

Robust and LPV Control

- Robust controller design
- Parameter Elimination and Dualization
- LPV controller synthesis with multipliers
- An illustrative missile example

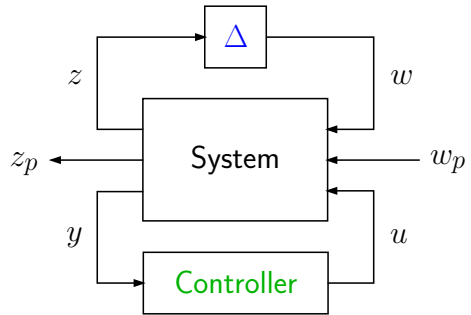
Configuration for Robust Controller Synthesis

Design **controller** guaranteeing:

- robust stability
- robustly desired performance specification on $w_p \rightarrow z_p$.

Consider following approach:

- Use robust performance characterization with **multipliers**
- Try to satisfy the multiplier characterization with suitable controller



Just for notational simplicity concentrate on **robust stabilization**.
Consider time-varying parametric uncertainty and quadratic stability.

System Descriptions

Uncontrolled LTI part:

$$\begin{aligned}\dot{x} &= Ax + B_1w + Bu \\ z &= C_1x + D_1w + Eu \\ y &= Cx + Fw\end{aligned}$$

w : uncertainty input

z : uncertainty output

u : control input

y : measured output

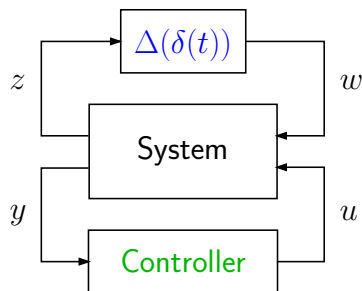
Controller:

$$\begin{aligned}\dot{x}_c &= A_c x_c + B_c y \\ u &= C_c x_c + D_c y\end{aligned}$$

Controlled LTI part:

$$\begin{aligned}\dot{\xi} &= \mathcal{A}\xi + \mathcal{B}w \\ z &= \mathcal{C}\xi + \mathcal{D}w\end{aligned}$$

Uncertainty: $w(t) = \Delta(\delta(t))z(t)$.



Robust Stability Analysis Inequalities

Assume $\delta(t) \in \mathcal{D} = \text{co}\{\delta^1, \dots, \delta^N\}$ (polytope) containing zero.

Robust stability guaranteed if exist \mathcal{X} and Q, R, S with

$$Q \prec 0, \quad \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix}^T \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix} \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix} \succ 0, \quad k = 1, \dots, N$$

$$\mathcal{X} \succ 0, \quad \begin{pmatrix} I & 0 \\ \mathcal{X}A & \mathcal{X}B \\ 0 & I \\ C & D \end{pmatrix}^T \left(\begin{array}{cc|cc} 0 & I & 0 & 0 \\ I & 0 & 0 & 0 \\ \hline 0 & 0 & Q & S \\ 0 & 0 & S^T & R \end{array} \right) \begin{pmatrix} I & 0 \\ \mathcal{X}A & \mathcal{X}B \\ 0 & I \\ C & D \end{pmatrix} \prec 0.$$

Apply standard procedure to step from analysis to synthesis.

Robust Synthesis Inequalities

Exists controller guaranteeing robust stability if exist v , Q , R , S :

$$Q \prec 0, \quad \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix}^T \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix} \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix} \succ 0, \quad k = 1, \dots, N$$
$$X(v) \succ 0, \quad \begin{pmatrix} I & 0 \\ A(v) & B(v) \\ 0 & I \\ C(v) & D(v) \end{pmatrix}^T \left(\begin{array}{cc|cc} 0 & I & 0 & 0 \\ I & 0 & 0 & 0 \\ \hline 0 & 0 & Q & S \\ 0 & 0 & S^T & R \end{array} \right) \begin{pmatrix} I & 0 \\ A(v) & B(v) \\ 0 & I \\ C(v) & D(v) \end{pmatrix} \prec 0.$$

Unfortunately **not convex** in all variables v and Q , R , S !

No technique known how to convexify in general!

Dualization Lemma

Suppose that $R \succ 0$ and $Q - S^T R S \prec 0$. Then

$$\begin{pmatrix} I \\ W \end{pmatrix}^T \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix} \begin{pmatrix} I \\ W \end{pmatrix} \prec 0$$

is equivalent to

$$\begin{pmatrix} W^T \\ -I \end{pmatrix}^T \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix}^{-1} \begin{pmatrix} W^T \\ -I \end{pmatrix} \succ 0.$$

Note that $\text{im} \begin{pmatrix} W^T \\ -I \end{pmatrix}$ equals orthogonal complement of $\text{im} \begin{pmatrix} I \\ W \end{pmatrix}$.

In general: Let $P = P^*$ be nonsingular with k negative eigenvalues. If the subspace \mathcal{S} with dimension k is P -negative then \mathcal{S}^\perp is P -positive.

Dual Robust Synthesis Inequalities

Exists controller guaranteeing robust stability if exist v , \tilde{Q} , \tilde{R} , \tilde{S} :

$$\tilde{Q} \succ 0, \quad \begin{pmatrix} -I \\ \Delta(\delta^k)^T \end{pmatrix}^T \begin{pmatrix} \tilde{Q} & \tilde{S} \\ \tilde{S}^T & \tilde{R} \end{pmatrix} \begin{pmatrix} -I \\ \Delta(\delta^k)^T \end{pmatrix} \prec 0, \quad k = 1, \dots, N$$

$$\mathbf{X}(v) \succ 0, \quad \begin{pmatrix} \mathbf{A}(v)^T & \mathbf{C}(v)^T \\ -I & 0 \\ \mathbf{B}(v)^T & \mathbf{D}(v)^T \\ 0 & -I \end{pmatrix}^T \left(\begin{array}{cc|cc} 0 & I & 0 & 0 \\ I & 0 & 0 & 0 \\ \hline 0 & 0 & \tilde{Q} & \tilde{S} \\ 0 & 0 & \tilde{S}^T & \tilde{R} \end{array} \right) \begin{pmatrix} \mathbf{A}(v)^T & \mathbf{C}(v)^T \\ -I & 0 \\ \mathbf{B}(v)^T & \mathbf{D}(v)^T \\ 0 & -I \end{pmatrix} \succ 0.$$

Note that multipliers are related as $\begin{pmatrix} \tilde{Q} & \tilde{S} \\ \tilde{S}^T & \tilde{R} \end{pmatrix} = \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix}^{-1}$.

No progress in general. However it helps for **state-feedback synthesis**

Static State-Feedback Synthesis - Lucky Case!

Recall block substitution:

$$\begin{pmatrix} A(v) & B(v) \\ C(v) & D(v) \end{pmatrix} = \begin{pmatrix} AY + BM & B_1 \\ C_1Y + EM & D_1 \end{pmatrix}.$$

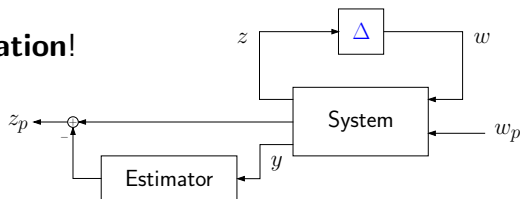
Last column does not depend on v ...

... dual inequalities are **affine** in all variables ...

... robust state-feedback synthesis possible with LMI's!

Similar results for **robust estimation**!

Very good exercise.



Elimination of Transformed Controller Parameters

Unstructured matrix variables in **one** LMI can often be eliminated.

For example let us recall the particular structure

$$\begin{pmatrix} \mathbf{A}(v) & \mathbf{B}(v) \\ \mathbf{C}(v) & \mathbf{D}(v) \end{pmatrix} = \left(\begin{array}{cc|c} AY & A & B_1 \\ 0 & XA & XB_1 \\ \hline C_1Y & C_1 & D_1 \end{array} \right) + \begin{pmatrix} 0 & B \\ I & 0 \\ 0 & E \end{pmatrix} \begin{pmatrix} K & L \\ M & N \end{pmatrix} \begin{pmatrix} I & 0 & 0 \\ 0 & C & F \end{pmatrix}.$$

Can eliminate $\begin{pmatrix} K & L \\ M & N \end{pmatrix}$ in synthesis inequalities. How?

Elimination Lemma

Consider the following quadratic matrix inequality in Z :

$$\begin{pmatrix} I \\ U^T Z V + W \end{pmatrix}^T \underbrace{\begin{pmatrix} Q & S \\ S^T & R \end{pmatrix}}_P \begin{pmatrix} I \\ U^T Z V + W \end{pmatrix} \prec 0$$

Let U_\perp, V_\perp be basis matrices of $\ker(U), \ker(V)$.

Suppose that $R \succ 0$ and $Q - S^T R S \prec 0$. Then the quadratic matrix inequality has a solution Z iff

$$V_\perp^T \begin{pmatrix} I \\ W \end{pmatrix}^T P \begin{pmatrix} I \\ W \end{pmatrix} V_\perp \prec 0 \quad \& \quad U_\perp^T \begin{pmatrix} W^T \\ -I \end{pmatrix}^T P^{-1} \begin{pmatrix} W^T \\ -I \end{pmatrix} U_\perp \succ 0.$$

Can explicitly construct solution Z if solvability conditions satisfied.

Application to Quadratic Performance Synthesis

Exists controller that renders \mathcal{A} Hurwitz and the QP spec for

$$P_p = \begin{pmatrix} Q_p & S_p \\ S_p^T & R_p \end{pmatrix} \quad \text{with } R_p \succcurlyeq 0$$

satisfied iff there exists \mathbf{v} with

$$\mathbf{X}(\mathbf{v}) \succ 0, \quad \left(\begin{array}{cc} I & 0 \\ 0 & I \\ \hline \mathbf{A}(\mathbf{v}) & \mathbf{B}(\mathbf{v}) \\ \mathbf{C}(\mathbf{v}) & \mathbf{D}(\mathbf{v}) \end{array} \right)^T \left(\begin{array}{cc|cc} 0 & 0 & I & 0 \\ 0 & Q_p & 0 & S_p \\ \hline I & 0 & 0 & 0 \\ 0 & S_p^T & 0 & R_p \end{array} \right) \left(\begin{array}{cc} I & 0 \\ 0 & I \\ \hline \mathbf{A}(\mathbf{v}) & \mathbf{B}(\mathbf{v}) \\ \mathbf{C}(\mathbf{v}) & \mathbf{D}(\mathbf{v}) \end{array} \right) \prec 0.$$

Suppose P is non-singular and partition $\tilde{P} := P^{-1}$ as P .

Let Φ, Ψ be basis matrices of $\ker \begin{pmatrix} B^T & E^T \end{pmatrix}, \ker \begin{pmatrix} C & F \end{pmatrix}$.

QP Synthesis Inequalities after Elimination

$$\begin{aligned}
 & \begin{pmatrix} \textcolor{red}{Y} & I \\ I & \textcolor{red}{X} \end{pmatrix} \succ 0, \\
 & \Psi^T \left(\begin{array}{cc} I & 0 \\ 0 & I \\ \hline \textcolor{red}{X}A & \textcolor{red}{X}B_1 \\ C_1 & D_1 \end{array} \right)^T \left(\begin{array}{cc|cc} 0 & 0 & I & 0 \\ 0 & Q_p & 0 & S_p \\ \hline I & 0 & 0 & 0 \\ 0 & S_p^T & 0 & R_p \end{array} \right) \left(\begin{array}{cc} I & 0 \\ 0 & I \\ \hline \textcolor{red}{X}A & \textcolor{red}{X}B_1 \\ C_1 & D_1 \end{array} \right) \Psi \prec 0, \\
 & \Phi^T \left(\begin{array}{cc} \textcolor{red}{Y}A^T & \textcolor{red}{Y}C_1^T \\ B_1^T & D_1^T \\ \hline -I & 0 \\ 0 & -I \end{array} \right)^T \left(\begin{array}{cc|cc} 0 & 0 & I & 0 \\ 0 & \tilde{Q}_p & 0 & \tilde{S}_p \\ \hline I & 0 & 0 & 0 \\ 0 & \tilde{S}_p^T & 0 & \tilde{R}_p \end{array} \right) \left(\begin{array}{cc} \textcolor{red}{Y}A^T & \textcolor{red}{Y}C_1^T \\ B_1^T & D_1^T \\ \hline -I & 0 \\ 0 & -I \end{array} \right) \Phi \succ 0.
 \end{aligned}$$

Much fewer variables! Nice system theoretic interpretation!

Alternative Robust Synthesis Inequalities

Synthesis inequalities with **non-convex coupling** ($k = 1, \dots, N$):

$$\begin{pmatrix} \tilde{Q} & \tilde{S} \\ \tilde{S}^T & \tilde{R} \end{pmatrix} = \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix}^{-1}, \quad \begin{pmatrix} Y & I \\ I & X \end{pmatrix} \succ 0, \quad Q \prec 0, \quad \tilde{R} \succ 0$$

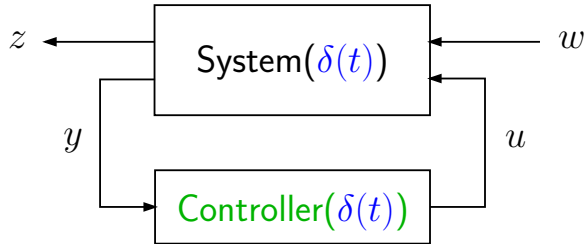
$$[*] \left(\begin{array}{cc|cc} 0 & X & 0 & 0 \\ X & 0 & 0 & 0 \\ \hline 0 & 0 & Q & S \\ 0 & 0 & S^T & R \end{array} \right) \begin{pmatrix} I & 0 \\ A & B_1 \\ \hline 0 & I \\ C_1 & D_1 \end{pmatrix} \Psi \prec 0, \quad [*] \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix} \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix} \succ 0$$

$$[*] \left(\begin{array}{cc|cc} 0 & Y & 0 & 0 \\ Y & 0 & 0 & 0 \\ \hline 0 & 0 & \tilde{Q} & \tilde{S} \\ 0 & 0 & \tilde{S}^T & \tilde{R} \end{array} \right) \begin{pmatrix} A^T & C_1^T \\ -I & 0 \\ \hline B_1^T & D_1^T \\ 0 & -I \end{pmatrix} \Phi \succ 0, \quad [*] \begin{pmatrix} \tilde{Q} & \tilde{S} \\ \tilde{S}^T & \tilde{R} \end{pmatrix} \begin{pmatrix} -I \\ \Delta(\delta^k)^T \end{pmatrix} \prec 0$$

Main Points

- General procedure: Genuine multi-objective synthesis with Youla
- General difficulties in robust controller synthesis
 - Illustrated for specific uncertainty/multiplier class
 - Show specific information structure and remedy
- Useful Technical Lemmas:
 - Linearization, Dualization, Elimination

Gain-Scheduling Control for LPV Systems



Given a **parameter-dependent** system, design a **controller** that stabilizes and achieves optimal performance, with the extra advantage (in contrast to a robust controller) that it can take on-line measurements of the parameters as information into account.

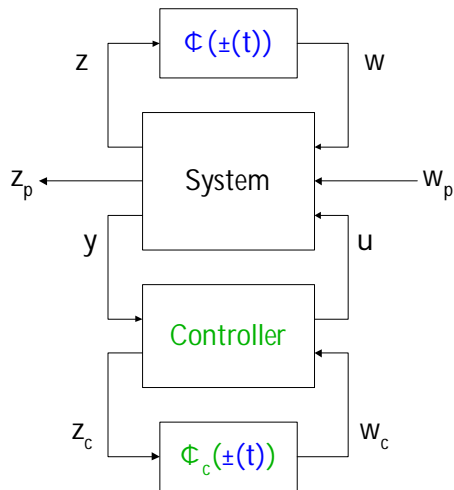
Example application: **Gain-scheduling**

Configuration for Multiplier LPV Synthesis

Design **parameter-dependent controller** guaranteeing

- exponential stability
- quadratic performance

If some parameters coincide with state-components procedure leads to **nonlinear controller!**



Again concentrate on **stability** - no performance channel.

System Descriptions

Uncontrolled LTI part:

$$\begin{aligned}\dot{x} &= Ax + B_1w + Bu \\ z &= C_1x + D_1w + Eu \\ y &= Cx + Fw\end{aligned}$$

w : uncertainty input

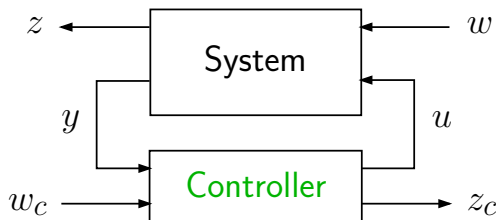
z : uncertainty output

u : control input

y : measured output

Controller LTI part:

$$\begin{aligned}\dot{x}_c &= A_c x_c + B_c \begin{pmatrix} y \\ w_c \end{pmatrix} \\ \begin{pmatrix} u \\ z_c \end{pmatrix} &= C_c x_c + D_c \begin{pmatrix} y \\ w_c \end{pmatrix}\end{aligned}$$



Fundamental Trick to Solve LPV Problem

The interconnection can be seen to result from

$$\begin{pmatrix} \dot{x} \\ z \\ z_c \\ y \\ w_c \end{pmatrix} = \begin{pmatrix} A & B_1 & 0 & B_2 & 0 \\ C_1 & D_1 & 0 & D_{12} & 0 \\ 0 & 0 & 0 & 0 & I \\ C_2 & D_{21} & 0 & D_2 & 0 \\ 0 & 0 & I & 0 & 0 \end{pmatrix} \begin{pmatrix} x \\ w \\ w_c \\ u \\ z_c \end{pmatrix}$$

controlled with

$$\dot{x}_c = A_c x_c + B_c \begin{pmatrix} y \\ w_c \end{pmatrix}, \quad \begin{pmatrix} u \\ z_c \end{pmatrix} = C_c x_c + D_c \begin{pmatrix} y \\ w_c \end{pmatrix}$$

Can solve the **robust control problem** for this interconnection.

LPV Synthesis: Step I

Synthesis inequalities **without non-convex coupling**: ($k = 1, \dots, N$):

$$\begin{pmatrix} Y & I \\ I & X \end{pmatrix} \succ 0, \quad Q \prec 0, \quad \tilde{R} \succ 0$$

$$[*] \left(\begin{array}{cc|cc} 0 & X & 0 & 0 \\ X & 0 & 0 & 0 \\ \hline 0 & 0 & Q & S \\ 0 & 0 & S^T & R \end{array} \right) \begin{pmatrix} I & 0 \\ A & B_1 \\ \hline 0 & I \\ C_1 & D_1 \end{pmatrix} \Psi \prec 0, \quad [*] \begin{pmatrix} Q & S \\ S^T & R \end{pmatrix} \begin{pmatrix} \Delta(\delta^k) \\ I \end{pmatrix} \succ 0$$

$$[*] \left(\begin{array}{cc|cc} 0 & Y & 0 & 0 \\ Y & 0 & 0 & 0 \\ \hline 0 & 0 & \tilde{Q} & \tilde{S} \\ 0 & 0 & \tilde{S}^T & \tilde{R} \end{array} \right) \begin{pmatrix} A^T & C_1^T \\ -I & 0 \\ \hline B_1^T & D_1^T \\ 0 & -I \end{pmatrix} \Phi \succ 0, \quad [*] \begin{pmatrix} \tilde{Q} & \tilde{S} \\ \tilde{S}^T & \tilde{R} \end{pmatrix} \begin{pmatrix} -I \\ \Delta(\delta^k)^T \end{pmatrix} \prec 0$$

LPV Synthesis: Step II

Find extension

$$\begin{pmatrix} Q_e & S_e \\ S_e^T & R_e \end{pmatrix} = \left(\begin{array}{cc|cc} \textcolor{red}{Q} & * & \textcolor{red}{S} & * \\ * & * & * & * \\ \hline \textcolor{red}{S}^T & * & \textcolor{red}{R} & * \\ * & * & * & * \end{array} \right) \quad \text{with inverse} \quad \left(\begin{array}{cc|cc} \textcolor{red}{\tilde{Q}} & * & \textcolor{red}{\tilde{S}} & * \\ * & * & * & * \\ \hline \textcolor{red}{\tilde{S}}^T & * & \textcolor{red}{\tilde{R}} & * \\ * & * & * & * \end{array} \right)$$

such that

$$\begin{pmatrix} \textcolor{red}{Q} & * \\ * & * \end{pmatrix} \prec 0 \quad \text{and} \quad \begin{pmatrix} \textcolor{red}{\tilde{R}} & * \\ * & * \end{pmatrix} \succ 0.$$

This is **always** possible.

Size of extension determines size of scheduling function $\Delta_c(\cdot)$.

LPV Synthesis: Step III

Find scheduling function $\Delta_c(\delta)$ such that for all $\delta \in \mathcal{D}$

$$\begin{pmatrix} \Delta(\delta) & 0 \\ 0 & \Delta_c(\delta) \\ \hline I & 0 \\ 0 & I \end{pmatrix}^T \left(\begin{array}{cc|cc} \tilde{Q} & * & \tilde{S} & * \\ * & * & * & * \\ \hline \tilde{S}^T & * & \tilde{R} & * \\ * & * & * & * \end{array} \right) \begin{pmatrix} \Delta(\delta) & 0 \\ 0 & \Delta_c(\delta) \\ \hline I & 0 \\ 0 & I \end{pmatrix} \preceq 0$$

or equivalently

$$\begin{pmatrix} -I & 0 \\ 0 & -I \\ \hline \Delta(\delta)^T & 0 \\ 0 & \Delta_c(\delta)^T \end{pmatrix}^T \left(\begin{array}{cc|cc} \tilde{Q} & * & \tilde{S} & * \\ * & * & * & * \\ \hline \tilde{S}^T & * & \tilde{R} & * \\ * & * & * & * \end{array} \right) \begin{pmatrix} -I & 0 \\ 0 & -I \\ \hline \Delta(\delta)^T & 0 \\ 0 & \Delta_c(\delta)^T \end{pmatrix} \preceq 0.$$

Have explicit formula for $\Delta_c(\delta)$.

LPV Synthesis: Step IV

Design LTI-part of controller

$$\dot{x}_c = A_c x_c + B_c \begin{pmatrix} y \\ w_c \end{pmatrix}, \quad \begin{pmatrix} u \\ z_c \end{pmatrix} = C_c x_c + D_c \begin{pmatrix} y \\ w_c \end{pmatrix}$$

as quadratic performance controller for extended system

$$\begin{pmatrix} \dot{x} \\ z \\ z_c \\ y \\ w_c \end{pmatrix} = \left(\begin{array}{c|ccc} A & B_1 & 0 & B_2 & 0 \\ \hline C_1 & D_1 & 0 & D_{12} & 0 \\ 0 & 0 & 0 & 0 & I \\ \hline C_2 & D_{21} & 0 & D_2 & 0 \\ 0 & 0 & I & 0 & 0 \end{array} \right) \begin{pmatrix} x \\ w \\ w_c \\ u \\ z_c \end{pmatrix}$$

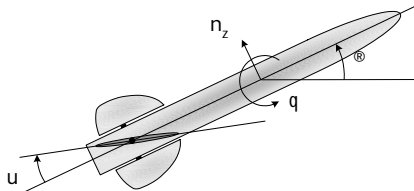
with performance index $\begin{pmatrix} Q_e & S_e \\ S_e^T & R_e \end{pmatrix}$.

Some Remarks on Sketched Procedure

- Obtained solution of LPV problem with full block scalings.
Can further enlarge scaling set. Requires behavioral controller [1].
- Comparison to direct implemented version in LMI-toolbox:
 - Allows rational parameter dependence
 - matrices $B(\delta)$, $E(\delta)$, $C(\delta)$, $F(\delta)$ allowed to depend on parameters
 - Care-free scheduling without determination of $\lambda_k(t)$.
- Exists extension to parameter dependent Lyapunov functions.

[1] C.W. Scherer, LPV control with full block multipliers, *Automatica* **37** (2001) 361-375.

High-Performance Aircraft System



u : Control input

α : Measurable parameter

n_z : Tracked output

Nonlinear system description with aerodynamic effects:

$$\dot{\alpha} = KM \left[(a_n \alpha^2 + b_n \alpha + c_n (2 - M/3)) \alpha + d_n u \right] + q$$

$$\dot{q} = M^2 \left[(a_m \alpha^2 + b_m \alpha - c_m (7 - 8M/3)) \alpha + d_m u \right]$$

$$n_z = M^2 \left[(a_n \alpha^2 + b_n \alpha + c_n (2 - M/3)) \alpha + d_n u \right]$$

Main Idea

Rewrite as linear parameter-varying system

$$\dot{\alpha} = K\delta_1 \left[(a_n\delta_2^2 + b_n\delta_2 + c_n(2 - \delta_1/3)) \alpha + d_n u \right] + q$$

$$\dot{q} = \delta_1^2 \left[(a_m\delta_2^2 + b_m\delta_2 - c_m(7 - 8\delta_1/3)) \alpha + d_m u \right]$$

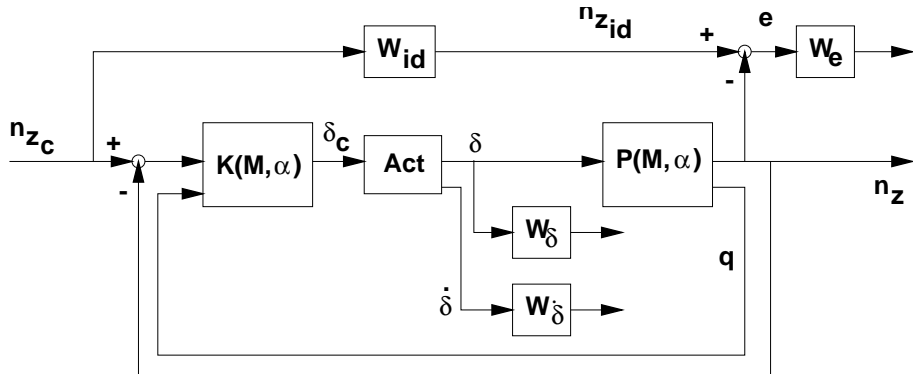
$$n_z = \delta_1^2 \left[(a_n\delta_2^2 + b_n\delta_2 + c_n(2 - \delta_1/3)) \alpha + d_n u \right]$$

with bounds $2 \leq \delta_1(t) \leq 4$ and $-20 \leq \delta_2(t) \leq 20$.

Design good controller scheduled with $\delta_1(t)$, $\delta_2(t)$

→ Is good controller for nonlinear system

Interconnection Structure



Model-Matching

Let controlled system approximately act like ideal model W_{id} .

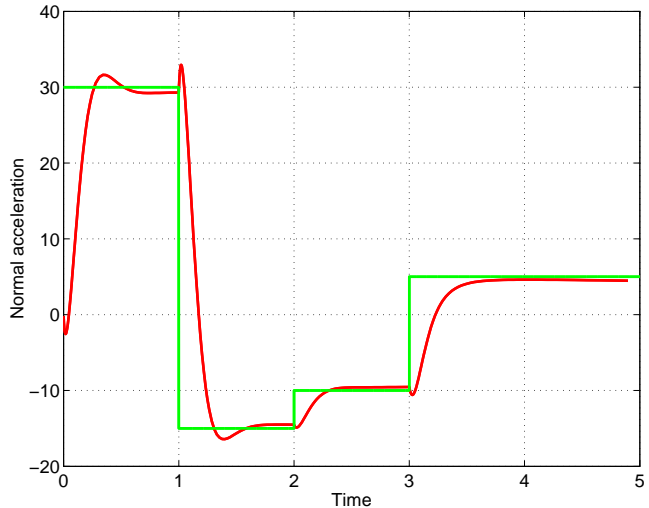
Synthesis with Convex Hull Relaxation

$M(t)$ decreases in 5 seconds from 4 to 2.

**Normal
acceleration**

Reference

Response



Some Book References

- [1] S.P. Boyd, G.H. Barratt, Linear Controller Design - Limits of Performance, Prentice-Hall, Englewood Cliffs, New Jersey (1991).
- [2] S.P. Boyd, L. El Ghaoui et al., Linear matrix inequalities in system and control theory, Philadelphia, SIAM (1994).
- [3] L. El Ghaoui, S.I. Niculescu, Eds., Advances in Linear Matrix Inequality Methods in Control, Philadelphia, SIAM (2000).
- [4] A. Ben-Tal, A. Nemirovski, Lectures on Modern Convex Optimization. Philadelphia, SIAM Publications (2001).
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- [6] Manuals for “LMI Control Toolbox” and “ μ Analysis and Synthesis Toolbox”
- [7] C.W. Scherer, S. Weiland, DISC Lecture Notes “LMI’s in Control”.