

Lab 7 Report 403

François-Eliott Rousseau 260670000

Mohamed Reda El Khili 260678513

Ismail Faruk 260663521

1)

```
1 %% Q1
2 - Km=0.0077;
3 - Rm=2.6;
4 - Lm=180;
5 - Vmax=5.0;
6 - Kg=3.7;
7 - rg=0.0064;
8 - mc=0.526;
9 - Kx=2.28*(10^-5);
10 - Kth=0.00153;
11 - mp=0.106*2;
12 - lp=0.168*2;
13 - g=9.81;
14
15 - b = (Kg*Km) / (rg);
16
17 - A = [0, 1, 0, 0;
18       0, (-b^2) / (mc*Rm), -(mp*g) / (mc), 0;
19       0, 0, 0, 1;
20       0, (b^2) / (mc*lp*Rm), ((mc+mp)*g) / (mc*lp), 0];
21
22 - B = [0;
23       b / (mc*Rm);
24       0;
25       -b / (mc*lp*Rm)];
26
27 - C = eye(4)
28
29 - D = [0, 0, 0, 0]'
```

Figure1: Code to define the state space system of the cart

A =

0	1.0000	0	0
0	-14.4899	-3.9538	0
0	0	0	1.0000
0	43.1247	40.9638	0

B =

0
3.2550
0
-9.6876

C =

1	0	0	0
0	1	0	0
0	0	1	0
0	0	0	1

D =

0
0
0
0

Figure 2: Matrices A, B, C and D

```

31      %% Q2
32 -      sys = ss(A,B,C,D);
33 -      tf_sys = ss2tf(A,B,C,D);
34
35 -      s = tf('s');
36 -      G = tf(sys);

tf_sys =

    0         0    3.2550   -0.0000  -95.0349
    0    3.2550   -0.0000  -95.0349         0
    0         0   -9.6876    0.0000         0
    0   -9.6876    0.0000         0         0

G =

From input to output...
      3.255 s^2 - 2.891e-15 s - 95.03
1:  -----
      s^4 + 14.49 s^3 - 40.96 s^2 - 423.1 s

      3.255 s^2 - 95.03
2:  -----
      s^3 + 14.49 s^2 - 40.96 s - 423.1

      -9.688 s
3:  -----
      s^3 + 14.49 s^2 - 40.96 s - 423.1

      -9.688 s^2
4:  -----
      s^3 + 14.49 s^2 - 40.96 s - 423.1

Continuous-time transfer function.

```

Figure 3: Definition of the transfer function on Matlab

```

38      %% Q3
39 -      pzmap(G);

```

Figure 4: pzmap function for pole placement

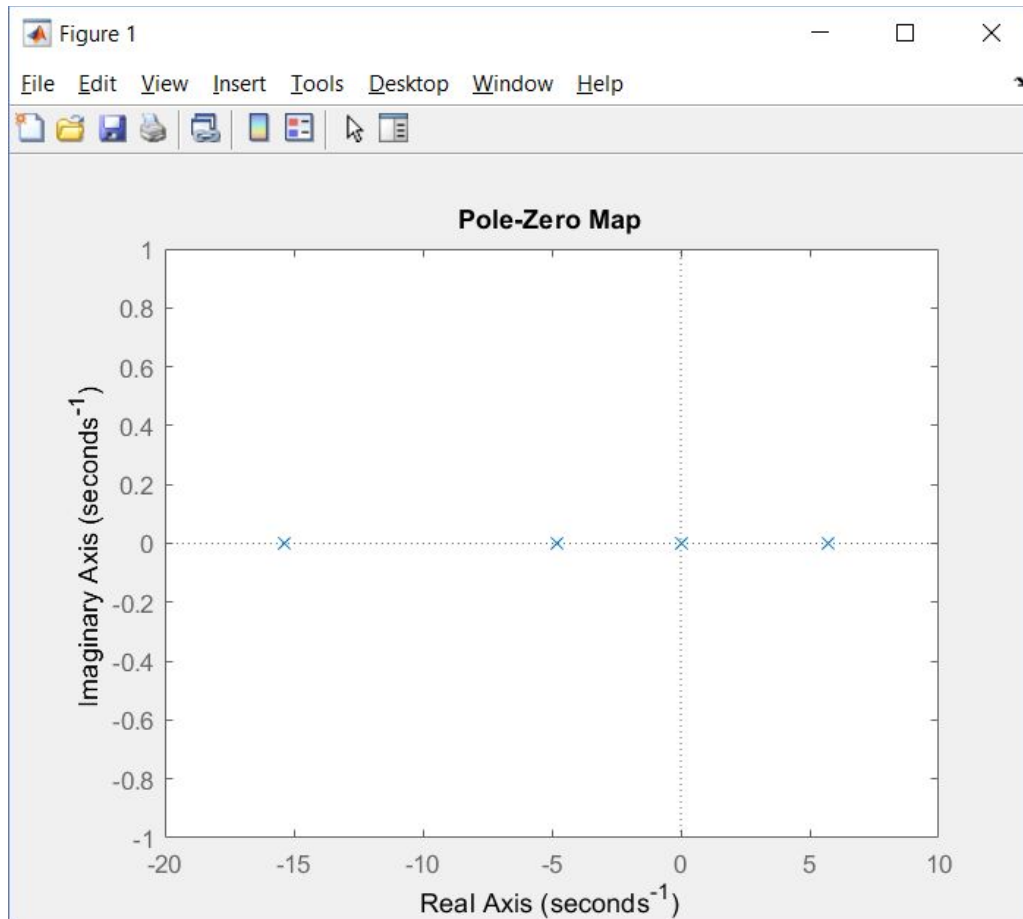


Figure 5: Pole placement

```

41 %% Q4
42 - eig(A)

ans =

     0
-15.3639
 -4.8286
  5.7026

```

Figure 6: Eigenvalues of A

The eigenvalues of the system are equal to the poles of the system.

```

44 %% Q5
45 Co = ctrb(A,B);
46 k = rank(Co);

Co =

1.0e+04 *

    0    0.0003   -0.0047    0.0722
    0.0003   -0.0047    0.0722   -1.1013
    0   -0.0010    0.0140   -0.2431
   -0.0010    0.0140   -0.2431    3.6874

k =

4

```

Figure 7: Controllability matrix

The system is controllable. The controllability has full rank.

We do not need to check for Observability because matrix C is full rank

```

52 %% Q7
53 Mp = 8.5; % ~6 deg
54 Z = ((log(Mp)^2)/(pi^2+(log(Mp)^2)))^0.5;
55 Ts = 2;
56 Wn = 3.9/(Z*Ts);
57 Pd1 = -(Z*Wn) + 1i*Wn*(1-Z^2)^0.5;
58 Pd2 = -(Z*Wn) - 1i*Wn*(1-Z^2)^0.5;

```

$M_p =$

11.5000

$Z =$

0.6138

$T_s =$

2.5000

$W_n =$

2.5417

$Pd1 =$

-1.5600 + 2.0066i

$Pd2 =$

-1.5600 - 2.0066i

```
61 %% Q8
62 - Nd1 = 10*Pd1;
63 - Nd2 = 10*Pd2;
```

Nd1 =

-15.6000 +20.0663i

Nd2 =

-15.6000 -20.0663i

```
66 %% Q9 PP for Con
67 - p = [Pd1, Pd2, Nd1, Nd2];
68 - K = place(A,B,p);
...
p =
-1.5600 + 2.0066i -1.5600 - 2.0066i -15.6000 +20.0663i -15.6000 -20.0663i

K =
-43.9140 -27.7812 -96.3840 -11.3814
```

The scope of position, velocity, angle and angle velocity were identical to that of the small stick, shown in *Lab 6 Report*.

2-6)

```
67 - Q=15*[20,0,0,0;
68 -     0,0.666,0,0;
69 -     0,0,15,0;
70 -     0,0,0,0];
71 - R= 0.01;
72 - N= 0;
73
74 - [K,S,e] = lqr(A,B,Q,R,N);
```

By gradually changing the values of R and Q, and with the help of the Lab Instructor we were able to narrow down on the values. We conducted tests with 20 different values.

Q =

300.0000	0	0	0
0	9.9900	0	0
0	0	225.0000	0
0	0	0	0

R =

0.0100

N =

0

K =

-173.2051 -99.7644 -232.0346 -23.0626

S =

165.0867	40.3286	39.9456	6.8646
40.3286	18.1040	15.1170	3.0930
39.9456	15.1170	22.3361	2.6594
6.8646	3.0930	2.6594	0.5315

e =

-99.6244 + 0.0000i
-31.6622 + 0.0000i
-2.6544 + 1.8414i
-2.6544 - 1.8414i

Increasing the value of K will decrease rise time and steady state error of the four quantities - position, velocity, angle, angle velocity - but increase the overshoot.

7)

$Q =$

150.0000	0	0	0
0	9.9900	0	0
0	0	150.0000	0
0	0	0	0

$R =$

0.1000

$N =$

0

$K =$

-38.7298 -32.8067 -90.8924 -14.9673

$S =$

109.8190	34.2149	57.9682	11.8960
34.2149	18.9486	30.5441	6.7054
57.9682	30.5441	63.9329	11.2011
11.8960	6.7054	11.2011	2.4075

$e =$

-33.9073 + 0.0000i
-13.8529 + 0.0000i
-2.4701 + 1.3171i
-2.4701 - 1.3171i

8) Matrix Q contains the costs and matrix K has the feedback gains.

System behavior after adding LQR gains, with the longer pendulum.

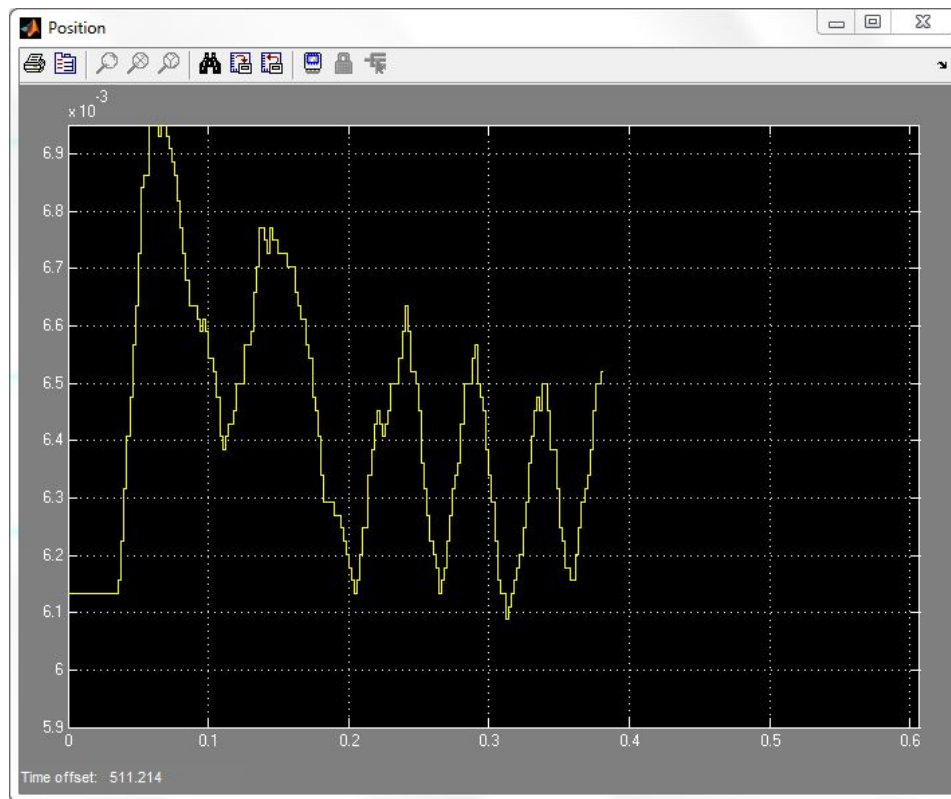


Figure: Position

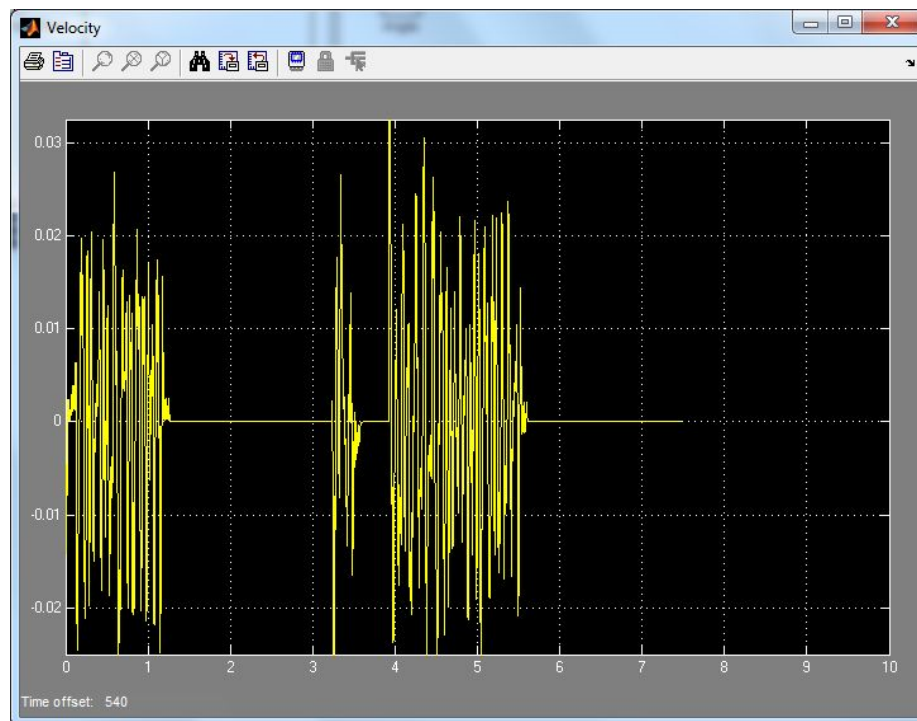


Figure: Velocity

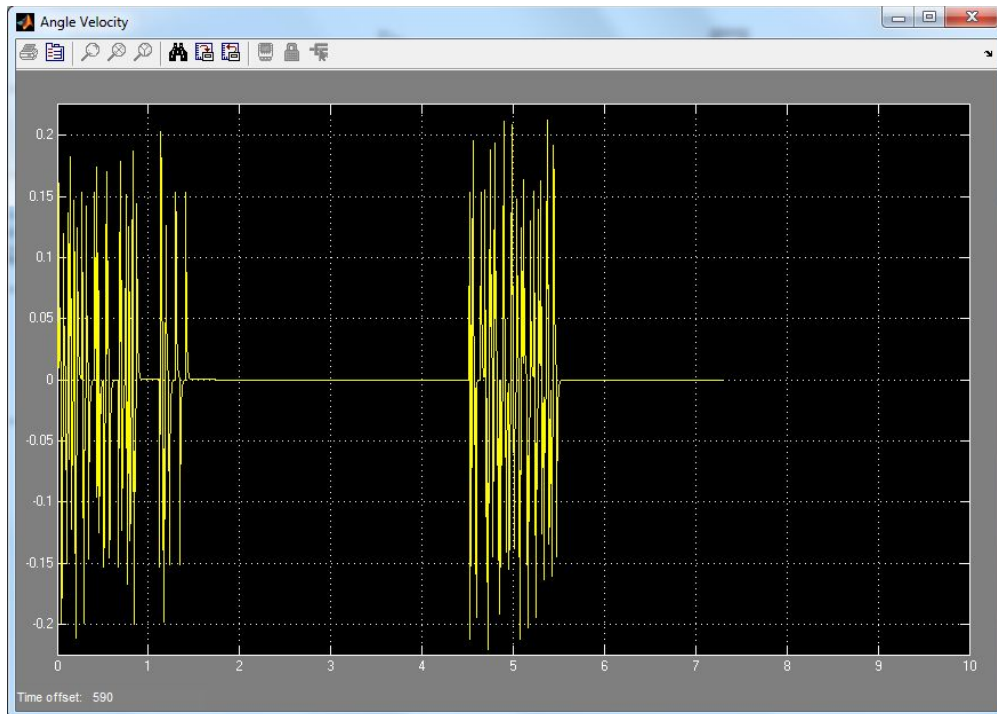


Figure: Angle

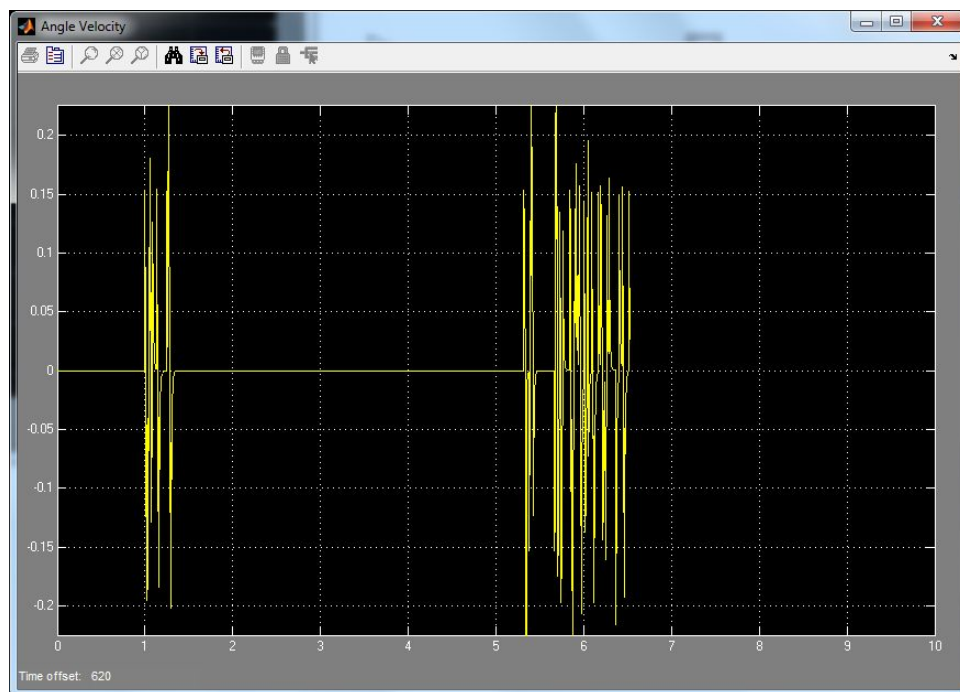


Figure: Angle Velocity

Appendix

```
%% Q1
Km=0.0077;
Rm=2.6;
Lm=180;
Vmax=5.0;
Kg=3.7;
rg=0.0064;
mc=0.526;
Kx=2.28*(10^-5);
Kth=0.00153;
mp=0.106*2;
lp=0.168*2;
g=9.81;

b = (Kg*Km)/(rg);

A = [0,1,0,0;
     0,(-b^2)/(mc*Rm),-(mp*g)/(mc),0;
     0,0,0,1;
     0,(b^2)/(mc*lp*Rm),((mc+mp)*g)/(mc*lp),0];

B = [0;
     b/(mc*Rm);
     0;
     -b/(mc*lp*Rm)];

C = eye(4);

D = [0,0,0,0]';

%%
sys = ss(A,B,C,D);
tf_sys = ss2tf(A,B,C,D);

s = tf('s');
G = tf(sys);

%%
pzmap(G);

%%
eig(A);
```

```

%% Q
Co = ctrb(A,B);
k = rank(Co);

%%
Mp = 11.5; % 11.5
Z = ((log(Mp)^2)/(pi^2+(log(Mp)^2)))^0.5;
Ts = 2.5;
Wn = 3.9/(Z*Ts);
Pd1 = -(Z*Wn) + 1i* Wn*(1-Z^2)^0.5;
Pd2 = -(Z*Wn) - 1i* Wn*(1-Z^2)^0.5;

%%
Nd1 = 10*Pd1;
Nd2 = 10*Pd2;

%% PP for Con
p = [Pd1, Pd2, Nd1, Nd2];
K = place(A,B,p);

Q=15*[10,0,0,0;
      0,0.666,0,0;
      0,0,10,0;
      0,0,0,0];
R= 0.1;
N= 0;

[K,S,e] = lqr(A,B,Q,R,N);

```