Applied Computational Fluid Dynamics with OpenFOAM

Day - 4



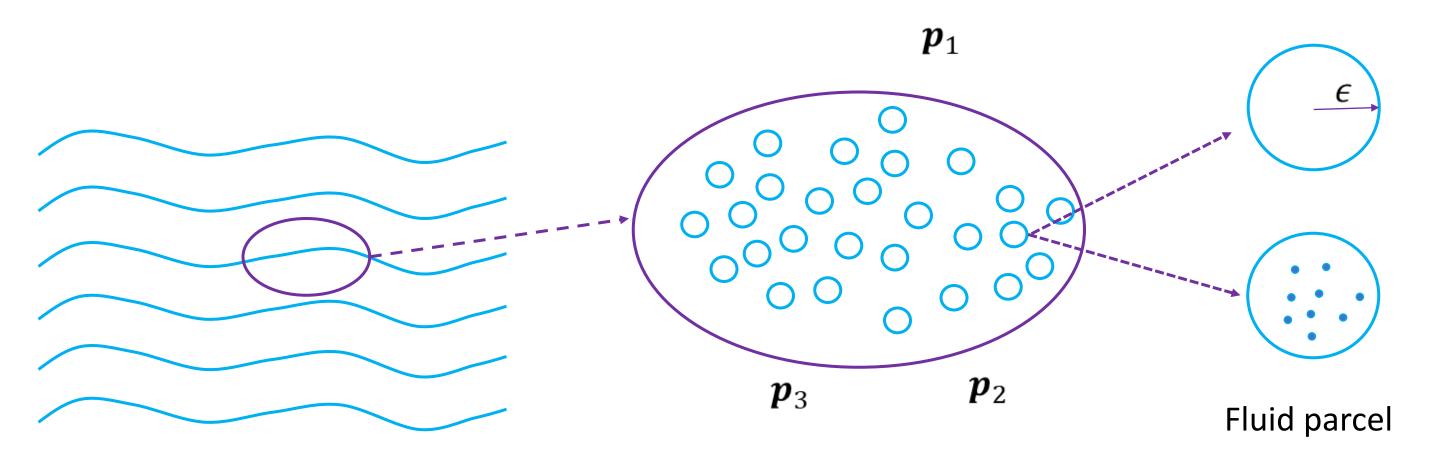


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- > Gauss Divergence Theorem
- > Reynolds Transport Theorem



Lagrangian frame – discrete phase

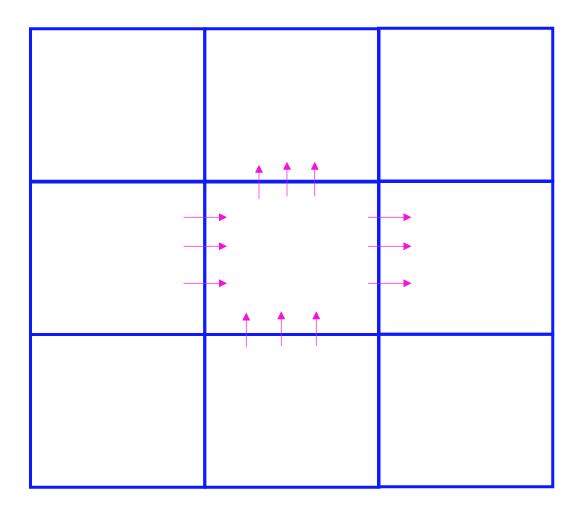


- Follow a fluid parcel $(p_1, p_2, ...)$
- Flow property at a location is obtained from the fluid parcel that happens to be at that location at that time
- Useful to derive conservation laws



Eulerian frame

- Conservation laws are applied around a fixed "control volume" in space
- Flux of quantities through the boundary is used to estimate the flow variables
- Useful to take observations at fixed locations





Governing Equations

• The general equation can be written in the form as:

$$\frac{\partial}{\partial t} (\rho \phi) + \nabla \cdot (\rho u \phi) = \nabla \cdot (\Gamma \nabla \phi) + S$$

$$\left| \frac{\partial}{\partial t} (\rho \phi) + \frac{\partial}{\partial x_j} (\rho u_j \phi) \right| = \frac{\partial}{\partial x_j} \left(\Gamma \frac{\partial \phi}{\partial x_j} \right) + S$$

$$\frac{\partial}{\partial t} \iiint \rho \phi dV + \iint \rho \phi (u.dA) = \iint \Gamma \nabla \phi. dA + \iiint S dV$$

1

2

3

4

1 Unsteady/Transient term

3 Diffusion term

Advection/Convection term

4



Governing Equations

$$\left| \frac{\partial}{\partial t} (\rho \phi) + \nabla \bullet (\rho u \phi) = \nabla \bullet (\Gamma \nabla \phi) + S \right|$$

Continuity equation:
$$\phi = 1, S = 0 \implies \left| \frac{\partial}{\partial t} (\rho) + \nabla \cdot (\rho u) = 0 \right|$$

Momentum equation:
$$\phi = u, \Gamma = \mu, S = S_u \Rightarrow \left| \frac{\partial}{\partial t} (\rho u) + \nabla \cdot (\rho u u) = \nabla \cdot (\mu \nabla u) - \nabla p + S_u \right|$$

Energy equation:
$$\phi = h, \Gamma = k / C_p, S = S_h$$
 $\Rightarrow \left| \frac{\partial}{\partial t} (\rho h) + \nabla \cdot (\rho u h) = \nabla \cdot \left(\frac{k}{C_p} \nabla h \right) + S_h \right|$

Species equation:
$$\phi = h, \Gamma = \Gamma_l, S = S_m$$
 \Rightarrow $\left| \frac{\partial}{\partial t} (\rho m_l) + \nabla \cdot (\rho u m_l) = \nabla \cdot (\Gamma_l \nabla m_l) + S_m \right|$

Turbulence equation:
$$\phi = k(\text{or})\varepsilon$$
, $\Gamma = \Gamma_k(\text{or})\Gamma_\varepsilon$, $S = S_k(\text{or})S_\varepsilon$ \Rightarrow $\left|\frac{\partial}{\partial t}(\rho k) + \nabla \cdot (\rho u k)\right| = \nabla \cdot (\Gamma_k \nabla k) + S_k$

Ideal Gas equation:
$$p = \rho RT$$



Integral Form – Differential Form

Conservation of mass

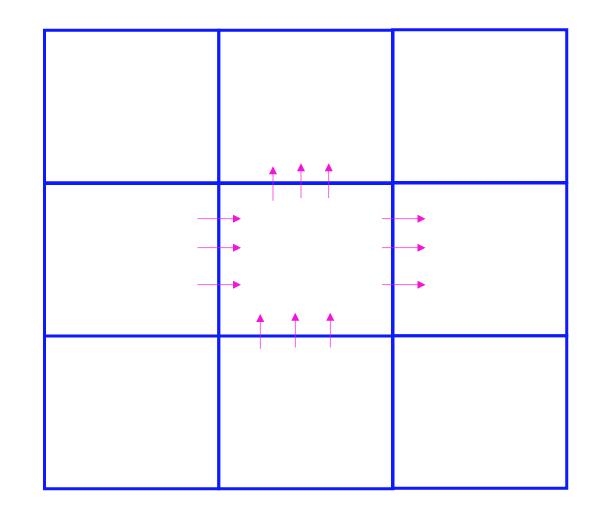
$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0$$

• Integrate over a control volume

$$\int_{V} \left(\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \boldsymbol{u}) \right) dV = 0$$

$$\int_{V} \frac{\partial \rho}{\partial t} dV + \int_{V} \nabla \cdot (\rho \boldsymbol{u}) dV = 0$$

$$\frac{\partial}{\partial t} \int_{V} \rho dV + \oint_{S} \rho \boldsymbol{u} \cdot d\boldsymbol{S} = 0$$





Gauss – divergence theorem

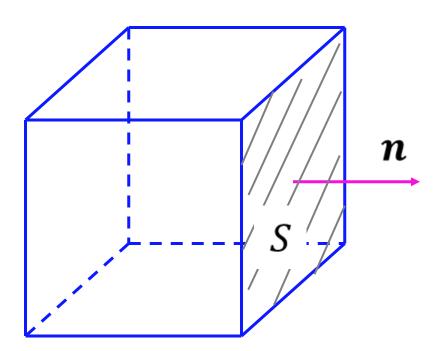


• Surface Area Vector: **S**

$$S = |S|n$$

$$S = Sn$$

- Extensive property: Φ
- Intensive property: ϕ
 - Conserved property calculated by mass
- Mass: $\rho V = \int_{V} \rho dV$

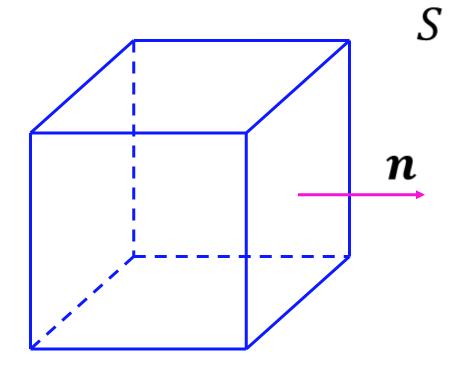




• For a vector: **F**

$$\int (\nabla \cdot \mathbf{F}) d\mathbf{V} \approx \sum \mathbf{F} \cdot \mathbf{S}$$

• Rate of change of a quantity over a control volume = Rate of flow through control surface.





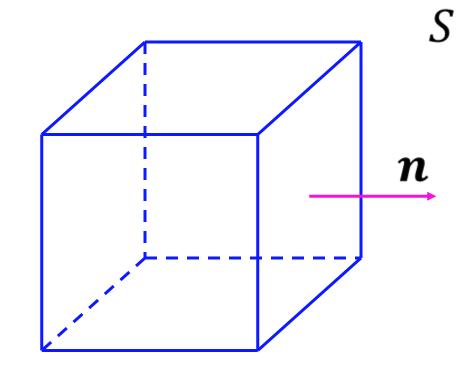
• For a vector: **F**

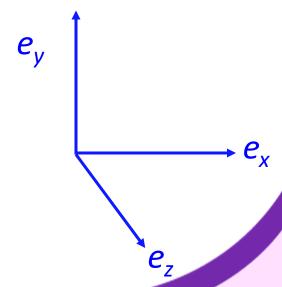
$$\int (\nabla \cdot F) dV \approx \sum F \cdot S$$

Let:
$$\mathbf{F} = x\mathbf{e}_x + y\mathbf{e}_y$$

$$\int (\nabla \cdot \mathbf{F}) d\mathbf{V} = \int \left(\frac{\partial}{\partial x} \mathbf{e}_x + \frac{\partial}{\partial y} \mathbf{e}_y \right) \cdot \left(x \mathbf{e}_x + y \mathbf{e}_y \right) d\mathbf{V}$$

$$\nabla \cdot \boldsymbol{u} = \left(\frac{\partial}{\partial x}\boldsymbol{e}_x + \frac{\partial}{\partial y}\boldsymbol{e}_y + \frac{\partial}{\partial z}\boldsymbol{e}_z\right) \left(u\boldsymbol{e}_x + v\boldsymbol{e}_y + w\boldsymbol{e}_z\right) = \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z}\right)$$







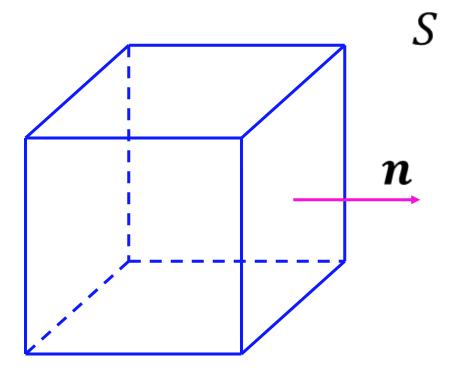
$$\int (\nabla \cdot F) dV \approx \sum F \cdot S$$

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$$\mathbf{F} = x\mathbf{e}_x + y\mathbf{e}_y$$

$$\int (\nabla \cdot \mathbf{F}) d\mathbf{V} = \int \left(\frac{\partial}{\partial x} \mathbf{e}_x + \frac{\partial}{\partial y} \mathbf{e}_y \right) \cdot \left(x \mathbf{e}_x + y \mathbf{e}_y \right) d\mathbf{V}$$

$$= \int \left(\frac{\partial x}{\partial x} + \frac{\partial y}{\partial y}\right) dV = 2V$$

$$\sum \mathbf{F} \cdot \mathbf{S} = \sum (x\mathbf{e}_x + y\mathbf{e}_y) \cdot \mathbf{S}$$

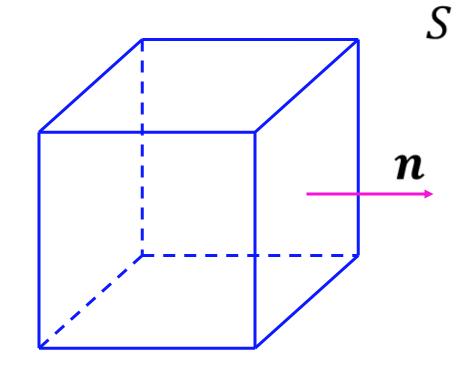


$$S = \Delta x \Delta y$$

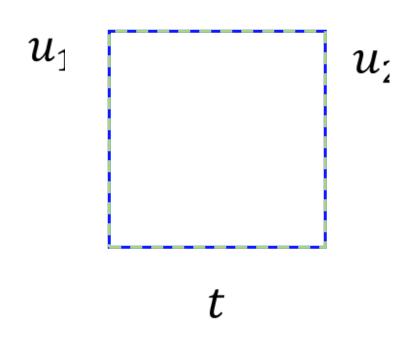
$$V = \Delta x \Delta y \Delta z$$

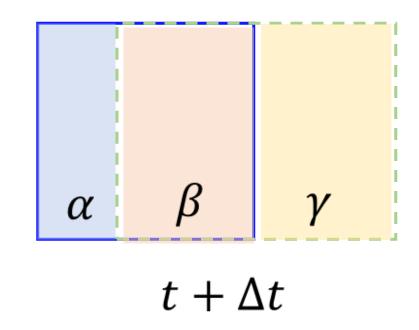


$$\frac{d}{dt} \int_{V_S} \rho \phi dV = \frac{d}{dt} \int_{V_{CV}} \rho \phi dV + \int_{S} \rho \phi \mathbf{u} \cdot \mathbf{n} dS$$









$$\frac{d}{dt} \int_{V_S} \rho \phi dV = \frac{d}{dt} \int_{V_{CV}} \rho \phi dV + \int_{S} \rho \phi \mathbf{u} \cdot \mathbf{n} dS$$

Control volume

System

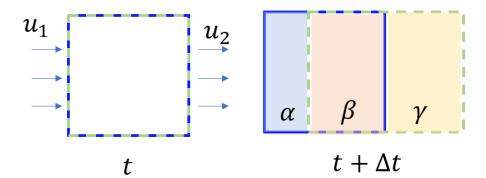
(1)
$$\Phi_S(t) = \Phi_{CV}(t)$$

(2)
$$\Phi_S(t + \Delta t) = \Phi_{CV}(t + \Delta t) - \Phi_\alpha + \Phi_\gamma$$



(1)
$$\Phi_S(t) = \Phi_{CV}(t)$$

(2)
$$\Phi_{\mathcal{S}}(t + \Delta t) = \Phi_{\mathcal{C}\mathcal{V}}(t + \Delta t) - \Phi_{\alpha} + \Phi_{\gamma}$$



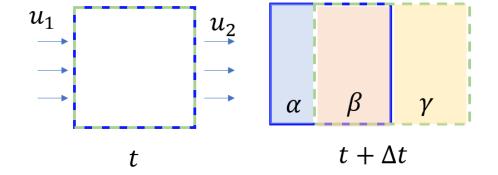
Subtract (1) from (2)

$$\left(\frac{\Phi(t+\Delta t)-\Phi(t)}{\Delta t}\right)_{System} = \left(\frac{\Phi(t+\Delta t)-\Phi(t)}{\Delta t}\right)_{CV} - \dot{\Phi}_{\alpha} + \dot{\Phi}_{\gamma}$$

$$\frac{d\Phi}{dt}_{System} = \frac{d\Phi}{dt}_{CV} - \dot{\Phi}_{\alpha} + \dot{\Phi}_{\gamma}$$



$$\frac{d\Phi}{dt}_{System} = \frac{d\Phi}{dt}_{CV} - \dot{\Phi}_{\alpha} + \dot{\Phi}_{\gamma}$$



$$\Phi_{\alpha} = \phi_{\alpha} m_{\alpha} = \phi_{\alpha} \rho_{\alpha} V_{\alpha} = \phi_{\alpha} \rho_{\alpha} (u_1 \Delta t) S$$

Using $\Phi = \int \rho \phi dV$ and net flux as $\int \rho \phi u \cdot n dS$

$$\frac{d}{dt} \int_{V_S} \rho \phi dV = \frac{d}{dt} \int_{V_{CV}} \rho \phi dV + \int_{S} \rho \phi \mathbf{u} \cdot \mathbf{n} dS$$



Conservation Laws

Conservation of mass

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \boldsymbol{u}) = 0; \quad \nabla \cdot (\rho \boldsymbol{u}) = \frac{\partial (\rho u)}{\partial x} + \frac{\partial (\rho v)}{\partial y} + \frac{\partial (\rho w)}{\partial z}$$

$$\frac{d}{dt} \int_{V_S} \rho \phi dV = \frac{d}{dt} \int_{V_{CV}} \rho \phi dV + \int_{S} \rho \phi \mathbf{u} \cdot \mathbf{n} dS$$

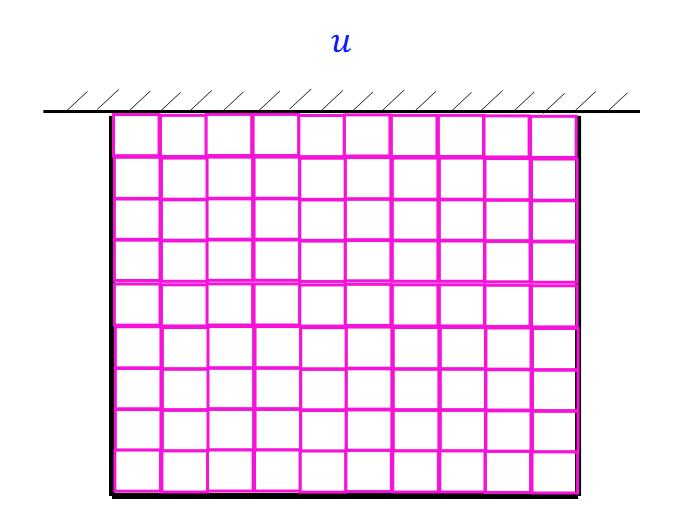
Conservation of momentum

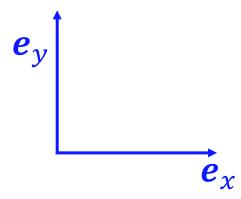
$$\frac{\partial \rho \boldsymbol{u}}{\partial t} + \nabla \cdot (\rho \boldsymbol{u} \boldsymbol{u}) = -\nabla p + \nabla \cdot \boldsymbol{\sigma} + \rho \boldsymbol{g};$$

Scalar conservation law

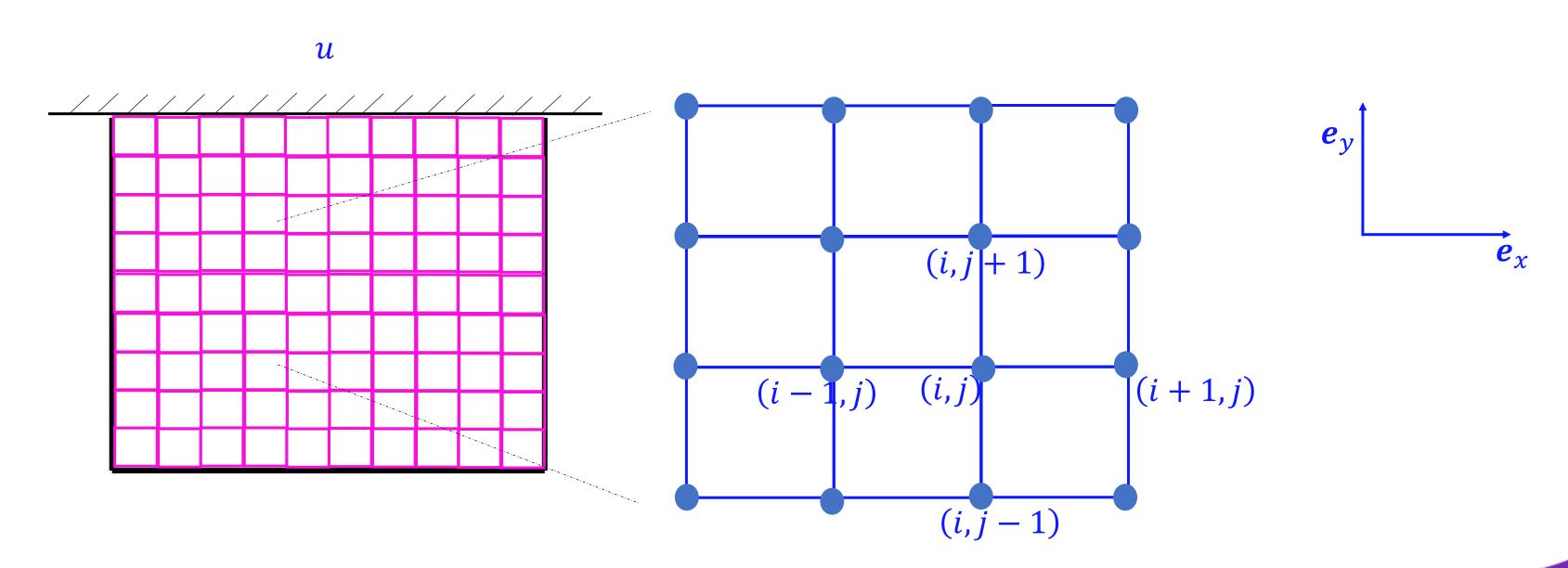
$$\frac{\partial \rho \phi}{\partial t} + \nabla \cdot (\rho \boldsymbol{u} \phi) = \nabla \cdot (\Gamma \nabla \phi) + S_{\phi}$$





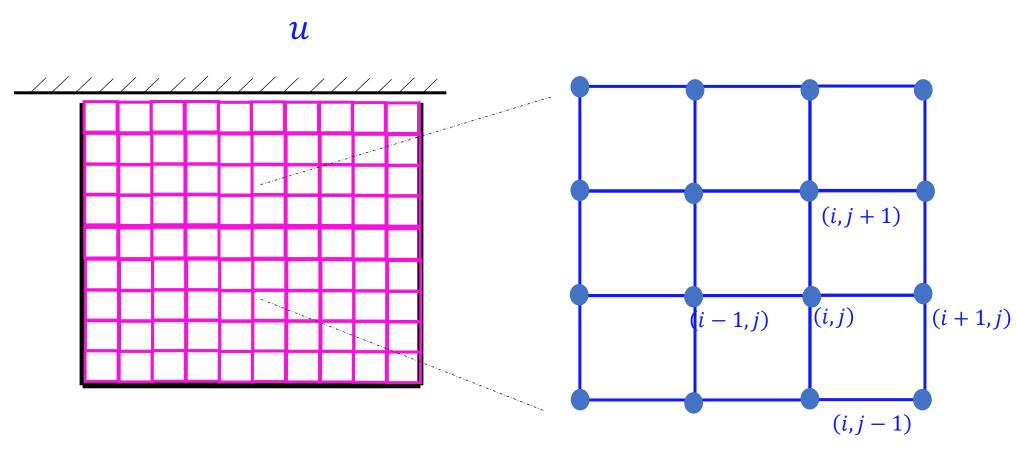


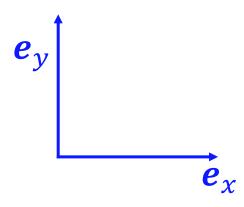


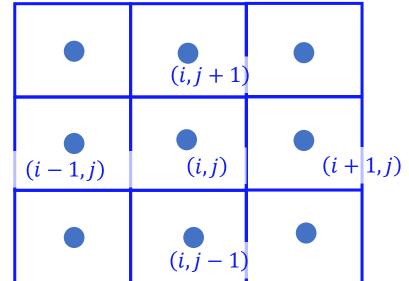


• Flow properties are assigned at each grid locations.





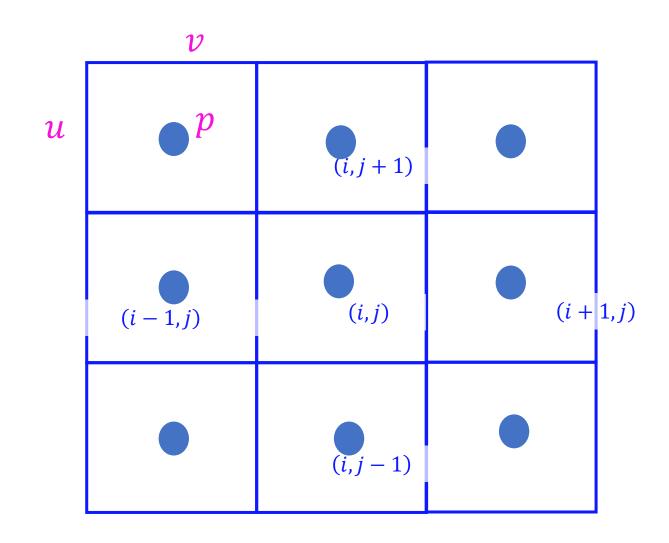




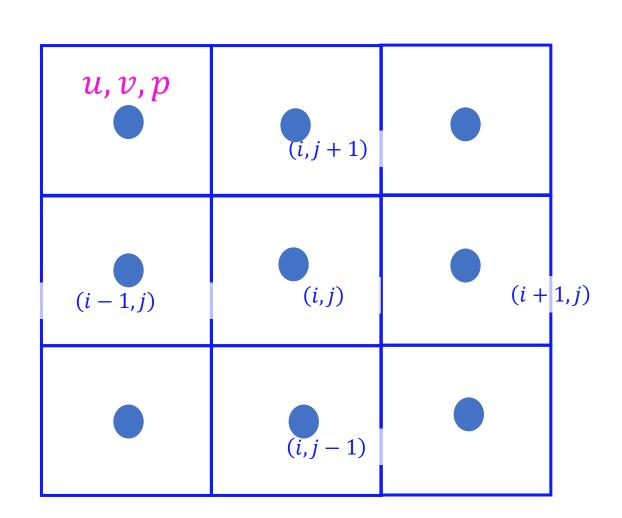
Cell centered representation



Cell centered representation



Staggered grid



Collocated grid

 \boldsymbol{e}_{χ}

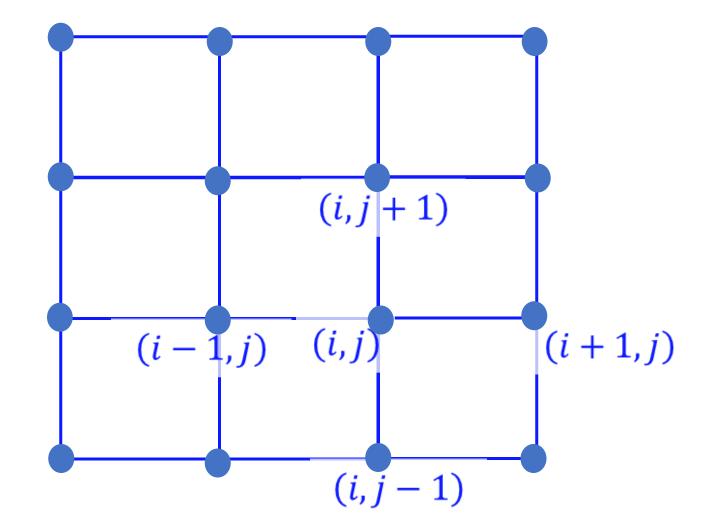
 \boldsymbol{e}_{y}



Finite Difference – Finite Volume

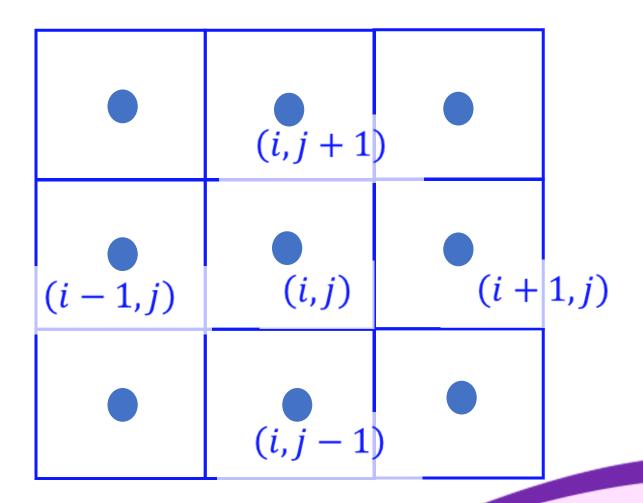
Differential form

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \boldsymbol{u}) = 0$$



Integral form

$$\frac{\partial}{\partial t} \int_{V} \rho dV + \oint_{S} \rho \boldsymbol{u} \cdot d\boldsymbol{S} = 0$$







Taylor series and FDM

Taylor series:

$$\rho(x_{i+1}) = \rho(x_i) + \Delta x_i \left(\frac{\partial \rho}{\partial x}\right)_i + O(\Delta x_i^2)$$

$$\left(\frac{\partial \rho}{\partial x}\right)_{i} = \frac{\rho(x_{i+1}) - \rho(x_{i})}{\Delta x_{i}} + \frac{1}{\Delta x_{i}} O(\Delta x_{i}^{2})$$

$$\left(\frac{\partial \rho}{\partial x}\right)_{i} = \frac{\rho(x_{i+1}) - \rho(x_{i})}{\Delta x_{i}} + \frac{O(\Delta x_{i})}{\Delta x_{i}}$$

$$\left(\frac{\partial \rho}{\partial x}\right)_{i} \approx \frac{\rho(x_{i+1}) - \rho(x_{i})}{\Delta x_{i}}$$

Finite difference

$$\nabla \rho = \left(\frac{\partial}{\partial x} \boldsymbol{e}_x + \frac{\partial}{\partial y} \boldsymbol{e}_y + \frac{\partial}{\partial z} \boldsymbol{e}_z\right) \rho = \left(\frac{\partial \rho}{\partial x} \boldsymbol{e}_x + \frac{\partial \rho}{\partial y} \boldsymbol{e}_y + \frac{\partial \rho}{\partial z} \boldsymbol{e}_z\right)$$

$$\left(\frac{\partial \rho}{\partial x}\right)_{i} = \frac{\rho(x_{i+1}) - \rho(x_{i})}{\Delta x_{i}} + \frac{O(\Delta x_{i})}{\Delta x_{i}}$$

First order forward difference scheme



Taylor series expansion

$$x_{i-1}$$
 x_i x_{i+1} x_{i+2}

$$\rho(x_{i+1}) = \rho(x_i) + (x_{i+1} - x_i) \left(\frac{d\rho}{dx}\right)_i + (x_{i+1} - x_i)^2 \left(\frac{d^2\rho}{dx^2}\right)_i + (x_{i+1} - x_i)^3 \left(\frac{d^3\rho}{dx^3}\right)_i + \cdots$$

(1)
$$\rho(x_{i+1}) = \rho(x_i) + \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \Delta x_i^2 \left(\frac{d^2\rho}{dx^2}\right)_i + O(\Delta x_i^3)$$

(2)
$$\rho(x_{i-1}) = \rho(x_i) - \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \Delta x_i^2 \left(\frac{d^2\rho}{dx^2}\right)_i + O(\Delta x_i^3)$$



Taylor series: Central Difference Scheme (2nd order)

(1)
$$\rho(x_{i+1}) = \rho(x_i) + \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \Delta x_i^2 \left(\frac{d^2\rho}{dx^2}\right)_i + O(\Delta x_i^3)$$

$$x_{i-1}$$
 x_i x_{i+1} x_{i+2}

(2)
$$\rho(x_{i-1}) = \rho(x_i) - \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \Delta x_i^2 \left(\frac{d^2\rho}{dx^2}\right)_i + O(\Delta x_i^3)$$

Subtract (2) from (1)

$$\rho(x_{i+1}) - \rho(x_{i-1}) = 2\Delta x_i \left(\frac{d\rho}{dx}\right)_i + O(\Delta x_i^3)$$

$$\left(\frac{d\rho}{dx}\right)_{i} = \frac{\rho(x_{i+1}) - \rho(x_{i-1})}{2\Delta x_{i}} + O(\Delta x_{i}^{2})$$

Second order central difference scheme



Taylor series: Backward Difference Scheme (1st order)

$$\rho(x_{i-1}) = \rho(x_i) - \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \Delta x_i^2 \left(\frac{d^2\rho}{dx^2}\right)_i + O(\Delta x_i^3)$$

$$\rho(x_{i-1}) = \rho(x_i) - \Delta x_i \left(\frac{d\rho}{dx}\right)_i + O(\Delta x_i^2)$$

$$\rho(x_{i-1}) = \rho(x_i) - \Delta x_i \left(\frac{d\rho}{dx}\right)_i + O\left(\Delta x_i^2\right)$$

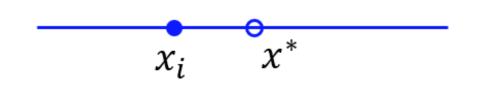
$$\left(\frac{d\rho}{dx}\right)_{i} = \frac{\rho(x_{i}) - \rho(x_{i-1})}{\Delta x_{i}} + O\left(\Delta x_{i}\right)$$

$$\left(\frac{d\rho}{dx}\right)_{i} \approx \frac{\rho(x_{i}) - \rho(x_{i-1})}{\Delta x_{i}}$$

First order backward difference scheme



Taylor series: Summary



$$\rho(x^*) = \rho(x_i) + (x^* - x_i) \left(\frac{d\rho}{dx}\right)_i + (x^* - x_i)^2 \left(\frac{d^2\rho}{dx^2}\right)_i + (x^* - x_i)^3 \left(\frac{d^3\rho}{dx^3}\right)_i + \cdots$$

$$x_{i-1}$$
 x_i x_{i+1} x_{i+2}

First order forward difference

$$\left(\frac{d\rho}{dx}\right)_i \approx \frac{\rho(x_{i+1}) - \rho(x_i)}{\Delta x_i}$$

Second order central difference

$$\left(\frac{d\rho}{dx}\right)_i \approx \frac{\rho(x_{i+1}) - \rho(x_{i-1})}{2\Delta x_i}$$

First order backward difference

$$\left(\frac{d\rho}{dx}\right)_{i} \approx \frac{\rho(x_{i}) - \rho(x_{i-1})}{\Delta x_{i}}$$



Approximating Second Order Derivative

$$\left(\frac{d^2\rho}{dx^2}\right)_i$$

$$x_{i-1}$$
 x_i x_{i+1} x_{i+2}

$$\rho(x_{i+1}) = \rho(x_i) + (x_{i+1} - x_i) \left(\frac{d\rho}{dx}\right)_i + \frac{(x_{i+1} - x_i)^2}{2} \left(\frac{d^2\rho}{dx^2}\right)_i + \frac{(x_{i+1} - x_i)^3}{6} \left(\frac{d^3\rho}{dx^3}\right)_i + \frac{(x_{i+1} - x_i)^4}{24} \left(\frac{d^4\rho}{dx^4}\right)_i + \cdots$$

$$(1) \quad \rho(x_{i+1}) = \rho(x_i) + \Delta x_i \left(\frac{d\rho}{dx}\right)_i + \frac{\Delta x_i^2}{2} \left(\frac{d^2\rho}{dx^2}\right)_i + \frac{\Delta x_i^3}{6} \left(\frac{d^3\rho}{dx^3}\right)_i + \frac{\Delta x_i^4}{24} \left(\frac{d^4\rho}{dx^4}\right)_i + O(\Delta x_i^5)$$



Approximating Second Order Derivative

$$x_{i-1}$$
 x_i x_{i+1} x_{i+2}

Add (1) and (2)

$$\rho(x_{i+1}) + \rho(x_{i-1}) = 2\rho(x_i) + \Delta x_i^2 \left(\frac{d^2 \rho}{dx^2}\right)_i + \frac{\Delta x_i^4}{12} \left(\frac{d^4 \rho}{dx^4}\right)_i + \cdots$$

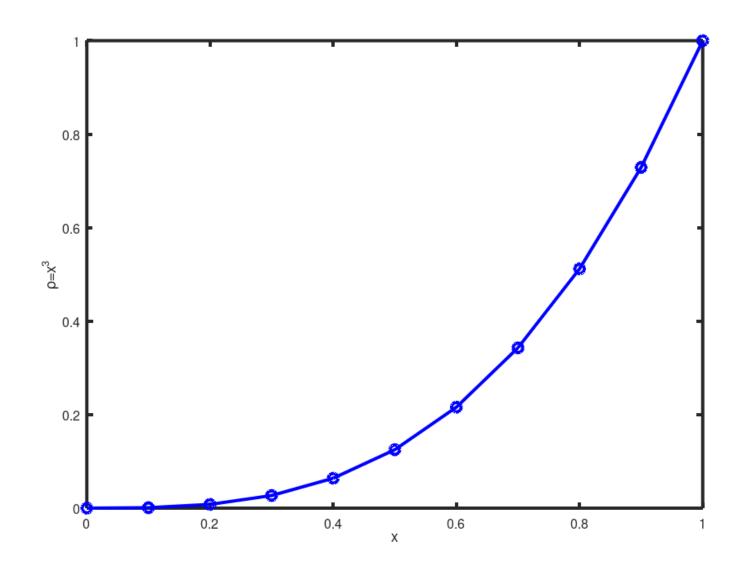
$$\left(\frac{d^{2}\rho}{dx^{2}}\right)_{i} = \frac{\rho(x_{i+1}) - 2\rho(x_{i}) + \rho(x_{i-1})}{\Delta x_{i}^{2}} + \frac{\Delta x_{i}^{4}}{12\Delta x_{i}^{2}} \left(\frac{d^{4}\rho}{dx^{4}}\right)_{i} + \cdots$$

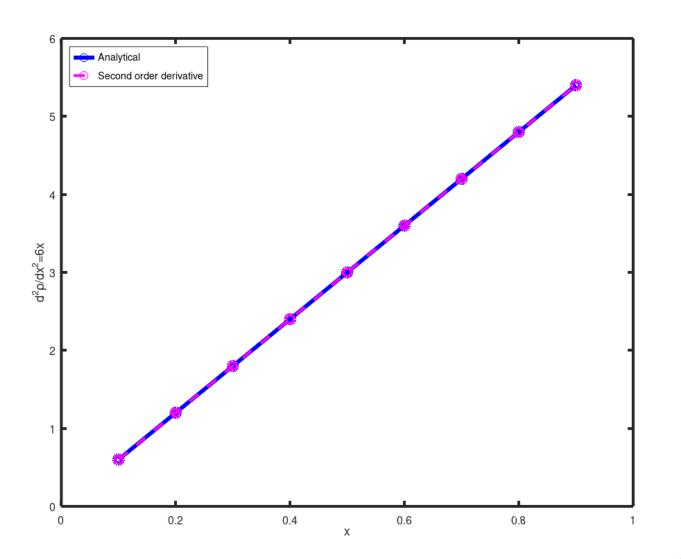
$$\left(\frac{d^{2}\rho}{dx^{2}}\right)_{i} = \frac{\rho(x_{i+1}) - 2\rho(x_{i}) + \rho(x_{i-1})}{\Delta x_{i}^{2}} + O(\Delta x_{i}^{2})$$



Approximating Second Order Derivative

$$\left(\frac{d^{2}\rho}{dx^{2}}\right)_{i} = \frac{\rho(x_{i+1}) - 2\rho(x_{i}) + \rho(x_{i-1})}{\Delta x_{i}^{2}} + O(\Delta x_{i}^{2})$$





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