

$$J_{con} = \frac{1 + \cos \alpha}{2} (m \dot{u}_e + (p_e - p_a) A_e)$$

$$\approx \frac{1 + \cos \alpha}{2} \underbrace{(m \dot{u}_e + (p_e - p_a) A_e)}_{J_{1D}}$$

J_{1D}

Line in fig. 11.5 is $\frac{1 + \cos \alpha}{2} \cdot 0.995$

viscous losses

small losses compared to ideal

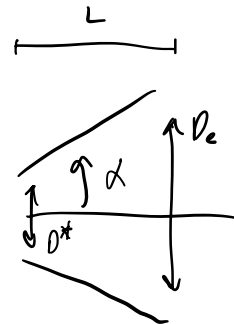
Temptation: make α very small

But: must keep $\frac{A_e}{A^*}$ necessary to achieve
desired $\left\{ \begin{array}{l} M_e \rightarrow \\ p_e/p_0 \rightarrow \frac{J}{p_0 A^*} \end{array} \right.$

Then we need a long (heavy) nozzle

For given $\left\{ \begin{array}{l} \text{throat } (A^*, D^*) \\ \text{exit } (A_e, D_e) \end{array} \right.$

Relate length to semi-angle α



$$\rightarrow \frac{A_e}{A^*} = \left(\frac{D_e}{D^*} \right)^2 = \left(\frac{D^* + 2L \tan \alpha}{D^*} \right)^2 \quad D_e = D^* + 2L \tan \alpha$$

$$\frac{L}{D^*} = \frac{\sqrt{A_e/A^*} - 1}{2 \tan \alpha}$$

$A_e/A^* = 100$
$\alpha = 30^\circ \rightarrow L/D^* = 7.8$
$\alpha = 15^\circ \rightarrow L/D^* = 16.8$

To mitigate thrust loss by making α smaller,
huge price in weight

Better approach: Shape the nozzle so as to have
"as purely axial as possible" velocity @ exit

Nozzle design criteria

- prevent formation of normal & oblique shocks inside nozzle
(2D Gas dynamics)
- Have purely axial velocity on as large a fraction of exit area as possible

Approaches:

Best "Bell" shape

- method of characteristics (applied math)
- calculus of variations (paper by Rao)

Good

- parabola (paper by Allman & Hoffman)

Poor

- conical nozzle (typical $\alpha = 15^\circ$)

common to all of these:

throat section is described by circular arcs

Improvement from conical to parabolic nozzle:

→ class figures

combustion chambers

→ present some sizing criteria

suppose T , P_0 , C_g , C^* , I_{sp} selected based on performance/requirements/past designs

Then A^* also determined → can be used as starting point of CC design

In general, CC's are surfaces of revolution
(pressure vessels) b/c they yield the minimum
thickness (weight) for an assigned pressure

convention: CC volume, V_c , is space between injector & plane
of throat area

Requirement: cc volume must be large (long) enough to allow complete
fuel/oxidizer

- atomization
- evaporation
- mixing
- reaction