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## Original Article

# Investigating the elastic modulus and hardness properties of a high entropy alloy coating using nanoindentation



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#### ABSTRACT

Using a focused ion beam TTX-NHT3 Nanoindentation tester with a load of 500 mN, we study the micromechanical properties of laser-deposited AlCoCrFeNiCu high entropy alloy coatings. The indentation tests conducted were used to examine the influence of laser power and scan speed on the elastic modulus and hardness of the alloy coatings using the Oliver & Pharr method. There were several indentation points impressed and used to extract the mechanical properties of the alloys, and the results of the alloy were compared with that of the A301 steel substrate. The results showed the mean value of the NanoHardness and Elastic modulus of the high entropy alloy were 2.769 GPa and 149 GPa, respectively. The Vickers hardness showed a 60% decline as the laser power increased from 1200 W to 1600 W. The hardness and the elastic modulus were proportional to each other, both increasing with a decrease in the indentation depth and laser power. The laser-deposited high entropy alloys were more resistant to plastic deformation and had improved mechanical properties than the steel substrate attributed to the solid-solution hardening and lattice distortion effect of the BCC phase structure and aluminium contents, respectively.

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number of elements in the HEAs system causes a lattice distortion

attributed to uneven atomic radius by distinct elements in the alloy

system [4]. The lattice distortion core effect that hinders the

dislocation movement results in excellent hardness and strength,

even at elevated temperatures. Diao et al. [5] reported that the

Vickers hardness values of HEAs range from 140 HV to 900 HV.

However, these values depend on the alloying composition and the

manufacturing route. According to Zhang et al. [6], HEAs are

variations in both techniques to the difference in thermal gradient and solidification rates [11]. Xiang et al. [12] reported that the laser

metal deposition technique (LMD) had better mechanical proper-

ties than the as-cast HEAs at cryogenic temperatures and that the

LMD allows the adjustments of the mechanical properties of the

#### 4.0/).

#### 1. Introduction

Materials used for aerospace applications must be lightweight, corrosion-resistant, having good thermal behaviour at elevated temperatures and mechanical features such as toughness, ductility, hardness, creep and strength. Materials used in the aerospace industry with these characteristic features are nickel superalloys, steels, titanium alloys, cobalt-based alloys and more recently high entropy alloys (HEAs) [1].

Yeh et al. [2] reported that high entropy alloys comprise principal elements from the periodic table where the compositional concentration ranges between 5% and 35%, and these elements must at least be five or more for the equilibrium phases in the alloy system to form solid solution phases as opposed to intermetallic compounds [3]. This high entropy effect triggered by an increased

fabricated via powder metallurgy, casting, mechanical alloying and additive manufacturing, to mention a few [7]. In the literature, most HEAs are manufactured through arc melting and casting technique, nonetheless, the mechanical properties of as-cast HEAs are generally lower than other manufacturing routes, particularly when compared to materials fabricated via laser additive manufacturing [8,9]. Joseph et al. [10] did a comparative study on HEAs fabricated via arc melting and laser deposition and the authors attributed the

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HEAs by changing the laser power attributed to the heat flux direction and rapid solidification of the technique. This suggests that the laser parameters have a significant influence on the mechanical properties of laser-deposited HEAs. Sun et al. [13] proposed that the optimization of the laser parameters results in improved mechanical properties and homogeneous microstructures, thus, the importance of studying the effect of these parameters on the mechanical properties of laser-deposited HEAs [14]. Several conventional methods have been used to test the mechanical properties of HEAs, namely; compression, tension and Charpy impact tests [15], however, the lack of in-situ measurements without damaging effects or the demand for standard or bulk sample sizes makes these methods unsuitable [16].

Nanoindentation is a non-destructive technique that provides in-situ indentation measurements on a small area but a wide range of geometries for several materials, including coatings. Thus, avoiding the effect of the substrate at penetration depths as low as 25 nm [17]. This technique uses a Berkovich three-sided pyramid indenter or a four-sided pyramid Vickers indenter. In this study, a Berkovich indenter is used as opposed to a Vickers indenter because the former provides a sharp pointed tip compared to the 0.5 µm slight offset of the Vickers indenter [18]. The Berkovich indenter also probes many points on the surface of the sample through a line, point or grid method [19]. For this reason, this indenter was selected for this study, nonetheless, both Vickers and Berkovich indenters give adequate results accepted by the ISO/FDIS 14577-1 standard [20]. The Nanoindentation technique has been used to examine the creep characteristics [21–23], assessing phase stabilities [24], investigating plastic deformation [25,26], measuring residual stresses [27,28], fracture toughness [29,30], hardness and elastic modulus of polymers, metals and alloys [31-34]. Table 1 shows earlier studies on the nanoindentation of high entropy alloys as having been reported in the literature on thin films via magnetic sputtering [35-39], HEAs via selective laser melting [40-45] or other manufacturing techniques and alloy composition [46]. To our best knowledge, reports on the Nanoindentation of LMD AlCoCrFeNiCu HEA coatings on a steel substrate are limited in the literature. Thus, this study investigates the micromechanical properties of LMD AlCoCrFeNiCu HEAs coating by utilizing indentation testing for aerospace applications.

#### 2. Material and methods

#### 2.1. Laser metal deposition

In this study, the nominal and actual composition of AlCoCrFe-NiCu high entropy alloy powder is shown in Table 2. The

 Table 2

 Chemical composition of AlCoCrFeNiCu HEA in atomic percentage.

Element	Al (at.%)	Co (at.%)	Cr (at.%)	Fe (at.%)	Ni (at.%)	Cu (at.%)
Nominal	16.6	16.6	16.6	16.6	16.6	16.6
Actual	42.95	11.09	10.24	13.52	10.36	11.84

AlCoCrFeNiCu powder blend used to produce the samples A-E was purchased from F.J Brodmann & CO., L.L.C each having purity of 99.9%. The as-received powder with particle sizes distribution which ranges from 45 to 106 um was deposited on a preheated A301 steel substrate of geometric size 10 cm  $\times$  5 cm at 400 °C via a laser engineering net shaping (LENS) system at the Council for Scientific and Industrial Research, Pretoria, South Africa and used to fabricate five samples. The LMD process was optimized using the following parameters presented in Table 3, thus, varying the laser power from 1200 to 1600 W and scan speed from 8 mm/s to 12 mm/s while other parameters such as the beam diameter 2 mm, powder feed rate 2 rpm, 50% overlap and 0.5 mm layer thickness on an A301 austenitic steel plate (7% nickel and 17% chromium) remains constant. The microstructure of the alloy was detected using an XPERT-PRO X-ray diffraction system (XRD) and a JEOL-JSM-6010/LA Plus Scanning electron microscope (SEM).

#### 2.2. NanoIndentation testing

After deposition, the samples were cut into smaller pieces using a cutting blade machine. The cross-sections of the samples were mounted to expose the surface of the samples for grinding and polishing. The grinding of the surface with grit sizes 80, 320 and 1200 and the polishing to a mirror-finish surface was achieved using Tripoli, intermediates and finishing rouge. The elastic modulus and microhardness were measured on the polished surface of the mounted samples using an Anton-Paar TTX-NHT<sup>3</sup> Nanoindentation tester. The indenter is equipped with a diamond Berkovich three-sided pyramid indenter via the Oliver-Pharr method at the Surface Engineering Research Laboratory (SERL), Pretoria, South Africa.

The equipment shown in Fig. 1 prints an impression in grid form on the surface of the samples through a diamond Berkovich

**Table 3**The laser processing parameters.

Sample	Α	В	С	D	Е
Laser power (W)	1200	1200	1400	1600	1600
Scan speed (mm/s)	8	12	12	10	12

**Table 1**Summary of Nanoindentation studies of HEAs.

HEAs	Structure	Property	Process	Substrate	Ref.
AlCoCrFeNiCu	FCC, BCC	Creep, hardness and modulus, fracture toughness and Vickers hardness	LENS	A301 steel	This Study
Al <sub>x</sub> CoCrFeNi	FCC, Disordered BCC and Ordered B2	Hardness and modulus	LENS	CoCrFeNi	[11]
Al <sub>0.5</sub> CoCrFeNi	FCC + BCC	Plastic deformation	Arc Melting	_	[47]
CoCrFeNiMn	FCC and BCC	Hardness and elastic modulus	Magnetron Sputtering	Silicon	[48]
AlCoCuCrFeNiTi	FCC, BCC, Ordered BCC and Tetragonal	Phases, hardness and elastic modulus	Mechanical Alloying	-	[49]
CoCrFeMnNi	FCC	Creep	SLM	_	[40]
FeMnCoCrSiCu	FCC and HCP	Elastoplastic deformation	Casting	_	[26]
AlCrCuFeNi <sub>2</sub>	FCC and BCC	Indentation deformation of the phases	Arc Melting	_	[50]
AlCoCrFeNi	FCC, BCC, Ordered BCC	Pile-up and Sink-in Characteristics	Arc Melting	_	[51]
FeCoNiCrMn and FeCoNiCrMnAl	FCC and FCC + BCC	Elevated temperature creep	Arc Melting	_	[52]



Fig. 1. The nanoindentation equipment illustrating the main components.

indenter with an indentation load of 500 mN. The loading and unloading time was placed at 20 s and the holding time between the loading and unloading stage for each test was at 5 s shown in Fig. 2.

#### 2.2.1. Nanoindentation theory

The ratio of the peak force  $F_{max}$  to the indentation area  $A_c$  is defined by the hardness and calculated from the equation [53]:

$$H = \frac{F_{\text{max}}}{A_c} \tag{1}$$

The area of indentation  $A_c$  is derived from Ref. [53] as:

$$A_c = f\left(h_c^2\right) \tag{2}$$

where the contact depth at the peak load is  $h_c$  and can be calculated as in equation [53]:

$$h_c = h_{\text{max}} - \varepsilon \frac{f_{\text{max}}}{S} \tag{3}$$

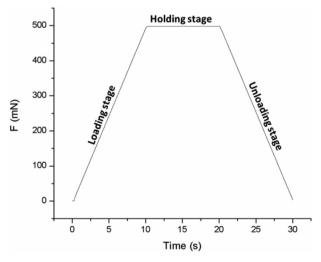


Fig. 2. The applied Load-Time Curve.

where the displacement at the maximum load is  $h_{max}$ , S is the contact stiffness and  $\varepsilon$  is a constant for the indenter geometry at 1.5 for Berkovich indenter [20].

The contact stiffness was calculated using equation (4):

$$S = \left(\frac{dF}{dh}\right)_m = \beta \frac{2}{\sqrt{\pi}} E_r \sqrt{A_c}$$
 (4)

where h and F are the penetration depth and applied load during the indentation tests,  $\beta = 1.034$ , which is the Triangular geometric factor of the diamond shape-dependent indent for a Berkovich indenter [54]. While the reduced modulus  $E_r$  is derived by the equation [55]:

$$\frac{1}{E_r} = \frac{1 - v^2}{E} + \frac{1 - v_i^2}{E_i} \tag{5}$$

where  $v_i$  and  $E_i$  are the elastic modulus and poisons ratio of the indenter and E and v are the elastic modulus and Poisson's ratio of the samples [56].

The formula by Evans and Wilshaw was used to calculate the fracture toughness ( $K_{IC}$ )as in Eqs. (6) and (7) [53]:

$$K_{IC} = 0.079 \left(\frac{P}{a^{3/2}}\right) \log\left(4.5 \cdot \frac{a}{c}\right) MPa \ m^{1/2}$$
 (6)

$$K_{IC} = \alpha \left(\frac{E}{H}\right)^{0.5} \left(\frac{P}{C^{1.5}}\right)$$

where a the half indentation is diagonal which is approximately 0.04 for Berkovich indenter [57]. P is the applied normal load, c is the crack length and half indentation diagonal.

#### 3. Results and discussion

#### 3.1. The load-displacement curves

Nanoindentation tests were employed to probe the time-dependent nanoscale mechanical properties of the AlCoCrFe-NiCu HEA at room temperature and the microstructural analysis of the alloy was performed using a JEOL-JSM-6010/LA plus scanning electron microscope (SEM) presented in Fig. 3. The

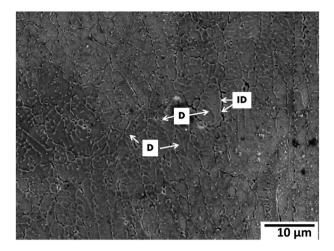


Fig. 3. SEM images showing the microstructure of AlCoCrFeNiCu high entropy alloy [68].

changes in laser parameters did not influence the structure of the high entropy alloy produced. Sample A-E exhibited a columnar dendrite BCC phase and a minor FCC phase observed in the interdendritic region made up of Cu. Copper segregated to this region because of its low binding energy and high mixing enthalpy with Ni, Fe, Cr and Co [58-60]. The chemical compositions of the alloy presented in Table 2 show that the actual content of Al is higher than the nominal contents, which acted as a stabilizer for the BCC phase. The structure of the phases was detected using an XPERT-PRO X-ray diffraction system and presented in Fig. 4. The diagram shows the laser-deposited HEA consists of an FCC and BCC phase with or without the changes in laser parameters, signifying that the laser processing parameters did not induce any new phase, however, the structure had dominant BCC phases with the highest intensity at 44.402° and an interplanar distance of 2.03859 Å attributed to Al, which is in agreement with Tang et al. [61]. The solid solution phases observed without traces of intermetallics can be attributed to the high entropy mixing effect. The AlCoCrFeNiCu compositional blend possesses the maximum near equiatomic ratio, solidifying to form solid solution phases [62]. The load-displacement (P-h)curves of the samples are shown in Fig. 5 with succession

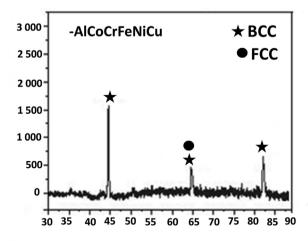


Fig. 4. XRD of the AlCoCrFeNiCu high entropy alloy.

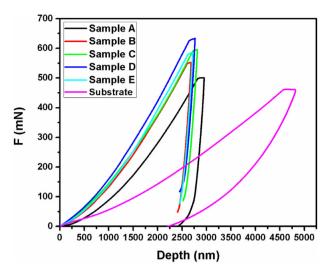


Fig. 5. The Nanoindentation force-depth curves of the samples.

loading and unloading stages. Interestingly, the tilt of the A301 steel unloading retract curve showed the baseplate was trying to recover from deformation, however, the indentation depth of the A301 steel substrate was greater than the AlCoCrFeNiCu coatings with a penetration depth of about 2500 nm. Thus, suggesting that the HEA coatings are more resistant to plastic deformation than the steel substrate, attributed to the BCC strengthening phase and the lattice distortion effect of the HEA [63]. Li et al. [64] reported that the plastic deformation for crystalline metals is attributed to the dislocation of slip. Notably, the BCC phase reduces the slipping plane and prevents interplanar dislocations, which stabilizes the microstructure of the alloy and resists plastic deformation [65]. Additionally, plastic deformation is attributed to the movement of the crystal lattice dislocations and the lattice distortion effect of the alloy attributed to the large atomic radius of the elements in the alloy system hinders dislocation movements. It was observed that sample B-E deformed less than sample A due to the low energy input of sample A at high-stress levels. The reduction in the indentation depth influenced an increase in the Nanohardness. Rodriguez et al. [66] also mentioned that the nanoindentation hardness is significantly influenced by the indentation depth, increasing with a decrease in the indentation depth [67].

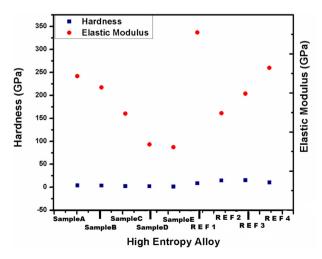
### 3.2. Modulus and nanohardness

The nominal composition of the laser-deposited high entropy alloys and their respective microstructures were the same: however, the alloys fabricated using several processing parameters resulted in each sample exhibiting different mechanical properties. The elastic modulus and hardness of each sample were calculated by analyzing the load-displacement curves based on the Oliver & Pharr method using a diamond Berkovich indenter with a centerline-to-face angle of 65.35°. The depths of the modulus and hardness were determined using the measurements from the contact stiffness, which is a function of the indentation depth. The results showed the young modulus and hardness were depthdependent. Therefore, in this depth-sensing indentation test, reduction of the influence of thermal drift on the indentation tests was achieved through machine calibration using fused silica [69]. The modulus responses and the nanohardness had comparable tendencies, decreasing with an increase in the indentation depth [70]. The modulus and hardness of the HEA also showed a 66% and

60% decrease respectively, as the laser parameters increased from 1200 W to 1600 W with no significant effect from the scanning speed. The data analysis showed that the mean value for the NanoHardness was 2.77 GPa and Elastic modulus 149 GPa, respectively, which were larger than that of the steel substrate (1.2 GPa and 140 GPa, respectively). According to Sun et al. [71], the nanoindentation point arrangement is divided into grid, line and points indentation imprints. The grid nanoindentation method used in this study proved to be effective in conducting the desired Nanoindentation data statistics and getting the mechanical properties of laser-deposited high entropy alloys [72]. Fig. 6 shows the comparison of the mean values of the modulus and hardness of this study with results from the literature. The graph shows that the hardness of sample A to E are in the same range compared with other high entropy alloys, however, the literature [26] showed slightly higher modulus attributed to the variation in the manufacturing process.

#### 3.3. Vickers hardness

The average Vickers hardness and the corresponding modulus as a function of the different parameters of the samples and compared with the substrate are plotted in Fig. 7. The Vickers hardness and the nanohardness were proportional to each other for the laser-deposited HEA, the Vickers hardness had a 60% decrease with an increase in laser power from 1200 W to 1600 W. Sun et al. [74] investigated the relationship among Young modulus, indentation hardness and tensile strength of pharmaceutical materials and the authors confirm that the Young modulus is proportional to the hardness for all materials. Nonetheless, the steel substrate showed quite the opposite when comparing the hardness and modulus. The modulus of the substrate showed higher values than the Vickers hardness of the substrate. Generally, the Vickers hardness of the high entropy alloys was 69% higher than the substrate and this is attributed to the BCC phase solid solution strengthening effect of the HEAs. The BCC phase attributed to the high Al content improved the hardness properties [65], in agreement with our previous studies [75]. Thus, the Al element also has one of the largest atomic radiuses, and the combined increment of the atomic radius in the HEA system resulted in lattice distortion, which also influenced the increment in hardness compared to the steel substrate.



**Fig. 6.** The comparison of the modulus and nanohardness of present high entropy alloys with the literature [26,49,64,73].

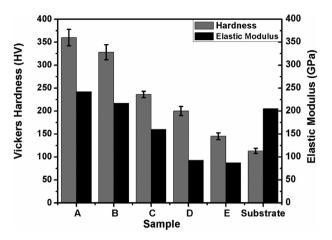


Fig. 7. The Vickers hardness and modulus.

#### 3.4. Nanoindentation creep deformation and fracture toughness

The nanoindentation creep time-dependent deformation curves are presented in Fig. 8. As the HEA was stressed at a constant load at room temperature, the number of mobile dislocations reduced and due to the interactions with other dislocations during creep with time, the dislocations were rendered immobile leading to the strain hardening of the alloy since the stress field around the dislocation clusters prevented any dislocation glide. Deformation was further induced when the stress applied increased thus letting the trapped immobile dislocation overcome their restraints and start moving again. The results showed the creep deformation rate of the laserdeposited HEAs was lower than the steel substrate, increasing with an increase in laser power. Moreso, an increase in hardness increased the alloy's capacity to relieve stresses under the applied load. However, the time taken for deformation to occur was relatively close for all samples between 30 and 35 s. Hence, suggesting that a holding time below 30 s suffices to avoid the effect of creep deformation on the unloading curves. Chudoba et al. [76] also proposed a minimum holding time of less than 30 s for highprecision measurements. The deformation was low as the scan speed increases from 8 to 12 mm/s. According to Yu et al. [77], a high scan speed decreases the energy density and reduces the creep of laser-deposited materials, which is due to the increased cooling rate that minimizes the residual stresses.

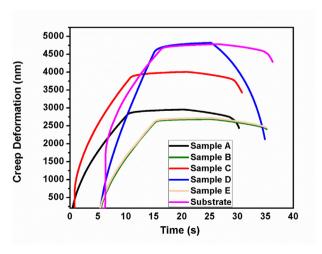


Fig. 8. The Nanoindentation creep deformation-time curve.

The fracture toughness is quantified by the nano-mechanic approach of the samples and characterized as  $K_{IC}$ , by using the ASTM standard E399. As recommended by Li et al. [78], the centerline-to-face angle was about 35.3%. This helped reduce the sample's cracking threshold value. The laser-deposited HEA had greater fracture toughness than the substrate and the fracture toughness decreased with an increase in scan speed at laser power 1200 W and 1600 W, respectively. Sample A and D had fracture toughness of 17.5 MPa m<sup>1/2</sup> and 17.29 MPa m<sup>1/2</sup>, while sample B, C and D had fracture toughness of 14.8 MPa m<sup>1/2</sup>, 12.5 MPa m<sup>1/2</sup> and 15.4 MPa m<sup>1/2</sup>, respectively. Nonetheless, both processing parameters significantly influenced the fracture toughness of the laser-deposited high entropy alloys. All samples were observed not to fail in a brittle manner as the mean fracture toughness values of all samples were more than 10 MPa.

#### 4. Conclusion

The AlCoCrFeNiCu High entropy alloy was fabricated via laser additive manufacturing on an A301 steel substrate, and the micromechanical properties of the alloy and its substrate using a nanoindentation tester were investigated. The alloy exhibited both FCC and BCC phases; however, the BCC phase structure was prominent. This phase was responsible for its strength and resistance to plastic deformation compared with the substrate. A reduction in the indentation depth and laser power estimated an increase in the Nanohardness values and modulus and vice versa. The mean value for the NanoHardness was 2.769 GPa, Elastic modulus 149 GPa and fracture toughness was 15.5 MPa  $m^{1/2}$  under a constant load of 500 mN. The laser processing power had the most influence on the nanomechanical properties of the laser-deposited high entropy alloy and the alloy had improved mechanical properties than the substrate. This suggests that AlCoCrFeNiCu High entropy alloy can be used as suitable coatings on A301 steel for protection against mechanical failures in aerospace structural applications.

#### **Conflicts of interest**

The authors declare that there is no conflicts of interest.

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