

SH2702  
Nuclear Reactor Technology

Project work Task 4

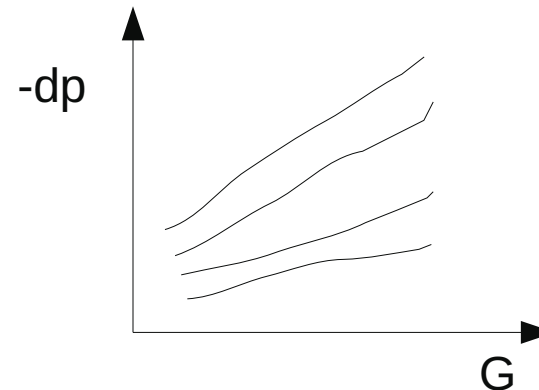
# Project work

Topic numbers	Topics
1	Design, operation and safety features of NuSCALE
2	Design, operation and safety features of ABWR
3	Design, operation and safety features of ESBWR
4	Design, operation and safety features of EPR
5	Design, operation and safety features of AP1000

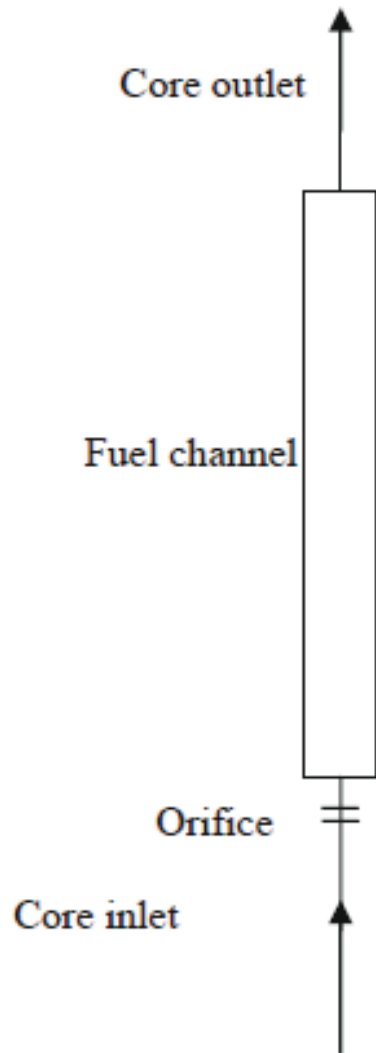
- Task 1 – General design specification of the nuclear power plant with selected reactor type
- Task 2 – Operational principles of the power plant
- Task 3 – Safety features of the power plant
- Task 4 – Calculation of selected core parameters
- Task 5 – Calculation of CHF margins in a hot channel
- Task 6 – Calculation of the maximum cladding and fuel pellet Temperature

## Task 4

- 1. Data collection
  - Tables are recommended
- 2. core-averaged thermal-hydraulic calculations
  - Axial enthalpy/temperature distribution
  - Axial void fraction distribution
    - BWRs, from subcooled to saturated
  - Axial pressure distribution
    - Inlet orifices pressure loss, BWRs (50%), PWRs (25%)
  - Flow characteristic of the core  $(-dp)=f(G)$ 
    - 0%, 50%, 100%, 150% power
    - 1% to 150% flow



## Task 4

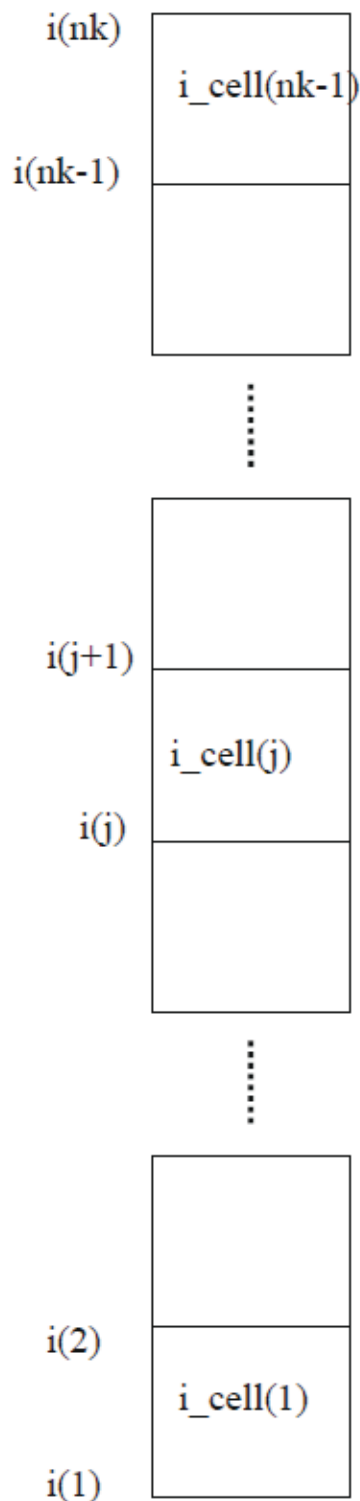


- Inlet orifices pressure loss
  - BWRs (50% at nominal operating conditions)
  - PWRs (25% at nominal operating conditions)

$$\Delta p = p_{out} - p_{in} = \Delta p_{FuelChannel} + \Delta p_{Orifice}$$

$$|\Delta p_{Orifice}| = \xi_{Orifice} \frac{\rho U^2}{2} = \xi_{Orifice} \frac{G^2}{2\rho}$$

## Task 4 Nodalization and numerical solution



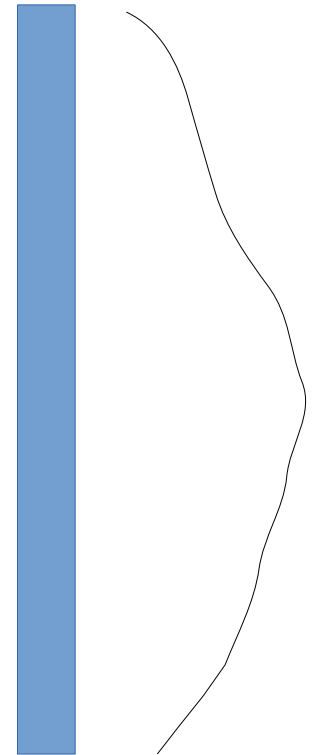
- for  $j = 2$  to  $nk$ 
  - $i(j) = i(j-1) + q\_cell(j-1) / W$  (energy balance)
- end for
- while p not converged
  - $p(1) = p_{in} + dp_{InletOrifice}$
  - for  $j = 2$  to  $nk$ 
    - $x_e(j), x_a(j), \alpha(j)$  (void fraction model)
    - $dpf\_cell(j-1), dp_g\_cell(j-1), dp_a\_cell(j-1), dp_l\_cell(j-1)$
    - $dp\_cell(j-1)$  (pressure drop calculation)
    - $p(j) = p(j-1) + dp\_cell(j-1)$
  - end for
- end while p
- $T(j)$ 
  - $f(p(j), i(j))$  for subcooled water
  - $T_{sat}(j)$  for saturated water
- Inlet orifices pressure loss coefficient (designed for nominal condition)
- Flow characteristic of the core  $(-dp)=f(G)$

## Power distribution

- If we assume the typical distribution for cylindrical core

$$q''(r, z) = q_0'' J_0 \left( \frac{2.405r}{\tilde{R}} \right) \cos \left( \frac{\pi z}{\tilde{H}} \right)$$

- Find peaking factor  $f_R, f_z$
- Example 1
  - $Q_{tot}$ , total reactor power from literature
  - $V_{tot}$ :  $A_{fa} * N_{fa} * H$ , or  $\pi * R^2 * H$  (power is distributed even in coolant)
  - $q'''_{ave} = Q_{tot} / V_{tot}$
  - $q'''_0 = q'''_{ave} * f_R * f_z$
  - For average channel,  $q'''(z) = q'''_0 / f_R * \cos(\pi * z / H_e)$
  - For hot channel,  $q'''(z) = q'''_0 * \cos(\pi * z / H_e)$
  - For each cell in the channel,  $q'''_c = q'''(z_{CellCenter})$
  - $q_{cell} = q'''_c * V_{cell(RodCoolant)} = q'''_c * A_{rodSurface}$
  - Check total power:  $SUM(q_{cell}) * N_{rod} * N_{fa} = Q_{tot}$
  - Otherwise  $q_{cellNew} = q_{cell} * Q_{tot} / (SUM(q_{cell}) * N_{rod} * N_{fa})$

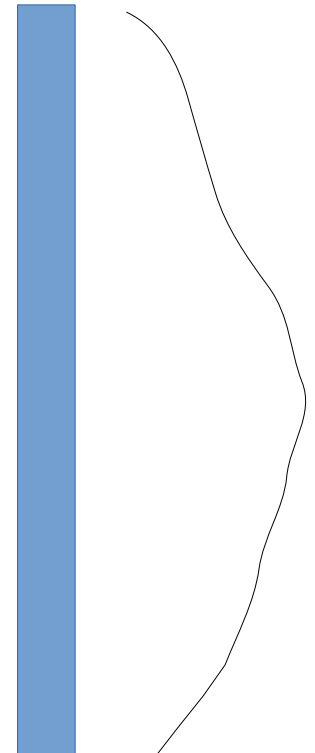


## Power distribution

- If we assume the typical distribution for cylindrical core

$$q''(r, z) = q_0'' J_0 \left( \frac{2.405r}{\tilde{R}} \right) \cos \left( \frac{\pi z}{\tilde{H}} \right)$$

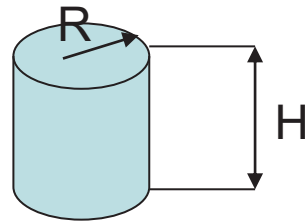
- Find peaking factor  $f_R$ ,  $f_z$
- Example 2
  - $Q_{tot}$ , total reactor power from literature
  - $q_{rodAve} = Q_{tot} / (N_{fa} * N_{FuelRod})$ , (assign total power to each fuel rod)
  - $q_{CellAve} = q_{rodAve} / N_{cell}$
  - For average channel,  $q_{cell}(z) = q_{CellAve} * f_z * \cos(\pi * z / H_e)$
  - For hot channel,  $q_{cell}(z) = q_{CellAve} * f_z * f_R * \cos(\pi * z / H_e)$
  - For each cell in the channel,  $q_{cell} = q_{cell}(z_{CellCenter})$
  - $q_{cell} = q''_c * A_{rodSurface}$
  - Check total power:  $SUM(q_{cell}) = q_{rodAve}$ ,
  - Otherwise  $q_{cellNew} = q_{cell} * q_{rodAve} / SUM(q_{cell})$



# Thermal Power Distribution in Fission Reactors

• Distribution of thermal power density in nuclear reactors depends on the shape of the reactor:

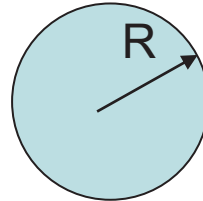
– Finite cylindrical with radius  $R$  and height  $H$ :



$$q'''(r, z) = q_0''' J_0\left(\frac{2.405r}{\tilde{R}}\right) \cos\left(\frac{\pi z}{\tilde{H}}\right)$$

$J_0(x)$  – Bessel function of first kind and zero order. See Compendium, Appendix B

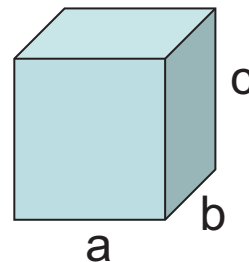
– Sphere with radius  $R$ :



$$q'''(r) = q_0''' \left(\frac{\tilde{R}}{\pi}\right) \frac{\sin \frac{\pi r}{\tilde{R}}}{r}$$

**Note:** dimensions with tilde are so-called extrapolated dimensions to avoid zero flux at reactor boundary

– Rectangular parallelepiped with sides  $a$ ,  $b$ ,  $c$ :



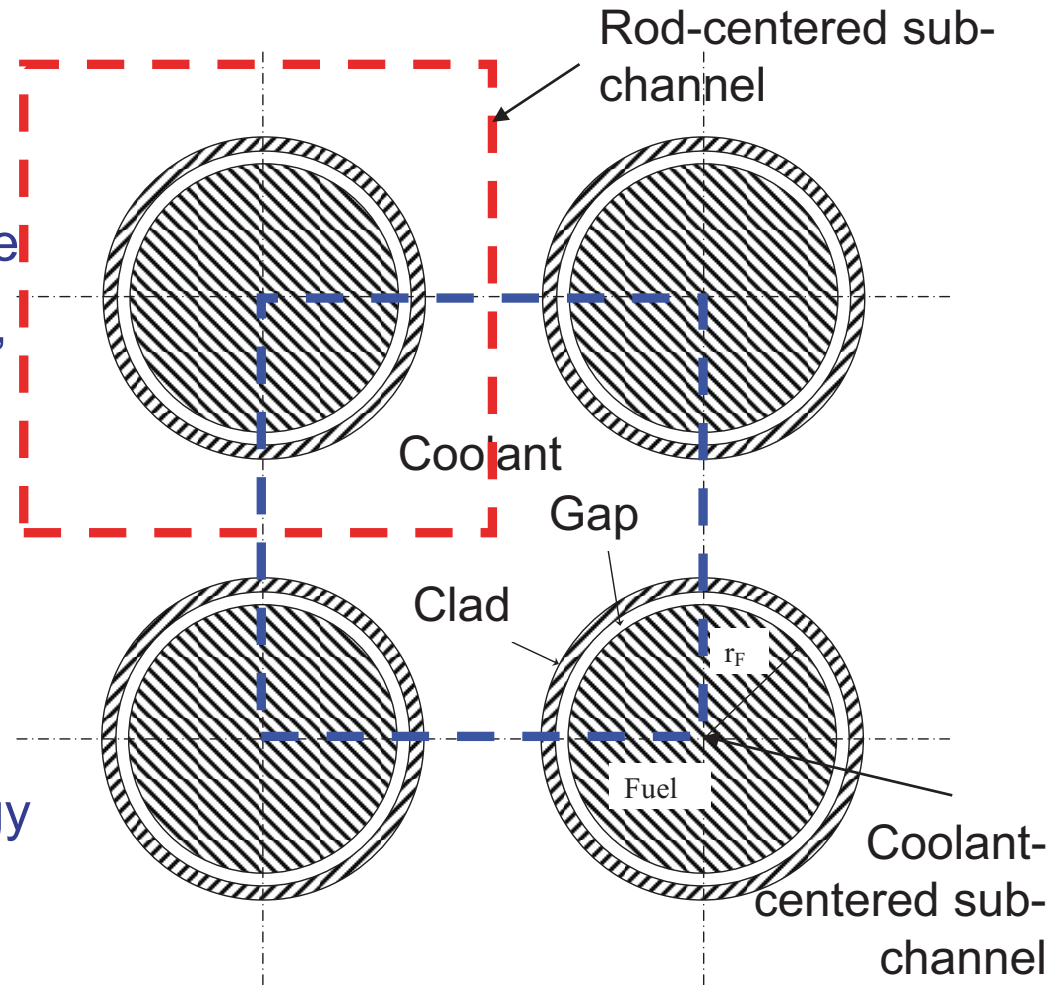
$$q'''(x, y, z) = q_0''' \cos\left(\frac{\pi x}{\tilde{a}}\right) \cos\left(\frac{\pi y}{\tilde{b}}\right) \cos\left(\frac{\pi z}{\tilde{c}}\right)$$

$q_0'''$  - power density at the core centre;  $r=0$ ,  $z=0$



# Isolated Sub-channel Model

- Cross-section over a square lattice with fuel pins
- Heat transfer calculations are performed in an averaged, representative “sub-channel”
- Heat conduction is considered in each rod separately
- Main assumption: no flow of mass, momentum and energy through sub-channel “walls”



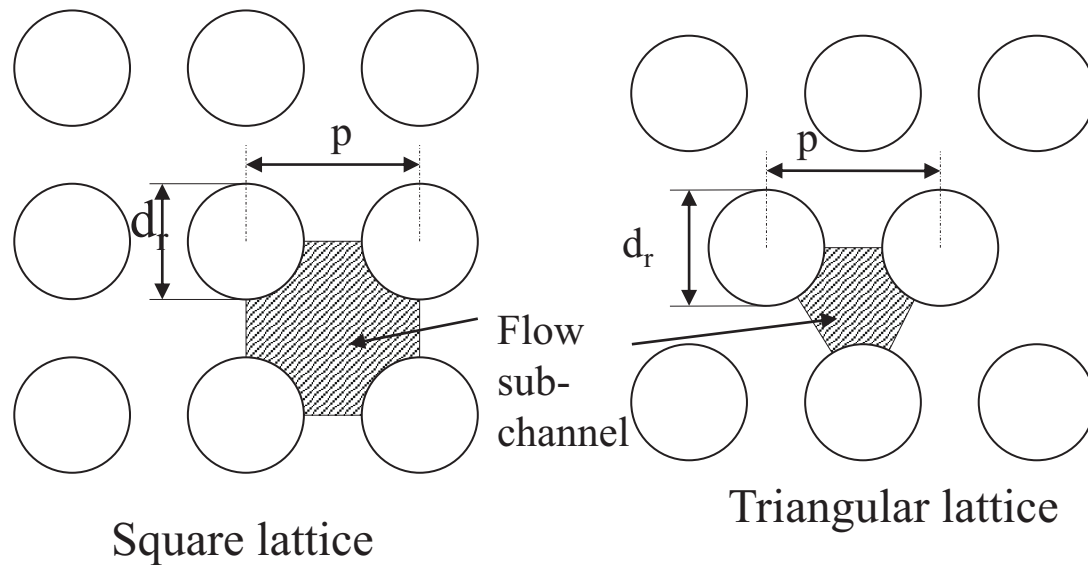
# Basic Parameters Describing Isolated Sub-channel (1)

- Hydraulic diameter
- Flow area
- Wetted perimeter

$$D_h = \frac{4A}{P_w}$$

$A$  – channel cross-section area

$P_w$  – channel wetted perimeter



$$D_h = \begin{cases} d_r \left[ \frac{4}{\pi} \left( \frac{p}{d_r} \right)^2 - 1 \right] & \text{for square lattice} \\ d_r \left[ \frac{2\sqrt{3}}{\pi} \left( \frac{p}{d_r} \right)^2 - 1 \right] & \text{for triangular lattice} \end{cases}$$

$p$  – lattice pitch

$d_r$  – rod diameter

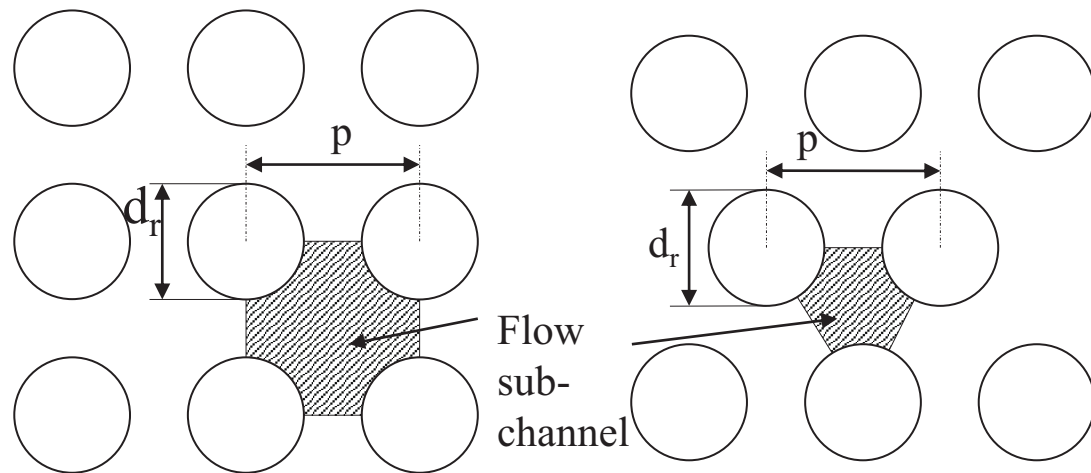
# Basic Parameters Describing Isolated Sub-channel (2)

- Heated diameter
- Flow area
- Heated perimeter

$$D_H = \frac{4A}{P_H}$$

$A$  – channel cross-section area

$P_H$  – sub-channel heated perimeter



Square lattice

Triangular lattice

For  
subchannels  
with all heated  
rods we have:

$$D_H = \begin{cases} d_r \left[ \frac{4}{\pi} \left( \frac{p}{d_r} \right)^2 - 1 \right] & \text{for square lattice} \\ d_r \left[ \frac{2\sqrt{3}}{\pi} \left( \frac{p}{d_r} \right)^2 - 1 \right] & \text{for triangular lattice} \end{cases}$$

$p$  – lattice pitch

$d_r$  – rod diameter

# Whole-Assembly Model

- This model is suitable to BWR fuel assemblies

- Basic parameters:

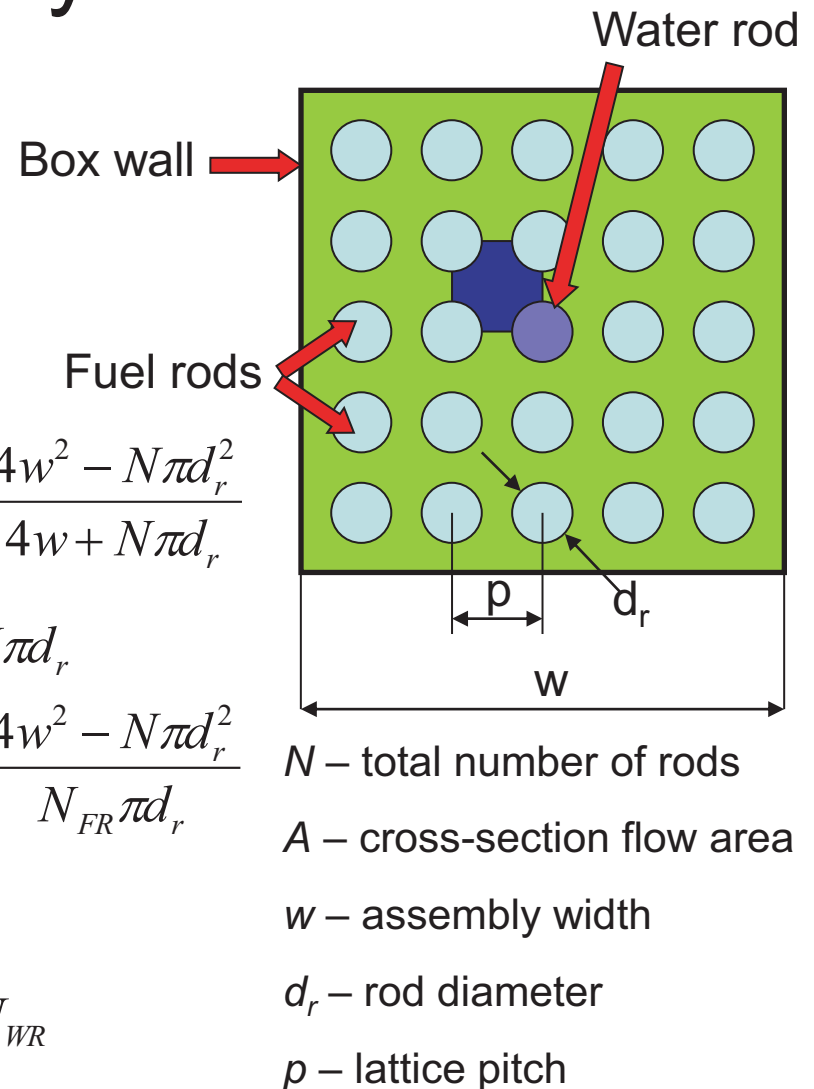
- hydraulic diameter  $D_h$  
$$D_h \equiv \frac{4A}{P_w} = \frac{4w^2 - N\pi d_r^2}{4w + N\pi d_r}$$

- wetted perimeter  $P_w$  
$$P_w = 4w + N\pi d_r$$

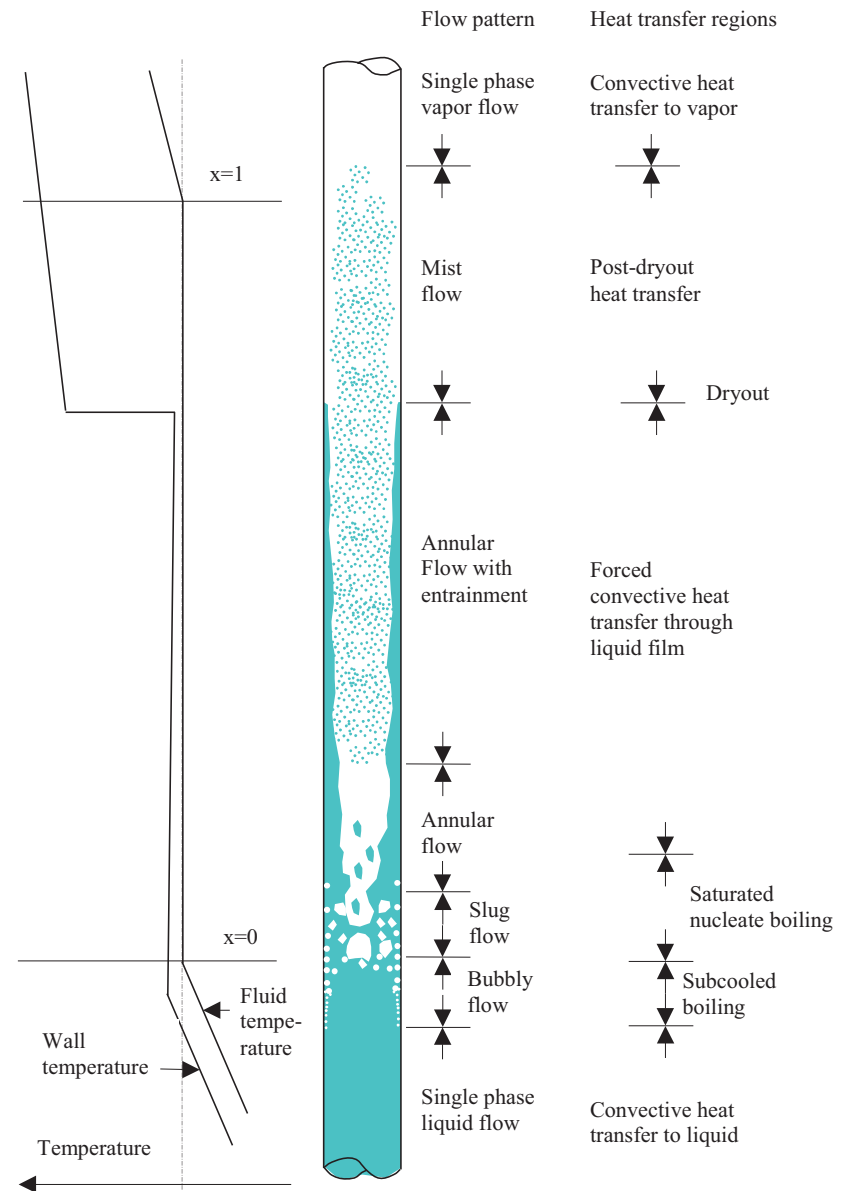
- heated diameter  $D_H$  
$$D_H \equiv \frac{4A}{P_H} = \frac{4w^2 - N\pi d_r^2}{N_{FR}\pi d_r}$$

- heated perimeter  $P_H$  
$$P_H = N_{FR}\pi d_r$$

- $N_{FR}, N_{WR}$  – nr of fuel /water rods 
$$N = N_{FR} + N_{WR}$$



# Flow and Heat Transfer Regimes in a Boiling Channel



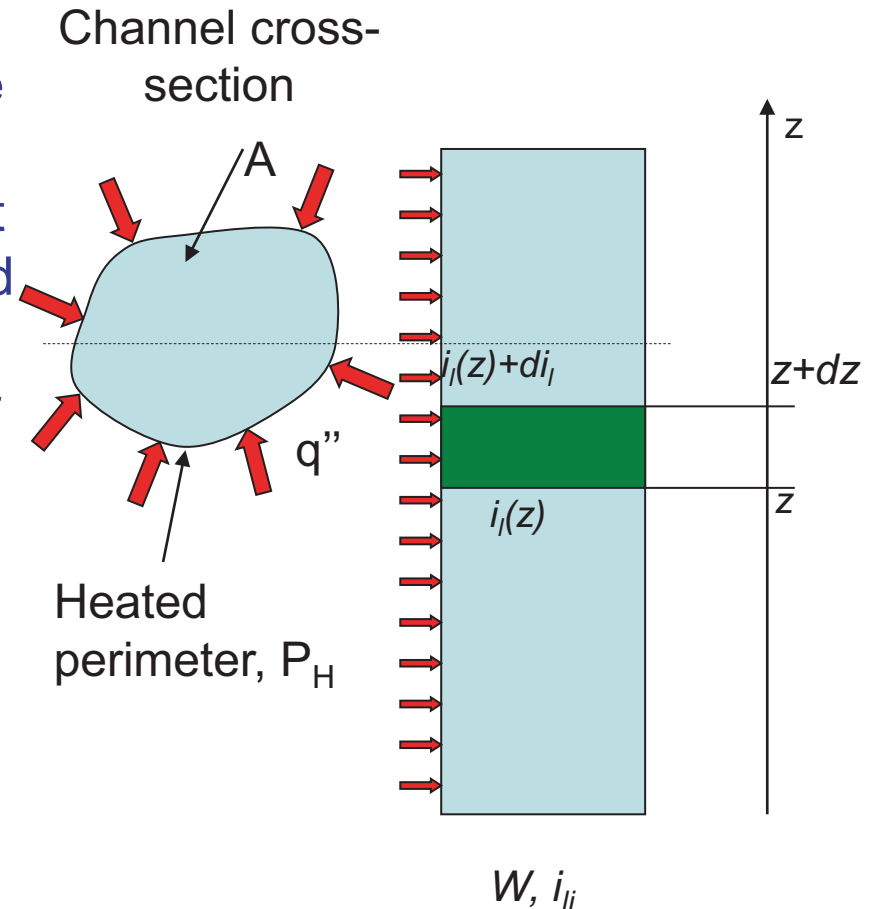
# Coolant Enthalpy Distribution in Heated Channels (1)

- Assume a heated channel as shown in the figure to the right. The channel is uniformly heated along its length with heat flux  $q''$  [W/m<sup>2</sup>], it has a flow cross-section area  $A$  and heated perimeter  $P_H$ .
- The energy balance for a portion of channel  $dz$  is as follows:

$$W \cdot i_l(z) + q''(z) \cdot P_H(z) \cdot dz = W \cdot [i_l(z) + di_l]$$

$$\frac{di_l(z)}{dz} = \frac{q''(z) \cdot P_H(z)}{W}$$

$$W = G \cdot A$$



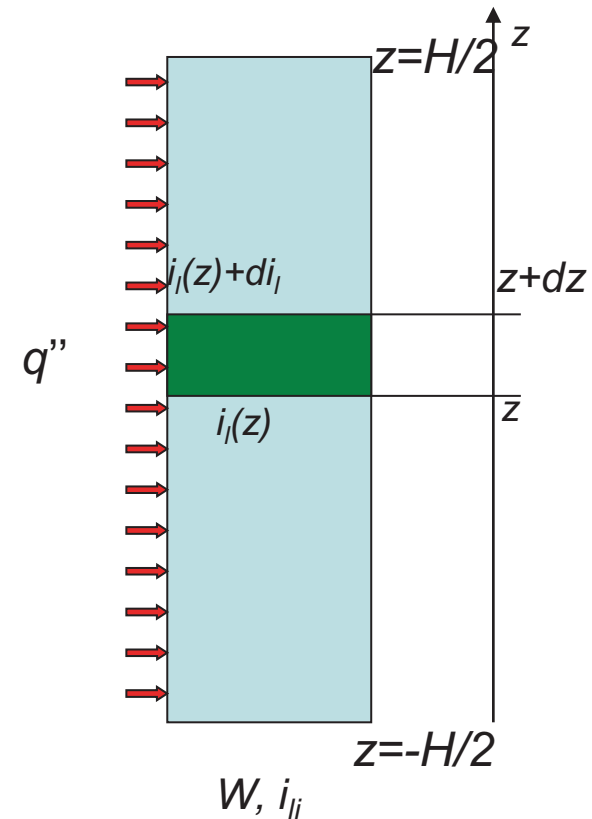
# Coolant Enthalpy Distribution in Heated Channels (2)

- Thus, the enthalpy distribution of coolant is described by the following differential equation:

$$\frac{di_l(z)}{dz} = \frac{q''(z) \cdot P_H(z)}{W}$$

- Integration yields

$$i_l(z) = i_{li} + \frac{1}{W} \int_{-H/2}^z q''(z) \cdot P_H(z) \cdot dz$$



# Coolant Enthalpy Distribution in Heated Channels (3)

- Assuming constant specific heat (calorically perfect fluid) the enthalpy increase can be expressed in terms of the temperature increase as follows:

$$di = c_p * dT$$

- Using  $W = G A$  and assuming a constant channel cross-section area and heat flux distribution, the coolant temperature can be found as,

$$T_{lb}(z) = T_{lbi} + \frac{q'' P_H (z + H / 2)}{c_p G A}$$

$T_{lb} = \frac{\int_A \rho_l c_{pl} v_l T_l dA}{\int_A \rho_l c_{pl} v_l dA}$

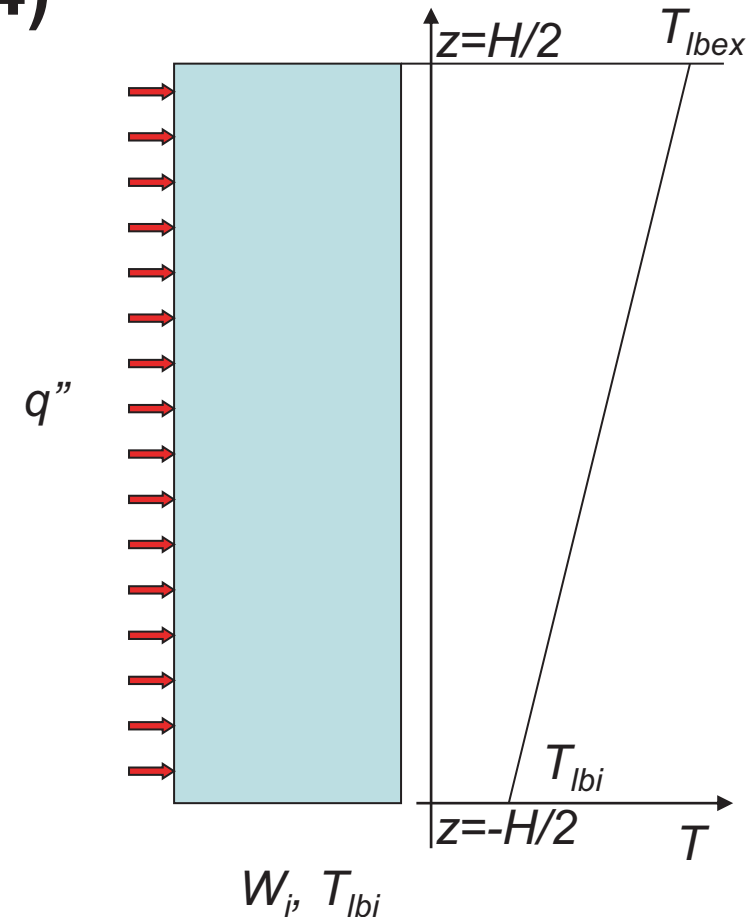
Definition of the bulk liquid temperature



# Coolant Enthalpy Distribution in Heated Channels (4)

- The temperature is thus linearly distributed between the inlet and the exit of the assembly
- The exit temperature becomes

$$T_{lbex} = T_{lbi} + \frac{q'' P_H H}{c_p G A}$$

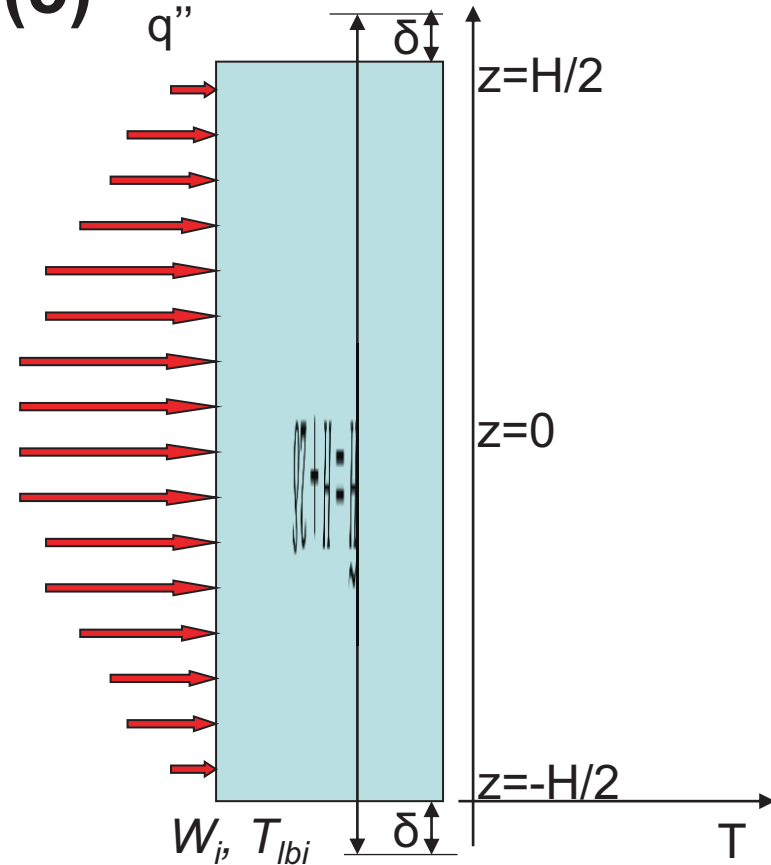


# Coolant Enthalpy Distribution in Heated Channels (5)

- Usually the axial power distribution is non-uniform. In a cylindrical reactor the axial power distribution is given by the cosine function:

$$q''(z) = q_0'' \cdot \cos\left(\frac{\pi z}{\tilde{H}}\right)$$

The differential equation for the enthalpy (temperature) distribution is now



$$\frac{di_l(z)}{dz} = \frac{q_0'' \cdot P_H(z)}{W} \cos\left(\frac{\pi z}{\tilde{H}}\right), \quad \text{or} \quad \frac{dT_{lb}(z)}{dz} = \frac{q_0'' \cdot P_H(z)}{W \cdot c_p} \cos\left(\frac{\pi z}{\tilde{H}}\right)$$

# Coolant Enthalpy Distribution in Heated Channels (6)

- After integration, ( $P_H = \text{const}$ ) the coolant enthalpy (temperature) distribution is as follows

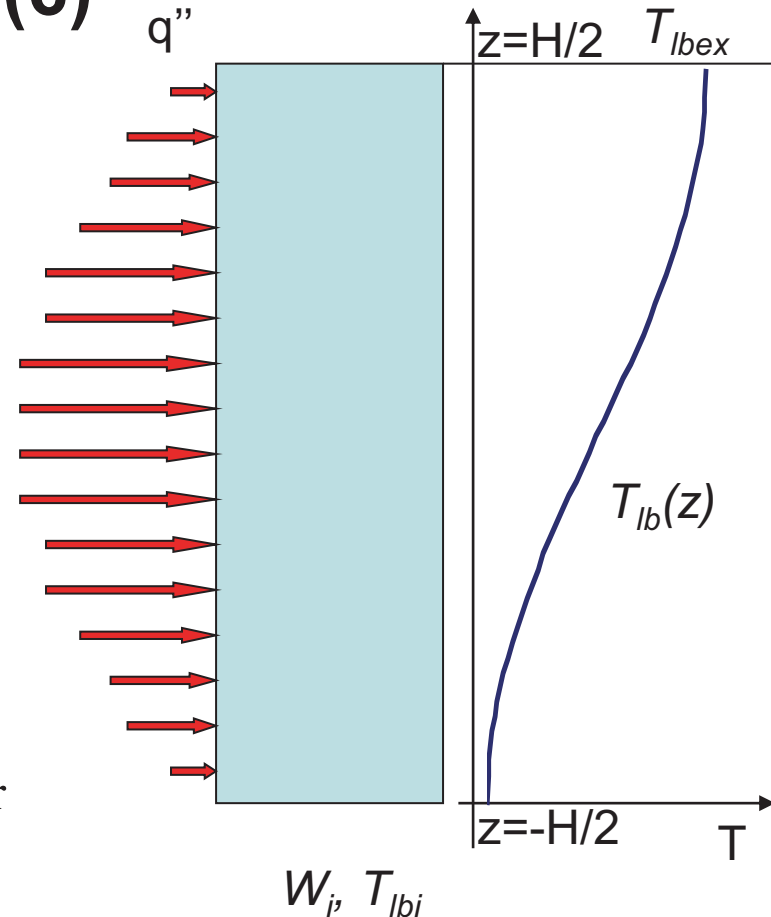
$$i_l(z) = \frac{q_0'' \cdot P_H}{W} \cdot \frac{\tilde{H}}{\pi} \left[ \sin\left(\frac{\pi z}{\tilde{H}}\right) + \sin\left(\frac{\pi H}{2\tilde{H}}\right) \right] + i_{li}, \quad \text{or}$$

$$T_{lb}(z) = \frac{q_0'' \cdot P_H}{W \cdot c_p} \cdot \frac{\tilde{H}}{\pi} \left[ \sin\left(\frac{\pi z}{\tilde{H}}\right) + \sin\left(\frac{\pi H}{2\tilde{H}}\right) \right] + T_{lbi}$$

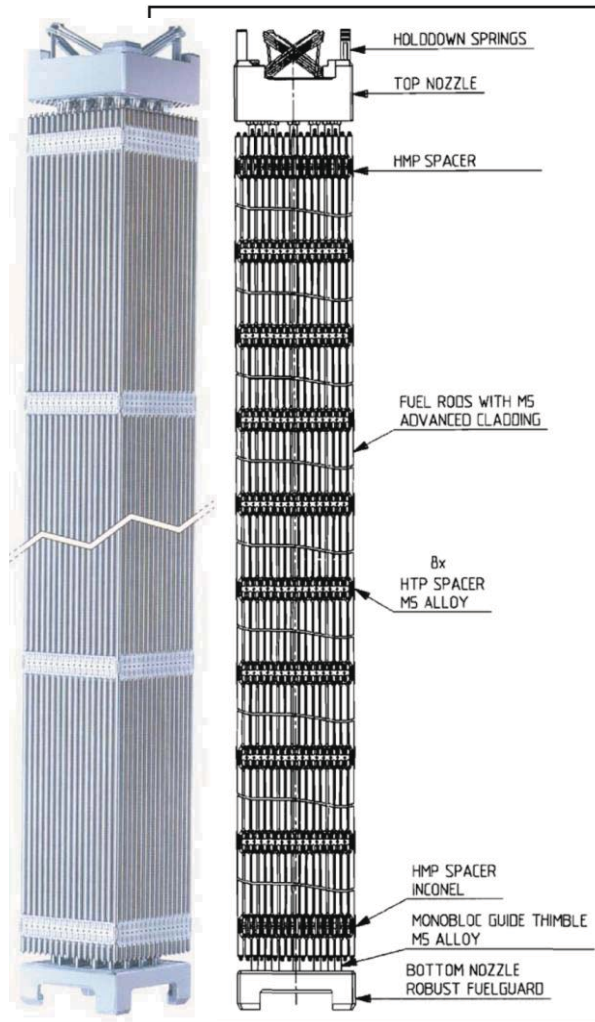
The exit enthalpy (temperature) can be found as:

$$i_{lex} = i_l(H/2) = \frac{2q_0'' \cdot P_H \cdot \tilde{H}}{\pi \cdot W} \sin\left(\frac{\pi H}{2\tilde{H}}\right) + i_{li}, \quad \text{or}$$

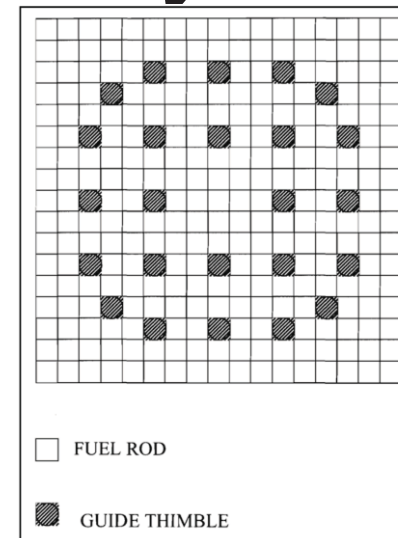
$$T_{lbex} = T_{lb}(H/2) = \frac{2q_0'' \cdot P_H \cdot \tilde{H}}{\pi \cdot W \cdot c_p} \sin\left(\frac{\pi H}{2\tilde{H}}\right) + T_{lbi}$$



# PWR Fuel Assembly

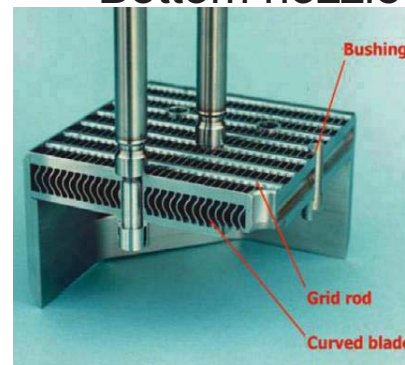


Top nozzle



Cross section

Bottom nozzle with debris filter



# Pressure Drop Calculation

- Calculation of pressure drop in single phase flows, including
  - Friction pressure losses
  - Local losses from the spacer grids
  - Local losses at the assembly inlet and exit
  - Local losses due to flow area change
  - Elevation pressure drop
- The total pressure drop in a vertical channel with length  $H$  and hydraulic diameter  $D_h$  can be calculated from the following equation ( $G = \text{const}$ )

$$-\Delta p_{tot} = -\Delta p_{fric} - \Delta p_{loc} - \Delta p_{elev} = \left( \frac{4C_f H}{D_h} + \sum_i \xi_i \right) \frac{G|G|}{2\rho} + H\rho g$$

# Friction Pressure Losses (1)

- Friction pressure losses in a channel with length  $H$  and hydraulic diameter  $D_h$  is calculated as:

$$-\Delta p_{fric} = \frac{4C_f H}{D_h} \frac{G|G|}{2\rho}$$

- where  $C_f$  is the (Fanning) friction coefficient, which depends on the Reynolds number and wall roughness, defined as

$$C_f \equiv \frac{\tau_w}{\frac{1}{2}\rho U^2}$$

$\tau_w$  – wall shear stress,  
 $U = G/\rho$  – flow velocity

# Friction Pressure Losses (2)

- Friction coefficient for pipes

- Laminar flow ( $\text{Re} < 2300$ )

$$C_f = \frac{16}{\text{Re}}$$

- Turbulent flow (Blasius formula,  $10^4 < \text{Re} < 10^5$ )

$$C_f = \frac{0.0791}{\text{Re}^{0.25}}$$

- Turbulent flow in commercial rough tubes (Colebrook formula)

$$\frac{1}{\sqrt{C_f}} = -4.0 \log_{10} \left( \frac{k / D_h}{3.7} + \frac{1.255}{\text{Re} \sqrt{C_f}} \right)$$

$k$  – wall roughness [m],

$D_h$  – hydraulic diameter [m]

# Friction Pressure Losses (3)

- Friction coefficient for pipes, cont'ed
  - Colebrook formula can be replaced with the Haaland formula (which does not require iterations)

$$\frac{1}{\sqrt{C_f}} = -3.6 \log_{10} \left[ \left( \frac{k / D_h}{3.7} \right)^{1.11} + \frac{6.9}{\text{Re}} \right]$$

$k$  – wall roughness [m],

$D_h$  – hydraulic diameter [m]



# Friction Pressure Losses (4)

- In fuel assemblies, friction coefficients are obtained experimentally and are in general expressed in the following form:

$$C_f = a \operatorname{Re}^{-b}$$

$a, b > 0$  – coefficients that depend on the fuel assembly design

# Friction Pressure Losses (5)

- For triangular lattice with  $1.0 < p/d_r < 1.5$  the following correlation can be used:

$$C_{f,b} = \frac{0.25 \left( 0.96 \frac{p}{d_r} + 0.63 \right)}{(1.82 \log_{10} \text{Re} - 1.64)^2} \quad \text{Re} > 4000$$

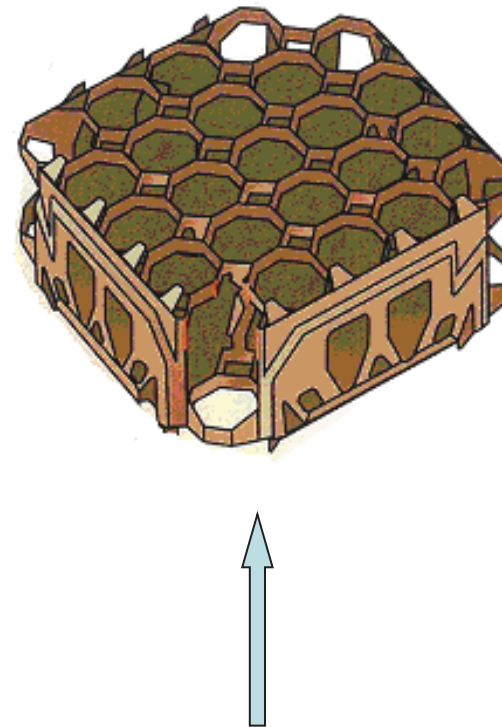
# Local Losses Due to Spacer Grids

- Spacer local pressure loss

- Geometry-dependent
- In general, the pressure loss can be calculated as

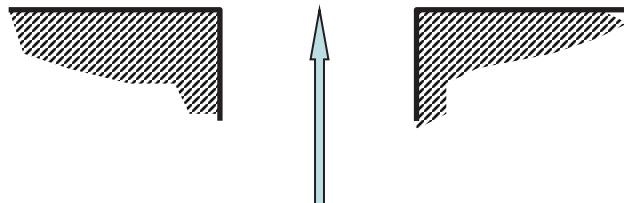
$$\xi_{\text{spac}} = a_1 + a_2 \cdot \text{Re}^{-b}$$

- Constants  $a_1$ ,  $a_2$  and  $b$  are usually obtained from experiments



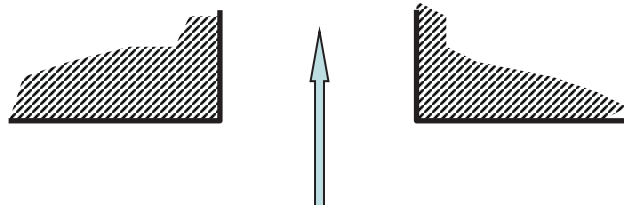
# Local Losses Due to Area Changes

Exit from fuel assembly



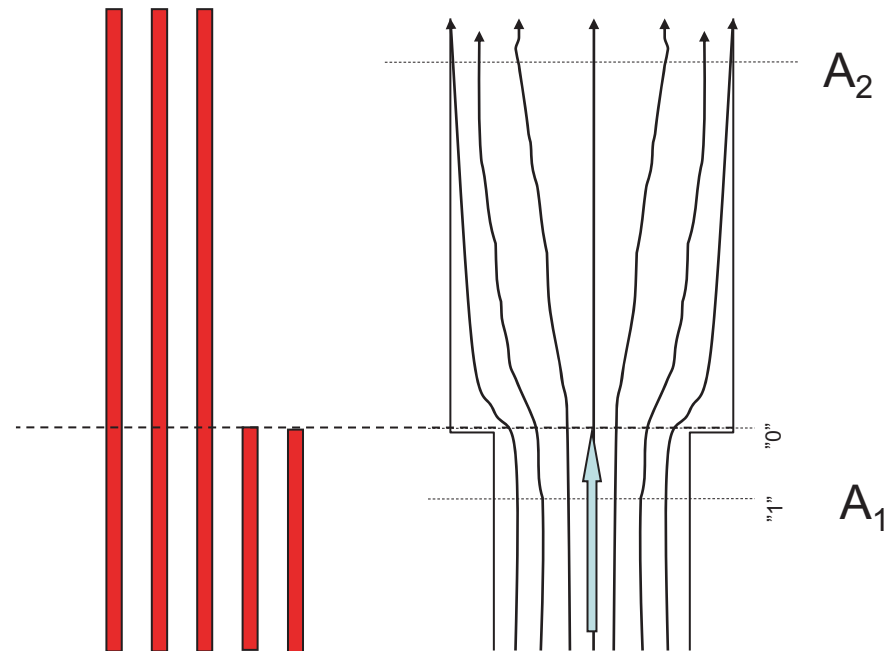
$$-\Delta p_I = \xi_{ex} \cdot \frac{G^2}{2\rho}; \quad \xi_{ex} = 1.0$$

Inlet to fuel assembly



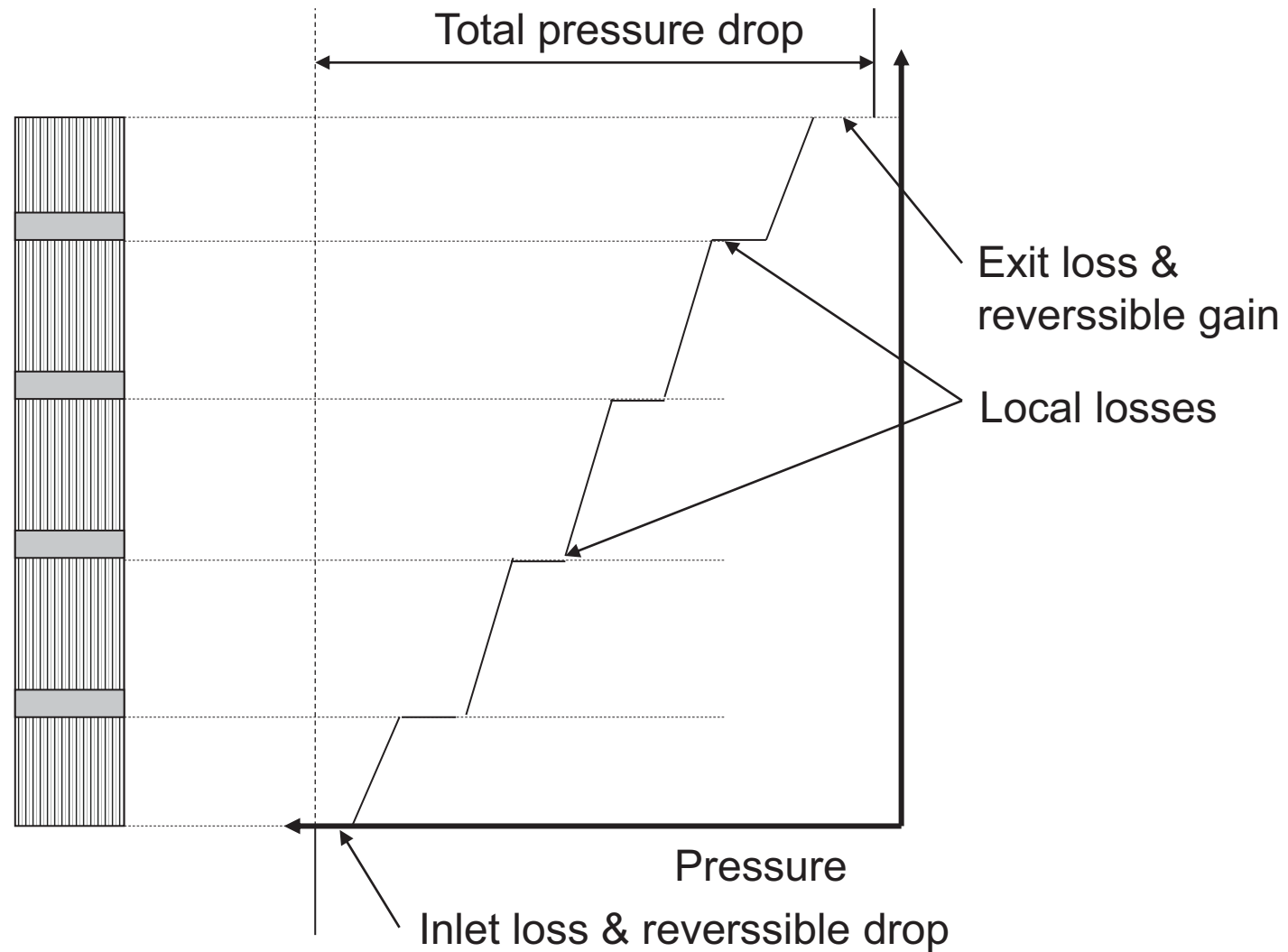
$$-\Delta p_I = \xi_{in} \cdot \frac{G|G|}{2\rho}; \quad \xi_{in} = 0.5$$

Area change due to part-length rods



$$-\Delta p_I = \left(1 - \frac{A_1}{A_2}\right)^2 \cdot \frac{G_1|G_1|}{2\rho}; \quad \xi_{enl} = \left(1 - \frac{A_1}{A_2}\right)^2$$

# Loss Distribution in Fuel Assembly



# Pressure Drop in Two-Phase Flows

- Steady-state momentum equation for a homogeneous two-phase mixture flow in a channel can be written as,

$$-\frac{dp}{dz} = \left( \frac{dp}{dz} \right)_w + \rho_m g \sin \varphi + \frac{1}{A} \frac{d}{dz} \left( \frac{G^2 A}{\rho_M} \right)$$

- Where two definitions of mixture density are introduced:

- Mixture static density  $\rho_m = \sum_k \rho_k \alpha_k$

- Mixture dynamic density  $\rho_M = \left( \sum_k \frac{x_k^2}{\rho_k \alpha_k} \right)^{-1}$

# Local Pressure Loss in Two-Phase Flows

- Local pressure losses in two-phase flows are calculated as:

$$-\Delta p_{loc} = \phi_{lo,d}^2 \xi \frac{G^2}{2\rho_f}$$

Here:

$G$  - total mass flux, kg/m<sup>2</sup>.s

$\xi$  - local (single-phase) loss coefficient

$$\phi_{lo,d}^2 = \left[ 1 + \left( \frac{\rho_f}{\rho_g} - 1 \right) x \right] \text{ - HEM local two-phase multipl.}$$

# Friction pressure loss in two-phase flows

- It can be shown that the ratio of two-phase friction loss to single-phase friction loss is as follows,

$$\left(\frac{dp}{dz}\right)_{w,tp} / \left(\frac{dp}{dz}\right)_{w,lo} = \frac{C_{f,tp}}{C_{f,lo}} \frac{\rho_l}{\rho_m}$$

- The above ratio is called a two-phase friction multiplier and is as follows

$$\phi_{lo}^2 = \left(\frac{dp}{dz}\right)_{w,tp} / \left(\frac{dp}{dz}\right)_{w,lo} = \frac{C_{f,tp}}{C_{f,lo}} \frac{\rho_l}{\rho_m}$$

- It should be noted that it is a local variable



# Two-Phase Friction Multiplier using HEM

- For Homogeneous Equilibrium Model, it can be shown that the two-phase friction multiplier is the following function of the local equilibrium quality:

$$\phi_{lo}^2 = \left[ 1 + \left( \frac{\mu_f}{\mu_g} - 1 \right) x \right]^{-0.25} \left[ 1 + \left( \frac{\rho_f}{\rho_g} - 1 \right) x \right]$$

- where it is assumed that mixture viscosity is given as:

$$\frac{1}{\mu_m} = \frac{x}{\mu_g} + \frac{1-x}{\mu_f}$$

- it should be noted that other models of mixture viscosity are used as well (see Compendium in Thermal-Hydraulics)

# Rod Bundle Correlations for $\phi_{lo}^2$

- Local two-phase friction multiplier in general depends on local conditions (pressure, mass flux, heat flux) and geometry (pipe, bundle)
- For a rod bundle geometry the following correlation has been obtained (FRIGG)

$$\phi_{lo}^2 = 1 + (2234 - 0.348G) \left( \frac{x}{p} \right)^{0.96}$$

x – quality

p – pressure (bar)

G – mass flux (kg/m<sup>2</sup>s)

- To capture the effect of heating:

$$\frac{(\phi_{lo}^2)_{diabatic}}{(\phi_{lo}^2)_{adiabatic}} = 1 + C \left( \frac{q''}{G} \right)^{0.7}$$

C – constant coefficient

q'' - heat flux (W/m<sup>2</sup>)

G – mass flux (kg/m<sup>2</sup>s)

# EPRI Correlation for $\phi_{lo}^2$

$$\phi_{lo}^2 = \left[ 1 + x \left( \frac{\rho_f}{\rho_g} - 1 \right) C \right]$$

$$C = \begin{cases} 1.02x^{-0.175} G_R^{-0.45} & \text{for } p > 4.137 \text{ MPa} \\ 0.357(1 + p_R)x^{-0.175} G_R^{-0.45} & \text{for } 2.068 < p \leq 4.137 \text{ MPa} \end{cases}$$

$$p_R = \frac{p}{p_{cr}}; G_R = \frac{G}{1356.2}$$

$x$  – equilibrium quality

$p$  – pressure (Pa)

$G$  – mass flux (kg/m<sup>2</sup>s)

$p_{cr}$  – critical pressure (22.1 MPa)

Parameter range:  $2.068 < p < 8.963$  MPa;  $0 < x < 1$ ;  $475 < G < 4475$  kg/m<sup>2</sup>s;  
 $5.08 < d < 15.24$  mm;  $127 < L < 2540$  mm; geometry: round tubes and vertical  
upflow; based on 1533 experimental points; RMS error: 9.7%

# Mean Value of $\phi_{lo}^2$ Over Channel Length

- Integration of  $\phi_{lo}^2$  along a channel length gives

$$r_3 = \frac{1}{L} \int_0^L \phi_{lo}^2 dz \quad \phi_{lo}^2 = \left[ 1 + \left( \frac{\mu_f}{\mu_g} - 1 \right) x \right]^{-0.25} \left[ 1 + \left( \frac{\rho_f}{\rho_g} - 1 \right) x \right]$$

- The integral to calculate  $r_3$  is thus a function of the quality distribution along the channel.
- In particular, if  $x = \text{const}$  (unheated channel):

$$r_3 = \phi_{lo}^2$$

# Enthalpy and Quality in Heated Channel

- For heated channel, we have:

$$di = \frac{q''(z)P_H dz}{W} \Rightarrow d\left(\frac{i - i_f}{i_{fg}}\right) \equiv dx = \frac{q''(z)P_H dz}{Wi_{fg}}$$

thus, assuming  $z = 0$  at the inlet:

$$x(z) - x_{in} = \frac{P_H}{Wi_{fg}} \int_0^z q''(z') dz'$$

For uniformly heated channel:

$$x(z) = x_{in} + \frac{P_H q''}{Wi_{fg}} z$$

# Total Pressure Drop in Boiling Channel

- Integration of the momentum eq. gives the total pressure drop for two-phase flows in channel with length L as:

$$-\Delta p = \underbrace{r_3 C_{f,lo} \frac{4L}{D} \frac{G^2}{2\rho_f}}_{\text{friction}} + \underbrace{r_4 L \rho_f g \sin \varphi}_{\text{gravity}} + \underbrace{r_2 \frac{G^2}{\rho_f}}_{\text{acceleration}} + \underbrace{\left( \sum_{i=1}^N \phi_{lo,d,i}^2 \xi_i \right) \frac{G^2}{2\rho_f}}_{\text{local}}$$

- where:

- friction multiplier:  $r_3 = \frac{1}{L} \int_0^L \phi_{lo}^2 dz$

- gravity multiplier:  $r_4 = \frac{1}{L \rho_f} \int_0^L [\alpha \rho_g + (1-\alpha) \rho_f] dz$

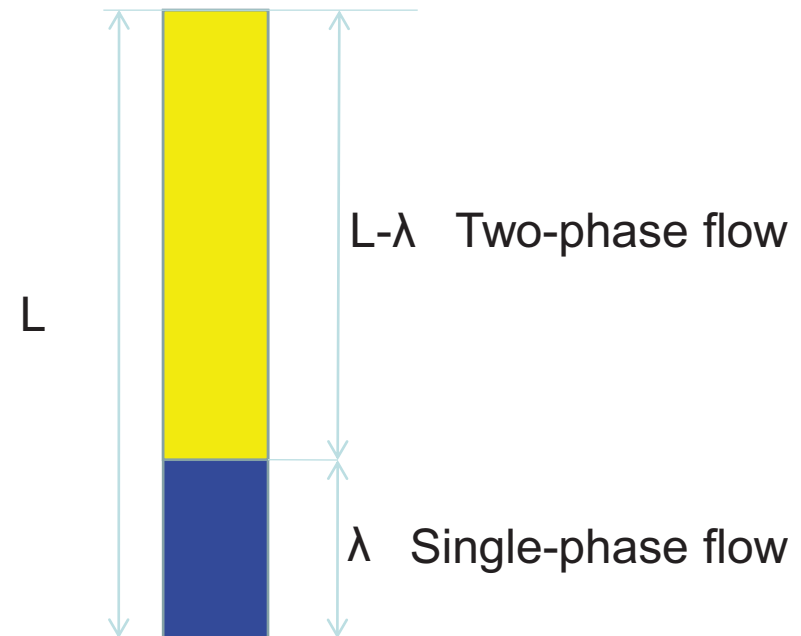
- acceleration multiplier:  $r_2 \equiv \rho_f \int_0^L \frac{d}{dz} \left[ \frac{x^2}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha) \rho_f} \right] dz = \left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_{ex} - \left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_{in}$

# Friction Loss in BWR Fuel Assembly

- Thus to find friction pressure drop in heated fuel assembly:
  - Find the location of the onset of two-phase flow. If HEM is used, it will be at location where  $x = 0$
  - Let  $z = \lambda = z_{\text{SUB}}$  where  $x = 0$

$$-\Delta p_{\text{fric}} = -\int_0^L \left( \frac{dp}{dz} \right)_{\text{fric}} dz =$$

$$-\int_0^{\lambda} \left( \frac{dp}{dz} \right)_{\text{fric}} dz - \int_{\lambda}^L \left( \frac{dp}{dz} \right)_{\text{fric}} dz$$



# Friction Loss in BWR Fuel Assembly

- Thus: 
$$-\Delta p_{fric} = \left( \frac{4C_f \lambda}{D_h} + \frac{4C_{f,lo}}{D_h} \int_{\lambda}^L \phi_{lo}^2 dz \right) \frac{G^2}{2\rho_f} = \left[ \frac{4C_f \lambda}{D_h} + r_3 \frac{4C_{f,lo}(L-\lambda)}{D_h} \right] \frac{G^2}{2\rho_f}$$

- where 
$$r_3 = \frac{1}{L-\lambda} \int_{\lambda}^L \phi_{lo}^2 dz$$

Assuming uniform power distributions with  $q''=\text{const}$

where 
$$x_{ex} = x_{in} + \frac{q'' P_H}{Wi_{fg}} L$$

$$r_3 = \int_0^1 \frac{1 + x_{ex} \left( \frac{\rho_f}{\rho_g} - 1 \right) \zeta}{\left[ 1 + x_{ex} \left( \frac{\mu_f}{\mu_g} - 1 \right) \zeta \right]^{0.25}} d\zeta$$

is the exit quality and  $x_{in}$  is the inlet quality



# Gravity Pressure Drop

- The gravity pressure drop multiplier is given as:

$$r_4 = \frac{1}{L\rho_f} \int_0^L [\alpha\rho_g + (1-\alpha)\rho_f] dz \quad \text{where using HEM}$$

the local void fraction is obtained as:

$$\alpha = \frac{1}{1 + \frac{\rho_g}{\rho_f} \cdot \left( \frac{1-x}{x} \right)} \quad \text{for } 0 < x < 1$$

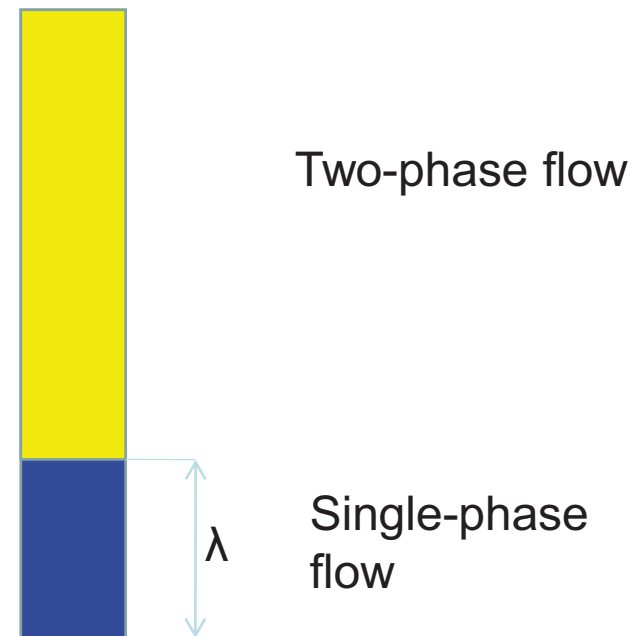
The integral to calculate  $r_4$  is thus a function of the quality distribution along the channel.

# Gravity Pressure Drop in BWR Fuel Assembly

- Thus to find the gravity pressure drop in a heated fuel assembly:
- Find the location of the onset of two-phase flow. If HEM is used, it will be at location where  $x = 0$
- Let  $z = \lambda$  where  $x = 0$

$$-\Delta p_{grav} = -\int_0^L \left( \frac{dp}{dz} \right)_{grav} dz =$$

$$-\int_0^{\lambda} \left( \frac{dp}{dz} \right)_{grav} dz - \int_{\lambda}^L \left( \frac{dp}{dz} \right)_{grav} dz$$



# Gravity Pressure Drop in BWR Fuel Assembly

- Thus:

$$-\Delta p_{grav} = \int_0^{\lambda} \rho_l g \sin \varphi dz + \int_{\lambda}^L [\alpha \rho_g + (1 - \alpha) \rho_f] g \sin \varphi dz =$$

$$\lambda \rho_l g \sin \varphi + r_4 (L - \lambda) \rho_f g \sin \varphi$$

where:

$$r_4 = \frac{1}{(L - \lambda) \rho_f} \int_{\lambda}^L [\alpha \rho_g + (1 - \alpha) \rho_f] dz$$

assuming uniform power  
distribution:

$$r_4 = 1 - x_{ex} \int_0^1 \frac{\zeta}{\rho_g / (\rho_f - \rho_g) + x_{ex} \zeta} d\zeta$$

where  $x_{ex}$  is the exit quality

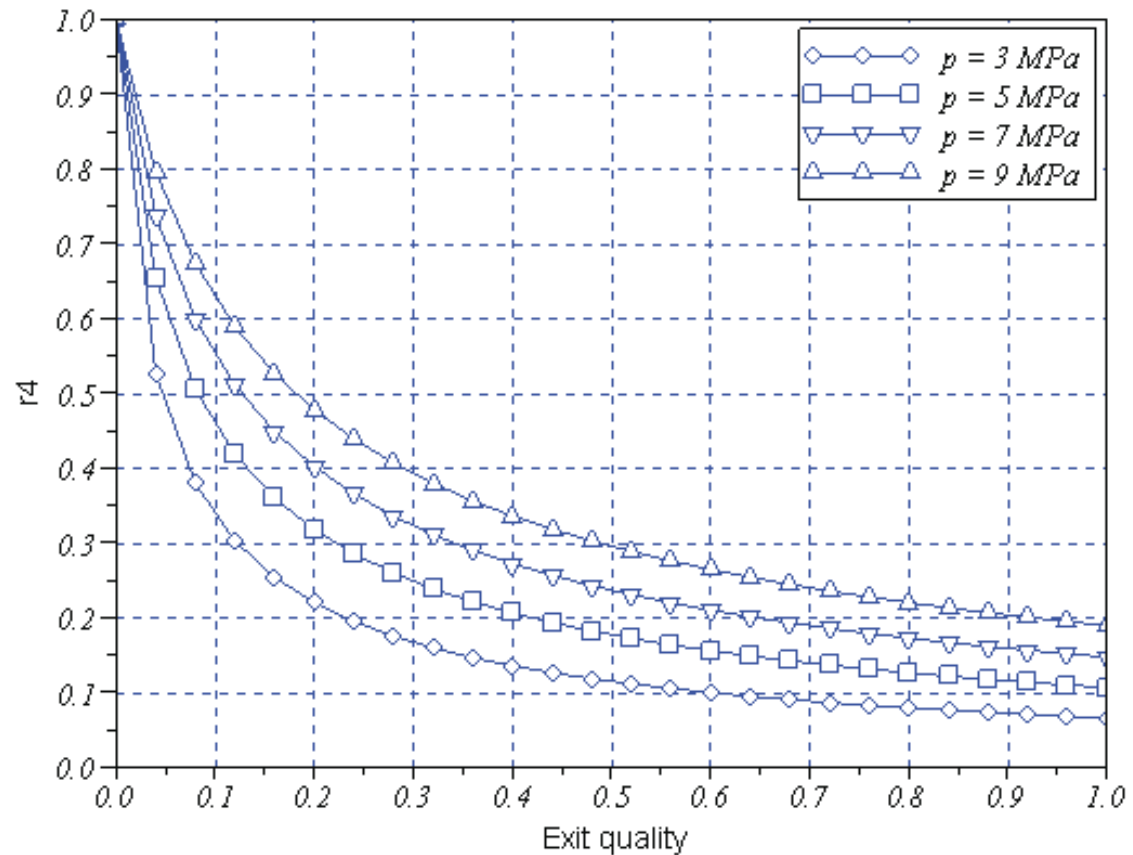
# Gravity Pressure Drop Multiplier

$$r_4 = 1 - x_{ex} \int_0^1 \frac{\zeta}{\rho_g / (\rho_f - \rho_g) + x_{ex}}$$

This graph can be used to find the value of the  $r_4$  multiplier for known exit quality and system pressure in uniformly heated channel and  $x_{in}=0$

The gravity pressure drop is then found as:

$$-\Delta p_{grav} = \lambda \rho_f g \sin \varphi + r_4 (L - \lambda) \rho_f \sin \varphi$$



# Acceleration Pressure Drop in Two-Phase Flows

For channel with subcooled water at inlet, the acceleration multiplier can be calculated as:

$$\begin{aligned}
 r_2 &\equiv \rho_f \int_0^L \frac{d}{dz} \left[ \frac{x^2}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha) \rho_f} \right] dz = \rho_f \int_0^\lambda \frac{d}{dz} \left[ \frac{x^2}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha) \rho_f} \right] dz + \\
 &\quad \rho_f \int_\lambda^L \frac{d}{dz} \left[ \frac{x^2}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha) \rho_f} \right] dz = \left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_{ex} - \underbrace{\left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_\lambda}_0 = \\
 &\quad \left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_{ex} - 1
 \end{aligned}$$

Thus:

$$r_2 = \left[ \frac{x^2 \rho_f}{\alpha \rho_g} + \frac{(1-x)^2}{(1-\alpha)} \right]_{ex} - 1$$

# Void Fraction Calculation

- Prediction of void fraction is important because it affects the moderator density, thus, it affects power generation in nuclear reactors
- Two models are widely used in saturated region:
  - Homogeneous Equilibrium Model (HEM)
  - Drift-Flux Model (DFM)
- Void fraction in subcooled region
  - Onset of Nucleate Boiling (ONB)
  - Onset of Significant Void (OSV)
  - Actual quality model

# Void Fraction - HEM (1)

- In HEM, it is assumed that both phases are in the thermodynamic equilibrium and flow with the same speed
- Void fraction is calculated in two steps:
  - first the value of the equilibrium quality ( $x_e$ ) is found as  $x_e(z) \equiv \frac{i(z) - i_f}{i_{fg}}$
  - next the value of void fraction is calculated from the following equation:
 
$$\alpha(z) = \begin{cases} 0 & \text{for } x_e \leq 0 \\ \frac{1}{1 + \frac{\rho_g}{\rho_f} \cdot \left( \frac{1 - x_e(z)}{x_e(z)} \right)} & \text{for } 0 < x_e < 1 \\ 1 & \text{for } x_e \geq 1 \end{cases}$$

# Void Fraction - DFM

- Drift flux model allows for:
  - different velocities for both phases
  - thermodynamic equilibrium/non-equilibrium
- The void fraction is calculated from the following relationship:

$$\alpha = \frac{J_v}{C_0 J + U_{vj}} \quad J = J_v + J_l$$
$$J_v = \frac{G_v}{\rho_v} = \frac{xG}{\rho_v}$$
$$J_l = \frac{G_l}{\rho_l} = \frac{(1-x)G}{\rho_l}$$

- here  $C_0$  and  $U_{vj}$  are the distribution parameter and the drift velocity, respectively. They are flow-regime dependent.
- $J_v$  and  $J$  are superficial velocities for vapor and for the mixture, respectively

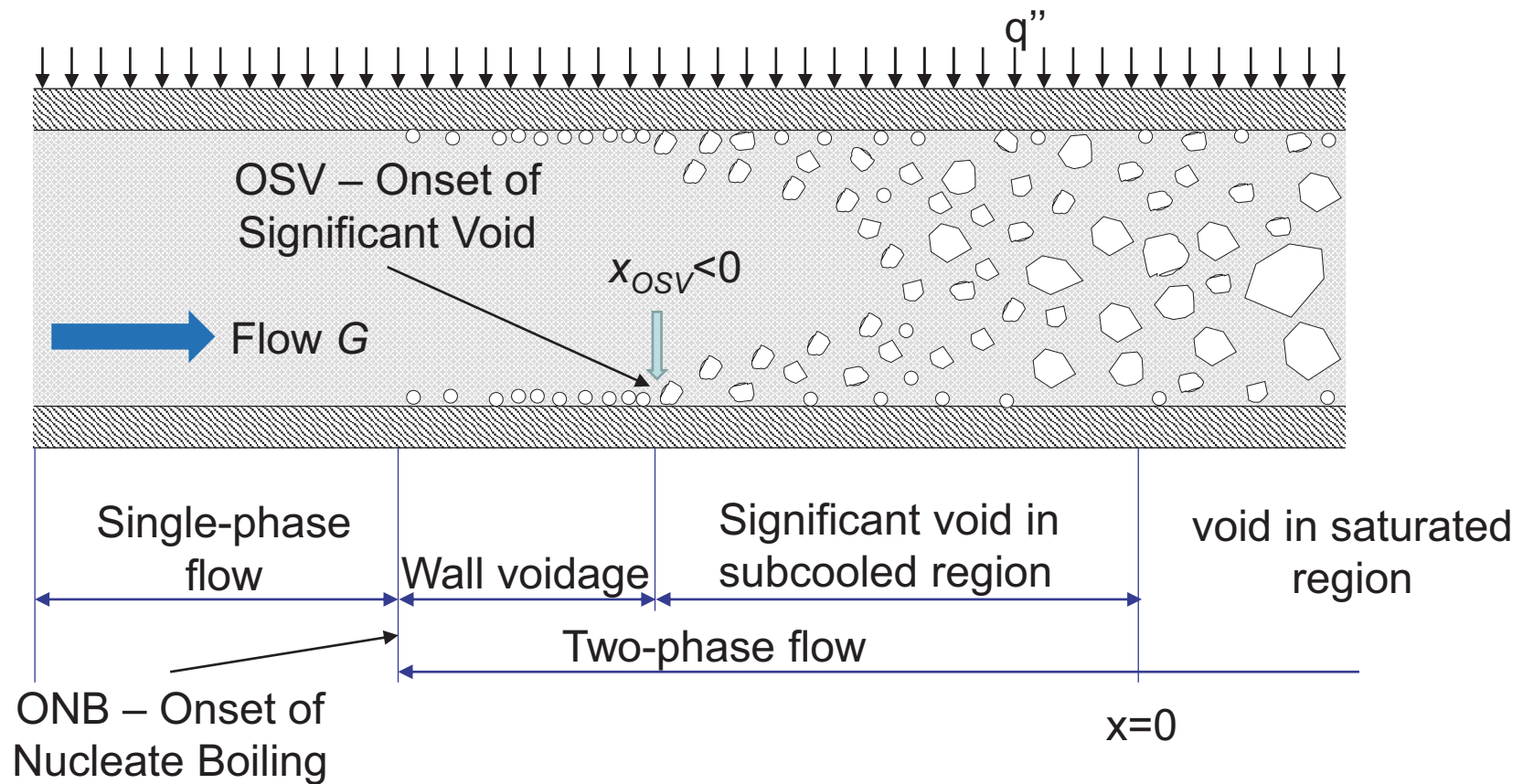


# DFM in Thermodynamic Equilibrium

Flow pattern	Distribution parameter	Drift velocity
Bubbly $0 < \alpha \leq 0.25$	$C_0 = \begin{cases} 1 - 0.5p/p_{\sigma} & D \geq 0.05m \\ 1.2 & p/p_{\sigma} < 0.5 \\ 1.4 - 0.4p/p_{\sigma} & p/p_{\sigma} \geq 0.5 \end{cases} \quad D < 0.05m$ <sup>1)</sup>	$U_{vj} = 1.41 \left( \frac{\sigma g (\rho_l - \rho_v)}{\rho_l^2} \right)^{0.25}$
Slug/churn $0.25 < \alpha \leq 0.75$	$C_0 = 1.15$	$U_{vj} = 0.35 \left( \frac{gD(\rho_l - \rho_v)}{\rho_l} \right)^{0.5}$
Annular $0.75 < \alpha \leq 0.95$	$C_0 = 1.05$	$U_{vj} = 23 \left( \frac{\mu_l j_l}{\rho_v D_h} \right)^{0.5} \frac{(\rho_l - \rho_v)}{\rho_l}$
Mist $0.95 < \alpha < 1$	$C_0 = 1.0$	$U_{vj} = 1.53 \left( \frac{\sigma g (\rho_l - \rho_v)}{\rho_v^2} \right)^{0.25}$

<sup>1)</sup>  $p_{\sigma}$  – critical pressure       $\sigma$  - surface tension       $D=D_h$  – hydraulic diameter

# Void Fraction in Subcooled Region



# Void Fraction – Subcooled Boiling (1)

- It can be assumed that void fraction is negligible up to the Onset of Significant Void (OSV) point.
- This point occurs at the location, where equilibrium quality becomes (Saha-Zuber model):

$$x_{e,OSV} = \begin{cases} -0.0022 \frac{q'' \cdot D_h \cdot c_{pf}}{i_{fg} \cdot \lambda_f} & \text{for } Pe < 70000 \\ -154 \frac{q''}{G \cdot i_{fg}} & \text{for } Pe \geq 70000 \end{cases}$$

– here Pe is the Peclet number, defined as:

$$Pe = Re \cdot Pr = \frac{G \cdot D_h \cdot c_{pf}}{\lambda_f}$$

$q''$  – heat flux, W/m<sup>2</sup>  
 $D_h$  – hydraulic diameter, m  
 $c_{pf}$  – fluid spec. heat, J/kgK  
 $\lambda_f$  – thermal conduct. W/mK

# Void Fraction – Subcooled Boiling (2)

- The actual quality is approximated as (Levy's model):

$$x_a(z) = x_e(z) - x_e(z_{OSV}) \cdot e^{\frac{x_e(z)}{x_e(z_{OSV})} - 1}$$

- The void fraction is then found as:

$$\alpha = \frac{J_v}{C_0 J + U_{vj}}$$

$$J_v = \frac{x_a G}{\rho_g} \quad \text{superficial velocity of vapour}$$

– where  $C_0 = \beta \left[ 1 + \left( \frac{1}{\beta} \right)^b \right]$

$$J_l = \frac{(1 - x_a) G}{\rho_f} \quad \text{Superficial velocity of liquid}$$

$$\beta = \frac{1}{1 + \frac{\rho_g}{\rho_f} \frac{1 - x_a(z)}{x_a(z)}}$$

$$b = \left( \frac{\rho_g}{\rho_f} \right)^{0.1}$$

$$U_{vj} = 2.9 \left( \frac{\sigma g (\rho_f - \rho_g)}{\rho_f^2} \right)^{0.25} \quad \sigma - \text{surface tension, N/m}$$