per
      constraints
  end
103
104
  function [beta] = beta_solver(gamma, M, theta)
      % accepts and returns degrees
106
      delta=1;
107
      theta=theta*pi/180;
108
      lamb = ((M^2-1).^2 - 3*(1 + (gamma-1)/2*M^2) * (1 + (gamma+1)/2*M.^2)
109
      * tan(theta)^2)^0.5;
      chi = ((M^2-1)^3 - 9 * (1 + (gamma-1)/2 * M^2) * (1 + (gamma-1)/2 * M^2)
110
      ^2 + (gamma+1)/4*M^4).*tan(theta)^2)/lamb^3;
      tan_beta = (M^2 - 1 + 2*lamb*cos((4*pi*delta+acos(chi))/3)) / (3 * (1)
111
     + (gamma-1)/2*M^2).*tan(theta));
      beta = atan(tan_beta)*180/pi;
112
113 end
```