

# 무인이동체 설계해석 소프트웨어 개발 및 성능시험

Unmanned Vehicle Design Software

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2022. 11. 8



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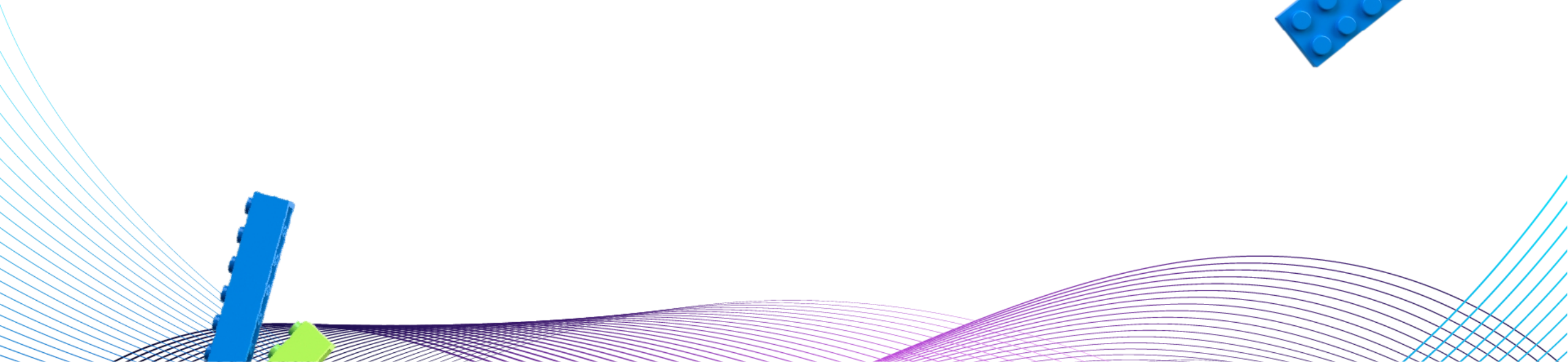
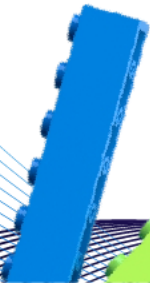
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# 01

## 무인항공기 시장





# 01

## 항공기 시장



- Airbus A320 family deliveries<sup>[39][40]</sup>
- Boeing 737 series deliveries<sup>[41][42]</sup>

(출처: Wikipedia)

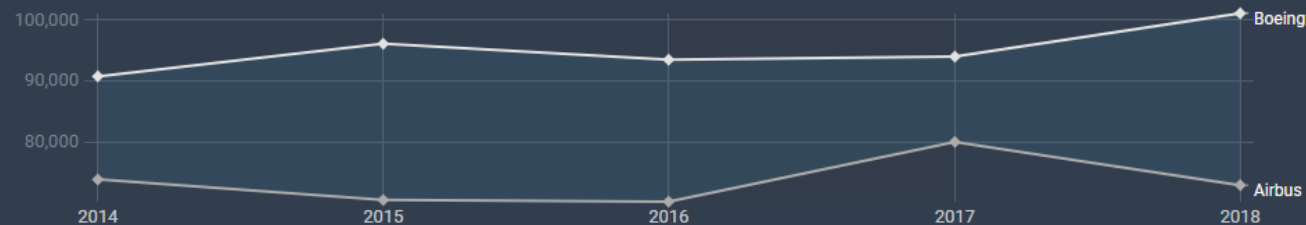


에어버스 A320(좌) 보잉 737(우)

- 전통적인 민항기 시장  
⇒ 소품종 소량 생산

### 5-year Revenue Comparison

Boeing beats Airbus both in revenues and margins. Airbus witnessed its highest growth in the last five years, in 2017.



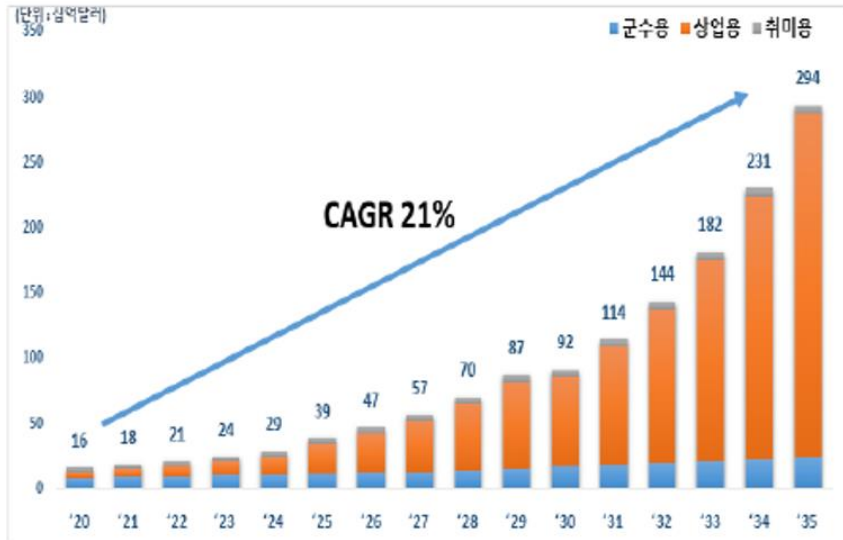
All revenues in \$m.

Chart: Praveen Duddu • Created with Datawrapper

# 01

## 무인기(드론) 시장

(공중) '20년 159억불 → '25년 385억불 → '30년 915억불



(출처: 무인이동체로드맵 공청회)

### 미국 드론 등록 대수 (2022, FAA)

- 865,505 drones registered
  - 314,689 commercial drones registered
  - 538,172 recreational drones registered



2021년 매출 US\$ 3.83B

- 무인기(드론) 시장  
⇒ 다품종 대량 생산

# 01

## 드론 오픈소스 생태계



### PX4 Autopilot

The PX4 Autopilot provides guidance, navigation, and control algorithms for autonomous fixed wing, multirotor, and VTOL airframes, along with estimators for attitude and position.



### MAVLink

Feature complete and fully customizable control station for MAVLink based UAVs. It can be used to update, configure, and tune an airframe. The control station has a mighty mission planner and tracks missions.



### MAVSDK

Feature complete and fully customizable control station for MAVLink based UAVs. It can be used to update, configure, and tune an airframe. The control station has a mighty mission planner and tracks missions.



### QGroundControl

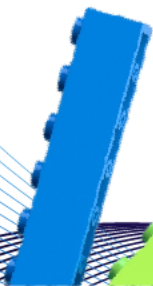
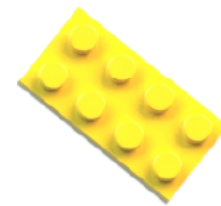
Feature complete and fully customizable control station for MAVLink based UAVs. It can be used to update, configure, and tune an airframe. The control station has a mighty mission planner and tracks missions.





02

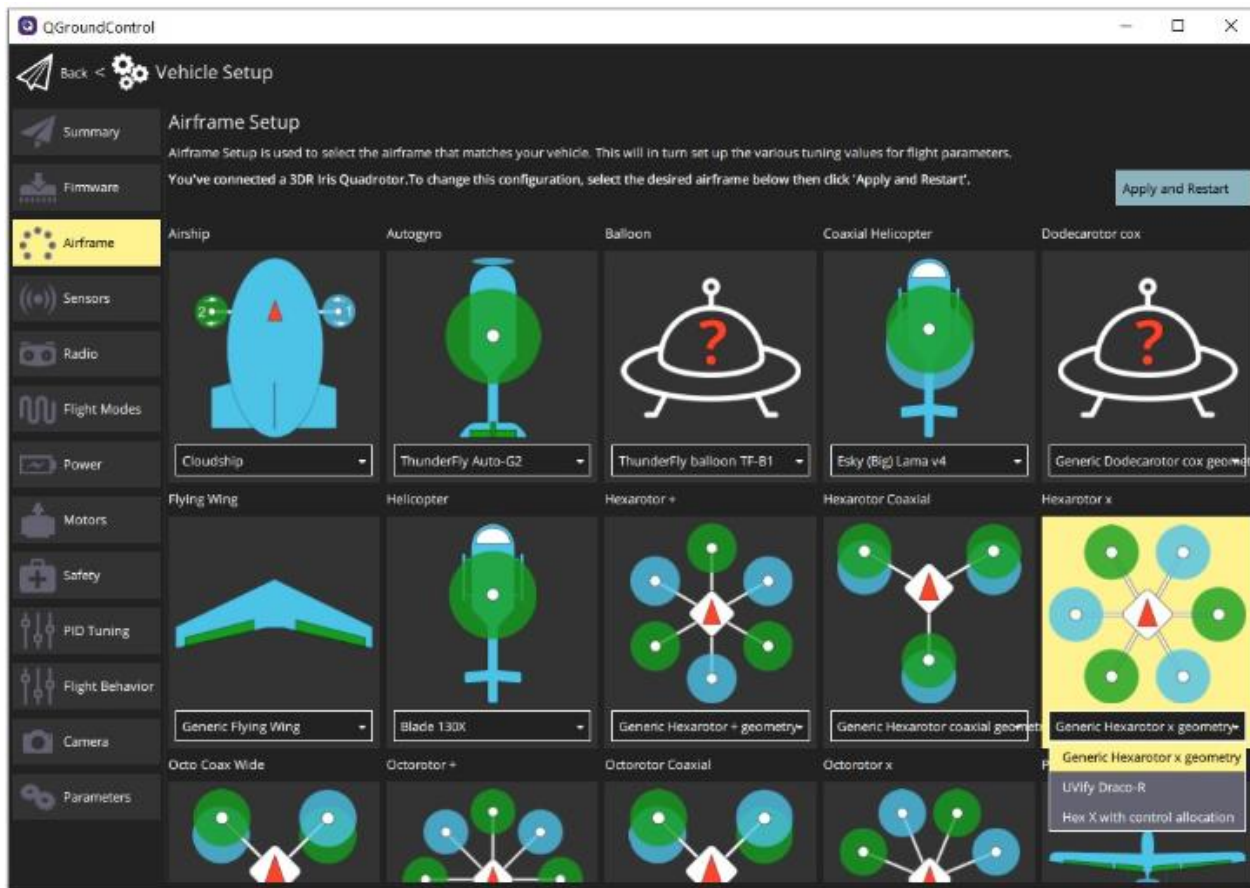
# PX4 비행제어



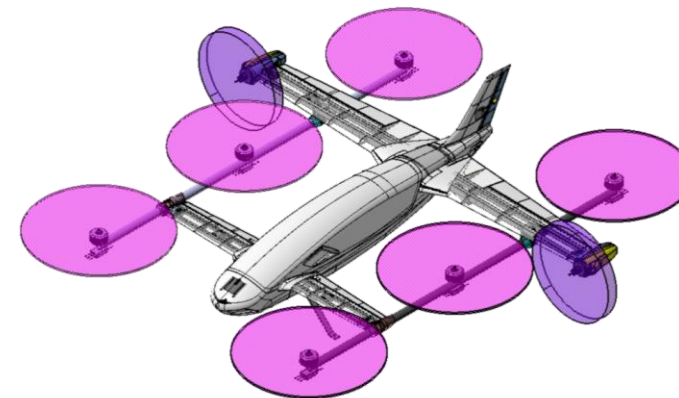


# 02

## PX4 비행제어 알고리즘 개선



드론 기체 형상 설정 (QGC)



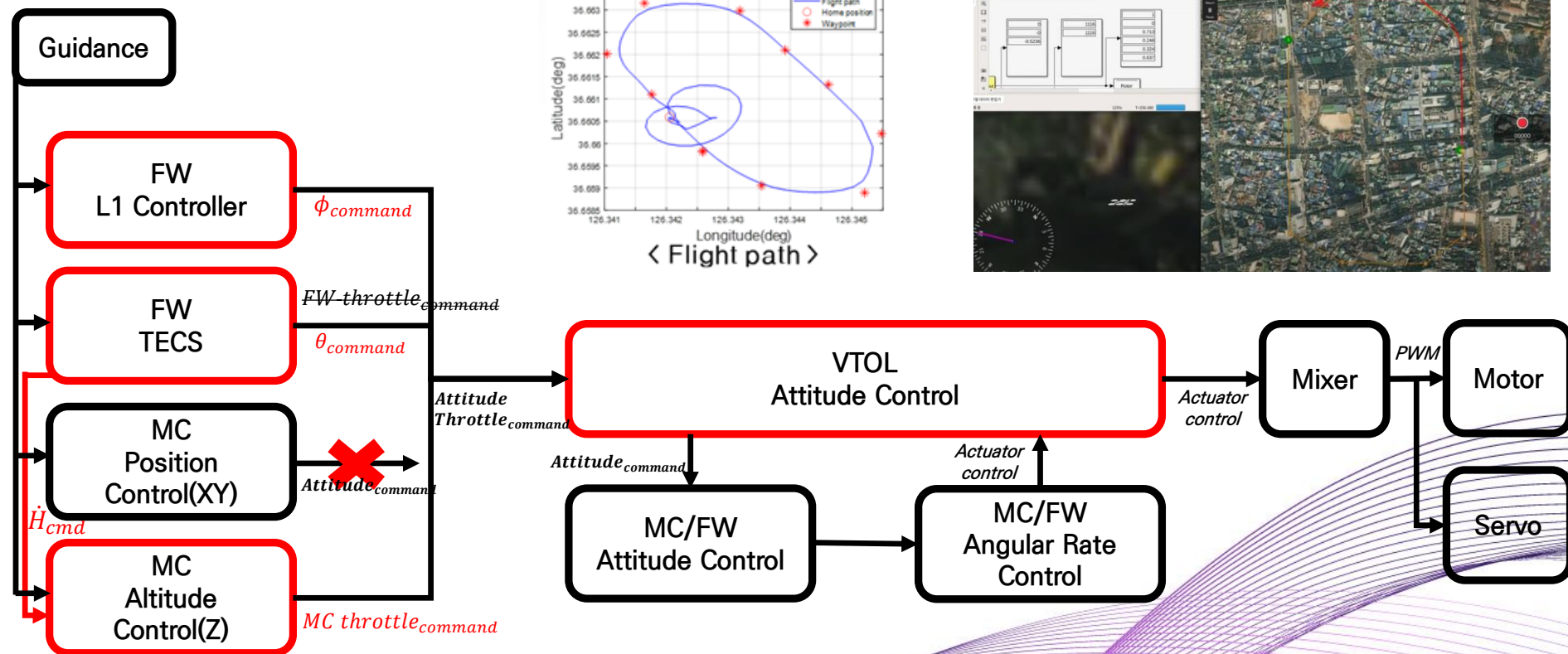
LC62-50B (항우연)



# 02

## PX4 비행제어 알고리즘 개선

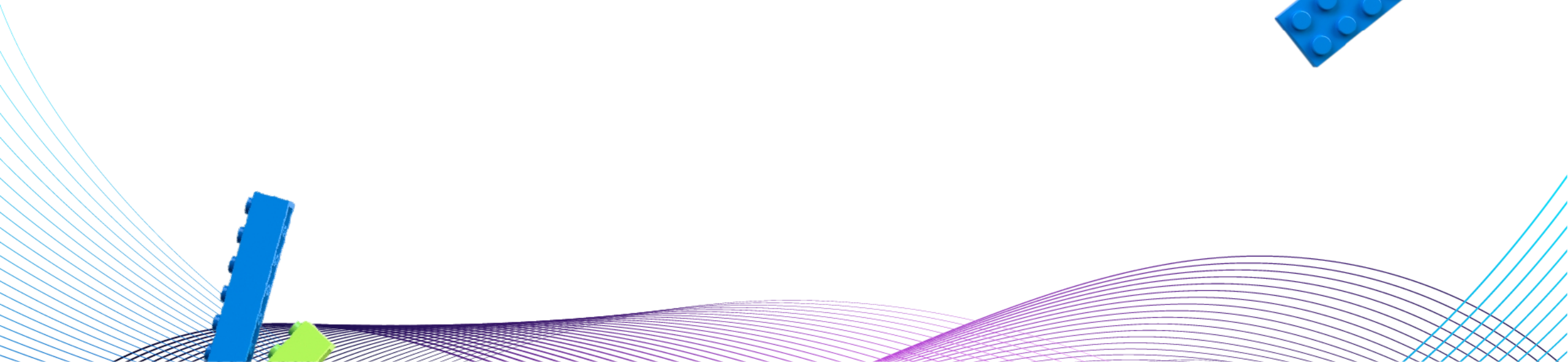
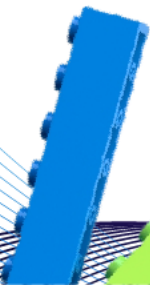
- 천이 비행을 위한 비행제어 알고리즘 개선
  - 천이 중 선회 비행이 가능하도록 비행제어 알고리즘 개선
  - 회전익+고정익 모드 혼합 시 비행제어 안정화





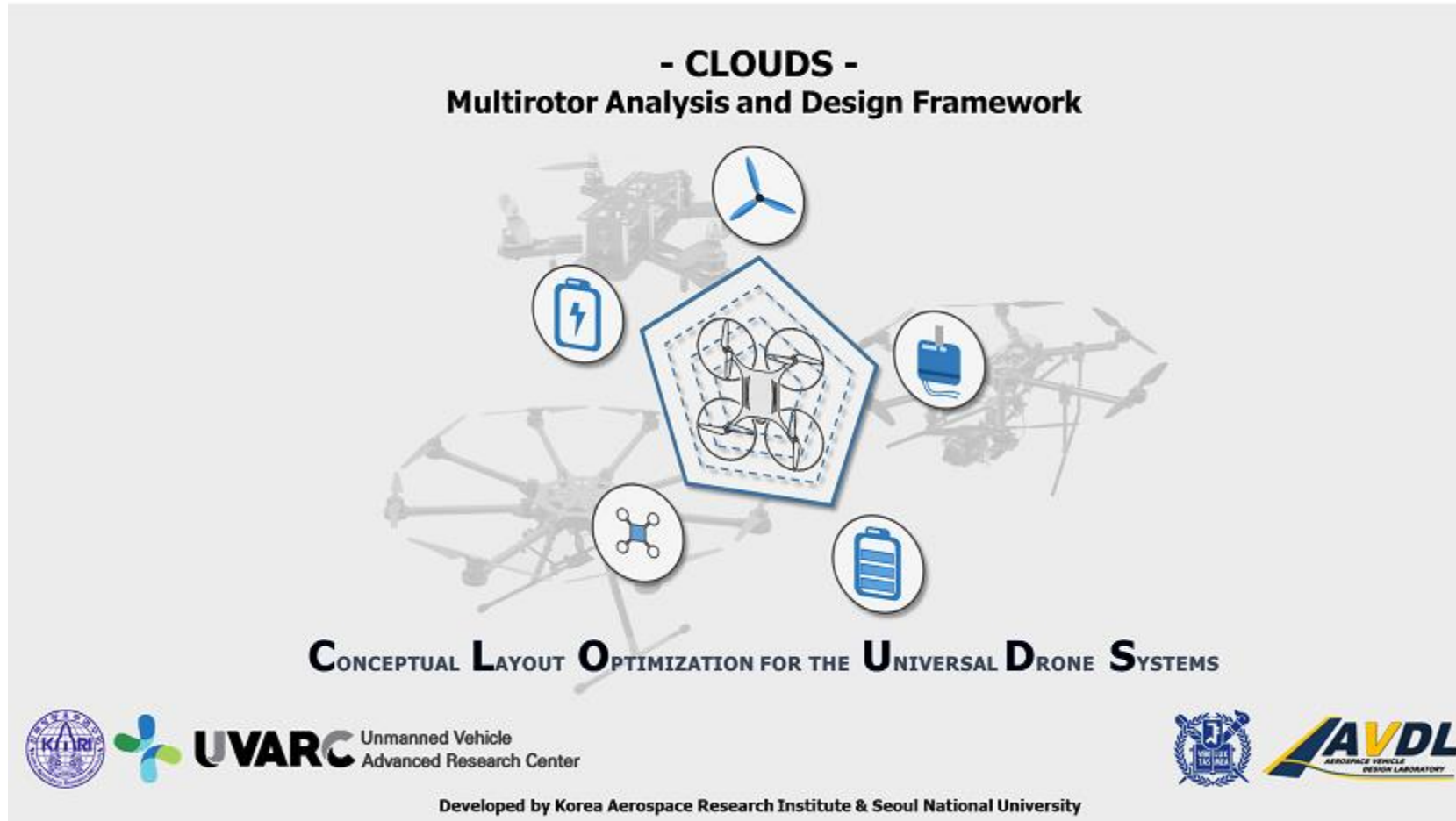
# 03

## 소형 드론 설계 SW



# 03

## 소형 드론 초기설계 SW



25kg 이하 소형드론 초기설계용 소프트웨어



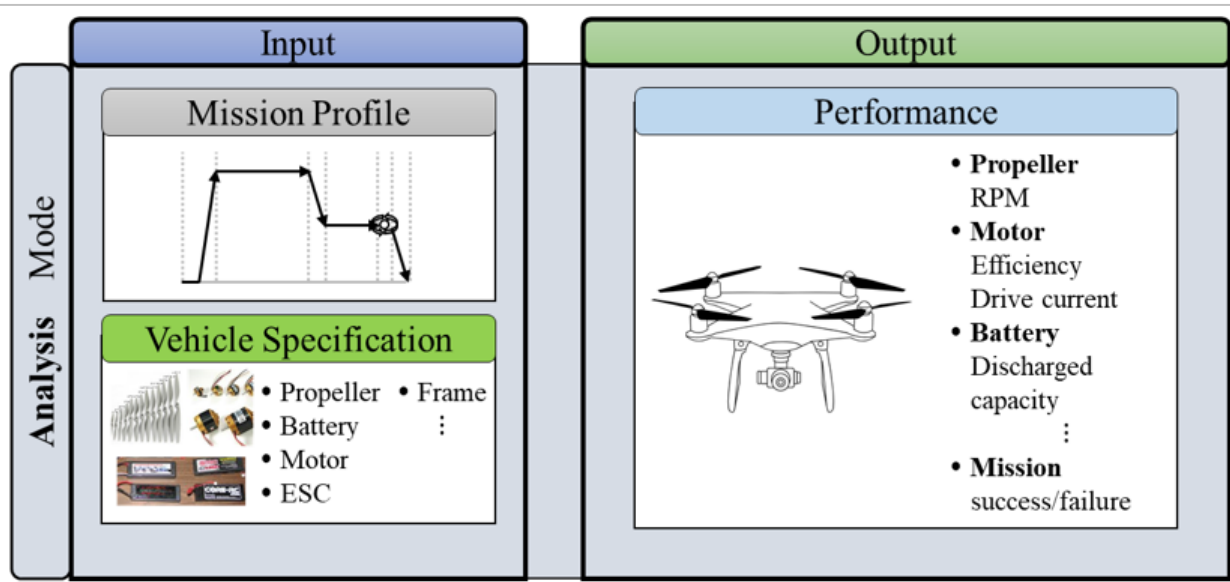
## 해석 모드

## ▶ 입력

- 사용자 요구 임무 형상
- 멀티콥터 구성품 제원

## ▶ 결과

- 멀티콥터 성능
- 임무 달성 가능 여부



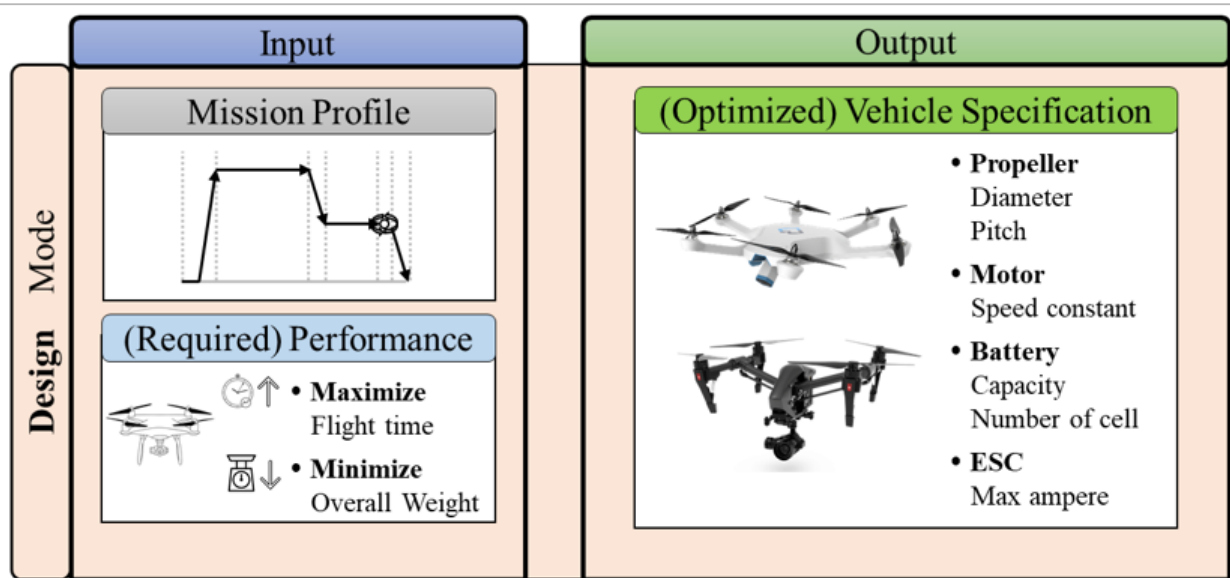
## 설계 모드

## ▶ 입력

- 사용자 요구 임무 형상
- 멀티콥터 요구 성능

## ▶ 결과

- 최적설계된 멀티콥터의 구성품 제원



# 03

## 드론 기술시연기 개발 및 성능검증

### 목적함수

### 비행시간 최대화

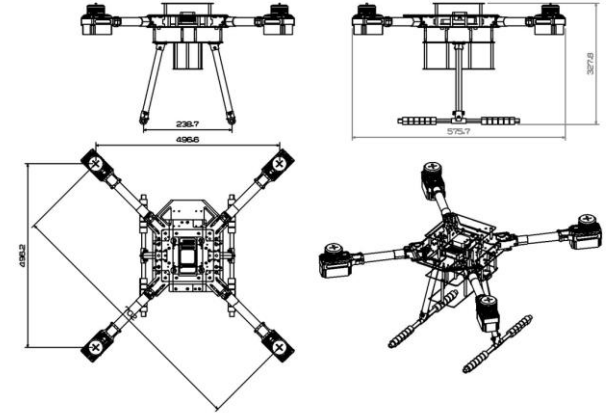
설계변수	단위	최소	최대
모터상수	RPM/V	200	600
프로펠러 지름	In	15	17
배터리 전압	V	4S	6S
배터리 용량	mAh	5,000	15,000
ESC 최대전류량	A	30	70
이륙중량*	g		5000

\* 계측장비중량 1250g 포함

### 설계결과 및 기술시연기 개발

	CLOUDS	기술시연기 적용*
모터	390	KDE 4014XF-380
프로펠러	15.2x5.0	Tmotor 16.0x5.4
배터리	6S x 12,000	Dinogy 11,000
ESC	30	Hobbywing Xrotor 50A
이륙중량	4923	4910
축간거리	800	700

\* 프로펠러 크기에 따른 축간거리 선정후 재계산



### 비행시험 결과

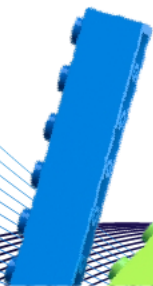
임무중량 (g)	CLOUDS	비행시험
0	15분 49초	16분 22초
500	12분 41초	12분 19초
1000	11분 10초	11분 19초





# 04

## 무인이동체 매개변수 모델링 & 해석

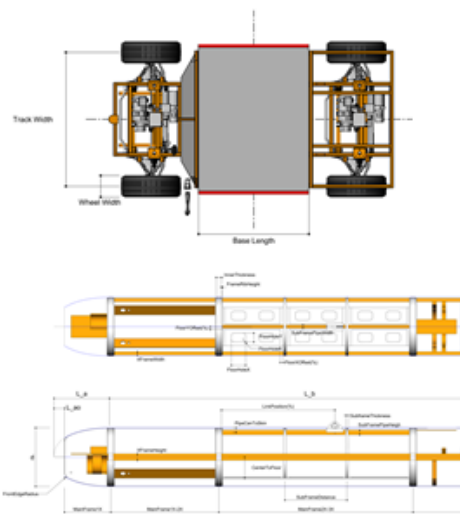




## 04

## 무인이동체 매개변수 형상 모델링

## Unmanned Vehicle Parameters (Excel)

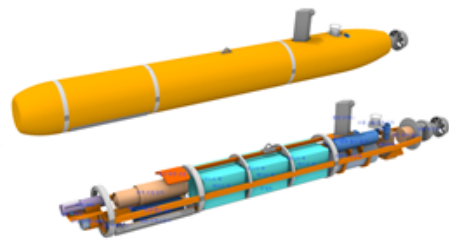


LandingPadLength(%)	101
LandingPadWidth(%)	105
LandingPadHeight	170
LandingPadLift	400
LifterLength(%)	90
BatteryLength	1200
BatteryWidth	600
BatteryHeight	200
BaseLength	1450
GroundClearance	330

Main Shape Parameters				
L_a	425	MainFrame1X	355	XFrameHe
L_b	3075	MainFrame1X-2X	880	XFrameTh
L_c	1005	MainFrame2X-3X	1490	SubFrame
L_d	80	MainFrame3X-Plate	1190	PipeCent
L_e	350	MainFrameWidth	32	SubFrame
d	450	FrameInnerHeight/2	166	SubFrame
theta	25	InnerThickness	40	SubFrame
n	2.5	FrameRibThickness	2	Subframe1
FrameWeight	75	FrameRibOffset	4	Subframe2
FairingWeight	38	FrameRibHeight	8	Subframe1

AFT Wing Parameters									
Position	X	2109	Z	164	Attack Angle	0			
	Airfoil	Chord	Thickness	Twist angle	Span	Sweepback	Dihedral	Buffer zone	Twist ref. chord
Root	sd7032	991	149	0	0				25%
Mid	sd7032	486	75	0	453	29	0	5	25%
Tip	sd7032	311	30	0	1300	0.3	0		25%
Buffer Zone Shape						AFT Control Surface		Use	TRUE
	W2-3				Span Start(%)	Span End(%)	Width(%)	Clearance	
Num Sections	10				5	45	23	1.5	
Blend Bulge	0.5								
Continuity	Tangent								
Front Wing Parameters									
Position	X	776	Z	50	Attack Angle	0			
	Airfoil	Chord	Thickness	Twist angle	Span	Sweepback	Dihedral	Buffer zone	Twist ref. chord
Root	sd7032	348	79	0	0				25%
Tip	sd7032	256	35	0	1120	0.35	0		25%
						BMD Control Surface		Use	TRUE

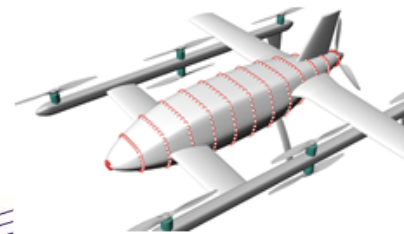
## Vehicle 3D CAD



## Model Data



Surface Volume  
Weight Area  
Volume Center (3D)  
Center of gravity (3D)



Fuselage XSection

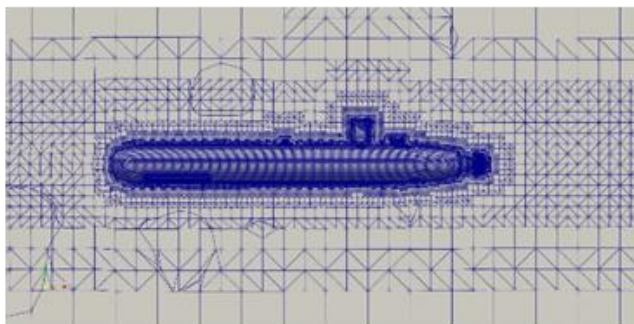
# 04

## 무인이동체 매개변수 유동해석

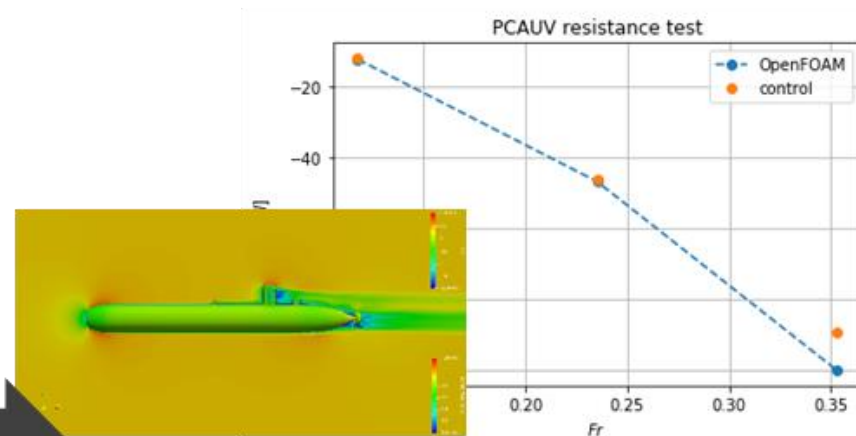
3D CAD Model



CFD Mesh



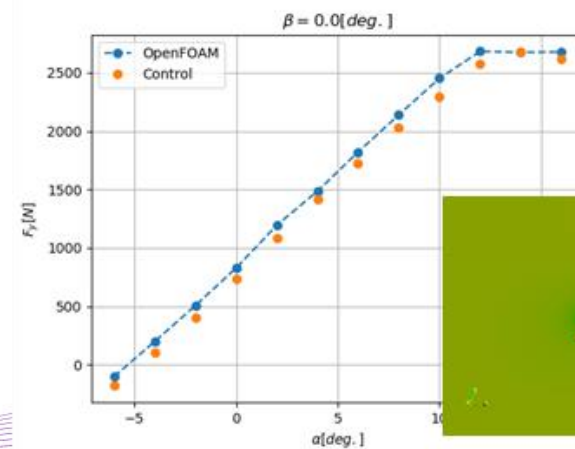
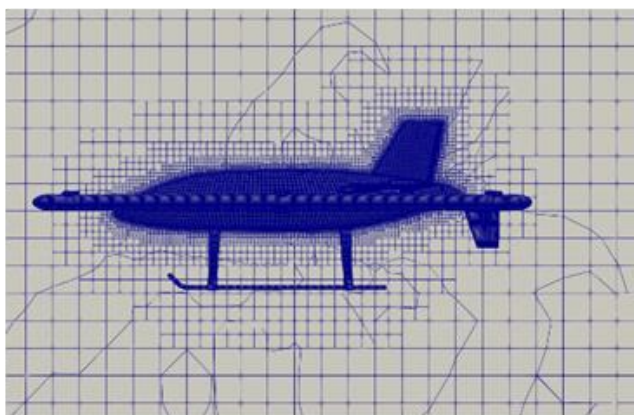
CFD Analysis



3D STL Files



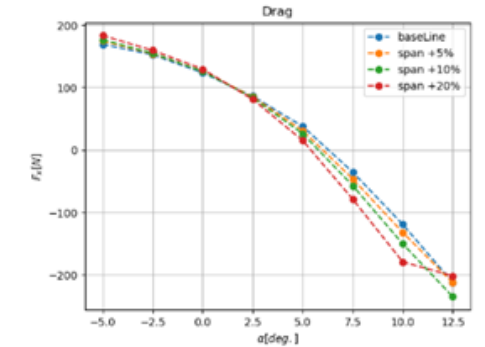
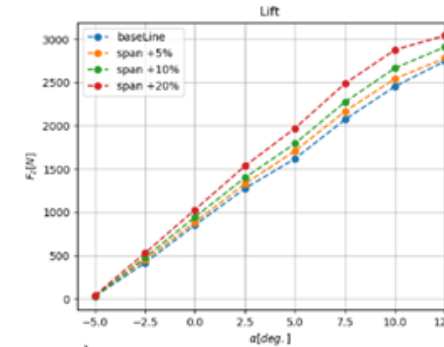
openFOAM SHM



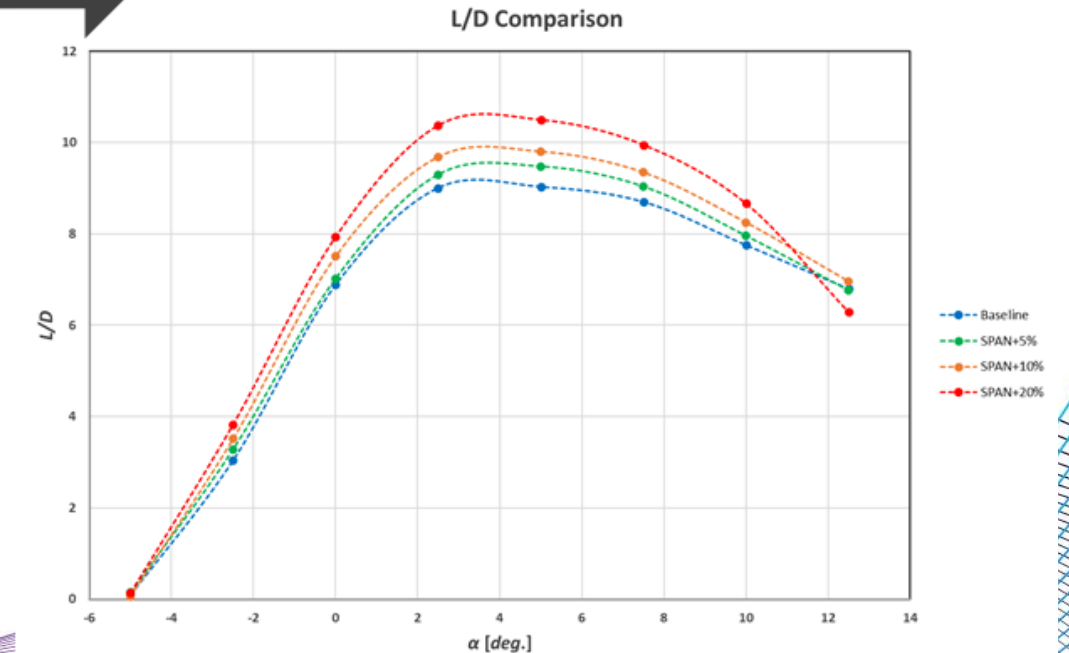
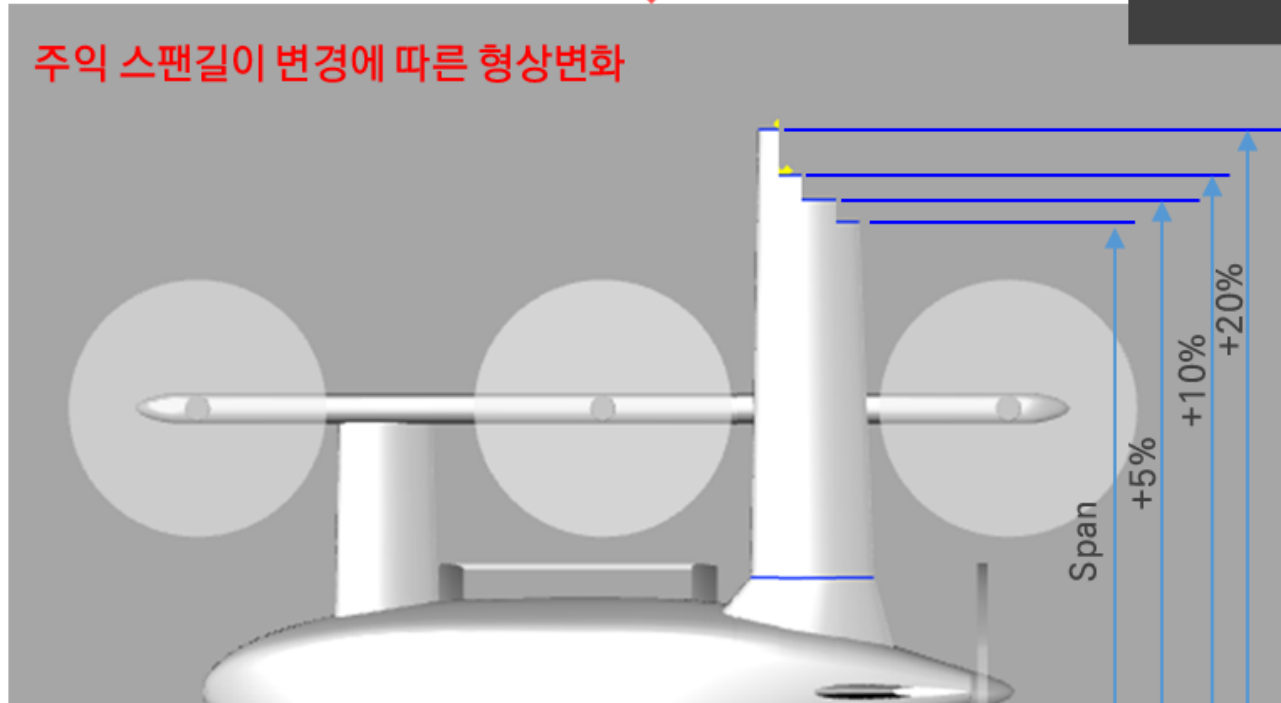
# 04

## 무인이동체 매개변수 최적설계 자동화

AFT Wing Parameters									
Position	X	2109	Z	164	Attack Angle	0			
	Airfoil	Chord	Thickness	Twist angle	Span	Sweepback	Dihedral	Buffer zone	Twist ref. chord
Root	sg6043	991	99	0	0	29	0	5	25%
Mid	sg6043	486	58	0	453				25%
Tip	sg6043	399	48	0	1400				0.3
Buffer Zone Shape		Chord_210811				AFT Control Surface		Use	FALSE
	W2-3	991	149						
Num Sections	10	486	75						
Blend Bulge	0.5	311	30						
Continuity	Tangent								



주익 스패ん길이 변경에 따른 형상변화





**감사합니다**

