XC User's Manual

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Chapter 1

Introduction

1.1 So... what's XC?

The "Finite Element Analysis (FEA)" or "Finite Element Method (FEM)" is a numerical method for solving problems of engineering and mathematical physics. This method has proven to be useful to solve problems with one of more of the following characteristics:

- complicated geometries
- complicated load patterns (in space and/or in time).
- complicated material properties.

XC is an open source FEA program designed to solve structural analysis problems. The program can solve various types of problems, from simple linear analysis to complex nonlinear simulations. It has a library of finite elements which allows to model various geometries, and multiple material models, allowing its application in various areas of structural analysis.

1.2 Motivation

Someone said that, when the French climber Lionel Terray was asked about his reason to climb a mountain, he simply said "because it's there".

Something similar happens with the development of this program. Since I began the study of the finite element method, after studying the analytic solutions to elastic problems (so limited), the scope of their employment to meet structural problems brought on me a great attraction. This, coupled with my love with computer science, made me decide to develop a finite element program that would be useful to calculate structures and that could be modified an expanded in any way the user wanted (a bit optimistic, yeah...). So, first I wrote a Pascal version of the program which could only work with bar-type elements. Afterwards, I wrote a C++ version "from scratch" that was never be able to solve any nontrivial problem. Finally I've discovered the possibilities offered by the calculation core of *Opensees* and I decided to modify it so it was able to be used in a "engineering office environment" (as opposed to academic use) so to speak.

To achieve this objective, the main modifications made to the original code were as follows:

1. Incorporation of some simple algorithms for generation of the finite element mesh. The modeler is able to create structured grids from the description of geometry by means of points, lines, surfaces and solids.

- 2. Generating graphics using VTK library. This is an open source library for generating graphics for scientific use.
- 3. At first I've used a macro language, built from the ground, which served to expose all the values from the model (displacements, internal forces, strains, stresses,...) in a way that allows the user to formulate a sentence like "get the ratio between the vertical displacement of the node closest to the center of the beam an the total span of the beam". In 2013 we stopped the development of that custom language and start the migration to Python which was finished in 2015.
- 4. Utilities for the construction and calculation of design load combinations prescribed by the building codes (EHE, ACI 318, EAE, Eurocodes ,...) so as to facilitate the verification of design requirements on each of them.
- 5. Ability to activate and deactivate elements to enable the analysis of structures built in phases, of geotechnical engineering problems and the strengthening of existing structures.
- 6. Writing macros to verify the structure and its elements according the criteria prescribed by building codes (axial and bending capacity, shear reinforcement,...).
- 7. Changing the code so it's linked with "standard" linear algebra libraries (BLAS, Arpack, LAPACK, SuperLU ,...). This eliminates the need to include in the program "ad-hoc" versions of these libraries.
- 8. Modification of the material models so that support prescribed strains. That makes possible to solve problems involving thermal and rheological actions.
- 9. Implementation of MP constraints that allows to fix a node to an arbitrary position of an element (the code is based mostly on the article Modélisation des câbles de précontrainte. This way we can attach the linear elements, used to model steel cables, to bi-dimensional o three-dimensional elements.

1.3 XC foundations (on the shoulders of giants)

The libraries and programs on which XC relies are the following:

- Python: user's programming language (boost.python interface with OpenSees based C++ kernel).
- OpenSees: finite element analysis kernel.
- VTK: visualization routines.
- CGAL: computational geometry algorithms library.
- NumPy: numerical computation.
- SciPy: scientific computing tools for Python.
- LaTeX: document typesetting.
- matplotlib: plotting all kind of math related figures...

1.3.1 Capabilities

The main capabilities of the program are:

- Geometry modeling and mesh generation tools.
- OD, 1D, 2D and 3D elements.
- Linear and non-linear analysis, static and dynamic.
- Fiber section models (modelization of RC members,...).
- Activation an deactivation of elements (construction phases,...).
- Tools for implementing structural codes (Eurocodes, ACI, AISC,...) verifications (in progress...).
- Interfaces with Salome and SCIA (in progress...).

XC inherits from OpenSees its advanced capabilities for modeling and analyzing the nonlinear response of systems using a wide range of material models, elements, and solution algorithms. The analysis kernel is designed for parallel computing¹ to allow scalable simulations on high-end computers or for parameter studies.

1.3.2 Model components

The programs allows the creation of bi or three-dimensional models with 2,3 or 6 degrees of freedom. The finite element library provides 1D elements (truss and beam-column elements), 2D and 3D elements for continuum mechanics and 2D shell elements.

1.3.3 Analysis

Nonlinear analysis requires a wide range of algorithms and solution methods. XC provides nonlinear static and dynamic methods, equation solvers, and methods for handling constraints.

1.4 Open source

XC is open-source. This manual provides information about the software architecture, access to the source code, and the development process. The open-source² movement allows researchers and users to build upon each others accomplishments using XC as community-based software.

1.5 Getting started

The natural way to get started is to download the software, compile it³ and run some examples. It goes as follows:

• Primo. Depending on the OS you use you'll need to make some preliminary work. MS Windows and Mac users can take a look here.

¹The Python interface and verifications tests for these capabilities are not yet implemented.

²See also The Cathedral and the Bazaar by Eric Raymond, which inspired Netscape's decision to release it's browser as open-source software.

³Yes, for the time being there is no DEB nor RPM binary packages available.

- Secondo. Go to xc repository, clone it and follow the instructions on install.linux.md.
- \bullet Terzo. You can test some of the examples in xc_examples and in tests.

You can find some useful links and introductory material here.



Chapter 2

Finite element model components

2.1 Introduction

XC is comprised of a set of Python modules and objects to perform:

- creation of the finite element model,
- specification of an analysis procedure,
- selection of quantities to be monitored during the analysis,
- and the output of results.

In each finite element analysis, an analyst constructs 5 main types of objects, as shown in figure 2.1:

Figure 2.1: Main Objects in an Analysis

Those main objects are:

- 1. **Preprocessor**: As in any finite element analysis, the analyst's first step is to subdivide the body under study into elements and nodes, to define loads acting on the elements and nodes, and to define constraints acting on the nodes. The Preprocessor is the object in the program responsible for building the Element, Node, LoadPattern, TimeSeries, Load and Constraint objects.
- 2. **Domain**: The Domain object is responsible for storing the objects created by the Preprocessor object and for providing the Analysis and Recorder objects access to these objects.
- 3. Analysis: Once the analyst has defined the model, the next step is to define the analysis that is to be performed on the model. This may vary from a simple static linear analysis to a transient non-linear analysis. The Analysis object is responsible for performing the analysis. In XC each Analysis object is composed of several component objects, which define how the analysis is performed. The component classes consist of the following: SolutionAlgorithm, Integrator, ConstraintHandler, DOF_Numberer, SystemOfEqn, Solver, and AnalysisModel.

- 4. Recorder: Once the model and analysis objects have been defined, the analyst has the option of specifying what is to be monitored during the analysis. This, for example, could be the displacement history at a node in a transient analysis or the entire state of the model at each step in the solution procedure. Several Recorder objects are created by the analyst to monitor the analysis.
- 5. **Post-processor**: Postprocessing may be defined as the "art of results representation". The post-processor is composed by the objects an modules that organize the output of the analysis such that it is easily understandable by the user. It can include checks on the codes and standards to which the construction must comply.

2.2 Nodes

2.2.1 Description

The nodes of a finite element mesh are the points where the degrees of freedom reside. Each node object has, at least, the following information:

- Coordinates which define its position in space. Typically (x,y,z) coordinates.
- Definition of the degrees of freedom in the node (displacements, rotations,...)

The nodes can also serve to define loads or masses that act over the model at its position.

2.2.2 Node creation

To create a node you can use the following commands:

```
\begin{array}{c} nodos.newNodeXY(x,y) \\ nodos.newNodeIDXY(tag,x,y) \\ nodos.newNodeXYZ(x,y,z) \\ nodos.newNodeIDXYZ(x,y,z) \end{array}
```

where:

nodos: is a node container obtained from the preprocessor.

tag: is an integer that identifies the node in the model.

(x,y) or (x,y,z): are the cartesian coordinates that define node's position.

2.2.3 Predefined spaces

Nodes definition in typical elastic FE models.

```
from model import predefined_spaces as ps
nodos= preprocessor.getNodeLoader
```

```
\begin{array}{ccc} \texttt{ps.gdls\_elasticidad2D(nodos)} & 2 \text{ node coordinates } (x,y) \\ & 2 \text{ node DOF } (u_x,u_y) \\ \texttt{ps.gdls\_resist\_materiales2D(nodos)} & 2 \text{ node coordinates } (x,y) \end{array}
```

```
\begin{array}{c} \text{3 node DOF } (u_x,u_y,\theta) \\ \text{ps.gdls\_elasticidad3D(nodos)} & \text{3 node coordinates } (x,y,z) \\ \text{3 node DOF } (u_x,u_y,u_z) \\ \text{ps.gdls\_resist\_materiales3D(nodos)} & \text{3 node coordinates } (x,y) \\ \text{6 node DOF } (u_x,u_y,u_z,\theta_x,\theta_y,\theta_z) \end{array}
```

2.3 Boundary conditions

In a finite element problem, the boundary conditions¹ are the specified values of the field variables (displacement, rotations, pore pressures,...).

2.3.1 Essential and natural boundary conditions

Essential boundary conditions are conditions that are imposed explicitly on the solution and natural boundary conditions are those that automatically will be satisfied after solution of the problem. Otherwise stated if boundary condition directly involves he nodal freedoms, such as displacements or rotations, it is essential.

This class of boundary conditions involve one or more degrees of freedom and are imposed by manipulating the left hand side (LHS) of the system of equations (the side of the stiffness matrix).

The natural boundary conditions are imposed by manipulating the right hand side (RHS) of the system of equations (the side of the force vector). Conditions involving applied loads are natural. This kind of constraints are treated in chapter 5.

2.3.2 Constraints

In XC all the classes that represent model constraints inherits from Constraint class

2.3.2.1 Classification of constraint conditions

In the previous description we have said that an essential boundary condition can involve one or more degrees of freedom.

2.3.2.1.1 Single-freedom constraints. When there is only one condition involved we call them *single-freedom constraints*. These conditions are mathematically expressible as constraints on individual degrees of freedom:

For example:

$$u_{x4} = 0, u_{y9} = 0.6 (2.1)$$

These are two single-freedom constraints. The first one is homogeneous while the second one is non-homogeneous.

¹The following explanation is based on the reference [1].

2.3.2.1.2 Multi-freedom constraints The next step up in complexity involves multifreedom equality constraints, or multifreedom constraints for short, the last name being acronymed to MFC. These are functional equations that connect two or more displacement components:

or with a more formal mathematical notation:

$$f(u_{x4}, u_{y9}, u_{y109}) = p (2.2)$$

Equation 2.2, in which all displacement components are in the left-hand side, is called the canonical form of the constraint.

An MFC of this form is called *multipoint* or *multinode* if it involves displacement components at different nodes. The constraint is called *linear* if all displacement components appear linearly on the left-hand-side, and *nonlinear* otherwise

The constraint is called *homogeneous* if, upon transferring all terms that depend on displacement components to the left-hand side, the right-hand side — the "prescribed value" in (8.3) — is zero. It is called *non-homogeneous* otherwise.

2.3.2.2 Methods for imposing the constraints

The methods for imposing the constraints are described in 6.1.1.

2.3.2.3 Single-point constraints.

A single-point constraint (SPC) enforces a single degree of freedom (normally associated with a node) to a specified value. In structural analysis we use single-point constraints to:

- Constrain or enforce translations and/or rotations of nodes that correspond to structure supports.
- Impose constraints for displacements that correspond to symmetric or antisymmetric boundary conditions.
- Removal of DOFs that correspond to frozen nodes in evolutive problems where some parts of the mesh are deactivated.

2.3.2.4 Multi-point constraints

Multipoint constraints are used to impose linear relationships between some of the degrees of freedom of the model as in:

$$\sum_{i} A_i u_i = 0 \tag{2.3}$$

where A_i are constant factors and u_i are degrees of freedom of the model.

This type of constraint allows considerable freedom in describing relations between degrees of freedom. They are used for example to create rigid elements and to link a node to an element.

2.3.2.4.1 Description An MP_Constraint represents a multiple point constraint in the domain. A multiple point constraint imposes a relationship between the displacement for certain dof at two nodes in the model, typically called the *retained* node and the *constrained* node:

$$U_c = C_{cr}U_r \tag{2.4}$$

An MP_Constraint is responsible for providing information on the relationship between the dof, this is in the form of a constraint matrix, C_{cr} , and two ID objects, retainedID and constrainedID indicating the dof's at the nodes represented by C_{cr} . For example, for the following constraint imposing a relationship between the displacements at node 1, the constrained node, with the displacements at node 2, the retainednode in a problem where the x,y,z components are identified as the 0,1,2 degrees-of-freedom:

$$u_{1,x} = 2u_{2,x} + u_{2,z} (2.5)$$

$$u_{1,y} = 3u_{2,z} (2.6)$$

the constraint matrix is:

$$C_{cr} = \begin{bmatrix} 2 & 1 \\ 0 & 3 \end{bmatrix} \tag{2.7}$$

and the vectors defining the dof's at the nodes are:

$$constrained ID = [0,1] (2.8)$$

$$retainedID = [0, 2] (2.9)$$



Chapter 3

Materials and sections

3.1 Standard uniaxial materials

3.1.1 defElasticMaterial

Constructs an elastic uniaxial material

from materials import typical_materials
typical_materials.defElasticMaterial(preprocessor,name,E)

preprocessor preprocessor name
name name identifying the material
E tangent in the stress-strain diagram (see figure 3.1)

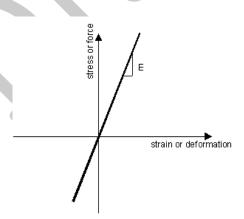


Figure 3.1: Elastic uniaxial material. Stress-strain diagram

3.1.2 defElasticPPMaterial

Constructs an elastic perfectly-plastic uniaxial material

from materials import typical_materials
typical_materials.defElasticPPMaterial(preprocessor,name,E,fyp,fyn)

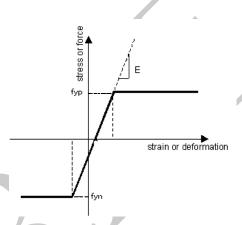


Figure 3.2: Elastic perfectly-plastic uniaxial material. Stress-strain diagram

3.1.3 defElastNoTensMaterial

Constructs a uniaxial elastic-no tension material

from materials import typical_materials
typical_materials.defElastNoTensMaterial(preprocessor,name,E)

preprocessor preprocessor name
name name identifying the material
E tangent in the elastic zone of the stress-strain diagram (see figure 3.3)

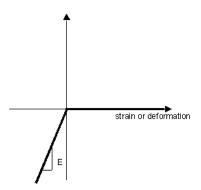


Figure 3.3: Elastic-no tension material. Stress-strain diagram

3.2 Steel and reinforcing steel materials

3.2.1 defCableMaterial

Constructs a uniaxial bilinear prestressed material. The stress strain ranges from slack (large strain at zero stress) to taught (linear with modulus E).

from materials import typical_materials
typical_materials.defCableMaterial(preprocessor,name,E,prestress,rho)

preprocessor name

name identifying the material

E elastic modulus

prestress prestress

rho effective self weight (gravity component of weight per vol-

ume transverse to the cable)

3.2.2 defSteel01

Constructs a uniaxial bilinear steel material object with kinematic hardening

```
from materials import typical_materials
typical_materials.defSteelO1(preprocessor,name,E,fy,b)
```

preprocessor	preprocessor name
name	name identifying the material
E	initial elastic tangent (see figure 3.4)
fy	yield strength (see figure 3.4)

b strain-hardening ratio: ratio between post-yield tangent

and initial elastic tangent (see figure 3.4)

3.2.3 defSteel02

Constructs a uniaxial Giuffre-Menegotto-Pinto steel material object with isotropic strain hard-ening

```
from materials import typical_materials
typical_materials.defSteel02(preprocessor,name,E,fy,b,initialStress)
```

preprocessor name

name name identifying the material

initial electic tengent (see figure 3.5)

b strain-hardening ratio: ratio between post-yield tangent and initial elastic tangent)

initialStress initial stress

The transition from elastic to plastic branches (see figure 3.5) is controlled by parameters R0, R1, R2. The default values R0=15, R1=0.925 and R2=0.15

3.2.4 ReinforcingSteel

This class constructs a bilinear stress-strain diagram to carry out the analysis of reinforced concrete according to Eurocode 2. Other national standards, like the spanish EHE and the swiss SIA also adopt this diagram.

Parameters

nmbMaterial	name identifying the material
${\tt nmbDiagK}$	name identifying the characteristic stress-strain diagram (default:
	"dgK"+nmbMaterial)
$\mathtt{matTagK}$	tag of the uniaxial material with the characteristic stress-strain
	diagram
nmbDiagD	name identifying the design stress-strain diagram (default:
	"dgD"+nmbMaterial)
$\mathtt{matTagD}$	tag of the uniaxial material with the design stress-strain diagram
fyk	characteristic value of the yield strength
gammaS	partial factor for material (default: 1.15)
Es	elastic modulus of the material (default: 2e11)
emax	maximum strain at failure point
k	ratio between characteristic ultimate stress and characteristic
	yield stress ⁽¹⁾ (default: 1.05)

^{(1):} according to annex C of EC2: for class A k $\geq 1,\!05,$ for class B k $\geq 1,\!08$

Methods

fmaxk()	characteristic value of the ultimate strength
fyd()	design value of the yield strength
eyk()	characteristic strain at yield point
eyd()	design strain at yield point
Esh()	post-yield tangent
bsh()	ratio between post-yield tangent and initial elastic tangent
defDiagK(preprocesso	rteturns XC uniaxial material (characteristic values)
defDiagD(preprocesso	r) returns XC uniaxial material (design values)

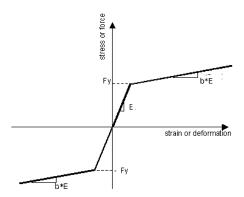


Figure 3.4: Steel001: uniaxial bilinear steel material with kinematic hardening. Stress-strain diagram

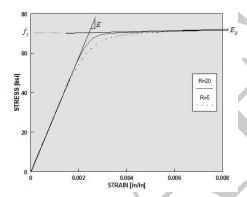


Figure 3.5: Steel002: uniaxial bilinear steel material with isotropic strain hardening. Stress-strain diagram

3.3 Concrete materials

3.3.1 defConcrete01

Constructs a uniaxial Kent-Scott-Park concrete material object with degraded linear unloading/reloading stiffness according to the work of Karsan-Jirsa and no tensile strength.

from materials import typical_materials
typical_materials.defConcreteO1(preprocessor,name,epsc0,fpc,fpcu,epscu)

preprocessor	preprocessor name
name	name identifying the material
fpc	concrete compressive strength at 28 days (compression is negative) ⁽¹⁾
epsc0	concrete strain at maximum strength (see figure 3.6) (2)
fpcu	concrete crushing strength (see figure 3.6)
epscu	concrete strain at crushing strength (see figure 3.6)

- (1): Compressive concrete parameters should be input as negative values (if input as positive, they will be converted to negative internally)
- (2): The initial slope for this model is 2 * fpc/epsc0 (see figure 3.6)

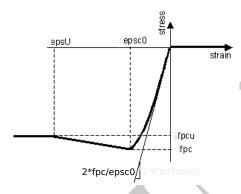


Figure 3.6: Concrete01: uniaxial Kent-Scott-Park concrete material. Stress-strain diagram

3.4 ND materials

An ND material is an object that represents the stress-strain relationship at the gauss-point of a continuum element.

3.4.1 defElasticIsotropic3d

Constructs an elastic isotropic material.

from materials import typical_materials
typical_materials.defElasticIsotropic3d(preprocessor,name,E,nu,rho)

preprocessor	preprocessor name
name	name identifying the material
E	elastic modulus
nu	Poisson's ratio
rho	mass density, optional (default $= 0.0$)

3.4.2 defElasticIsotropicPlaneStrain

Constructs an elastic isotropic plane-strain material.

```
from materials import typical_materials
typical_materials.defElasticIsotropicPlaneStrain(preprocessor,name,E,nu,rho)
```

preprocessor name

name identifying the material

E elastic modulus nu Poisson's ratio

rho mass density, optional (default = 0.0)

3.4.3 defElasticIsotropicPlaneStress

Constructs an elastic isotropic plane-stress material.

from materials import typical_materials
typical_materials.defElasticIsotropicPlaneStress(preprocessor,name,E,nu,rho)

preprocessor name

name identifying the material

E elastic modulus nu Poisson's ratio

rho mass density, optional (default = 0.0)

3.5 Sections

A section represents a force-deformation (or resultant stress-strain) relationship at beam-column or plate element.

Three types of sections are going to be considered:

Elastic: defined by material and geometric constants;

Resultant: general nonlinear description of force-deformation response, e.g. moment-curvature;

Fiber: section is discretized into smaller regions for which the material stress-strain response is integrated to give resultant behavior, e.g. reinforced concrete.

3.5.1 Sections definition

3.5.1.1 sectionProperties

It's an abstract class used to define the properties of a generic section.

from materials import sectionProperties
sectionProperties.sectionProperties(name,E,nu)

Parameters

name	name identifying the section
E	elastic modulus of material
nu	Poisson's ratio of material

Abstract methods

A()	cross-sectional area of the section (to override)
Iy()	second moment of area about the local y-axis (to override)
Iz()	second moment of area about the local z-axis (to override)
J()	torsional moment of inertia of the section (to override)
G()	transverse modulus of elasticity (defaults to $\frac{E}{2(1+\nu)}$
alphaY()	coefficient of distortion about the local y-axis
alphaZ()	coefficient of distortion about the local z-axis
Wyel()	elastic section modulus about the local y-axis
Wzel()	elastic section modulus about the local z-axis

Methods

sectionProperties.defSeccElastica3d(preprocessor)

returns an elastic section appropriate for 3D beam analysis, from the data of the section geometric properties

sectionProperties.defSeccShElastica3d(preprocessor)

returns an elastic section appropriate for 3D beam analysis including shear deformations, from the data of the section geometric properties

sectionProperties.defSeccElastica2d(preprocessor)

returns an elastic section appropriate for 2D beam analysis, from the data of the section geometric properties

sectionProperties.defSeccShElastica2d(preprocessor)

returns an elastic section appropriate for 2D beam analysis including shear deformations, from the data of the section geometric properties

3.5.1.2 defSeccElastica3d

Returns an elastic section appropriate for 3D beam analysis.

from materials import sectionProperties
sectionProperties.defSeccElastica3d(preprocessor,defSecc)

preprocessor

preprocessor name

defSecc name identifying the class object used to define the prop-

erties of the section

3.5.1.3 defSeccShElastica3d

Returns an elastic section appropriate for 3D beam analysis, including shear deformations.

from materials import sectionProperties
sectionProperties.defSeccShElastica3d(preprocessor,defSecc)

preprocessor

preprocessor name

defSecc

name identifying the class object used to define the prop-

erties of the section

3.5.1.4 defSeccElastica2d

Returns an elastic section appropriate for 2D beam analysis.

from materials import sectionProperties
sectionProperties.defSeccElastica2d(preprocessor,defSecc)

preprocessor

preprocessor name

defSecc

name identifying the class object used to define the prop-

erties of the section

3.5.1.5 defSeccShElastica2d

Returns an elastic section appropriate for 2D beam analysis, including shear deformations.

from materials import sectionProperties
sectionProperties.defSeccShElastica2d(preprocessor,defSecc)

preprocessor

preprocessor name

 ${\tt defSecc}$

name identifying the class object used to define the prop-

erties of the section

3.5.1.6 Circular Section

This subclass derived from superclass sectionProperties is used to define the properties of a circular section.

from materials import paramCircularSection
paramCircularSection.CircularSection(name,r,E,nu)

Parameters

name	name identifying the section	
r	radius	
E	elastic modulus of material	
nu	Poisson's ratioof material	

Methods

cross-sectional area of the section
second moment of area about the local y-axis
second moment of area about the local z-axis
torsional moment of inertia of the section
coefficient of distortion about the local y-axis
coefficient of distortion about the local z-axis

Functions for defining circular sections

```
paramCircularSection.setupSeccCircular(sectionName,r,E,nu)
returns a circular section
paramCircularSection.defSeccCircularElastica3d(preprocessor, defSecc)
returns an elastic circular section appropriate for 3D beam
analysis.
paramCircularSection.defSeccCircularShElastica3d(preprocessor, def-Secc)
```

returns an elastic circular section appropriate for 3D beam analysis, including shear deformations.

paramCircularSection.defSeccCircularElastica2d(preprocessor, defSecc) returns an elastic circular section appropriate for 2D beam analysis.

paramCircularSection.defSeccCircularShElastica2d(preprocessor, def-Secc)

returns an elastic circular section appropriate for 2D beam analysis, including shear deformations.

preprocessor : preprocessor name
defSecc : name identifying an object of the class Circu-

larSection with the properties of the section.

3.5.1.7 Rectangular Section

This subclass derived from superclass sectionProperties is used to define the properties of a rectangular section.

from materials import paramRectangularSection
paramRectangularSection.RectangularSection(,name,b,h,E,nu)

Parameters

name	name identifying the section	
b	width	
h	overall depth	
E	elastic modulus of material	
nu	Poisson's ratioof material	

Methods

A (cross-sectional area of the section)
Iy()	second moment of area about the local y-axis (Y= weak
	axis)
Iz()	second moment of area about the local z-axis (Z= strong
	axis)
J()	torsional moment of inertia of the section
Wyel()	elastic section modulus about the local y-axis
Wzel()	elastic section modulus about the local z-axis
alphaY()	coefficient of distortion about the local y-axis
alphaZ()	coefficient of distortion about the local z-axis

Functions for defining parameters of rectangular sections

3.5.1.8 sccRectang

This class is used to define a geometric rectangular section.

```
from materials import sccRectg
sccRectg.sccRectang()
```

Parameters

b	width
h	overall depth
nDivIJ	number of cells in IJ (width) direction
nDivJK	number of cells in JK (height) direction

Methods

area() rectangle area second moment of area about the axis through the cen of gravity and parallel to the width dimension (axis 1) i1() bending radius of the section with regard to the axis 1 well(fy) yield moment of the rectangular section about the axis fy = yield stress of the section material	
of gravity and parallel to the width dimension (axis 1) i1() bending radius of the section with regard to the axis 1 Me1(fy) yield moment of the rectangular section about the axis	
i1() bending radius of the section with regard to the axis 1 Me1(fy) yield moment of the rectangular section about the axis	tre
Me1(fy) yield moment of the rectangular section about the axis	
fy = yield stress of the section material	s 1
-	
S1PosG() first moment of the area of half a section with regard to t	the
axis 1	
Mp1(fy) plastic moment of the rectangular section about the axi	s 1
fy = yield stress of the section material	
12() second moment of area about the axis through the cen	tre
of gravity and parallel to the high dimension (axis 2)	
i2() bending radius of the section with regard to the axis 2	
Me2(fy) yield moment of the rectangular section about the axis	s 2
fy = yield stress of the section material	
S2PosG() first moment of the area of half a section with regard to t	the
axis 2	
Mp2(fy) plastic moment of the rectangular section about the axi	s 2
fy = yield stress of the section material	
discretization(gm,nmbMat)	
returns the discretized region	

gm: name identifying a section geometry nmbMat: name identifying the material

3.5.1.9 SteelProfile

This subclass derived from superclass sectionProperties is used to define the properties of a structural steel section.

```
from materials import steelProfile
steelProfile.SteelProfile(steel,name,table)
```

Parameters

steel	type of structural steel (e.g. S275JR)
table	file containing a table (dictionary) of steel profiles (e.g. per-
	files_metalicos.arcelor.perfiles_he_arcelor)
name	name of the profile in the table (e.g. $"HE_400_B"$)

Methods

A()	cross-sectional area of the section
Iy()	second moment of area about the local y-axis (Y= weak axis)
Iz()	second moment of area about the local z-axis (Z= strong axis)
J()	torsional moment of inertia of the section
EIy()	product of $E_{steel} \times I_y$
EIz()	product of $E_{steel} \times I_z$
GJ()	product of $E_{steel} \times G$
<pre>alphaY()</pre>	coefficient of distortion about the local y-axis
alphaZ()	coefficient of distortion about the local z-axis

3.5.2 Elastic sections

3.5.2.1 defElasticSection2d

Constructs an elastic section appropriate for 2D beam analysis.

from materials import typical_materials
typical_materials.defElasticSection2d(preprocessor,name,A,E,I)

preprocessor	preprocessor name
name	name identifying the section
A	cross-sectional area of the section
E	Young's modulus of material
I	second moment of area about the local z-axis

${\bf 3.5.2.2} \quad {\bf defElasticShearSection2d}$

Constructs an elastic section appropriate for 2D beam analysis, including shear deformations.

```
from materials import typical_materials
typical_materials.defElasticShearSection2d(preprocessor,name,A,E,G,I,alpha)
```

preprocessor	preprocessor name
name	name identifying the section
A	cross-sectional area of the section
E	Young's modulus of material
G	shear modulus
I	second moment of area about the local z-axis
alpha	shear shape factor

${\bf 3.5.2.3} \quad {\bf defElastic Section From Mech Prop 2d}$

Constructs an elastic section appropriate for 2D beam analysis, taking mechanical properties of the section form a MechProp2d object.

from materials import typical_materials
typical_materials.defElasticSectionFromMechProp2d(preprocessor,name,mechProp2d)

preprocessor	preprocessor name
name	name identifying the section
mechProp2d	object that contains mechanical properties of the section

3.5.2.4 defElasticSection3d

Constructs an elastic section appropriate for 3D beam analysis.

from materials import typical_materials
typical_materials.defElasticSection3d(preprocessor,name,A,E,G,Iz,Iy,J)

preprocessor	preprocessor name
name	name identifying the section
A	cross-sectional area of the section
E	elastic modulus of material
Iz	second moment of area about the local z-axis
Iy	second moment of area about the local y-axis
J	torsional moment of inertia of the section

3.5.2.5 defElasticShearSection3d

Constructs an elastic section appropriate for 3D beam analysis, including shear deformations.

```
from materials import typical_materials
typical_materials.defElasticShearSection3d(preprocessor,name,A,E,G,Iz,Iy,J,alpha)
```

preprocessor	preprocessor name
name	name identifying the section
A	cross-sectional area of the section
E	elastic modulusof material
G	transverse modulus of elasticity
Iz	second moment of area about the local z-axis
Iy	second moment of area about the local y-axis
J	torsional moment of inertia of the section
alpha	shear shape factor

${\bf 3.5.2.6} \quad {\bf defElasticSectionFromMechProp3d}$

Constructs an elastic section appropriate for 3D beam analysis, taking mechanical properties of the section form a MechProp3d object.

from materials import typical_materials
typical_materials.defElasticSectionFromMechProp3d(preprocessor,name,mechProp3d)

preprocessor	preprocessor name
name	name identifying the section
mechProp3d	object that contains mechanical properties of the section

3.5.2.7 defElasticMembranePlateSection

Constructs an an isotropic elastic section appropriate for plate and shell analysis.

from materials import typical_materials
typical_materials.defElasticMembranePlateSection(preprocessor,name,E,nu,rho,h)

preprocessor	preprocessor name
name	name identifying the section
E	elastic modulus
nu	Poisson's ratio
rho	mass density
h	overall depth of section

3.5.2.8 defElasticPlateSection

Constructs an an isotropic elastic section appropriate for plate analysis.

```
from materials import typical_materials
typical_materials.defElasticPlateSection(preprocessor,name,E,nu,rho,h)
```

preprocessor name

name identifying the section

E elastic modulus nu Poisson's ratio rho mass density

h overall depth of section

3.5.3 Fiber sections

3.5.3.1 FiberSet

This class constructs a set of all the fibers made of the same material from a fiber section

from materials.fiber_section import createFiberSets
createFiberSets.FiberSet(scc,setName,matTag)

Parameters

scc name identifying the fiber section

setName name of the set of fibers to be generated

matTag tag of the uniaxial material which forms the fibers

Methods

getFiberWithMinStrain()

returns the fiber with the minimum straim from the set of

fibers

getFiberWithMaxStrain()

returns the fiber with the maximum straim from the set of fibers

3.5.3.2 RCSets

This class constructs both the concrete and reinforced steel fiber sets from a reinforced concrete fiber section.

from materials.fiber_section import createFiberSets
createFiberSets.RCSets(scc,concrMatTag, concrSetName,reinfMatTag, reinfSetName)

Parameters

name identifying the fiber section

tag of the uniaxial material that makes up the concrete fibers of the section

concrSetName name of the set of fibers of concrete to be generated

reinfMatTag tag of the uniaxial material that makes up the reinforcing steel fibers of the section

reinfSetName name of the set of fibers of reinforcing steel to be generated

Methods

reselTensionFibers(scc,tensionFibersSetName)

returns a set with those fibers in tension from the total set

getConcreteArea(factor)

returns the cross section area of concrete in the set

getMaxConcreteStrain()

returns the maximum strain in the set of concrete fibers

getConcreteInitialTangent()

returns the initial tangent in the stress-strain diagram of

the material that makes up the fibers of concrete

getConcreteCompression()

returns the resultant of compressive stresses in concrete

fibers

getNumTensionRebars()

returns the number of reinforcing steel fibers in tension

3.5.3.3 fiberSectionSetupRCSets

Returns an object of the class RCSets

from materials.fiber_section import createFiberSets
createFiberSets.fiberSectionSetupRCSets(scc,concrMatTag, concrSetName,reinfMatTag, reinfSetName)

scc name identifying the fiber section

concrMatTag tag of the uniaxial material that makes up the concrete

fibers of the section

concrSetName name of the set of fibers of concrete to be generated reinfMatTag tag of the uniaxial material that makes up the reinforcing

steel fibers of the section

reinfSetName name of the set of fibers of reinforcing steel to be generated

3.5.3.4 createRCFiberSets

Constructs the sets of concrete fibers ("hormigon") and reinforcing steel fibers ("reinforcement") for all the elements included in a set of elements.

from materials.fiber_section import createFiberSets
createFiberSets.createRCFiberSets(preprocessor, setName, concrMatTag, reinfMatTag)

preprocessor name

setName name identifying the set of elements

concrMatTag tag of the uniaxial material that makes up the concrete

fibers of the section

reinfMatTag tag of the uniaxial material that makes up the reinforcing

steel fibers of the section

3.5.3.5 reselTensionFibers

Returns the fibers under tension included in a set of fibers.

from materials.fiber_section import createFiberSets
createFiberSets.reselTensionFibers(scc,fiberSetName,tensionFibersSetName)

tensionFibersSetName name of the set of tensioned fibers returned

3.5.3.6 fiberSectionSetupRC3Sets

Returns a set of tensioned fibers ("reinforcementTraccion") of a fiber section of reinforced concrete.

from materials.fiber_section import createFiberSets
createFiberSets.fiberSectionSetupRC3Sets(scc,concrMatTag, concrSetName,reinfMatTag, reinfSetName)

scc name identifying the fiber section

concrMatTag tag of the uniaxial material that makes up the concrete

fibers of the section

concrSetName name of the set of fibers of concrete to be generated

reinfMatTag tag of the uniaxial material that makes up the reinforcing

steel fibers of the section

reinfSetName name of the set of fibers of reinforcing steel to be generated

3.5.3.7 RecordRCColumnSection

This class is used to define the variables that make up a reinforced concrete section with reinforcement symmetric in both directions (as usual in columns)

from materials.fiber_section import defSeccionHAPilar
defSeccionHAPilar.RecordRCColumnSection()

Parameters

sectionName name identifying the section section description sectionDescr name identifying the geometric section gmSectionName type of concrete (e.g. hormigonesEHE.HA25) concrType concrDiagName name identifying the characteristic stress-strain diagram of the concrete material depth cross-section depth width cross-section width nDivIJ number of cells in IJ (width) direction number of cells in JK (height) direction nDivJKreinfSteelType type of reinforcement steel name identifying the characteristic stress-strain diagram of the reinfDiagName reinforcing steel material cover number of rebars in the width direction of the section (each face) nRebarsWidth cross sectional area of each rebar in width direction areaRebarWidth number of rebars in the height direction of the section (each face nRebarsDepthareaRebarDepth cross sectional area of each rebar in height direction shReinfZ record of .defRCSimpleSectype tion.RecordShearReinforcement() defining the shear reinforcement in Z direction shReinfY of .defRCSimpleSecrecord type tion.RecordShearReinforcement() defining the shear reinforcement in Y direction

Methods

defGeomRCColumnSection(matDiagType)

returns a reinforced concrete section with reinforcement symmetric in both directions
matDiagType ="k" for characteristic diagram, ="d" for design diagram

3.5.3.8 RecordShearReinforcement

This class is used to define the variables that make up a family of shear reinforcing bars.

 $\label{lem:condition} from \ materials.fiber_section \ import.defRCSimpleSection \\ defSeccionHASimple.RecordShearReinforcement()$

Parameters

familyName nShReinfBranches areaShReinfBranch	name identifying the family of shear reinforcing bars number of effective branches area of the shear reinforcing bar
shReinfSpacing	longitudinal distance between transverse reinforcements (defaults
1 0	to 0.2)
${\tt angAlphaShReinf}$	angle between the shear reinforcing bars and the axis of the mem-
	ber (defaults to $\Pi/2$)
${\tt angThetaConcrStruts}$	angle between the concrete's compression struts and the axis of
	the member (defaults to $\Pi/4$)

Methods

getAs()	returns the area per unit length of the family of shear rein-
	forcements

3.5.3.9 RecordRCSimpleSection

This class is used to define the variables that make up a rectangular reinforced concrete section with single reinforcement layers in top and bottom faces

from materials.fiber_section import.defRCSimpleSection
defSeccionHASimple.RecordRCSimpleSection()

Parameters

sectionName sectionDescr	name identifying the section section description
tipoHormigón	type of concrete (e.g. hormigonesEHE.HA25)
concrDiagName	name identifying the characteristic stress-strain diagram of the
	concrete material
depth	cross-section height
width	cross-section width
nDivIJ	number of cells in IJ (width) direction

Parameters ... continued from previous page

number of cells in JK (height) direction nDivJK type of reinforcement steel reinfSteelType name identifying the characteristic stress-strain diagram of the reinfDiagName reinforcing steel material number of longitudinal rebars in the negative face of the section nRebarsNeg area of each longitudinal rebar in the negative face areaRebarNeg rebarsDiamNeg diameter of the bars in the negative face of the section coverNeg cover of longitudinal reinforcement in the negative face lateral cover of longitudinal reinforcement in the negative face coverLatNeg number of longitudinal rebars in the positive face of the section nRebarsPos areaRebarPos area of each longitudinal rebar in the positive face diameter of the bars rebar in the positive face of the section rebarsDiamPos cover of longitudinal reinforcement in the positive face coverPos lateral cover of longitudinal reinforcement in the positive face coverLatPos coverMin minimal covering of the longitudinal reinforcement shReinfZ record of .defRCSimpleSectype tion.RecordShearReinforcement() defining the shear reinforcement in Z direction defRCSimpleSecshReinfY record type tion.RecordShearReinforcement() defining the shear reinforcement in Y direction

Methods

defGeomRCColumnSection(matDiagType)

returns a reinforced concrete section with reinforcement symmetric in both directions matDiagType ="k" for characteristic diagram, ="d" for design diagram

3.5.3.10 RecordRCSlabSection

This class is used to define the variables that make up a reinforced concrete slab section with single reinforcement layers in top and bottom faces

from materials.fiber_section import.defRCSimpleSection
defSeccionHASimple.RecordRCSlabSection(nmb,desc,depth,concrete,steel,basicCover)

Parameters

nmb basic name identifying the section

the name identifying the section in 1 direction is formed adding to nmb the suffix "L"; for the section in 2 direction the suffix added

to the basic name is "T"

desc basic section description

depth cross-section height (width is considered = 1) concrete type of concrete (e.g. hormigonesEHE.HA25)

steel type of reinforcement steel

basicCover minimal cover of longitudinal reinforcement

Methods

setMainReinf2neg(diam, area, spacing)

defines the rebar arrangement of the layer of main reinforcement bars corresponding to the negative face of the section

in 2 direction

diam = diameter of the bar
area = area of each bar

spacing = space between axis of bars

setMainReinf2pos(diam, area, spacing)

idem for the layer of main reinforcement bars corresponding to the positive face of the section in 2 direction

setMainReinf1neg(diam, area, spacing)

idem for the layer of main reinforcement bars corresponding to the negative face of the section in 1 direction

setMainReinf1pos(diam, area, spacing)

idem for the layer of main reinforcement bars corresponding to the positive face of the section in 1 direction

setShearReinfT(nShReinfBranches, areaShReinfBranch, spacing)

defines the rebar arrangement of the shear reinforcement bars corresponding to the section in 2 direction nShReinfBranches = number of effective branches areaShReinfBranch = area of the shear reinforcing bar spacing = longitudinal distance between shear reinforcements bars

setShearReinfL(nShReinfBranches, areaShReinfBranch, spacing)

idem for shear reinforcement bars corresponding to the section in 1 direction

3.5.3.11 getIMaxPropFiber

Returns the fiber in a set with the maximum value of a property.

from materials.fiber_section import fiberUtils
fiberUtils.getIMaxPropFiber(fibers,methodName)

fibers name of the set of fibers

methodName name of the method or property (e.g. "getArea")

3.5.3.12 getIMinPropFiber

Returns the fiber with the minimum value of a property from a set of fibers.

from materials.fiber_section import fiberUtils
fiberUtils.getIMinPropFiber(fibers,methodName)

fibers name of the set of fibers

methodName name of the method or property (e.g. "getArea")

3.5.3.13 Utils

3.5.3.13.1 tipoSolicitacion Returns the following values, depending on the state of stress in the section:

- 1 pure or combined tension where the entire section is under tension;
- 2 pure or combined bending (there are fibers in tension and in compression);
- 3 single or combined compression where all the fibers are in compression.

from materials import regimenSeccion
regimenSeccion.tipoSolicitacion(epsCMin, epsSMax)

epsCMin minimum strain in concrete
epsSMax maximum strain in steel

3.5.3.13.2 strTipoSolicitacion Returns:

"tracción simple o compuesta" in pure or combined tension state;

"flexotracción" in pure or combined bending state;

"compresión simple o compuesta" in single or combined compression state;

"falla" in all other cases.

from materials import regimenSeccion
regimenSeccion.strTipoSolicitacion(tipoSol)

tipoSol =1 for pure or combined tension state

=2 for pure or combined bending state=3 for single or combined compression state

3.5.3.14 gmHorizRowRebars

Returns a reinforcement layer parallel to the top and bottom sides of the rectangular section. The row of reinforcement bars is placed at a distance h over the horizontal axis of the section.

from materials.fiber_section import geomArmaduraSeccionesFibras
geomArmaduraSeccionesFibras.gmHorizRowRebars(sectionGeom, fiberMatName,
nRebars, areaRebar, depth, width, cover, h)

sectionGeom name identifying the geometric section

fiberMatName name identifying the uniaxial material which forms the

fibers

nRebars number of rebars in the layer

areaRebararea of each rebardepthcross-section heightwidthcross-section width

cover cover

h distance between the horizontal axis of the cross-section

and the reinforcement layer

3.5.3.15 gmBottomRowRebars

Returns a reinforcement layer in the bottom side of a rectangular section.

from materials.fiber_section import geomArmaduraSeccionesFibras
geomArmaduraSeccionesFibras.gmBottomRowRebars(sectionGeom,fiberMatName,
nRebars, areaRebar, depth, width, cover)

sectionGeom name identifying the geometric section

fiberMatName name identifying the uniaxial material which forms the

fibers

nRebars number of rebars in the layer

areaRebararea of each rebardepthcross-section heightwidthcross-section width

cover cover

${\bf 3.5.3.16} \quad {\bf ggmTopRowRebars}$

Returns a reinforcement layer in the top side of a rectangular section.

from materials.fiber_section import geomArmaduraSeccionesFibras geomArmaduraSeccionesFibras.ggmTopRowRebars(sectionGeom,fiberMatName,nRebars, areaRebar, depth, width, cover)

sectionGeom name identifying the geometric section

fiberMatName name identifying the uniaxial material which forms the

 $_{
m fibers}$

nRebars number of rebars in the layer

areaRebararea of each rebardepthcross-section heightwidthcross-section width

cover cover

3.5.3.17 gmSquareSection

Returns a square geometric region of fibers' material.

from materials.fiber_section import geomSeccionesFibras
geomSeccionesFibras.gmSquareSection(geomSection, fiberMatName, ld, nD)

geomSection name identifying the geometric section

fiberMatName name identifying the uniaxial material which forms the

fibers

ld side of the square

nD number of cells in each side of the square

3.5.3.18 gmRectangSection

Returns a rectangular geometric region of fibers' material.

from materials.fiber_section import geomSeccionesFibras
geomSeccionesFibras.gmRectangSection(geomSection,fiberMatName, h, b,
nDIJ, nDIK)

geomSection name identifying the geometric section

fiberMatName name identifying the uniaxial material which forms the

fibers

h side of the rectangle in IK direction
b side of the rectangle in IJ direction
nDIJ number of cells in IJ direction
nDIK number of cells in IK direction

3.5.3.19 RecordFamMainReinforcement

This class constructs a family of main reinforcing bars.

materials.fiber_section import informeGeomSeccion informeGeomSeccion.RecordFamMainReinforcement(reinfLayer)

Parameters

${\tt reinfLayer}$	name identifying the family of reinforcing bars
nRebars	number of rebars in the layer

rebarsDiam diameter of the bars

areaRebar total area of the bars in the family

coverMec cover of the reinforcement

cdgBarras position of the reinforcement centre of gravity

Methods

texWrite(archTex)

writes the value of parameters in a file named archTex

${\bf 3.5.3.20} \quad imprime Main Rein forcement$

The function writes in the specified file the characteristics (diameter, area, cover, position of gravity centre, ...) of each main reinforcement family provided in a list of family names.

from materials.fiber_section import informeGeomSeccion
informeGeomSeccion.imprimeMainReinforcement(listaFamMainReinforcement,
areaHorm, archTex)

listaFamMainReinforcemente identifying a list of names of reinforcement families areaHorm area of the concrete section archTex name of the output file

3.5.3.21 imprimeArmaduraCortante

The function writes in the specified file the characteristics (diameter, area, distances, angles, \dots) of a family of shear reinforcement bars.

 $\label{lem:condition} from \ materials.fiber_section \ import \ informeGeomSeccion.imprimeArmaduraCortante(recordShearReinf, archTex)$

recordShearReinf name identifying a family of shear reinforcement

archTex name of the output file

3.5.3.22 informeGeomSeccion

This function prepares a report with the characteristics of a sections, as illustrated in table 3.65

from materials.fiber_section import informeGeomSeccion
informeGeomSeccion.informeGeomSeccion(preprocessor,scc, archTex, pathFigura)

preprocessor name

name identifying the section archTex name of the output file

pathFigura path to place the figure file representing the reinforcement

arrangement

3.5.4 Sections calculation

This class is used to perform the stress calculations in a rectangular reinforced concrete section with single reinforcement layers in top and bottom faces.

from materials import stressCalc
stressCalc.stressCalc(b,h,r,rp,As,Asp,Ec,Es)

Parameters

- b cross-section width
- h cross-section height
- r cover of longitudinal reinforcement in the positive face
- ${\tt rp} {\tt cover}$ of longitudinal reinforcement in the negative face
- As area of longitudinal reinforcement in the positive face
- Asp area of longitudinal reinforcement in the negative face
- Ec concrete elastic modulus
- Es reinforcing steel elastic modulus
- N normal force
- M bending moment

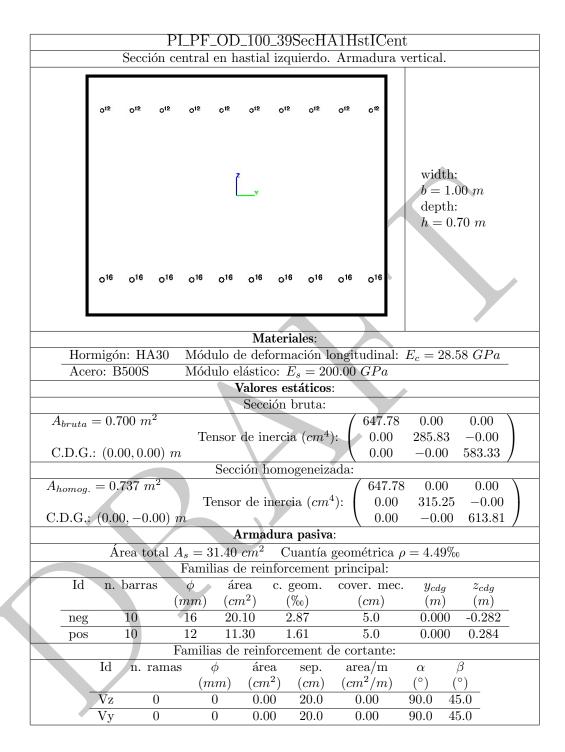


Table 3.65: Sección central en hastial izquierdo. Armadura vertical. (PI_PF_OD_100_39SecHA1HstICent).

Chapter 4

Elements

4.1 Zero-length elements

4.1.1 ZeroLength

The ZeroLength class represents an element defined by two nodes at the same geometric location, hence it has zero length.

The nodes are connected by means of uniaxial materials to represent the force-deformation relationship for the element.

ZeroLength elements are constructed with a tag in a domain of dimension 1, 2, or 3, connected by nodes Nd1 and Nd2. The vector \vec{x} defines the local x-axis for the element and the vector \vec{yp} lies in the local x-y plane for the element. The local z-axis is the cross product between \vec{x} and \vec{yp} , and the local y-axis is the cross product between the local z-axis and \vec{x} .

The force-deformation relationship for the element is given by a pointer *theMaterial* to a **UniaxialMaterial** model acting in local *direction*.

The local direction is 0, 1, 2 for translation in the local x, y, z axes or 3, 4, 5 for rotation about the local x, y, z axes.

```
preprocessor=xc.ProblemaEF().getModelador
ZeroLengthElement=preprocessor.getElementLoader.newElement("zero_length",
xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics
getDimension	element dimension
${\tt getVtkCellType}$	cell type for Vtk graphics
getIVector	vector in the element local x-axis direction

Parameters ... continued from previous page

getJVector vector in the element local y-axis direction vector in the element local z-axis direction

Methods

the element is to commit its current state; returns 0 if succommitState() cessful, a negative number if not the element is to set its current state to the last committed revertToLastCommit() state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at revertToStart before the analysis started; returns 0 if successful, a negative number if not returns the number of DOF associated with the element; getNumDOF this should equal the sum of the DOFs at each of the external nodes getResistingForce() returns the resisting force vector for the element; this is equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation returns cell type for Vtk graphics getVtkCellType() getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry shape geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape reset the tributary length, area and volume of connected resetTributarias() nodes vuelcaTributarias

calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the Node given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter <code>geomInicial = True</code> to consider the initial geometry shape in the calculation or <code>geomInicial = False</code> for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter <code>geomInicial</code> = True to consider the initial geometry shape in the calculation or <code>geo-mInicial</code> = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMinCooNod(axisIdx) returns the minimum value among the coordinates in the axisIdx axis of the external nodes associated with the element (axisIdx adopts the values 0,1,...)

setupVectors(x,yp) establish orientation of element for the transformation matrix

 ${\bf x}$ vector in global coordinates defining local x-axis ${\bf yp}$ vector in global coordinates which lies in the local x-y plane for the element

local z-axis is defined by the vector $\vec{z} = \vec{x} \times y\vec{p}$

clears all material definition of the element clearMaterials() setMaterial(dir,matName) assigns uni-axial materials to the different directions of the element

dir: integer representing the direction in which the uniaxial material acts: dir is 0,1,2 for translation in the local x, y, z axes or 3, 4, 5 for rotation about the local x, y, z

axes.

matName: string representing the name of the material

getMaterials

ZeroLengthSection 4.1.2

The ZeroLengthSection class represents an element defined by two nodes at the same geometric location, hence it has zero length.

The nodes are connected by a SectionForceDeformation object which represents the forcedeformation relationship for the element.

ZeroLength elements are constructed with a tag in a domain of dimension 1, 2, or 3, connected by nodes Nd1 and Nd2. The vector \vec{x} defines the local x-axis for the element and the vector \vec{yp} lies in the local x-y plane for the element. The local z-axis is the cross product between \vec{x} and $y\bar{p}$, and the local y-axis is the cross product between the local z-axis and \vec{x} .

The force-deformation relationship for the element is obtained by invoking getCopy() on the SectionForceDeformation pointer the Section. The section model acts in the local space defined by the \vec{x} and $y\vec{p}$ vectors. The section axial force-deformation acts along the element local x-axis and the section y-z axes directly corresponds to the local element y-z axes.

```
preprocessor=xc.ProblemaEF().getModelador
ZeroLengthElement=preprocessor.getElementLoader.newElement(
"zero_length_section",xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag

tags of the nodes connected by the element

Parameters

getIdxNodes vector containing the node index to be used in Vtk graphics getDimension element dimension cell type for Vtk graphics getVtkCellType getIVector vector in the element local x-axis direction vector in the element local y-axis direction getJVector

Parameters . . . continued from previous page

getKVector

vector in the element local z-axis direction

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not revertToLastCommit() the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at revertToStart before the analysis started; returns 0 if successful, a negative number if not getNumDOF returns the number of DOF associated with the element; this should equal the sum of the DOFs at each of the exreturns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation returns cell type for Vtk graphics getVtkCellType() returns cell type for MED file writing. getMEDCellType() getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape resetTributarias() reset the tributary length, area and volume of connected nodes vuelcaTributarias calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the Node given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the Node given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMaxCooNod(axisIdx) returns the maximum value among the coordinates in the axisIdx axis of the external nodes associated with the element (axisIdx adopts the values 0,1,...)

getMinCooNod(axisIdx) returns the minimum value among the coordinates in the axisIdx axis of the external nodes associated with the element (axisIdx adopts the values 0,1,...)

setupVectors(x,yp)

establish orientation of element for the transformation ma-

x vector in global coordinates defining local x-axis

yp vector in global coordinates which lies in the local x-y plane for the element

local z-axis is defined by the vector $\vec{z} = \vec{x} \times y\vec{p}$

getSection()

returns the section axial force-deformation associated with the element

getMaterial() returns the section axial force-deformation associated with the element

4.1.3 ZeroLengthContact2D, ZeroLengthContact3D

These classes are used to constructs a zeroLengthContact2D element or a zeroLengthContact3D element, which are Node-to-node frictional contact element used in two dimensional analysis and three dimensional analysis.

The contact element is node-to-node contact. Contact occurs between two contact nodes when they come close. The relation follows Mohr-coulomb law: $T = \mu \cdot N + c$, where T is tangential force and N is normal force across the interface; μ is friction coefficient and c is total cohesion (summed over the effective area of contact nodes).

The contact node pair in node-to-node contact element is termed \blacksquare master node \blacksquare and \blacksquare slave node \blacksquare , respectively. Master/slave plane is the contact plane which the master/slave node belongs to. The discrimination is made solely for contact detection purpose. User need to specify the corresponding out normal of the master plane, and this direction is assumed to be unchanged during analysis. For simplicity, 3D contact only allows 3 options to specify the directions of the contact plane. The convention is: out normal of master plane always points to positive axial direction (+X or +Y, or +Z)

For 2D contact, slave nodes and master nodes must be 2 DOF. For 3D contact, slave nodes and master nodes must be 3 DOF.

The resulted tangent from the contact element is non-symmetric. Switch to non-symmetric matrix solver.

```
preprocessor=xc.ProblemaEF().getModelador
ZeroLengthElement=preprocessor.getElementLoader.newElement(
"zero_length_contact_2d",xc.ID([Nd1Tag,Nd2Tag]))
"zero_length_contact_3d",xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag tags of master and slave nodes

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics
getDimension	element dimension
getVtkCellType	cell type for Vtk graphics
getIVector	vector in the element local x-axis direction
getJVector	vector in the element local y-axis direction
getKVector	vector in the element local z-axis direction

Parameters ... continued from previous page

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not revertToLastCommit() the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at revertToStart before the analysis started; returns 0 if successful, a negative number if not getNumDOF returns the number of DOF associated with the element: this should equal the sum of the DOFs at each of the external nodes returns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation getVtkCellType() returns cell type for Vtk graphics getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry shape geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape resetTributarias() reset the tributary length, area and volume of connected nodes vuelcaTributarias calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the ${\tt Node}$ given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

setupVectors(x,yp) establish orientation of element for the transformation matrix

x vector in global coordinates defining local x-axis yp vector in global coordinates which lies in the local x-y plane for the element local z-axis is defined by the vector $\vec{z} = \vec{x} \times y\vec{p}$

4.2 Truss elements

4.2.1 Truss

This class is used to constructs a truss element object defined by two nodes connected by means of a previously defined uniaxial material. The truss element does not include geometric nonlinearities, even when used with beam-columns utilizing P-Delta or Corotational transformations. The truss element considers strain-rate effects, and is thus suitable for use as a damping element.

preprocessor=xc.ProblemaEF().getModelador
trussElement=preprocessor.getElementLoader.newElement(
"truss",xc.ID([Nd1Tag,Nd2Tag]))

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes vector containing the node index to be used in Vtk graphics
getDimension element dimension
getVtkCellType cell type for Vtk graphics
getCoordTransf() returns the identifier of the coordinate-transformation associated

with the element

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not revertToLastCommit() the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not revertToStart the element is to set its current state to the state it was at before the analysis started; returns 0 if successful, a negative number if not getNumDOF returns the number of DOF associated with the element; this should equal the sum of the DOFs at each of the external nodes returns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation

getVtkCellType() returns cell type for Vtk graphics
getMEDCellType() returns cell type for MED file writing.
getPosCentroid(geomInicial)

returns the element centroid position.

geomInicial = True to consider the initial geometry
shape

geomInicial = False to consider the deformed geometry
shape

getCooCentroid(geomInicial)

returns the element centroid coordinates

geomInicial = True to consider the initial geometry
shape

geomInicial = False to consider the deformed geometry
shape

getPoints(ni,nj,nk,geomInicial)

returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions

getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape

resetTributarias()

reset the tributary length, area and volume of connected nodes

vuelcaTributarias

calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the ${\tt Node}$ given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter <code>geomInicial = True</code> to consider the initial geometry shape in the calculation or <code>geomInicial = False</code> for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the ${\tt Node}$ given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the Node given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

${\tt strainLoad}$ (PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

getCooPuntos(ndiv) returns ndi

returns ndiv - 1 equally-spaced points on the element

getDim() returns element dimension

getMaterial() returns the material associated with the element

 ${\tt getL} \hspace{1cm} {\tt returns} \ {\tt the} \ {\tt element} \ {\tt length}$

area cross-sectional area of the element

getN

Returns the internal axial force N in the element

4.2.2 **TrussSection**

This class is used to constructs a truss element object defined by two nodes connected by means of a previously defined section.

preprocessor=xc.ProblemaEF().getModelador trussSectionElement=preprocessor.getElementLoader.newElement("truss_section",xc.ID([Nd1Tag,Nd2Tag]))

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

vector containing the node index to be used in Vtk graphics getIdxNodes element dimension getDimension cell type for Vtk graphics getVtkCellType getCoordTransf()

returns the identifier of the coordinate-transformation associated

with the element

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not

revertToLastCommit() the element is to set its current state to the last committed

state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at

revertToStart before the analysis started; returns 0 if successful, a nega-

tive number if not

getNumDOF returns the number of DOF associated with the element;

this should equal the sum of the DOFs at each of the ex-

ternal nodes

Methods ... continued from previous page getResistingForce() returns the resisting force vector for the element; this is equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRFassigns Stress Reduction Factor for element deactivation getVtkCellType() returns cell type for Vtk graphics getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape reset the tributary length, area and volume of connected resetTributarias() nodes vuelcaTributarias calculaLongsTributarias(geomInicial) returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape getLongTributaria(Node) returns the tributary length associated with the Node given

as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome
vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

 $\begin{array}{lll} \verb|getCooPuntos(ndiv)| & \text{returns } ndiv-1 \text{ equally-spaced points on the element} \\ \verb|getDim()| & \text{returns element dimension} \\ \verb|getMaterial()| & \text{returns the material associated with the element} \\ \verb|getL| & \text{returns the element length} \\ \end{array}$

4.2.3 CorotTruss

This class is used to constructs a corotational truss element object defined by two nodes connected by means of a previously defined uniaxial material.

When constructed with a UniaxialMaterial object, the corotational truss element considers strain-rate effects, and is thus suitable for use as a damping element.

```
preprocessor=xc.ProblemaEF().getModelador
corotTrussElement=preprocessor.getElementLoader.newElement(
"corot_truss",xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics
getDimension	element dimension
getVtkCellType	cell type for Vtk graphics
<pre>getCoordTransf()</pre>	returns the identifier of the coordinate-transformation associated
	with the element
area	cross-sectional area of the element

Methods

commitState()	the element is to commit its current state; returns 0 if suc-
	cessful, a negative number if not
<pre>revertToLastCommit()</pre>	the element is to set its current state to the last committed
	state; returns 0 if successful, a negative number if not
revertToStart	the element is to set its current state to the state it was at
	before the analysis started; returns 0 if successful, a nega-
	tive number if not

returns the number of DOF associated with the element; getNumDOF this should equal the sum of the DOFs at each of the external nodes returns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation getVtkCellType() returns cell type for Vtk graphics getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry shape geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape resetTributarias() reset the tributary length, area and volume of connected nodes vuelcaTributarias calculaLongsTributarias(geomInicial) returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape getLongTributaria(Node) returns the tributary length associated with the Node given as argument getLongTributariaByTag(tag) returns the tributary length associated with the node labelled with the tag given as argument calculaAreasTributarias(geomInicial) returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1, PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

<pre>getCooPuntos(ndiv)</pre>	returns $ndiv - 1$ equally-spaced points on the element
<pre>getDim()</pre>	returns element dimension
<pre>getMaterial()</pre>	returns the material associated with the element
getN	Returns the internal axial force N in the element

4.2.4 CorotTrussSection

This class is used to constructs a corotational truss element object defined by two nodes connected by means of a previously defined section.

preprocessor=xc.ProblemaEF().getModelador
corotTrussSectionElement=preprocessor.getElementLoader.newElement(
"corot_truss_section",xc.ID([Nd1Tag,Nd2Tag]))

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics
getDimension	element dimension
getVtkCellType	cell type for Vtk graphics
<pre>getCoordTransf()</pre>	returns the identifier of the coordinate-transformation associated
	with the element

Methods

commitState()	the element is to commit its current state; returns 0 if successful, a negative number if not
<pre>revertToLastCommit()</pre>	the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not
revertToStart	the element is to set its current state to the state it was at before the analysis started; returns 0 if successful, a nega-
•	tive number if not

returns the number of DOF associated with the element; getNumDOF this should equal the sum of the DOFs at each of the external nodes returns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation getVtkCellType() returns cell type for Vtk graphics getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry shape geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape resetTributarias() reset the tributary length, area and volume of connected nodes vuelcaTributarias calculaLongsTributarias(geomInicial) returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape getLongTributaria(Node) returns the tributary length associated with the Node given as argument getLongTributariaByTag(tag) returns the tributary length associated with the node labelled with the tag given as argument calculaAreasTributarias(geomInicial) returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the ${\tt Node}$ given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome vector2dUniformLoadGlobal(v)

obal(v)
applies a uniform surface load to the element; the value and

direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1, PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

 $\begin{array}{lll} \verb|getCooPuntos(ndiv)| & \text{returns } ndiv-1 \text{ equally-spaced points on the element} \\ \verb|getDim()| & \text{returns element dimension} \\ \verb|getMaterial()| & \text{returns the material associated with the element} \\ \end{array}$

4.3 Beam-column elements

4.3.1 ElasticBeam2d

This class is used to constructs a uniaxial element with tension, compression, and bending capabilities. The element has three degrees of freedom at each node: translations in the nodal x and y directions and rotation about the nodal z-axis.

The element is defined by two 2D nodes, a previously-defined coordinate-transformationt object and a previously-defined 2D elastic section. The initial strain in the element (initialStrain) is given by Δ/L , where Δ is the difference between the element length, L (as defined by the 1 and 2 node locations), and the zero strain length.

```
preprocessor=xc.ProblemaEF().getModelador
elasticBeam2dElement=preprocessor.getElementLoader.newElement(
"elastic_beam_2d",xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics				
getDimension	element dimension				
getVtkCellType	cell type for Vtk graphics				
<pre>getCoordTransf()</pre>	returns the identifier of the coordinate-transformation associated				
	with the element				
sectionProperties	· ·				
rho	mass density				
h	overall depth				
initialStrain	initial strain in the element				
getV	returns the shear force at the central section of the element				
getV1 returns the shear force at the back end of the element					
getV2	returns the shear force at the front end of the element				
getN1	returns the axial force at the back end of the element				
getN2	returns the axial force at the front end of the element				
getM1 returns the bending moment at the back end of the el					

continued on next page ...

Parameters . . . continued from previous page

getM2

returns the bending moment at the front end of the element

Methods

commitState()	the element is to commit its current state; returns 0 if successful, a negative number if not					
revertToLastCommit()	the element is to set its current state to the last committed					
100010102020000000000000000000000000000	state; returns 0 if successful, a negative number if not					
revertToStart	the element is to set its current state to the state it was at					
1000101020410	before the analysis started; returns 0 if successful, a nega-					
	tive number if not					
getNumDOF	returns the number of DOF associated with the element;					
ge on ambor	this should equal the sum of the DOFs at each of the ex					
	ternal nodes					
motDogistingEorgo()						
<pre>getResistingForce()</pre>	returns the resisting force vector for the element; this is					
	equal to the applied load due to element loads minus the					
	loads at the nodes due to internal stresses in the element due					
. D. 1007	to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$					
setDeadSRF	assigns Stress Reduction Factor for element deactivation					
<pre>getVtkCellType()</pre>	returns cell type for Vtk graphics					
<pre>getMEDCellType()</pre>	returns cell type for MED file writing.					
getPosCentroid(geomI						
	returns the element centroid position.					
	geomInicial = True to consider the initial geometry					
	shape					
	<pre>geomInicial = False to consider the deformed geometry</pre>					
	shape					
getCooCentroid(geomI						
	returns the element centroid coordinates					
	<pre>geomInicial = True to consider the initial geometry</pre>					
	shape					
	<pre>geomInicial = False to consider the deformed geometry</pre>					
	shape					
<pre>getPoints(ni,nj,nk,g</pre>	eomInicial)					
	returns a uniform grid of points over the element.					
	ni,nj,nk number of divisions in i,j,k directions					
	getLongTributaria True to consider the initial geometry					
	shape, False for the deformed geometry shape					
<pre>resetTributarias()</pre>	reset the tributary length, area and volume of connected					
	nodes					
vuelcaTributarias						
calculaLongsTributar	ias(geomInicial)					

continued on next page . . .

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the Node given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the Node given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

getCooPuntos(ndiv)

returns ndiv - 1 equally-spaced points on the element

4.3.2 ElasticBeam3d

This class is used to constructs a uniaxial element with tension, compression, torsion, and bending capabilities. The element has six degrees of freedom at each node: translations in the nodal x, y, and z directions and rotations about the nodal x, y, and z axes.

The element is defined by two 3D nodes, a previously-defined coordinate-transformationt object and a previously-defined 3D elastic section. The element x-axis is oriented from node I toward node J. The initial strain in the element (initialStrain) is given by Δ/L , where Δ is the difference between the element length, L (as defined by the 1 and 2 node locations), and the zero strain length.

```
preprocessor=xc.ProblemaEF().getModelador
elasticBeam2dElement=preprocessor.getElementLoader.newElement(
"elastic_beam_3d",xc.ID([Nd1Tag,Nd2Tag]))
```

Nd1Tag, Nd2Tag

tags of the nodes connected by the element

Parameters

getIdxNodes	vector containing the node index to be used in Vtk graphics					
getDimension	element dimension					
getVtkCellType	cell type for Vtk graphics					
<pre>getCoordTransf()</pre>	returns the identifier of the coordinate-transformation associated					
	with the element					
sectionProperties						
rho	mass density					
initialStrain	initial strain in the element					
getAN2	returns the axial force applied to the front end of the element					
getN1	returns the axial force at the back end of the element					
getN2	returns the axial force at the front end of the element					
getN	returns the mean value of the axial force at the element $N =$					
G	$N_1 + N_2$					
	2					
getAMz1	returns the bending moment about the local Z axis applied to the					
	back end of the element					
getAMz2	returns the bending moment about the local Z axis applied to the					
	front end of the element					
getMz1	returns the bending moment about the local Z axis at the back					
	end of the element					
getMz2	returns the bending moment about the local Z axis at the front					
	end of the element					
getMy1	returns the bending moment about the local Y axis at the back					
	end of the element					
getMy2	returns the bending moment about the local Y axis at the front					
O V	end of the element					
getVy	returns the element mean shear force in the local Y axis direction					
getVy1	returns the shear force in local Y axis direction at the back end					
8	of the element					
getVy2	returns the shear force in local Y axis direction at the front end					
87-	of the element					
getAVy1	returns the shear force in local Y axis applied on the back end					
5001171	of the element					
getAVy2	returns the shear force in local Y axis direction applied on the					
gethvyz	front end of the element					
getVz	returns the element mean shear force in the local Z axis direction					
	returns the element mean shear force in the local Z axis direction at the back end of					
getVz1						
+-	the element					
getVz2	returns the shear force in local Z axis direction at the front end					
1.477.4	of the element					
getAVz1	returns the shear force in local Z axis applied on the back end of					
	the element					
getAVz2	returns the shear force in local Z axis direction applied on the					
	front end of the element					

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not revertToLastCommit() the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at revertToStart before the analysis started; returns 0 if successful, a negative number if not getNumDOF returns the number of DOF associated with the element; this should equal the sum of the DOFs at each of the external nodes returns the resisting force vector for the element; this is getResistingForce() equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation returns cell type for Vtk graphics getVtkCellType() returns cell type for MED file writing. getMEDCellType() getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getCooCentroid(geomInicial) returns the element centroid coordinates geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry shape getPoints(ni,nj,nk,geomInicial) returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions getLongTributaria True to consider the initial geometry shape, False for the deformed geometry shape resetTributarias() reset the tributary length, area and volume of connected nodes vuelcaTributarias calculaLongsTributarias(geomInicial) returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape getLongTributaria(Node) returns the tributary length associated with the Node given

as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the ${\tt Node}$ given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

getMEDCellType() interface with MED data format for Salome vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

continued on next page ...

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end.

getCooPuntos(ndiv)
 getVDirEjeFuerteLocales()

returns ndiv-1 equally-spaced points on the element

returns a vector, expressed in local coordinates, that defines the strong axis orientation

getVDirEjeDebilLocales()

returns a vector, expressed in local coordinates, that defines the weak axis orientation

getAnguloEjeFuerte() returns the angle between the strong axis and the plane XZ of the element

getAnguloEjeDebil() returns the angle between the weak axis and the plane XZ of the element

getVDirEjeFuerteGlobales()

returns a vector, expressed in global coordinates, that defines the strong axis orientation

getVDirEjeDebilGlobales()

returns a vector, expressed in global coordinates, that defines the weak axis orientation

4.3.3 ForceBeamColumn2d

This command is used to constructs a 2D forceBeamColumn element object, which is based on the iterative force-based formulation. A variety of numerical integration options can be used in the element state determination and encompass both distributed plasticity and plastic hinge integration. More details on the available numerical integration options can be found in the paper titled "Numerical Integration Options for the Force-Based Beam-Column Element in OpenSees", by Michael H. Scott.

The element is defined by two 2D nodes, a previously-defined coordinate-transformation object and a previously-defined 2D elastic section.

```
preprocessor=xc.ProblemaEF().getModelador
elasticBeam2dElement=preprocessor.getElementLoader.newElement()
```

"force_beam_column_2d",xc.ID([Nd1Tag,Nd2Tag]))

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes vector containing the node index to be used in Vtk graphics

getDimension element dimension getVtkCellType cell type for Vtk graphics

getCoordTransf() returns the identifier of the coordinate-transformation associated

with the element

rho mass density

Methods

commitState() the element is to commit its current state; returns 0 if suc-

cessful, a negative number if not

revertToLastCommit() the element is to set its current state to the last committed

state; returns 0 if successful, a negative number if not

revertToStart the element is to set its current state to the state it was at

before the analysis started; returns 0 if successful, a nega-

tive number if not

 ${\tt getNumDOF} \qquad \qquad {\tt returns} \ \ {\tt the} \ \ {\tt number} \ \ {\tt of} \ \ {\tt DOF} \ \ {\tt associated} \ \ {\tt with} \ \ {\tt the} \ \ {\tt element};$

this should equal the sum of the DOFs at each of the ex-

ternal nodes

getResistingForce() returns the resisting force vector for the element; this is

equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ assigns Stress Reduction Factor for element deactivation

setDeadSRF assigns Stress Reduction Factor for elementary elements cell type for Vtk graphics

getMEDCellType() returns cell type for MED file writing.

getPosCentroid(geomInicial)

returns the element centroid position.

geomInicial = True to consider the initial geometry

shape

 ${\tt geomInicial = False}\ to\ consider\ the\ deformed\ geometry$

 $_{\mathrm{shape}}$

getCooCentroid(geomInicial)

returns the element centroid coordinates

geomInicial = True to consider the initial geometry
shape

geomInicial = False to consider the deformed geometry
shape

getPoints(ni,nj,nk,geomInicial)

returns a uniform grid of points over the element. ni,nj,nk number of divisions in i,j,k directions

getLongTributaria *True* to consider the initial geometry shape, *False* for the deformed geometry shape

resetTributarias()

reset the tributary length, area and volume of connected nodes

vuelcaTributarias

calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the ${\tt Node}$ given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the ${\tt Node}$ given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

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getMEDCellType() interface with MED data format for Salome
vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end. returns ndiv - 1 equally-spaced points on the element

getCooPuntos(ndiv)
 getNumSections

getSections Returns element sections

4.3.4 ForceBeamColumn3d

This command is used to constructs a 3D forceBeamColumn element object, which is based on the iterative force-based formulation. A variety of numerical integration options can be used in the element state determination and encompass both distributed plasticity and plastic hinge integration. More details on the available numerical integration options can be found in the paper titled Numerical Integration Options for the Force-Based Beam-Column Element in OpenSees, by Michael H. Scott.

The element is defined by two 3D nodes, a previously-defined coordinate-transformationt object and a previously-defined 3D elastic section.

preprocessor=xc.ProblemaEF().getModelador
elasticBeam2dElement=preprocessor.getElementLoader.newElement(
"force_beam_column_3d",xc.ID([Nd1Tag,Nd2Tag]))

Nd1Tag, Nd2Tag tags of the nodes connected by the element

Parameters

getIdxNodes vector containing the node index to be used in Vtk graphics

getDimension element dimension getVtkCellType cell type for Vtk graphics

getCoordTransf() returns the identifier of the coordinate-transformation associated

with the element

rho mass density

Methods

commitState() the element is to commit its current state; returns 0 if successful, a negative number if not revertToLastCommit() the element is to set its current state to the last committed state; returns 0 if successful, a negative number if not the element is to set its current state to the state it was at revertToStart before the analysis started; returns 0 if successful, a negative number if not getNumDOF returns the number of DOF associated with the element; this should equal the sum of the DOFs at each of the external nodes getResistingForce() returns the resisting force vector for the element; this is equal to the applied load due to element loads minus the loads at the nodes due to internal stresses in the element due to the current trial displacement, i.e. $R_e = P_e - f_{R_e}(U_{trial})$ setDeadSRF assigns Stress Reduction Factor for element deactivation getVtkCellType() returns cell type for Vtk graphics getMEDCellType() returns cell type for MED file writing. getPosCentroid(geomInicial) returns the element centroid position. geomInicial = True to consider the initial geometry geomInicial = False to consider the deformed geometry getCooCentroid(geomInicial)

returns the element centroid coordinates

geomInicial = True to consider the initial geometry
shape

geomInicial = False to consider the deformed geometry
shape

getPoints(ni,nj,nk,geomInicial)

returns a uniform grid of points over the element.

ni,nj,nk number of divisions in i,j,k directions
getLongTributaria True to consider the initial geometry

shape, False for the deformed geometry shape

resetTributarias()

reset the tributary length, area and volume of connected nodes

vuelcaTributarias

calculaLongsTributarias(geomInicial)

returns the tributary length associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getLongTributaria(Node)

returns the tributary length associated with the ${\tt Node}$ given as argument

getLongTributariaByTag(tag)

returns the tributary length associated with the node labelled with the tag given as argument

calculaAreasTributarias(geomInicial)

returns the tributary area associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getAreaTributaria(Node)

returns the tributary area associated with the ${\tt Node}$ given as argument

getAreaTributariaByTag(tag)

returns the tributary area associated with the node labelled with the tag given as argument

calculaVolsTributarios(geomInicial)

returns the tributary volume associated with each node of the element; the parameter geomInicial = True to consider the initial geometry shape in the calculation or geomInicial = False for the deformed geometry shape

getVolTributario(Node)

returns the tributary volume associated with the ${\tt Node}$ given as argument

getVolTributarioByTag(tag)

returns the tributary volume associated with the node labelled with the tag given as argument

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getMEDCellType() interface with MED data format for Salome
vector2dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the global coordinate system

vector2dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 2D vector \vec{v} , expressed in the element local coordinate system

vector2dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector2dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 2D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local system

vector2dPointLoadGlobal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the global coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation (in global coordinates)

vector2dPointLoadLocal(p,v)

applies a punctual force to the element; 2D vector \vec{p} defines the coordinates of the point of application of the force; 2D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

vector3dUniformLoadGlobal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the global coordinate system

vector3dUniformLoadLocal(v)

applies a uniform surface load to the element; the value and direction of the load is defined by the 3D vector \vec{v} , expressed in the element local coordinate system

vector3dPointByRelDistLoadGlobal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the global system

vector3dPointByRelDistLoadLocal(d,v)

applies a punctual force to the element; scalar d specifies the offset distance from node 2 (toward node 1) where the force is applied, this distance is input as a length fraction (its value varies between 0 and 1); 3D vector \vec{v} defines the force value and orientation, its coordinates are expressed in the element local-coordinate system

vector3dPointLoadGlobal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the global coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation (in global coordinates)

vector3dPointLoadLocal(p,v)

applies a punctual force to the element; 3D vector \vec{p} defines the coordinates of the point of application of the force; 3D vector \vec{v} defines the force value and orientation; both vectors are expressed in the element local-coordinate system

strainLoad(PlanoDeformacion1,PlanoDeformacion2)

This function creates an imposed strain load in the current load case. The first argument defines the deformation at element start and the second at the element's end. returns ndiv - 1 equally-spaced points on the element

getCooPuntos(ndiv)
getNumSections

getSections Returns element sections

getVDirEjeFuerte-Locales()

> returns a vector, expressed in local coordinates, that defines the strong axis orientation

getVDirEjeDebilLocales()

returns a vector, expressed in local coordinates, that defines the weak axis orientation

getAnguloEjeFuerte() returns the angle between the strong axis and the plane XZ

of the element

getAnguloEjeDebil() returns the angle between the weak axis and the plane XZ of the element

continued on next page ...

getVDirEjeFuerteGlobales()

returns a vector, expressed in global coordinates, that defines the strong axis orientation

getVDirEjeDebilGlobales()

returns a vector, expressed in global coordinates, that defines the weak axis orientation

4.3.5 Numerical integration options for the forceBeamColumn elements.

The following paragraph are based on the Michael H. Scott article in OpenSees.

To specify the numerical integration options to represent distributed plasticity or non-prismatic section details in force-based beam-column elements, i.e., across the entire element domain [0,L] we can use one of the following schemes.



Chapter 5

Loads

5.1 Time series

Time Series objects represents the relationship between the time in the domain, t, and the load factor applied to the loads, λ , in the load pattern with which the Time Series object is associated, i.e.

$$\lambda = F(t) \tag{5.1}$$

- 5.1.1 Constant TimeSeries
- 5.1.2 Linear TimeSeries
- 5.1.3 Trigonometric TimeSeries
- 5.1.4 Triangular TimeSeries
- 5.1.5 Rectangular TimeSeries
- 5.1.6 Pulse TimeSeries
- 5.1.7 Path TimeSeries
- 5.1.8 PeerMotion
- 5.1.9 PeerNGAMotion



Chapter 6

Solution

6.1 Analysis and its components

In XC (just like in OpenSees) an analysis is an object which is composed by the aggregation of component objects. It is the component objects which define the type of analysis that is performed on the model. The component classes consist of the following:

Constraint Handler : determines how the constraint equations are enforced in the analysis : how it handles the boundary conditions/imposed displacements

DOF_Numberer: determines the mapping between equation numbers and degrees-of-freedom

Integrator: determines the predictive step for time t+dt

SolutionAlgorithm : determines the sequence of steps taken to solve the non-linear equation at the current time step

SystemOfEqn/Solver: within the solution algorithm, it specifies how to store and solve the system of equations in the analysis

Convergence Test: determines when convergence has been achieved.

6.1.1 Constraint handlers

The ConstraintHandler objects determine how the constraint equations are enforced in the analysis. Constraint equations enforce a specified value for a DOF, or a relationship between DOFs.

6.1.1.1 Constraint types

The types of constraints supported by the software are defined in 2.3.2.3 and 2.3.2.4.

6.1.1.2 Constraint handler types

The available constraint handler types are the following:

- Plain Constraints
- Lagrange Multipliers

- Penalty Method
- Transformation Method

6.1.1.3 Plain Constraints

A plain constraint handler can only enforce homogeneous single point constraints (fix command) and multi-point constraints constructed where the constraint matrix is equal to the identity (equalDOF command).

6.1.1.4 Lagrange multipliers

A Lagrange multiplier constraint handler enforces the constraints by introducing Lagrange multipliers to the system of equations.

In an object of this type the following parameters can be defined:

 α_S : factor on single point, optional, default = 1.0

 α_M : factor on multi-points, optional default = 1.0

The Lagrange multiplier method introduces new unknowns to the system of equations. The diagonal part of the system corresponding to these new unknowns is 0.0. This ensure that the system IS NOT symmetric positive definite.

6.1.1.4.1 LagrangeMP_FE LagrangeMP_FE is a subclass of FE_Element used to enforce a multi point constraint, of the form $U_c = C_{cr}U_r$, where U_c are the constrained degrees-of-freedom at the constrained node, U_r are the retained degrees-of-freedom at the retained node and C_{cr} a matrix defining the relationship between these degrees-of-freedom.

To enforce the constraint the following are added to the tangent and the residual:

$$\left[\begin{array}{cc} 0 & \alpha C^t \\ \alpha C & 0 \end{array}\right], \left\{\begin{array}{c} 0 \\ 0 \end{array}\right\}$$

at the locations corresponding to the constrained degree-of-freedoms specified by the MP_Constraint, i.e. $[U_c \ U_r]$, and the lagrange multiplier degrees-of-freedom introduced by the LagrangeConstraintHandler for this constraint, $C = [-I \ C_{cr}]$. Nothing is added to the residual.

To constructs a LagrangeMP_FE element to enforce the constraint specified by the MP_Constraint theMP using a default value for α of alpha. The FE_Element class constructor is called with the integers 3 and the two times the size of the retainedID plus the size of the constrainedID at the MP_Constraint theMP plus . A Matrix and a Vector object are created for adding the contributions to the tangent and the residual. The residual is zeroed. If the MP_Constraint is not time varying, then the contribution to the tangent is determined. Links are set to the retained and constrained nodes. The DOF_Group tag ID is set using the tag of the constrained Nodes DOF_Group, the tag of the retained Node Dof_group and the tag of the LagrangeDOF_Group, theGroup. A warning message is printed and the program is terminated if either not enough memory is available for the Matrices and Vector or the constrained and retained Nodes of their DOF_Groups do not exist.

virtual void setID(void);

Causes the LagrangeMP_FE to determine the mapping between it's equation numbers and the

degrees-of-freedom. This information is obtained by using the mapping information at the DOF_Group objects associated with the constrained and retained nodes and the LagrangeDOF_Group, the Group. Returns 0 if successful. Prints a warning message and returns a negative number if an error occurs: -2 if the Node has no associated DOF_Group, -3 if the constrained DOF specified is invalid for this Node (sets corresponding ID component to -1 so nothing is added to the tangent) and -4 if the ID in the DOF_Group is too small for the Node (again setting corresponding ID component to -1).

virtual const Matrix &getTangent(Integrator *theIntegrator);

If the MP_Constraint is time-varying, from the MP_Constraint theMP it obtains the current C_{cr} matrix; it then adds the contribution to the tangent matrix. Returns this tangent Matrix.

 $virtual\ const\ \ Vector\ \mathscr{E}getResidual(Integrator\ ^*theIntegrator);$

Returns the residual, a zero Vector.

6.1.1.5 Penalty method

A penalty constraint handler enforces the constraints using the penalty method. These handlers allows the user to choose the penalty factors:

 α_S : penalty factor on single point constraints

 α_M : penalty factor on multimple point constraints

The degree to which the constraints are enforced is dependent on the penalty values chosen. Problems can arise if these values are too small (constraint not enforced strongly enough) or too large (problems associated with conditioning of the system of equations).

6.1.1.6 Transformation method

A transformation constraint handler enforces the constraints using the transformation method.

The single-point constraints when using the transformation method are done directly. The matrix equation is not manipulated to enforce them, rather the trial displacements are set directly at the nodes at the start of each analysis step.

Great care must be taken when multiple constraints are being enforced as the transformation method does not follow constraints:

- 1. If a node is fixed, constrain it with the fix command and not equalDOF or other type of constraint.
- 2. If multiple nodes are constrained, make sure that the retained node is not constrained in any other constraint.

And remember if a node is constrained to multiple nodes in your model it probably means you have messed up.

6.1.2 DOF_Numberer: mapping between equation numbers and degrees of freedom

The DOF_Numberer object determines the mapping between equation numbers and degrees of freedom i.e. how degrees of freedom are numbered.

6.1.2.1 Plain Numberers

WIP Work in progress...

6.1.2.2 Reverse Cuthill-McKee Numberers

WIP Work in progress...

6.1.3 System of equation and its solution

This objects are used to construct the LinearSOE and LinearSolver objects to store and solve the system of equations in the analysis.

6.1.3.1 Band general system of equations

BandGeneralSOE WIP Work in progress...

6.1.3.2 Band symmetric positive definite system of equations

BandSPDSOE WIP Work in progress...

6.1.3.3 Profile symmetric positive definite system of equations

ProfileSPDSOE WIP Work in progress...

6.1.3.4 Sparse general linear system of equations (SuperLU)

SuperLUSOE WIP Work in progress...

6.1.3.5 Sparse general linear system of equations (UmfPack)

UmfPackSOE WIP Work in progress...

6.1.3.6 Full general linear system of equations

UmfPackSOE WIP Work in progress...

6.1.3.7 Sparse symmetric system of equations

SparseSYM WIP Work in progress...

6.1.4 Integrator

The Integrator object determines the meaning of the terms in the system of equation object Ax=B. The Integrator object is used for the following:

- determine the predictive step for time t+dt
- specify the tangent matrix and residual vector at any iteration
- determine the corrective step based on the displacement increment dU

- 6.1.4.1 Static integrators
- **6.1.4.1.1 Load Control** WIP Work in progress...
- **6.1.4.1.2 Displacement Control** WIP Work in progress...
- **6.1.4.1.3 Minimum Unbalanced Displacement Norm** WIP Work in progress...
- 6.1.4.1.4 Arc-Length Control
- 6.1.4.2 Transient integrators
- **6.1.4.2.1 Central Difference** WIP Work in progress...
- 6.1.4.2.2 Newmark Method WIP Work in progress...
- **6.1.4.2.3** Hilber-Hughes-Taylor Method WIP Work in progress...
- **6.1.4.2.4 Generalized Alpha Method** WIP Work in progress...
- 6.1.4.2.5 TRBDF2

6.1.5 Convergence test

The convergence tests are used to allow certain SolutionAlgorithm objects to determine if convergence has been achieved at the end of an iteration step. The convergence test is applied to the matrix equation, AX=B stored in the LinearSOE.

6.1.5.1 Norm Unbalance Test

WIP Work in progress...

6.1.5.2 Norm Displacement Increment Test

WIP Work in progress...

6.1.5.3 Energy Increment Test

WIP Work in progress...

6.1.5.4 Relative Norm Unbalance Test

WIP Work in progress...

6.1.5.5 Relative Norm Displacement Increment Test

WIP Work in progress...

${\bf 6.1.5.6} \quad {\bf Total} \ {\bf Relative} \ {\bf Norm} \ {\bf Displacement} \ {\bf Increment} \ {\bf Test}$

WIP Work in progress...

6.1.5.7 Relative Energy Increment Test

WIP Work in progress...

6.1.5.8 Fixed Number of Iterations

WIP Work in progress...

6.1.6 Solution algorithm

A SolutionAlgorithm object is used to determine the sequence of steps taken to solve the non-linear equation.

6.1.6.1 Linear Algorithm

WIP Work in progress...

6.1.6.2 Newton Algorithm

WIP Work in progress...

6.1.6.3 Newton with Line Search Algorithm

WIP Work in progress...

6.1.6.4 Modified Newton Algorithm

WIP Work in progress...

6.1.6.5 Krylov-Newton Algorithm

WIP Work in progress...

6.1.6.6 Secant Newton Algorithm

WIP Work in progress...

6.1.6.7 BFGS Algorithm

WIP Work in progress...

6.1.6.8 Broyden Algorithm

WIP Work in progress...

6.1.7 Analyze method

All analysis objects have an 'analyze' method that is used to perform the analysis. This command can receive one or more of the following parameters:

numSteps: number of analysis steps to perform.

dt: time-step increment. Required if transient or variable transient analysis

 ${\rm dtMin,dtMax:}$ minimum and maximum time steps. Required if a variable time step transient analysis was specified.

Jd: number of iterations user would like performed at each step. The variable transient analysis will change current time step if last analysis step took more or less iterations than this to converge. Required if a variable time step transient analysis was specified.

This command returns a zero if successful or a negative value otherwise.





Chapter 7

Check routines for structural steel

7.1 Introduction

This chapter describes the routines that can be used to check the design following the specifications of different design codes.

7.2 Lateral torsional buckling of steel beams (EC3)

Flexural buckling check, according to article 5.5.2 of EC3, can be written as:

$$F = \frac{M_d}{M_{b,Rd}} \le 1 \tag{7.1}$$

where:

 M_d Design value of bending moment.

 $M_{b,Rd}$ Buckling resistance

7.2.1 Design lateral torsional buckling resistance $M_{b,Rd}$

Design value of lateral torsional buckling resistance can be determined as follows:

$$M_{b,Rd} = \chi_{LT} \cdot W_y \cdot \frac{f_y}{\gamma_{M1}} \tag{7.2}$$

where γ_{M1} is the partial factor for member buckling resistance and f_y is the characteristic value of the yield strength.

7.2.1.1 Cross section modulus W_y

$$W_{y} = \begin{cases} W_{pl,y} & : 1 \text{ or } 2 \text{ class cross section (plastic section modulus)} \\ W_{el,y} & : \text{ class } 3 \text{ cross section (elastic section modulus)} \\ W_{eff,y} & : \text{ class } 4 \text{ cross section (effective section modulus)} \end{cases}$$
(7.3)

7.2.1.2 Reduction factor χ_{LT}

The reduction factor χ_{LT} for the appropriate non-dimensional slenderness λ_{LT} may be determined from:

$$\chi_{LT} = \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \overline{\lambda}_{LT}^2}} \tag{7.4}$$

7.2.1.2.1 Intermediate factor ϕ_{LT} calculated as:

$$\phi_{LT} = 0.5[1 + \alpha_{LT} \cdot (\overline{\lambda}_{LT} - 0.2) + \overline{\lambda}_{LT}^{2}]$$
 (7.5)

7.2.1.2.2 Imperfection factor α_{LT} Depending on the buckling curves ¹:

$$\alpha_{LT} = \begin{cases} 0.21 & : \text{ curve a} \\ 0.34 & : \text{ curve b} \\ 0.49 & : \text{ curve c} \\ 0.76 & : \text{ curve d} \end{cases}$$
 (7.6)

7.2.1.2.3 Non-dimensional beam slenderness $\bar{\lambda}_{LT}$ calculated as:

$$\bar{\lambda}_{LT} = \sqrt{\frac{W_y f_y}{M_{cr}}} \tag{7.7}$$

Where the non-dimensional slenderness $\bar{\lambda}_{LT} \leq 0.4$ no allowance for lateral-torsional buckling is necessary²

7.2.1.2.4 Critical bending moment M_{cr} Following the [2] reference, the elastic critical moment is directly affected by the following factors:

- Material properties (elastic modulus, poisson ratio,...).
- Properties of the cross section (torsion constant, warping constant³ and moment of inertia about the minor axis).
- Properties of the beam such as its length, lateral bending and warping restrictions at supports.
- Loading: lataeral-torsional buckling is greatly affected by moment diagram and loading position with respect to the section shear centre.

As a generally accepted procedure, consideration of the bending moment diagram is taken into account by means of the equivalent uniform moment factor (EUMF), also called the moment gradient correction factor. The elastic critical moment for any bending moment diagram is obtained by multipliying this factor by the elastic critical moment of the simply supported beam with uniform moment.

The expression of the elastic critical moment is the calculated as:

¹Selection of flexural buckling curve for a cross section can be made from EC3 tables.

²This value can be adapted in the national annex.

³The effect of torsional loading can be split into two parts, the first part causing twist and the second, warping.

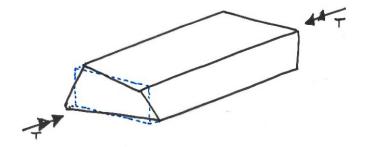


Figure 7.1: Warping on a rectangular section due to pure torsion

$$M_{cr} = C_1 \frac{\pi^2 E I_z}{(k_z L)^2} \sqrt{(\frac{k_z}{k_w})^2 \frac{I_w}{I_z} + \frac{(k_z L)^2 G I_t}{\pi^2 E I_z}}$$
(7.8)

where the lateral bending coefficient k_z and the warping coefficient k_w are introduced to consider support conditions different from the simply supported beam. These coefficients are equal to 1 for free lateral bending and free warping and 0.5 for prevented lateral bending and prevented warping. There rest of the parameters are defined as follows:

E Elastic modulus.

G Shear modulus.

 $k_z L$ Lateral torsional buckling length of the beam: length between points which have lateral restraint.

 I_z Moment of inertia about the minor axis.

 I_t Torsional constant.

 I_w Warping constant.

Finally, the moment moment gradient factor C_1 is calculated as:

$$C_1 = \frac{\sqrt{\sqrt{k}A_1 + \left[\frac{(1-\sqrt{k})}{2}A_2\right]^2 + \frac{(1-\sqrt{k})}{2}A_2}}{A_1}$$
 (7.9)

where:

 $k = \sqrt{k_1 k_2}$ depends on the lateral bending and warping conditions coefficients k_1 and k_2 .

 M_i and M_{max} moments which represent moment diagram of the beam as in figure 7.2. M_{max} is the maximum moment and M_1 , M_2 , M_3 , M_4 and M_5 are the values of the moment at the different sections of the beam as indicated int the figure, each of them with the corresponding sign.

$$A_1 = \frac{M_{max}^2 + \sum\limits_{i=1..5} \alpha_i M_i^2}{(1 + \sum\limits_{i=1..5} \alpha_i) \cdot M_{max}^2}$$

$$A_2 = \frac{M_1 + 2M_2 + 3M_3 + 2M_4 + M_5}{9 \cdot M_{max}}$$

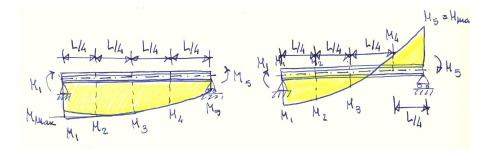


Figure 7.2: Moment diagrams and moment values for calculation of C_1

 α_i factor calculated as follows:

$$\alpha_1 = (1 - k_2); \ \alpha_2 = 5\frac{k_1^3}{k_2^2}; \ \alpha_3 = 5(\frac{1}{k_1} + \frac{1}{k_2}); \ \alpha_4 = 5\frac{k_2^3}{k_1^2}; \ \alpha_5 = (1 - k_1)$$
 (7.10)

 k_1 and k_2 support conditions at each end of the beam as follows:

 $k_1 = 1.0$ Free lateral bending and warping at left end.

 $k_1 = 0.5$ Prevented support conditions at left end.

 $k_2 = 1.0$ Free lateral bending and warping at right end.

 $k_2 = 0.5$ Prevented support conditions at right end.

Chapter 8

Check routines for reinforced concrete

8.1 Introduction

This chapter describes the routines that can be used to check the design following the specifications of different design codes.

8.2 Verification of RC sections

8.2.1 Sections definition

8.2.2 Limit State at Failure under normal stresses verification

$\bf 8.2.2.1 \quad lanza Calculo TNF rom XCD at a$

This function carries out the verification of the limit state at failure under normal stresses. It takes as input the internal forces and bending moments calculated for the shell elements for every ULS combinations, the sections definition and the interaction diagrams to be employed.

The function returns two files with the verification results: outputFileName.py: XC file that assigns each shell element the capacity factor (worst-case) FCC calculated for reinforcement in directions 1 and 2, together with the concomitant axial force and bending moments N My Mz. outputFileName.py: LATEX file containing a table with the following items:

Section 1

Element number	Section name	ULS name	Axial force NCP1	Bending moment MyCP1	Bending moment MzCP1	Capacity factor FCCP1
Section 2						
Element number	Section name	ULS name	Axial force NCP2	Bending moment MyCP2	Bending moment MzCP2	Capacity factor FCCP2

.

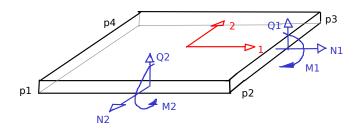


Figure 8.1: Positive directions of forces and moments in shell elements

from materials.xLamina import calculo_tn
calculo_tn.lanzaCalculoTNFromXCData(preprocessor,analysis,intForcCombFileName,outputFileName,
mapSectionsForEveryElement,mapSectionsDefinition, mapInteractionDiagrams)

preprocessor
analysis

preprocessor name

type of analysis to be performed. Some predefined types

can be selected:

 ${\tt from \ solution \ import \ predefined_solutions \ as \ ps}$

pEF=xc.ProblemaEF()

ps.simple_static_linear(pEF) for linear static analysis ps.simple_newton_raphson(pEF) for non linear analysis using Newton Raphson algorithm to solve the equation

ps.simple_newton_raphson_band_gen(pEF)

 ${\tt ps.simple_static_modified_newton(pEF)}$ for non linear analysis using the modified Newton Raphson algorithm to

solve the equation
ps.penalty_newton_raphson(pEF)

ps.frequency_analysis(p)

intForcCombFileName

name of the file where are to be found the shell element forces and moments obtained for each ULS combination, to be used in the verification of the ULS under normal stresses. Directory path, file name and extension must be specified. The file contains a list of the following items (see the positive directions of the forces and moments in figure 8.1): $ULS_{id}\ elem_{id}\ N_1\ N_2\ N_{12}\ M_1M_2\ M_{12}\ Q_1\ Q_2$

name of the output files (directory path and file name with-

out extension) where to write the results of the verification

 ${\tt mapSectionsForEveryElement}$

data structure, such that, for each shell element: mapSectionsForEveryElement[tagElem] = [nmbScc1,nmbScc2], where nmbScc1 and nmbScc2 are the names identifying the sections to be used in 1 and 2 directions respectively

 ${\tt mapSectionsDefinition}$

outputFileName

data structure that links each section name and direction with the respective record that contains the section definition parameters

mapInteractionDiagrams

data structure that links each section name and direction with the interaction diagram object to be used for its verification

8.2.3 Limit State of Failure due to shear verification

This function carries out the verification of the limit state at failure under normal stresses. It takes as input the internal forces and bending moments calculated for the shell elements for every ULS combinations, the sections definition and the interaction diagrams to be employed.

The function returns two files with the verification results:

outputFileName.py: XC file that assigns each shell element the capacity factor (worst-case) FCC calculated for reinforcement in directions 1 and 2, together with the concomitant forces and moments N My Mz Vy Vz and the ultimate shear forces and moment Mu Vcu Vsu Vu outputFileName.py: LATEX file containing a table with the following items:

Section 1										
Element	Section	ULS	Axial	Bending	Bending	Cracking	Shear	Shear	Ultimate	Capacity
number	name	name	$_{ m force}$	moment	moment	moment	force	force	$_{ m shear}$	factor
			NCP1	MyCP1	MzCP1	MuCP1	VyCP1	VyCP1	force VuCP1	FCCP1
	•									
Section 2										
Element	Section	ULS	Axial	Bending	Bending	Cracking	Shear	Shear	Ultimate	Capacity
number	name	name	force	moment	moment	moment	force	force	shear	factor
			NCP2	MyCP2	MzCP2	MuCP2	VyCP2	VyCP2	force VuCP2	FCCP2

from materials.xLamina import calculo_v
calculo_v.lanzaCalculoV(preprocessor,analysis,intForcCombFileName,outputFileName,
mapSectionsForEveryElement,mapSectionsDefinition,mapInteractionDiagrams,
procesResultVerifV)

preprocessor
analysis

preprocessor name

type of analysis to be performed. Some predefined types can be selected:

from solution import predefined_solutions as ps
pEF=xc.ProblemaEF()

ps.simple_static_linear(pEF) for linear static analysis ps.simple_newton_raphson(pEF) for non linear analysis using Newton Raphson algorithm to solve the equation ps.simple_newton_raphson_band_gen(pEF)

ps.simple_static_modified_newton(pEF) for non linear analysis using the modified Newton Raphson algorithm to solve the equation

ps.penalty_newton_raphson(pEF) ps.frequency_analysis(p)

intForcCombFileName

name of the file where are to be found the shell element forces and moments obtained for each ULS combination, to be used in the verification of the ULS due to shear. Directory path, file name and extension must be specified. The file contains a list of the following items (see the positive directions of the forces and moments in figure 8.1): $ULS_{id} \ elem_{id} \ N_1 \ N_2 \ N_{12} \ M_1M_2 \ M_{12} \ Q_1 \ Q_2$

outputFileName

name of the output files (directory path and file name without extension) where to write the results of the verification

mapSectionsForEveryElement

data structure, such that, for each shell element: mapSectionsForEveryElement[tagElem] = [nmbScc1,nmbScc2], where nmbScc1 and nmbScc2 are the names identifying the sections to be used in 1 and 2 directions respectively

mapSectionsDefinition

data structure that links each section name and direction with the respective record that contains the section definition parameters

mapInteractionDiagrams

data structure that links each section name and direction with the interaction diagram object to be used for its verification

procesResultVerifV

name of the function to be used for carrying out the section verification, e.g. shearSIA262.procesResultVerifV

Cracking limit state verification

from materials.xLamina import calculo_fis calculo_fis.lanzaCalculoFISFromXCDataPlanB(preprocessor,analysis,intForcCombFileName, outputFileName, mapSectionsForEveryElement, mapSectionsDefinition, procesResultVerifFIS)

preprocessor analysis

preprocessor name

type of analysis to be performed. Some predefined types

can be selected:

from solution import predefined_solutions as ps

pEF=xc.ProblemaEF()

ps.simple_static_linear(pEF) for linear static analysis ps.simple_newton_raphson(pEF) for non linear analysis using Newton Raphson algorithm to solve the equation

ps.simple_newton_raphson_band_gen(pEF) ps.simple_static_modified_newton(pEF) for non linear analysis using the modified Newton Raphson algorithm to

solve the equation

ps.penalty_newton_raphson(pEF)

ps.frequency_analysis(p)

intForcCombFileName

name of the file where are to be found the shell element forces and moments obtained for each quasi permanent SLS or frequent SLS combination, to be used in the verification of the cracking limit state. Directory path, file name and extension must be specified. The file contains a list of the following items (see the positive directions of the forces and moments in figure 8.1): $ULS_{id} \ elem_{id} \ N_1 \ N_2 \ N_{12} \ M_1M_2 \ M_{12} \ Q_1 \ Q_2$

outputFileName

name of the output files (directory path and file name without extension) where to write the results of the verification

mapSectionsForEveryElement

data structure, such that, for each shell element: mapSectionsForEveryElement[tagElem] = [nmbScc1, nmbScc2], where nmbScc1 and nmbScc2 are the names identifying the sections to be used in 1 and 2 directions respectively

mapSectionsDefinition

data structure that links each section name and direction with the respective record that contains the section definition parameters

mapInteractionDiagrams

data structure that links each section name and direction with the interaction diagram object to be used for its verification

procesResultVerifFIS name of the function to be used for carrying out the section verification, materie.g.

als.sia262.crackControlSIA262.procesResultVerifFISSIA262PlanB

Verification of beam sections

Pending.

Punching shear calculation

Punching shear calculation according to EC2

The design procedure for punching shear in EN 1992-1-1:2004 is defined in its clause 6.4.3. The procedure is base on checks at the face of the column and at the basic control perimeter u_1 . The following design shear stresses (Pa) along the control sections, are defined:

 $v_{Rd,c}$: design value of the punching shear resistance of a slab without punching shear reinforcement along the control section considered.

 $v_{Rd,cs}$: design value of the punching shear resistance of a slab with punching shear reinforcement along the control section considered.

 $v_{Rd,max}$: design value of the maximum punching shear resistance along the control section considered.

The following checks should be carried out:

- 1. At the column perimeter, or the perimeter of the loaded area, the maximum punching shear stress should not be exceeded: $v_{Ed} \leq v_{Rd,max}$.
- 2. Punching shear reinforcement is not necessary if: $v_{Ed} \leq v_{Rd,c}$.
- 3. Where v_{Ed} exceeds the value $v_{Rd,c}$ for the control section considered, punching shear reinforcement should be provided according to 6.4.5.

8.4.1.1 Related shear stress $v_{Ed,u1}$

To determine the design shear stress $v_{Ed,u1}$ related to the control perimeter u_1 , there are the following two options:

- Compute the maximum design shear stress $v_{Ed,u1}$ according to equation (6.38) of EC2. To do so, the β factor in this equation can be obtained using the equations (6.39) of EC2.
- Compute the maximum of the shear stresses at the critical perimeter using the results of the finite element model. This method takes into account the effect of the uneven rotational symmetric load by using the maximum value. Therefore, an additional increment of the shear force by the factor β in equation (6.38) can be omitted.

For the shake of simplicity we will use the second method that gives the value of $v_{Ed,u1}$ directly from the analysis results. Although the utilization of the maximum shear force value in the control perimeter is the most accurate method for determining the design value of the punching load, it is also the method most susceptible to the singularity effects. In particular, the user should pay attention to sufficient refinement of the mesh in the punching shear areas. As a rule of thumb, at least two or three elements must be arranged between the punching nodes and the control perimeter¹.

8.4.1.2 Punching shear resistance without punching shear reinforcement

The punching shear resistance without shear reinforcement $v_{Rd,c}$ has to be determined according equation (6.47) of EC2 clause 6.4.4 (1) as follows:

$$v_{Rd,c} = \max(C_{Rd,c} \cdot k \cdot (100 \cdot \rho_l \cdot f_{ck}/10^6)^{1/3}, v_{min}) + k_1 \sigma_{cp}$$
(8.1)

where:

 f_{ck} : characteristic compressive cylinder strength of concrete at 28 days expressed in Pa.

d: mean of effective depths of the slab cross-sections.

¹You can use mesh refinement techniques if needed.

k: is given by the equation: $k = min(1 + \sqrt{0.2/d}, 2.0)$ with d expressed in m.

 ρ_l : reinforcement ratio for longitudinal reinforcement computed as $\rho_l = min(\sqrt{\rho_{ly} \cdot \rho_{lz}}, 0.02)$. ρ_{ly} and ρ_{lz} relate to the bonded tension steel in y and z directions respectively. Their values should be calculated as mean values taking into account a slab width equal to the column width plus $3 \cdot d$ each side.

 σ_{cp} : given by the equation: $\sigma_{cp} = (\sigma_{cy} + \sigma_{cz})/2$, where:

 $\sigma_{cy} = N_{Ed,y}/A_{cy}$ and $\sigma_{cz} = N_{Ed,z}/A_{cz}$: are the normal concrete stresses in the critical section in y and z directions (Pa, positive if compression).

 $N_{Ed,y}$ and $N_{Ed,z}$: are the longitudinal forces across the full bay for internal columns and the longitudinal force across the control section for edge columns.

 A_c : is the area of concrete according to the definition of N_{Ed} .

 $C_{Rd,c}$: its value depend on the National Annex. The recommended value is $C_{Rd,c} = 0.18/\gamma_c$.

 v_{min} : its value is given by the expression (6.3N): $v_{min} = 0.035 \cdot k^{3/2} f_{ck}^{1/2}$ (k is defined above). The National Annex can provide a different value.

 k_1 : its value depends also of the National Annex. The recommended value is 0.1.

8.4.1.3 Maximum punching shear resistance $v_{Rd,max}$

If the punching shear design is impossible without punching reinforcement, the maximum punching shear resistance $v_{Rd,max}$ is calculated in the next step.

According to paragraph (3) of EC2 clause 6.4.5, the maximum punching shear resistance is performed on a column periphery. The considered periphery length u_0 encloses the basic control perimeter and can be determined directly on the load surface. The maximum punching shear resistance on the column periphery $v_{Rd,max}$ is determined as follows²:

$$v_{Rd,max} = 0.4 \cdot \nu f_{cd} \tag{8.2}$$

$$v_{Rd,max} = 0.4 \cdot \nu f_{cd}$$

$$\nu = 0.6 \cdot \left(1 - \frac{f_{ck}}{250 \times 10^6}\right)$$
(8.2)

The acting design shear force on the column periphery results from:

$$v_{Ed,u0} = \frac{\beta V_{Ed}}{u_0 \cdot d} \tag{8.4}$$

To avoid the computation of β , if we compare the expression with the epression (6.38) of EC2:

$$v_{Ed,u1} = \frac{\beta V_{Ed}}{u_1 \cdot d} \tag{8.5}$$

it is easy to deduce that:

$$v_{Ed,u0} = \frac{u_1}{u_0} v_{Ed,u1} \tag{8.6}$$

So, with the value of $v_{Ed,u1}$ computed according to 8.4.1.1 we can obtain $v_{Ed,u0}$ multiplying by u_1/u_0 .

The design is fullfilled³ if $v_{Ed,u0} \leq v_{Rd,max}$.

 $^{^{2}}$ The value of $v_{Rd,max}$ for use in a Country may be found in its National Annex.

³In the German National Annex, the design of the maximum punching shear resistance is not performed

8.4.1.4 Punching shear resistance with punching shear reinforcement

If the design of $v_{Rd,max}$ has been performed successfully, the required punching shear reinforcement is determined in the next step. The required punching shear reinforcement is calculated according to the adjusted Expression (6.52) of EC2. The required reinforcement A_{sw} results from the following equation:

$$A_{sw} = \frac{(v_{Ed,u1} - 0.75 \cdot v_{Rd,c}) \cdot d \cdot u_1}{1.5 \cdot \frac{d}{s_n} \cdot f_{ywd,ef} \cdot \sin \alpha}$$

$$(8.7)$$

where:

 $v_{Ed,u1}$: shear stress in the critical perimeter u_1 calculated according to section 8.4.1.1.

 $v_{Rd,c}$: shear resistance without shear reinforcement calculated according to expression (8.1).

d: mean of effective depths of the slab cross-sections.

 s_r : the radial spacing of perimeters of shear reinforcement.

 $f_{ywd,ef}$: is given by the equation: $f_{ywd,ef} = min(250 + 0.25 \cdot d, f_{ywd})$

 f_{uvd} : design yield of shear reinforcement

 α : the angle between the shear reinforcement and the slab plane.

According to the German National Annex, the amount of reinforcement in the first shear reinforcement perimeter must be increased by the factor $k_{sw,1} = 2.5$, and by $k_{sw,2} = 1.4$ in the second shear reinforcement perimeter.

The punching shear reinforcement should be placed up to a distance of $1.5 \cdot d$ from the outermost perimeter. The required length of the outermost perimeter is $u_{out,ef}$, which is defined according to equation (6.54) of EC2:

$$u_{out,ef} = \beta \cdot \frac{V_{Ed}}{v_{Rd,c} \cdot d} \tag{8.8}$$

where:

 V_{Ed} : design value of the applied shear force.

 $v_{Rd.c}$: shear resistance without shear reinforcement calculated according to expression (8.1).

d: mean of effective depths of the slab cross-sections.

 β : the load increment factor for considering the asymmetrical distribution of shear force in the control perimeter.

The value of V_{Ed} can be computed integrating the stresses obtained in 8.4.1.1 along the critical perimeter u_1 . Once the value of V_{Ed} , β can be obtained from the expression (6.38) of EC2 as follows:

$$\beta = \frac{v_{Ed,u1} \cdot u_1 \cdot d}{V_{Ed}} \tag{8.9}$$

on the column periphery, but in the basic control perimeter u_1 according to Expression NA6.53.1 as follows: $v_{Ed,u1} \leq v_{Rd,max} = 1.4 \cdot v_{Rd,c,u1}$.

Appendix A

Rough calculations

One way to verify the results of your finite element analysis is to compare them with analytical solutions when available. Sometimes there is no analytical solution for the problem at hand, but you can use some approximation that can give us a clue about the accuracy of your analysis. This chapter describes the routines in the rough_calculations package, that are written for that purpose.

A.1 Punching shear

```
import rough_calculations.ng_punzonamiento as punch
```

```
punch.esfuerzoPunzonamiento(qk,A)
rough estimation of the load on the slab over a support (HL.3 num. gordos)
punch.punzMaximo(fck,d,a,b)
rough estimation of the maximum punching force with no need of reinforcement for punching shear (HL.3 num. gordos)
punch.reinforcementPunz(Vd,fck,d,a,b,h,fyd)
rough estimation of the reinforcement for punching shear (HL.3 num. gordos)
```

where:

qk	characteristic uniformly distributed load on the slab
A	slab surface covered by the support
fck	characteristic compressive strength of concrete
d	effective depth of the slab
a,b	section dimensions of the support
h	overall depth of the slab
fydof the rein-	design value of the yield strength
forcement steel	

A.1.1 Beam deflections

import rough_calculations.flechas_vigas as beamDefl

beamDefl.deflCantBeamPconcentr(1,EI,P,a)

maximum deflection in a cantilever beam with a concentrated load at any point

beamDefl.deflCantBeamQunif(1,EI,q)

maximum deflection in a cantilever beam with a uniformly distributed load

beamDefl.deflCantBeamMend(1,EI,M)

maximum deflection in a cantilever beam with a couple moment at the free end

beamDefl.deflSimplSupBeamPconcentr(1,EI,P,b)

maximum deflection in a beam simply supported at ends with a concentrated load at any point

beamDefl.deflSimplSupBeamQunif(1,EI,q)

maximum deflection in a beam simply supported at ends with a uniformly distributed load

beamDefl.deflSimplSupBeamMend(1,EI,M)

maximum deflection in a beam simply supported at ends with a couple moment at the right end

1	span
EI	flexural rigidity of the section
P	concentrated load
q	uniformly distributed load
M	coupled moment
a	distance from the left end of the beam
b	distance from the right end of the beam

BEAM DEFLECTION FORMULAE

BEAM TYPE	SLOPE AT FREE END	DEFLECTION AT ANY SECTION IN ERMS OF	MAXIMUM DEFLECTION
1. Cantilever Bea	am – Concentrated load P at	the free end	
$\begin{array}{c c} P & \downarrow & x \\ \hline \theta & \delta_{\text{max}} \end{array}$	$\theta = \frac{Pl^2}{2EI}$	$y = \frac{Px^2}{6F}(sl - x)$	$\delta_{\text{max}} = \frac{Pl^3}{3EI}$
2. Cantilever Bea	am – Concentrated load P at	any point	
$ \begin{array}{c c} a & P & b \\ \hline & \delta_{\text{max}} \end{array} $	$\theta = \frac{Pa^2}{2EI}$	$y = \frac{Px^2}{6EI} (3a - x) 1 < x < a$ $y = \frac{Pa^2}{6EI} (3x - a) \text{for } a < x < a$	$\delta_{\max} = \frac{Pa^2}{6EI} (3l - a)$
3. Cantilever Bea	am – Uniformly distributed lo	oad ω (N/m)	
\mathcal{Y}	$\theta = \frac{\omega l^3}{6EI}$	$y = \frac{\omega x^2}{24EI} \left(x^2 + 6l - 4lx \right)$	$\delta_{ m max} = rac{\omega I^4}{8EI}$
4. Cantilever Bea	am – Unifor aly varying loa	Maximum intensity ω_o (N/m)	
$\omega = \frac{\omega_{o}}{l}(l-x)$ ω_{o} λ λ λ δ_{max}	$\theta = \frac{l^3}{2}$	$v = \frac{a_0 x^2}{20lEI} \left(10l^3 - 10l^2 x + 5lx^2 - x^3 \right)$	$\delta_{\rm max} = \frac{\omega_{\rm o} I^4}{30 EI}$
5. Canti ¹ ver Bea	am – le moment M	e free end	
$\begin{array}{c c} I & & x \\ \hline & & \delta_{\max} \\ \end{array}$	$\theta = \frac{\eta}{1}$	$y = \frac{Mx^2}{2EI}$	$\delta_{\max} = \frac{Ml^2}{2EI}$

BEAM DEFLECTION FORMULAS

BEAM TYPE	SLOPE AT ENDS	DEFLECTION AT ANY SECTION IT TERMS OF A	MAXIMUM AND CENTER DEFLECTION
6. Beam Simply	Supported at Ends – Concent	trated load P at the center	DEFLECTION
$\begin{array}{c c} \theta_1 & P & \theta_2 & X \\ \hline & & & \\ \hline \end{array}$	$\theta_1 = \theta_2 = \frac{Pl^2}{16EI}$ Supported at Ends – Concen	$y = \frac{Px}{12EI} \left(\frac{3l^2}{4} - x^2 \right) \text{ for } 0 < x < \frac{l}{2}$	$\delta_{\text{max}} = \frac{Pl^3}{48EI}$
7. Beam Simply	Supported at Ends = Concen		
· · ·	$\theta_1 = \frac{Pb(l^2 - b^2)}{6lEI}$ $\theta_2 = \frac{Pab(2l - b)}{6lEI}$	$\int_{c}^{c} \left \frac{1}{b} (x-a)^{3} + (l^{2}-b^{2}) \lambda \right $ for $a < x < l$	$\delta_{\text{max}} = \frac{Pb(l^2 - b^2)^{3/2}}{9\sqrt{3} lEI} \text{ at } x = \sqrt{(l^2 - b^2)/3}$ $\delta = \frac{Pb}{48EI} (3l^2 - 4b^2) \text{ at the center, if } a > b$
8. Beam Simply	Supported at Ends – Uniforn	nly distributed oad ω (*/m)	
$\begin{array}{c c} \omega & \lambda & \lambda \\ \lambda & \delta_{\text{max}} & \lambda \\ \lambda & \delta_{\text{max}} & \lambda \end{array}$	2121	$y = \frac{\omega x}{24EI} \left(l^3 - 2lx^2 + x^3 \right)$	$\delta_{\text{max}} = \frac{5\omega l^4}{384EI}$
9. Beam Simply	Supported Fnds – Coup!	*M at the rig'. end	
$\begin{array}{c c} \theta_1 \downarrow & f \theta_2 \stackrel{M}{\longrightarrow} x \\ \hline \downarrow y & I & \end{array}$	$\theta_1 = \frac{1}{6EI}$ $= \frac{Ml}{3EI}$	$y = \frac{Mlx}{6EI} \left(1 - \frac{x^2}{l^2} \right)$	$\delta_{\text{max}} = \frac{Ml^2}{9\sqrt{3} EI} \text{ at } x = \frac{l}{\sqrt{3}}$ $\delta = \frac{Ml^2}{16EI} \text{ at the center}$
Peam Simply	Supported at ds – Unifc.	mly varying load: Maximum intensity ω _o (N/m)	
$ \frac{\theta_1}{V} = \frac{\omega_0}{l} x \qquad \theta_2 $	$\theta_1 = \frac{7}{36} \frac{l^3}{EI}$ $\theta_2 = \frac{0}{45EI}$	$y = \frac{\omega_0 x}{360 lEI} \left(7l^4 - 10l^2 x^2 + 3x^4 \right)$	$\delta_{\text{max}} = 0.00652 \frac{\omega_{\text{o}} l^4}{EI} \text{ at } x = 0.519 l$ $\delta = 0.00651 \frac{\omega_{\text{o}} l^4}{EI} \text{ at the center}$

A.2 Masonry bridge

Functions for the assessment of a masonry arch bridge by means of a deterministic analytical method.

In these routines, the mechanism method programmed is used for a single-span masonry in a limit analysis, giving a load carrying capacity and a failure mode of the structure. The algorithm of the method applies the kinematic approach. It uses the assumption that a masonry arch becomes a mechanism when at least four plastic hinges appears in the arch barrel. A strict and efficient calculation for unknown position of hinges is based on application of a linear programming algorithm with a target function minimising a live load factor. In this way the most probable mechanism mode is found automatically.

A.3 Soil thrust





Appendix B

Generation of combinations to consider in the structural calculation

B.1 Introduction

This Appendix has the object of defining the actions, weighting coefficients and the combination of actions which shall be taken into account when designing structures.

Checking the structures through design is the most used method to guarantee their safety ¹.

B.1.1 The Limit States design method

The usual method prescribed by the codes for checking the safety of a structure is the so-called *Method of limit states*. A *limit state* is a situation in which, when exceeded, it may be considered that the structure does not fulfil one of the functions for which it has been designed.

The limit states are classified in:

- Ultimate Limit States (ULS);
- Serviceability Limit States (SLS), and
- Durability Limit States (DLS).

B.1.2 Design situations

The concept of *design situation* is useful to sort the checks performed on the project or study of a structure. A design situation is a simplified representation of the reality that is amenable to analysis.

Thus, it can be considered design situations those that correspond to the different phases of construction of the structure, the normal use of the structure, its reparation, exceptional conditions,

For each of the design situations, it must be checked that the structure doesn't exceed any of the Limit States previously laid down in paragraph B.1.1

¹Other procedures are also acceptable such as the reduced model tests, full-scale tests of the structure or its elements, extrapolation of the behaviour of similar structures, . . .

Type of structure	Design working life			
Temporary structures (*)	3 to 10 years (*)			
Replaceable structural elements that are	10 to 25 years			
not part of the main structure (eg,	-			
handrails, pipe supports)				
Agricultural or industrial buildings (or in-	15 to 50 years			
stallations) and maritime works				
Residential buildings or offices, bridges or	50 years			
crossings of a total length of less than 10				
meters and civil engineering structures (ex-				
cept maritime works) having a low or av-				
erage economic impact				
Public buildings, health and education.	75 years			
Monumental buildings or having a special	100 years			
importance				
Bridges of total length equal to or greater	100 years			
than 10 meters and other civil engineering				
structures of high economic impact				
(*)In accordance with the purpose of the structure (temporary ex-				
posure, etc.). Under no circumstances shall structures with a design				
working life greater than 10 years be regarded as temporary struc-				

tures

Table B.1: Design working life of the various types of structure (according reference [4]).

B.1.3 Actions

Action is defined as any cause capable of producing stress states in a structure, or modifying the existing one. Weight coefficients can be different according to the codes that apply for verification of the different structural elements (IAP, EHE, Eurocodes,...).

B.1.4 Working life

The working life of a structure is the period of time from the end of its execution, during which must maintain the requirements of security and functionality of project and an acceptable aesthetic appearance. During that period it will require conservation in accordance with the maintenance plan established for that purpose.

The design working life depends on the type of structure and must be fixed by the Owners at the start of the design. In any case its duration will be lower than that indicated in the regulations applicable or, in the absence of these, than the values laid down in Table B.1.

When a structure consists of different members, different working life values may be adopted for such members, always in accordance with the type and characteristics of the construction thereof.

Risk level B.1.5

The level of risk of an infrastructure defines the consequences of a structural failure during its construction or service (public building, private store, bridge, ...)

B.1.6 Control level

Regardless of the rigor with which the checking calculations of the structure are made during the project, its safety also depend on careful construction of it. Different standards establish the influence that the level of control during the execution of the work has on safety factors to be used in the execution of the same.

B.1.7 Combination of actions

When designing a structure or a structural member by the limit state method, load combinations shall be considered as the sum of the products of the load effect corresponding to the basic value of each load and the load factor.

Load factors shall be determined appropriately considering the limit state, the target reliability index, the variability in the load effect of each load and resistance, the probability of load coincidence, etc.

B.1.8 Verification of the structure

From the discussion in the previous sections, the verification procedure of the structure will consist of performing the following tasks:

- 1. identify the design situations to be considered when checking the structure;
- 2. identify the load criterions hypotheses for each of those design situations;
- 3. define the combinations of actions to be considered when checking the ULS and SLS, depending on:
 - (a) materials composing the structure or the element to check: rolled steel, reinforced concrete, wood, . . . ;
 - (b) risk level of the infrastructure
 - (c) level of control with which the construction work is performed;
 - (d) design situation (persistent, transient or accidental)
- 4. obtain the calculation value of the effect of actions for each combination.
- 5. verify all the limit states.

B.2 Actions

An action is a set of forces applied to the structure or a set of imposed deformations or accelerations, that has an effect on structural members (e.g. internal force, moment, stress, strain) or on the whole structure (e.g. defection, rotation)

B.2.1 Classification of actions

Actions can be classified by their variation over time, their nature, their origin, their spatial variation, ...

B.2.1.1 By their nature

- Direct actions: loads applied to the structure (e.g. self-weight, dead load, live load, ...)
- **Indirect actions**: imposed deformations or accelerations caused for example by temperature changes, moisture variation,...

B.2.1.2 By their variation over time

Actions shall be classified by their variation in time, by reference to their service life², as follows:

- Permanent actions G: actions that are likely to act throughout a given reference period and for which the variation in magnitude with time is negligible, or for which the variation is always in the same direction (monotonic) until the action attains a certain limit value, e.g. self-weight of structures, fixed equipment and road surfacing, and indirect actions caused by shrinkage and uneven settlement.
- Permanents of a non-constant value G*: are those which act at any time but whose magnitude is non constant. This group include those actions whose variation is a function of elapsed time and are produced in a single direction, tending towards a certain limit value (rheological actions, pretensioning, subsidence of the ground under the foundations, ...). They also include other actions originating from the ground whose magnitude does not vary as a function of time but as a function of the interaction between the ground and the structure (for example, thrusts on vertical elements).
- Variables Q: action for which the variation in magnitude with time is neither negligible nor monotonic. E.g. imposed loads on building floors, beams and roofs, wind actions or snow loads.
- Accidental actions A: action, usually of short duration but of significant magnitude, that is unlikely to occur on a given structure during the design working life. E.g. explosions, or impact from vehicles.
- seismic action AS: action that arises due to earthquake ground motions.

B.2.1.3 By their origin

- **Gravitational**: which has its origin in the earth's gravitational field (self-weight, dead load, earth pressure, . . .)
- Climatic: whose origin is in the climate (thermal action and wind actions³)
- **Rheological**: which has its origin in the response of material with plastic flow rather than deforming elastically when a force is applied (e.g. shrinkage of concrete).
- **Seismic**: due to earthquake ground motions.

²See section B.1.4.

³thermal and wind actions can not be due to climate, such as in the case of an oven or structures subjected to the thrust of jet engines of aircraft

B.2.1.4 By the structural response which they produce

- static action: action that does not cause significant acceleration of the structure or structural members:
- **dynamic action**: action that causes significant acceleration of the structure or structural members;
- quasi-static action: dynamic action represented by an equivalent static action in a static model.

B.2.1.5 By their spatial variation

- fixed action: action that has a fixed distribution and position over the structure or structural member such that the magnitude and direction of the action are determined unambiguously for the whole structure or structural member if this magnitude and direction are determined at one point on the structure or structural member;
- free action: action that may have various spatial distributions over the structure.

B.2.1.6 By their relation with other actions

- Compatible actions: two actions are compatible when it's possible for them to act simultaneously.
- **Incompatible actions**: two actions are incompatible when it's impossible for them to act at the same time (e.g. one crane acting simultaneously in two different positions).
- Synchronous actions: two actions are synchronous when the act necessarily together, at the same time (e.g. the braking load of a crane bridge will be synchronised with the action of the weight of the crane).

B.2.1.7 By their participation in a combination

- Leading action: in a combination of actions, the leading variable action is the one which produces the largest design load effect; its characteristic value is used.
- Accompanying action: variable action that accompanies the leading action in a combination; its characteristic value is reduced by using a factor Ψ .

B.2.2 Values of actions

B.2.2.1 Characteristic value of an action F_k

It is the principal representative value of an action; it is chosen so as to correspond to a 5% probability of not being exceeded on the unfavourable side during a "reference period" taking into account the design working life of the structure and the duration of the design situation.

B.2.2.2 Combination value of a variable action F_{r0}

Value chosen so that the probability that the effects caused by the combination will be exceeded is approximately the same as by the characteristic value of an individual action. It may be expressed as a determined part of the characteristic value by using a factor $\Psi_0 \leq 1$

B.2.2.3 Frequent value of a variable action F_{r1}

Value determined so that either the total time, within the reference period, during which it is exceeded is only a small given part of the reference period, or the frequency of it being exceeded is limited to a given value. It may be expressed as a determined part of the characteristic value by using a factor $\Psi_1 \leq 1$.

B.2.2.4 Quasi-permanent value of a variable action F_{r2}

Value determined so that the total period of time for which it will be exceeded is a large fraction ⁴ of the reference period. It may be expressed as a determined part of the characteristic value by using a factor $\Psi_1 \leq 2$.

B.2.2.5 Representative value F_r of the actions. Factors of simultaneity

The representative value of an action is the value of it that is used to verify the limit states. By multiplying this representative value by the the corresponding partial coefficient γ_f , the calculation value shall be obtained.

The principal representative value of the actions is their characteristic value. Usually, for permanent and accidental actions, a single representative value is considered, that matches the characteristic value ($\psi = 1$) ⁵. Other representative values are considered for the variable actions, in accordance with the verification involved and the type of action:

- Characteristic value F_k : this value is used for leading actions in the verification of ultimate limit states in a continuous or temporary situation and of irreversible serviceability limit states
- Combination value $\psi = \psi_0 F_k$ this value is used for accompanying actions in the verification of ultimate limit states in a continuous or temporary situation and of irreversible serviceability limit states.
- Frequent value $\psi = \psi_1 F_k$: this value is used for the leading action in the verification of ultimate limit states in an accidental situations and of reversible serviceability limit states.
- Quasi-permanent value $\psi = \psi_2 F_k$: this value is used for accompanying actions in the verification of ultimate limit states in an accidental situation and of reversible serviceability limit states as well as in the assessment of the postponed effects.

In short, the representative value of an action depends on:

- its variation over time (G,G*,Q,A,AS);
- its participation in the combination as leading action or accompanying action;
- the type of situation (accidental, ...);
- the origin of the load (climate, use, water, ...).

B.2.2.5.1 Values of Ψ factors of simultaneity The value of the simultaneity factors ψ are different depending on the action that is involved.

⁴according to *Documento Nacional de Aplicación español del Eurocódigo de Hormigón (UNE ENV 1992-1-1)* more than half of the service life of the structure

⁵The IAP instruction (reference [5]) makes some exceptions to this rule)

CLIMATIC ACTIONS	ψ_0	ψ_1	ψ_2
Snow loads	0.6	0.2	0.0
Wind loads	0.6	0.5	0.0
Temperature $(non-fire)$	0.6	0.5	0.0

Table B.2: Recommended values of Ψ factor for climatic actions, according to EHE

According to EHE: the recommended values of factors of simultaneity ψ_0, ψ_1, ψ_2 according to the *Documento Nacional de Aplicación español del Eurocódigo de Hormigón* (UNE ENV 1992-1-1) can be seen in tables B.2 y B.3.

According to EAE [4]: see tables B.5 y B.4.

According to IAP [5]: see table B.6.

B.2.2.6 Calculation value F_d of the actions

The calculation value of an action is obtained by multiplying its characteristic value by the corresponding partial coefficient γ_f :

$$F_d = \gamma_f \cdot F_r \tag{B.1}$$

The values of the coefficients γ_f takes into account one or more of the following uncertainties:

- 1. uncertainties in the estimation of the representative value of the actions, in fact, the characteristic value is chosen admitting a 5% probability of being exceeded during the working life of the structure;
- 2. uncertainties in the calculations results, due to simplifications in the models and to certain numeric errors (rounding, truncation, ...)
- 3. Uncertainty in the geometric and mechanical characteristics of the built structure. During the execution of the structure some errors can be committed ⁶ that can make the dimensions of the sections, the position of the reinforcement, the position of the axes, the mechanical characteristics of the materials, ..., be different from the theoretical.

B.2.2.6.1 Values of the partial coefficients The coefficients γ_f have different values in accordance with:

- 1. the limit state to be verified;
- 2. the design situation that is involved (see section B.3);
- 3. the variation of the action over time (according to classification in B.2.1.2);
- 4. the effect favourable o unfavourable of the action in the limit state that is verified;
- 5. the control level.

According to EHE: the values of the partial coefficients γ_f are specified in table B.7 for serviceability limit states and in table B.8 for ultimate limit states.

⁶It is understood that these errors are within the tolerances established in the regulations

LIVE LOADS	ψ_0	ψ_1	ψ_2
Roofs			
Inaccessible or accessible only for maintenance	0.7	0.5	0.3
Accessible	by use	by use	by use
Residential buildings			
Rooms	0.7	0.5	0.3
Stairs and public accesses	0.7	0.5	0.3
Cantilevered balconies	0.7	0.5	0.3
Hotels, hospitals, prisons,			
Bedrooms	0.7	0.5	0.3
Public areas, stairs and accesses	0.7	0.7	0.6
Assembly and areas	0.7	0.7	0.6
Cantilevered balconies	by use	by use	by use
Office and commercial buildings			
Private premises	0.7	0.5	0.3
Public offices	0.7	0.5	0.3
Shops	0.7	0.7	0.6
Commercial galleries, stairs and access	0.7	0.7	0.6
Storerooms	1.0	0.9	0.8
Cantilevered balconies	by use	by use	by use
Educational buildings			
Classrooms, offices and canteens	0.7	0.7	0.6
Stairs and access	0.7	0.5	0.6
Cantilevered balconies	by use	by use	by use
Churches, buildings for assembly and public performances			
Halls with fixed seatings	0.7	0.7	0.6
Halls without fixed seatings, tribunes, stairs	0.7	0.7	0.6
Cantilevered balconies	by use	by use	by use
Driveways and garages			
Traffic areas with vehicles under 30 kN in weight	0.7	0.7	0.6
Traffic areas with vehicles of 30 to 160 kN in weight	0.7	0.5	0.3

Table B.3: Recommended values of Ψ factors of simultaneity for climatic loads, according to EHE

USE OF AREA	ψ_0	ψ_1	ψ_2
Domestic, residential areas	0.7	0.5	0.3
Office areas	0.7	0.5	0.3
Congregation areas	0.7	0.7	0.6
Shopping areas	0.7	0.7	0.6
Storage areas	1.0	0.9	0.8
Traffic areas, weight of vehicle $\leq 30 \ kN$	0.7	0.7	0.6
Traffic areas, 30 kN < weight of vehicle $\leq 160 \ kN$	0.7	0.5	0.3
Inaccessible Roofs	0.0	0.0	0.0

Table B.4: Recommended values of Ψ factors for buildings, according to EAE

CLIMATIC ACTIONS	ψ_0	ψ_1	ψ_2
Snow loads in buildings set over a thousand meters above	0.7	0.5	0.2
sea level.			
Snow loads in buildings set under a thousand meters	0.5	0.2	0.0
above sea level.			
Wind loads	0.6	0.2	0.0
Thermal action	0.6	0.5	0.0

Table B.5: Recommended values of Ψ factors of simultaneity, according to EAE

Variable actions	ψ_0	ψ_1	ψ_2
Traffic load model fatigue	1.0	1.0	1.0
Other variable actions	0.6	0.5	0.2

Table B.6: Values of Ψ factors of simultaneity according to IAP.

ACTION	Effect		
	favourable	unfavourable	
Permanent	$\gamma_G = 1.00$	$\gamma_G = 1.00$	
Prestressing (pre-tensioned concrete)	$\gamma_P = 0.95$	$\gamma_{P} = 1.05$	
Prestressing (post-tensioned concrete)	$\gamma_P = 0.90$	$\gamma_P = 1.10$	
Permanent of a non-constant value	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$	
Variable	$\gamma_{Q} = 0.00$	$\gamma_Q = 1.00$	
NOTATION:			
G: Permanent action.			
P. Prestressing			

Table B.7: Partial factor for actions in serviceability limit states according to EHE.

Action	Control level	Effect in persistent or transient		Effect in acc	idental or seismic
		design situations		design situations	
		favourable	unfavourable	favourable	unfavourable
	intense	$\gamma_G = 1.00$	$\gamma_G = 1.35$	$\gamma_G = 1.00$	$\gamma_G = 1.00$
G	normal	$\gamma_G = 1.00$	$\gamma_G = 1.50$	$\gamma_G = 1.00$	$\gamma_G = 1.00$
	low	$\gamma_G = 1.00$	$\gamma_G = 1.60$	$\gamma_G = 1.00$	$\gamma_G = 1.00$
	intense	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.50$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
G*	normal	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.60$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
	low	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.80$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
	intense	$\gamma_Q = 0.00$	$\gamma_Q = 1.50$	$\gamma_{Q} = 0.00$	$\gamma_Q = 1.00$
Q	normal	$\gamma_Q = 0.00$	$\gamma_Q = 1.60$	$\gamma_Q = 0.00$	$\gamma_Q = 1.00$
	low	$\gamma_Q = 0.00$	$\gamma_{Q} = 1.80$	$\gamma_Q = 0.00$	$\gamma_Q = 1.00$
A	-	-	-	$\gamma_A = 1.00$	$\gamma_A = 1.00$
Nomanion.					

NOTATION:

Table B.8: Partial factor for actions in ultimate limit states according to EHE.

G*: Permanent action of a non-constant value.

Q: Variable action.

A: Accidental action.

G: Permanent action.

G*: Permanent action of a non-constant value.

Q: Variable action.

A: Accidental action

ACTION	Effect		
	favourable	unfavourable	
Permanent	$\gamma_G = 1.00$	$\gamma_G = 1.00$	
Permanent of a non-constant value	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$	
Variable	$\gamma_{Q} = 0.00$	$\gamma_{Q} = 1.00$	

Table B.9: Partial factor for actions in serviceability limit states according to EAE.

Action	Effect in persistent or transient		Effect in accidental or seismic	
	design situations		design situations	
	favourable	unfavourable	favourable	unfavourable
G	$\gamma_G = 1.00$	$\gamma_G = 1.35$	$\gamma_G = 1.00$	$\gamma_G = 1.00$
G*	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.50$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Q	$\gamma_Q = 0.00$	$\gamma_Q = 1.50$	$\gamma_Q = 0.00$	$\gamma_Q = 1.00$
A	-	-	$\gamma_A = 1.00$	$\gamma_A = 1.00$
NOTATION:				
G: Permanent action.				
G*: Permanent action of a non-constant value.				
Q: Variable action.				
A: Accidental action.				

Table B.10: Partial factor for actions in ultimate limit states according to EAE.

According to EAE: the values of the partial coefficients γ_F to be used are specified int tables B.9 for serviceability limit states and in table B.10 for ultimate limit states.

According to IAP: the values of the partial coefficients γ_F to be used are specified int tables B.11 for serviceability limit states and in table B.12 for ultimate limit states.

B.3 Design situations

Design situations, that take into account the circumstances under which the structure can be required during its execution and use, shall be classified as follows:

- 1. Persistent design situations, which refer to the conditions of normal use.
- 2. transient design situations, which refer to temporary conditions applicable to the structure, e.g. during execution or repair.
- 3. Accidental design situations, which refer to exceptional conditions applicable to the structure or to its exposure, e.g. to fire, explosion, impact or the consequences of localised failure.

B.4 Level of quality control

A two level system for control during execution has been adopted:

- Intense control.
- Normal control.

ACTION	Ef	FECT
	favourable	unfavourable
Permanent	$\gamma_G = 1.00$	$\gamma_G = 1.00$
Internal prestressing (post-tensioned concrete)	$\gamma_{P_1} = 0.9$	$\gamma_{P_1} = 1.1$
Internal prestressing (pre-tensioned concrete)	$\gamma_{P_1} = 0.95$	$\gamma_{P_1} = 1.05$
External prestressing	$\gamma_{P_2} = 1.0$	$\gamma_{P_2} = 1.0$
Other prestressing actions	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Rheological	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Thrust of the site	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Variable	$\gamma_{Q} = 0.00$	$\gamma_{Q} = 1.00$
NOTATION:	-	
G: Permanent action.		
P_1 : Internal prestressing.		
P_2 : External prestressing.		
G*: Permanent action of a non-constant value.		
Q: Variable action.		
A: Accidental action.		

Table B.11: Partial factor for actions in serviceability limit states according to IAP.

Action	Effect in persistent or transient		Effect in accidental or seismic	
	design situations		design situations	
	favourable	unfavourable	favourable	unfavourable
Permanent	$\gamma_G = 1.00$	$\gamma_G = 1.35$	$\gamma_G = 1.00$	$\gamma_G = 1.00$
Internal prestressing	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
External prestressing	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.35$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Other prestressing actions	$\gamma_{G*} = 0.95$	$\gamma_{G*} = 1.05$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Rheological	$\gamma_{G*} = 1.0$	$\gamma_{G*} = 1.35$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Thrust of the site	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.50$	$\gamma_{G*} = 1.00$	$\gamma_{G*} = 1.00$
Variable	$\gamma_Q = 0.00$	$\gamma_Q = 1.50$	$\gamma_{Q} = 0.00$	$\gamma_{Q} = 1.00$
Accidental	-	-	$\gamma_A = 1.00$	$\gamma_A = 1.00$
			,	,

Table B.12: Partial factor for actions in ultimate limit states according to IAP.

As will be seen later, the partial factors for a material or a member resistance depend on the level of inspection during construction.

B.5 Limit states

They can be defined as those states beyond which the structure no longer fulfils the relevant design criteria.

The design of the structure will be right when:

- 1. it is verified that no ultimate limit state is exceeded for the design situations and load cases defined in B.6.1, and
- 2. it is verified that no serviceability limit state is exceeded under the design situations and load cases defined in B.6.2.

B.5.1 Ultimate limit states

They are states associated with collapse or with other similar forms of structural failure. They generally correspond to the maximum load-carrying resistance of a structure or structural member

The following ultimate limit states shall be verified where they are relevant: - failure caused by fatigue or other time-dependent effects.

- 1. loss of equilibrium of the structure or any part of it, considered as a rigid body;
- failure by excessive deformation, transformation of the structure or any part of it into a mechanism, rupture, loss of stability of the structure or any part of it, including supports and foundations;
- 3. failure caused by fatigue or other time-dependent effects.

B.5.2 Serviceability limit states

They can be defined as states that correspond to conditions beyond which specified service requirements for a structure or structural member are no longer met. These service requirements can concern:

- functionality.
- comfort.
- durability.
- aesthetics.

The verification of serviceability limit states should be based on criteria concerning the following aspects:

- 1. deformations that affect:
 - the appearance,
 - the comfort of users, or

• the functioning of the structure (including the functioning of machines or services),

or that cause damage to finishes or non-structural members;

2. vibrations

- that cause discomfort to people, or
- that limit the functional effectiveness of the structure:
- 3. damage that is likely to adversely affect
 - the appearance,
 - the durability, or
 - the functioning of the structure.

B.6 Combination of actions

When the verification of a structure is carried out by the partial factor method, it shall be verified than, in all relevant design situations, no relevant limit state is exceeded when design values for actions or effects of actions and resistances are used in the design models.

In order to eliminate the combinations that are not possible (or do not make sense), the following criteria will be considered:

- When an action is involved in a combination, none of its incompatible actions will be involved in that combination.
- When an action is involved in a combination, all of its synchronous actions must be involved in that combination ⁷

In what follows, we will consider any structure, under the following actions:

- n_G permanent actions: G_i^8 .
- n_{G*} permanent actions of a non-constant value: $G*_i$.
- n_Q variable actions: Q_l .
- n_A accidental actions: Q_m .
- n_{AS} seismic actions: Q_n .

B.6.1 Combinations of actions for ultimate limit states

For the selected design situations and the relevant ultimate limit states the individual actions for the critical load cases should be combined as detailed in this section.

⁷See synchronous action and compatible action definitions in section B.2.1.6.

⁸The subscript refers to each of permanent actions on the structure $G_1, G_2, G_3, G_4, \ldots, G_{n_G}$

B.6.1.1 Combinations of actions for persistent or transient design situations

For each variable action, a group of combinations with this action as *leading variable action* will be considered ⁹.

$$\sum_{i=1}^{n_G} \gamma_G \cdot G_{k,i} + \sum_{j=1}^{n_{G*}} \gamma_{G*} \cdot G *_{k,j} + \gamma_Q \cdot Q_{k,d} + \sum_{l=1}^{d-1} \gamma_Q \cdot Q_{r0,l} + \sum_{l=d+1}^{n_Q} \gamma_Q \cdot Q_{r0,l}$$
(B.2)

where:

 $\gamma_G \cdot G_{k,i}$: design value of the permanent action i, obtained from its characteristic value;

 $\gamma_{G*} \cdot G*_{k,j}$: design value of the permanent action of a non-constant value j, obtained from its characteristic value;

 $\gamma_Q \cdot Q_{k,d}$: design value of the leading variable action d, obtained from its characteristic value;

 $\gamma_Q \cdot Q_{r0,l}$: design value of la variable action l, obtained from its accompanying value.

B.6.1.1.1 Number of combinations to be considered: According to section B.2.2.6:

- The permanent actions, in ULS combinations corresponding to persistent or transient design situations, will have two non-zero partial factors.
- In the same case, the permanent actions of a non-constant value will have two non-zero partial factors that, in some cases, can be equal (see the case of internal prestressing on the table B.12).
- The variable actions will have a single non-zero partial factor.

therefore, assuming that:

 n_{G2} is the number of permanent actions that have two different partial factors;

 n_{G1} is the number of permanent actions that have a single partial factor¹⁰;

 n_{G*2} is the number of permanent actions of a non-constant value that have two different partial factors;

 n_{G*1} the number of permanent actions of a non-constant value that have a single partial factor, and

 n_Q is the number of variable actions, all of then have a single partial factor.

If, by now, incompatibility or synchronicity of actions is ignored, for each variable action we'll have:

- $2^{n_{G2}}$ combinations of permanent actions in the set G2;
- 1 combination of permanent actions in the set G1;
- $2^{n_{G*2}}$ combinations of permanent actions in the set G*2;

⁹See section B.2.1.7.

¹⁰Because both factors are equal.

- 1 combination of permanent actions in the set G * 1, and
- 2^{n_Q-1} combinations of accompanying variable actions.

As, for each leading action two partial factors must be considered, the total number of combinations $n_{comb,spt}$ for persistent or transient design situations will be equal to the cartesian product of the previous combinations by $2^{n_{Qd}}$, where Qd is the number of variable actions that can be leading:

$$n_{comb,ULS,spt} = 2^{n_{G2}} \cdot 2^{n_{G*2}} \cdot 2^{n_{Q*2}} \cdot 2^{n_{Q}-1} \cdot 2^{n_{Qd}} = 2^{n_{G2}+n_{G*2}+n_{Q}+n_{Qd}-1}$$
(B.3)

Among these combinations, those that are incompatibles must be eliminated. For synchronic actions, the following procedure can be followed: Let a be a synchronic action of the action b:

- 1. a is eliminated from the list of variable actions;
- 2. the action a + b is added to the list of variable actions:
- 3. incompatibility between a + b and b actions is set.

B.6.1.2 Combinations of actions for accidental design situations

For each variable action Q_l , n_A combinations with that action as leading are formed.

$$\sum_{i=1}^{n_G} \gamma_G \cdot G_{k,i} + \sum_{i=1}^{n_{G*}} \gamma_{G*} \cdot G *_{k,j} + A_{k,m} + \gamma_Q \cdot Q_{r1,d} + \sum_{l=1}^{d-1} \gamma_Q \cdot Q_{r2,l} + \sum_{l=d+1}^{n_Q} \gamma_Q \cdot Q_{r2,l}$$
(B.4)

where:

 $A_{k,m}$: design value of the accidental action m, obtained from its characteristic value;

 $\gamma_Q \cdot Q_{r1,d}$: design value of the leading variable action d, obtained from its representative frequent value;

 $\gamma_Q \cdot Q_{r2,l}$: design value of la variable action l, obtained from its representative quasi-permanent value.

B.6.1.2.1 Number of combinations to be considered: it results the same number of combinations for each sum than in the case solved in the paragraph B.6.1.1 (see B.3 expression), though, in this case, the representative values of the variable actions are other ones. If, as usual, the partial factors for seismic actions are equal for favourable and unfavourable actions, it suffices to multiply by the number of accidental actions n_A .

$$n_{comb,ULS,acc} = 2^{n_{G2} + n_{G*2} + n_Q + n_{Qd} - 1} \cdot n_A$$
(B.5)

For incompatible actions, the procedure provided for in section B.6.1.1 is applicable.

B.6.1.3 Combinations of actions for seismic design situations

For each seismic action one combination will be formed:

$$\sum_{i=1}^{n_G} \gamma_G \cdot G_{k,i} + \sum_{j=1}^{n_{G*}} \gamma_{G*} \cdot G *_{k,j} + AS_{k,n} + \sum_{l=1}^{n_Q} \gamma_Q \cdot Q_{r2,l}$$
 (B.6)

where:

 $A_{k,m}$ is the design value of the accidental action m, and

 $\gamma_Q \cdot Q_{r2,l}$ is the design value of the variable action l, obtained from its representative quasipermanent value.

B.6.1.3.1 Number of combinations to be considered:

$$n_{comb.ULS.sism} = 2^{n_{G2} + n_{G*2} + n_Q} \cdot n_{AS}$$
 (B.7)

For incompatible actions, the procedure provided for in section B.6.1.1 is applicable.

B.6.2 Combinations of actions for serviceability limit states

For the selected design situations and the relevant serviceability limit states the individual actions for the critical load cases should be combined as detailed in this section.

B.6.2.1 Rare combinations:

For each variable action, one combination with this action as *leading variable action* will be considered.

$$\sum_{i=1}^{n_G} G_{k,i} + \sum_{j=1}^{n_{G*}} G *_{k,j} + Q_{k,d} + \sum_{l=1}^{d-1} Q_{r0,l} + \sum_{l=d+1}^{n_Q} Q_{r0,l}$$
(B.8)

In a general case, with no incompatible or concomitant combinations, the following combinations will be formed (see notation in section B.6.1.1):

$$n_{comb,SLS,pf} = 2^{n_{G2} + n_{G*2} + n_Q + n_{Qd} - 1}$$
(B.9)

Since the partial factors are for serviceability limit states, the sets G2 y G*2 generally will not match those for ultimate limit states. Given that in many cases both partial factors are equal to the unity, the cardinality of these sets will be much lower than the equivalent in paragraph B.6.1.1.

For incompatible actions, the procedure provided for in section B.6.1.1 is applicable.

B.6.2.2 Frequent combinations:

For each variable action, one combination in which this action acts as leading will be formed.

$$\sum_{i=1}^{n_G} G_{k,i} + \sum_{j=1}^{n_{G*}} G_{k,j} + Q_{r1,d} + \sum_{l=1}^{d-1} Q_{r2,l} + \sum_{l=d+1}^{n_Q} Q_{r2,l}$$
(B.10)

the number of combinations will be the same as the precedent case, since only the combination factors can vary.

B.6.2.3 Quasi-permanent combinations:

$$\sum_{i=1}^{n_G} G_{k,i} + \sum_{j=1}^{n_{G*}} G *_{k,j} + \sum_{l=1}^{n_Q} Q_{r2,l}$$
(B.11)

the number of combinations will be:

$$n_{comb,SLS,cp} = 2^{n_{G2} + n_{G*2} + n_Q}$$
 (B.12)

B.6.3 Combinations to be considered in the calculation

According to the discussion in the previous sections, the number of combinations for a general calculations will be the following:

Tileing to limit states	
Ultimate limit states	number of combinations
Persistent or transient design situations	$2^{(n_G+n_{G*}+n_Q)}\cdot n_Q$
Accidental design situations	$\frac{2^{(n_G+n_{G*}+n_Q)} \cdot n_Q}{2^{(n_G+n_{G*}+n_Q)} \cdot n_Q \cdot n_A}$ $2^{(n_G+n_{G*}+n_Q)} \cdot n_{AS}$
Seismic design situations	$2^{(n_G+n_{G*}+n_Q)}\cdot n_{AS}$
Total ULS	$2^{(n_G+n_{G*}+n_Q)}\cdot (n_Q(1+n_A)+n_{AS})$
Serviceability limit states	
Rare combinations	n_Q
Frequent combinations	n_Q
Quasi-permanent combination	1
Total SLS	$2n_Q + 1$
Total combinations	$2^{(\mathbf{n_{\mathbf{G}}}+\mathbf{n_{\mathbf{G}*}}+\mathbf{n_{\mathbf{Q}}})}\cdot(\mathbf{n_{\mathbf{Q}}}(1+\mathbf{n_{A}})+\mathbf{n_{\mathbf{AS}}})+2\mathbf{n_{\mathbf{Q}}}+1$

For example, if we had:

- 2 permanent actions;
- 1 permanent action of a non-constant value;
- 3 variable actions;
- 1 accidental action, and
- 2 seismic actions

the number of combinations will be:

Ultimate limit states	number of combinations
Persistent or transient design situations	$2^{(2+1+3)} \times 3 = 192$
Accidental design situations	$2^{(2+1+3)} \times 3 \times 1 = 192$
Seismic design situations	$2^{(2+1+3)} \times 2 = 128$
Total ULS	$2^{(2+1+3)} \times (3 \times (1+1) + 2) = 512$
Serviceability limit states	
Rare combinations	3
Frequent combinations	3
Quasi-permanent combination	1
Total SLS	6+1=7
Total combinations	519

Algorithm to write the complete list of combinations **B.6.4**

B.6.4.1 Combinations for ultimate limit states

Each of the sums in expressions (B.2),(B.4) y (B.6) appears as follows:

$$\sum_{i=1}^{n} \gamma_f \cdot F_{r,i} \tag{B.13}$$

For each action F_i the partial factor can take two values, depending on the effect favourable or unfavourable of the action¹¹.

The design value of the action $F_{r,i}$ depends on:

- its variation in time (G,G*,A,A,AS);
- its role in the combination, as leading or accompanying action;
- if there is or not accidental actions in the combination:
- the nature of the action (climatic or live loads).

in any case, for any combination, the value of $F_{r,i}$ is known.

Moreover, the value of n is known for each sum.

Following this, the summands of (B.13) correspond to the variations with repetition ¹² of two elements 13 taken n by n.

To write the variations with repetition of expression (B.13), proceed as follows:

Let $\gamma_{\mathbf{f}_v}$ be the row vector whose components are the partial factors of the variation v (1 \leq $v \leq 2^n$):

$$\gamma_{\mathbf{f}_v} = [\gamma_{f,1}, \gamma_{f,2}, \cdots, \gamma_{f,i}, \cdots, \gamma_{f,n}]$$
(B.14)

that's to say, the element $\gamma_{f,i}$ is the partial factor (favourable or unfavourable) of action $F_{r,i}$. Let $\mathbf{F_r}$ be the column vector whose components are the actions $F_{r,i}$ of the expression (B.13):

$$\mathbf{F_r}^T = [F_{r,1}, F_{r,2}, \cdots, F_{r,i}, \cdots, F_{r,n}]$$
 (B.15)

then, the expression (B.13) is equivalent to the scalar product:

$$\sum_{i=1}^{n} \gamma_f \cdot F_{r,i} = \gamma_{\mathbf{f}_v} \cdot \mathbf{F_r} \tag{B.16}$$

and it must be formed as many scalar products as variations with repetition can be arranged, that's to say, 2^n .

Let $S_{F,v}$ be the sum that corresponds to variation v,

$$S_{F_{\mathbf{r}},v} = \gamma_{\mathbf{f}v} \cdot \mathbf{F_r} \tag{B.17}$$

then each of sums (B.2),(B.4) and (B.6) gives rise to set of variations:

 $^{^{11}}$ We assume a priory unknown the effect favourable or unfavourable of the action for the limit state and structural element in analysed

¹²The variations with repetition of n elements taken k by k are the arranged groups formed by k elements from A (which may be repeated)

13The partial factors corresponding to favourable and unfavourable effects

$$S_{F_r,1} = \gamma_{\mathbf{f}_1} \cdot \mathbf{F_r}$$

$$S_{F_r,2} = \gamma_{\mathbf{f}_2} \cdot \mathbf{F_r}$$

$$\cdots$$

$$S_{F_r,v} = \gamma_{\mathbf{f}_v} \cdot \mathbf{F_r}$$

$$\cdots$$

$$S_{F_r,n_F} = \gamma_{\mathbf{f}_{n_F}} \cdot \mathbf{F_r}$$

where n_F is the number of actions in each case, that's to say n_G , n_{G*} , n_Q , n_A , or n_{AS} . Therefore, the summands (B.2),(B.4) and (B.6) will be one of the following scalar products:

- Summand corresponding to permanent actions: S_{G_r,v_G} $(1 \le v_G \le 2^{n_G})$.
- Summand corresponding to permanent actions of a non-constant value: $S_{G*_r,v_{G*}}$ (1 $\leq v_{G*} \leq 2^{n_{G*}}$).
- Summand corresponding to variable actions: S_{Q_r,v_Q} $(1 \le v_Q \le 2^{n_Q})$.
- Summand corresponding to accidental actions: S_{A_r,v_A} $(1 \le v_A \le 2^{n_A})$.
- Summand corresponding to seismic actions: $S_{AS_r,v_{AS}}$ $(1 \le v_{AS} \le 2^{n_{AS}})$.

B.6.4.1.1 Combinations of actions for persistent or transient design situations With this notation, the expression (B.2) can be written as follows:

$$CQ_{v_G,v_{G*},v_Q,d} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r_0,d},v_Q}$$
 (B.18)

where:

 v_G is the variation corresponding to the permanent actions;

 v_{G*} is the variation corresponding to the permanent actions of a non-constant value;

 v_Q is the variation corresponding to the variable actions;

d is the index that corresponds to the leading variable action, and

$$\mathbf{Q}_{r0,d}$$
 is the vector $[Q_{r0,1}, Q_{r0,2}, \cdots, Q_{r0,d-1}, Q_{k,d}, Q_{r0,d+1}, \cdots, Q_{r0,n_Q}]$

B.6.4.1.2 Combinations of actions for accidental design situations Similarly, the expression (B.4) can be written as follows:

$$CA_{v_G,v_{G*},v_Q,d,m} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r_2,d},v_Q} + A_{k,m}$$
(B.19)

where:

 v_G is the variation corresponding to the permanent actions;

 v_{G*} is the Variation corresponding to the permanent actions of a non-constant value;

 v_Q is the variation corresponding to the variable actions;

d is the index corresponding to the leading variable action;

 $\mathbf{Q}_{r2,d}$ is the vector $[Q_{r2,1}, Q_{r2,2}, \cdots, Q_{r2,d-1}, Q_{r1,d}, Q_{r2,d+1}, \cdots, Q_{r2,n_Q}];$

m is the index that corresponds to the accidental action considered, and

 $A_{k,m}$ is the design value of the accidental action m.

B.6.4.1.3 Combinations for seismic design situations Similarly, the expression (B.6) can be written as follows:

$$CS_{v_G,v_{G*},v_Q,n} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r_2},v_Q} + AS_{k,n}$$
(B.20)

where

 v_G is the variation corresponding to the permanent actions;

 v_{G*} is the variation corresponding to the permanent actions of a non-constant value;

 v_Q is the variation corresponding to the variable actions;

 \mathbf{Q}_{r2} is the vector $[Q_{r2,1}, Q_{r2,2}, \cdots, Q_{r2,n_Q}];$

n is the index of the seismic action considered, and

 $AS_{k,n}$ is the design value of the seismic action n.

B.6.4.1.4 Calculation algorithm The proposed algorithm for writing all the combinations for ultimate limit states is as follows:

- 1. calculation of all the variations corresponding to actions G: γ_{q,v_G} $(1 \le v_G \le 2^{n_G})$
- 2. calculation of all the variations corresponding to actions G*: $\gamma_{q*,v_{G*}}$ $(1 \le v_{G*} \le 2^{n_{G*}})$
- 3. calculation of all the variations corresponding to actions Q: γ_{q,v_Q} $(1 \leq v_Q \leq 2^{n_Q})$
- 4. from d=1 to $d=n_q$
 - (a) calculation of all the combinations $CQ_{v_G,v_{G*},v_Q,d}$.
- 5. from d = 1 to $d = n_Q$
 - (a) from m = 1 to $m = n_A$

i. calculation of all the combinations $CA_{v_G,v_{G*},v_O,d,m}$.

- 6. from n = 1 to $n = n_{AS}$
 - (a) calculation of all the combinations $CS_{v_G,v_{G*},v_{O},n}$.
- 7. end

refinement of step 4a:

- 1. from $v_G = 1$ to $v_G = 2^{n_G}$
 - (a) calculate S_{G_k,v_G}
 - (b) from $v_{G*} = 1$ to $v_{G*} = 2^{n_{G*}}$
 - i. calculate $S_{G*_k,v_{G*}}$

ii. from
$$v_Q = 1$$
 to $v_Q = 2^{n_Q}$

A. calculate
$$S_{Q_{r_0,d},v_Q}$$

B. calculate
$$CQ_{v_G,v_{G*},v_O,d} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r_0,d},v_O}$$

2. end

refinement of step 5(a)i:

- 1. from $v_G = 1$ to $v_G = 2^{n_G}$
 - (a) calculate S_{G_k,v_G}
 - (b) from $v_{G*} = 1$ to $v_{G*} = 2^{n_{G*}}$
 - i. calculate $S_{G*_k,v_{G*}}$
 - ii. from $v_Q = 1$ to $v_Q = 2^{n_Q}$
 - A. calculate $S_{Q_{r_2,d},v_Q}$.

B. calculate
$$CA_{v_G,v_{G*},v_Q,d,m} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r_2,d},v_Q} + A_{k,m}$$

2. end

refinement of step 6a:

- 1. from $v_G = 1$ to $v_G = 2^{n_G}$
 - (a) calculate S_{G_k,v_G}
 - (b) from $v_{G*} = 1$ to $v_{G*} = 2^{n_{G*}}$
 - i. calculate $S_{G*_k,v_{G*}}$
 - ii. from $v_Q = 1$ to $v_Q = 2^{n_Q}$
 - A. calculate S_{Q_{r2},v_Q} .

B. calculate
$$CS_{v_G,v_{G*},v_Q,n} = S_{G_k,v_G} + S_{G*_k,v_{G*}} + S_{Q_{r2},v_Q} + AS_{k,n}$$

2. end

B.6.4.2 Combinations for serviceability limit states

Taking into account the partial factors for serviceability limit states, if:

$$S_{G_k} = \sum_{i=1}^{n_G} G_{k,i} \tag{B.21}$$

$$S_{G*_k} = \sum_{j=1}^{n_{G*}} G*_{k,j}$$
 (B.22)

$$S_{Q_{r0},d} = \sum_{l=1}^{d-1} Q_{r0,l} + Q_{k,d} + \sum_{l=d+1}^{n_Q} Q_{r0,l}$$
(B.23)

$$S_{Q_{r2},d} = \sum_{l=1}^{d-1} Q_{r2,l} + Q_{r1,d} + \sum_{l=d+1}^{n_Q} Q_{r2,l}$$
(B.24)

and

$$S_{Q_{r2}} = \sum_{l=1}^{n_Q} Q_{r2,l} \tag{B.25}$$

then: the n_Q rare combinations will be:

$$CPF_d = S_{G_k} + S_{G*_k} + S_{Q_{r0},d}$$
(B.26)

the n_Q frequent combinations will be:

$$CF_d = S_{G_k} + S_{G*_k} + S_{Q_{r2},d} (B.27)$$

and the quasi-permanent combination will be:

$$CCP = S_{G_k} + S_{G*_k} + S_{Q_{r_2}}$$
(B.28)

B.6.4.2.1 Calculation algorithm The calculation algorithm of all the combinations for service-ability limit states would be expressed as follows:

- 1. calculation of S_{G_k}
- 2. calculation of S_{G*_k}
- 3. from d = 1 to $d = n_Q$
 - (a) calculate $S_{Q_{r0},d}$
 - (b) calculate $CPF_d = S_{G_k} + S_{G_{k}} + S_{Q_{r0},d}$
- 4. from d = 1 to $d = n_Q$
 - (a) calculate $S_{Q_{r_2},d}$
 - (b) calculate $CF_d = S_{G_k} + S_{G_{*_k}} + S_{Q_{r_2},d}$
- 5. calculation of $S_{Q_{r2}}$
- 6. calculate $CCP = S_{G_k} + S_{G_{*_k}} + S_{Q_{r_2}}$
- 7. end

Bibliography

- [1] Carlos A. Felippa, Introduction To Finite Element Methods (ASEN 5007). (Department of Aerospace Engineering Sciences University of Colorado at Boulder).
- [2] A. López, D. J. Yong, M. A. Serna, Lateral-torsional buckling of steel beams: a general expression for the moment gradient factor. (Lisbon, Portugal: Stability and ductility of steel structures, 2006).
- [3] Ministerio de Fomento, *EHE; Instrucción de hormigón estructural.* (España: Comisión Permanente del Hormigón.Ministerio de Fomento. 1998).
- [4] Ministerio de Fomento, *EAE; Instrucción de acero estructural.* (España: Comisión Permanente de estructuras de acero.Ministerio de Fomento. 2004).
- [5] Ministerio de Fomento, IAP; Instrucción sobre las acciones a considerar en el proyecto de puentes de carretera. (España: Dirección General de Carreteras. Ministerio de Fomento. 1998).
- [6] Ministerio de Fomento, NCSE-02; Norma de construcción sismorresistente: parte general y edificación. (España: Comisión permanente de Normas Sismorresistentes. Ministerio de Fomento. 2002).
- [7] Ministerio de Fomento, NBE-AE-88; Acciones en la edificación. (España: Ministerio de Fomento. 1988).