Orientation Tracking

Chao Qu*

University of Pennsylvania quchao@seas.upenn.edu

I. Introduction

ORientation Tgracking

II. Pre-processing

The raw IMU data we get is 10-bit ADC values, which technically ranges from 0 to 1023. Since it's difficult to use those values directly in our filter, we have to convert them so that they have real-world physical units and meanings. Our goal here is to convert the readings from the micro-controller to acceleration with a unit of m/s² and angular velocity with a unit of rad/s. The equations used for converting raw values to real values are shown in Eq.1 and Eq.2.

$$\vec{a} = (\vec{a}_{raw} - \vec{a}_{hias}) * s_a \tag{1}$$

$$\vec{\omega} = (\vec{\omega}_{raw} - \vec{\omega}_{bias}) * s_{\omega}$$
 (2)

where \vec{a}_{bias} is [511.5, 511.5, 511.5] and $\vec{\omega}_{bias}$ is determined empirically by averaging the first 100 readings of one of the IMU data and is [373.63,375.20,369.66]. s_a and s_ω are the scale factors and their values are 0.0106 and 0.0171.

III. Methods

For the main unscented kalman filter algorithm, I basically followed the steps outlined in Kraft's paper [1]. For calculating quaternion mean, I used the method proposed by Cheon [2] instead of gradient descent. Another difference is that Kraft uses only 2n Sigma points unform weights, while Cheon uses 2n + 1 Sigma

points with weights determined by parameters such as α , β and κ . Here I will describe the algorithm in detail. The following content is either from Kraft or from Cheon.

I. Process Model

Process model in kalman filter takes the from of

$$x_{k+1} = A(x_k, w_k) \tag{3}$$

where x is the state vector and w is the process noise.

- II. Measurement Model
- III. Initialization
- IV. Prediction
- V. Correction

IV. Discussion

REFERENCES

- [1] Edgar Kraft, *A quaternion-based unscented kalman filter for orientation tracking*. Information Fusion, 2003. Proceedings of the Sixth International Conference on Information Fusion, Vol. 1 (2003), pp. 47-54.
- [2] Y.-J. Cheon and J.-H. Kim, Unscented filtering in a unit quaternion space for spacecraft attitude estimation. IEEE International Symposium on Industrial Electronics (ISIE 2007) (2007), pp. 66-71.

^{*}A thank you or further information