

# main

May 11, 2023

0.0.1 San Francisco State University: School of Engineering

0.0.2 ENGR 463: Final Exam Project (Spring 2023)

0.0.3 By Elon Goliger Mallimson

0.0.4 Professor: Dr. Douglas Couldron

0.0.5 Date: 5/10/2023

Part A: Variable Speed, Constant Load Test

Setup variables and dependencies as well as given data

```
[ ]: #dependencies
from pint import UnitRegistry
import numpy as np
import pandas as pd
import matplotlib.pyplot as plt

#unit setup
ureg = UnitRegistry()
m = ureg.meter
g = ureg.gram
kg = ureg.kilogram
s = ureg.second
min = ureg.minute
hr = ureg.hour
rpm = ureg.revolution / min
turns = ureg.turn
kPa = ureg.kilopascal
W = ureg.watt
N = ureg.newton
K = ureg.kelvin

#variable array setup
speed = ureg.Quantity(np.array([1500, 2000, 2500, 3000, 3500, 4000]), rpm)
torque = ureg.Quantity(np.array([1.8, 1.8, 1.8, 1.8, 1.8, 1.8]), N * m)
Qdot_shaft = ureg.Quantity(np.array([None, None, None, None, None, None]), W) #
↳Qdot_shaft
```

```

Qdot_in = ureg.Quantity(np.array([None, None, None, None, None, None]), W) #
    ↳heat input
efficiency = np.array([None, None, None, None, None, None]) # n_th percent
MEP = ureg.Quantity(np.array([None, None, None, None, None, None]), kPa) # mean
    ↳effective pressure
bsfc = ureg.Quantity(np.array([None, None, None, None, None, None]), g / (W *
    ↳hr)) # brake specific fuel consumption
Qdot_exhaust = ureg.Quantity(np.array([None, None, None, None, None, None]), W)
    ↳# exhaust heat
Qdot_fins = ureg.Quantity(np.array([None, None, None, None, None, None]), W) #
    ↳fins heat
mdot_air = ureg.Quantity(np.array([None, None, None, None, None, None]), kg/s)
    ↳# mass flow rate of air
mdot_fuel = ureg.Quantity(np.array([5.55E-05, 5.64E-05, 6.83E-05, 8.26E-05, 0.
    ↳000104, 0.000118]), kg / s) # mass flow rate of fuel
deltaT = ureg.Quantity(np.array([247, 247, 247, 291, 330, 350]), K) #
    ↳temperature difference

#given value
D = 10**-4 * m**3 # displacement of the engine: given 100cc
#constants
rho_air = 1.2 * ureg.kilogram / ureg.meter**3
Cp_air = 1.006 * ureg.kilojoule / (ureg.kilogram * ureg.kelvin)
LHV_gas = 45.2 * ureg.kilojoule / ureg.gram

```

### Calculations

```

[ ]: mdot_air = (rho_air * D * speed) / (2*turns) # The 2 is present in the
    ↳denominator because engine will only draw air every second revolution
mdot_air = mdot_air.to(kg / s)
Qdot_exhaust = mdot_air * Cp_air * deltaT
Qdot_in = mdot_fuel * LHV_gas
Qdot_in = Qdot_in.to(W)
Qdot_shaft = speed*torque
Qdot_shaft = Qdot_shaft.to(W)
Qdot_fins = Qdot_in - Qdot_exhaust - Qdot_shaft
efficiency = (Qdot_shaft / Qdot_in)*100 # n_th percent
bsfc = mdot_fuel / Qdot_shaft
bsfc = bsfc.to(g / (W * hr))
MEP = (4 * np.pi * torque) / D
MEP = MEP.to(kPa)

```

### Table A

```

[ ]: # Create a dictionary with the column names and data
data = {
    'Speed (RPM)': speed.magnitude,
    'Torque (N * m)': torque.magnitude,

```

```

'Power Shaft (W)': Qdot_shaft.magnitude,
'Power Input (W)': Qdot_in.magnitude,
'Efficiency (%)': efficiency.magnitude,
'MEP (kPa)': MEP.magnitude,
'BSFC (g/(W*h))': bsfc.magnitude,
'Exhaust Heat (W)': Qdot_exhaust.magnitude,
'Fins Heat (W)': Qdot_fins.magnitude,
'Air Mass Flow (kg/s)': mdot_air.magnitude,
'Fuel Mass Flow (kg/s)': mdot_fuel.magnitude,
'Delta T (delta_K)': deltaT.magnitude
}

# Create a DataFrame from the dictionary and add a caption
df = pd.DataFrame(data)
df

```

```

[ ]:
Speed (RPM)  Torque (N * m)  Power Shaft (W)  Power Input (W)
0           1500             1.8           282.743339       2508.60 \
1           2000             1.8           376.991118       2549.28
2           2500             1.8           471.238898       3087.16
3           3000             1.8           565.486678       3733.52
4           3500             1.8           659.734457       4700.80
5           4000             1.8           753.982237       5333.60

Efficiency (%)  MEP (kPa)  BSFC (g/(W*h))  Exhaust Heat (W)
0       11.270961  226.194671         0.706648         0.372723 \
1       14.788141  226.194671         0.538580         0.496964
2       15.264479  226.194671         0.521774         0.621205
3       15.146207  226.194671         0.525848         0.878238
4       14.034514  226.194671         0.567501         1.161930
5       14.136460  226.194671         0.563408         1.408400

Fins Heat (W)  Air Mass Flow (kg/s)  Fuel Mass Flow (kg/s)
0       1853.133661             0.0015         0.000056 \
1       1675.324882             0.0020         0.000056
2       1994.716102             0.0025         0.000068
3       2289.795322             0.0030         0.000083
4       2879.135543             0.0035         0.000104
5       3171.217763             0.0040         0.000118

Delta T (delta_K)
0           247
1           247
2           247
3           291
4           330
5           350

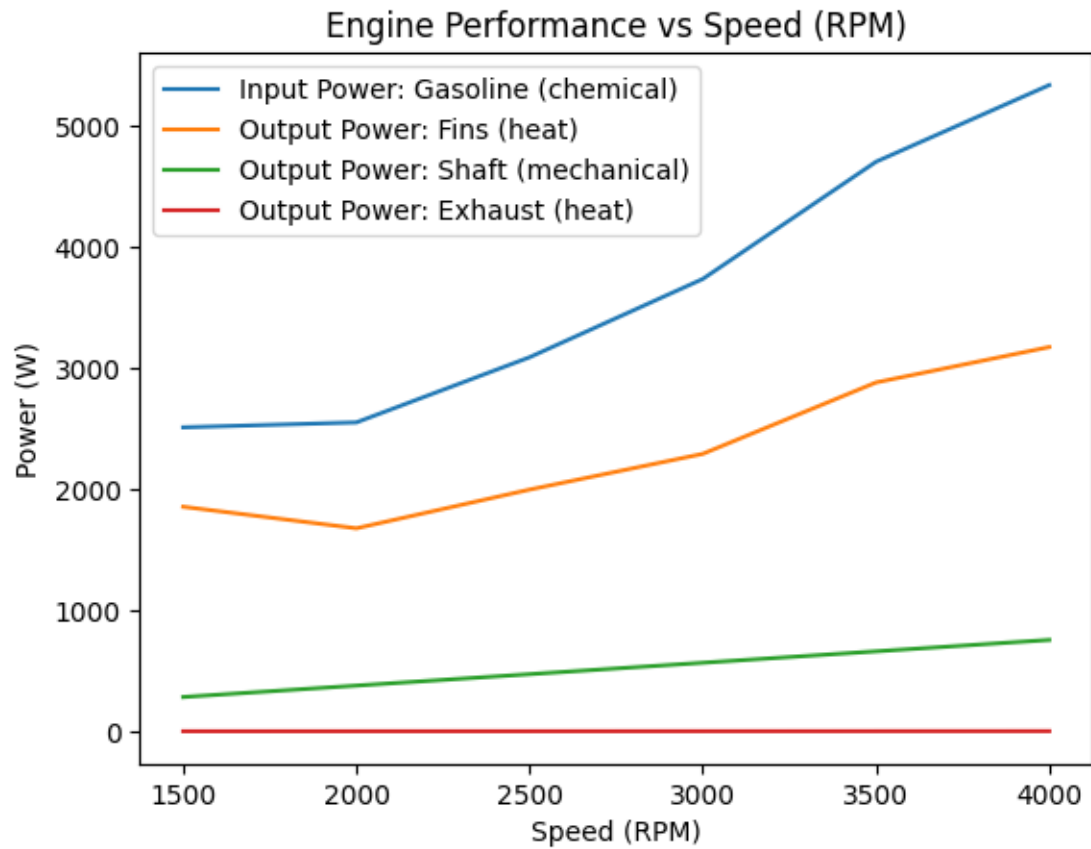
```

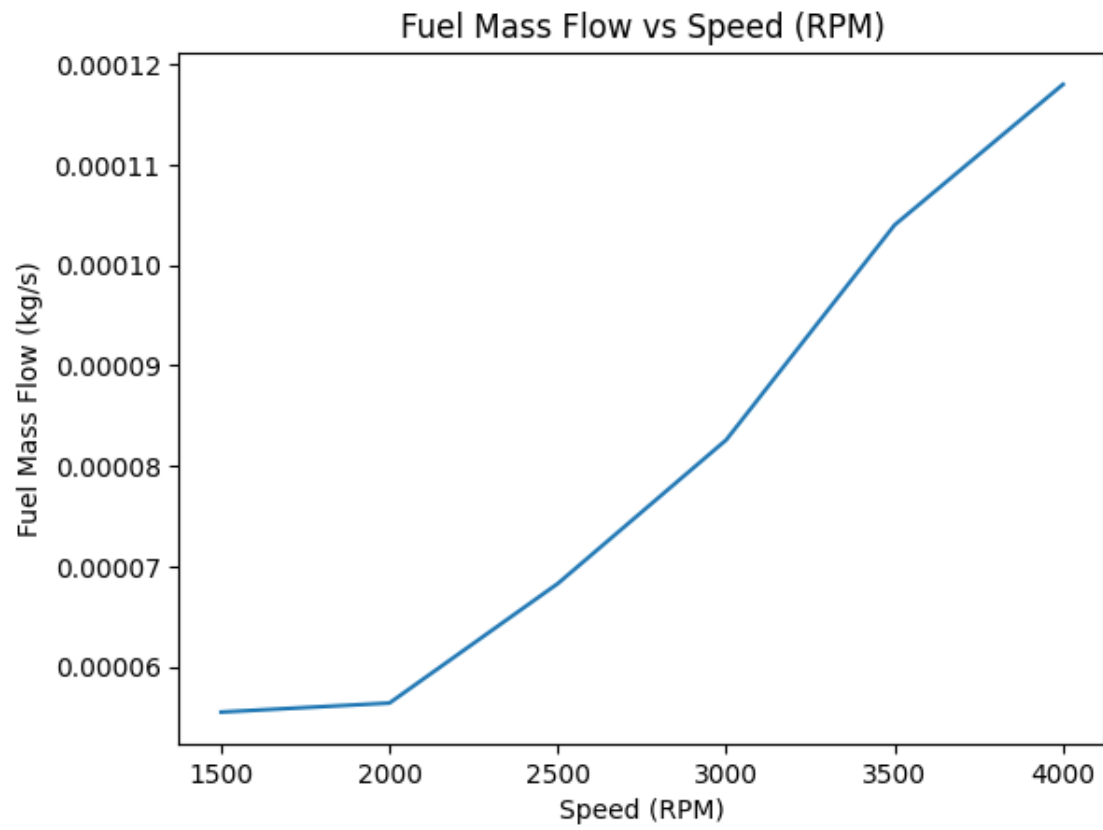
## Graphs

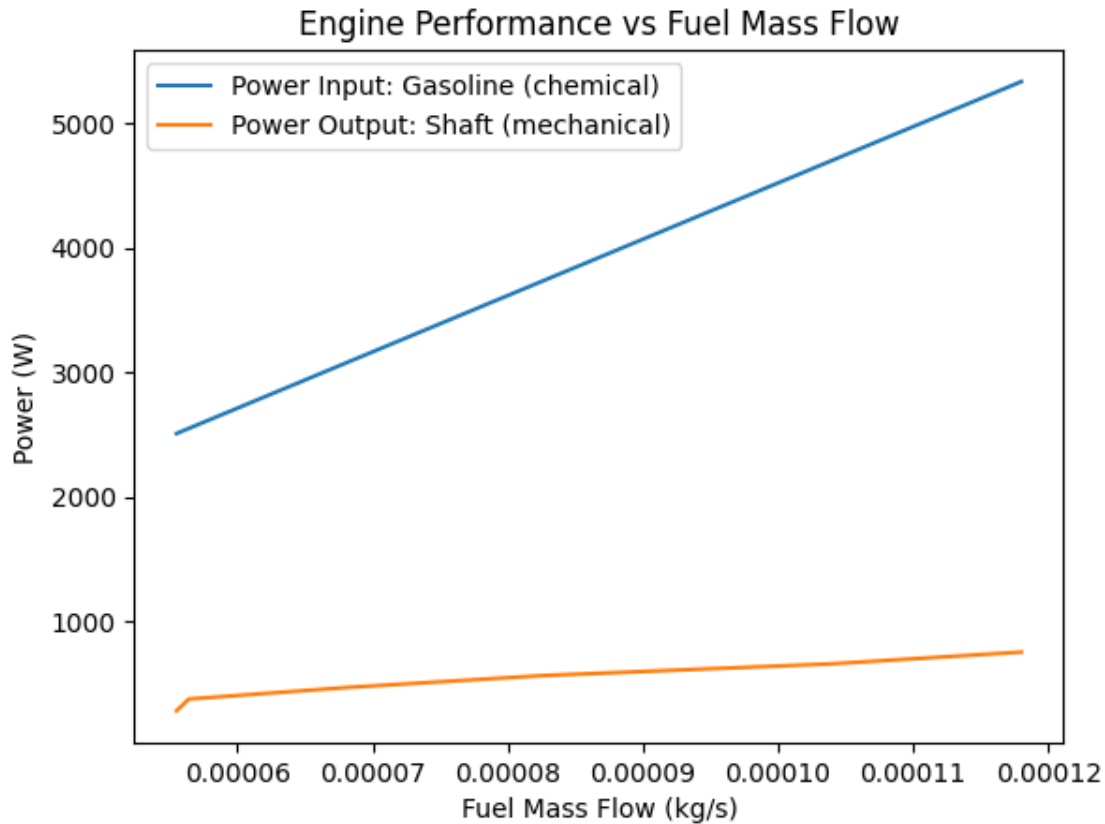
```
[ ]: # Plot Qdot_shaft, Qdot_in, Qdot_exhaust, Qdot_fins vs RPM
plt.figure()
plt.plot(df['Speed (RPM)'], df['Power Input (W)'], label="Input Power: Gasoline_
↳(chemical)")
plt.plot(df['Speed (RPM)'], df['Fins Heat (W)'], label="Output Power: Fins_
↳(heat)")
plt.plot(df['Speed (RPM)'], df['Power Shaft (W)'], label="Output Power: Shaft_
↳(mechanical)")
plt.plot(df['Speed (RPM)'], df['Exhaust Heat (W)'], label="Output Power:
↳Exhaust (heat)")
plt.xlabel('Speed (RPM)')
plt.ylabel('Power (W)')
plt.legend()
plt.title('Engine Performance vs Speed (RPM)')
plt.show()

# Plot fuel flow rate vs rpm
plt.figure()
plt.plot(df['Speed (RPM)'], df['Fuel Mass Flow (kg/s)'])
plt.xlabel('Speed (RPM)')
plt.ylabel('Fuel Mass Flow (kg/s)')
plt.title('Fuel Mass Flow vs Speed (RPM)')
plt.show()

# Plot Qdot_shaft, Qdot_in vs fuel flow rate
plt.figure()
plt.plot(df['Fuel Mass Flow (kg/s)'], df['Power Input (W)'], label="Power Input:
↳ Gasoline (chemical)")
plt.plot(df['Fuel Mass Flow (kg/s)'], df['Power Shaft (W)'], label="Power_
↳Output: Shaft (mechanical)")
plt.xlabel('Fuel Mass Flow (kg/s)')
plt.ylabel('Power (W)')
plt.legend()
plt.title('Engine Performance vs Fuel Mass Flow')
plt.show()
```







## Part B: Constant Speed (1500 RPM), Variable Load Test

Setup variables and dependencies as well as given data

```
[ ]: #variable array setup
percent_load = np.array([0.75, 1, 1.25, 1.5, 1.75, 2]) # percent
speed = ureg.Quantity(np.array([1500, 1500, 1500, 1500, 1500, 1500]), rpm) # rpm
torque = ureg.Quantity(np.array([1.35, 1.8, None, None, None, None]), N * m)
Qdot_shaft = ureg.Quantity(np.array([None, None, None, None, None, None]), W) # Qdot_shaft
Qdot_in = ureg.Quantity(np.array([None, None, None, None, None, None]), W) # Qdot_in
efficiency = np.array([None, None, None, None, None, None]) # n_th percent
MEP = ureg.Quantity(np.array([169.64, 226.19, None, None, None, None]), kPa) # MEP
bsfc = ureg.Quantity(np.array([None, None, None, None, None, None]), g / (W * hr)) # bsfc
Qdot_exhaust = ureg.Quantity(np.array([None, None, None, None, None, None]), W) # Qdot_exhaust
```

```

Qdot_fins = ureg.Quantity(np.array([None, None, None, None, None, None]), W) # fins heat
mdot_air = ureg.Quantity(np.array([None, None, None, None, None, None]), kg/s) # mass flow rate of air
mdot_fuel = ureg.Quantity(np.array([4.931E-05, 5.547E-05, 8.452E-05, 8.452E-05, 9.342E-05, 0.000118]), kg / s) # mass flow rate of fuel
deltaT = ureg.Quantity(np.array([235, 247, 250, 288, 314, 345]), K) # temperature difference

```

## Calculations

```

[ ]: # Calculate slope of load-torque line
slope = (torque[1].magnitude - torque[0].magnitude) / (percent_load[1] - percent_load[0])

# Calculate torque for remaining percent loads
for i in range(2, len(percent_load)):
    torque[i] = (slope * (percent_load[i] - percent_load[1]) + torque[1].magnitude) * N * m

mdot_air = (rho_air * D * speed) / (2*turns) # The 2 is present in the denominator because engine will only draw air every second revolution
mdot_air = mdot_air.to(kg / s)
Qdot_exhaust = mdot_air * Cp_air * deltaT
Qdot_in = mdot_fuel * LHV_gas
Qdot_in = Qdot_in.to(W)
Qdot_shaft = speed*torque
Qdot_shaft = Qdot_shaft.to(W)
Qdot_fins = Qdot_in - Qdot_exhaust - Qdot_shaft
efficiency = (Qdot_shaft / Qdot_in)*100 # n_th percent
bsfc = mdot_fuel / Qdot_shaft
bsfc = bsfc.to(g / (W * hr))
MEP = (4 * np.pi * torque) / D
MEP = MEP.to(kPa)

```

## Table B

```

[ ]: # Create a dictionary with the column names and data
data = {
    'Speed (RPM)': speed.magnitude,
    'Torque (N * m)': torque.magnitude,
    'Power Shaft (W)': Qdot_shaft.magnitude,
    'Power Input (W)': Qdot_in.magnitude,
    'Efficiency (%)': efficiency.magnitude,
    'MEP (kPa)': MEP.magnitude,
    'BSFC (g/(W*h))': bsfc.magnitude,
    'Exhaust Heat (W)': Qdot_exhaust.magnitude,
    'Fins Heat (W)': Qdot_fins.magnitude,
}

```



```

    'Air Mass Flow (kg/s)': mdot_air.magnitude,
    'Fuel Mass Flow (kg/s)': mdot_fuel.magnitude,
    'Delta T (delta_K)': deltaT.magnitude
}

# Create a DataFrame from the dictionary and add a caption
df = pd.DataFrame(data)
df

```

```

[ ]:   Speed (RPM) Torque (N * m) Power Shaft (W) Power Input (W) Efficiency (%) \
0      1500      1.35      212.057504      2228.812      9.514374 \
1      1500      1.8      282.743339      2507.244      11.277057
2      1500      2.25      353.429174      3820.304      9.251336
3      1500      2.7      424.115008      3820.304      11.101604
4      1500      3.15      494.800843      4222.584      11.717963
5      1500      3.6      565.486678      5333.600      10.602345

      MEP (kPa) BSFC (g/(W*h)) Exhaust Heat (W) Fins Heat (W) \
0  169.646003      0.837113      0.354615  1662.139496 \
1  226.194671      0.706266      0.372723  1851.777661
2  282.743339      0.860914      0.377250  3089.624826
3  339.292007      0.717428      0.434592  2961.596992
4  395.840674      0.679692      0.473826  3253.957157
5  452.389342      0.751211      0.520605  4247.508322

      Air Mass Flow (kg/s) Fuel Mass Flow (kg/s) Delta T (delta_K)
0          0.0015          0.000049          235
1          0.0015          0.000055          247
2          0.0015          0.000085          250
3          0.0015          0.000085          288
4          0.0015          0.000093          314
5          0.0015          0.000118          345

```

## Graphs

```

[ ]: # Plot Qdot_shaft, Qdot_in, Qdot_exhaust, Qdot_fins vs Torque
plt.figure()
plt.plot(df['Torque (N * m)'], df['Power Input (W)'], label="Input Power:␣
↳Gasoline (chemical)")
plt.plot(df['Torque (N * m)'], df['Fins Heat (W)'], label="Output Power: Fins␣
↳(heat)")
plt.plot(df['Torque (N * m)'], df['Power Shaft (W)'], label="Output Power:␣
↳Shaft (mechanical)")
plt.plot(df['Torque (N * m)'], df['Exhaust Heat (W)'], label="Output Power:␣
↳Exhaust (heat)")
plt.xlabel('Torque (N * m)')
plt.ylabel('Power (W)')
plt.legend()

```

```

plt.title('Engine Performance vs Speed (RPM)')
plt.show()

# Plot fuel flow rate vs rpm
plt.figure()
plt.plot(df['Torque (N * m)'], df['Fuel Mass Flow (kg/s)'])
plt.xlabel('Torque (N * m)')
plt.ylabel('Fuel Mass Flow (kg/s)')
plt.title('Fuel Mass Flow vs Torque (N * m)')
plt.show()

# Plot Qdot_shaft, Qdot_in vs fuel flow rate
plt.figure()
plt.plot(df['Fuel Mass Flow (kg/s)'], df['Torque (N * m)'], label="Torque")
plt.xlabel('Fuel Mass Flow (kg/s)')
plt.ylabel('Torque (N m)')
plt.legend()
plt.title('Torque vs Fuel Mass Flow')
plt.show()

```

