

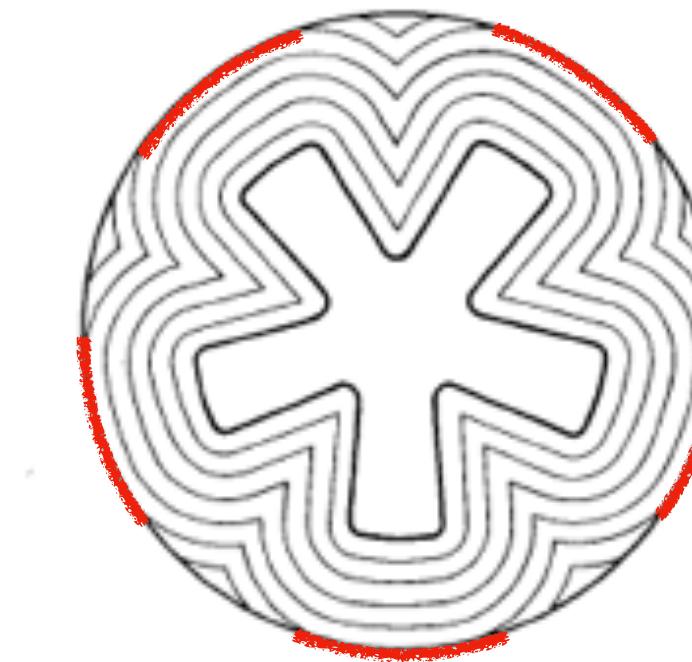
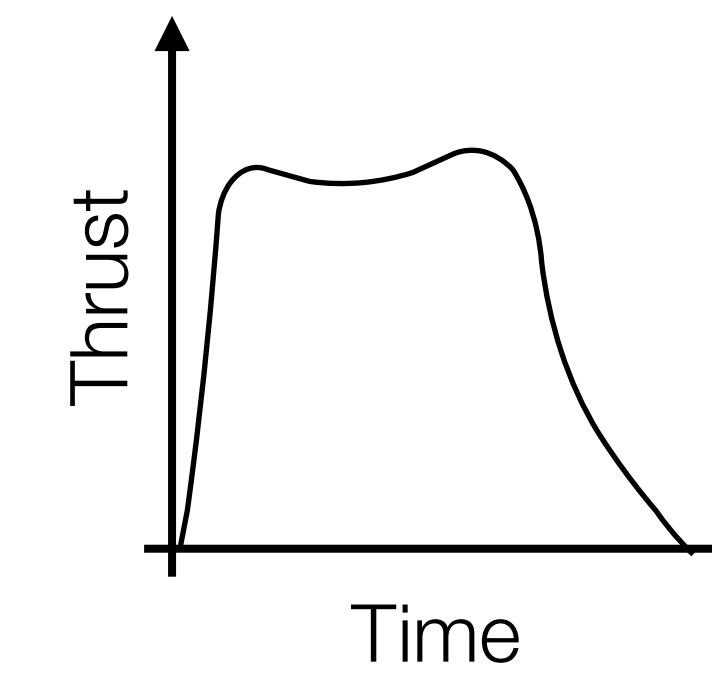
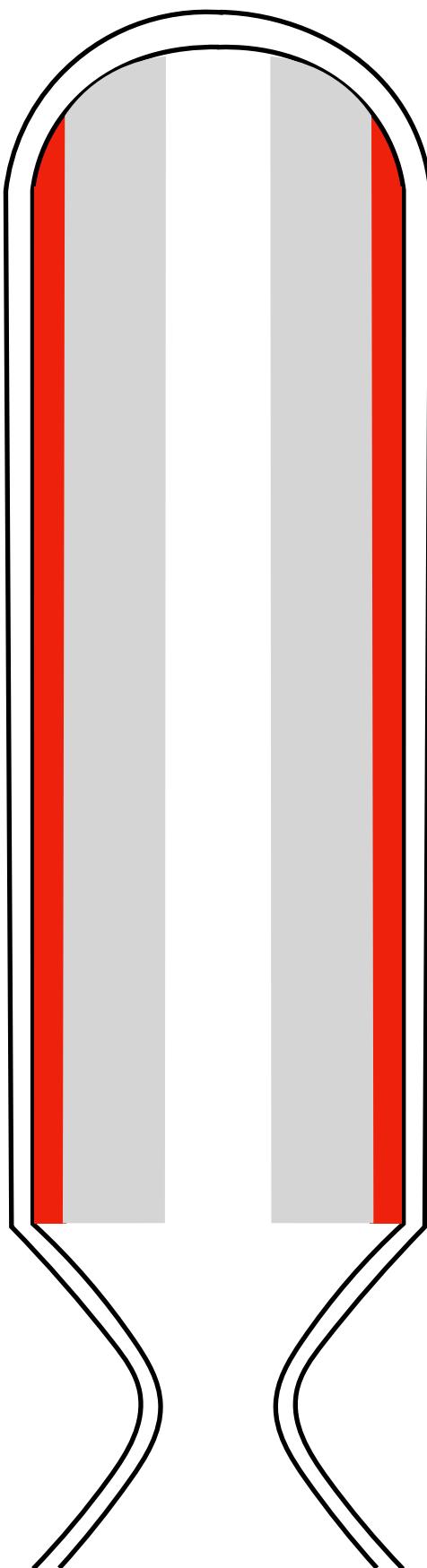
# **Sliding Basis Optimization for Heterogeneous Material Design**

**Nurcan Gecer Ulu, Svyatoslav Korneev, Erva Ulu, Saigopal Nelaturi**

Palo Alto Research Center



# Solid Rocket Propellant Design

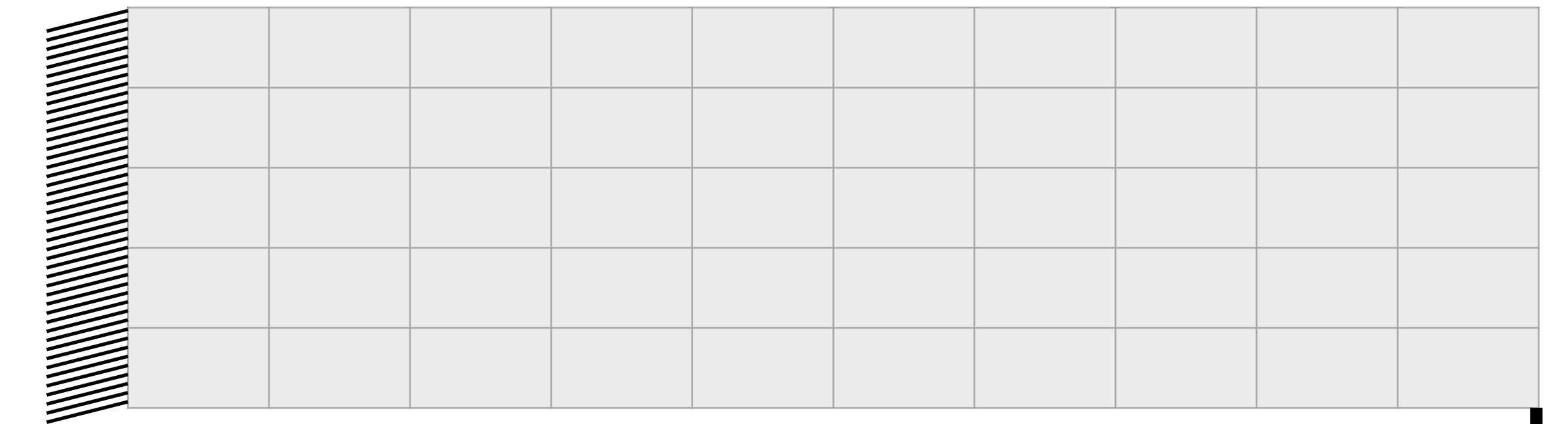


Geometry design with single material is not enough to achieve both desired thrust and eliminate insulation !!

# General Material Design Optimization Pipeline

$$\min_{\mathcal{F}} \quad f(\mathcal{F})$$

$$\text{s.t.} \quad g_i(\mathcal{F}) \leq 0$$



$$\Phi(\mathcal{F}) = 0$$

Costly analysis!

Complex physics!

Maybe black box!

Difficult to derive analytical gradients

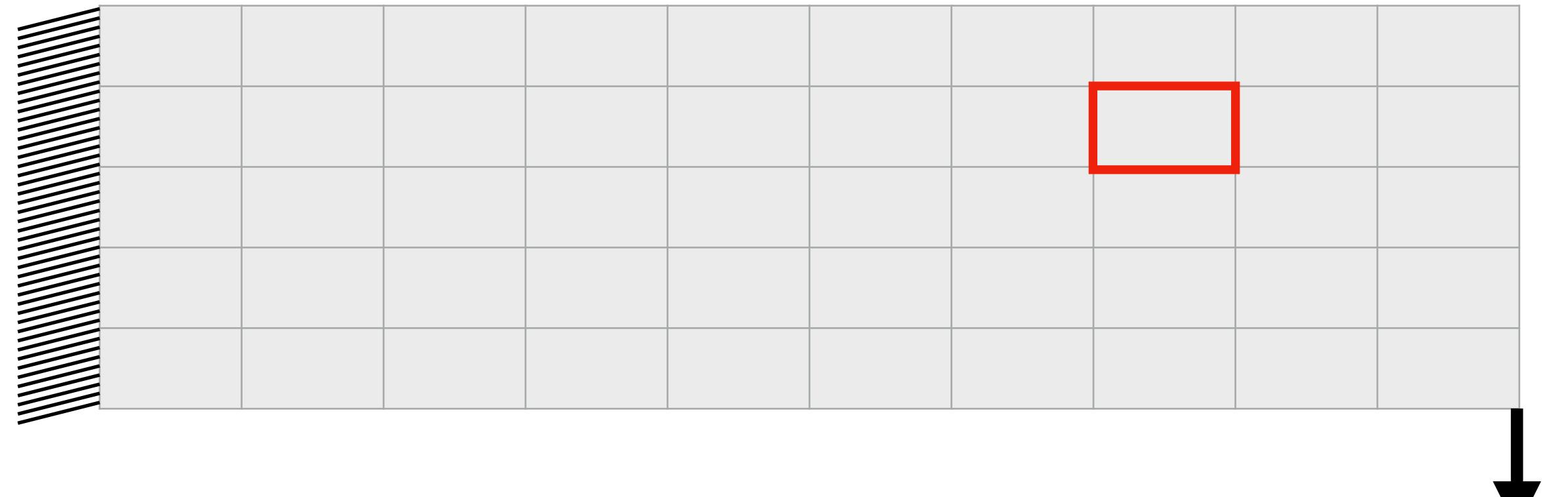
Numerical gradients are not practical  
for large scale problems

# Model reduction approach to reduce number of optimization variables

$$\begin{aligned} \min_{\mathcal{F}} \quad & f(\mathcal{F}) \\ \text{s.t.} \quad & g_i(\mathcal{F}) \leq 0 \end{aligned} \quad \rightarrow \quad \begin{aligned} \min_w \quad & f(w) \\ \text{s.t.} \quad & g_i(w) \leq 0 \\ \Phi(\mathcal{F}) = 0 \quad & \end{aligned}$$

thousand-millions

tens-hundreds



Represent the field as a combination of  
small set of basis functions!

$$\mathcal{F} = Bw$$

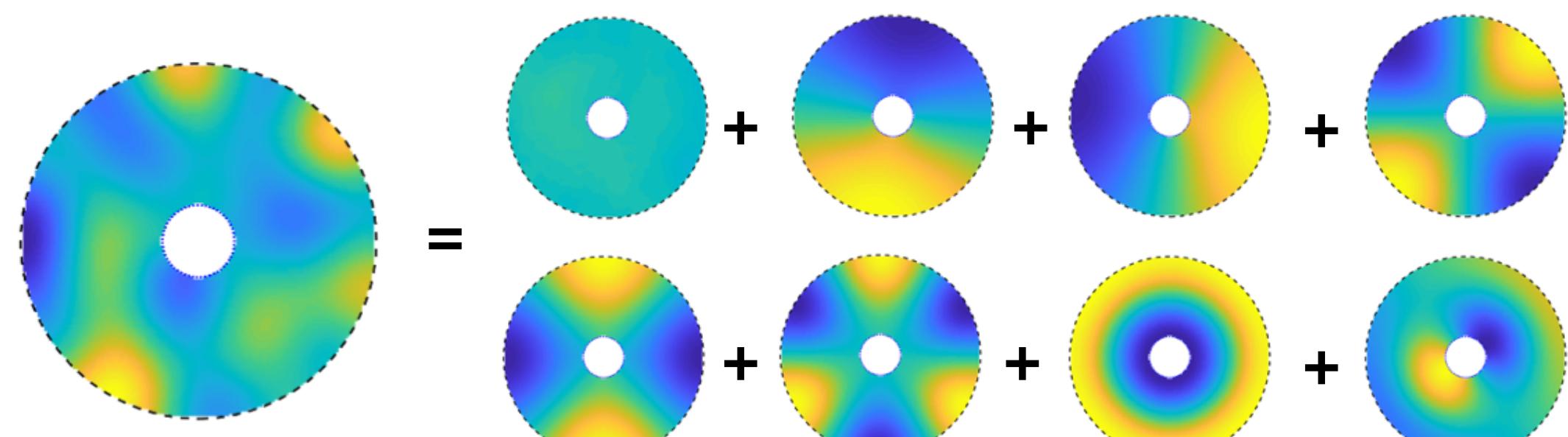
Reduce number of design variables  
without compromising analysis quality!

# Representing Material Distributions Using Shape Harmonics

- Fourier expansion of signal into harmonics ( $\sin + \cos$  functions) can be applied to manifold decomposition
- Fourier bases are eigenfunctions of the Laplacian on the unit interval
- Spherical harmonics are eigenfunctions of the Laplacian on the sphere

$$\lambda_i e_i = \mathcal{L} e_i \quad B = [e_1, e_2, \dots, e_k]$$

- Idea can be generalized to harmonics over any manifold
- Weighted sum of manifold harmonic basis can be used to describe shape and material in a ‘frequency domain’



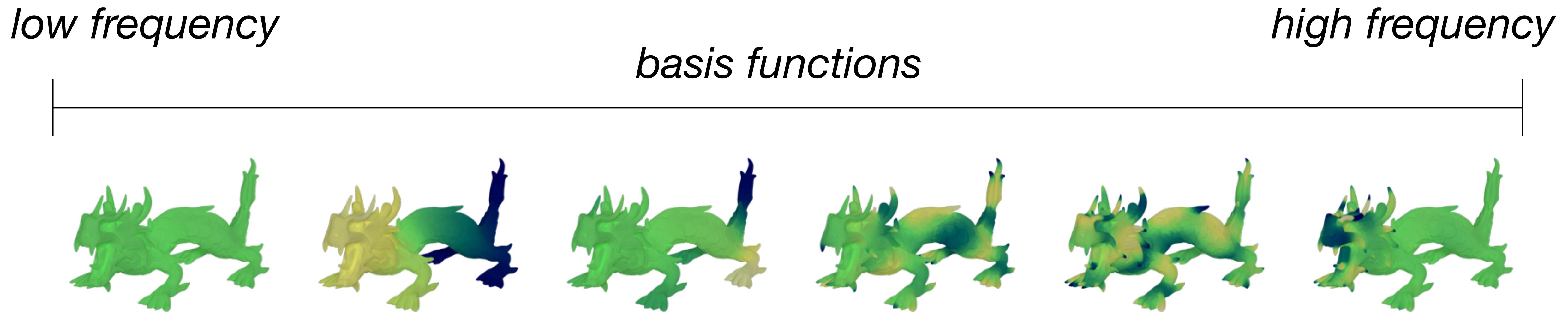
$$\mathcal{F} = Bw$$

10k quad mesh

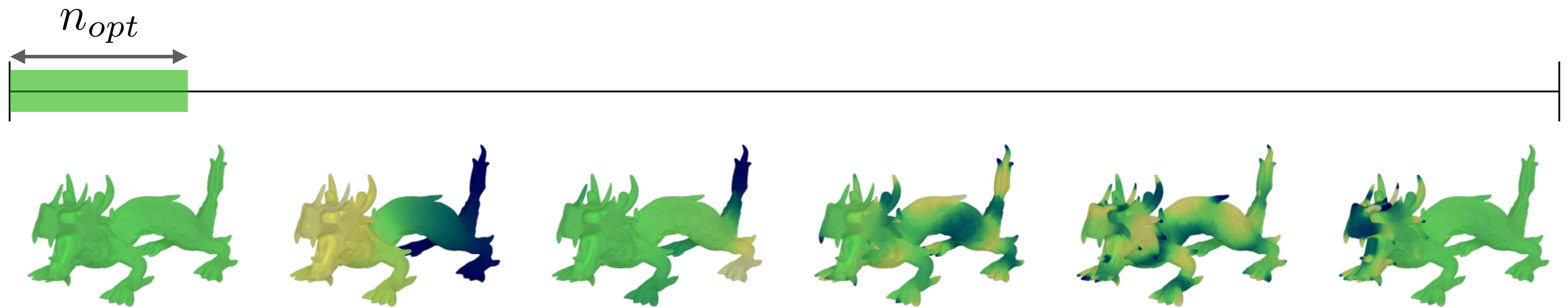
8 weights

We represent a field with fewer parameters  
using Laplacian basis

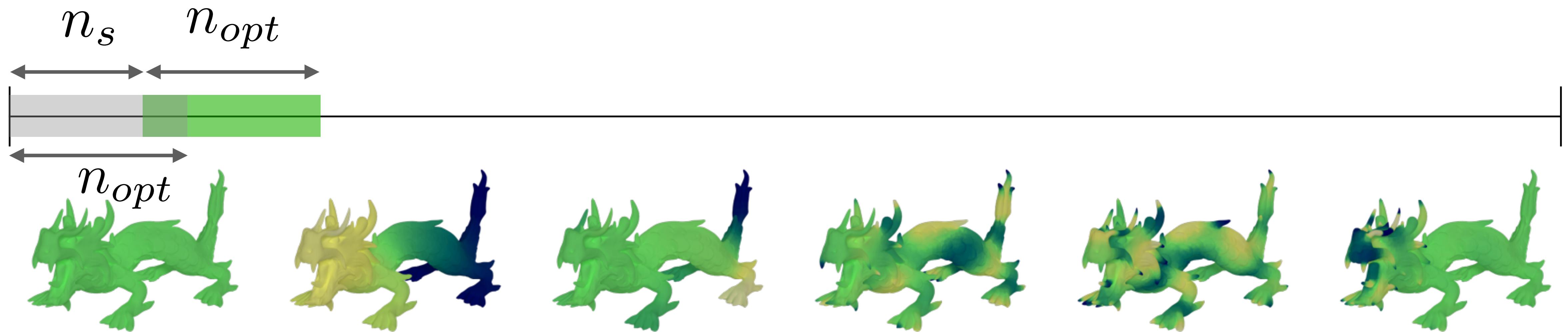
# Key Observation



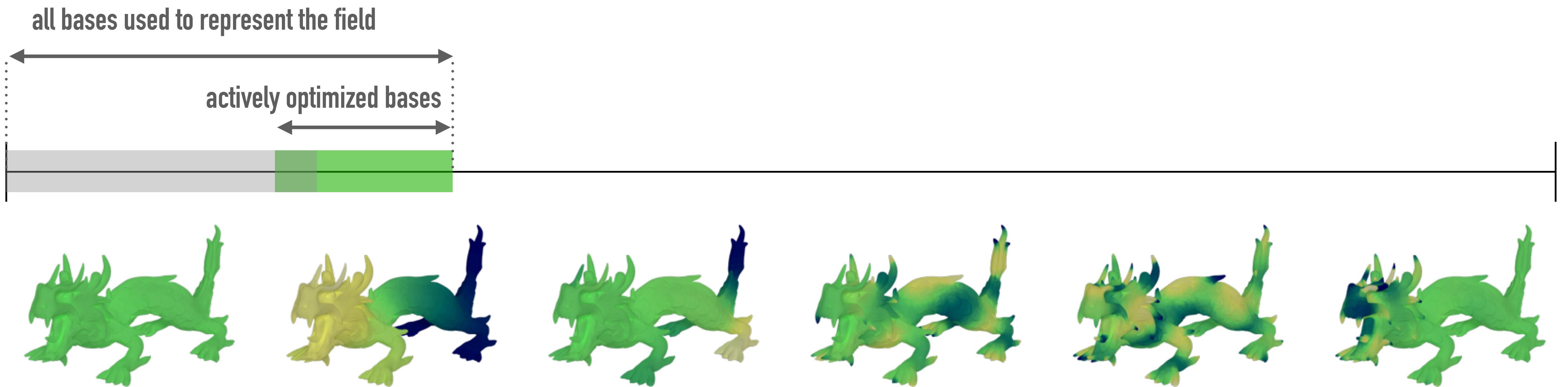
# Sliding Basis Optimization



# Sliding Basis Optimization



# Sliding Basis Optimization



We usually get convergence using only a small set of basis functions!

# Sliding basis optimization is a top level framework that works with existing optimization methods

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## Algorithm 1: Sliding basis optimization

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**Input:**  $n_{opt}, n_s, s_{max}$

**Output:** Optimized basis weights,  $w$

$i_{sb} \leftarrow 0$

▷ Index for the first active basis set

$it_s \leftarrow 0$

▷ Sliding iteration

$f \leftarrow 1/\epsilon$

▷ A large number

$w \leftarrow \emptyset$

▷ Optimized basis weights

**while** *not converged or*  $it_s < s_{max}$  **do**

$w_s \leftarrow \text{Initialize}()$       ▷ Weights for active basis functions

$(w_s, f_s) \leftarrow \text{Optimize}(i_{sb}, n_{opt})$

**if**  $f - f_s \geq \epsilon$  **then**

$w \leftarrow [w[0 : i_{sb}], w_s]$

$f \leftarrow f_s$

$it_s \leftarrow 0$

**else**

$w \leftarrow [w, \mathbf{0}]$

$it_s \leftarrow it_s + 1$

**end**

$i_{sb} \leftarrow i_{sb} + n_s$

**end**

This optimize step can be implemented  
using general nonlinear optimizers

# Sliding basis optimization speeds up differentiable problems, too

$$\begin{aligned} \min_{\boldsymbol{w}} \quad & f(\boldsymbol{w}) \\ \text{s.t.} \quad & g_i(\boldsymbol{w}) \leq 0 \end{aligned}$$

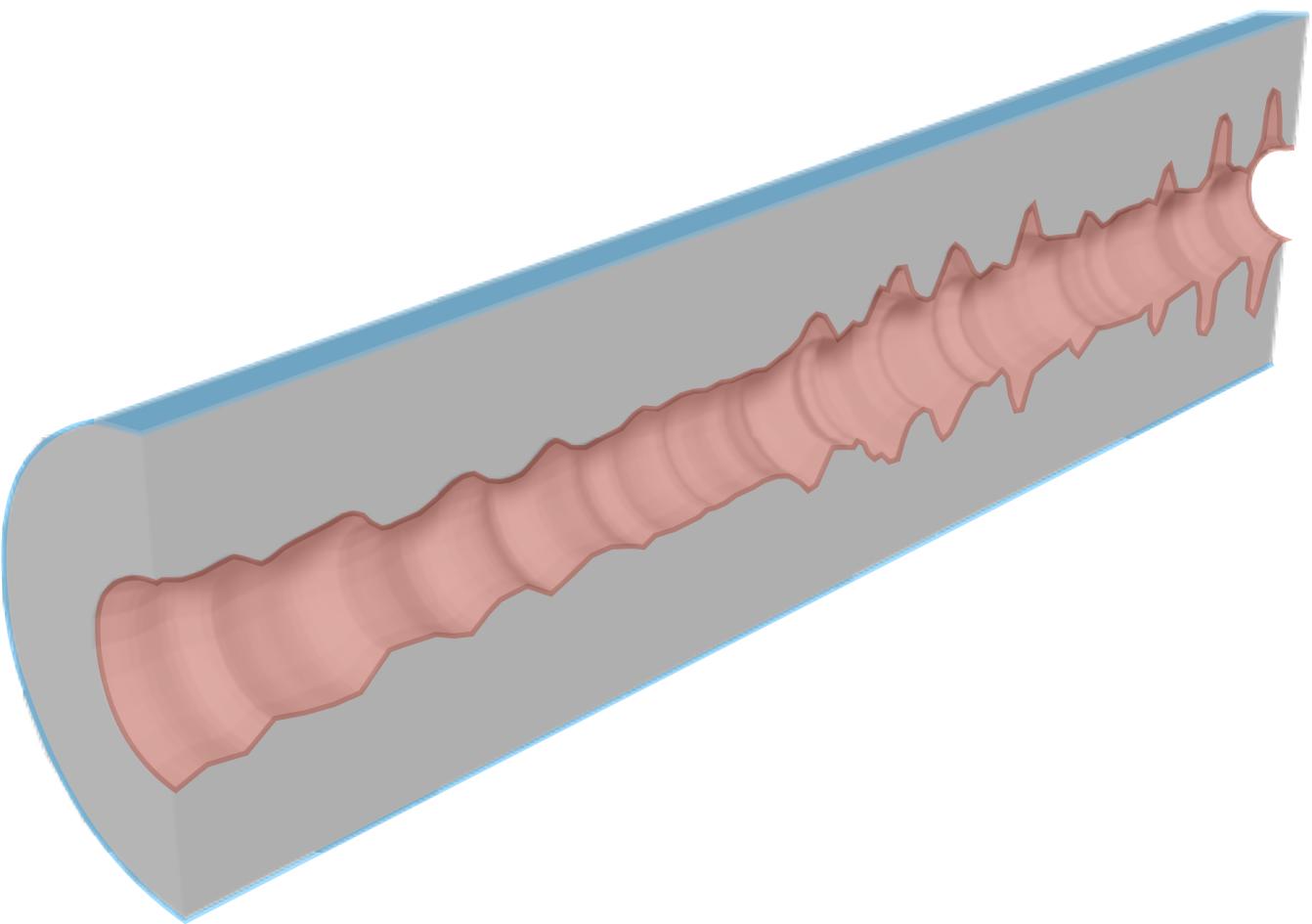
If gradients are derived  
for the full resolution

$$\frac{\partial f}{\partial \boldsymbol{w}} = \boxed{\frac{\partial f}{\partial \mathcal{F}} \frac{\partial \mathcal{F}}{\partial \boldsymbol{w}}}$$

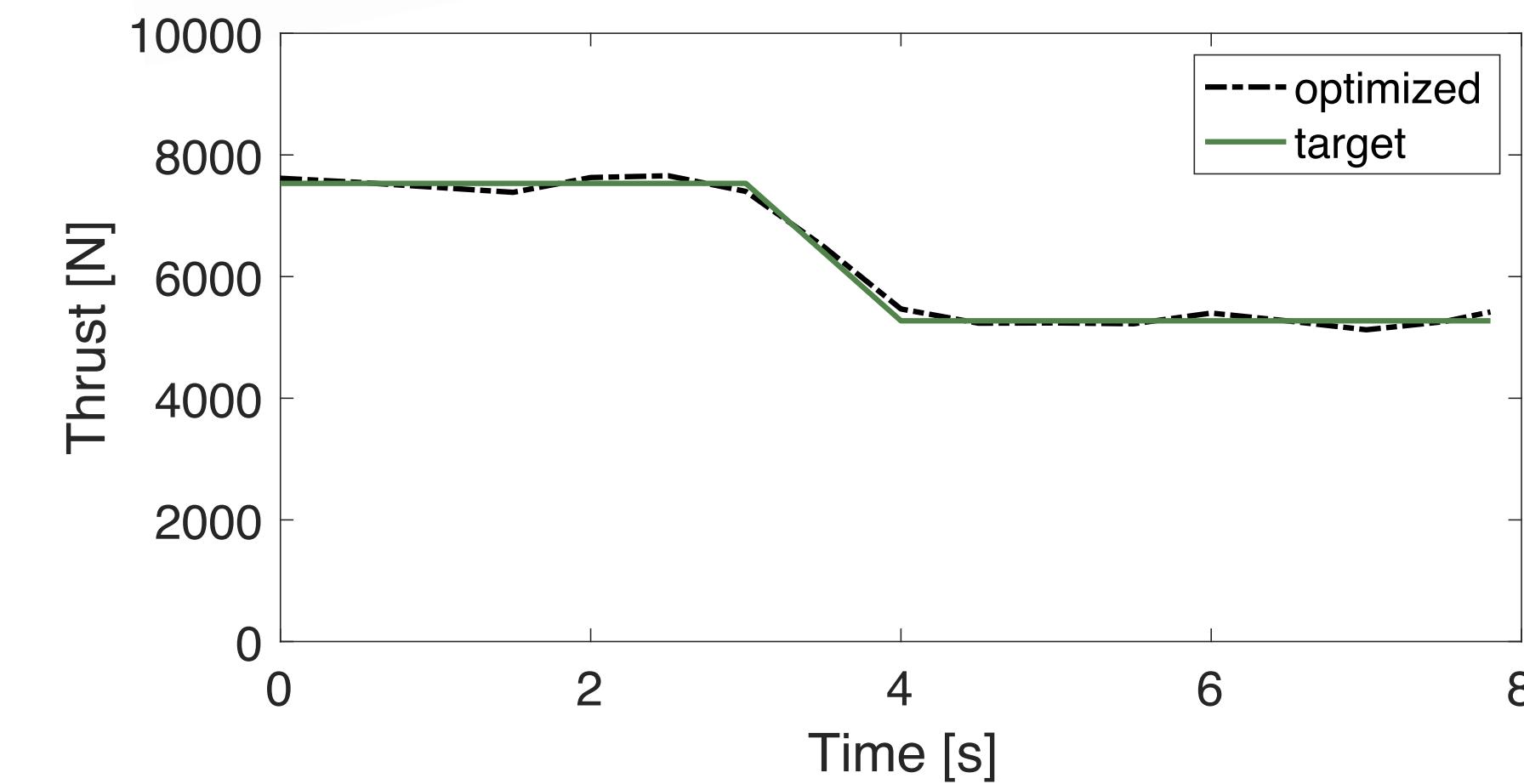
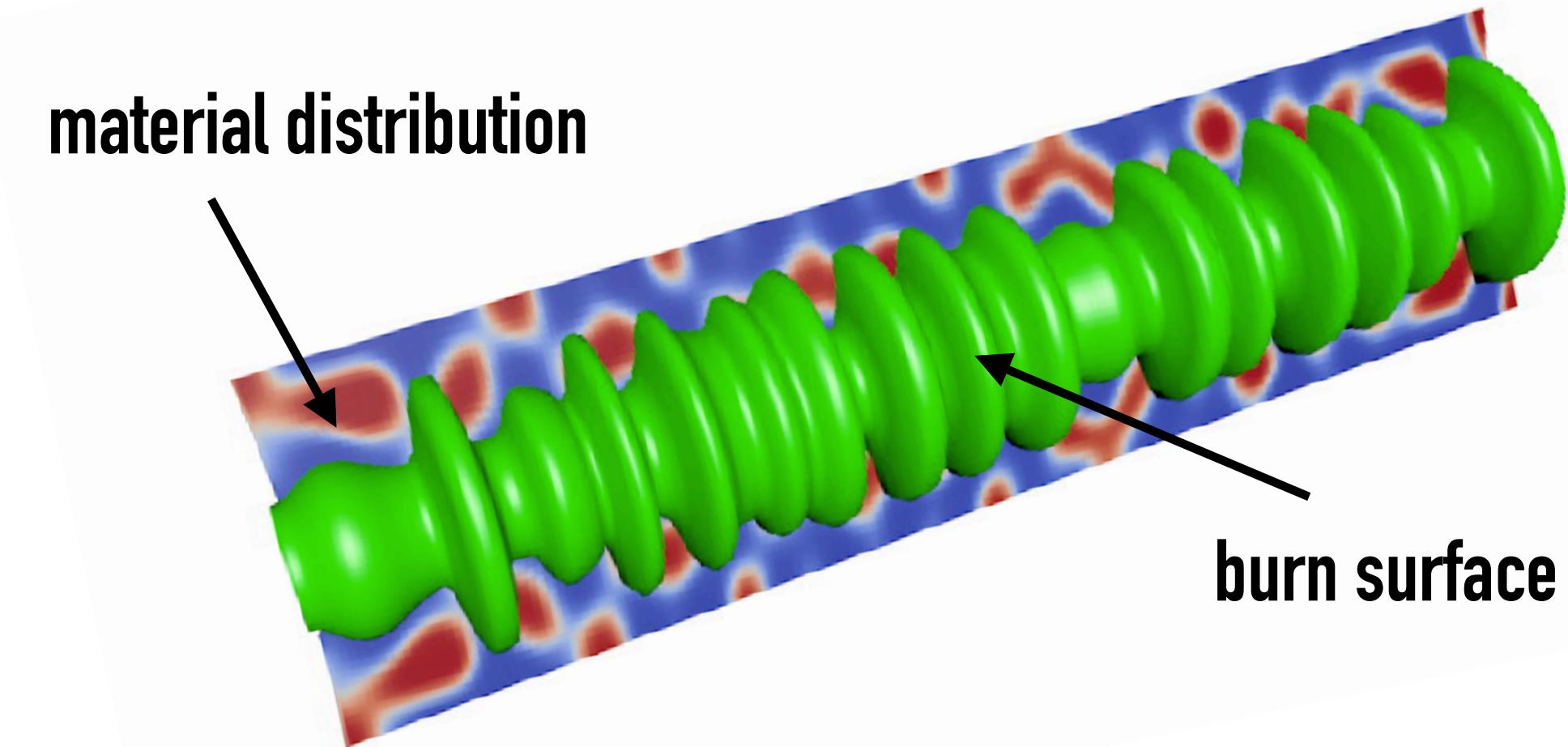
Gradients w.r.t. basis weights can be found  
through simple matrix multiplication with  $\mathbf{B}$

$$\mathcal{F} = \mathbf{B}\boldsymbol{w}$$

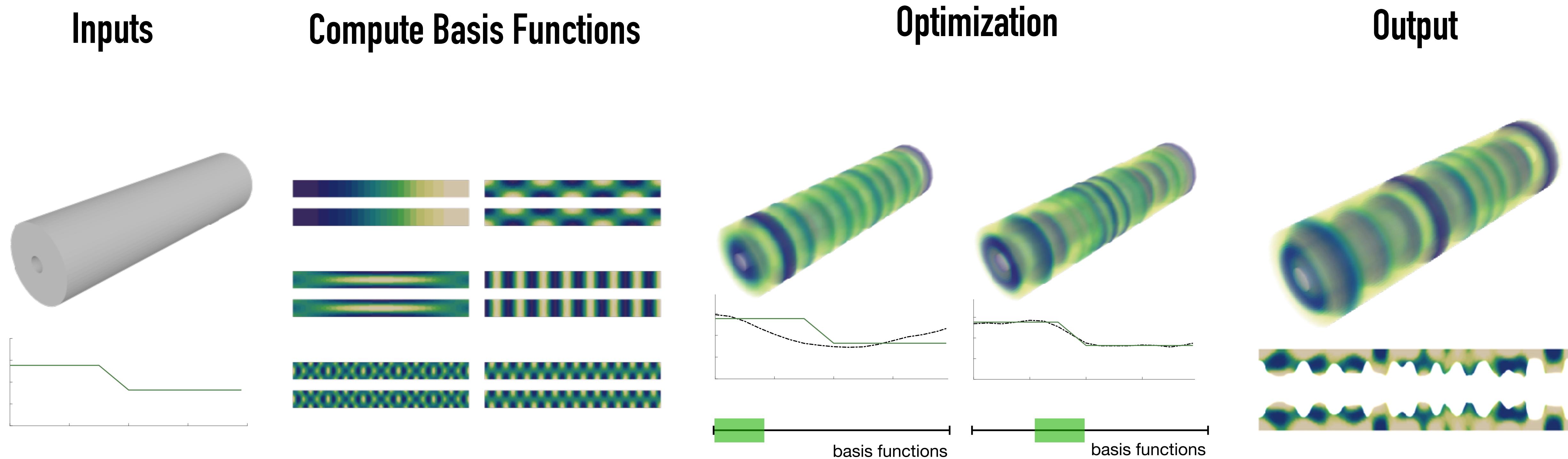
# Solid Rocket Fuel Design



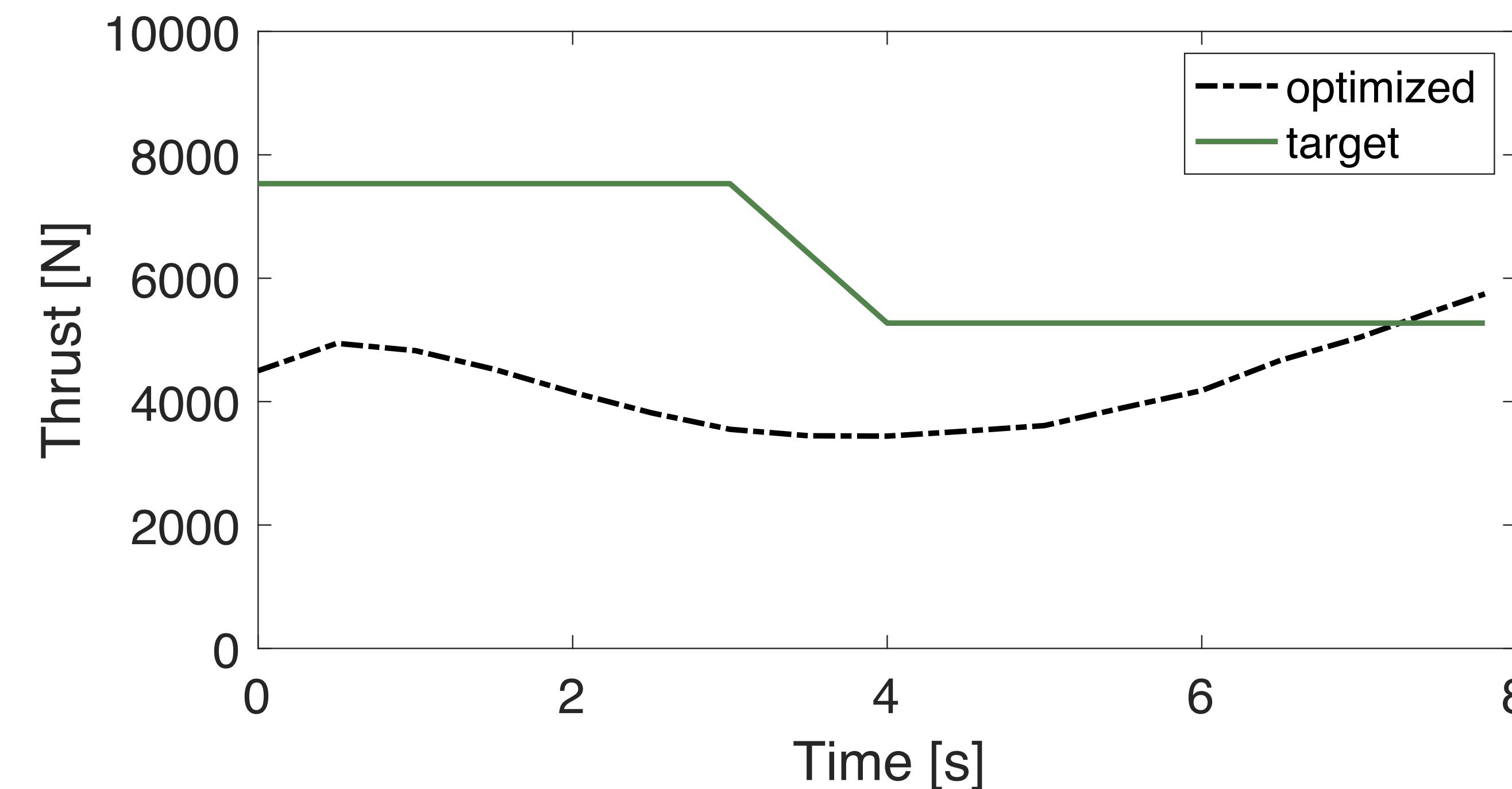
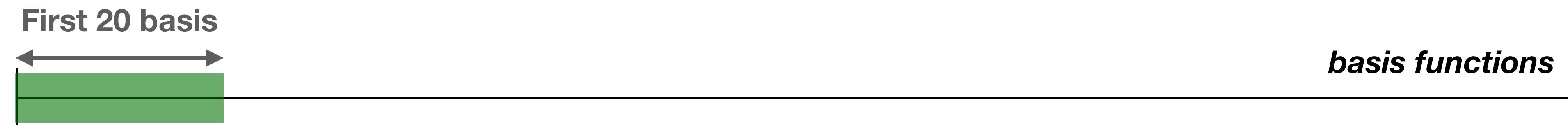
$$\begin{aligned} \min_{\boldsymbol{w}} \quad & \sum_t (th(\boldsymbol{w}) - th_{target})^2 \\ \text{s.t.} \quad & r_b(\boldsymbol{w})^i > r_{in} \end{aligned}$$



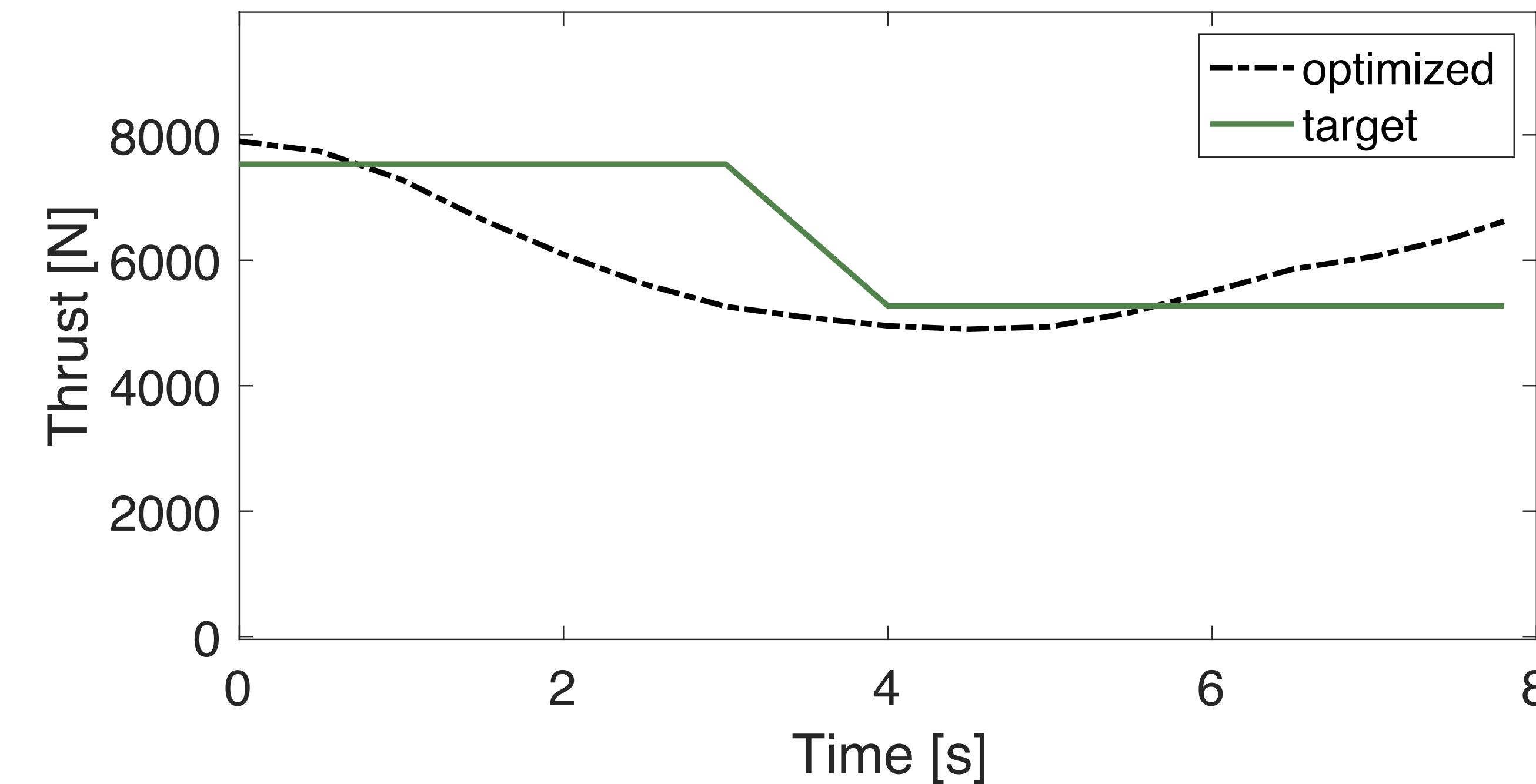
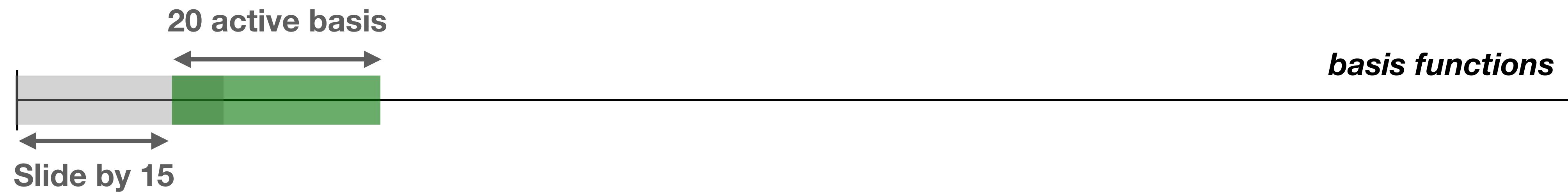
# Sliding Basis Optimization



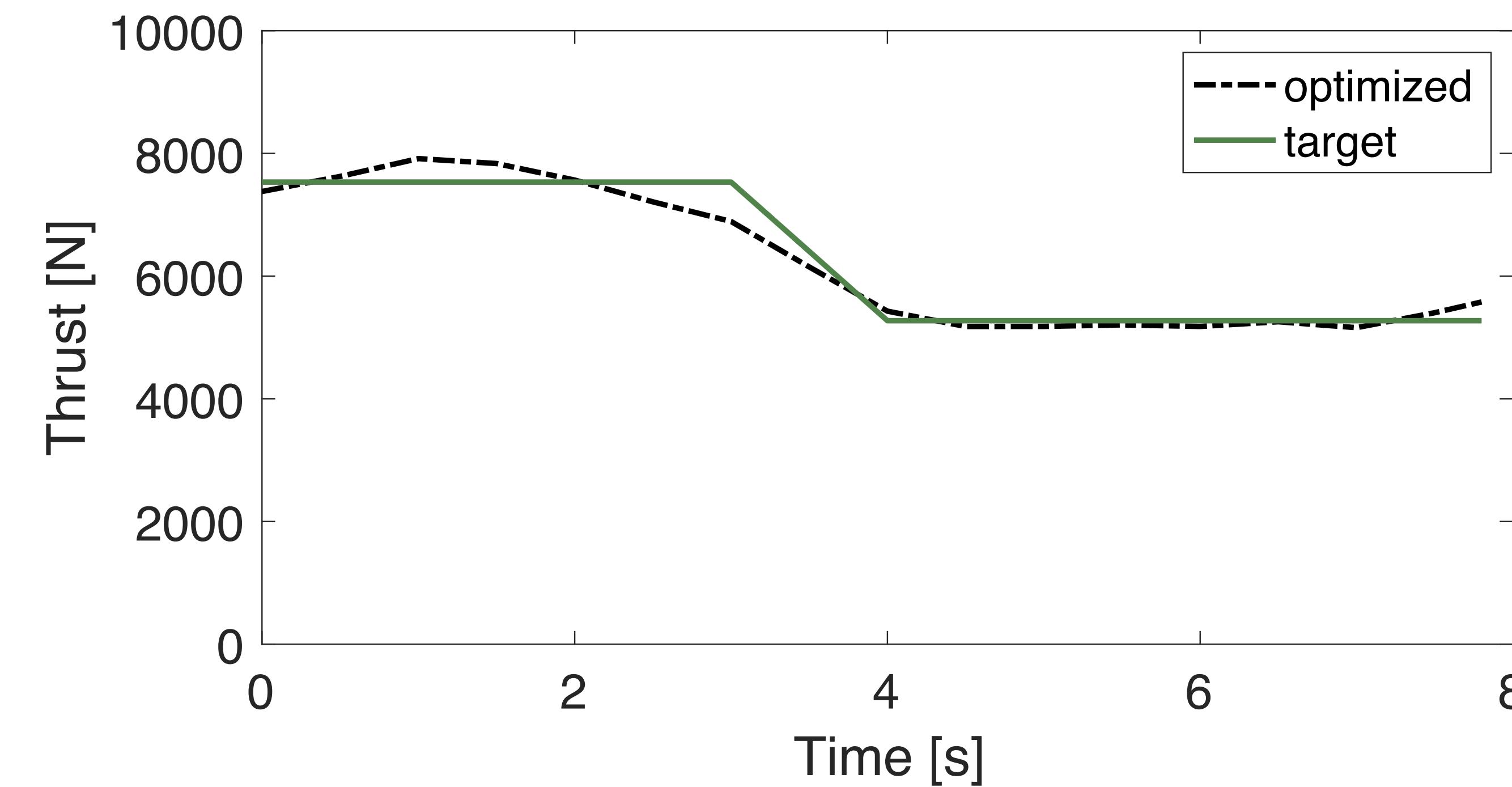
# Progression of thrust profile match through the sliding basis optimization



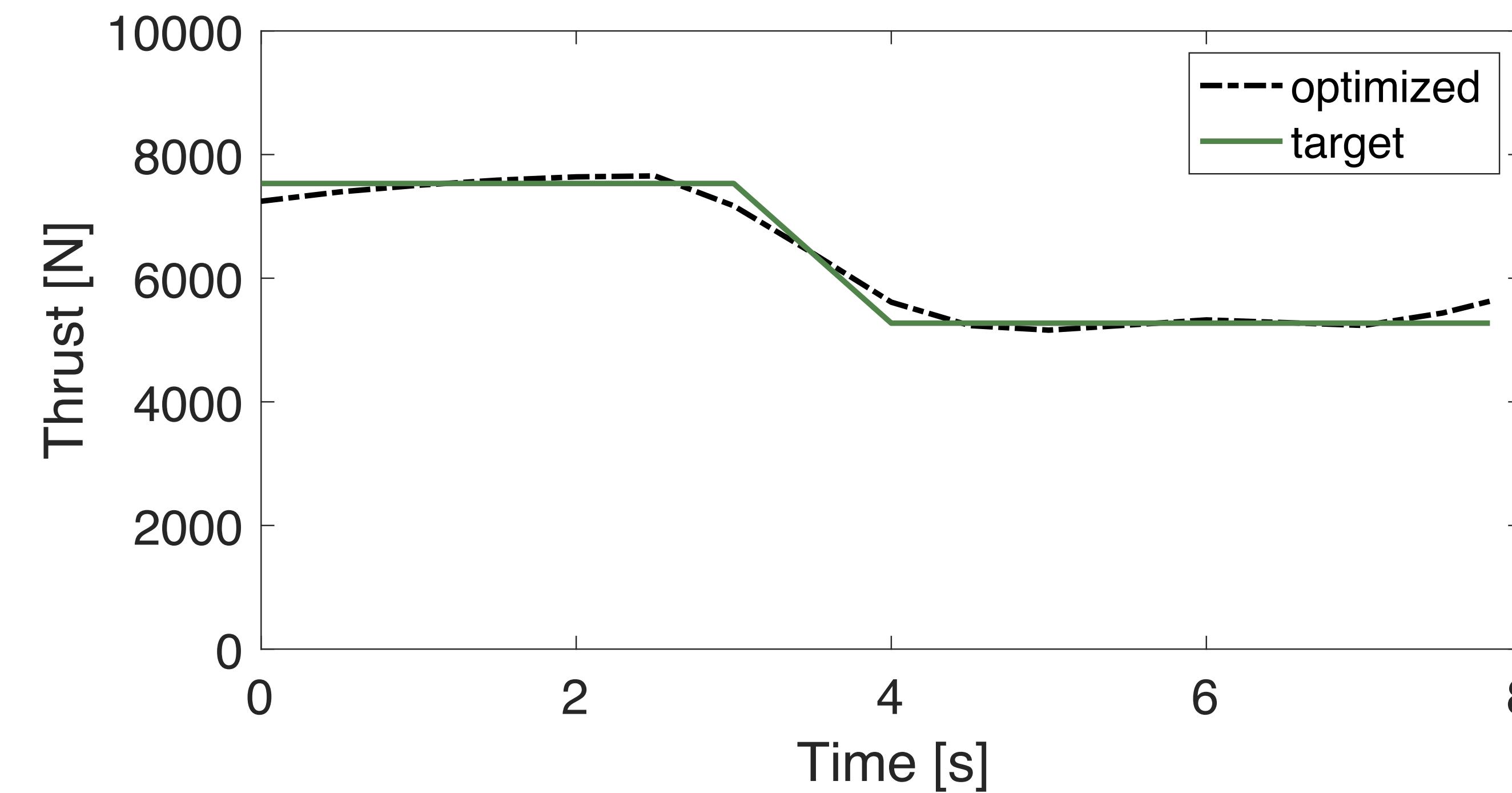
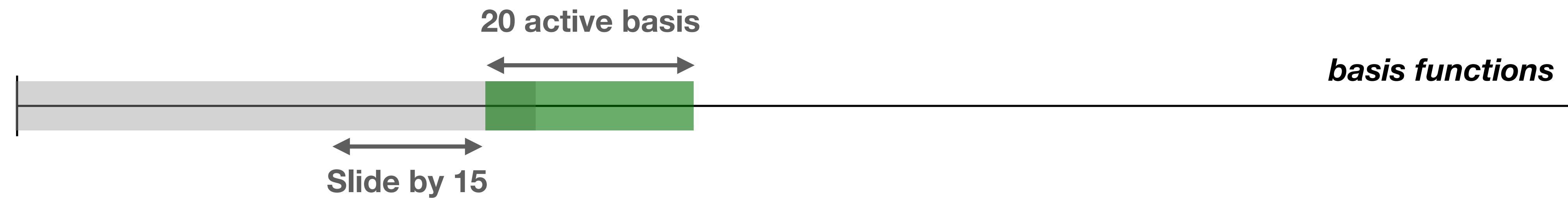
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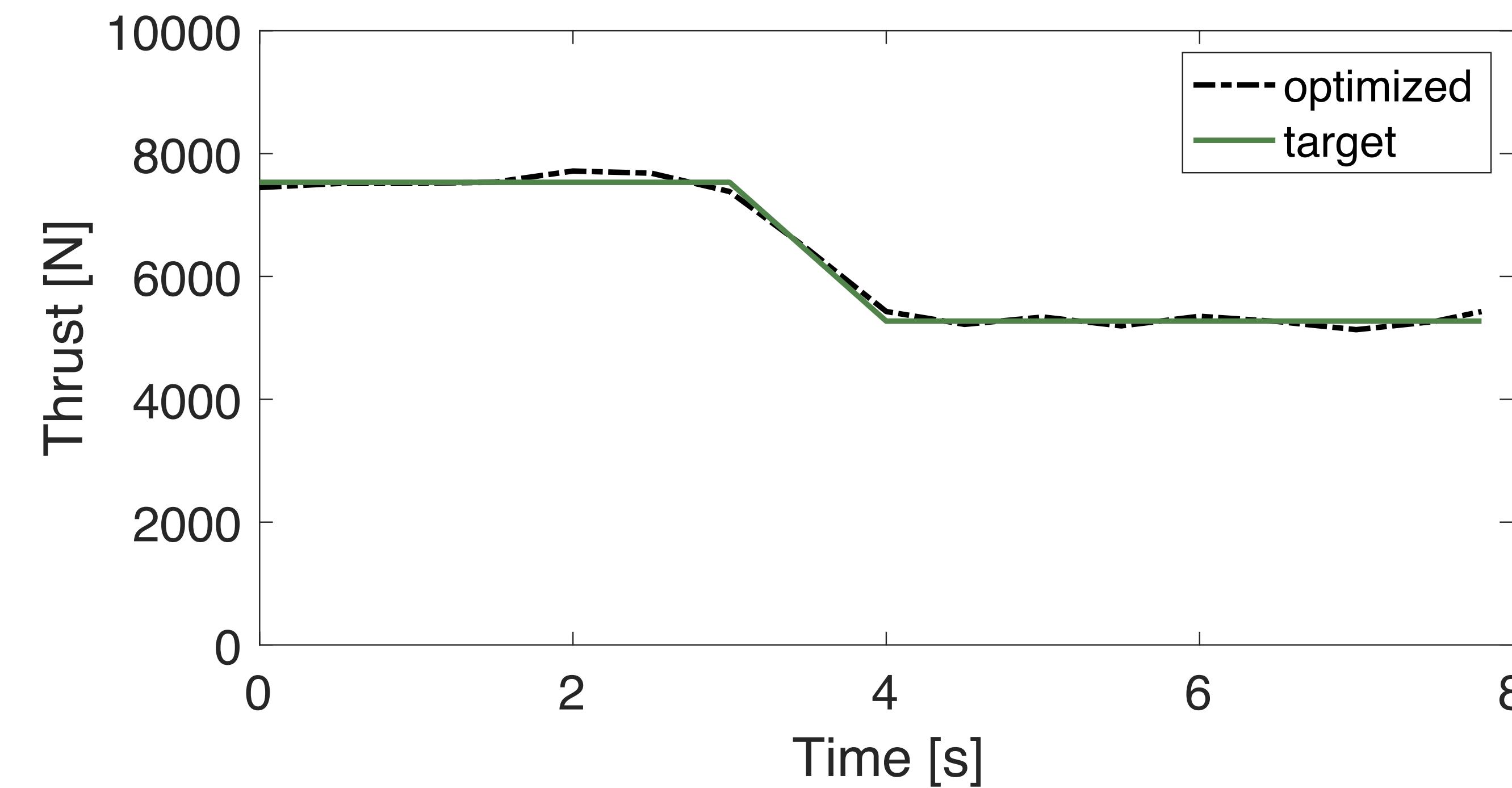
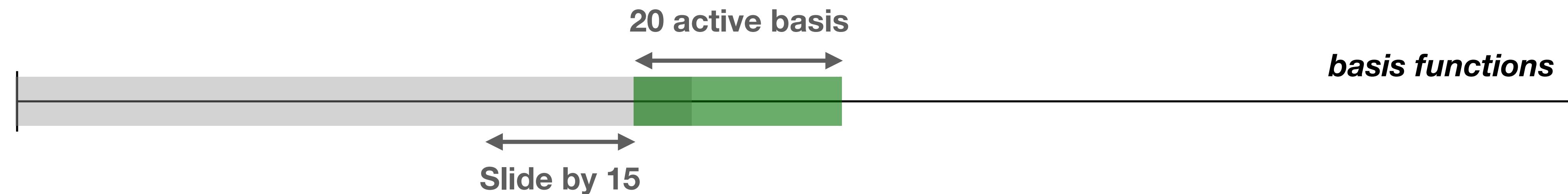
# Progression of thrust profile match through the sliding basis optimization



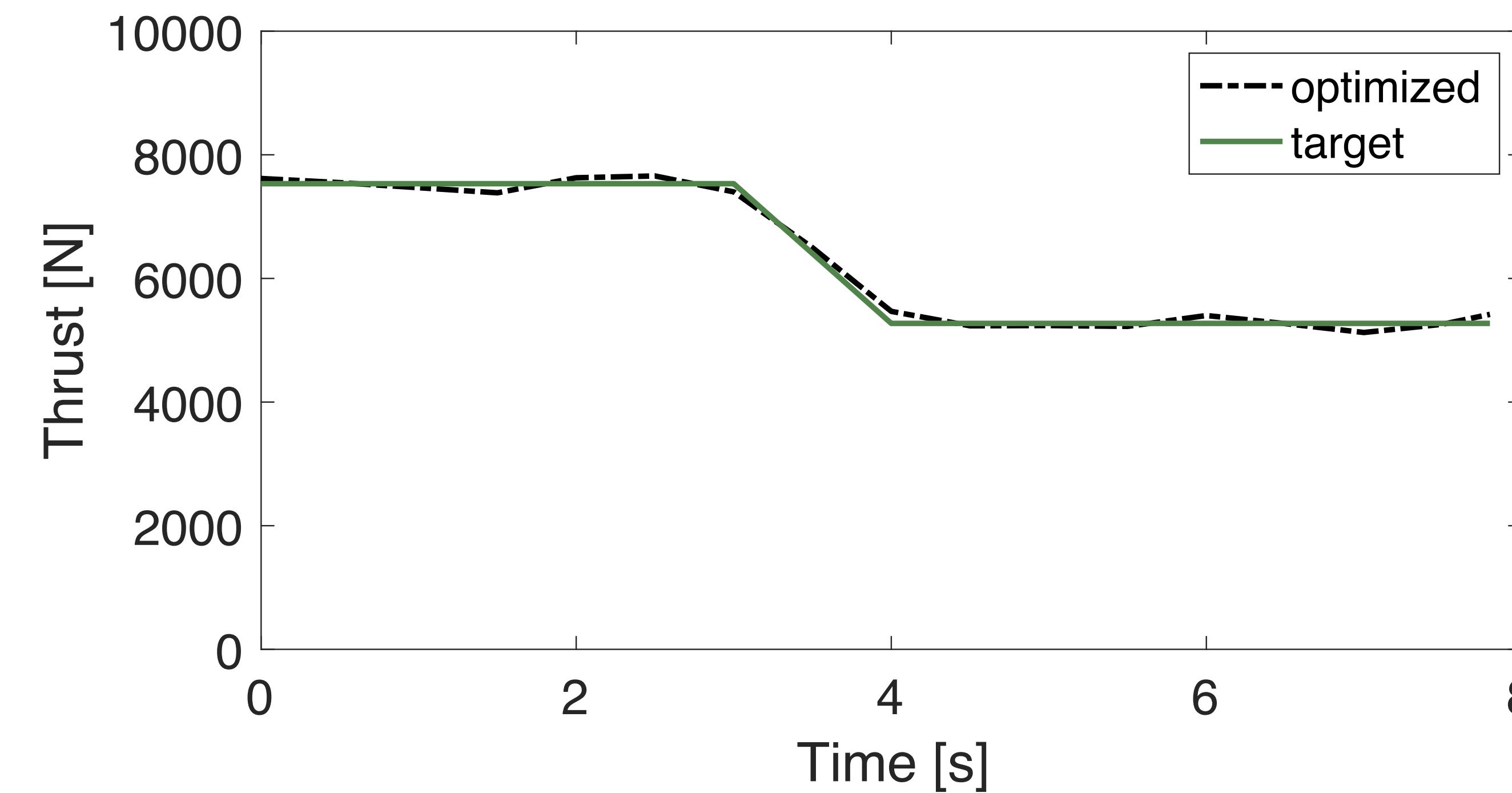
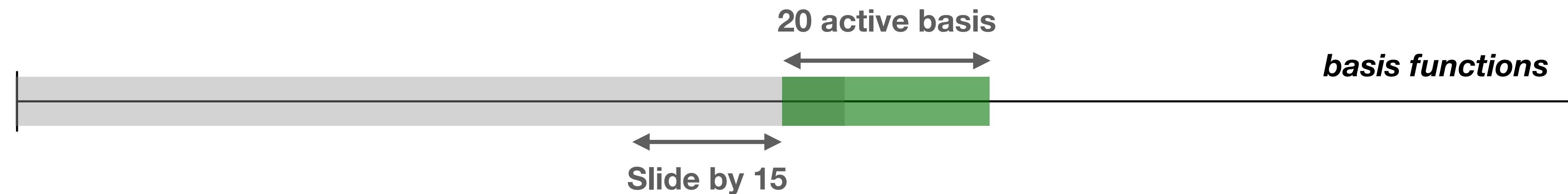
# Progression of thrust profile match through the sliding basis optimization



# Progression of thrust profile match through the sliding basis optimization

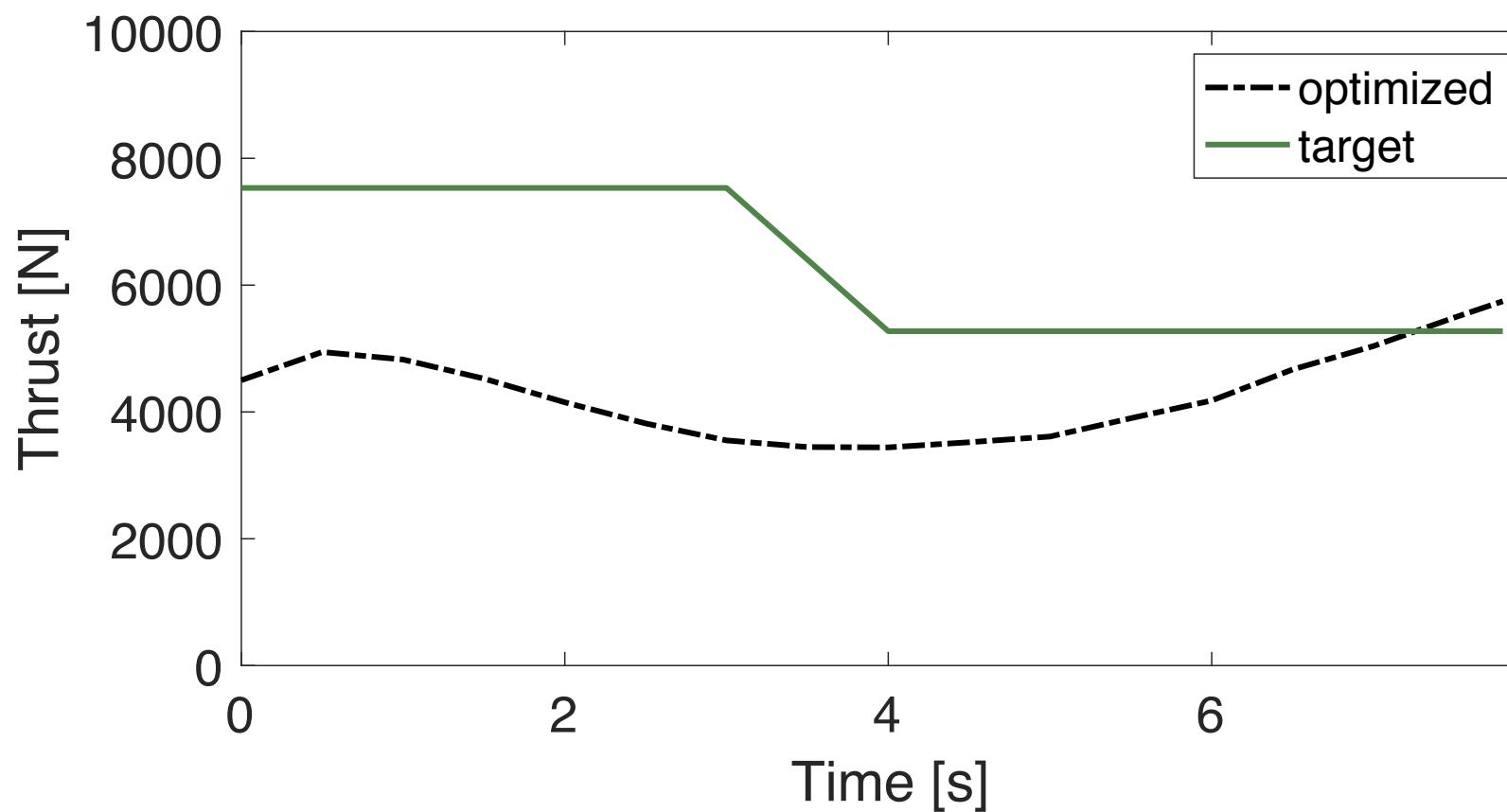


# Progression of thrust profile match through the sliding basis optimization

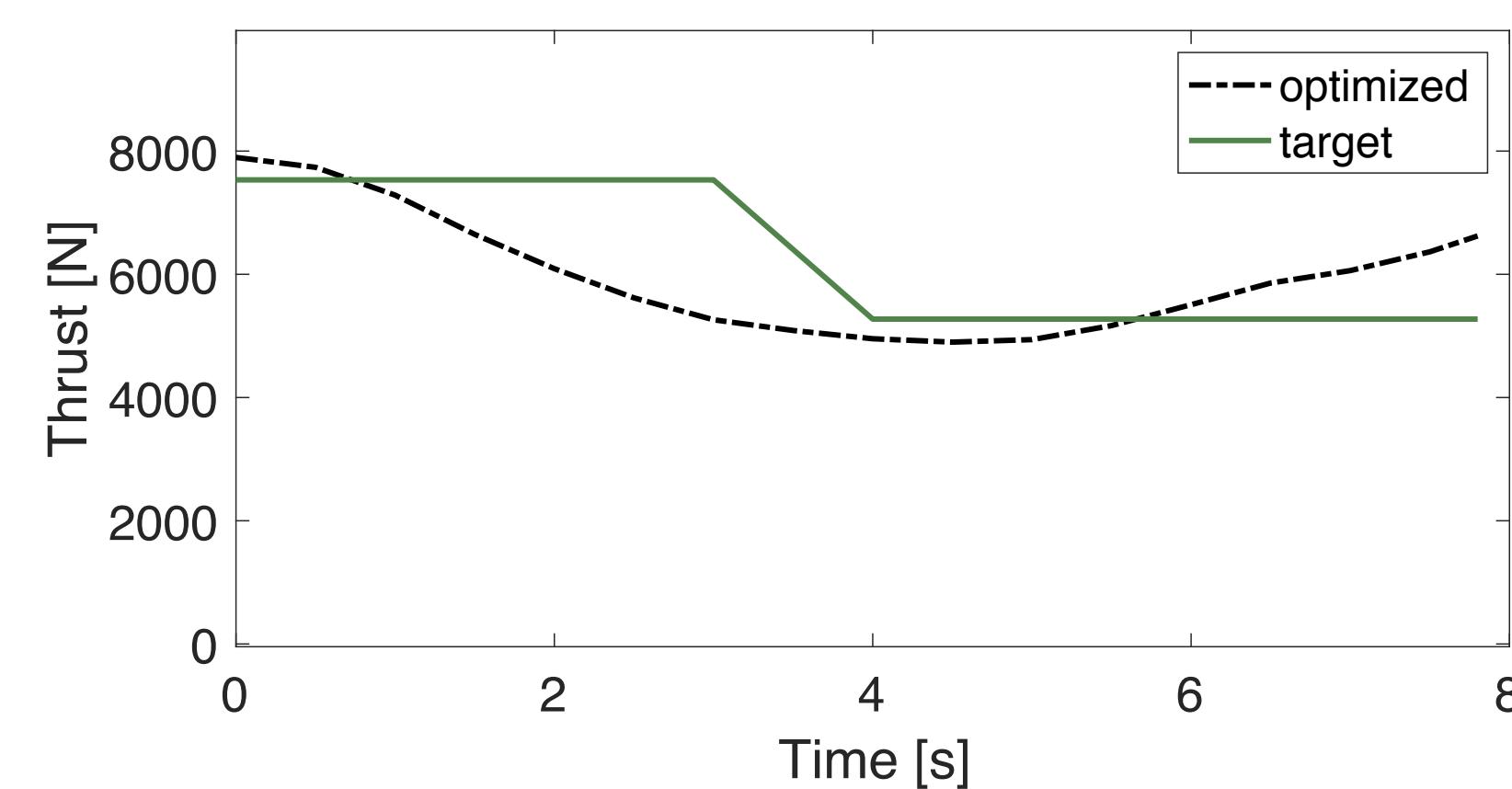


# Progression of thrust profile match through the sliding basis optimization

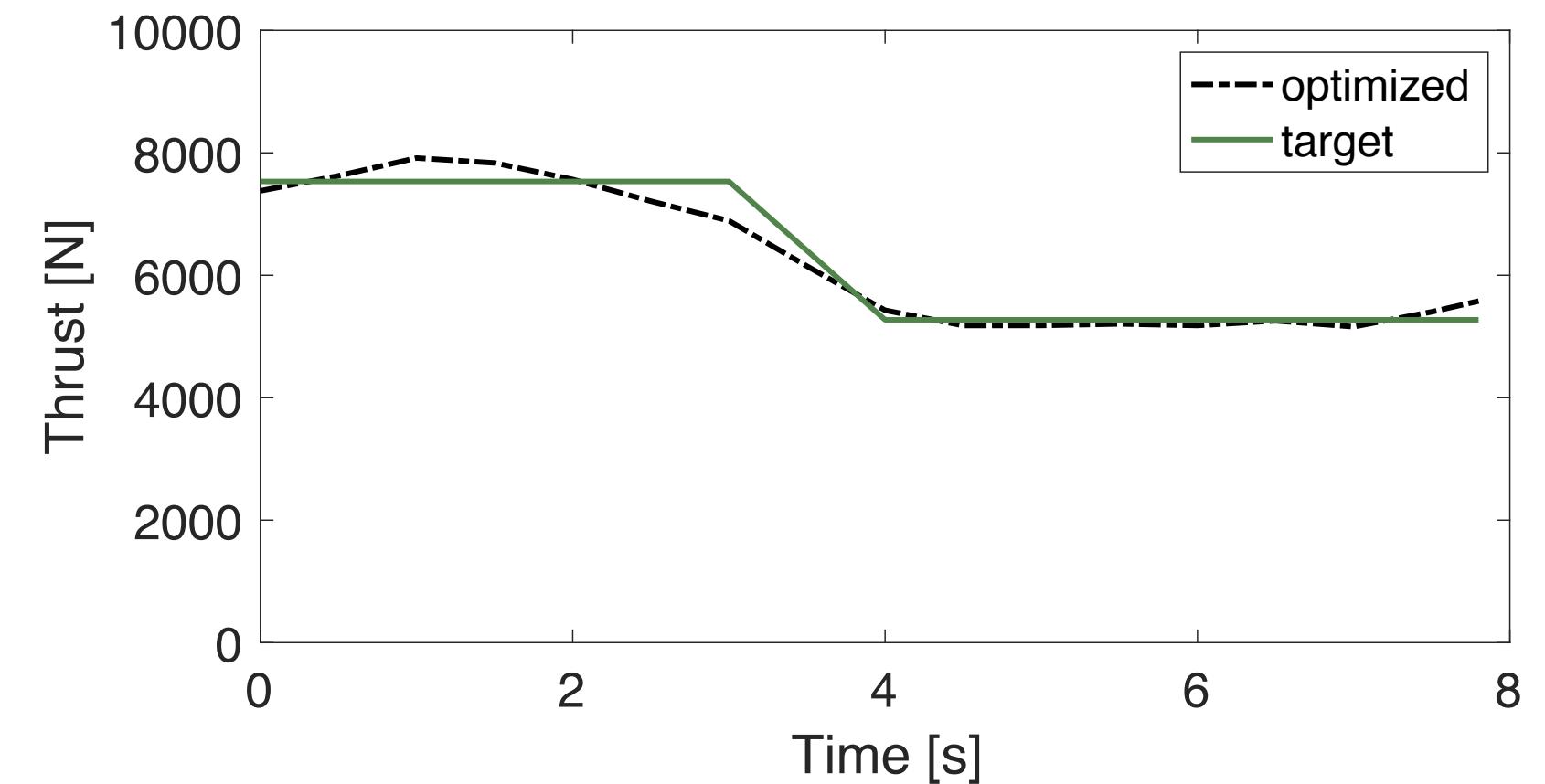
First 20 basis



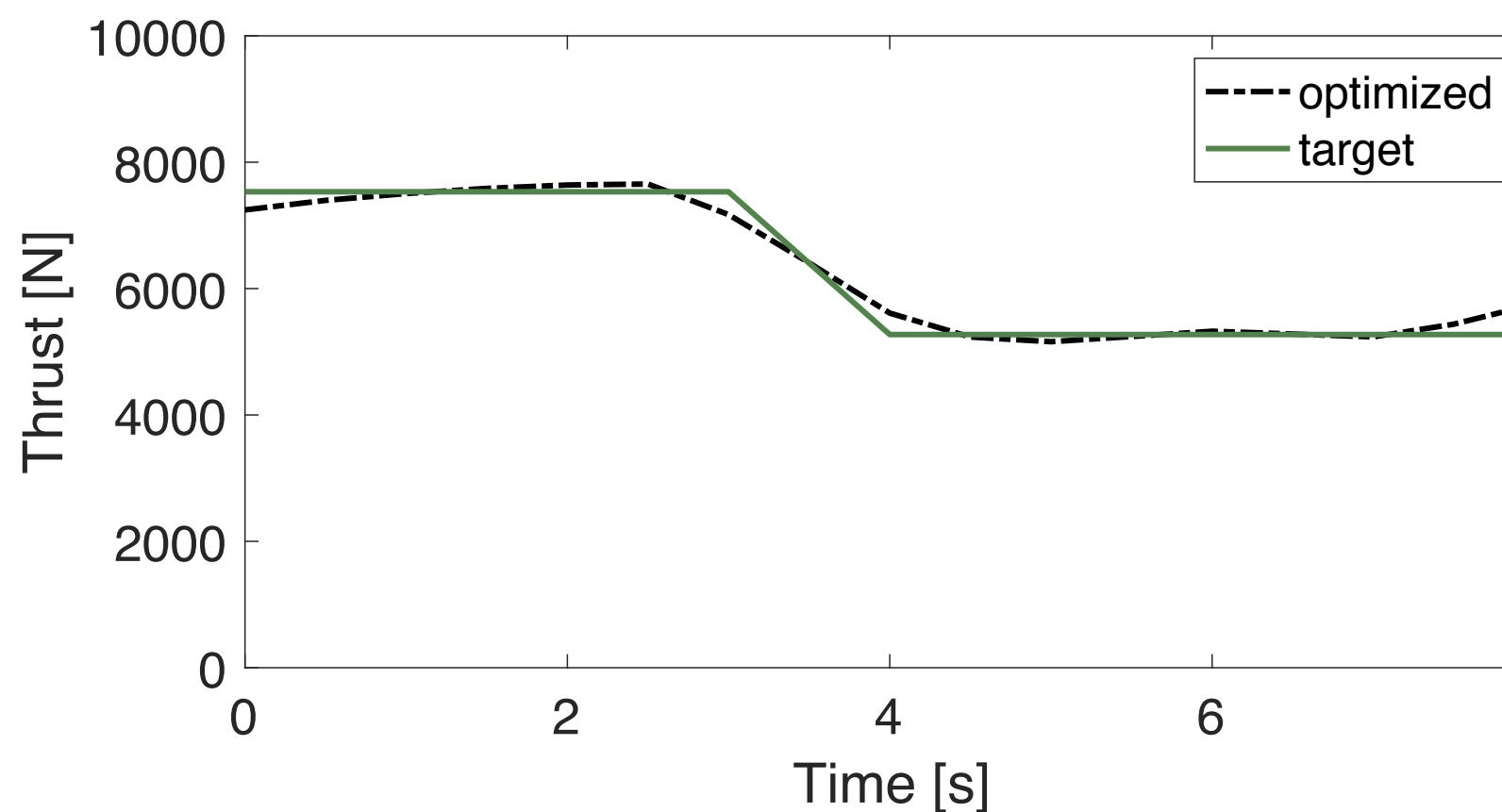
Slide by 15



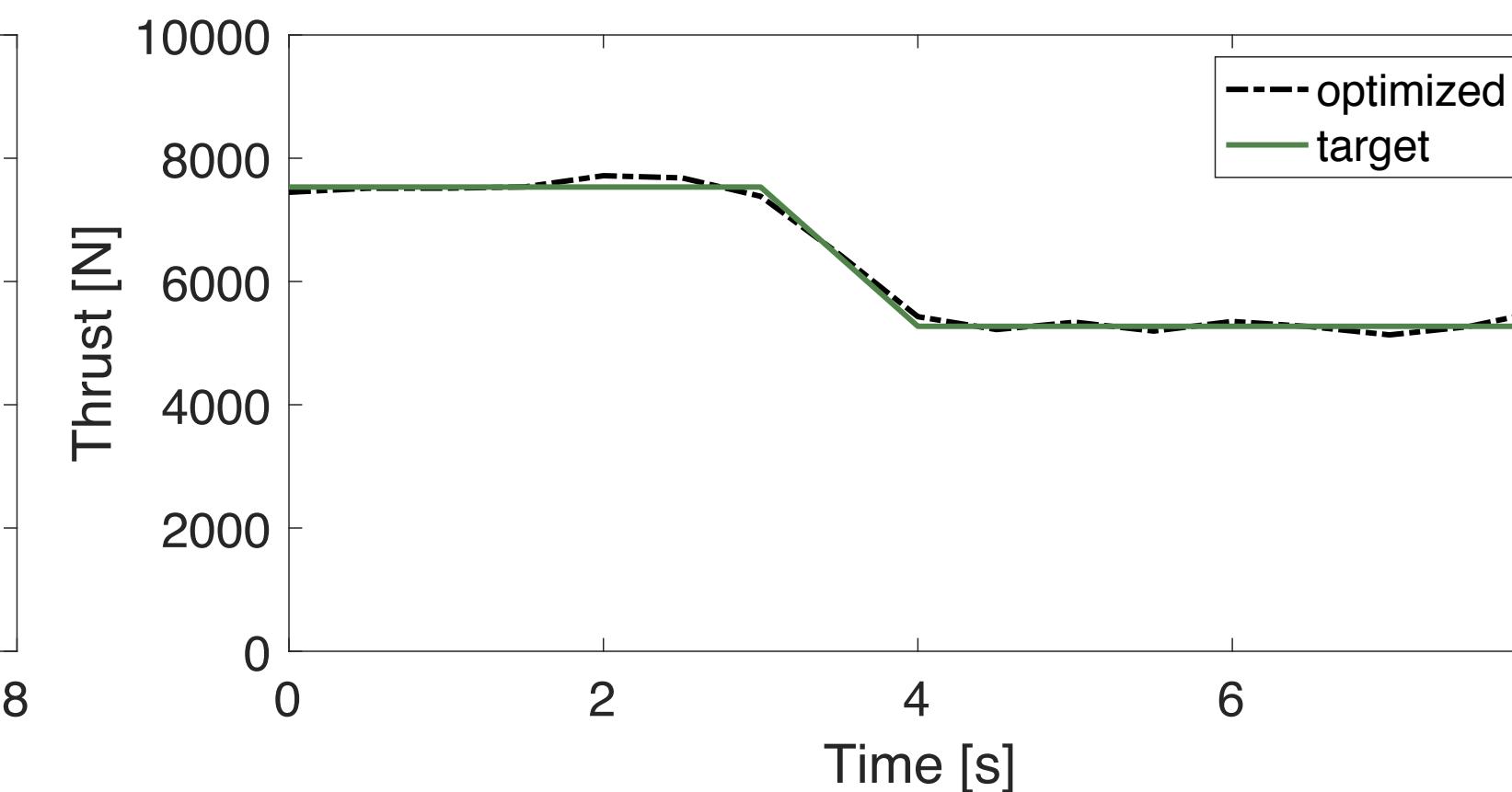
Slide by 15



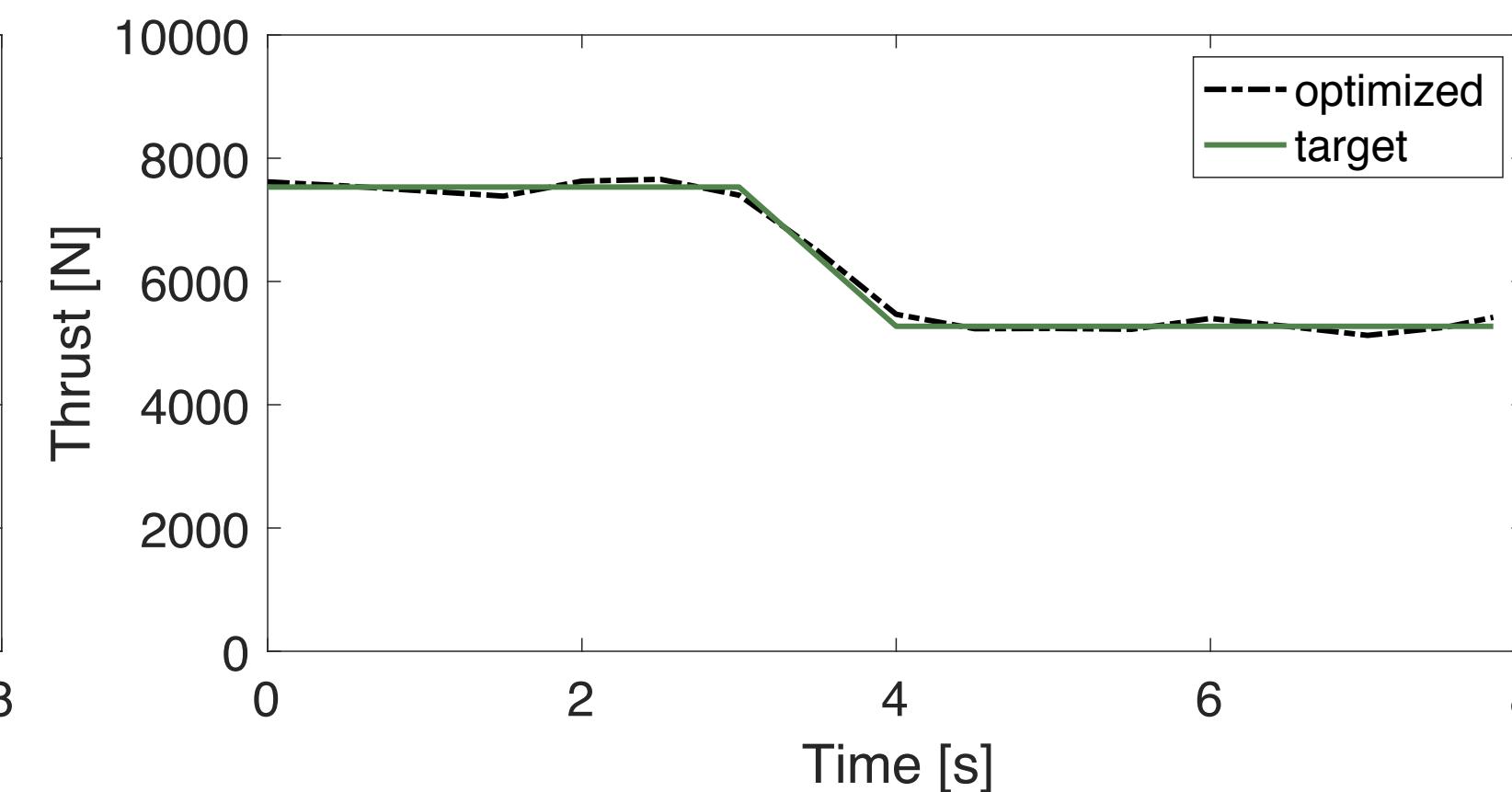
Slide by 15



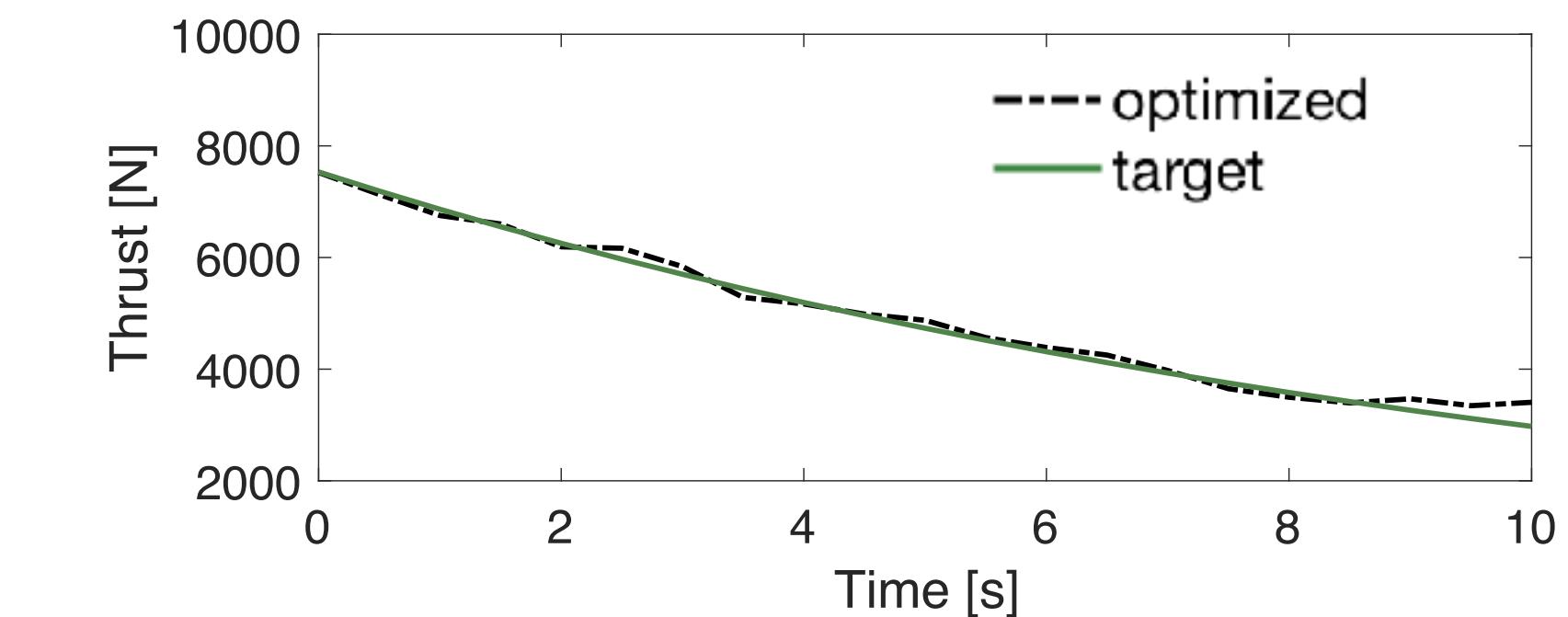
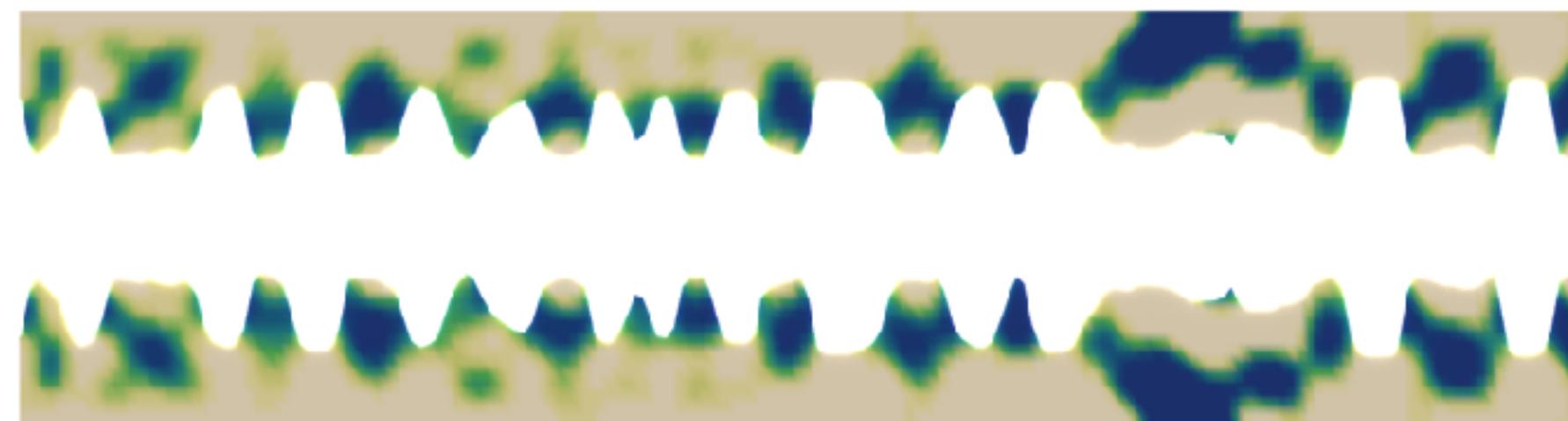
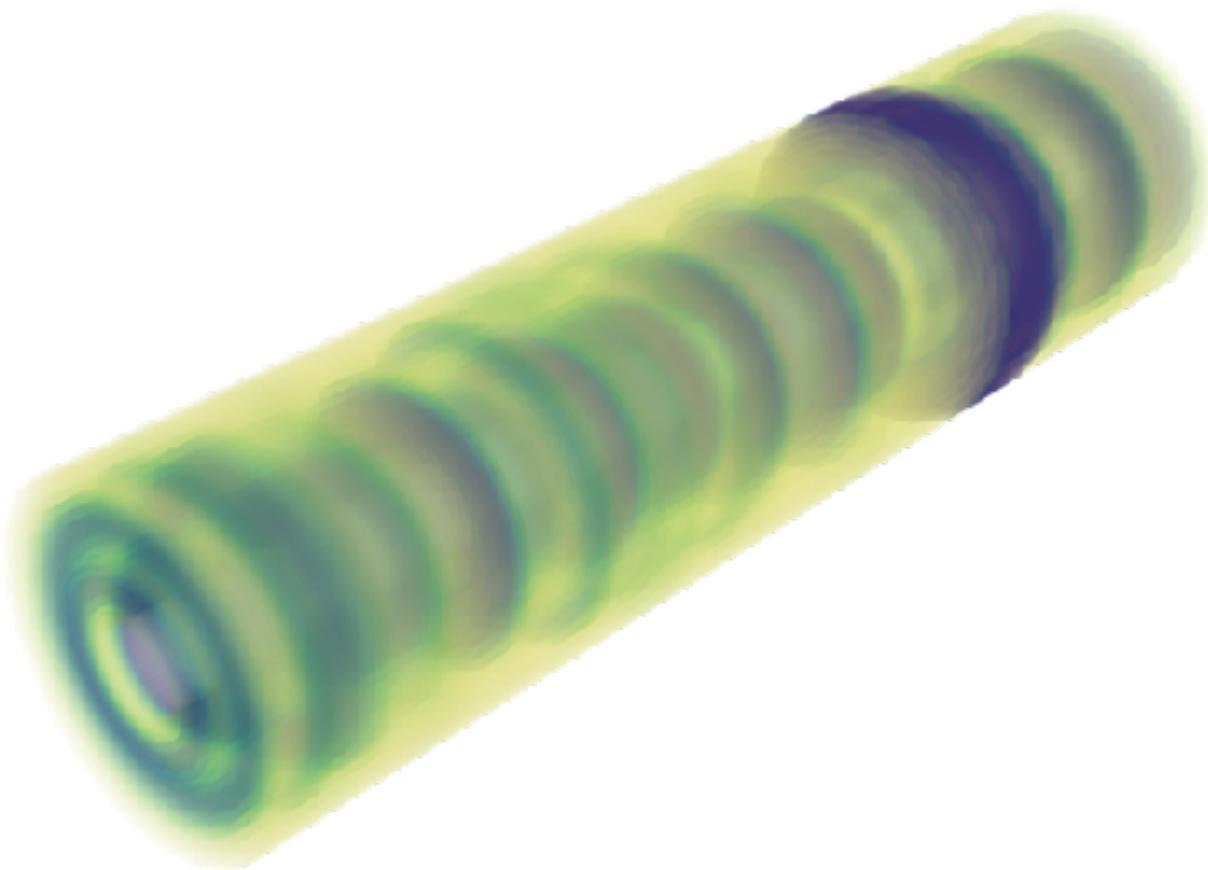
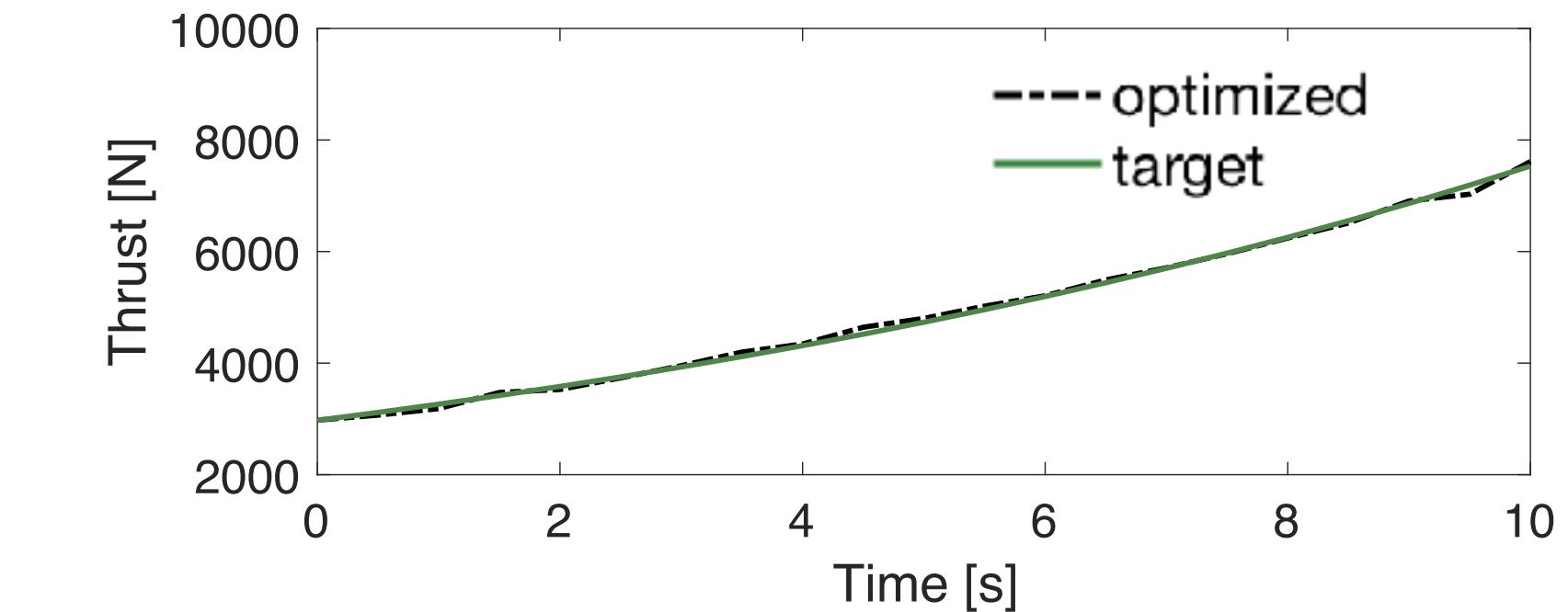
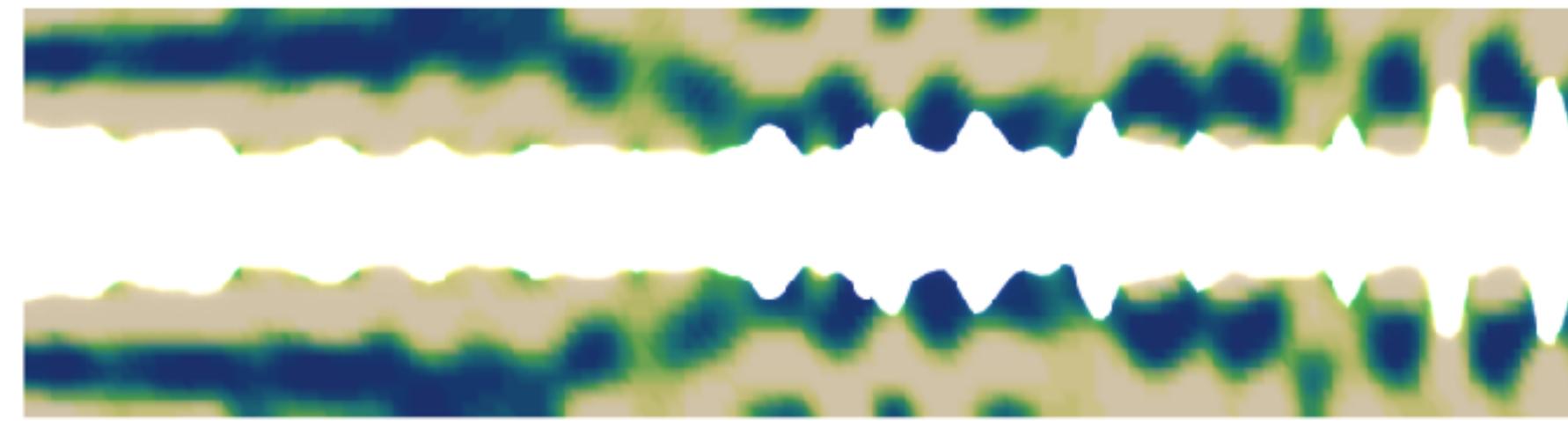
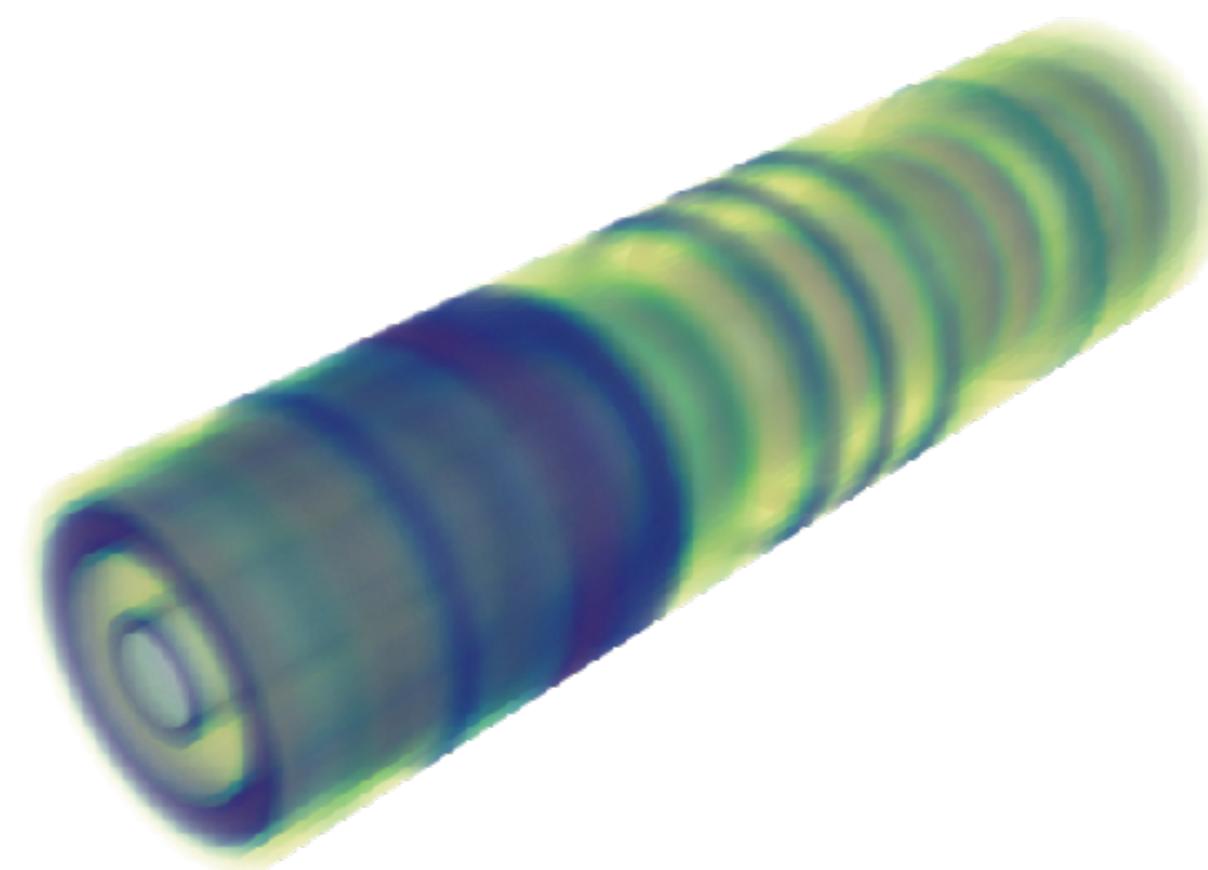
Slide by 15



Slide by 15



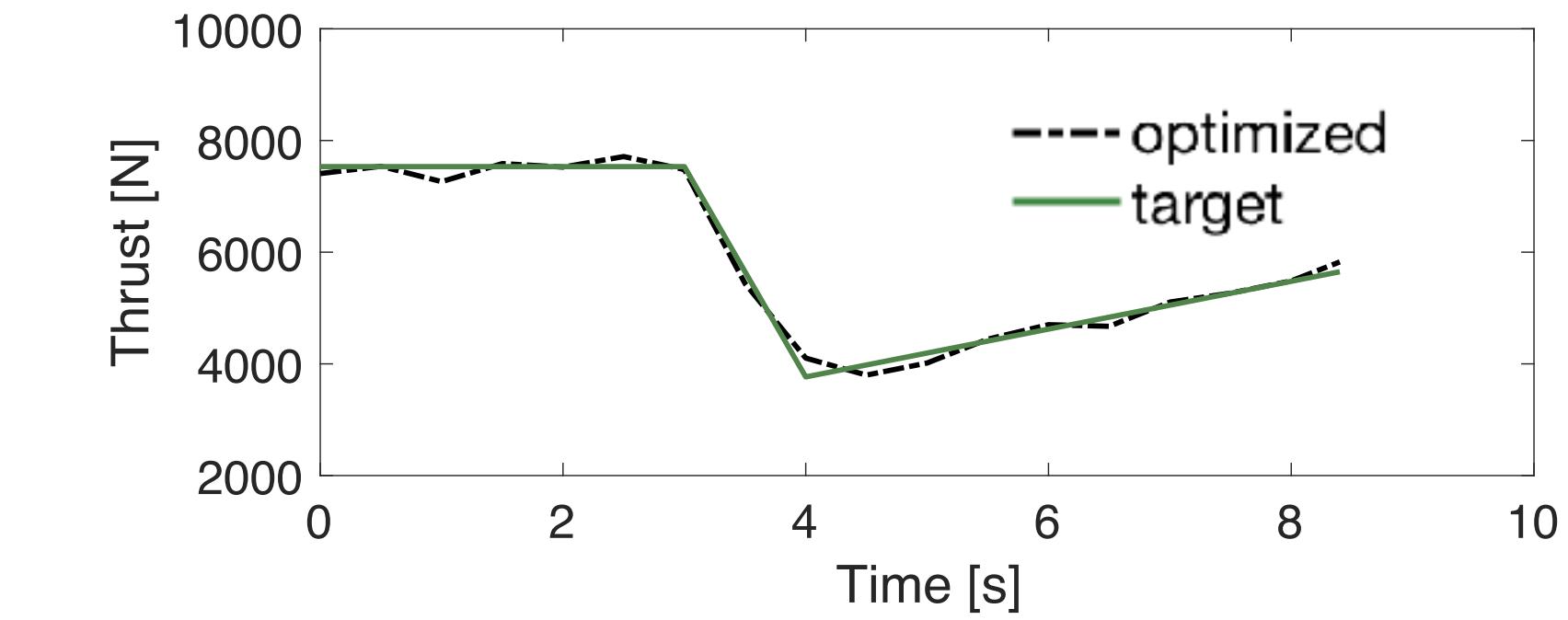
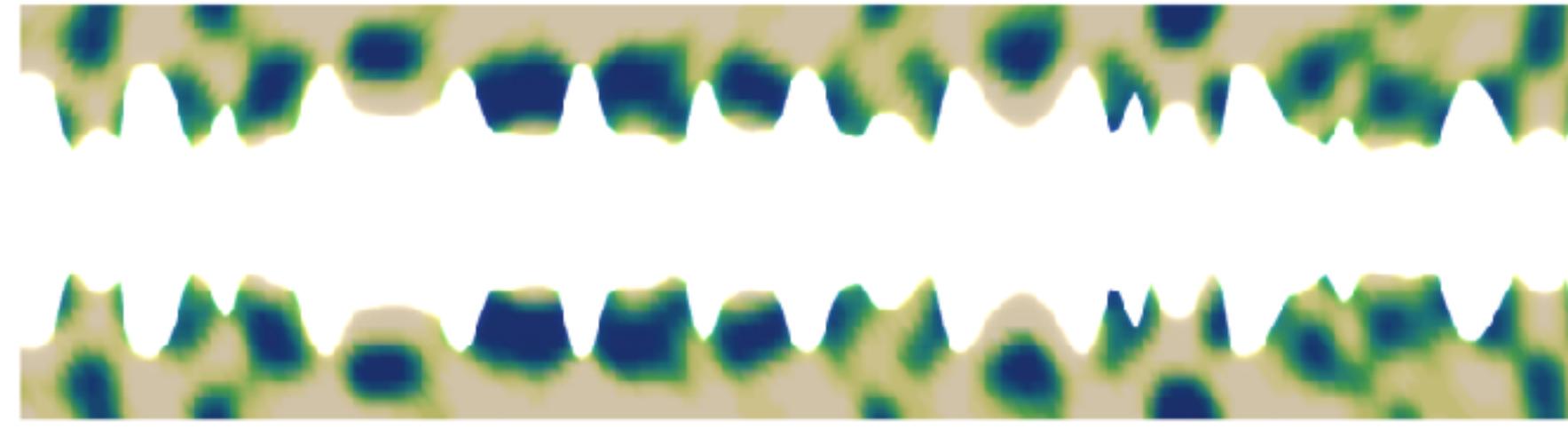
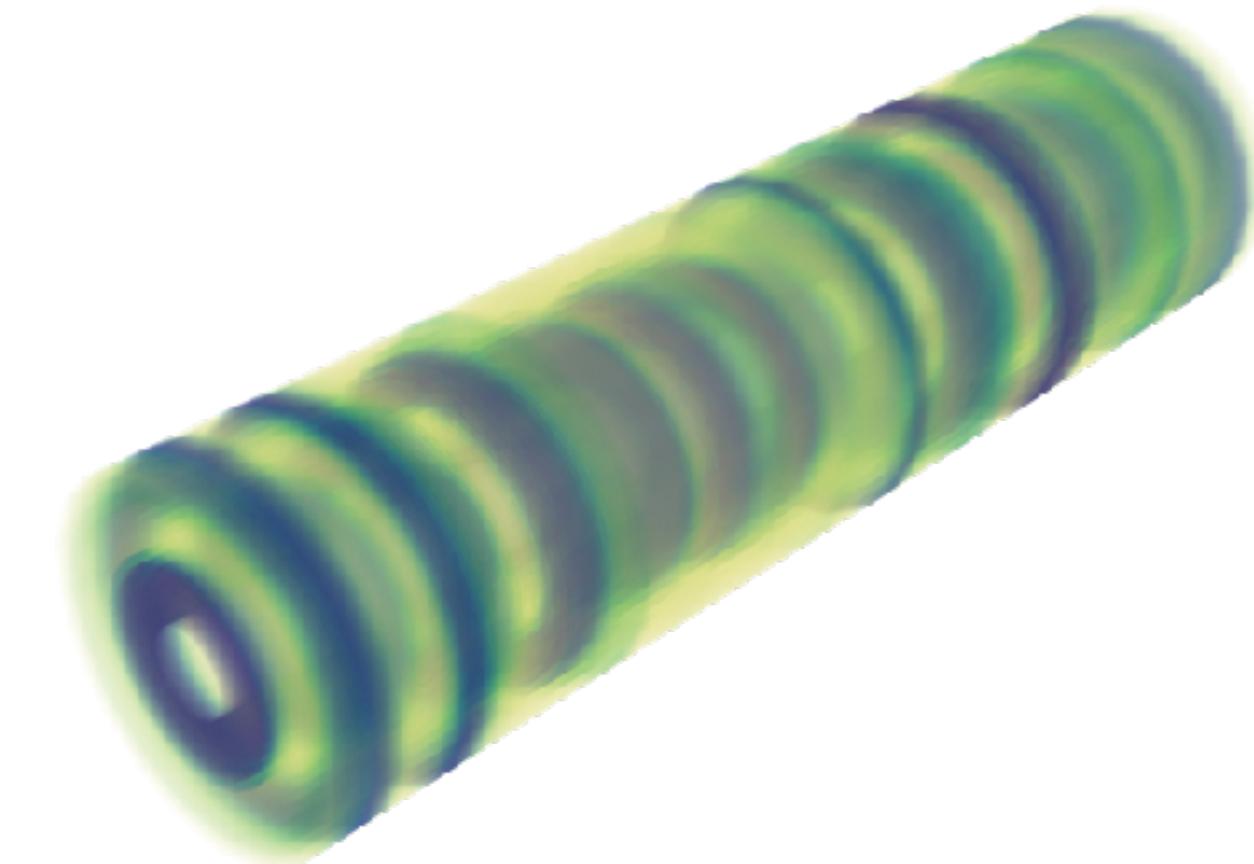
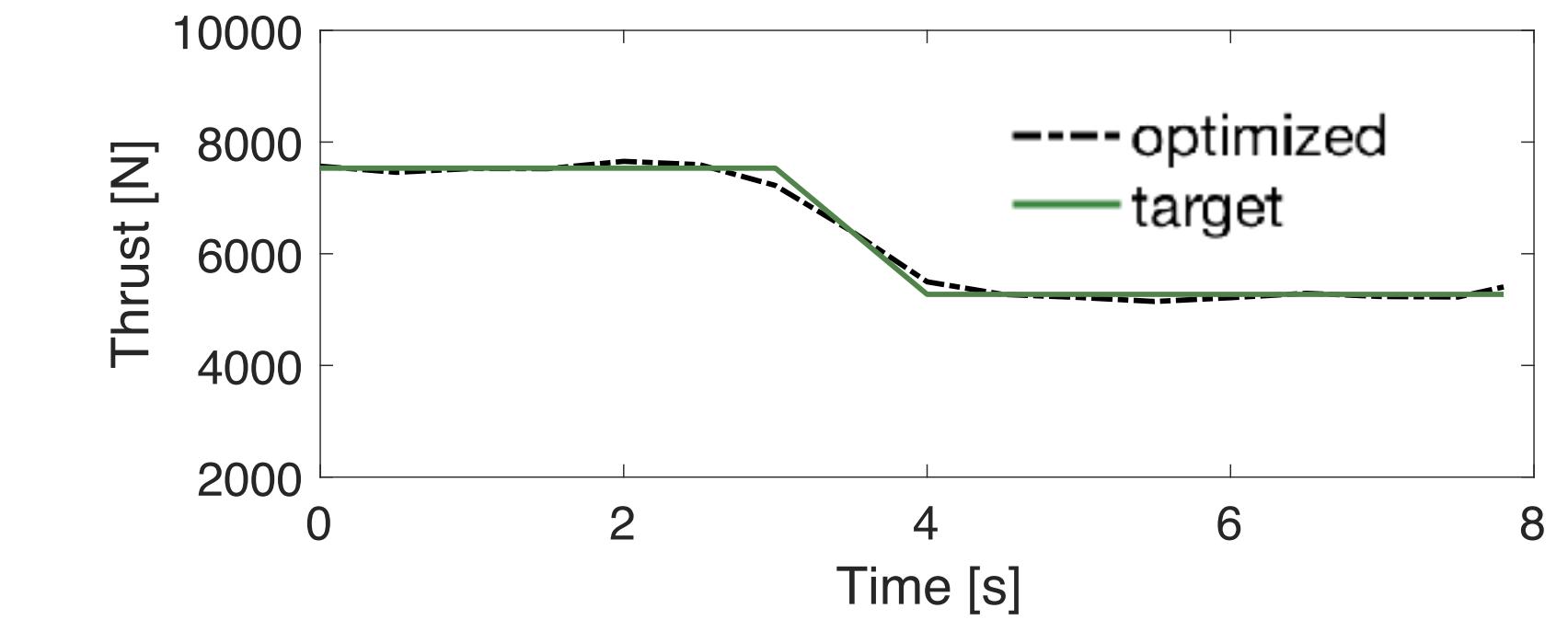
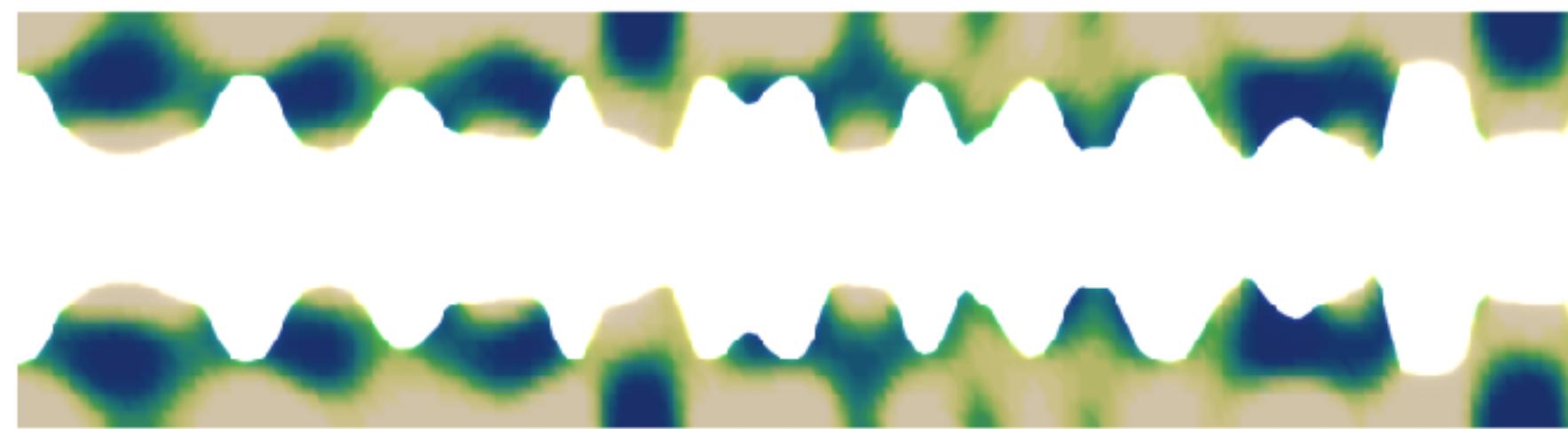
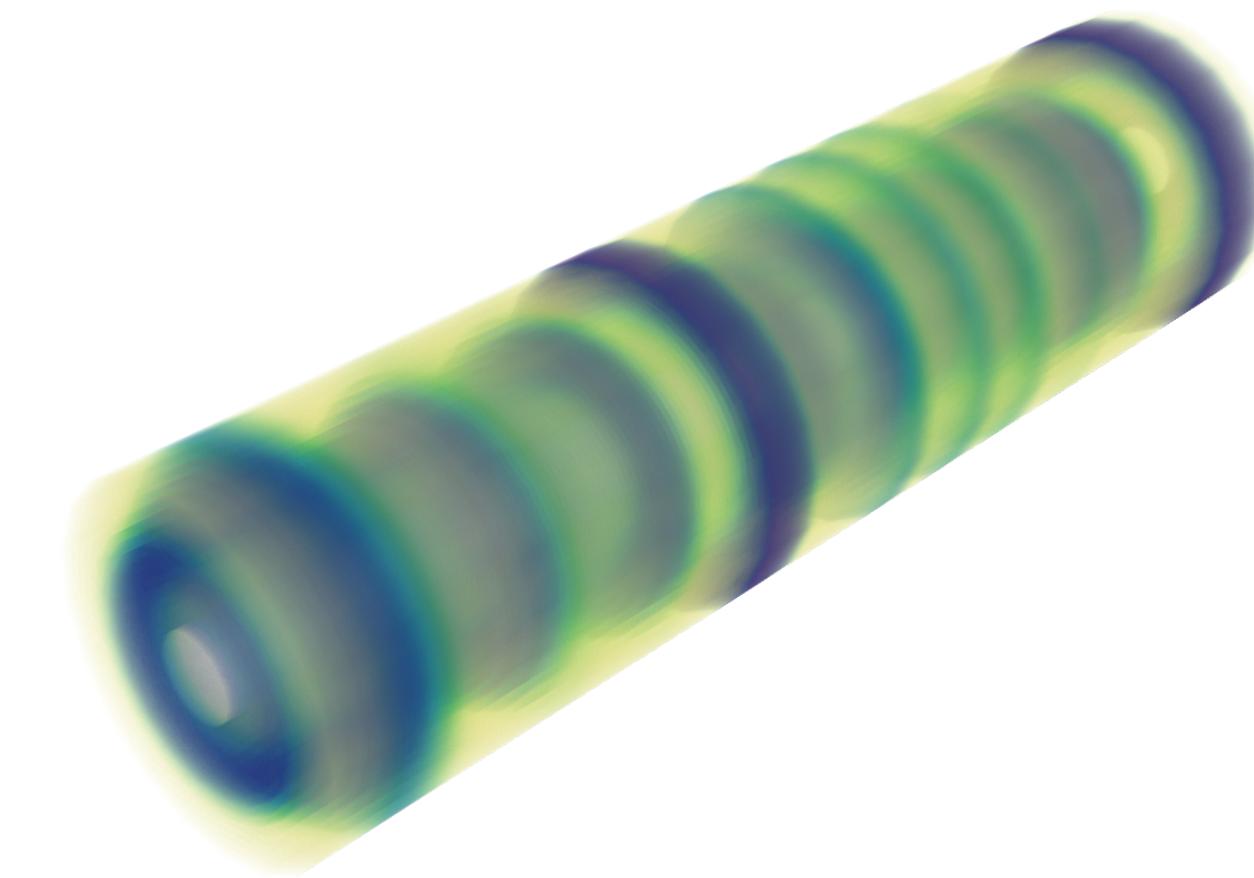
# Optimized Solid Rocket Fuel Designs



$0.254 \times 10^{-2}$  burn rate [m/sec]  $1.52 \times 10^{-2}$



# Optimized Solid Rocket Fuel Designs



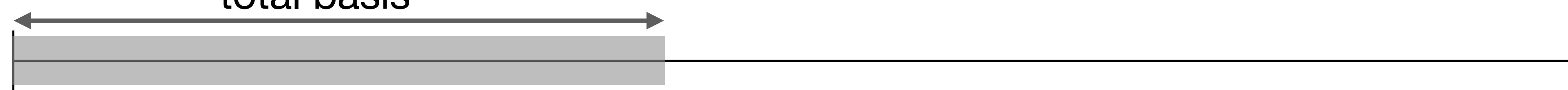
$0.254 \times 10^{-2}$  burn rate [m/sec]  $1.52 \times 10^{-2}$

Graded material fields!

# Performance



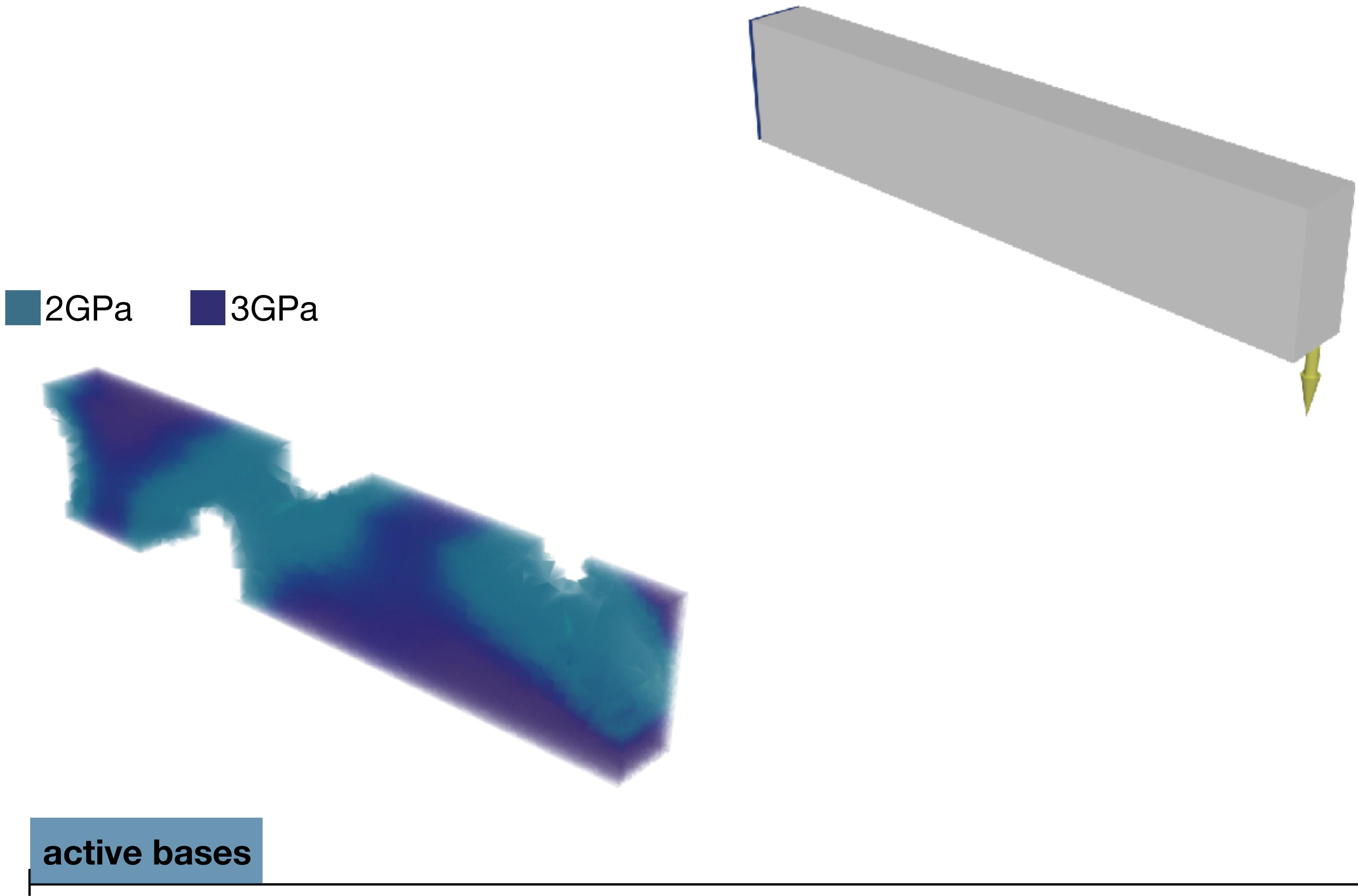
Fixed Basis (Already Reduced Order!!)



Thrust Profile	Fixed Basis		Sliding Basis	
	Time	Objective/Error	Time	Objective/Error
Constant Acceleration	1178s	349k/2.3%	288s	86k/1.1%
Constant Deceleration	4896s	867k/3.4%	621s	452k/2.7%
Two Step	191s	102k/1.1%	69s	217k/1.4%
Bucket	1006s	272k/1.8%	596s	272k/1.8%

Up to 8x speed up Comparable objective values

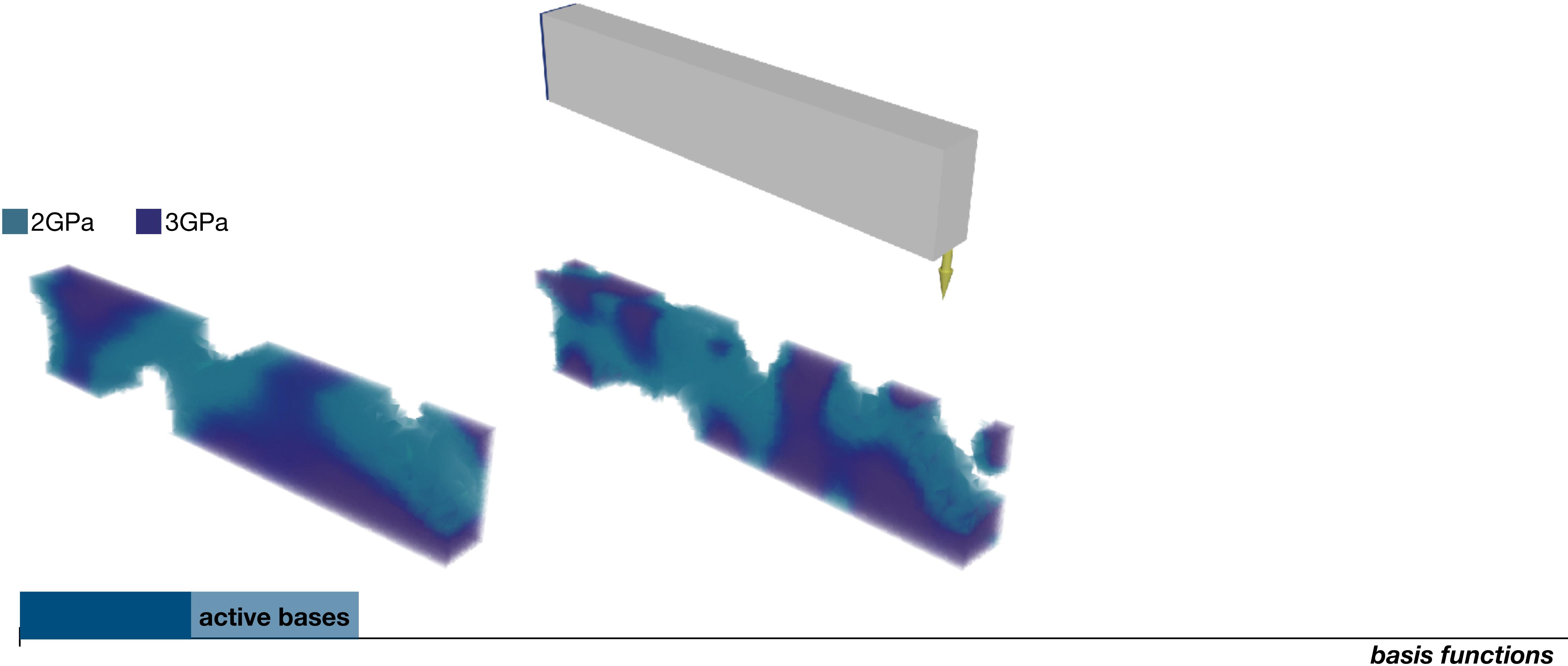
# Multi-material Topology Optimization Through Sliding Optimization Steps



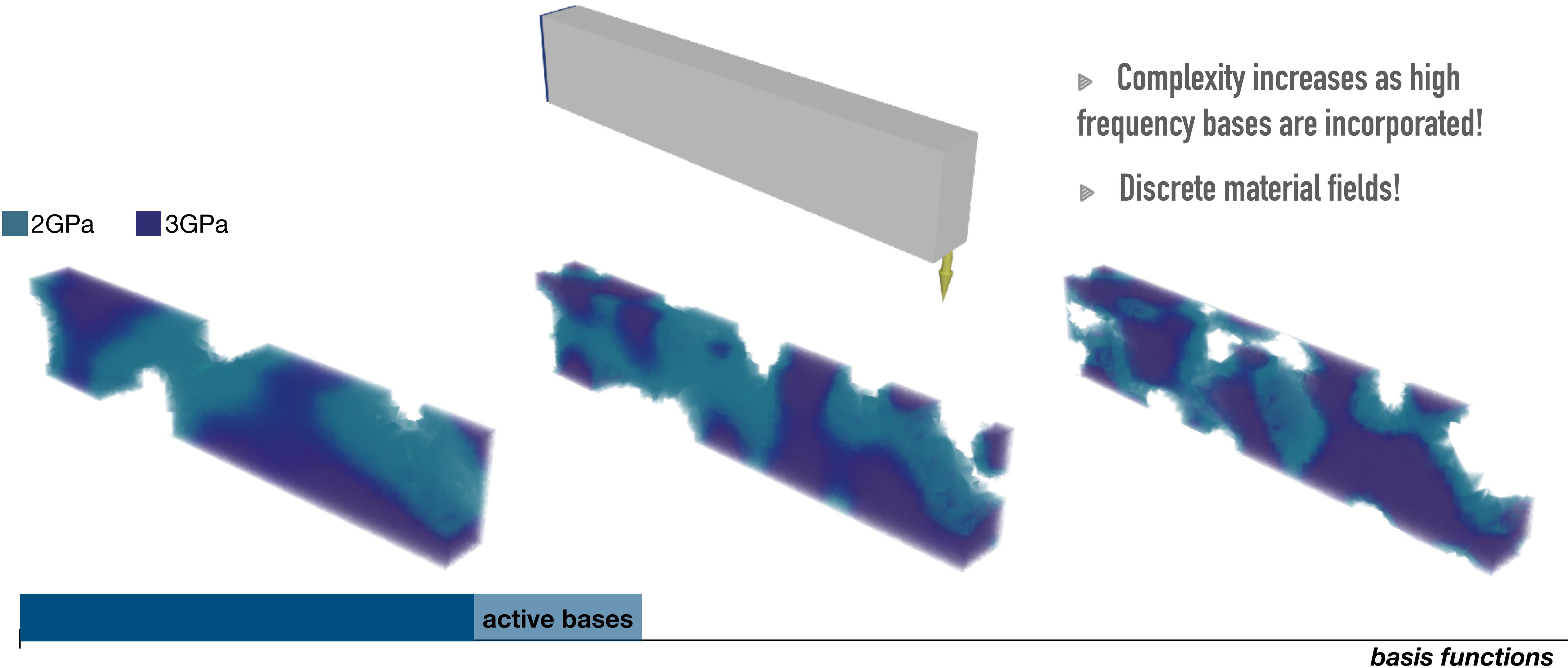
$$\begin{aligned} \min_{\boldsymbol{w}} \quad & \boldsymbol{u}^T \boldsymbol{K}(\boldsymbol{w}) \boldsymbol{u} \\ \text{s.t.} \quad & m(\boldsymbol{w})/m_0 \leq m_{frac} \\ & \boldsymbol{K}(\boldsymbol{w}) \boldsymbol{u} = \boldsymbol{F} \end{aligned}$$

*basis functions*

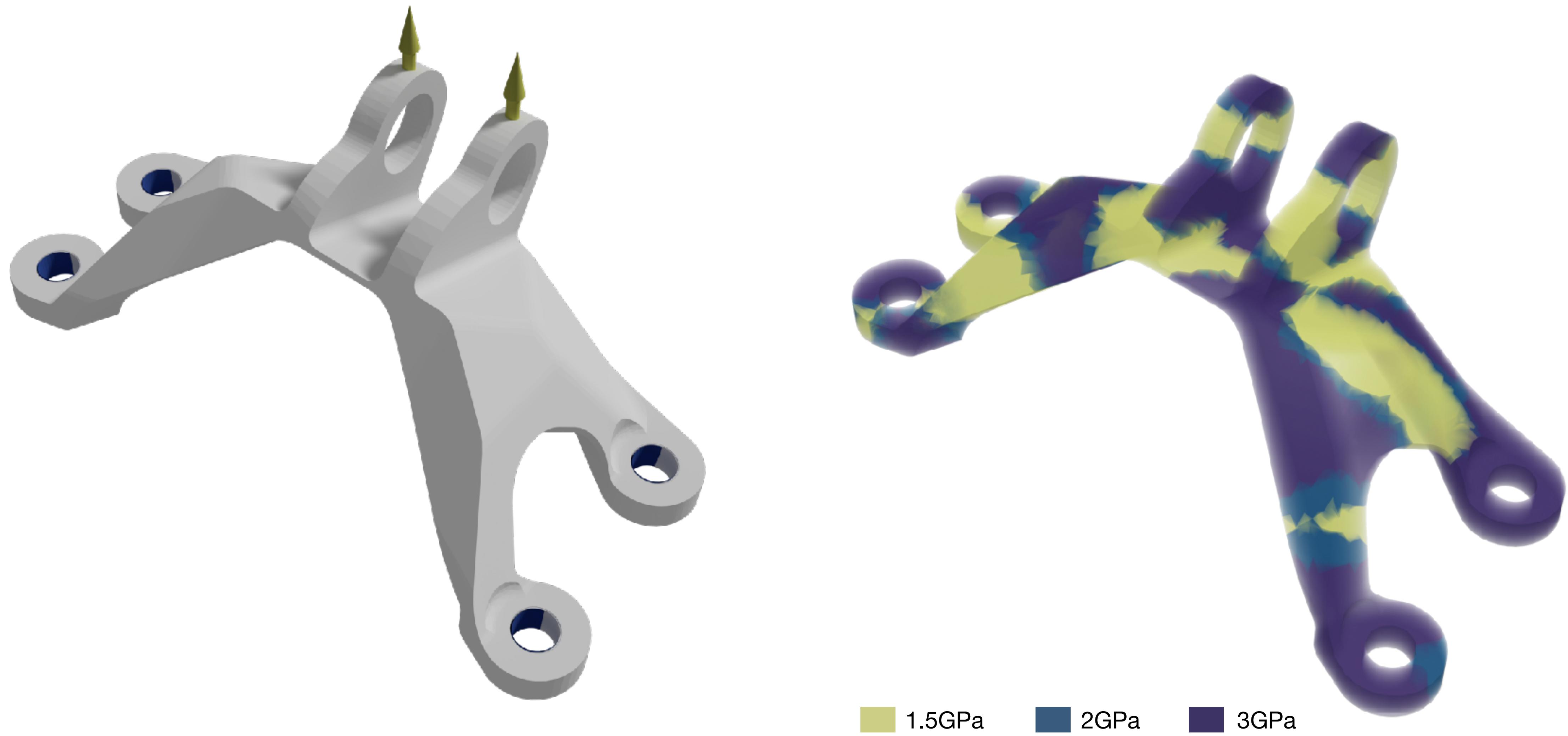
# Multi-material Topology Optimization Through Sliding Optimization Steps



# Multi-material Topology Optimization Through Sliding Optimization Steps

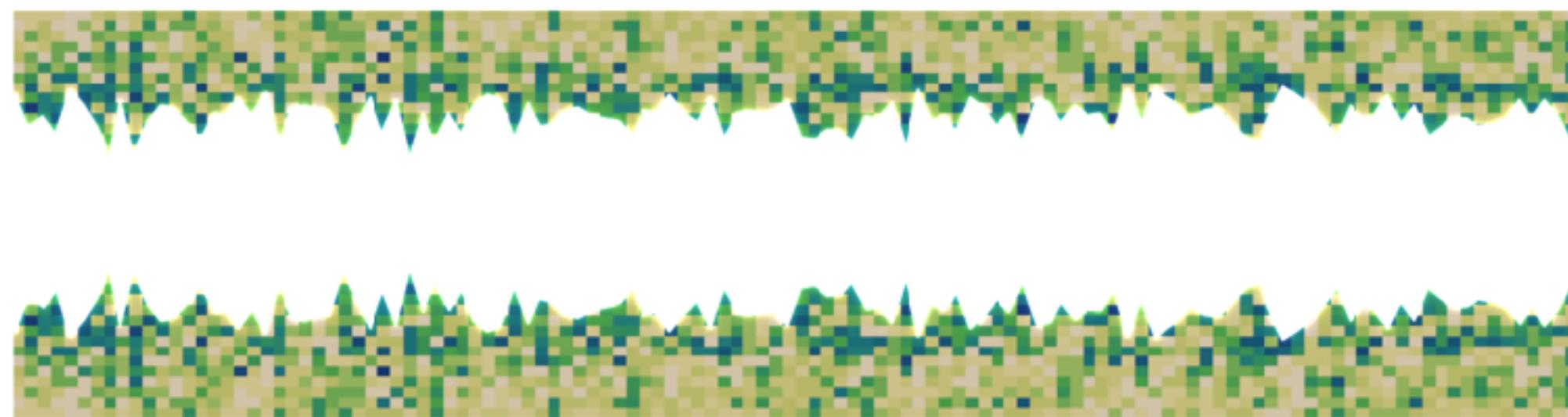


# Material Distribution Optimization



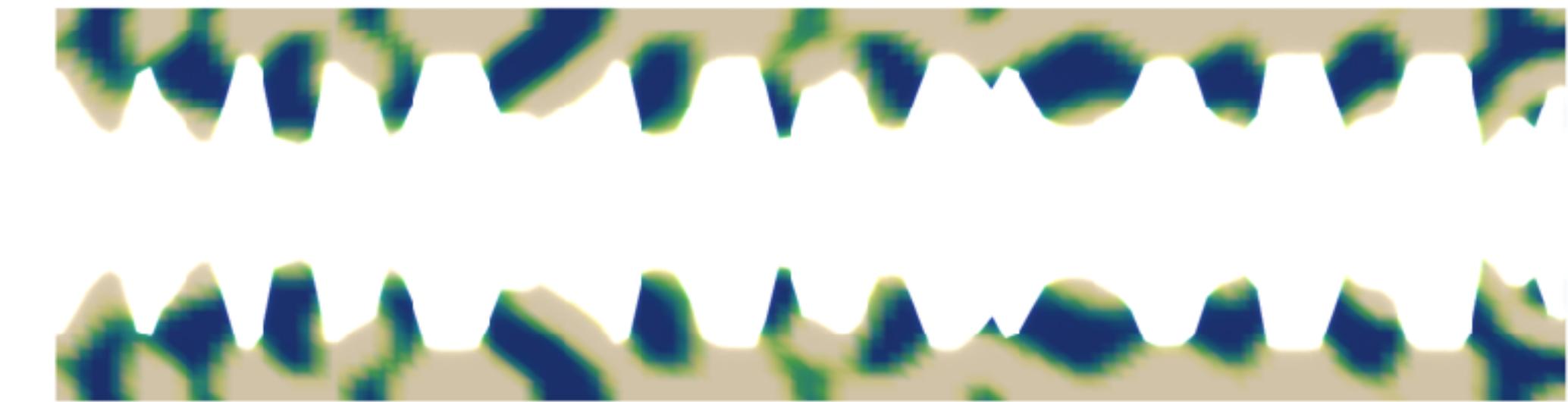
# Conventional VS Reduced Order

Optimize for each pixel

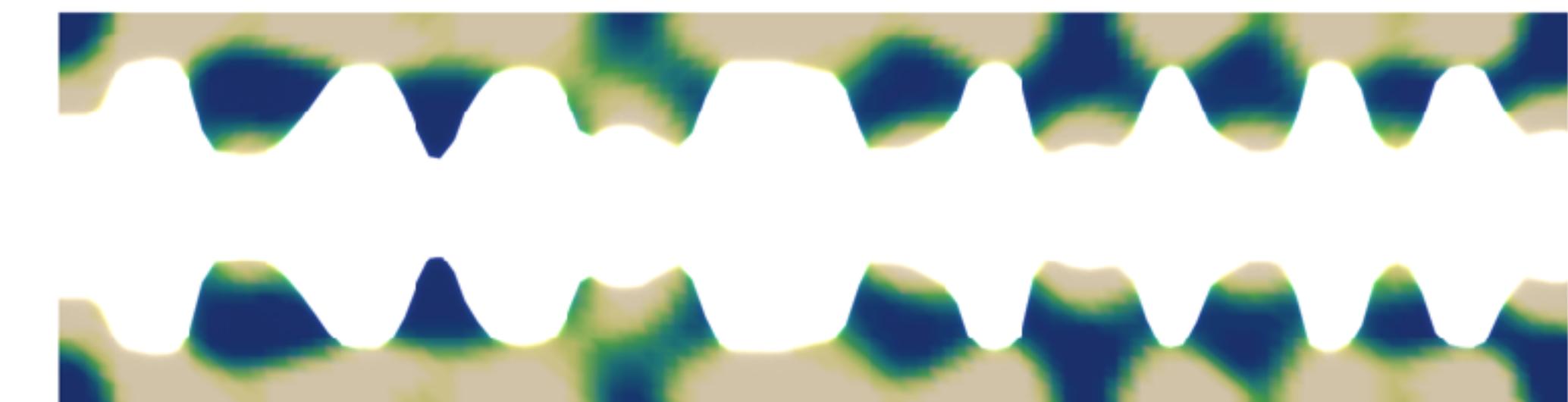


Objective: 252316 Time: 1945sec

Optimize for weights of the Laplacian basis  
(Ours)



Sliding Basis — Objective: 253040 Time: 42sec

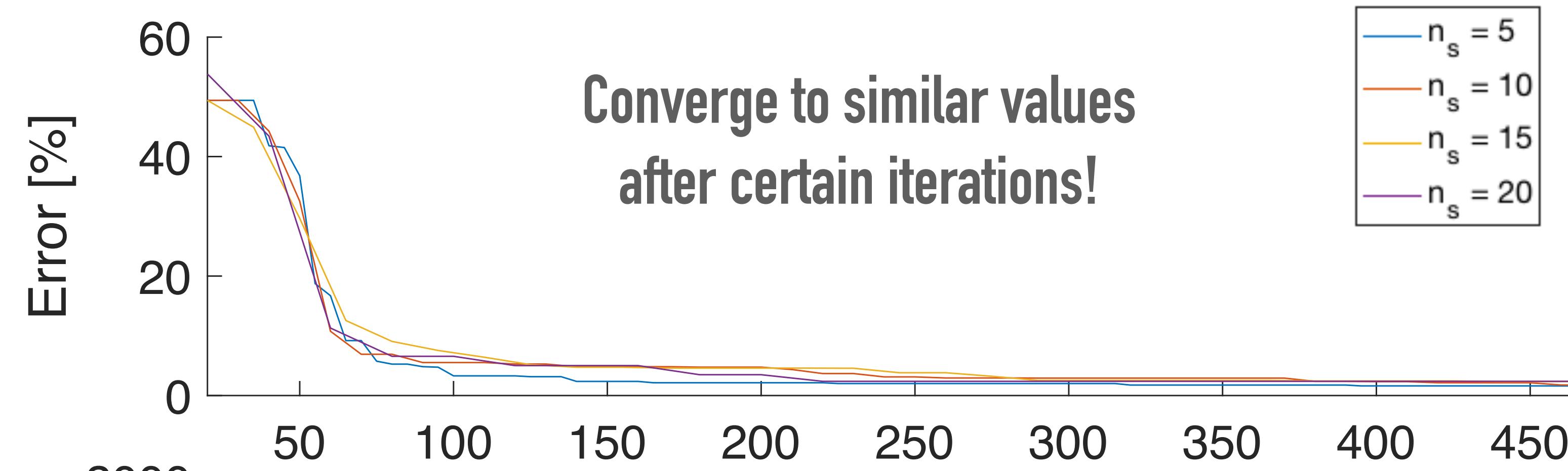
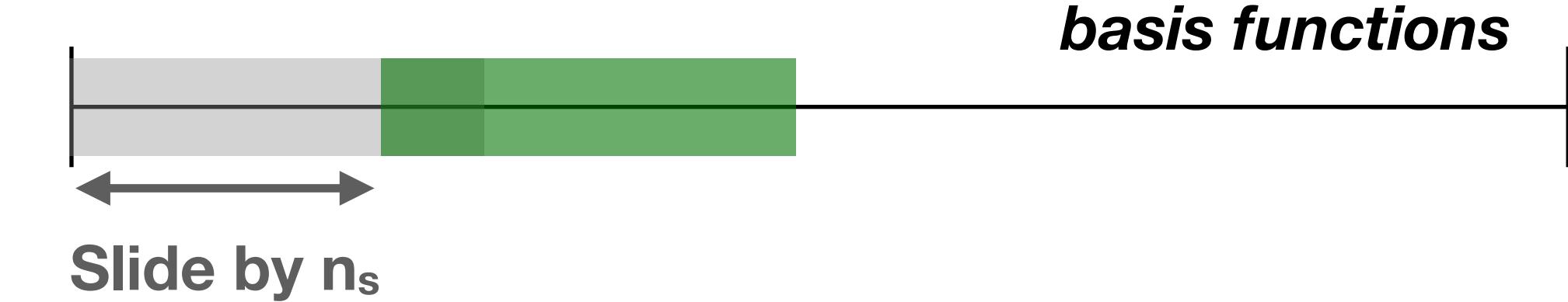


Fixed Basis — Objective: 287876 Time: 173sec

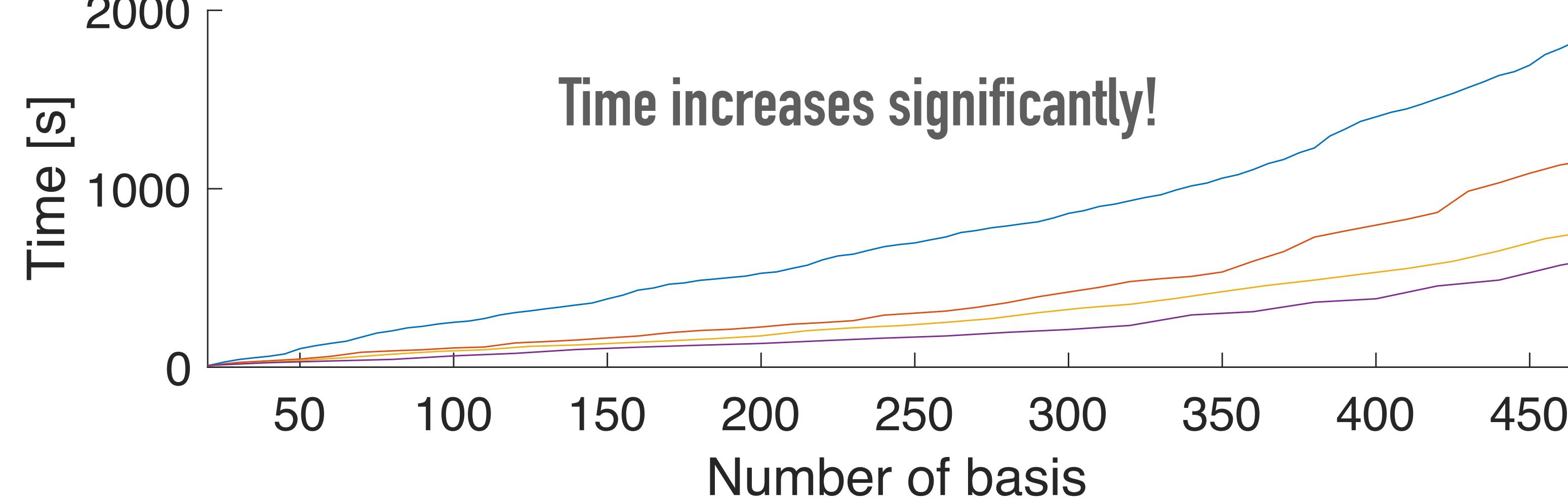
Comparable Objective Values

Reduced Order Faster

# Effect of sliding amount, $n_s$



Converge to similar values  
after certain iterations!



Time increases significantly!

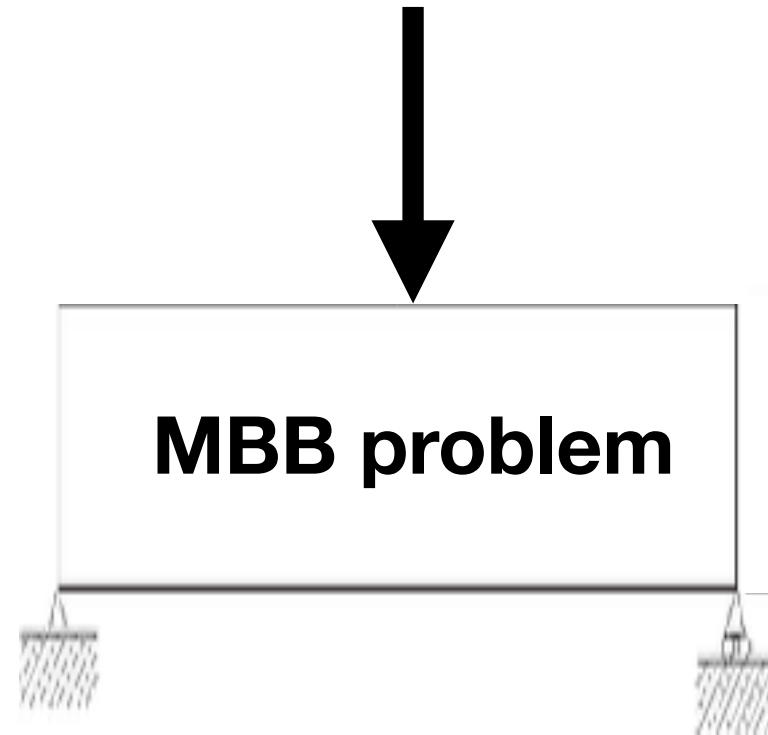
# A Versatile Design Optimization Tool

The main **contributions** of the presented work:

- An optimization technique we call **sliding basis optimization** to **efficiently explore** parameterized design space
- **Practical material design method** with prescribed bounds using Laplacian basis
- **Enabling** optimization of material distributions for new applications coupled with **black-box analysis**

# Recent Developments

Sliding basis topology optimization - a modular system



Sliding Basis Topology Optimization  
Using **50 Bases**  
**Compliance objective: 302.2**



**Conventional** Topology Optimization (**TOP88**)  
Using **10k elements**  
**Compliance objective: 287.8**

**Goal is not to match the geometry but achieve comparable performance faster!!**

# Thank you!

**nulu@parc.com**

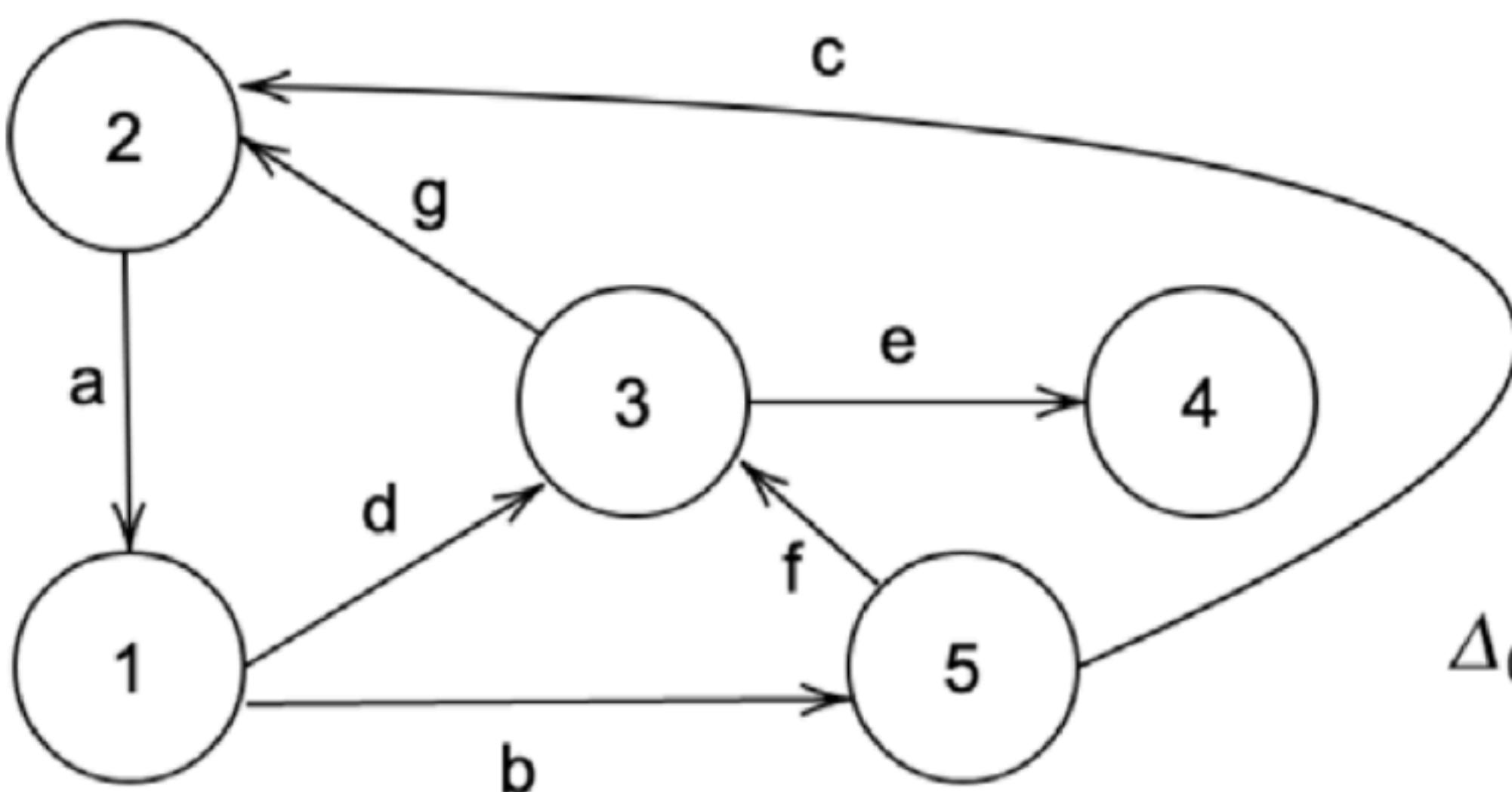
Sliding Basis Optimization for Heterogeneous Material Design

# **Supplementary**

# Acknowledgements

The authors would like to thank NASA Jacobs Space Exploration Group for providing the solid rocket fuel design problem with the target thrust profile. This research was developed with funding from the Defense Advanced Research Projects Agency (DARPA). The views, opinions and/or findings expressed are those of the authors and should not be interpreted as representing the official views or policies of the Department of Defense or U.S. Government. 3D models: dragon by XYZ RGB Inc and GE bracket by WilsonWong on GrabCAD.

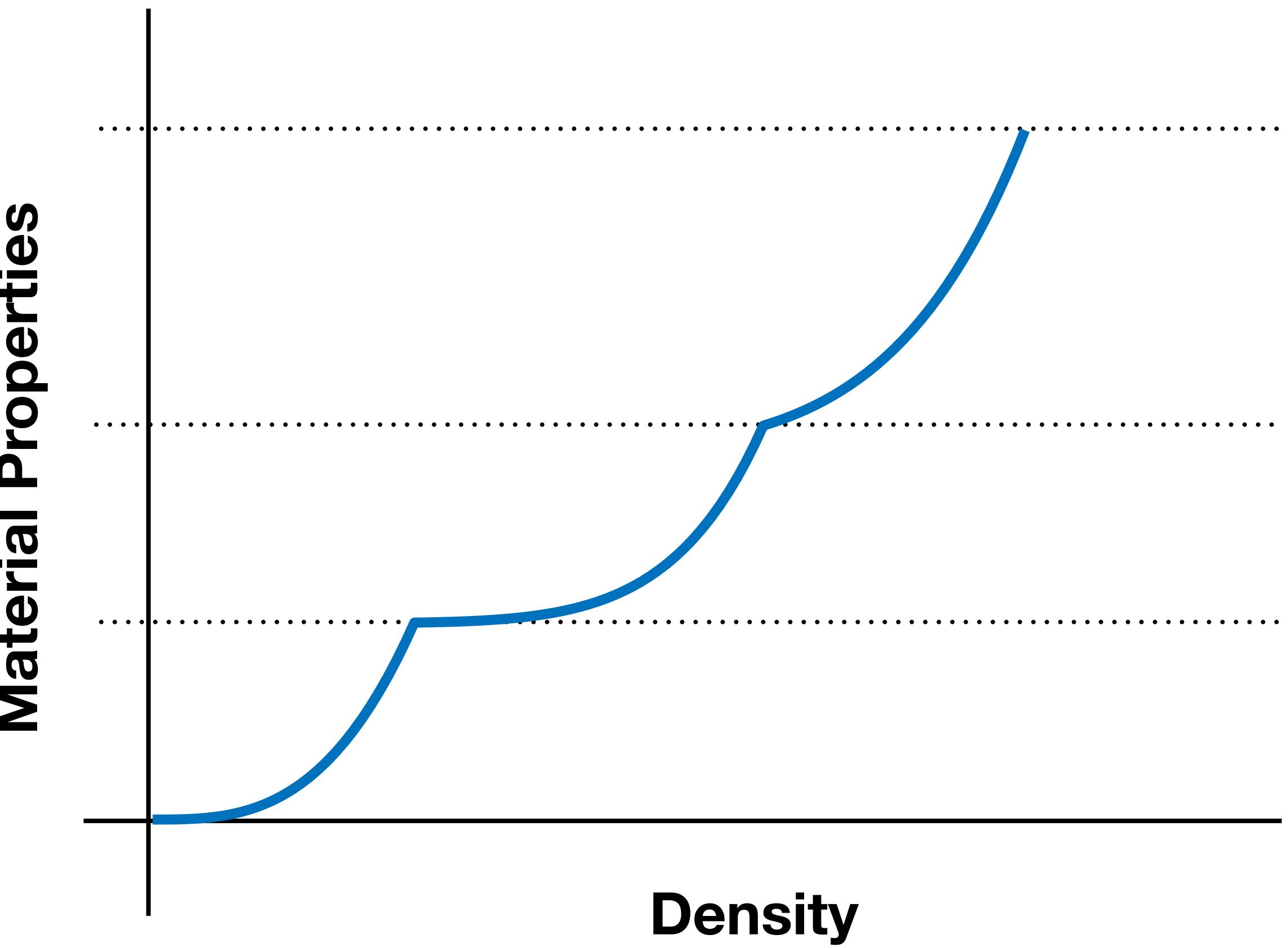
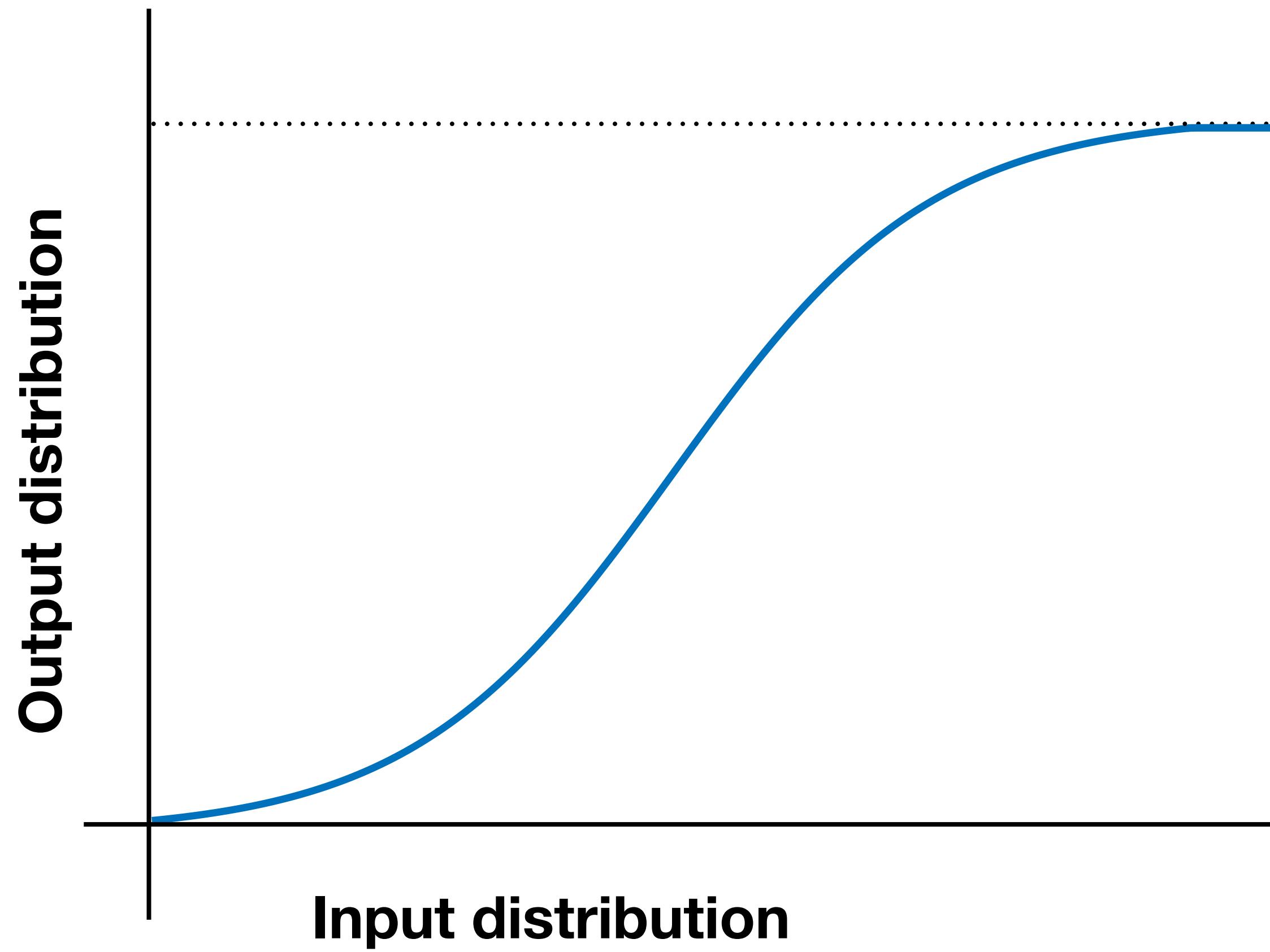
# Graph Laplacian

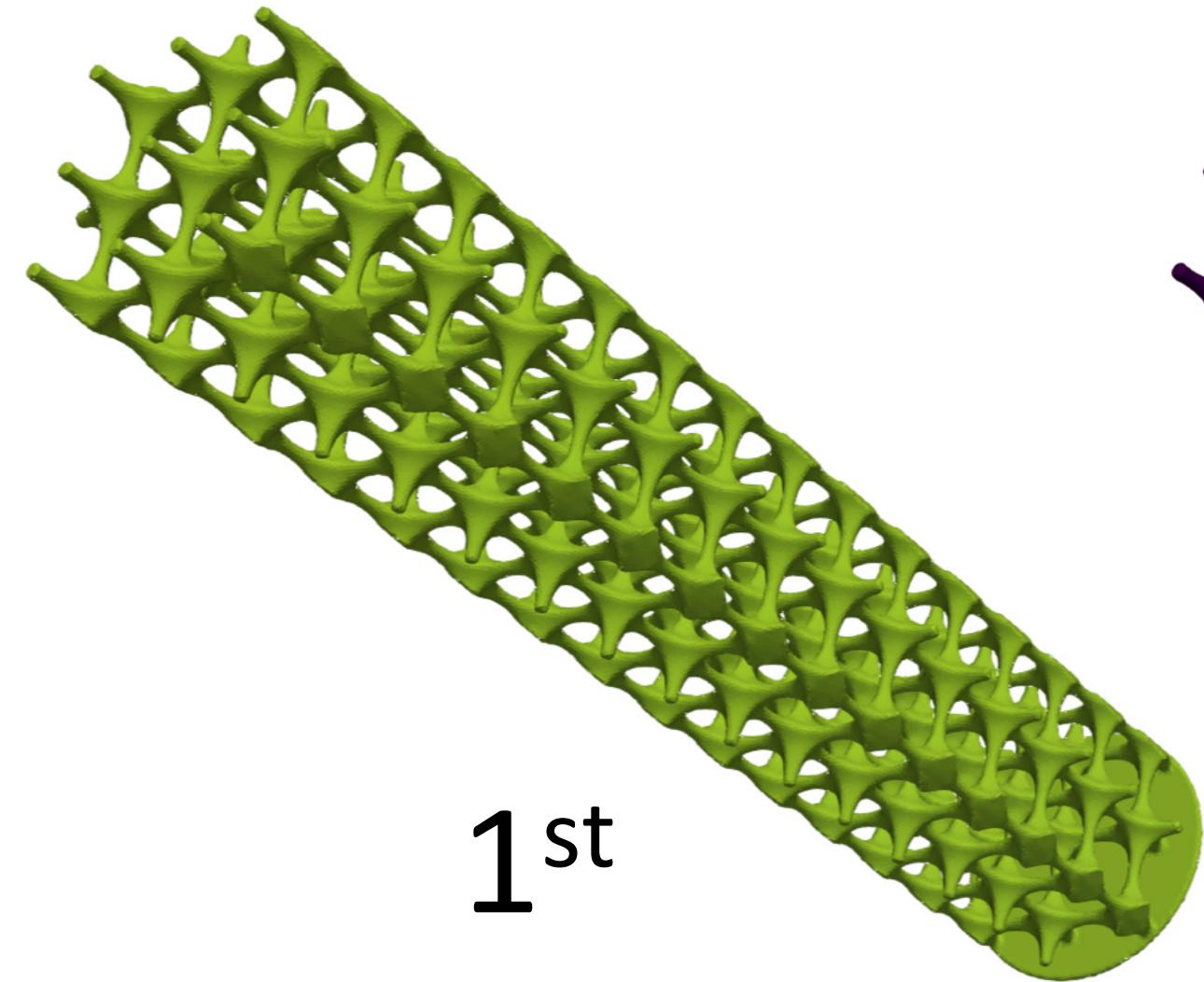


$$\partial_1 = \begin{pmatrix} -1 & 1 & 0 & 1 & 0 & 0 & 0 \\ 1 & 0 & -1 & 0 & 0 & 0 & -1 \\ 0 & 0 & 0 & -1 & 1 & -1 & 1 \\ 0 & 0 & 0 & 0 & -1 & 0 & 0 \\ 0 & -1 & 1 & 0 & 0 & 1 & 0 \end{pmatrix}$$

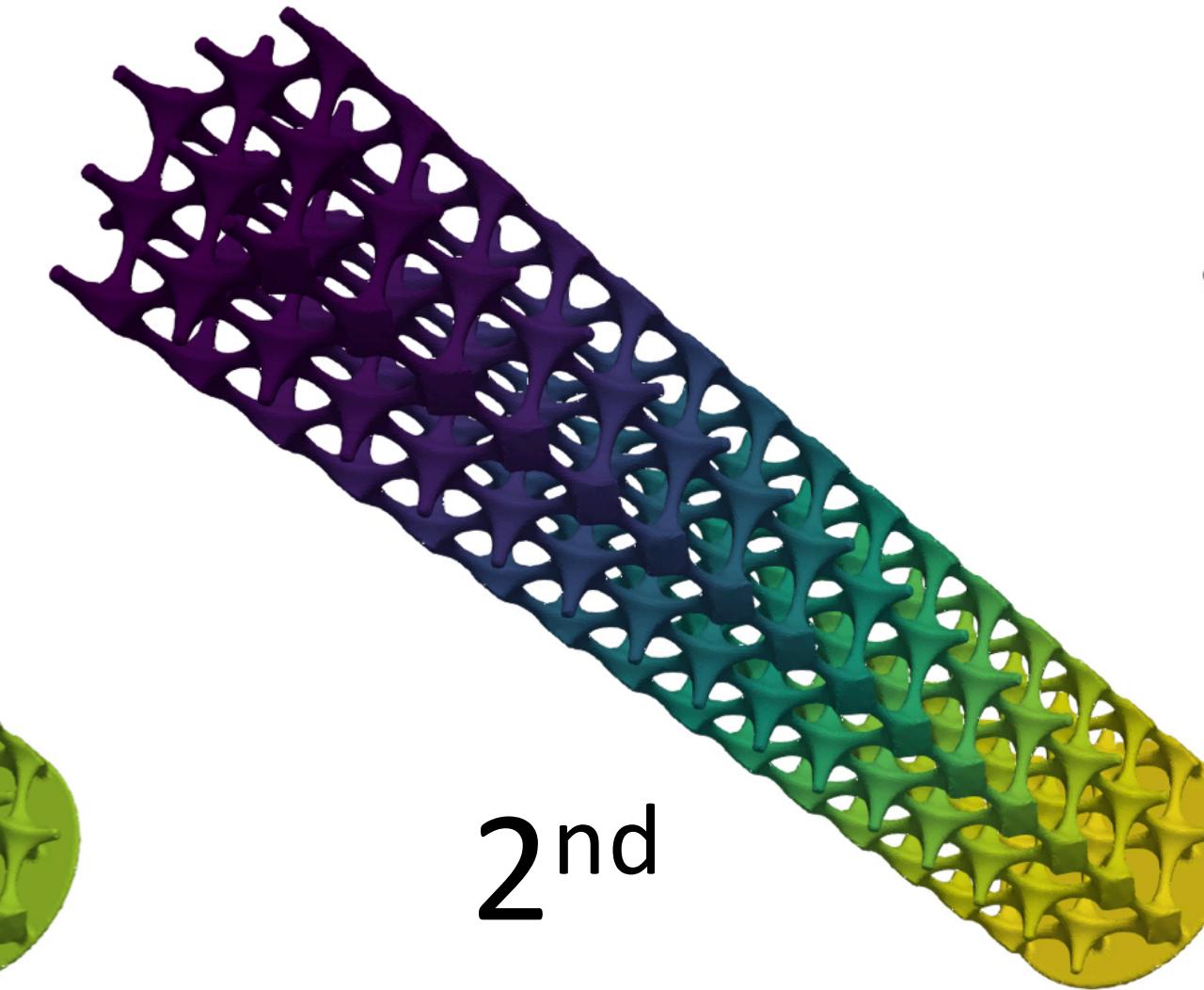
$$\Delta_0 = \partial_1 \partial_1^* = \begin{pmatrix} 3 & -1 & -1 & 0 & -1 \\ -1 & 3 & -1 & 0 & -1 \\ -1 & -1 & 4 & -1 & -1 \\ 0 & 0 & -1 & 1 & 0 \\ -1 & -1 & -1 & 0 & 3 \end{pmatrix}$$

# Filters

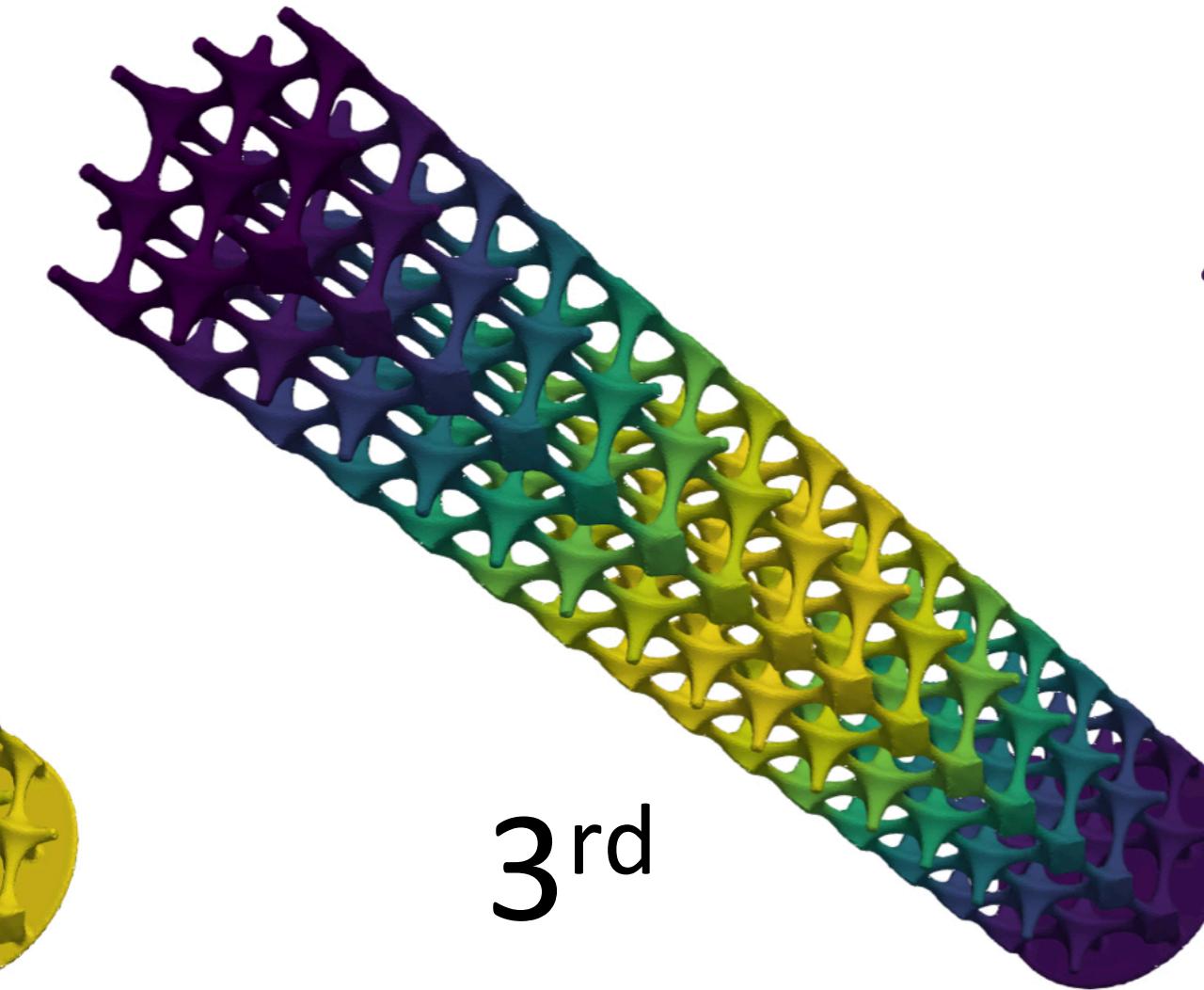




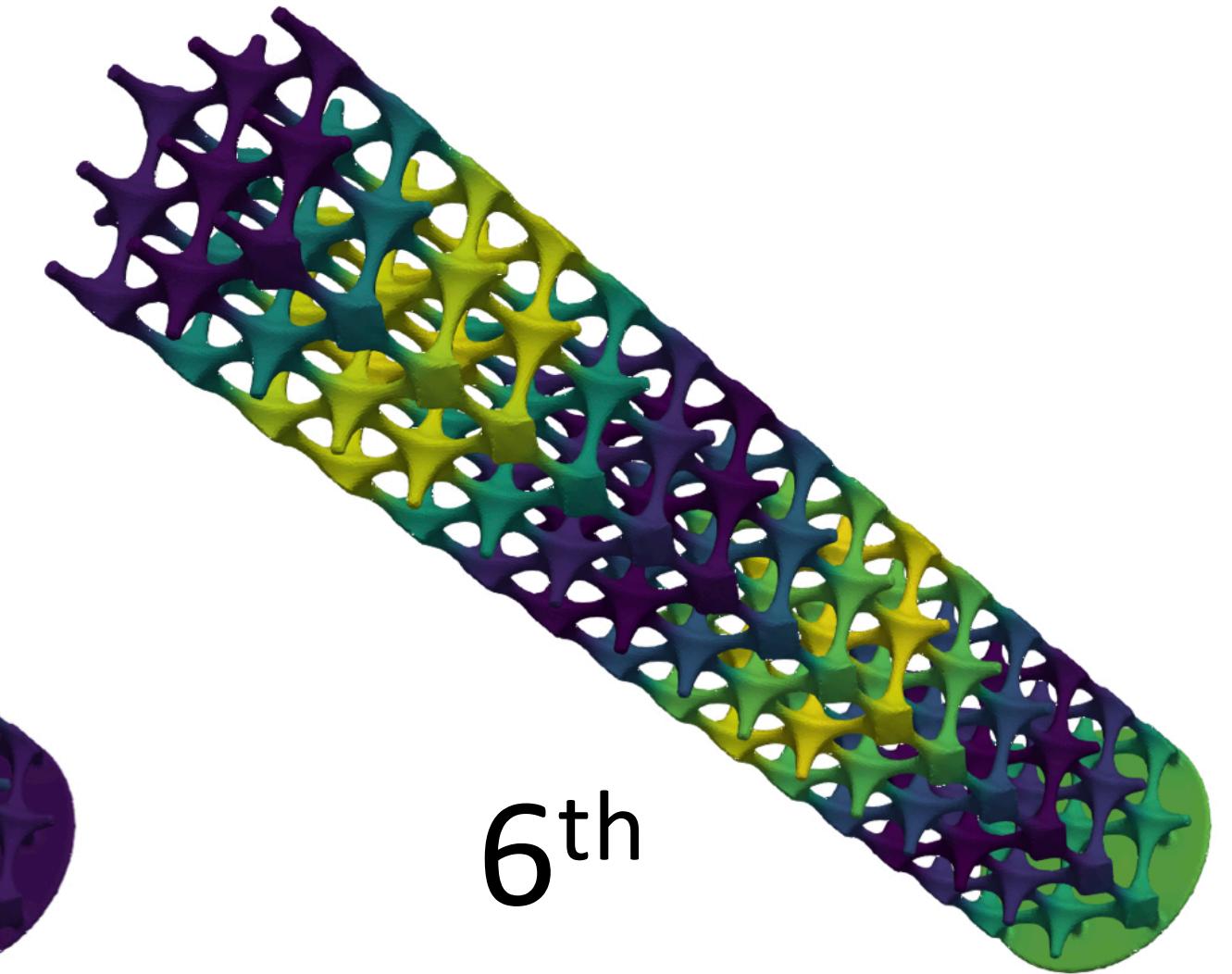
1<sup>st</sup>



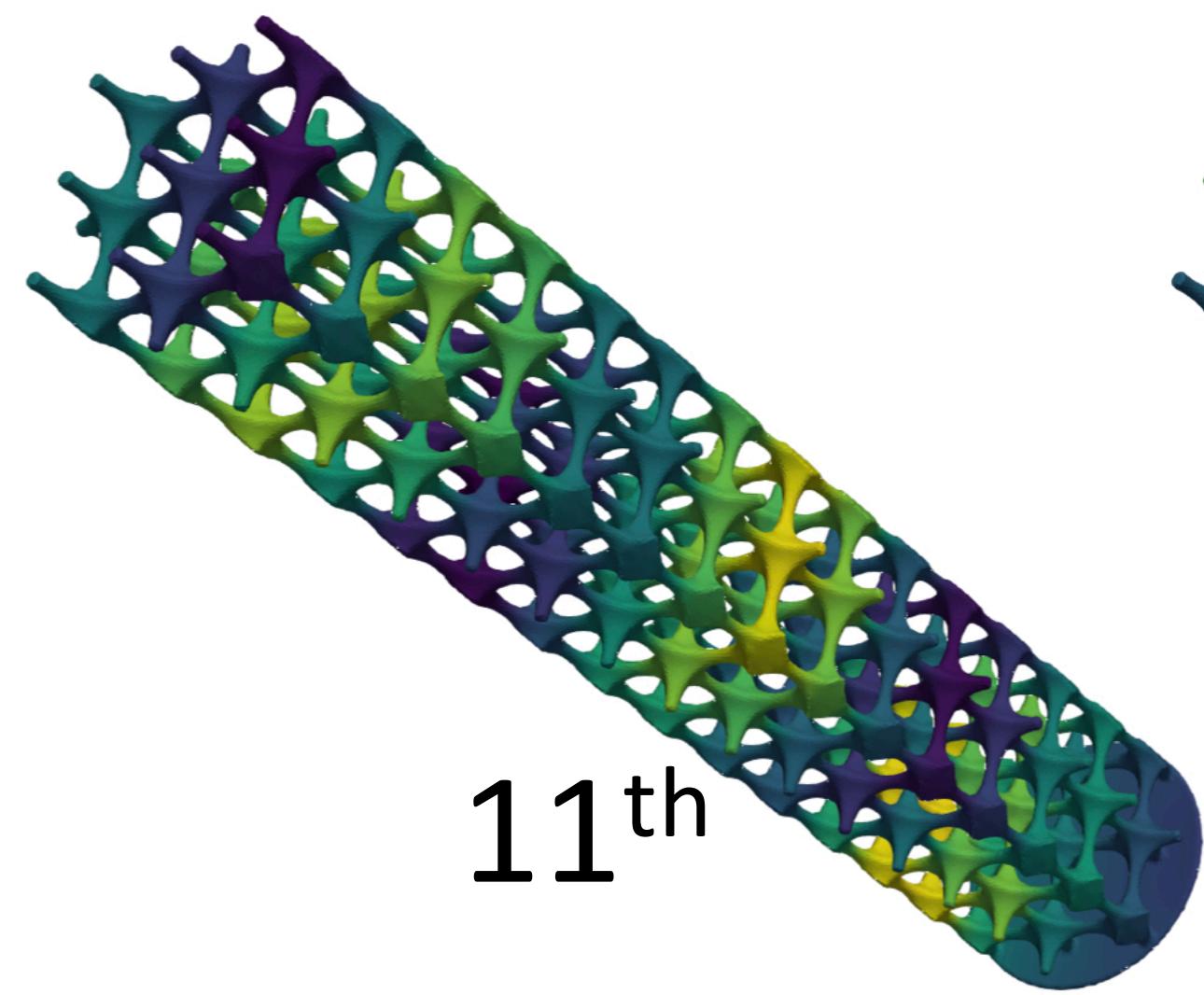
2<sup>nd</sup>



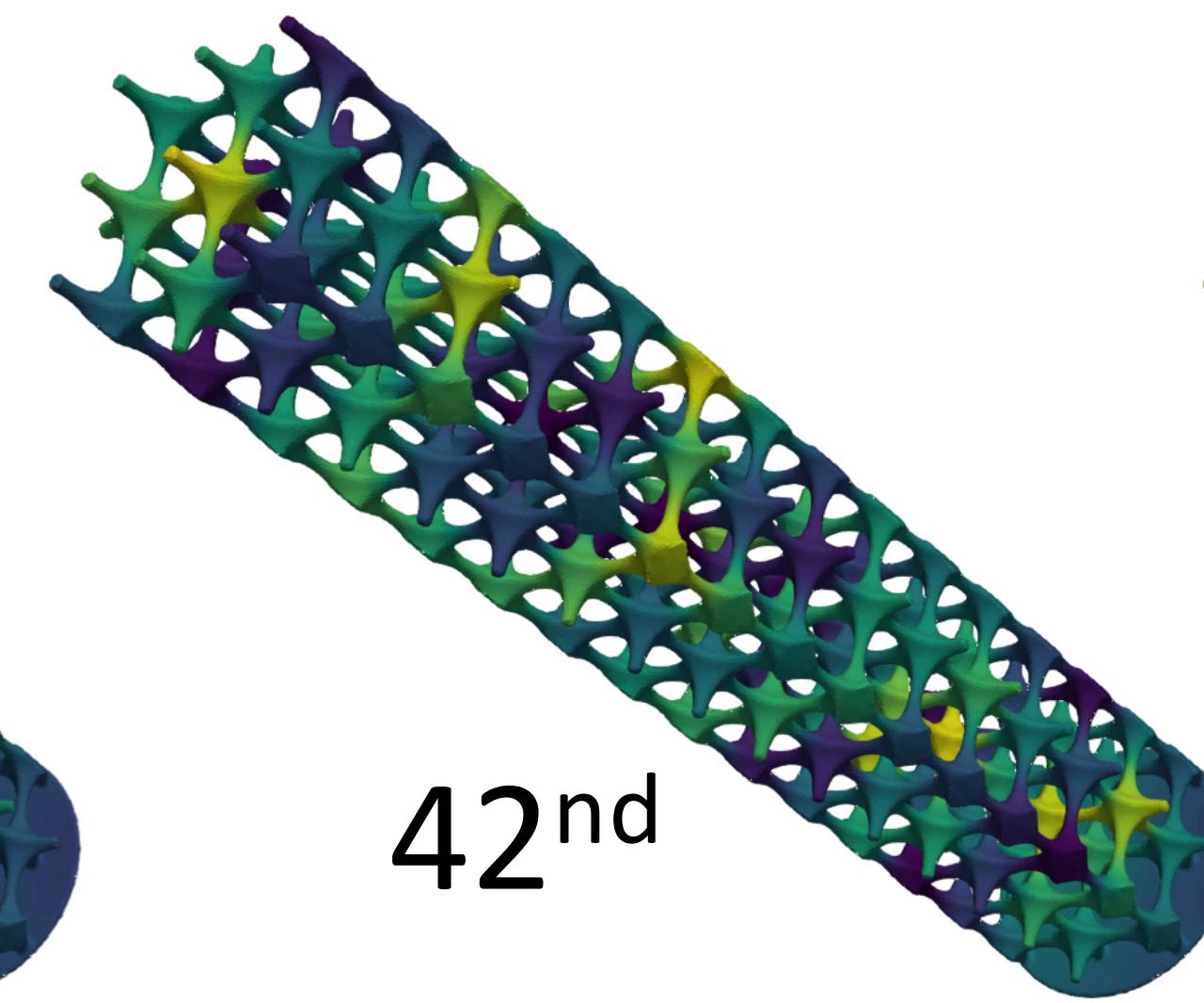
3<sup>rd</sup>



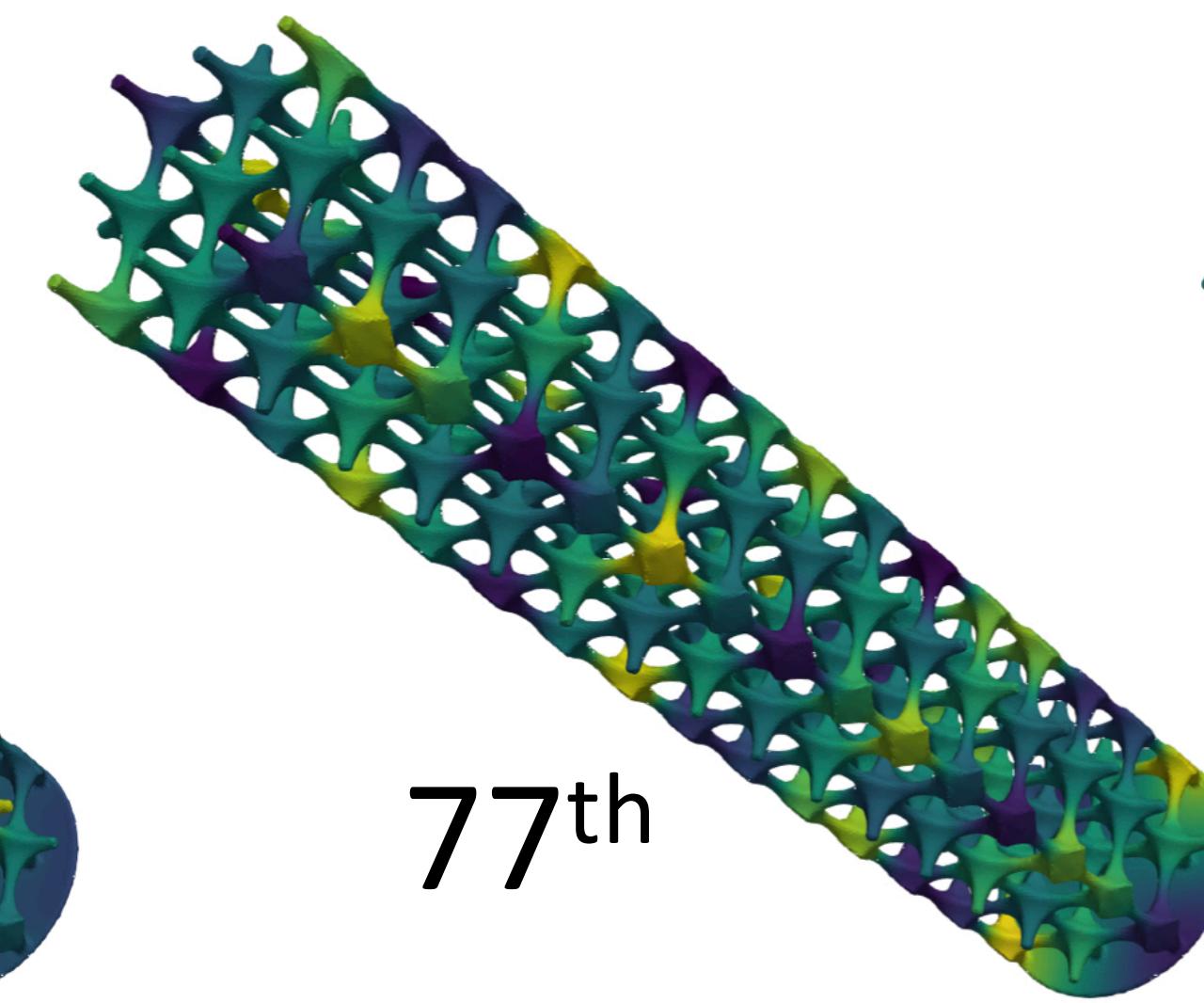
6<sup>th</sup>



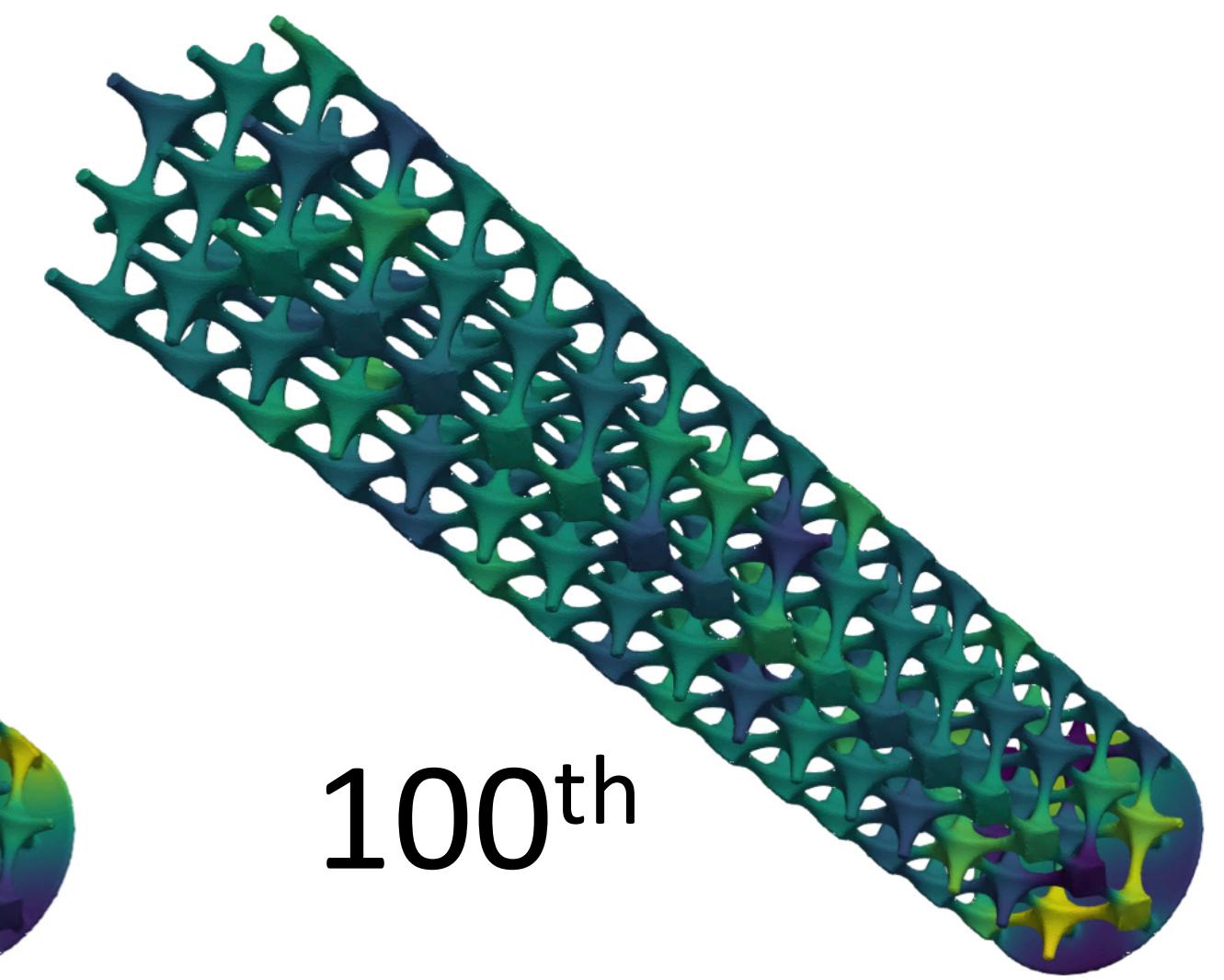
11<sup>th</sup>



42<sup>nd</sup>



77<sup>th</sup>

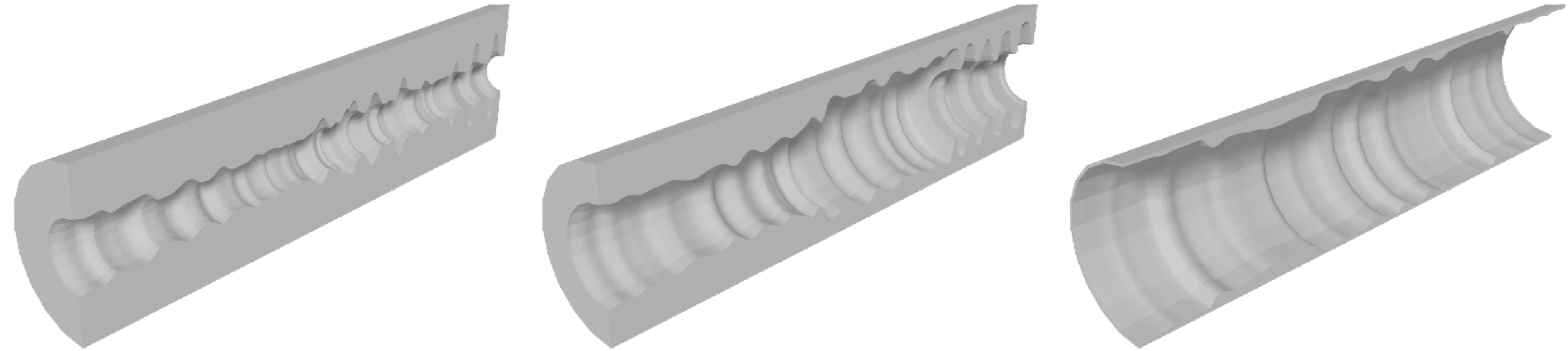
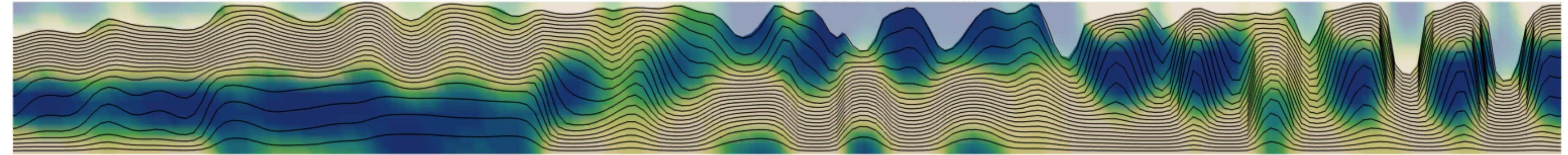
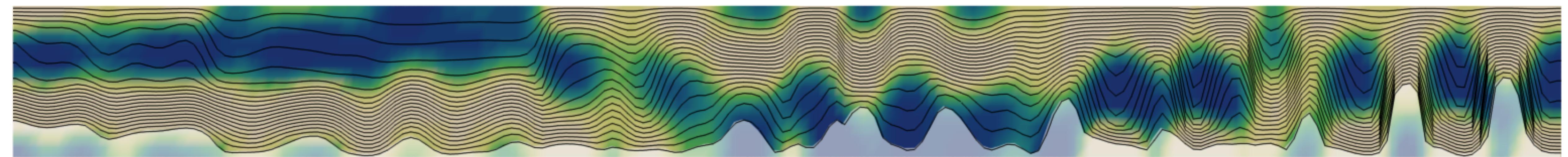


100<sup>th</sup>

$0.254 \times 10^{-2}$

burn rate [m/sec]

$1.52 \times 10^{-2}$



Thrust Profile	$n_{opt}$	$n_s$	$n_{slides}$	Total Basis	Fixed Basis		Sliding Basis	
					Time	Objective/Error	Time	Objective/Error
Constant Acceleration	20	15	14	230	1178s	349k/2.3%	288s	86k/1.1%
Constant Deceleration	50	40	7	320	4896s	867k/3.4%	621s	452k/2.7%
Two Step	20	15	7	125	191s	102k/1.1%	69s	217k/1.4%
Bucket	20	15	24	380	1006s	272k/1.8%	596s	272k/1.8%

# Why not automatic differentiation?

- Numerical differentiation is already implemented and default option in many optimization software.
- But our approach can also work with automatic differentiation.
- One disadvantage of automatic differentiation is that it cannot be used with truly black-box components.