

# PHYS454 HW2

by Edward Sanchez

References: <https://ssd.jpl.nasa.gov/horizons/manual.html>  
<https://ssd.jpl.nasa.gov/horizons/manual.html#observer-table>  
<https://ssd.jpl.nasa.gov/horizons/manual.html#obsquan>  
<https://ssd.jpl.nasa.gov/horizons/tutorial.html>  
<https://astroquery.readthedocs.io/en/latest/jplhorizons/jplhorizons.html>

All files available in GitHub <https://github.com>

## Prove Kepler's second law numerically

```
In [1]: # 1a. Use JPL Horizons Web app (https://ssd.jpl.nasa.gov/horizons/app.html#/) to retrieve
# about an object orbiting the sun. Try to show Kepler's 2nd law by numerically estimating
# the area swept out by the orbit during a short time duration and comparing that area to an equal time interval.
# Location along the orbit.
from astropy.coordinates import SkyCoord
from astroquery.jplhorizons import Horizons
from astropy.time import Time

# Mercury has an orbital period of 88 days so I use a 90-day period
mercury_start_time = '2024-01-01T00:00:00.00'
mercury_end_time = '2024-03-29T23:59:59.00'

# Generate ephemeris for Mercury
# https://ssd.jpl.nasa.gov/horizons/tutorial.html
mercury = Horizons('199', location='@sun', epochs={'start':mercury_start_time,'stop':mercury_end_time})
table_mercury = mercury.vectors(refplane='ecliptic')
coord_mercury = SkyCoord(table_mercury['x'].quantity, table_mercury['y'].quantity, table_mercury['z'].quantity,
                        representation_type='cartesian', frame='icrs', obstime=mercury_start_time)

print(coord_mercury)
```

```

<SkyCoord (ICRS): (x, y, z) in AU
[(-0.27463037, 0.20036221, 4.15640302e-02),
 (-0.29593486, 0.17809692, 4.16986213e-02),
 (-0.31516174, 0.15457935, 4.15403085e-02),
 (-0.33227883, 0.13002522, 4.11037681e-02),
 (-0.3472754 , 0.1046397 , 4.04047825e-02),
 (-0.36015893, 0.07861603, 3.94598307e-02),
 (-0.37095213, 0.05213478, 3.82857501e-02),
 (-0.37969021, 0.02536349, 3.68994632e-02),
 (-0.38641843, -0.00154315, 3.53177634e-02),
 (-0.39119005, -0.02844298, 3.35571523e-02),
 (-0.39406447, -0.05520578, 3.16337205e-02),
 (-0.39510575, -0.08171261, 2.95630637e-02),
 (-0.39438133, -0.10785505, 2.73602286e-02),
 (-0.39196101, -0.13353441, 2.50396816e-02),
 (-0.3879161 , -0.15866096, 2.26152963e-02),
 (-0.38231873, -0.18315316, 2.01003539e-02),
 (-0.37524137, -0.20693692, 1.75075548e-02),
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 (-0.3569358 , -0.25211609, 1.21364039e-02),
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 (-0.33357242, -0.29373013, 6.59267598e-03),
 (-0.32017 , -0.31307633, 3.78236124e-03),
 (-0.30571254, -0.33139123, 9.59552223e-04),
 (-0.29026798, -0.34863648, -1.86638334e-03),
 (-0.27390339, -0.36477702, -4.68642954e-03),
 (-0.25668499, -0.37978079, -7.49189169e-03),
 (-0.23867826, -0.39361845, -1.02743669e-02),
 (-0.21994797, -0.40626311, -1.30257158e-02),
 (-0.20055834, -0.41769014, -1.57380356e-02),
 (-0.18057311, -0.42787697, -1.84036342e-02),
 (-0.16005568, -0.4368029 , -2.10150062e-02),
 (-0.13906925, -0.444444903, -2.35648097e-02),
 (-0.11767694, -0.4507981 , -2.60458449e-02),
 (-0.09594194, -0.45583443, -2.84510337e-02),
 (-0.07392766, -0.45954385, -3.07734008e-02),
 (-0.05169791, -0.46191364, -3.30060564e-02),
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 (-0.00685003, -0.4625908 , -3.71750054e-02),
 ( 0.01563714, -0.46088009, -3.90978086e-02),
 ( 0.03807752, -0.45779367, -4.09038952e-02),
 ( 0.06040292, -0.45332641, -4.25865914e-02),
 ( 0.08254372, -0.44747494, -4.41392363e-02),
 ( 0.10442872, -0.44023776, -4.55551766e-02),
 ( 0.12598494, -0.43161544, -4.68277637e-02),
 ( 0.14713742, -0.42161079, -4.79503540e-02),
 ( 0.16780908, -0.41022916, -4.89163124e-02),
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 ( 0.20738971, -0.38337074, -5.03518888e-02),
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 ( 0.2610846 , -0.33307094, -5.11664214e-02),
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 ( 0.29204366, -0.29314021, -5.07429129e-02),
 ( 0.30578327, -0.27135946, -5.02232141e-02),
 ( 0.31822788, -0.24843038, -4.94908887e-02),
 ( 0.32927285, -0.22440958, -4.85409670e-02),
 ( 0.33881131, -0.19936301, -4.73690377e-02),
 ( 0.34673468, -0.17336712, -4.59713862e-02),
 ( 0.35293337, -0.14650996, -4.43451582e-02),

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```
( 0.35729778, -0.11889249, -4.24885493e-02),
( 0.35971951, -0.09062979, -4.04010236e-02),
( 0.36009302, -0.06185233, -3.80835608e-02),
( 0.35831755, -0.03270711, -3.55389326e-02),
( 0.35429956, -0.00335863, -3.27720037e-02),
( 0.3479556 , 0.02601035, -2.97900518e-02),
( 0.33921558, 0.0551986 , -2.66030968e-02),
( 0.32802661, 0.0839866 , -2.32242234e-02),
( 0.31435709, 0.11213767, -1.96698764e-02),
( 0.29820105, 0.13939982, -1.59601017e-02),
( 0.27958262, 0.16550886, -1.21187007e-02),
( 0.2585601 , 0.19019263, -8.17326326e-03),
( 0.23522953, 0.2131765 , -4.15504161e-03),
( 0.20972722, 0.23419008, -9.86352091e-05),
( 0.18223086, 0.25297488, 3.95853485e-03),
( 0.15295878, 0.26929258, 7.97696511e-03),
( 0.12216726, 0.28293337, 1.19160039e-02),
( 0.09014563, 0.29372377, 1.57349363e-02),
( 0.05720938, 0.30153321, 1.93941535e-02),
( 0.02369172, 0.30627885, 2.28563214e-02),
(-0.01006591, 0.30792811, 2.60874623e-02),
(-0.0437237 , 0.30649883, 2.90578637e-02),
(-0.07695284, 0.30205696, 3.17427557e-02),
(-0.10944382, 0.29471223, 3.41227176e-02),
(-0.14091325, 0.28461219, 3.61838116e-02),
(-0.17110871, 0.27193521, 3.79174636e-02),
(-0.19981179, 0.25688313, 3.93201344e-02),
(-0.22683933, 0.23967393, 4.03928359e-02),
(-0.252043 , 0.22053502, 4.11405491e-02),
(-0.27530786, 0.19969726, 4.15715958e-02)]>
```

## Observer & Vector Tables

If an observer or vector table has been requested, the @ symbol may be dropped; the Earth will be assumed if only an integer like 675 or a name fragment like Palom is input.

However, if you are trying to specify an observing site not on Earth, you MUST use the @ symbol for correct interpretation. For example, if an observer table as seen from the Sun is desired, it must be specified as @10 or @sun. Specifying 10 only will select the Caussols site.

For location 0 and ssb refer to solar system barycenter (SSB).

For id 1 = Mercury Barycenter and 199 = Mercury.

For Mercury and Venus, there is no difference between planet-center and system barycenter (1=199, 2=299) as far as Horizons selection is concerned because there is only the planet: no satellites, so no offset between planet center and planetary system center-of-mass.

If you request orbital elements of the Earth (399) with respect to Sun (10), the resulting elements will contain short-period oscillations due to the Earth's motion with respect to the Earth-Moon barycenter, as well as the Sun's motion with respect to the solar system barycenter. Unless these short period motions are desired, you might want to instead request 3 with respect to 0 (Earth-Moon barycenter with respect to solar system barycenter).

<https://ssd.jpl.nasa.gov/horizons/manual.html>

```
In [2]: import matplotlib.pyplot as plt

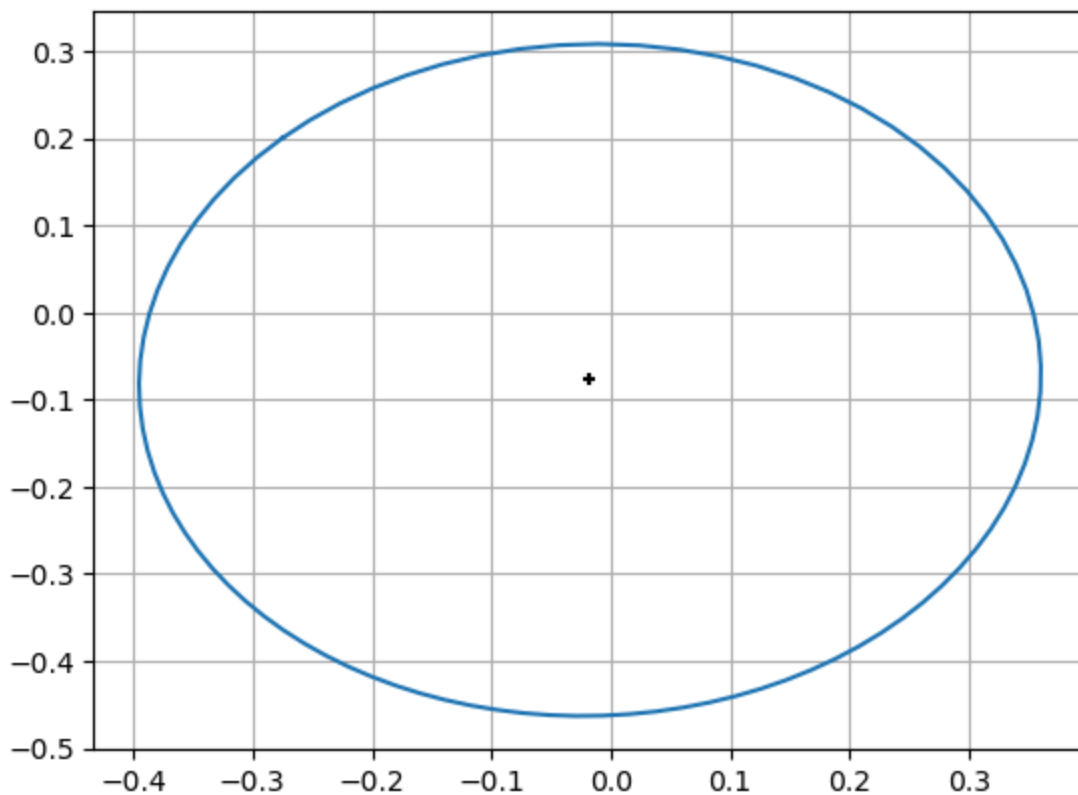
# Solar system barycenter at x = 0, y = 0
ssb_xcenter = (max(table_mercury['x']) + min(table_mercury['x'])) / 2
ssb_ycenter = (max(table_mercury['y']) + min(table_mercury['y'])) / 2
print (ssb_xcenter, ssb_ycenter)

fig1 = plt.figure(1)

def plot1():
    plt.plot(table_mercury['x'], table_mercury['y'])
    plt.scatter(ssb_xcenter, ssb_ycenter, s=25, marker='+', color='k')
    plt.grid()

plot1()
plt.show()
```

-0.01750636558649063 -0.07750222055156711



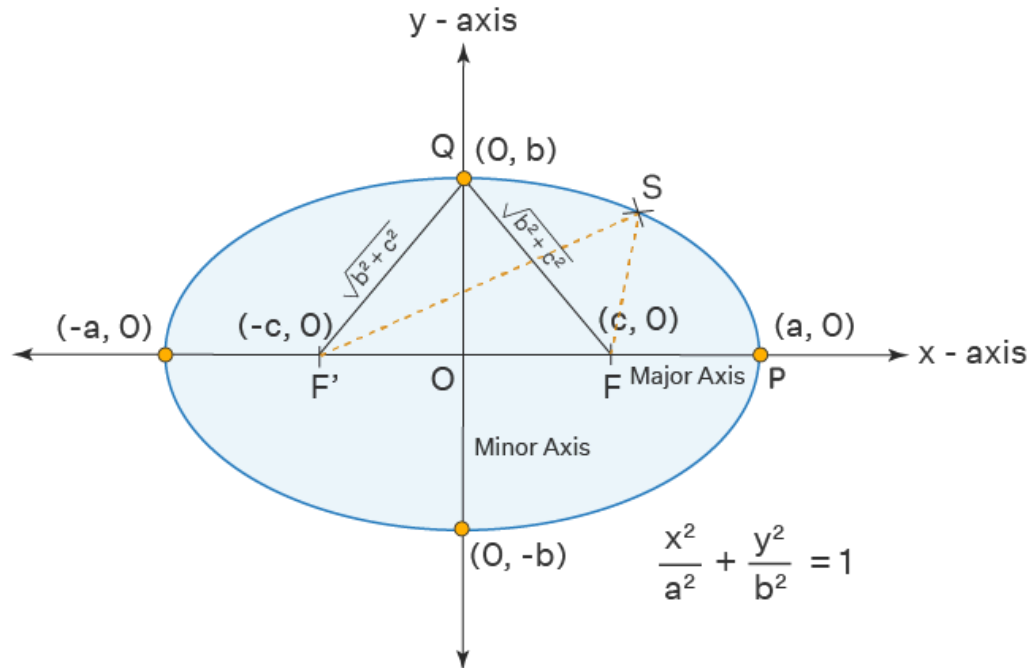
```
In [3]: # https://ssd.jpl.nasa.gov/planets/approx_pos.html
# Keplerian elements and rates
# Table 1
# Mercury semi-major axis (AU) and eccentricity
mercury_sma = 0.38709927
mercury_ecc = 0.20563593

# Calculate semi-minor axis in AU
b = mercury_sma * (1 - mercury_ecc**2)
print(f'The eccentricity of Mercury is e = {mercury_ecc:.5f}.')
print(f'The semi-major axis is {mercury_sma:.5f} AU and the semi-minor axis is {b:.5f}')
```

The eccentricity of Mercury is  $e = 0.20564$ .

The semi-major axis is 0.38710 AU and the semi-minor axis is 0.37073 AU.

## Derivation of Eccentricity of Ellipse



```
In [4]: import numpy as np

# Finding the ellipse foci
c = np.sqrt(mercury_sma**2 - b**2)
left_focus = ssb_xcenter - c
right_focus = ssb_xcenter + c
print(f'The two foci lie on ({left_focus:.2f}, {ssb_ycenter:.2f}) and ({right_focus:.2f}, {ssb_ycenter:.2f}).')
```

The two foci lie on (-0.13, -0.08) and (0.09, -0.08).

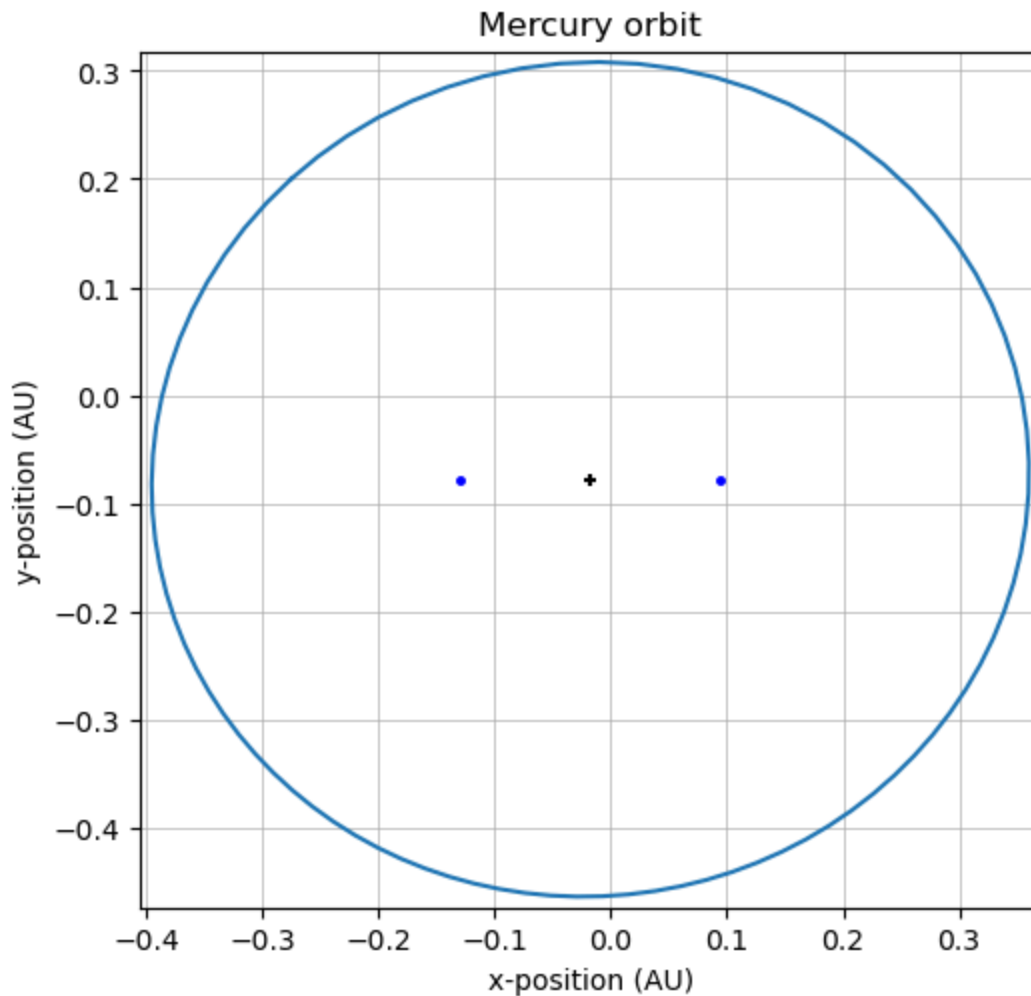
```
In [5]: # Foci Location
foci_locations = np.array([[left_focus, ssb_ycenter],
                           [right_focus, ssb_ycenter]])

# Plot Mercury orbit with new information and more accuracy
fig2 = plt.figure(2, figsize=(round(15*mercury_sma,2), round(15*b,2)))

def plot2():
    plot1()
    plt.scatter(foci_locations[:, 0], foci_locations[:, 1], s=25, marker='.', color='b')

    plt.title('Mercury orbit')
    plt.xlabel('x-position (AU)')
    plt.ylabel('y-position (AU)')
    plt.xlim(min(table_mercury['x'] - 0.01), max(table_mercury['x'] + 0.01))
    plt.ylim(min(table_mercury['y'] - 0.01), max(table_mercury['y'] + 0.01))
    plt.grid(linewidth=0.5)
```

```
plot2()
plt.show()
```



```
In [13]: # Visualization of triangles used for area calculations
import matplotlib.patches as patches

start_point = 0

# Defining the first triangle
vertices1 = [(left_focus, ssb_ycenter), (table_mercury['x'][start_point], table_mercury['y'][start_point]),
             (table_mercury['x'][start_point+10], table_mercury['y'][start_point+10])]
triangle1 = patches.Polygon(vertices1, closed=True, edgecolor='g', facecolor='none')

triangle_base = np.array([[right_focus, ssb_ycenter],
                           [table_mercury['x'][start_point], table_mercury['y'][start_point]]])
triangle_height = np.array([[table_mercury['x'][start_point], table_mercury['y'][start_point]],
                             [table_mercury['x'][start_point+10], table_mercury['y'][start_point+10]]])
triangle_hypotenuse = np.array([[table_mercury['x'][start_point+10], table_mercury['y'][start_point+10]],
                                 [right_focus, ssb_ycenter]])

# Defining the second triangle
vertices2 = [(left_focus, ssb_ycenter), (table_mercury['x'][start_point+45], table_mercury['y'][start_point+45]),
             (table_mercury['x'][start_point+55], table_mercury['y'][start_point+55])]
triangle2 = patches.Polygon(vertices2, closed=True, edgecolor='g', facecolor='none')

vertices2b = [(right_focus, ssb_ycenter), (table_mercury['x'][start_point+45], table_mercury['y'][start_point+45]),
              (table_mercury['x'][start_point+55], table_mercury['y'][start_point+55])]
```

```

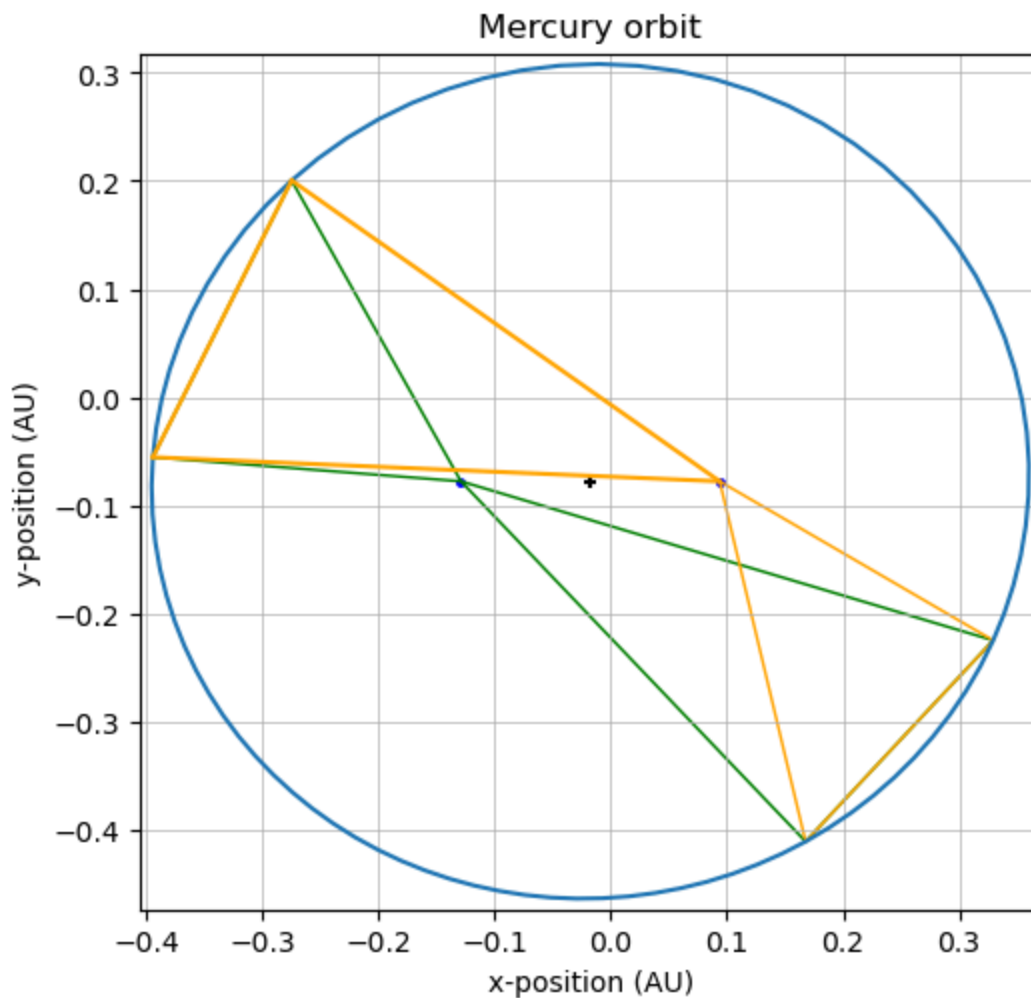
triangle2b = patches.Polygon(vertices2b, closed=True, edgecolor='orange', facecolor='r')

fig3 = plt.figure(3, figsize=(round(15*mercury_sma,2), round(15*b,2)))

def plot3():
    plot2()
    plt.plot(triangle_base[:, 0], triangle_base[:, 1], color='orange')
    plt.plot(triangle_height[:, 0], triangle_height[:, 1], color='orange')
    plt.plot(triangle_hypotenuse[:, 0], triangle_hypotenuse[:, 1], color='orange')
    ax = plt.gca()
    ax.add_patch(triangle1)
    ax.add_patch(triangle2)
    ax.add_patch(triangle2b)

plot3()
plt.show()

```



Those triangles do not look like they have equal areas...

## Setting the center at (0,0)

```

In [7]: # Function to calculate the swept area using triangles
def calculate_area(x, y):
    return 0.5 * np.sum((x[1:] - x[:-1]) * (y[:-1] + y[1:]))

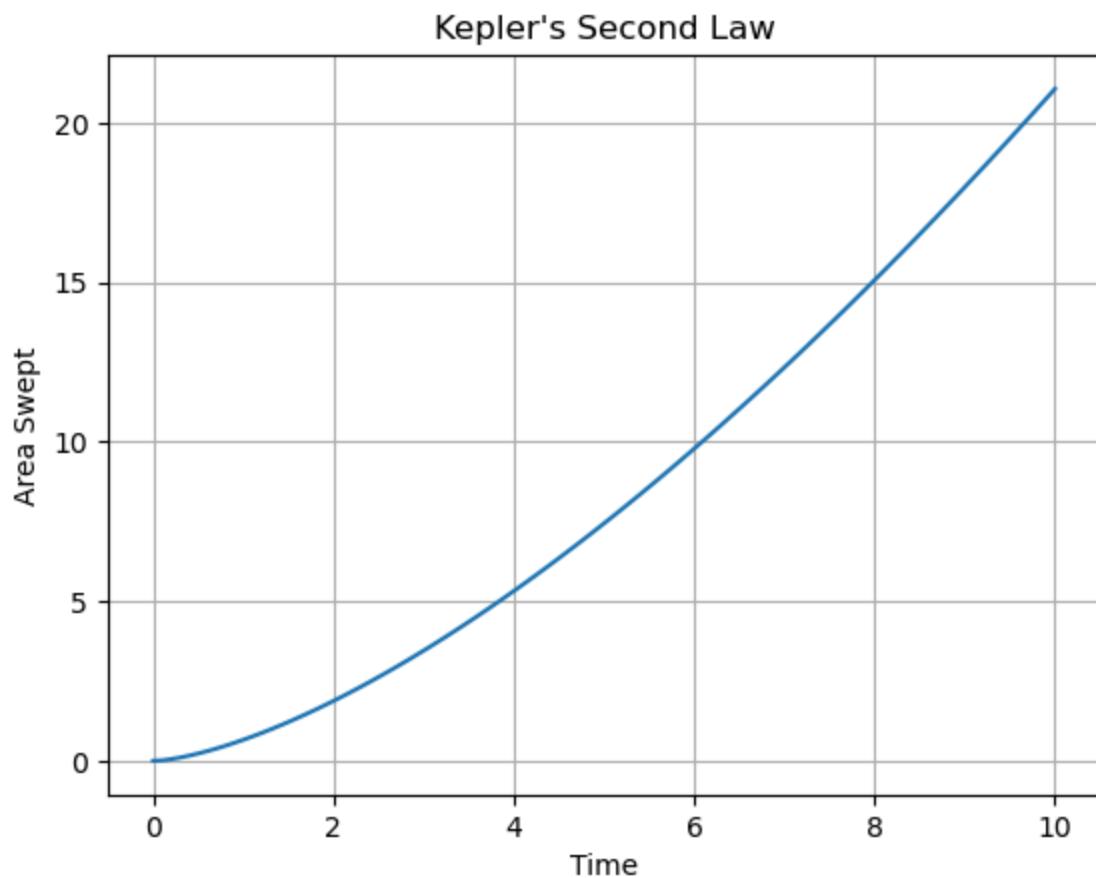
# Starting parameters for first period

```

```
time = np.linspace(0, 10, 100) # Time intervals
radius = np.sqrt(time) # Example radius (for demonstration purposes)

# Calculate areas swept by the line segment
areas = np.zeros_like(time)
for i in range(1, len(time)):
    area = calculate_area(time[:i+1], radius[:i+1])
    areas[i] = area

# Plotting
plt.plot(time, areas)
plt.xlabel('Time')
plt.ylabel('Area Swept')
plt.title("Kepler's Second Law")
plt.grid(True)
plt.show()
```



In [ ]: