


CHAPTER 4 - AIRWORTHINESS LIMITATIONS SCHEDULE**CONTENTS — MAINTENANCE PROCEDURES**

| Paragraph Number | Title | Chapter/Section Number | Page Number |
|-----------------------------|---|-----------------------------------|------------------------|
| 4-1 | Airworthiness limitations schedule..... | 4-00-00 | 5 |
| 4-2 | Calculating flight hours on 204-011-102 yoke | 4-00-00 | 12 |

TABLES

| Table Number | Title | Page Number |
|-------------------------|--|------------------------|
| 4-1 | Mandatory airworthiness limitations schedule | 6 |

| REVISION NO. | DATE OF SIGNATURE | F.A.A. SIGNATURE |
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(TABLE I.D. 922277-t)

AIRWORTHINESS LIMITATIONS SCHEDULE

4-1. AIRWORTHINESS LIMITATIONS SCHEDULE.

The Mandatory Airworthiness Limitations Schedule summarizes, in tabular form, the maximum hours life of various components before mandatory retirement from service. Parts not listed on the schedule have unlimited airworthiness life.

Refer to United Aircraft of Canada Ltd. Service Bulletins, 5000 series, for power plant components Airworthiness Limitations.



AIRWORTHINESS LIFE FOR KIT COMPONENT AND/OR PARTS ARE NOT COVERED IN THIS AIRWORTHINESS SCHEDULE. REFER TO APPLICABLE SERVICE INSTRUCTIONS FOR MANDATORY AIRWORTHINESS SCHEDULE.

NOTE

The operating time specified for retirement for any given part number contained in this Airworthiness Limitations Schedule applies to all successive dash numbers for that item unless otherwise specified.

Airworthiness lives assigned to helicopter components and assemblies are based upon experience, testing, and engineering judgment and are subject to change at the sole discretion of Bell Helicopter Textron or an appropriate government agency.


















WARNING

ALL PARTS REMOVED DUE TO REACHING THEIR LIFE LIMITS ARE DEEMED UNAIRWORTHY AND SHALL BE PERMANENTLY MARKED AS SCRAP OR PHYSICALLY DESTROYED TO THE EXTENT THAT THERE IS NO CHANCE OF REPAIR OR REINSTALLATION ON ANY HELICOPTER OR COMPONENT.







NOTE

Neither assignment of a retirement life to a component, nor failure to assign a retirement life, constitutes a warranty of any kind. The only warranty applicable to the helicopter and any component is that warranty included in the Purchase Agreement for the helicopter or the component.


Mandatory airworthiness limitations schedule

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|--|---|---|
| MAIN ROTOR BLADE AND HUB ASSEMBLY | | |
| Blade | 204-012-001-023 | 1500 Hours |
| Blade | 204-012-001-031 | 4000 Hours |
| Blade | 204-012-001-033 | 1500/4000 Hours  |
| Blade | 212-015-501-005 and -115 | 4000 Hours |
| Retention Strap | 204-012-122-001, -005 and 204-310-101-101 | 1200 Hours or 2 years, whichever occurs first.  |
| Outboard Strap Fitting | 204-012-103-005 | 3600 Hours |
| Inboard Strap Fitting | 212-010-103-005 | 1200 Hours  |
| Inboard Strap Fitting | 212-010-103-007 | 2400 Hours |
| Strap Pin | 204-012-104-003 | 2400 Hours |
| Main Rotor Yoke | 204-011-102-(All) | 3600 Hours  |
| Main Rotor Yoke | 212-011-102-105 | 10,000 Hours  |
| Main Rotor Yoke | 212-011-102-109 | 6000 Hours  |
| Pillow Block | 204-011-108-113 | Conditional  |
| Pillow Block Bushing | 204-011-135-003 | 2400 Hours |
| Pillow Block Bushing | 204-011-135-105 | 3600 Hours |
| Main Rotor Mast | 204-011-450-007 and -105 | 15,000 Hours or RIN = 300000; whichever occurs first.    |
| Main Rotor Mast | 204-011-450-113 | 13,000 Hours or RIN=275000; whichever occurs first.   |
| Main Rotor Mast | 204-011-450-119 | 13,000 Hours or RIN=275000; whichever occurs first.   |
| Main Rotor Trunnion | 204-011-105-103 | 13,000 Hours or RIN=275000; whichever occurs first.   |

Mandatory airworthiness limitations schedule (Cont)

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|--------------------------------|---|--|
| Main Rotor Trunnion | 204-011-105-001 | 15,000 Hours or RIN = 300000; whichever occurs first.    |
| MAIN ROTORS CONTROLS | | |
| Pitch Horn | 204-011-120-005 | 3000 Hours |
| Swashplate Drive Link | 204-011-407-001 | 9000 Hours |
| Swashplate Outer Ring | 204-011-403-001 | 9000 Hours |
| Swashplate Support | 204-011-404-009 | 1000 Hours |
| Swashplate Support | 204-011-404-121 and -125 | 1000 Hours |
| Stabilizer Bar Centerframe | 204-011-307-001 and -105 | 10,000 Hours |
| Stabilizer Bar Tube | 204-011-328-001 | 2400 Hours or 3 years; whichever occurs first. |
| Stabilizer Bar Tube | 204-011-328-011 | 5000 Hours or 5 years; whichever occurs first. |
| Mixing Lever | 204-011-301-001 | 9000 Hours |
| Mixing Lever | 212-010-302-001 and -105 | 9000 Hours |
| Mixing Lever Pivot Bearing | MS27641-6  | 100 Hours |
| Gimbal Ring | 204-010-404-001 | 9000 Hours |
| Scissors Hub | 204-011-405-013 | 9000 Hours |
| Scissors Single Pivot Bearing | MS20201KP8A  | 100 Hours |
| Scissors Tube | 212-010-404-005 | 9000 Hours |
| Pitch Link | 204-011-127-001 and -003 | 9000 Hours |
| Collective Sleeve | 204-011-408-003, -105 and -107 | 9000 Hours |
| TAIL ROTOR AND CONTROLS | | |
| Yoke (Hog-out) | 212-010-704-001, -005 and -107 | 5000 Hours |
| Yoke (Forging) | 212-010-744-001, -005 and -107 | 5000 Hours |

Mandatory airworthiness limitations schedule (Cont)

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|-----------|---|--------------------|
| Yoke | 212-011-702-001 | 5000 Hours |
| Blade | 212-010-750-009 and -105 | 5000 Hours |


POWER TRAIN DRIVE SYSTEM COMPONENTS

| | | |
|--|-----------------|---|
| Spider | 204-040-785-003 | Conditional  |
| Spider | 412-040-785-101 | 2500  |
| Spider | 412-040-785-103 | Unlimited |
| Mast Bearing | 204-040-136-009 | 1000 Hours |
| Bearing (When used in rotor brake quill) | 204-040-424 | 600 Hours  |
| Tail Rotor Hanger Bearing | 204-040-623-001 | 100 Hours |
| Tail Rotor Hanger Bearing | 204-040-623-005 | 1000 Hours |
| Pinion — Offset Accessory Drive | 212-040-202-001 | 1000 Hours |





POWER PLANT RELATED SYSTEMS

| | | |
|--------------------|--------------------|-----------|
| Blower, Oil Cooler | 212AA3192 (Benson) | 300 Hours |
|--------------------|--------------------|-----------|

LANDING GEAR

| | | |
|------------|--------------------------------------|--|
| Crosstubes | 205-050-400-007, -029, -035 and -705 | 1000 Hours  |
|------------|--------------------------------------|--|







FLOAT LANDING GEAR (KIT P/N 205-706-050-001, -007, -011, -101)

| | | |
|--------------------|-----------------------|--|
| Crosstube Assembly | 205-050-114-001 | 500 Hours  |
| Crosstube Assembly | 205-050-114-011 | 1000 Hours  |
| Crosstube Assembly | 205-050-114-023, -025 | Unlimited |
| Crosstube Assembly | 205-706-050-005 | 500 Hours  |
| Crosstube Assembly | 205-706-050-013 | 1000 Hours  |
| Crosstube Assembly | 205-706-050-015 | Unlimited |


Mandatory airworthiness limitations schedule (Cont)


| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|---|---|--------------------|
| CONTROL SYSTEM BOLTS (KIT P/N 212-704-092-001)  | | |
| Swashplate Support to Collective Lever Pivot Bolts (2) | AN178-22A | 1000 Hours |
| Pitch Horn to Pitch Link (2) | 20-057-6-31D | 1000 Hours |
| Pitch Link to Universal (2) | 20-057-6-27D | 1000 Hours |
| Universal to Mixing Lever (2) | 20-057-6-34D | 1000 Hours |
| Mixing Lever to Scissors Tube (2) | 20-057-5-27D | 1000 Hours |
| Scissors Tube to Scissors (2) | 20-057-5-27D | 1000 Hours |
| Scissors (204-011-406) Pivot Bolt (2) | 20-057-8S90D or 20-057-8-86D | 1000 Hours |
| Scissors (212-010-407) Pivot Bolt (2) | 212-010-411-5 or -3 | 1000 Hours |
| Scissors to Drive Link (2) | 20-057-8S69D | 1000 Hours |
| Drive Link to Rotating Swashplate (2) | 20-057-5-30D | 1000 Hours |
| Fixed Swashplate to Right Cyclic Boost Tube (1) | 20-057-5-24D | 1000 Hours |
| Fixed Swashplate to Left Cyclic Boost Tube (1) | 20-057-5-24D | 1000 Hours |
| Collective Lever to Collective Boost Tube (1) | 20-057-5-24D | 1000 Hours |
| Hydraulic Cylinder Tube to Swashplate Universal (3) | 20-057-5-24D | 1000 Hours |
| Universal to Hydraulic Cylinder (3) | 20-057-5-24D | 1000 Hours |
| Hydraulic Cylinder to Lower Support (3) | 212-001-304-003 | 1000 Hours |
| Hydraulic Cylinder to Lower Support (3) | 212-001-323-001 | 2500 Hours |
| Gimbal to Inner Ring (2) | 204-011-463-001 | 1000 Hours |


Mandatory airworthiness limitations schedule (Cont)

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|---|---|--|
| Gimbal to Swashplate Support (2) | 204-011-463-003 | 1000 Hours |
| Stabilizer Bar Pivot Bolts (2) | 20-057-10S27D or 20-057-10S29D | 1000 Hours |
| Mixing Lever (204-011-301) Pivot Bolts (4) | 20-057-6S20D or 20-057-6S23D | 1000 Hours |
| Mixing Lever (212-010-302) Pivot Bolts (4) | 20-057-6S23D or 20-057-6S24D | 1000 Hours |
| MISCELLANEOUS | | |
| Battery (Blade Inspection System) | MN1604 or 522 | 500 Hours or 6 Months; whichever occurs first. |
| Cartridge, Fire Extinguisher | 209-062-908-13 | 6 Years  |
| Cartridge, Fire Extinguisher | 209-062-908-17 | 6 Years  |
| Cartridge, Fire Extinguisher | 209-062-908-15 | 4 Years In-Service Life (6 Years Total)  |
| Cartridge, Fire Extinguisher | 209-062-908-19 | 4 Years In-Service Life (6 Years Total)  |
| Cartridge, Fire Extinguisher | 209-062-908-109 | 6 Years  |

NOTES:


 All subsequent dash number changes have the same airworthiness life presented unless otherwise noted.

 Rotor blades with serial numbers not listed as follows have a 1500 hour airworthiness limitation schedule. Rotor blades listed as follows have a 4000 hour airworthiness limitation schedule: AMR-04017 through AMR-04047, AMR-04053 through AMR-04074, AMR-54001, AMR-54002, AMR-54005, AMR-54006, AMR-54008 through AMR-54073, AMR-54097, and AMR-54099 through AMR-54256.

 Repeat heavy lift operators must factor their flight time according to the following table for 212 main rotor yoke:














| Number of events/hr* | Factor |
|----------------------|--------|
| 1.0 — 5.0 | 1.0 |
| 5.1 — 8.0 | 1.5 |

Mandatory airworthiness limitations schedule (Cont)





| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|-----------|---|--------------------|
| | 8.1 — 12.0 | 2.0 |
| | 12.1 — 18.0 | 3.0 |
| | 18.1 — 32.0 | 5.0 |
| | 32.1 — 48.0 | 7.0 |
| | 48.1 — 62.0 | 9.0 |

*An event is a lift operation or takeoff.


*A logging lift counts as two events.

-  The airworthiness life for this bearing and any authorized replacement bearing is applicable if installed in 204-011-301 mixing lever.
-  The airworthiness life for this bearing and any authorized replacement bearing is applicable if installed in 204-011-406 scissors.
-  Inspect control system bolts each 24 months (Chapter 5).
-  Cartridge service/shelf life starts from date of manufacture.
-  All tension-torsion straps must be retired after 24 months calendar time in service. In this application, calendar time in service begins when new straps installed in main rotor hub and blade assembly are subjected to powered rotation.
-  Not authorized for use with Kit, 212-704-153-101.
-  Perform a dye penetrant inspection of pillow block every 2400 hours.
-  Perform magnetic particle inspection of spider every 3100 hours per ASB 212-91-66A.
-  Upon retirement of the 212-010-103-005 strap fitting, the -005 fitting shall be replaced with the 212-010-103-007 strap fitting.
-  Helicopters with T.B.212-91-138 incorporated shall not use stainless steel main rotor yoke assembly P/N 212-011-102-105.
-  Overhaul schedule for rotor brake quill is 2400 hours; however, the 204-040-424 bearing used in the 205-040-300 quill shall be replaced each 600 hours. The 222-342-420 bearing used in the 412-040-125 quill does not have a finite life.
-  Successive dash nos. (-103 and sub.) do not have a finite life.
-  Retirement Index Number (RIN) is the retirement life based on fatigue damage from normal helicopter lifts and takeoffs. New components will begin with an accumulated RIN of zero that will be increased as lifts and takeoffs are performed. Operators must record the number of lifts and takeoffs and increase the accumulated RIN accordingly.

Mandatory airworthiness limitations schedule (Cont)

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|---|--|--------------------|
| When the maximum RIN or retirement flight hours is reached, whichever occurs first, the component will be removed from service. | | |
|  | Increase RIN count by 5 for each takeoff/lift performed. If logging, increase RIN count by 10 for each takeoff/lift performed. | |
|  | Per ASB 212-80-18 and ASB 212-76-3. | |
|  | Per ASB 212-77-17. | |

4-2. CALCULATING FLIGHT HOURS ON 204-011-102 YOKE.

Calculate flight hours on the yoke using the table given in note  to determine the correct factor based on the number of events per flight hour:

1. If flight hours cannot be determined, use the following:

Enter on yoke Historical Service Record 900 hours per year from date of helicopter delivery or date yoke was installed.

2. If number of lift events per hour cannot be determined, use the following:

Enter on yoke Historical Service Record five

hours for each flight hour of external operation, or two hours for each flight hour of internal operation for which the number of events cannot be determined. Use five hours for each flight hour if time actually spent in external or internal operation cannot be determined.

3. Perform the following operations following calculation or approximation of flight hours/lift events:

If main rotor yoke flight hours exceed 3300, remove yoke from service within the next 300 hours.

Retire main rotor yoke at 3600 hours.