SH2702 Nuclear Reactor Technology

Project work Task 6

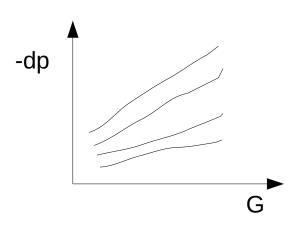
Project work

Topic numbers	Topics
1	Design, operation and safety features of NuSCALE
2	Design, operation and safety features of ABWR
3	Design, operation and safety features of ESBWR
4	Design, operation and safety features of EPR
5	Design, operation and safety features of AP1000

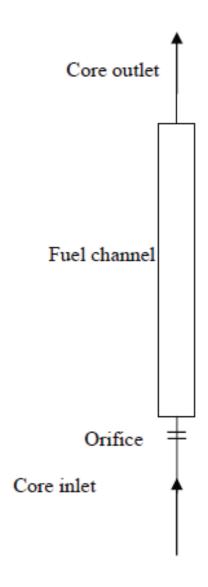
- Task 1 General design specification of the nuclear power plant with selected reactor type
- Task 2 Operational principles of the power plant
- Task 3 Safety features of the power plant
- Task 4 Calculation of selected core parameters
- Task 5 Calculation of CHF margins in a hot channel
- Task 6 Calculation of the maximum cladding and fuel pellet Temperature

Task 4

- 1. Data collection
 - Tables are recommended
- 2. core-averaged thermal-hydraulic calculations
 - Axial enthalpy/temperature distribution
 - Axial void fraction distribution
 - BWRs, from subcooled to saturated
 - Axial pressure distribution
 - Inlet orifices pressure loss, BWRs (50%), PWRs (25%)
 - Flow characteristic of the core (-dp)=f(G)
 - 0%, 50%, 100%, 150% power
 - 1% to 150% flow



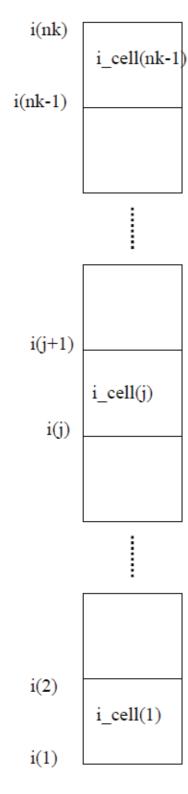
Task 4



- Inlet orifices pressure loss
 - BWRs (50% at nominal operating conditions)
 - PWRs (25% at nominal operating conditions)

$$\Delta p = p_{out} - p_{in} = \Delta p_{FuelChannel} + \Delta p_{Orifice}$$

$$\left| \Delta p_{Orifice} \right| = \xi_{Orifice} \frac{\rho U^2}{2} = \xi_{Orifice} \frac{G^2}{2\rho}$$



Task 4 Nodalization and numerical solution

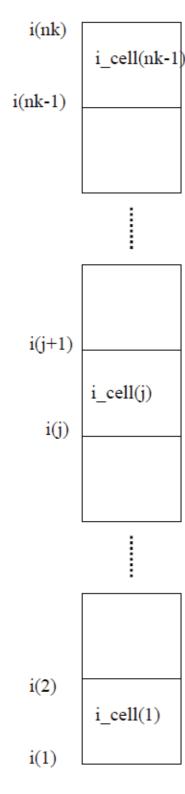
- for j = 2 to nk
 - $-i(j) = i(j-1) + q_cell(j-1) / W$ (energy balance)
- end for
- while p not converged
 - p(1) = pin + dpInletOrifice
 - for j = 2 to nk
 - xe(j), xa(j), alpha(j) (void fraction model)
 - dpf_cell(j-1), dpg_cell(j-1), dpa_cell(j-1), dpl_cell(j-1)
 - dp_cell(j-1) (pressure drop calculation)
 - $p(j) = p(j-1) + dp_cell(j-1)$
 - end for
- end while p
- T(j)
 - f(p(j), i(j)) for subcooled water
 - Tsat(j) for saturated water
- Inlet orifices pressure loss coefficient (designed for nominal condition)
- Flow characteristic of the core (-dp)=f(G)

Task 5

- 1. Hot channel
 - Find data on power distribution, otherwise use the simplified shape

$$q''(r,z) = q_0'' J_0 \left(\frac{2.405r}{\tilde{R}} \right) \cos \left(\frac{\pi z}{\tilde{H}} \right)$$

- Find peaking factor in radial direction
- 2. CHF
 - Find CHF, DNB for PWRs, Dryout for BWRs
 - Calculate thermal margin parameters
 - MDNBR, MCPR
- 3. Hot channel result
 - Axial enthalpy/temperature distribution
 - Axial void fraction distribution
 - BWRs, from subcooled to saturated
 - Axial pressure distribution
 - Axial distribution of DNBR and location of MDNBR, for PWRs

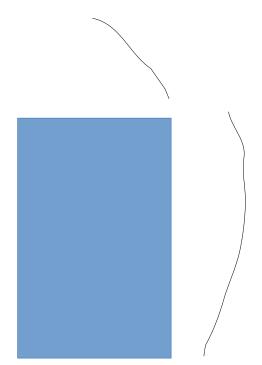


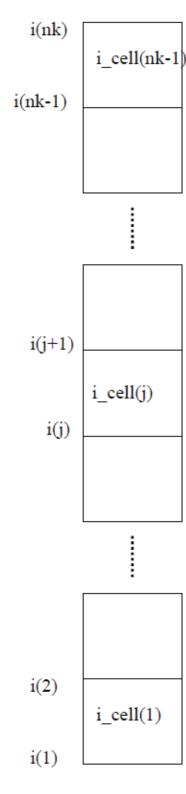
Task 5 Nodalization and numerical solution

- for j = 2 to nk
 - q_cell = q_cell * fR
 - $-i(j) = i(j-1) + q_cell(j-1) / W$ (energy balance)
- end for
- while p not converged
 - p(1) = pin + dpInletOrifice
 - for j = 2 to nk
 - xe(j), xa(j), alpha(j) (void fraction model)
 - dpf_cell(j-1), dpg_cell(j-1), dpa_cell(j-1), dpl_cell(j-1)
 - dp_cell(j-1) (pressure drop calculation)
 - $p(j) = p(j-1) + dp_cell(j-1)$
 - end for
- end while p
- T(j)
 - f(p(j), i(j)) for subcooled water
 - Tsat(j) for saturated water
- q2cr(j), xcr(j), DNBR, CPR

Task 6

- 1. Find design and regulatory limit of fuel/clad temperatures
- 2. Identify hot spots (usually in hot channel)
 - Fuel temperature
 - Fuel material properties (temperature dependent)
 - Peak temperature and location
 - Clad temperature
 - Clad material properties (temperature dependent)
 - Peak temperature and location





Task 6 Nodalization and numerical solution

- for j = 2 to nk
 - q_cell = q_cell * fR
 - $-i(j) = i(j-1) + q_cell(j-1) / W$ (energy balance)
- end for
- while p not converged
 - p(1) = pin + dpInletOrifice
 - for j = 2 to nk
 - xe(j), xa(j), alpha(j) (void fraction model)
 - dpf_cell(j-1), dpg_cell(j-1), dpa_cell(j-1), dpl_cell(j-1)
 - dp_cell(j-1) (pressure drop calculation)
 - $p(j) = p(j-1) + dp_cell(j-1)$
 - end for
- end while p
- T(j)
- Re(j), Pr(j), Nu(j), HTC(j), Tw(j)
- Tco(j), Tci(j), Tfo(j), Tfc(j)
- TciMax, TfcMax

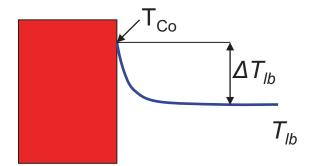
Clad-Coolant Heat Transfer in Channels with Single Phase Flows (1)

- In Light Water Reactors, coolant is sub-cooled at the inlet to the reactor core
- The subcooling is defined as the difference between the saturation temperature and the actual coolant bulk temperature: $\Delta T_{sub} = T_f T_{lb}$
- For example, if the inlet temperature and pressure of the water coolant are 549 K and 7 MPa, respectively, then the inlet subcooling is equal to 559 K – 549 K = 10 K, since the saturation temperature of water at 7 MPa pressure is equal to 559 K

Clad-Coolant Heat Transfer in Channels with Single Phase Flows (2)

• In the single-phase region, when $z_{in} < z < z_{ONB}$, the clad surface temperature T_{Co} of the heated wall and the liquid bulk temperature T_{lb} are related to each other as follows,

$$T_{Co} - T_{lb} \equiv \Delta T_{lb} = q''/h$$



• where h is the heat transfer coefficient and ΔT_{lb} is the temperature difference between the surface of the heated wall and the bulk liquid

*) z_{in} – inlet coordinate; ONB – Onset of Nucleate Boiling

Clad-Coolant Heat Transfer in Channels with Single Phase Flows (3)

- The heat transfer coefficient h is evaluated from correlations, which, in turn, are based on experimental data and are using the principles of the dimensionless analysis
- The following general relationships are employed

Nu =
$$f$$
 (Re, Pr, ...), where: Nu = $\frac{hD_h}{\lambda}$ Nusselt number Re = $\frac{GD_h}{\mu}$ Reynolds number, Pr = $\frac{c_p\mu}{\lambda}$, Pr_w = $\frac{c_p\mu}{\lambda}$ Prandtl number

Clad-Coolant Heat Transfer in Channels with Single Phase Flows (4)

 For flows in pipes, rectangular channel and annuli, and with 10⁴ < Re, 0.7 < Pr < 160 and L/D_h > 60, the following correlation can be used (Colburn):

$$Nu = 0.023 \cdot Re^{0.8} Pr^{0.33}$$

 Another correlation frequently used for heat transfer calculations in pipes was given by Dittus&Boelter:

$$Nu = 0.023 \cdot Re^{0.8} Pr^n$$
 n=0.4 for heating n=0.3 for cooling

valid for $L/D_h > 60$, Re $> 10^4$ and 0.7 < Pr < 100

Heat Transfer in Rod Bundles (1)

Heat transfer in the entire bundle is calculated from a single correlation including effects of:

flow conditions

fluid properties

geometry

Typically the correlation is of the form:

 $Nu = F(Re, Pr, D_h/d_r, p/d_r,...)$

Heat Transfer in Rod Bundles (2)

The influence of flow/fluid conditions and geometry factors can be separated:

Nu =
$$F_1(Re, Pr,...) \times F_2(D_h/d_r, p/d_r,...)$$

p – lattice pitch

d_r – rod diameter

Example: the Weisman (1959) correlation:

$$Nu = A \cdot Re^{0.8} Pr^{1/3}$$

$$A = \begin{cases} 0.026 \, p/d_r - 0.006 & \text{triangular } 1.1 < p/d_r < 1.5 \\ 0.042 \, p/d_r - 0.024 & \text{square } 1.1 < p/d_r < 1.3 \end{cases}$$

Heat Transfer in Rod Bundles (3)

 Subotin et al. (1975) recommended for heat transfer to liquids in bundles

$$Nu = A \cdot \text{Re}^{0.8} \text{ Pr}^{0.4}$$
 $A = 0.0165 + 0.02 \left[1 - \frac{0.91}{(p/d_r)^2} \right] \left(\frac{p}{d_r} \right)^{0.15}$

Triangular lattice with $1.1 < p/d_r < 1.8$; 1.0 < Pr < 20; $5.10^3 < Re < 5.10^5$

For gas flow in tight rod bundles Ajn and Putjkov (1964) give

$$\frac{Nu_{bundle}}{Nu_{DB}} = 1.184 + 0.351 \cdot \log_{10}(p/d_r - 1) \qquad \text{1.03
$$\text{Nu}_{DB} - \text{Dittus-Boelter correlation} \qquad \qquad \text{p-lattice pitch}$$$$

d_r – rod diameter

Heat Transfer in Rod Bundles (4)

- In the cited correlations it is assumed that the flow/fluid conditions and the geometry effect are separable
- This, however, seems not to be valid based on an extensive study done by Markoczy (1972)
- He suggested the following form of the correlation

$$Nu_{bundle} = Nu_{pipe} \times F_{geom}(p/d_r, Re, Pr)$$

In other words, the geometry effect is flow/propertydependent

Heat Transfer in Rod Bundles (5)

Markoczy (1972) performed study of experimental data (over 63 bundles of different geometry)

He proposed the following correlation:

$$\frac{\text{Nu}_{bundle}}{\text{Nu}_{DB}} = 1 + 0.91 \,\text{Re}^{-0.1} \,\text{Pr}^{0.4} \left(1 - 2e^{-B}\right) \quad B = \begin{cases} \frac{2\sqrt{3}}{\pi} \left(\frac{p}{d_r}\right)^2 - 1 & \text{triangular} \\ \frac{4}{\pi} \left(\frac{p}{d_r}\right)^2 - 1 & \text{square} \end{cases}$$

Validity region: 3 10³<Re<10⁶; 0.66<Pr<5; 1.02<p/d_r<2.5

Heat Transfer in Rod Bundles (6)

In summary, the bundle-wide approach is based on:
 base correlation, which typically takes into account dependence of
 the heat transfer coefficient on flow/property conditions
 geometry factor, which takes into account the dependence on
 pitch/rod-diameter

$$Nu_{bundle} = F_{geo}(p/d_r,...) \times Nu_{base}(Re, Pr, ...)$$

Heat Transfer in Rod Bundles (7)

- Occasionally another approach can be encountered in the literature:
- Osmachkin (1974) recommended to use a correlation valid for pipes (e.g. Dittus-Boelter), replacing the hydraulic diameter with the "effective" one:

$$D_{eff} = \frac{2}{(1-\varepsilon)^2} \left(\frac{\varepsilon - 3}{2} - \frac{\ln \varepsilon}{1-\varepsilon} \right) D_h$$

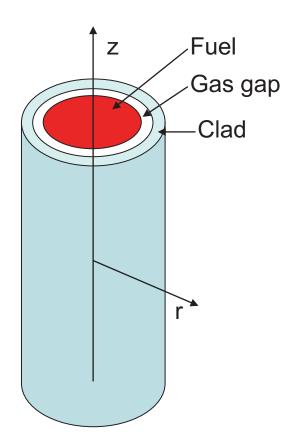
 ϵ – fraction of the bundle crosssection occupied by rods: ϵ = A_r/A_{tot}; A_r – rod cross-section area, A_{tot} – total (rod+coolant) cross section area

Heat conduction in reactor fuel elements (1)

 In the cylindrical coordinate system, for a fuel rod as shown in figure, the conduction equation can be written as

$$\nabla \cdot \lambda \nabla T = -q'''(\mathbf{r})$$

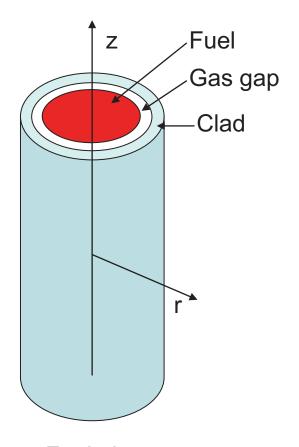
$$\frac{1}{r} \frac{\partial}{\partial r} \left(r \lambda \frac{\partial T(r,z)}{\partial r} \right) + \frac{\partial}{\partial z} \left[\lambda \frac{\partial T(r,z)}{\partial z} \right] = -q'''(r,z)$$



Fuel element

Heat conduction in reactor fuel elements (2)

- The conduction equation can be further simplified:
 - Heat conduction in the z-direction can be neglected, since temperature gradient dT/dz is much lower than dT/dr
 - In fuel region q''' = q'''(z)
 - In gas gap and clad regions q"=0



Fuel element

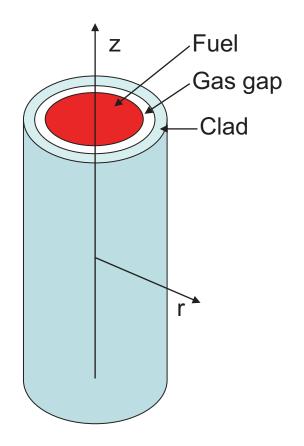
Heat conduction in reactor fuel elements (3)

 The conduction equation can be thus written for each region separately as:

- Fuel
$$\frac{1}{r}\frac{d}{dr}\left(r\lambda_F\frac{dT_F(r)}{dr}\right) = -q'''(z)$$

- Gap
$$\frac{1}{r}\frac{d}{dr}\left(r\lambda_G\frac{dT_G(r)}{dr}\right) = 0$$

- Clad
$$\frac{1}{r}\frac{d}{dr}\left(r\lambda_C\frac{dT_C(r)}{dr}\right) = 0$$

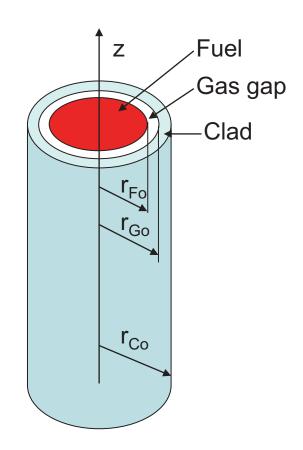


Fuel element

Heat conduction in reactor fuel elements (4)

- To solve the ordinary differential equations we need boundary conditions:
 - Finite temperature at r = 0
 - 4th kind b.c. at $r = r_{Fo}$ $T_F \Big|_{r=r_{Fo}} = T_G \Big|_{r=r_{Fo}} \lambda_F \frac{dT_F}{dr} \Big|_{r=r_{Fo}} = \lambda_G \frac{dT_G}{dr} \Big|_{r=r_{Fo}}$
 - 4th kind b.c. at $r = r_{Go}$ $T_G \Big|_{r=r_{Go}} = T_C \Big|_{r=r_{Go}} \lambda_G \frac{dT_G}{dr} \Big|_{r=r_{Go}} = \lambda_C \frac{dT_C}{dr} \Big|_{r=r_{Go}}$
 - 3rd kind b.c. at r = r_{Co}

$$-\lambda_C \left. \frac{dT_C}{dr} \right|_{r=r_{Co}} = h \left(T_{Co} - T_{lb} \right)$$



Fuel element

Heat conduction in reactor fuel elements (5)

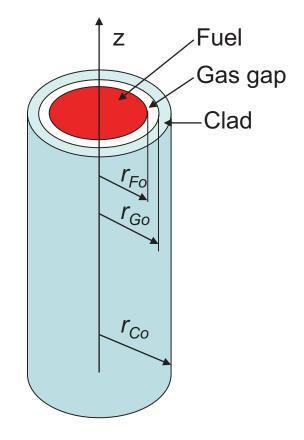
Solution in the fuel region

$$\frac{1}{r}\frac{d}{dr}\left(r\lambda_{F}\frac{dT_{F}(r)}{dr}\right) = -q'''(z)$$

$$\downarrow \qquad \qquad \downarrow \qquad \qquad \qquad \downarrow \qquad \qquad \qquad \downarrow \qquad \qquad \downarrow \qquad \qquad$$

- To limit $T_{Fc} = T_F(0)$, the constant C must be equal to zero: C = 0, thus

$$\lambda_F \frac{dT_F(r)}{dr} = -\frac{q'''(z) \cdot r}{2}$$



Fuel element

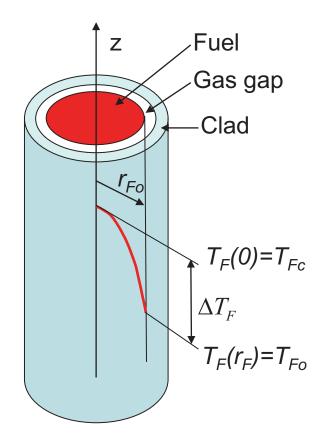
Heat conduction in reactor fuel elements (6)

 If the conductivity of the fuel material is assumed constant, the integration is straightforward as

$$T_F(r_F) - T_F(0) \equiv -\Delta T_F = -\int_0^{r_{Fo}} \frac{q'''(z) \cdot r}{2 \cdot \lambda_F} dr$$

or, after integration the temperature rise in fuel region is as follows

$$\Delta T_F(z) \equiv T_F(0) - T_F(r_F) = T_{Fc} - T_{Fo} = \frac{q'''(z) \cdot r_{Fo}^2}{4 \cdot \lambda_F}$$



Fuel element

Heat conduction in reactor fuel elements (7)

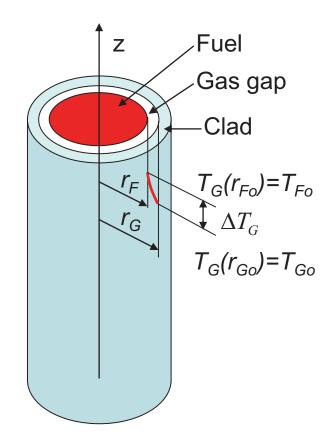
Solution in the gas gap

$$\frac{1}{r}\frac{d}{dr}\left(r\lambda_G\frac{dT_G(r)}{dr}\right) = 0$$

$$\lambda_G \frac{dT_G(r)}{dr} = \frac{C'}{r} \Rightarrow T_G(r) = \frac{C'}{\lambda_G} \ln(r) + C''$$

- Where C' and C" are constants
- Temperature drop in gap is

$$\Delta T_G \equiv T_G(r_{Fo}) - T_G(r_{Go}) = -\frac{C'}{\lambda_G} \ln \left(\frac{r_{Go}}{r_{Fo}}\right)$$



Fuel element

Heat conduction in reactor fuel elements (8)

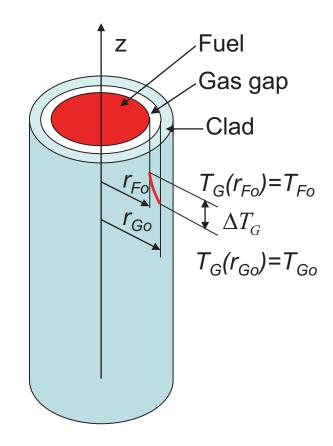
 The constant C' can be found from the energy balance at the fuel-gap interface:

$$q''|_{r_{Fo}} = -\lambda_{G} \frac{dT_{G}(r)}{dr}|_{r_{Fo}} = -\frac{C'}{r_{Fo}}$$

$$q''|_{r_{Fo}} \cdot 2\pi r_{Fo} \cdot dz = q''' \cdot \pi r_{Fo}^{2} \cdot dz$$

$$\Rightarrow C' = -\frac{q''' r_{Fo}^{2}}{2}$$

$$\Delta T_G = \frac{q''' r_{Fo}^2}{2\lambda_G} \ln \left(\frac{r_{Go}}{r_{Fo}}\right)$$



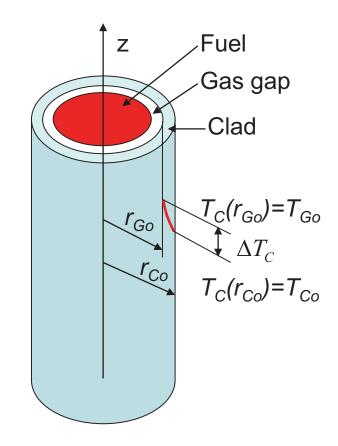
Fuel element

Heat conduction in reactor fuel elements (9)

 Since the conduction equation is the same in the clad region, the temperature rise in the clad is found as

$$\begin{aligned} q''|_{r_{Go}} &= -\lambda_{C} \frac{dT_{C}(r)}{dr} \Big|_{r_{Go}} = -\frac{C'}{r_{Go}} \\ q''|_{r_{Go}} \cdot 2\pi r_{Go} \cdot dz = q''' \cdot \pi r_{Fo}^{2} \cdot dz \end{aligned} \Rightarrow C' = -\frac{q''' r_{Fo}^{2}}{2}$$

$$\Delta T_C = \frac{q''' r_{Fo}^2}{2\lambda_C} \ln \left(\frac{r_{Co}}{r_{Go}} \right)$$



Fuel element

Heat conduction in reactor fuel elements (10)

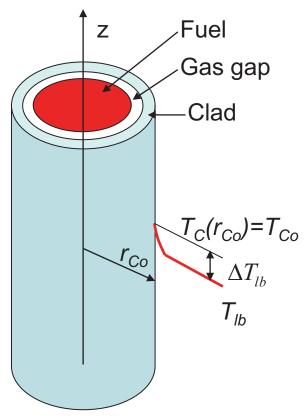
 Finally, the temperature rise in the thermal boundary layer in coolant can be found from the Newton equation for the convective heat transfer:

$$q''|_{r_{Co}} = h \cdot (T_{Co} - T_{lb}) = h \cdot \Delta T_{lb}$$

since
$$q''|_{r_{Co}} \cdot 2\pi r_{Co} \cdot dz = q''' \cdot \pi r_{Fo}^2 \cdot dz \Rightarrow q''|_{r_{Co}} = \frac{q''' r_{Fo}^2}{2r_{Co}}$$

thus

$$\Delta T_{lb} = \frac{q''' r_{Fo}^2}{2r_{Co}h}$$



Fuel element

Heat conduction in reactor fuel elements (11)

 The total temperature rise in the fuel element is thus

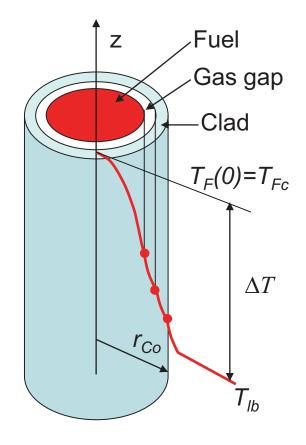
$$\Delta T = \Delta T_F + \Delta T_G + \Delta T_C + \Delta T_{lb} = T_{Fc} - T_{lb}$$

$$\Delta T = \frac{q'''r_{Fo}^2}{4\lambda_F} + \frac{q'''r_{Fo}^2}{2\lambda_G} \ln\left(\frac{r_{Go}}{r_{Fo}}\right) + \frac{q'''r_{Fo}^2}{2\lambda_C} \ln\left(\frac{r_{Co}}{r_{Go}}\right) + \frac{q'''r_{Fo}^2}{2r_{Co}h} =$$

$$\frac{q'''r_{Fo}^2}{4} \left[\frac{1}{\lambda_F} + \frac{2}{\lambda_G} \ln \left(\frac{r_{Go}}{r_{Fo}} \right) + \frac{2}{\lambda_C} \ln \left(\frac{r_{Co}}{r_{Go}} \right) + \frac{2}{r_{Co}h} \right]$$

Since $q'''\pi r_{F_0}^2 = q'$ (linear power density)

$$\Delta T = \frac{q'}{4\pi} \left[\frac{1}{\lambda_F} + \frac{2}{\lambda_G} \ln \left(\frac{r_{Go}}{r_{Fo}} \right) + \frac{2}{\lambda_C} \ln \left(\frac{r_{Co}}{r_{Go}} \right) + \frac{2}{r_{Co}h} \right]$$



Fuel element

Fuel Thermal Analysis

 For solid UO₂ with 95% density the recommended equation for the thermal conductivity is

$$\lambda_F(T) = \frac{100}{7.5408 + 17.692t + 3.6142t^2} + \frac{6400}{t^{5/2}} \exp\left(-\frac{16.35}{t}\right)$$

• where λ_F is in W/m·K, t = T/1000 and T is temperature in K. For porosity different from 5%, the thermal conductivity is found as:

$$\lambda_0 = \frac{\lambda_p}{1 - (2.6 - 0.5t)p}$$
 Here λ_0 is the thermal conductivity of fully dense UO₂ (that is p = 0) and λ_p is the thermal conductivity of UO₂ with porosity p.

$$\lambda_p = \lambda_0 \left[1 - \left(2.6 - 0.5t \right) p \right] = \lambda_F(T) \frac{1 - \left(2.6 - 0.5t \right) p}{1 - \left(2.6 - 0.5t \right) 0.05}$$

Fuel Thermal Analysis

 For mixed oxide fuel (80% U, 20% Pu) at 95% theoretical density and O/M (oxigen/metal)=2.0, the fuel thermal conductivity can be given as (Washington, 1973)

$$\lambda_F(T) = (0.042 + 2.71 \times 10^{-4} T)^{-1} + 6.9 \times 10^{-11} T^3$$

where λ_F is in W/m·K and T in K. For porosity different from 5%, the thermal conductivity is found as:

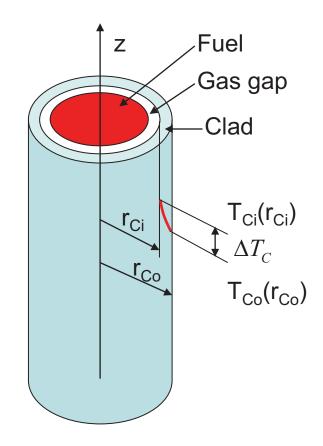
$$\lambda_{Fp}(T) = \begin{cases} \lambda_F(T) \frac{1 - 2.5p}{0.875} & p \le 0.1\\ \lambda_F(T) \frac{1 - p}{0.875(1 + 2p)} & p > 0.1 \end{cases}$$

Cladding Thermal Analysis

 For Zircaloy-2 and Zircaloy-4 (αphase), the thermal conductivity can be found as

$$\lambda_C = 12.6 + 0.0118T$$

- here: T [°C] temperature, λ_C [W/mK] thermal conductivity
- valid for 20 < T < 800 °C
- uncertainty ±1.01 W/mK



Fuel element