

TTK4190 Guidance and Control of **B**oats

Assignment 2, Pt. 3

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## Problem 1 - Propeller Revolution and Speed Control

a)

The propeller thrust and torque coefficients  $K_T$  and  $K_Q$  can be computed by using the `wageningen` function found in the MSS toolbox.

```
1 Ja          = 0;      % advance number
2 PD          = 1.5;    % pitch/diameter ratio
3 AEAO        = 0.65;   % blade area ratio
4 z           = 4;      % number of propeller blades
5 [KT,KQ]     = wageningen(Ja,PD,AEAO,z);
```

b)/c)

$\tau = 0$  and looking at the figure  $Y = \frac{Q_d}{K_m}$  Converting equation to time domain

$$\begin{aligned}\frac{Q_m}{Y} &= \frac{K_m}{T_m s + 1} e^{-\tau s} \\ \mathcal{L}^{-1}(Q_m T_m s + Q_m = Q_d) \\ \dot{Q}_m T_m + Q_m &= Q_d \\ \dot{Q}_m &= -\frac{Q_m}{T_m} + \frac{Q_d}{T_m} \\ &= -\frac{1}{T_m} * Q_m + \frac{K_m}{T_m} * Y\end{aligned}$$

which is implemented in `ship.m` as follows

```
1 %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
2 % Part 3, 1b,1c) compute torque-based shaft dynamics here
3 Im      = 10^5;          % (kg m^2)
4 Km      = 0.6;           % (s^-1)
5 Tm      = 10;            % (s)
6 tau     = 0;             % time delay (s)
7
8 thr = rho * Dia^4 * KT * abs(n) * n; % thrust command (N)
9 Q = rho * Dia^5 * KQ * abs(n) * n; % torque command
10
11 persistent Qm
12 if isempty(Qm)
13     Qm = 0;
14 end
15
16 Qd = rho * Dia^5 * KQ * abs(n_c) * n_c;
17 Qm_dot = (Qd - Qm) / Tm;
18
19 Y = Qd / Km; % control input
20
21 A = -1 / Tm; % transfer function to state space
22 B = Km / Tm;
23
24 Qm_dot = A * Qm + B * Y;
25
26 Qm = euler2(Qm_dot, Qm, h);
27
28 n_dot = (1 / Im) * (Qm - Q);
29 %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
```

Thus, the formula for the shaft speed dynamics  $\dot{n}$  is found.

d)

The linear surge equation for the surge subsystem from [1] can be expressed as

$$U = \frac{(t-1)T}{X_u}$$

through this derivation:

$$\begin{aligned} (m - X_{\dot{u}})\dot{u}_r - X_u u_r &= -X_{\delta\delta}\delta^2 + (1-t)T \\ \dot{u}_r &= \delta = 0 \\ \downarrow \\ -X_u u_r &= (1-t)T \\ u_r &= U \\ \downarrow \\ U &= \frac{(t-1)T}{X_u} \end{aligned}$$

The assumptions are that the Boat must not accelerate and the rudder angle must be zero ( $\dot{u}_r = 0$ ,  $\delta = 0$ ).

e)

The desired thrust  $T_d$  can be computed from d) as

$$T_d = \frac{U_{ref} X_u}{(t-1)} \quad (1)$$

and propeller revolution  $n_d$  as

$$n_d = \text{sgn}(T_d) \sqrt{\frac{|T_d|}{\rho D^4 K_T(0)}} \quad (2)$$

```

1 function n_c = open_loop_speed_control(U_ref)
2 % Ship variables
3 m = 17.0677e6; % mass (kg)
4 X_u_dot = -8.9830e5; % added mass in surge
5 T1 = 20; % linear damping time constant
6 X_u = -(m-X_u_dot)/T1; % linear damping in surge
7
8 % Propeller variables
9 Dia = 3.3; % propeller diameter (m)
10 rho = 1025; % density of water (m/s^3)
11 Ja = 0; % advance number
12 PD = 1.5; % pitch/diameter ratio
13 AEAO = 0.65; % blade area ratio
14 z = 4; % number of propeller blades
15 [KT, tau] = wageningen(Ja, PD, AEAO, z);
16
17 t = 0.05; % thrust deduction number p. 164 in Fossen
18
19 % Combining the equation in task d) and inserting T (eq. 9.7 in Fossen)
20 n_c_squared = U_ref * X_u / ((t-1) * rho * Dia^4 * KT);
21 n_c = sign(n_c_squared) * sqrt(abs(n_c_squared));
22 end

```

The Boat does not reach the desired velocity when maintaining a constant heading angle, as we are not utilizing closed-loop control to regulate speed.

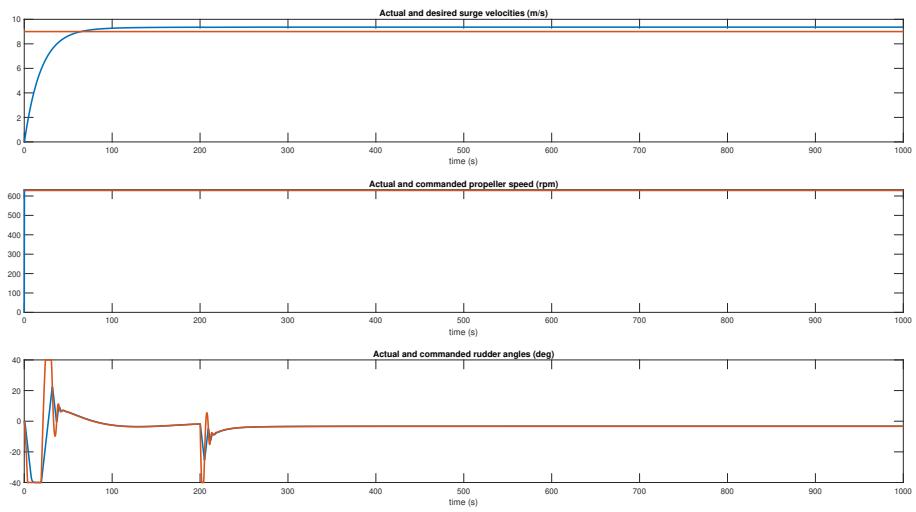


Figure 1: The result of open-loop speed control

## References

- [1] T. Fossen, *Handbook of Marine Craft Hydrodynamics and Motion Control*. John Wiley & Sons, 2011.