



Preparing for Potential Closure of European Airspaces due to Re-entering Space Objects

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Summary



- What this study is about
- Why performing it
- Who: institutional and geographical scope
- How: method, assumptions and simplifications
- Reproducibility and Thanks
- Future work
- Q&A

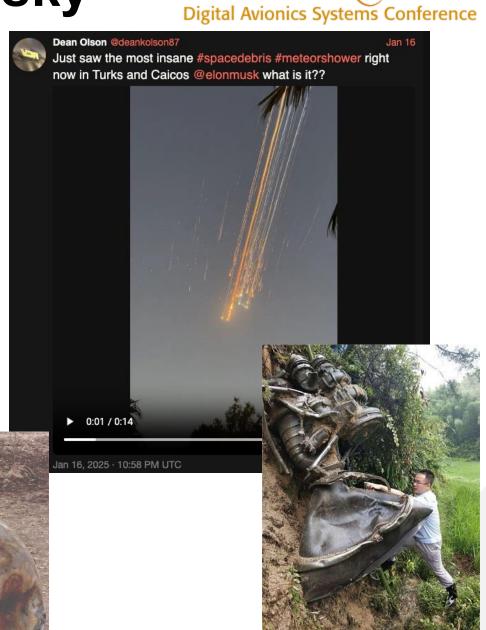


What: debris falling from the sky

impact on (EU) Aviation

- Space missions result in debris reentering Earth's atmosphere
- Re-entries can be <u>controlled</u> or uncontrolled (launch failure, satellite end of life, ...)
- We focus on impact on <u>airborne flights</u>, not on people/buildings/... on the ground

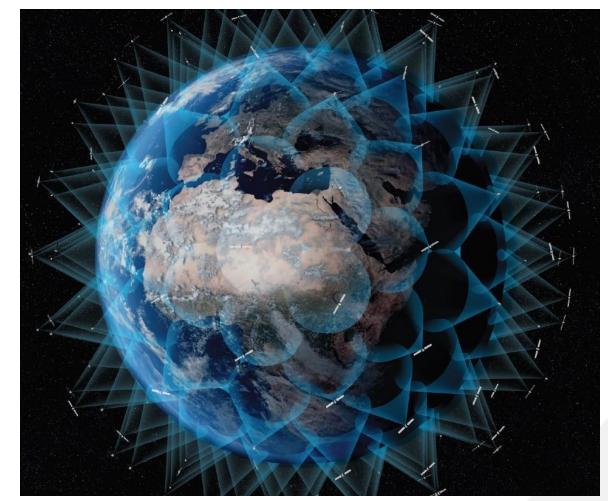




Why dealing with Space Debris in Aviation? 1/2



- Safety: Damage caused by re-entering debris to a flying aircraft could cause <u>hundreds</u> of casualties
- Awareness: growing number of LEO satellites (<u>constellations of thousands!</u>) increases the occurrence of uncontrolled re-entries.
 - Let's quantify the risk!
- Resilience: be ready to consistently put measures in place on short notice & rationally (rather than emotionally)





Why dealing with Space Debris

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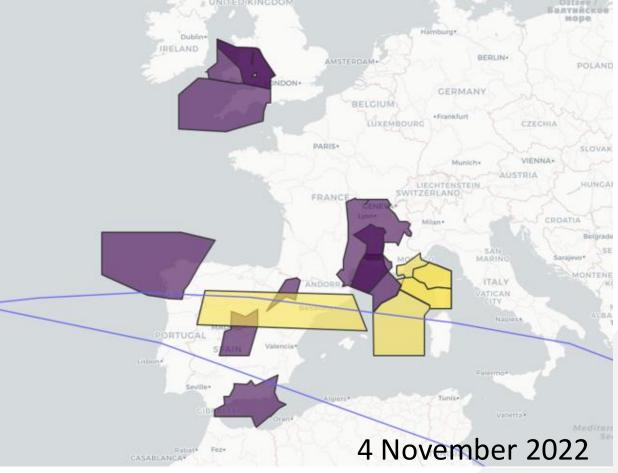
in Aviation? 2/2

 lack of harmonization of responses among States impacted by Long March 5B reentry on 4th Nov 2022

Solar panels fail Major breakup Subsequent breakup **♦ 18 km** Ground footprint 2000 km

Long March 5B reentry tracks (blue line) over EUzero rate (closure) NOTAMs in yellow

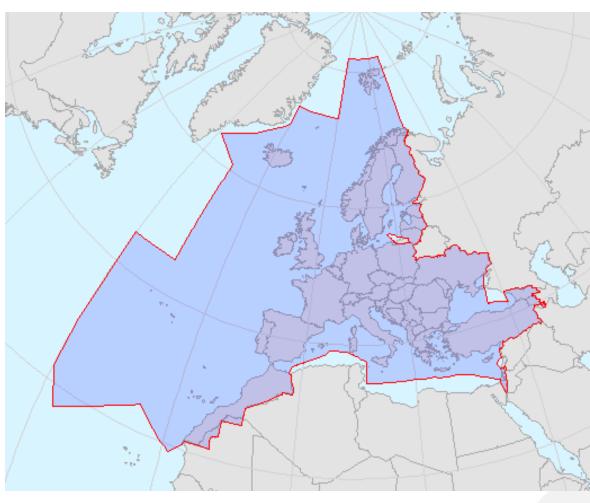
non-zero rate (advisory) NOTAMs in purple. [3]



Who: EUROCONTROL

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- 43 Member States' FIRs (NM Area, blue area)
- Analysis for bounding box containing all FIRs
- Use full flight trajectories as IFR flights filed/flown by airspace users crossing NM Area; past works used as proxies for trajectories
 - Population density
 - ADS-B position reports (including on ground and no flight type distinction, IFR/VFR, GA, ...)



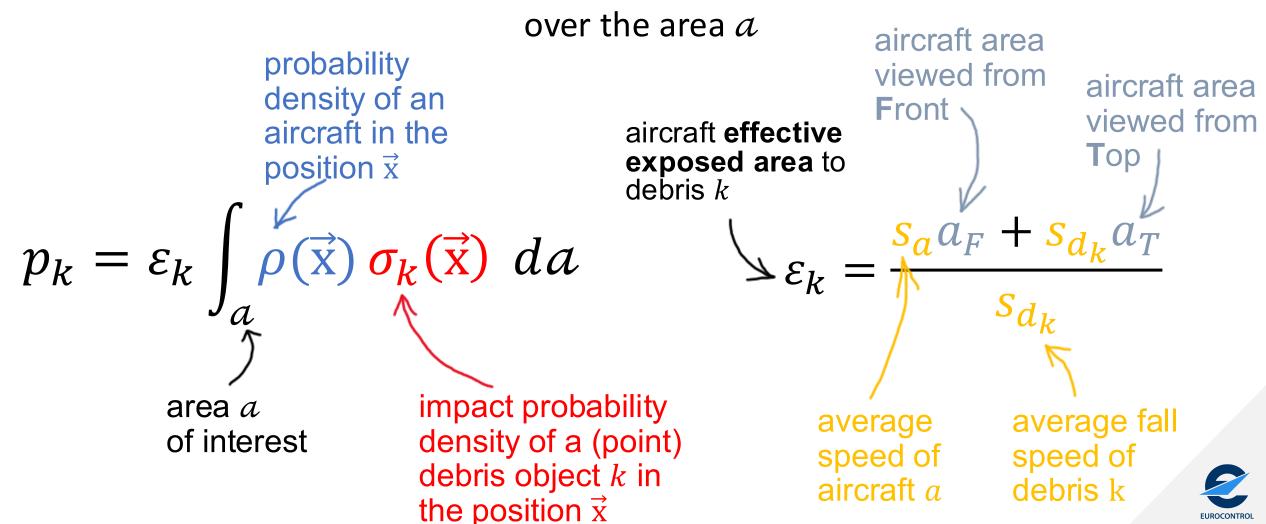
Area, α , of the study



Method: Step 1, general formulation



probability p_k of a debris (point) object k impacting an aircraft (point) object

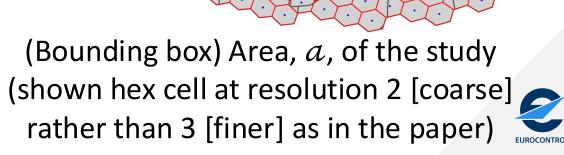


Step 2.1, discretize & assumptions

- Discretize using Uber's H3 cells resolution 3 hex:
 - edge ~69 km / 37.3 NM
 - area $\sim 14000 \text{ km}^2 / 5400 \text{ mi}^2$
- Trajectories sampled every 30s

<u>Assumptions</u>

- 1 debris, 1 piece
- Constant aircraft speed = cruise speed for relevant aircraft type
- Point debris all falling vertically at 145 mph [2]



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Step 2.2, discretize + assumptions



debris
$$k$$
 impact probability for cell h density of an aircraft being in cell h $\rho(h)\sigma_k(h)$

impact probability density of debris k in cell h



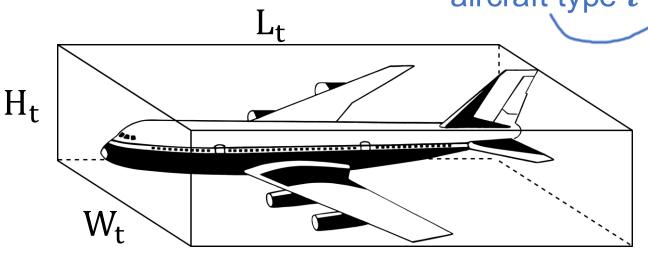
Step 2.3, discretize + assumptions

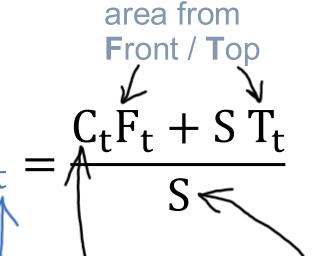


aircraft effective exposed area

$$\varepsilon_k = \frac{s_a a_F + s_{d_k} a_T}{s_{d_k}}$$

from debris *k* to aircraft type *t*





cruise speed for aircraft type *t*

constant fall speed for any debris

$$F_t = W_t \cdot H_t$$

$$T_t = W_t \cdot L_t$$



Method: step 2







Step 3, aircraft probability density

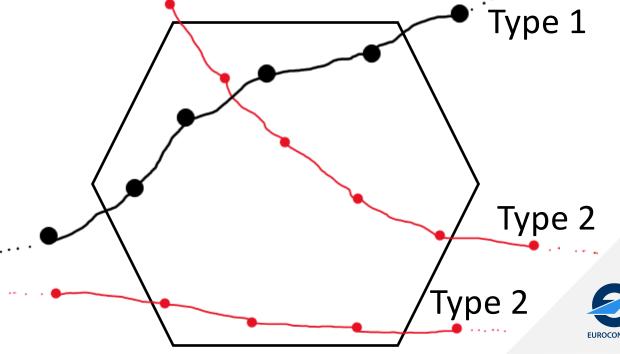


$$p_k = \sum_{h} p_k(h) = \varepsilon_k \sum_{h} \rho(h) \sigma_k(h) = \sum_{h} \sigma_k(h) \sum_{t} \varepsilon_t \rho_t(h)$$

 $\rho_t(h)$ is JUST the sum of the number of points per type t in cell h (per interval of time, i.e. hour)

Each trajectory point is 30s apart

probability of aircraft of type t being in cell h



Step 4.1, impact probability density



- A debris k in **uncontrolled** re-entry tends to fly **circular orbit** of inclination i
- uncertainties in atmospheric drag and the gravitational effects of the Equatorial

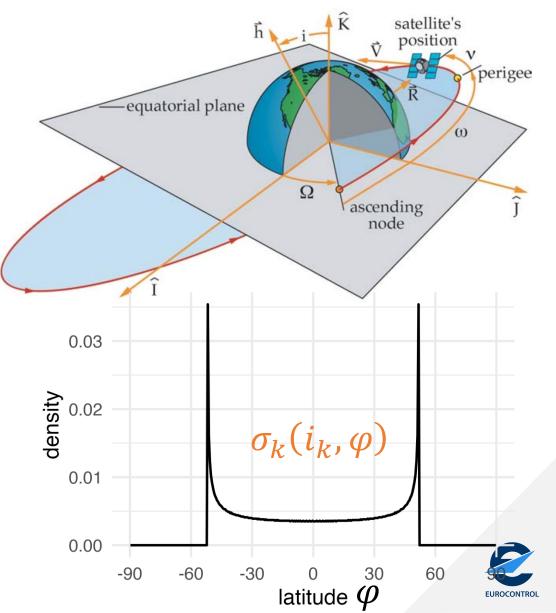
bulge
$$\rightarrow \sigma(\theta) = \frac{1}{2\pi}$$

complete uncertainty on longitude θ

• Density distribution for latitude φ only depends on orbit inclination i

$$\sigma_k(h) = \sigma_k(h, i, \theta, \varphi)$$

= $\sigma_k(h, \theta) \cdot \sigma_k(h, i, \varphi)$



Step 4.2, impact probability density

-90

-30

30

latitude



30

latitude

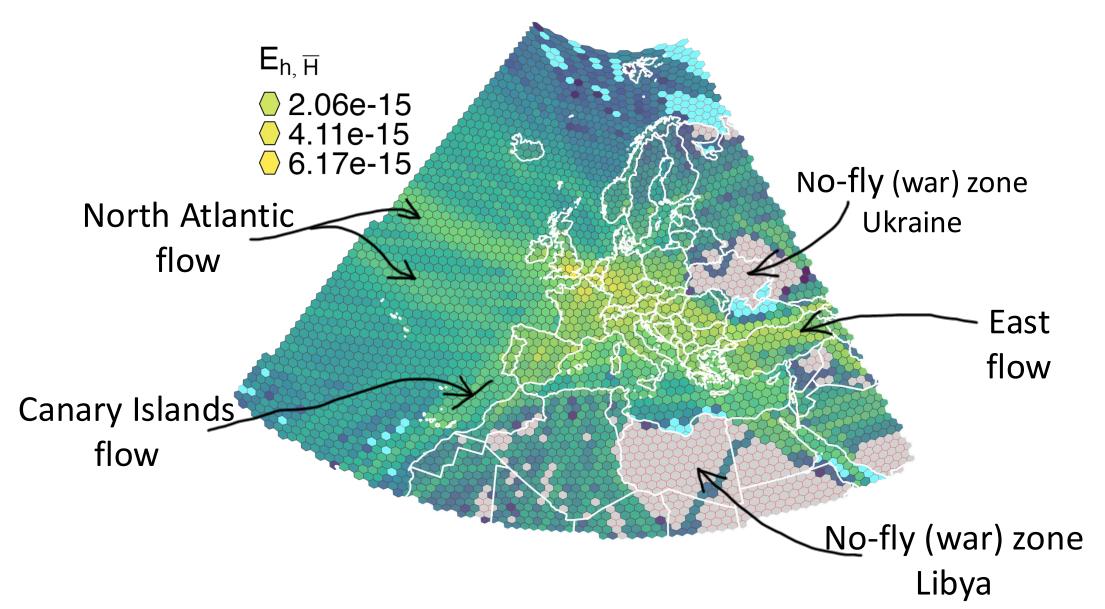
60

1968 debris 4.03e-12 re-entries • We learn from the past 25 years: .48e-11 2.56e-11 3.64e-11 combine all past debris densities by latitude φ 0.03 $\rightarrow \Sigma \rightarrow$ 0.03 0.006 density 0.02 density density 0.01 60 90 0.01 0.002 0.00 atitude 90 0.00 latitude

0.000

All combined, 2024-07-05 hourly

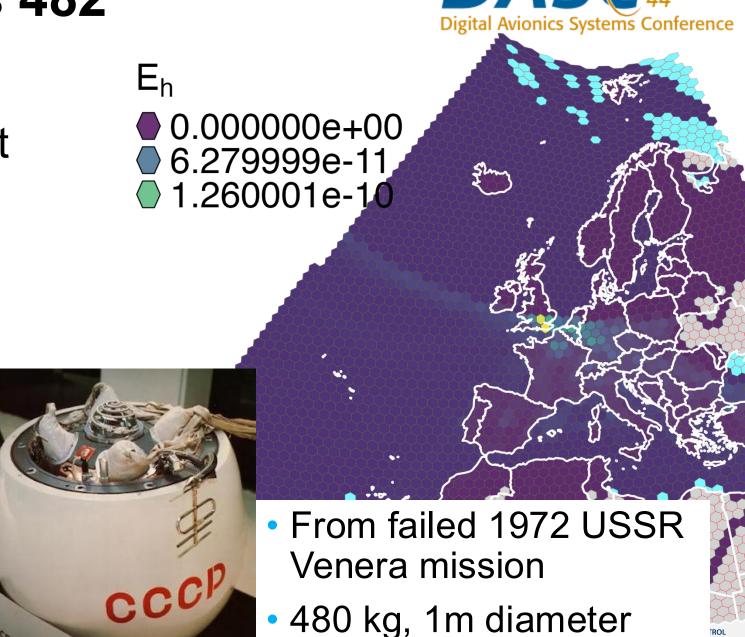






Use Case: Kosmos 482

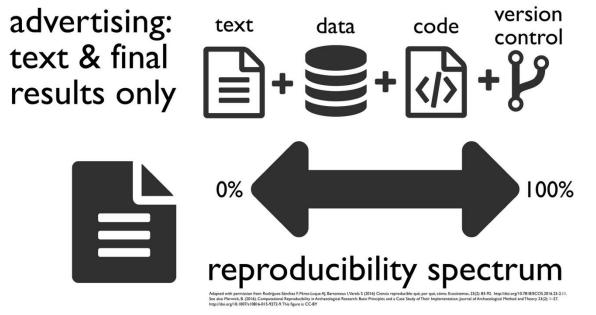
- Inclination $i = 51.95^{\circ}$
- Use trajectories of busiest day in 2024
- Highest probability collision risk of 2.63 · 10⁻¹⁰ at a cell overlapping with London Heathrow
- Ten times less probable than being struck by lightining



Reproducible research, 1/2



- Paper source (in Quarto format) and Figures reproduction (in R) (1)
- aviodebris R package for processing (2)
- (partial) trajectory data (1)
 - (1) https://github.com/euctrl-pru/flight density space debris
 - (2) https://github.com/euctrl-pru/aviodebris



science: text, code & data available, linked & licensed





Reproducible research, 2/2



We are particularly indebted to

- E. Wright, A. Boley, and M. Byers for their work and availability of Python code that made our learning and understanding of their research much easier
- Jonathan C. McDowell for his tireless effort in keeping alive and up to date the "General Catalog of Artificial Space Objects"



Future work



- Consider planned growing traffic and mega-constellations (with a certain rate of uncontrolled re-entries)
- Compute casualties probabilities using PAX per aircraft type
- Develop procedure for measures (airspace closure, re-routing) to deal with future uncontrolled re-entries or launch failures
- (Consider people densities at airports ?)







Thank you!

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