

a) condición del diseño

Flujo isoentropico dentro el sistema

$$P_0 = P_{e0}$$

$$T_0 = T_{e0}$$

$$A^* = A_e^* = A_t = 0.01 \text{ m}^2$$

$$\frac{A_e}{A^*} = \frac{0.04 \text{ m}^2}{0.01 \text{ m}^2} = 4 = \frac{(1 + 0.2 Ma_e^2)^3}{1.728 Ma_e}$$

$$\Rightarrow Ma_e = 2.94 \leftarrow \text{Supersonico}$$

$$\frac{P_{e0}}{P_e} = 33.56 \rightarrow P_e = 2.97 \text{ kPa}$$

Condición del diseño :  $P_b = P_e = 2.97 \text{ kPa}$

b) • El flujo está estrengulado

$$\hookrightarrow Ma_T = 1$$

• El flujo en la zona divergente es subsonico

$$\hookrightarrow Ma_e < 1$$

• Todo el flujo es isoentropico

$$P_{e0} = P_0$$

$$T_{e0} = T_0$$

$$A_{e0}^* = A^* = 0.01 \text{ m}^2$$

$$\frac{A_e}{A^*} = \frac{0.04 \text{ m}^2}{0.01 \text{ m}^2} = 4 = \frac{(1 + 0.2 Ma_e^2)^3}{1.728 Ma_e}$$

$$\hookrightarrow Ma_e < 1$$

$$\hookrightarrow Ma_e = 0.1465$$

-- "ma<sub>e</sub> < 1"

$$\hookrightarrow Ma_e = 0.465$$

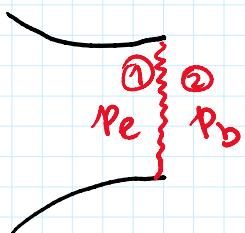
$$\hookrightarrow \frac{P_{e0}}{P_e} = 1.015$$

$$\Rightarrow P_e = \frac{100 \text{ kPa}}{1.015} = 98.5 \text{ kPa}$$

$$P_e = P_b = 98.5 \text{ kPa}$$

c) La presión mínima para que existan ondas de choque normales va a estar dada por la formación de una onda de choque normal en el plano de salida.

$\hookrightarrow$  Todo el flujo  $\rightarrow$  Isoentropia  
Hasta el Plano de Salida



$$\text{De a)} : P_e = 2.97 \text{ kPa}$$

$$Ma_e = 2.94$$

$$P_e = P_1 \\ Ma_e = Ma_1$$

$$Ma_2 = Ma_b \\ P_2 = P_b$$

$P_b$  estará dado por las relaciones de onda de choques normales:

$$\frac{P_2}{P_1} = \frac{P_b}{P_e} = \frac{1}{(k+1)} \left( 2k \frac{Ma_1^2}{Ma_e^2} - (k-1) \right)$$

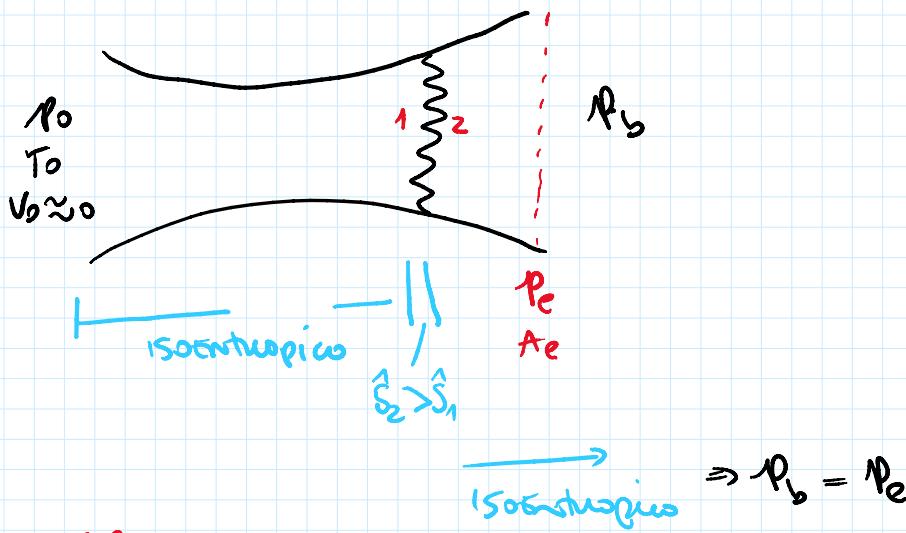
$$\underline{P_b} = 9.9175$$

$\overline{P_e}$

$$P_b = 9.9175 \cdot P_e = 9.9175 \cdot 2.94 \text{ kPa}$$

$$\underline{\underline{P_b = 29.2 \text{ kPa}}}$$

d)  $P_b = 50 \text{ kPa}$



$$\frac{P_e}{P_{01}} \cdot \frac{A_e}{A_t} = \left( \frac{P_e}{P_{0e}} \right) \cdot \left( \frac{A_e}{A_e^*} \right)$$

↑  $P_0 = 100 \text{ kPa}$   
 $\frac{P_e}{P_{01}}$   
 $\frac{A_e}{A_t}$   
 $0.04 \text{ m}^2$   
 $0.01 \text{ m}^2$

$\downarrow$   
 $0.04 \text{ m}^2$

$\downarrow$   
 $0.01 \text{ m}^2$

$\downarrow$   
 relaciones gases ideales

$$\frac{50 \text{ kPa}}{100 \text{ kPa}} \cdot \frac{0.01 \text{ m}^2}{0.01 \text{ m}^2} = 2 = \left( \frac{P_e}{P_{0e}} \right) \cdot \left( \frac{A_e}{A_e^*} \right)$$

$$2 = \left( 1 + \frac{k-1}{2} M_{ae}^2 \right)^{\frac{-k}{k-1}} \cdot \frac{1}{M_{ae}} \left( \frac{1 + \frac{1}{2}(k-1) M_{ae}^2}{1/2(k+1)} \right)^{\frac{k+1}{k-1}}$$

$$\downarrow k = 1.4$$

$$M_{ae} = 0.287$$

$$\hookrightarrow \frac{P_e}{P_{0e}} = 0.944$$

$$\hookrightarrow \frac{A_e}{A_{e^*}} = 2.118$$

Ahora:

$$\frac{P_2}{P_1} = \frac{A_t}{A_e} \cdot \frac{A_e}{A_{e^*}} = \frac{0.01 \text{ m}^2}{0.04 \text{ m}^2} \cdot 2.118 = 0.5295$$

$$\Rightarrow P_2 = 0.5295 \cdot P_1 = 0.5295 \cdot 100 \text{ kPa}$$

$$P_2 = 52.95 \text{ kPa}$$

$$\frac{P_e}{P_e} = \frac{52.95 \text{ kPa}}{50 \text{ kPa}} = 1.059 = \left( \frac{P_e}{P_{0e}} \right)^{-1} = (0.944)^{-1}$$

$$\frac{T_{0e}}{T_e} = \left( \frac{P_{0e}}{P_e} \right)^{\frac{k-1}{k}} = (1.059)^{\frac{0.4}{1.4}} = 1.016$$

$$T_{0e} = T_{01} \Rightarrow T_e = 392.96$$

Para calcular  $M_{a_1}$ :

$$\frac{P_{02}}{P_{01}} = 0.5295 = \left[ \frac{(k+1) M_{a_1}^2}{2 + (k-1) M_{a_1}^2} \right]^{\frac{k}{k-1}} \left[ \frac{k+1}{2k M_{a_1}^2 - (k-1)} \right]^{1/(k-1)}$$

↓ Al resolver

$$M_{a_1} = 2.425 \quad \leftarrow \text{Supersonico}$$

$$\frac{P_0}{P_1} = 15.2 \rightarrow P_1 = \frac{100 \text{ kPa}}{15.2} = 6.6 \text{ kPa}$$

flujos isentropicos

$$\frac{T_0}{T_1} = 2.18 \rightarrow T_1 = \frac{400 \text{ K}}{2.18} = 183.5 \text{ K}$$

$$Ma_2 = \sqrt{\frac{(k-1)Ma_1^2 + 2}{2kMa_1 - (k-1)}} \Rightarrow Ma_2 = 0.5205$$

$$Ma_1 \rightarrow \frac{P_2}{P_1} = 3.2428 \Rightarrow P_2 = P_1 \cdot 3.2428 = 21.4 \text{ kPa}$$

$\uparrow$   
6.6 kPa

$$\frac{T_2}{T_1} = 2.0643 \Rightarrow T_2 = T_1 \cdot 2.0643 = 378.9 \text{ K}$$

$\uparrow$   
183.5 K

Para el area transversal:

$$\left(\frac{A_1}{A^*}\right) = 2.425 \Rightarrow A_1 = A^* \cdot 2.425 = 0.1 \cdot 2.425$$

$A_1 = 0.2425 \text{ m}^2$

flujos isentropicos

considerando  $Ma_1$