

# Identification Model Motor DC

(Ini adalah contoh template PPT dengan LaTex)

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# Outline

1 Introduction

2 Diagram Blok Sistem

3 Kendali PID

# Pendahuluan

## Pengertian Motor DC

Motor DC atau dalam bahasa Indonesia disebut motor arus searah adalah ....

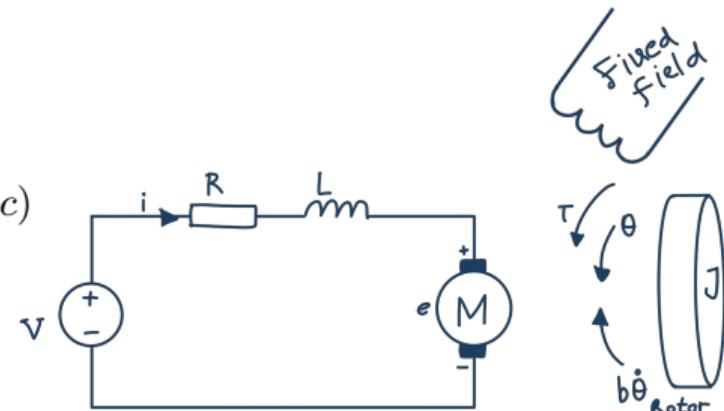
Ilustrasi Motor DC:

# Struktur Fisik

Parameter:

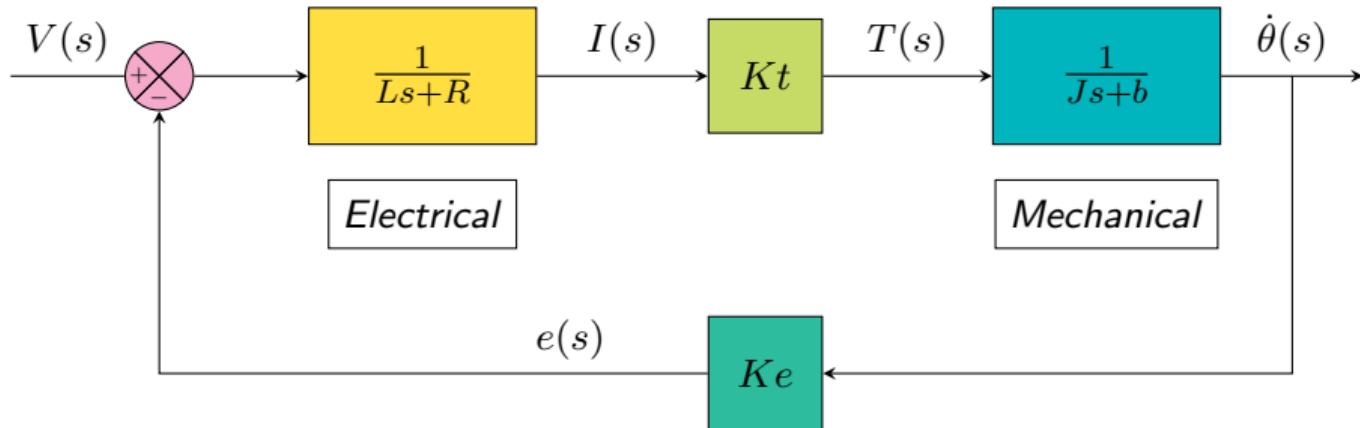
- $J$  : Momen inersia rotor ( $Kg.m^2$ )
- $b$  : Koefisien gaya gesek viskos ( $N.m.s$ )
- $Ke$  : Koefisien gaya elektromotif ( $V/rad/sec$ )
- $Kt$  : Koefisien torsi motor ( $N.m/Amp$ )
- $R$  : Resistansi kumparan ( $Ohm$ )
- $L$  : Induktansi kumparan ( $H$ )

Skematik Motor DC:



# Diagram Blok Plant Motor DC

Struktur motor DC:



# Fungsi Alih

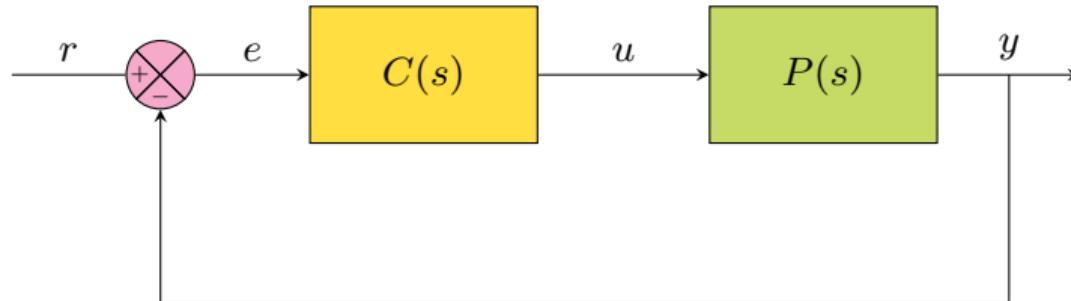
Diagram blok *plant* motor DC menghasilkan persamaan fungsi alih berikut:

$$\frac{\dot{\theta}(s)}{V(s)} = \frac{Kt}{(Js + b)(Ls + R) + KtKe} \quad \left[ \frac{\text{rad/sec}}{V} \right] \quad (1)$$

Persamaan di atas merupakan fungsi alih kecepatan motor DC. Dengan mengintegralkan fungsi alih tersebut, maka diperoleh fungsi alih untuk posisi motor DC:

$$\frac{\theta(s)}{V(s)} = \frac{Kt}{s((Js + b)(Ls + R) + KtKe)} \quad \left[ \frac{\text{rad}}{V} \right] \quad (2)$$

# Diagram Blok Kendali



## Keterangan:

$C(s)$  : Controller

$P(s)$  : Plant

$r(s)$  : Output yang diinginkan

$e(s)$  : Nilai error

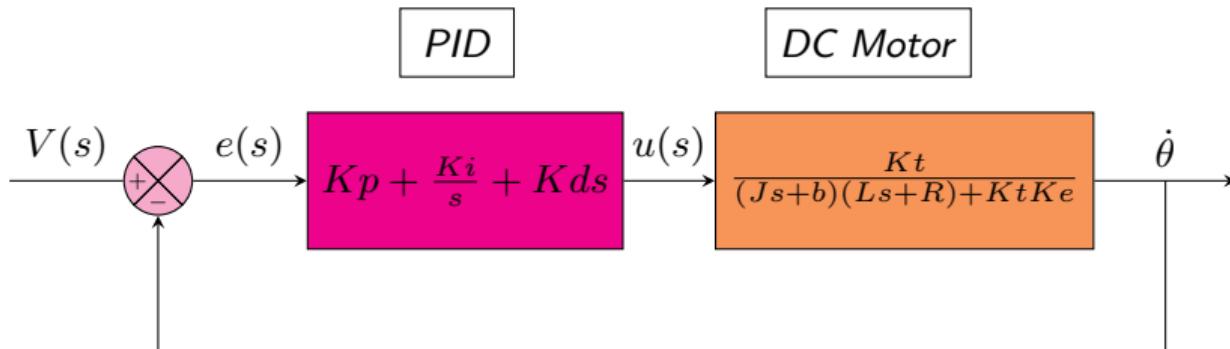
$u(s)$  : Sinyal kendali

$y(s)$  : Output sesungguhnya

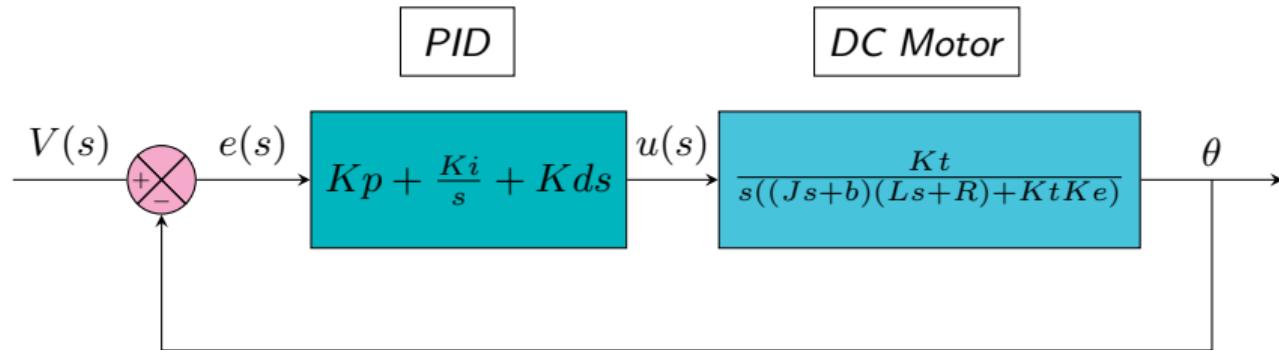
# Kendali PID

Contoh/ilustrasi perbandingan sistem dengan dan tanpa Kendali PID:

# Diagram Blok Kendali PID Motor DC: Kecepatan



## Diagram Blok Kendali PID Motor DC: Posisi



# Uji Perbandingan Sistem *Open-loop* dengan *Closed-loop* Motor DC

Program *Open-loop*:

```
J = 3.2284E-6;
b = 3.5077E-6;
Kt = 0.0274;
Ke = 0.0274;
R = 4;
L = 2.75E-6;;
s = tf('s');
P_motor = Kt/((J*s+b)*(L*s+R)+Kt*Ke);
rP_motor = 0.1/(0.5*s+1)
ltitiview('step', P_motor, 0:0.1:5);
```

Program *Closed-loop*:

```
J = 3.2284E-6;
b = 3.5077E-6;
Kt = 0.0274;
Ke = 0.0274;
R = 4;
L = 2.75E-6;
s = tf('s');
P_motor = Kt/((J*s+b)*(L*s+R)+Kt*Ke);
t = 0:0.001:0.2;
sys_cl = feedback(P_motor,1)
step(sys_cl,t)
```

# Kode Program Pada Matlab

```
% --- Outputs from this function are returned to the command line.
function varargout = LEDlagi2_OutputFcn(hObject, eventdata, handles)
varargout{1} = handles.output;
clear all;
global a;
a = arduino;

% --- Executes on button press in pushbutton1.
function pushbutton1_Callback(hObject, eventdata, handles)
global a;
writeDigitalPin(a, 'D9',1);

% --- Executes on button press in pushbutton2.
function pushbutton2_Callback(hObject, eventdata, handles)
global a;
writeDigitalPin(a, 'D9',0);
```

# Kendali PID: Posisi

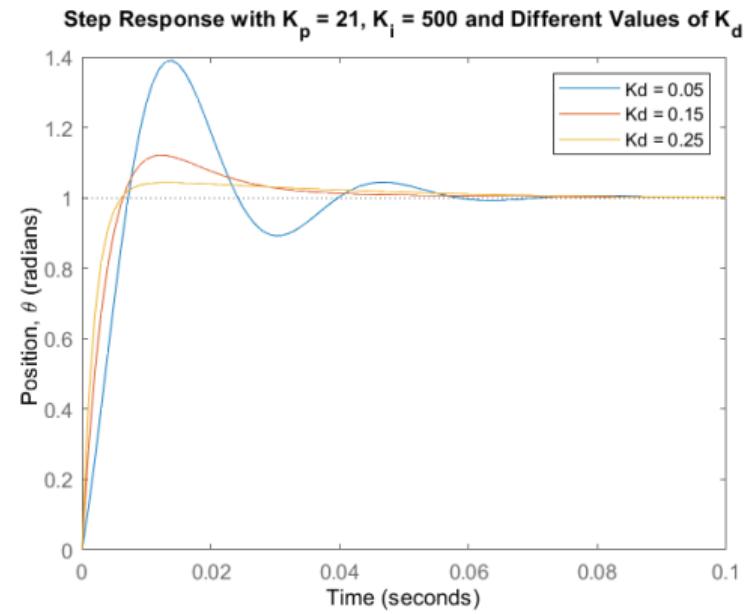
Program:

```
J = 3.2284E-6;
b = 3.5077E-6;
K = 0.0274;
R = 4;
L = 2.75E-6;
s = tf('s');
P_motor = K/(s*((J*s+b)*(L*s+R)+K^2));
Kp = 21;
Ki = 500;
Kd = 0.05;

for i = 1:3
    C(:,:,i) = pid(Kp,Ki,Kd);
    Kd = Kd + 0.1;
end

sys_cl = feedback(C*P_motor,1);
t = 0:0.001:0.1;
step(sys_cl(:,:,1), sys_cl(:,:,2), sys_cl(:,:,3), t)
ylabel('Position, \theta (radians)')
title('Step Response with K_p = 21, K_i = 500 and
        Different Values of K_d')
legend('Kd = 0.05', 'Kd = 0.15', 'Kd = 0.25')
```

Hasil:



# Embedded Video



# Embedded Video



## References

-  Fahmizal, H. A. Nugroho, A. I. Cahyadi, I. Ardiyanto *et al.*, "Attitude control of uav bicopter using adaptive lqg," *Results in Control and Optimization*, vol. 17, p. 100484, 2024.
-  ——, "Attitude control and low cost design of uav bicopter," *arXiv preprint arXiv:2309.08209*, 2023.
-  ——, "Trajectory tracking control of uav bicopter using linear quadratic gaussian," *arXiv preprint arXiv:2309.08226*, 2023.

QUESTIONS



ANSWERS

*thank  
you*



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