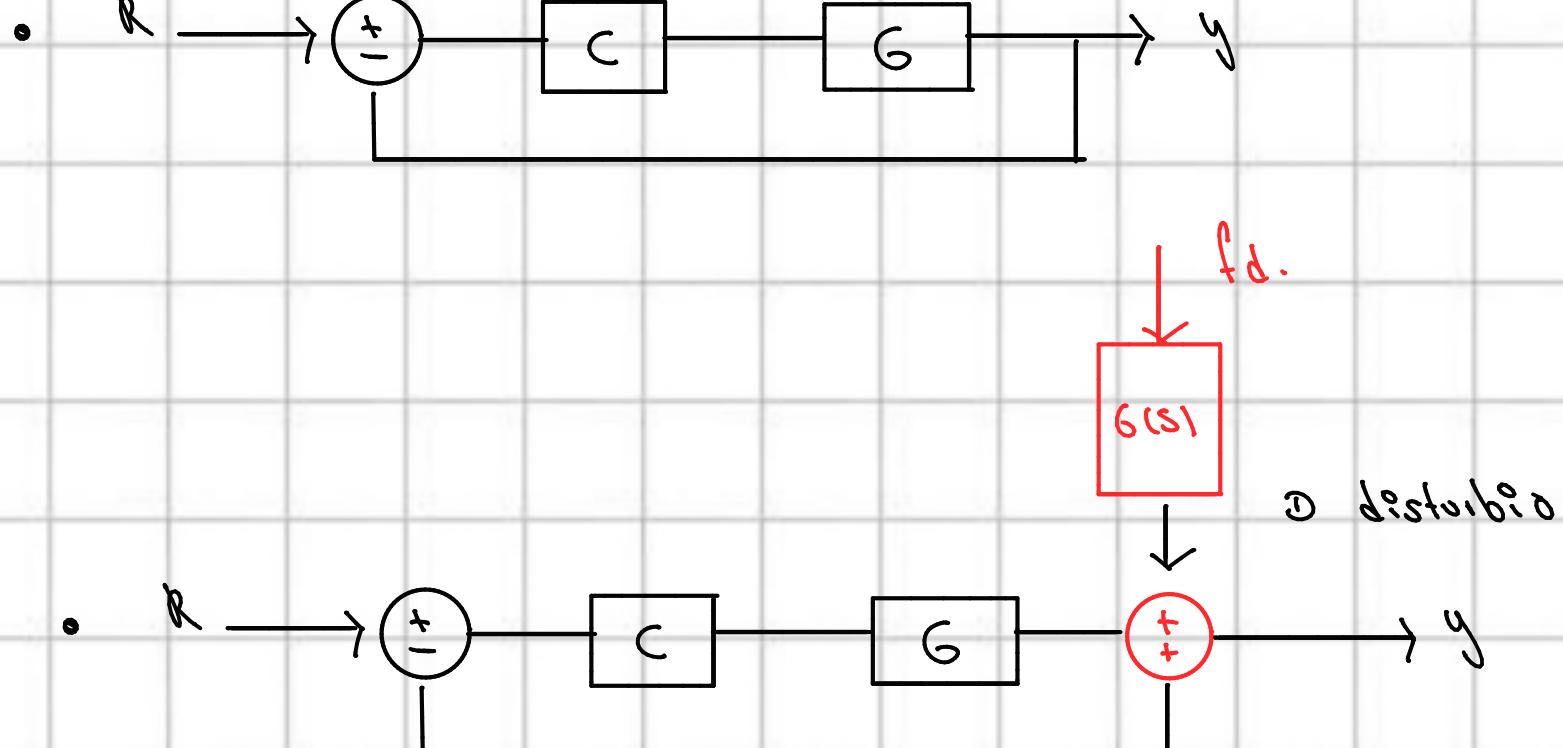


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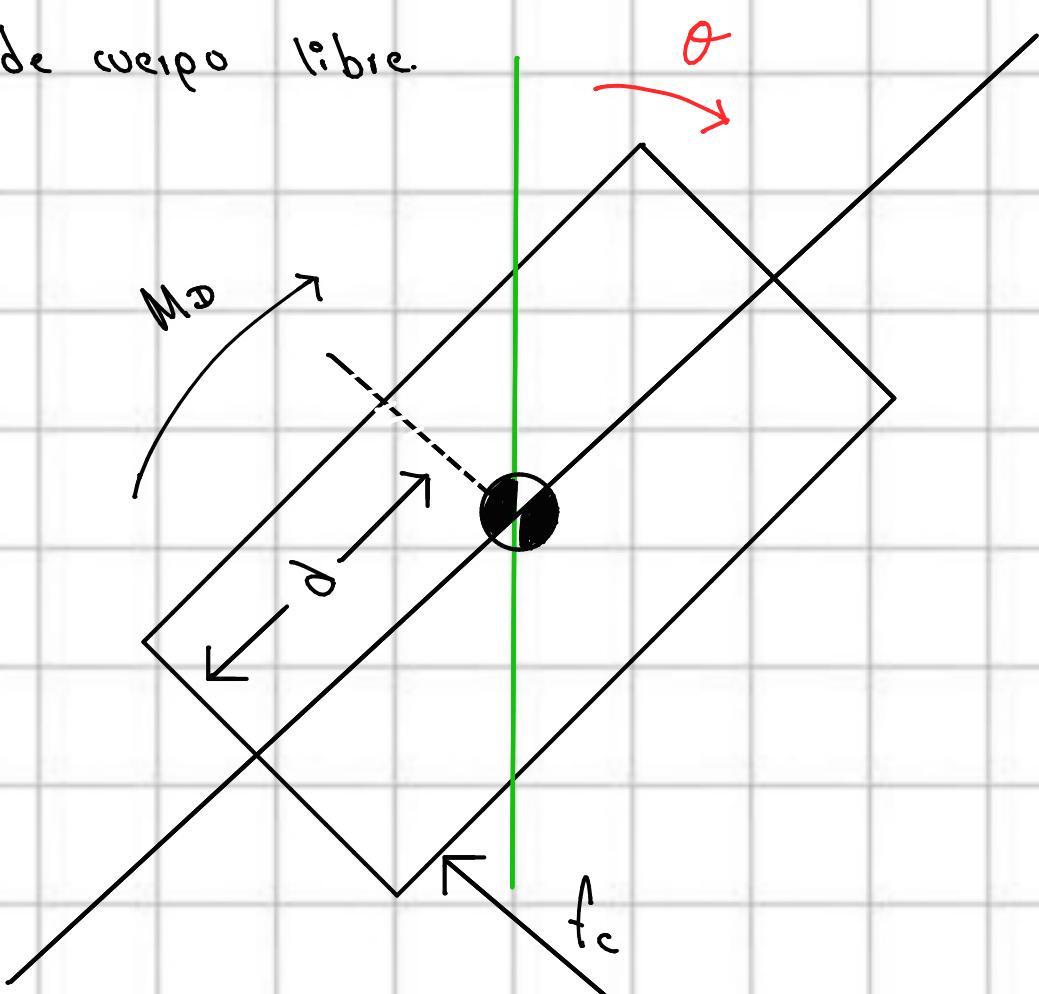
• Libro feedback control of dynamics Systems Franklin 8º Edición

• $M = I_2$ momento de inercia



• TAREA #3 Fabian Leonaldo Camargo Beinate.

• Diagrama de cuerpo libre



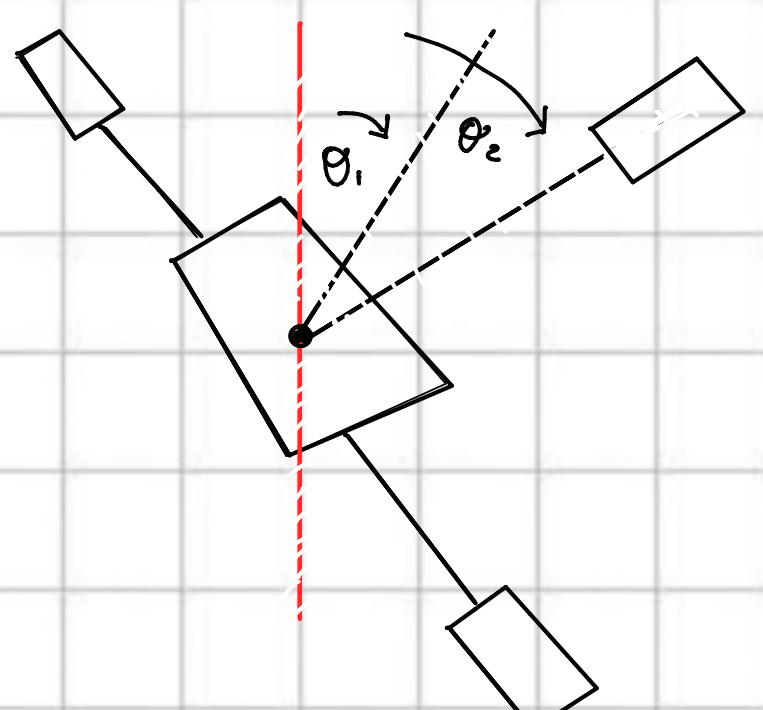
$$\underline{F_c d + M_0} = I \ddot{\theta}$$

U

$$\frac{U(s)}{U(s)} = \frac{1}{I s^2}$$

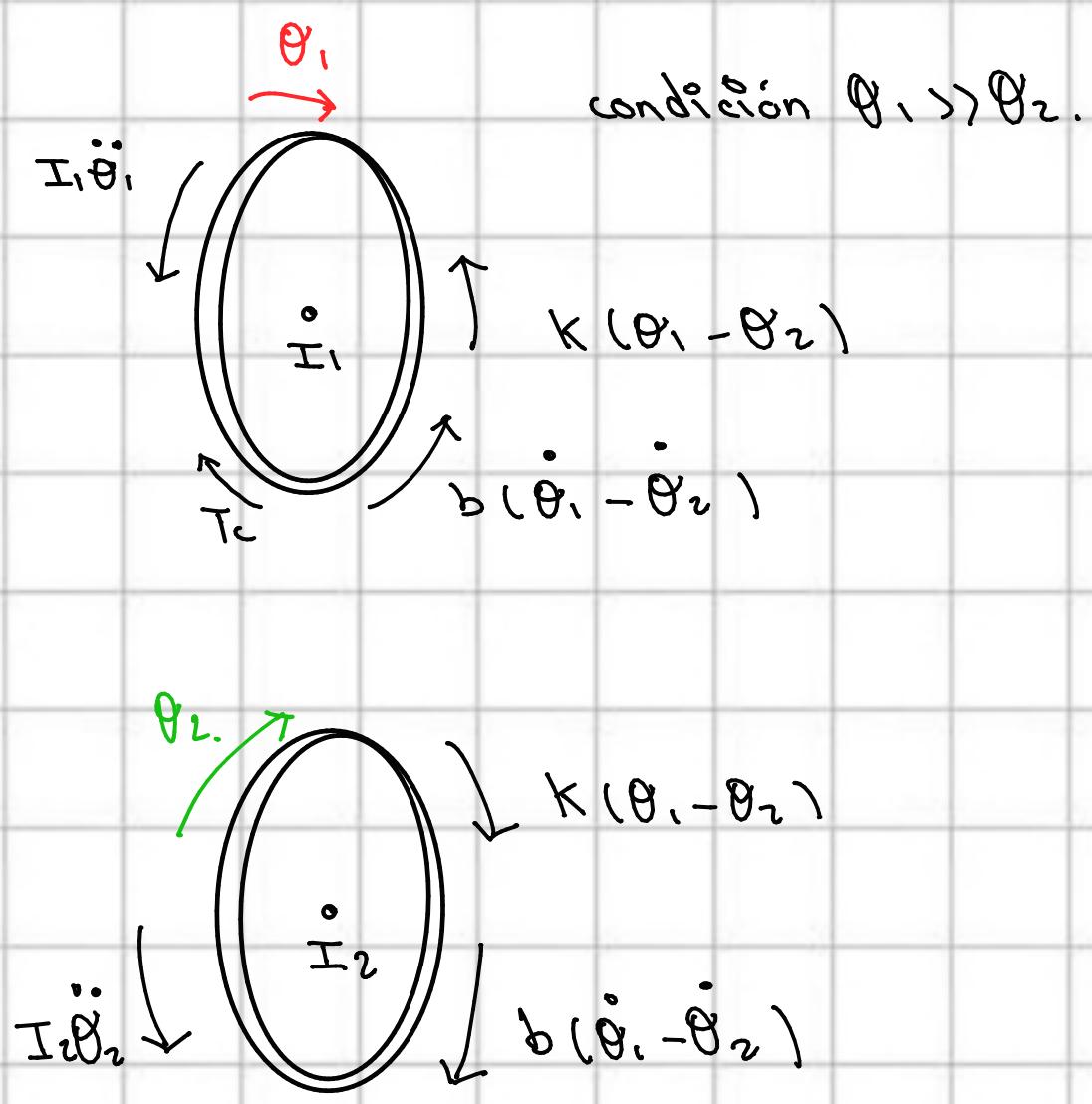
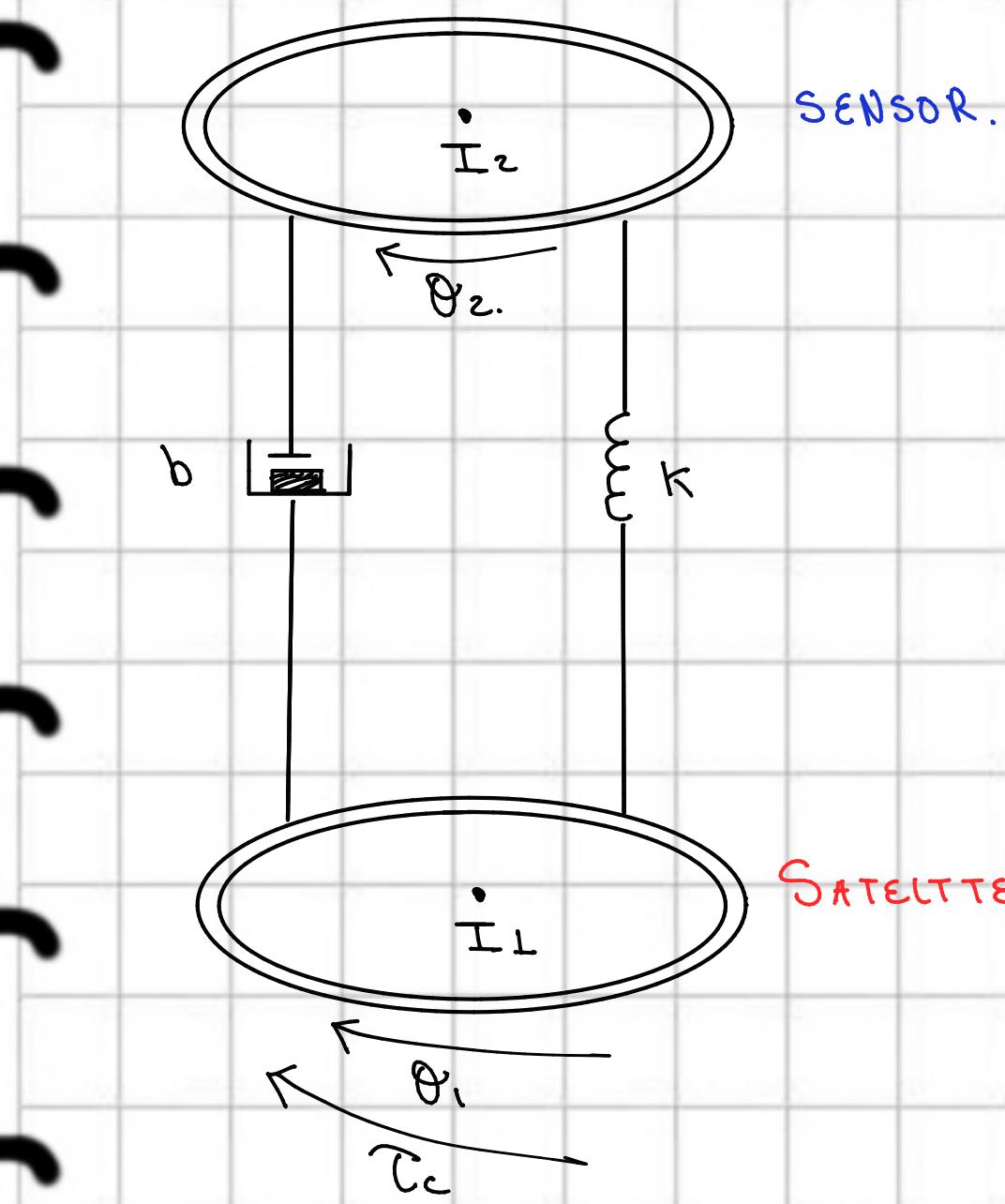
$$U(s) = I s^2 \theta(s)$$

Satélite.



θ_2 : con respecto al Sensor
 θ_1 : con respecto a la masa inercial.

Modelado del Sistema



• ecuaciones.

$$T_c - K(\theta_1 - \theta_2) - b(\dot{\theta}_1 - \dot{\theta}_2) = I_1 \ddot{\theta}_1$$

$$b(\dot{\theta}_1 - \dot{\theta}_2) + K(\theta_1 - \theta_2) = I_2 \ddot{\theta}_2$$

• Ecuaciones.

$$1. T = I_1 \ddot{\theta}_1 + K\theta_1 + b\dot{\theta}_1 - K\theta_2 - b\dot{\theta}_2 \rightarrow q_1 = \theta_1 \quad q_2 = \dot{\theta}_1$$

$$2. I_2 \ddot{\theta}_2 + b\dot{\theta}_2 + K\theta_2 - b\dot{\theta}_1 - K\theta_1 = 0 \rightarrow q_3 = \theta_2 \quad q_4 = \dot{\theta}_2$$

• Derivar variables de estado.

$$\dot{q}_1 = \dot{\theta}_1 = q_2$$

$$\dot{q}_2 = \ddot{\theta}_1$$

$$\dot{q}_3 = \dot{\theta}_2 = q_4 \quad y_1 = \theta_1 = q_1$$

$$\dot{q}_4 = \ddot{\theta}_2 \quad y_2 = \theta_2 = q_3$$

• Despejando $\ddot{\theta}_1$ y $\ddot{\theta}_2$.

$$1. \ddot{\theta}_1 = \frac{1}{I_1} (T - K\theta_1 - b\dot{\theta}_1 + K\theta_2 + b\dot{\theta}_2)$$

$$2. \ddot{\theta}_2 = \frac{1}{I_2} (b\dot{\theta}_1 + K\theta_1 - K\theta_2 - b\dot{\theta}_2)$$

• Reemplazando

$$1. \dot{q}_2 = \frac{1}{I_1} (T - Kq_1 - bq_2 + Kq_3 + bq_4)$$

$$2. \dot{q}_4 = \frac{1}{I_2} (bq_2 + Kq_1 - Kq_3 - bq_4)$$

- Expresiones en el espacio de estados.

$$\dot{q}_1 = \dot{\theta}_1 = q_2.$$

$$\dot{q}_2 = \frac{1}{I_1} (\tau - Kq_1 - bq_2 + Kq_3 + bq_4)$$

$$\dot{q}_3 = \dot{\theta}_2 = q_4.$$

$$\dot{q}_4 = \frac{1}{I_2} (bq_2 + Kq_1 - Kq_3 - bq_4)$$

$$\dot{q}_1 = \dot{\theta} = q_2$$

$\dot{q}_2 = \ddot{\theta}$ despejar.

• Reemplazando

$$\dot{q}_2 = \frac{\tau_c}{ml^2} - \frac{g}{l} q_1$$

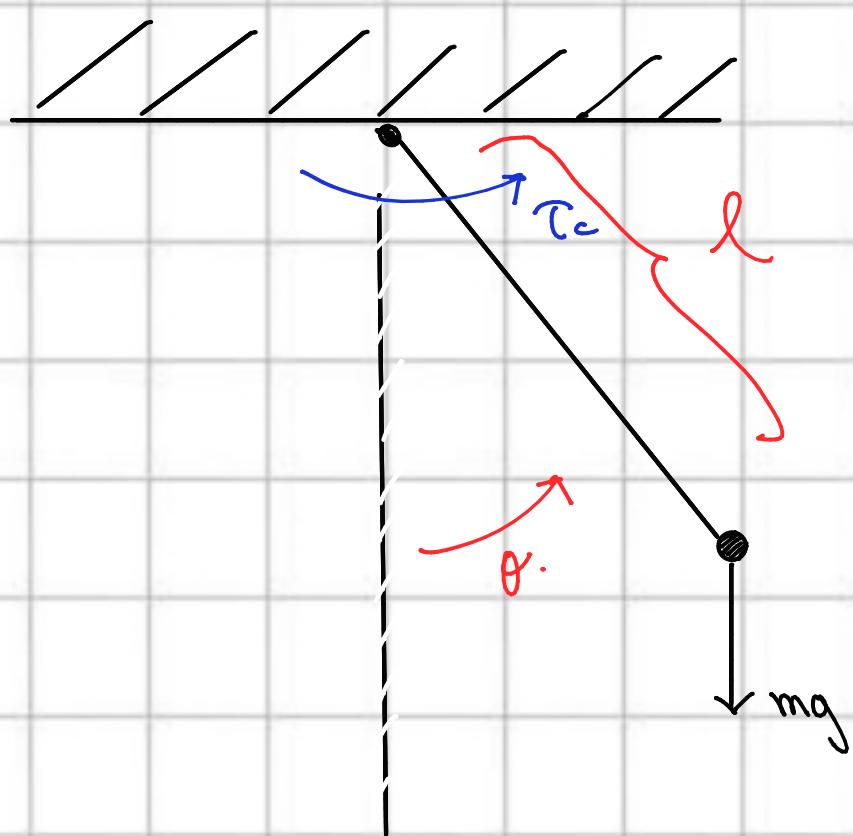
- formato matricial

$$\begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \\ \dot{q}_3 \\ \dot{q}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -K/I_1 & -b/I_1 & K/I_1 & b/I_1 \\ 0 & 0 & 0 & 1 \\ K/I_2 & b/I_2 & -K/I_2 & -b/I_2 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/I_1 \\ 0 \\ 0 \end{bmatrix} \tau$$

• Salida

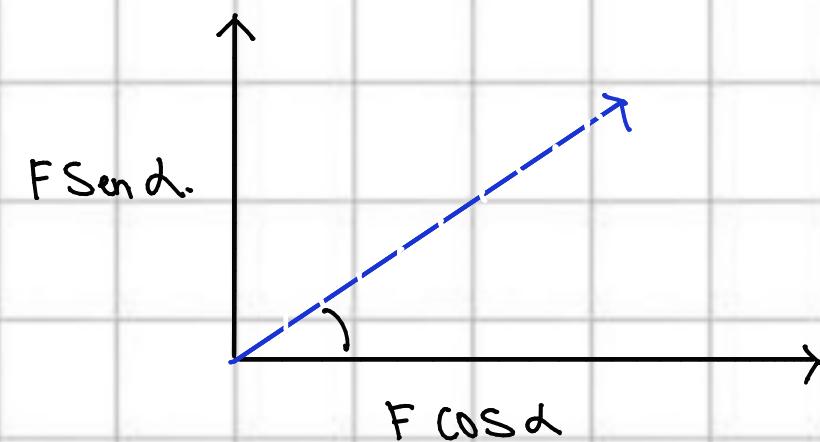
$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{bmatrix}$$

- Punto 2. Pendulo.



momento de inercia.

$$I = ml^2.$$



$$\ddot{\theta} = \frac{\tau_c - mgl \operatorname{Sen}\theta}{I}$$

$$\ddot{\theta} = \frac{\tau_c}{I} - \frac{mgl}{I} \operatorname{Sen}\theta.$$

$$\ddot{\theta} = \frac{\tau_c}{ml^2} - \frac{g}{l} \operatorname{Sen}\theta.$$

• Linearización con angulos pequeños $\operatorname{Sen}\theta \approx \theta$ y $\cos\theta \approx 1$

$$\therefore \ddot{\theta} = \frac{\tau_c}{ml^2} - \frac{g}{l} \theta$$

- Variables de estado.

$$\cdot q_1 = \theta$$

$$\cdot q_2 = \dot{\theta}$$

- Formato matricial Espacio de estados

$$\begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\frac{g}{l} & 0 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{\tau_c}{ml^2} \end{bmatrix} \tau_c$$

• Salida.

$$y = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix}$$