

Real nozzles:

- Must assess how "bad"
our theory is

How much does the actual nozzle flow depart
from 1-D theory?

For fixed A_e/A^* , how much does the nozzle
shape affect thrust/performance?

Is combustion really at $p_0 = \text{const}$?

Is the composition of combustion products
(propellant) fixed or varying in nozzle?

Departure from 1-D theory

In perfect 1-D case, all quantities only depend on x

Only important variable is $\frac{A(x)}{A^*}$

Since that sets $M(x)$ which sets everything else.

In real flow

Quantities depend on at least 2 variables (e.g. $p = p(x, r)$)

$$\underline{u} = u_x \hat{i} + u_r \hat{e}_r$$

Nozzle shape is very important

Rules of thumb {
Shape of converging section of nozzle is not
very important
Shape of divergent " " is extremely
important
Try to avoid shockwaves

Conical nozzle

purpose: show effects from non 1-D physics

conical: simplest departure from 1-D

$p, \rho, T = \text{const}$ on any spherical surface

$|\underline{u}| = \text{" " " " " "}$

Direction of \underline{u} varies over " "

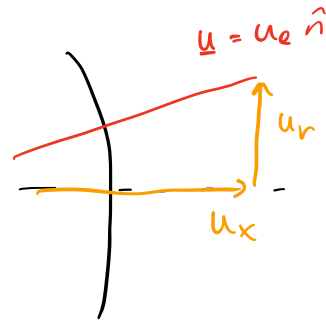
$$\underline{u} \perp A_e'$$

p_e acts $\perp A_e'$

physically: If \underline{u} has radial comp.

u_r @ exit, wasting momentum in dir.

\perp to motion of rocket



Force balance in x-direction

$$\text{In 1-D we had } J = \dot{m} u_e + (p_e - p_a) A_e$$

Here start from mom-eq

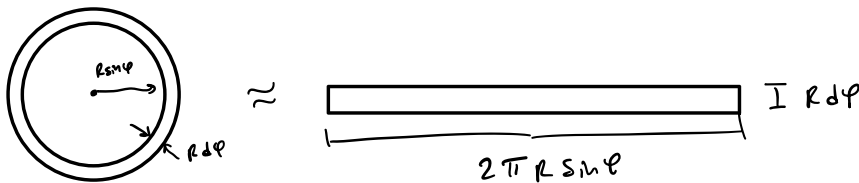
$$J = \int_{CS} u_x \rho \underline{u} \cdot d\underline{A} - \int_{CS} p d\underline{A} \cdot \hat{i} = \underbrace{\int_{CS} u_x \rho \underline{u} \cdot d\underline{A} \hat{n}}_1 - \underbrace{\int_{CS} p d\underline{A} \hat{n} \cdot \hat{i}}_2$$

About 2: only contribution from $p = p_a - p_e = \text{const}$ on A_e'

$$= - \int_{CS} p d\underline{A} \hat{n} \cdot \hat{i} = (p_e - p_a) A_e$$

\uparrow proj. of A_e' onto plane \perp to \hat{i}

About 1: Mass crosses CS on A_e'



$$dA = 2\pi R^2 \sin \phi d\phi$$

$$\underline{u} \cdot \hat{n} = u_e \hat{n} \cdot \hat{n} = u_e$$

$$u_x = u_e \cos \phi$$

$$\rightarrow J = \int_0^\alpha \underbrace{u_e \cos \phi}_{u_x} \underbrace{\rho_e u_e}_{\underline{u} \cdot \hat{n}} \underbrace{2\pi R^2 \sin \phi d\phi}_{dA} + (p_e - p_a) A_e$$

$$= \rho_e u_e^2 2\pi R^2 \int_0^\alpha \sin\varphi \cos\varphi d\varphi + (p_e - p_a) A_e$$

$$= - \frac{\cos^2\varphi}{2} \Big|_0^\alpha$$

$$\mathcal{J} = 2\pi R^2 \rho_e u_e^2 \cdot \left(\frac{1 - \cos^2\alpha}{2} \right) + (p_e - p_a) A_e$$

Now relate A_e' & A_e

$$\underbrace{A_e'}_{\text{on sphere}} = \int_0^\alpha dA = \int_0^\alpha 2\pi R^2 \sin\varphi d\varphi = -2\pi R^2 \cos\varphi \Big|_0^\alpha$$

$$= 2\pi R^2 (1 - \cos\alpha)$$

$$A_e = \pi r^2 \Big|_{\varphi=\alpha} = \pi R^2 \sin^2\alpha = \pi R^2 (1 - \cos^2\alpha)$$

$$= \pi R^2 (1 - \cos\alpha)(1 + \cos\alpha)$$

$$A_e = \frac{1 + \cos\alpha}{2} A_e'$$

Rewrite \mathcal{J}

$$\mathcal{J} = \underbrace{2\pi R^2 \rho_e u_e (1 - \cos\alpha) \left(\frac{1 + \cos\alpha}{2} \right)}_{\rho_e u_e A_e' = \dot{m}} u_e + (p_e - p_a) \left(\frac{1 + \cos\alpha}{2} \right) A_e'$$

$$\mathcal{J} = \frac{1 + \cos\alpha}{2} \left[\dot{m} u_e + (p_e - p_a) A_e' \right]$$

But since $\begin{cases} A_e \approx A_e' \\ (p_e - p_a) A_e' \ll \dot{m} u_e \end{cases}$

Can approximate

$$\mathcal{J} = \frac{1 + \cos\alpha}{2} \left[\underbrace{\dot{m} u_e + (p_e - p_a) A_e'}_{\mathcal{J}_{1-0}} \right]$$