# SA1: Interim Report 2

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### 1 Exercise 1

```
Listing 1: ueintbit.m
function f = ueintbit(xa, ua, xb, ub)
%calculates the integral part of Eq10 in handout
%Thwaite's solution of the momentum integral equation
%Assumes a linear variation in ue
    %Calculate u_{-}mean and delta x and u to simplify later
         equations
    u_{mean} = (ua+ub)*0.5;
    deltau = ub-ua;
    deltax = xb-xa;
    f = (u_mean^5 + 5/6 *u_mean^3*deltau^2 + 1/16 *u_mean
       *deltau^4)*deltax;
end
                      Listing 2: Exercise 1
clear
close all
%initialise basic parameters
ReL = 1e3;
npoints = 101;
x = linspace(0,1,npoints); \%actually x/L
ue = ones(1, npoints); %actually ue/U
theta = zeros(1, npoints); \%actually theta/L
I10 = 0; %integral part of equation 10 in handout
```

```
for j = 2: npoints %sum contributions to integral in
   equation 10 over all points
    intbit = ueintbit(x(j-1), ue(j-1), x(j), ue(j));
    I10 = I10 + intbit;
    theta(j) = \mathbf{sqrt}(I10*0.45/ReL/(ue(j)^6)); \% calculate
       mom thickness at each point
end
theta_blasius = 0.664/(ReL^0.5) * x.^0.5; \%Blasius
   solution for momentum thickness
figure("Name", "Momentum Thickness ue=U");
plot(x, theta,"-", "Color","r","LineWidth",1)
hold on
plot(x, theta_blasius, "-", "Color", "[0 0 1]", "LineWidth
   ", 1)
hold off
set (gca, 'Fontn', 'Times', 'FontSize', 12, 'linewidth', 1)
xlabel('x/L')
ylabel ('\theta/L')
legend("Thwaites","Blasius","Location","northwest")
\%print - deps2c \ fig\_ex1\_mom\_thickness.eps
```

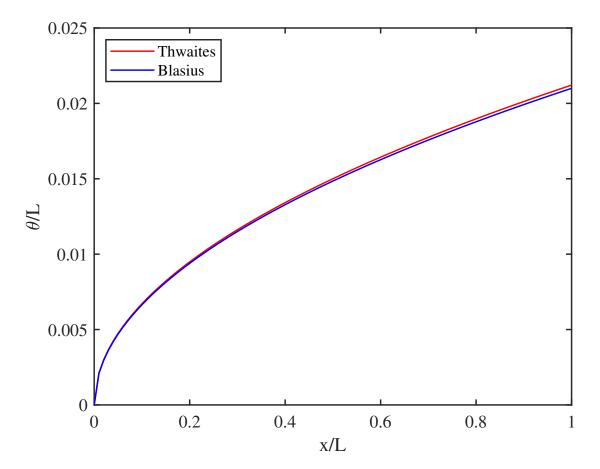


Figure 1: The momentum thickness,  $\theta$  of a laminar, zero pressure gradient boundary layer, from Thwaites' method and the exact Blasius solution

### 2 Exercise 2

clear close all

```
Listing 3: Exercise 2
```

```
theta = zeros(1, npoints);
laminar = true; % initializes boundary layer state flag
I10 = 0; %integral part of eq10 in handout
while laminar && i < npoints
    i = i + 1;
    intbit = ueintbit(x(i-1), ue(i-1), x(i), ue(i));
    I10 = I10 + intbit;
    theta(i) = sqrt(I10*0.45/ReL/((ue(i))^6));
    m = -ReL*(theta(i))^2*vel_grad;
    He = laminar_He(thwaites_lookup(m));
    Rethet = ReL*ue(i)*theta(i);
    \% test for laminar to turbulent transition
    if \log(\text{Rethet}) >= 18.4*\text{He}-21.74
        laminar = false;
        disp ([x(i) Rethet /1000])
    end
end
```

#### 2.1 Transition locations and $Re_{\theta}$ values

$Re_L [x10^6]$	$\frac{d(u_e/U)}{d(x/L)}$	Transition Location $\frac{x}{L}$	$Re_{\theta} [x10^3]$ at transition
	-0.1	0.48	1.08
5	0.0	0.74	1.29
	0.1	-	-
	-0.1	0.29	1.17
10	0.0	0.37	1.29
	0.1	0.55	1.51
	-0.1	0.17	1.25
20	0.0	0.19	1.31
	0.1	0.22	1.38

Table 1: Laminar to turbulent transition locations

### 3 Exercise 3

Listing 4: Exercise 3

clear

#### close all

```
%initialise basic parameters
ReL = 8.957e5;
%velocity\ gradient\ of\ d(ue/U)/d(x/L)
vel_grad = -0.25; %-0.25 for this case
npoints = 101;
x = linspace(0,1,npoints); \%actually x/L
ue = linspace(1,1+vel\_grad, npoints); \%acctually ue/U
theta = zeros(1, npoints); \%acctually theta/L
laminar = true; % initializes boundary layer state flag
i = 1;
int = 0; %natural transition point
ils = 0; %Laminar\ separation
I10 = 0; %equation 10 in handout
while laminar && i < npoints
    i = i + 1;
    intbit = ueintbit(x(i-1), ue(i-1), x(i), ue(i));
    I10 = I10 + intbit;
    theta(i) = sqrt(I10*0.45/ReL/(ue(i)^6));
    m = -ReL*(theta(i))^2*vel_grad;
    He = laminar_He(thwaites_lookup(m));
    Rethet = ReL*ue(i)*theta(i);
    %test for laminar to turbulent transition
    if \log(\text{Rethet}) >= 18.4*\text{He}-21.74
        laminar = false;
        %disp([x(i) (Rethet/1000])
        int = i;
    elseif m >= 0.09 \% test for laminar separation
        laminar = false;
        ils = i;
    end
end
if int = 0
    disp(['Natural transition at 'num2str(x(int))' with
        - Rethet - ' num2str(Rethet)])
end
if ils \tilde{}=0
    disp(['Laminar-seperation-at-' num2str(x(ils)) '-with
       Rethet - ' num2str(Rethet)])
end
```

#### 3.1 Laminar separation locations

$Re_L$	$\frac{d(u_e/U)}{d(x/L)}$	Separation Location $\frac{x}{L}$
$10^{3}$	-0.25	0.50
$10^{4}$	-0.25	0.50
$10^{5}$	-0.25	0.50

Table 2: Laminar separation locations

Transition supplants separation at  $Re_L = 9.0 \times 10^5$ , with  $\frac{d(u_e/U)}{d(x/L)} = -0.25$ .

#### 4 Exercise 4

```
Listing 5: thickdash.m
function dthickdx = thickdash(xmx0, thick)
    \%global variables used by thickdash
    global ReL ue0 vel_grad %#ok<GVMIS>
    theta = thick(1);
    de = thick(2);
    ue = ue0 + vel_grad*xmx0;
    Retheta = ReL*ue*theta;
    He = de/theta;
    if He >= 1.46
        H = (11*He+15)/(48*He-59);
    else
        H = 2.803;
    \mathbf{end}
    c_f = 0.091416*((H-1)*Retheta)^-0.232*exp(-1.26*H);
    c_{diss} = 0.010024*((H-1)*Retheta)^{-}(1/6);
    dthickdx = [(c_f/2 - (H+2)/ue*vel_grad*theta); (
       c_diss - 3/ue*vel_grad*de)];
```

Listing 6: Exercise 4

clear close all

```
\%global variables used by thickdash
global ReL ue0 vel_grad %#ok<GVMIS>
ue0 = 1.0;
vel_grad = 0.0;
ReL = 1e7;
%initialise turbulent boundary layer at first point
x0 = 0.01;
thick0(1) = 0.023*x0*(ReL*x0)^(-1/6);
thick0(2) = 1.83*thick0(1);
[delx, thickhist] = ode45(@thickdash, [0 0.99], thick0);
\% solve\ ODE\ (15)\ from\ handout\ up\ to\ x\!=\!\!L
x = x0 + delx;
theta7 = 0.037*x.*(ReL*x).^(-1/5); \%1/7th power law
   approximation
theta9 = 0.023*x.*(ReL*x).^(-1/6); \%1/9th power law
   approximation
figure ("Name", "Comparisons of Theta");
plot(x, thickhist(:,1),"-", "Color","r","LineWidth",1)
hold on
set (gca, 'Fontn', 'Times', 'FontSize', 12, 'linewidth', 1)
xlabel('x/L')
ylabel('\theta/L')
legend("\theta/L calculated","1/7th power estimate","1/9
   th power estimate", "Location", "northwest")
\%print\ -deps2c\ fig\_ex4\_mom\_thickness\_3methods.eps
```

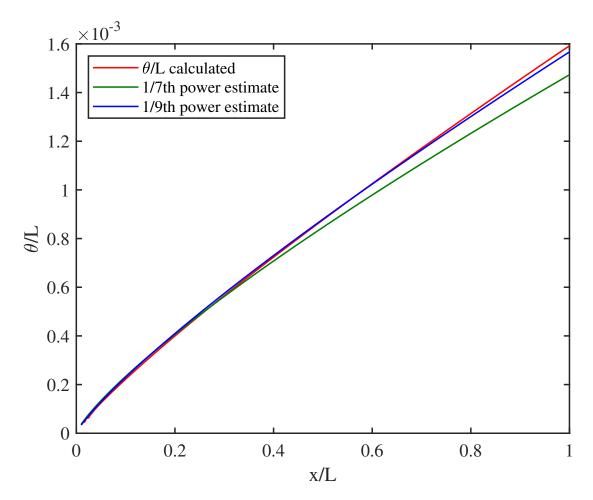


Figure 2: Momentum thickness, zero pressure gradient turbulent boundary layer

## 5 Exercise 5

```
Listing 7: Exercise 5

clear

close all

%global variables used by thickdash

global ReL ue0 vel_grad %#ok<GVMIS>

ue0 = 1.0;

vel_grad = -0.5;

ReL = [1e6, 1e7, 1e8];
```

```
ue0 = 1.0;
vel_grad = -0.5;
ReL = 1e7;
%initialise turbulent boundary layer at first point
x0 = 0.01:
thick0(1) = 0.023*x0*(ReL*x0)^(-1/6);
thick0(2) = 1.83*thick0(1);
[delx, thickhist] = ode45(@thickdash, [0 0.99], thick0);
\% solve\ ODE\ (15)\ from\ handout\ up\ to\ x\!=\!\!L
x = x0 + delx;
theta7 = 0.037*x.*(ReL*x).^(-1/5); \%1/7th power law
   approximation
theta9 = 0.023*x.*(ReL*x).^(-1/6); \%1/9th power law
   approximation
He = thickhist(:, 2)./thickhist(:, 1);
HeSep = 1.46.*ones(1, length(delx)); %turbulent separation
    below He=1.46
seploc = 0;
%check for turbulent separation
for i = 30:length(delx) %ignore numerical errors in first
    few points
%use plot below to decide size of loop
    if He(i-1) > 1.46 && He(i) < 1.46
        seploc = x(i);
        %could do linear interpolation for extra accuracy
        %less effect as number of points is increased
        {\bf break}~\% no~point~checking~further~downstream
            values
    end
end
if seploc = 0
    seploc %#ok<NOPTS>
end
figure ("Name", "H_e vs x/L");
plot (x, He," -", "Color"," r"," LineWidth",1)
```

```
hold on
plot(x, HeSep,"-", "Color", "[0 0 1]", "LineWidth",1)
hold off
set (gca, 'Fontn', 'Times', 'FontSize', 12, 'linewidth', 1)
xlabel('x/L')
ylabel('H_e')
figure("Name", "Comparisons of Theta and d_e");
plot(x, thickhist(:,1),"-", "Color","r","LineWidth",1)
hold on
{f plot}({f x}, {f thickhist}(:,2), "-", "Color", "[0 0.5 0]","
    LineWidth",1)
hold off
\mathbf{set}\,(\mathbf{gca}\,,\,\mathrm{'Fontn'}\,,\,\mathrm{'Times'}\,,\,\mathrm{'FontSize'}\,,12\,,\,\mathrm{'linewidth'}\,,1)
xlabel('x/L')
ylabel(' \land theta/L, \land delta_e/L')
legend ("\theta/L","\delta_e/L"," Location"," northwest")
\%print\ -deps2c\ fig\_ex5\_theta\_delta\_e.eps
```

#### 5.1 Turbulent separation locations

$Re_L$	$\frac{d(u_e/U)}{d(x/L)}$	Separation Location $\frac{x}{L}$
$10^{6}$	-0.5	0.895
$10^{7}$	-0.5	0.989
$10^{8}$	-0.5	-
$10^{7}$	-0.25	-
$10^{7}$	-0.95	0.523

Table 3: Turbulent separation locations

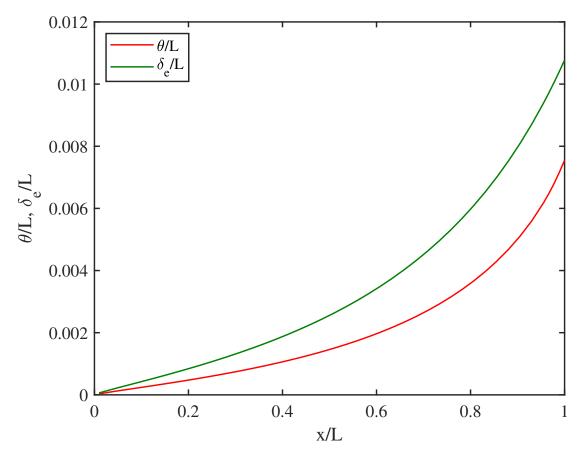


Figure 3:  $\frac{\theta}{L}$  and  $\frac{\delta}{L}$  against  $\frac{x}{L}$ , for a turbulent boundary layer at  $Re_L=10^7$ , and  $\frac{d(u_e/U)}{d(x/L)}=-0.5$ .

### 6 Exercise 6

#### 6.1 Flowchart

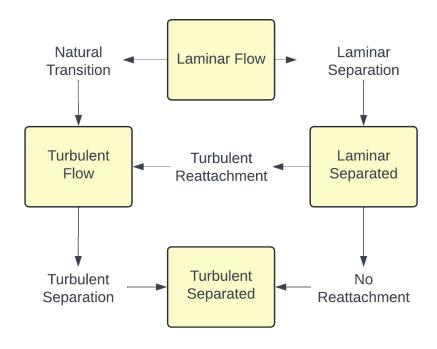


Figure 4: Boundary layer evolution flowchart

#### 6.2 Script

```
Listing 8: Exercise 6

clear
close all

npoints = 101;

global ReL ue0 vel_grad %#ok<GVMIS>
%initialise basic parameters
ReL = 1e5;
%velocity gradient of d(ue/U)/d(x/L)
vel_grad = -0.381;

x = linspace(0,1,npoints); %actually x/L
```

```
ue = linspace(1,1+vel_grad, npoints); %acctually ue/U
theta = zeros(1, npoints); \%acctually theta/L
He = 1.57258*ones(1, npoints);
laminar = true; % initializes boundary layer state flag
i = 1;
int = 0; %natural transition point
ils = 0; %Laminar\ separation
itr = 0; %Turbulent reattachement
its = 0; %Turbulent separation
I10 = 0; %equation 10 in handout
while laminar && i < npoints
    i = i + 1;
    intbit = ueintbit(x(i-1), ue(i-1), x(i), ue(i));
    I10 = I10 + intbit;
    theta(i) = sqrt(I10*0.45/ReL/(ue(i)^6));
    m = -ReL*(theta(i))^2*vel\_grad;
    He(i) = laminar_He(thwaites_lookup(m));
    Rethet = ReL*ue(i)*theta(i);
    %test for turbulence
    if \log(\text{Rethet}) >= 18.4*\text{He}(i) - 21.74
        laminar = false;
        %disp([x(i) (Rethet/1000])
        int = i;
    elseif m >= 0.09
        laminar = false;
        ils = i;
        He(i) = 1.51509;
    end
end
de = He(i) * theta(i);
while its == 0 && i < npoints
    thick0 = [theta(i), de];
    ue0 = ue(i);
    i = i + 1;
```

```
[delx, thickhist] = ode45(@thickdash, [0, x(i)-x(i-1)],
        thick0);
    theta(i) = thickhist(end, 1);
    de = thickhist(end, 2);
    He(i) = de/theta(i);
    %disp(He(i))
    if He(i) >= 1.58 && ils \tilde{}= 0 && itr == 0 % Turbulent
        reattachent
        itr = i;
    elseif He(i) < 1.46 % Turbulent seperation
        its = i;
    end
end
He(i:npoints) = He(i);
while its \tilde{}=0 && i < npoints
    i = i + 1:
    theta(i) = theta(its)*(ue(its)/ue(i))^(2.803+2);
end
if int = 0
    disp(['Natural-transition-at-' num2str(x(int)) '-with
        - Rethet - ' num2str(ReL*ue(int)*theta(int))])
end
if ils \tilde{}=0
    disp(['Laminar-seperation-at-' num2str(x(ils)) '-with
        -Rethet-' num2str(ReL*ue(ils)*theta(ils))])
end
if itr = 0
    disp(['Turbuent-Reattachment-at-' num2str(x(itr)) '-
        with Rethet 'num2str(ReL*ue(itr)*theta(itr))])
end
if its = 0
    \mathbf{disp}(['Turbulent \cdot seperation \cdot at \cdot 'num2str(x(its))']
        with Rethet 'num2str(ReL*ue(its)*theta(its))])
end
\%save\ 6-025.mat\ -v7.3\ \%save\ data\ to\ a\ relevant\ file
%example plotting
```

```
figure("Name", "\theta/L vs x/L");
plot(x,theta,"-", "Color","r","LineWidth",1)
set(gca,'Fontn','Times','FontSize',12,'linewidth',1)
xlabel('x/L')
ylabel('\theta/L')

figure("Name", "H_e vs x/L");
plot(x,He,"-", "Color","r","LineWidth",1)
set(gca,'Fontn','Times','FontSize',12,'linewidth',1)
xlabel('x/L')
ylabel('H_e')
```

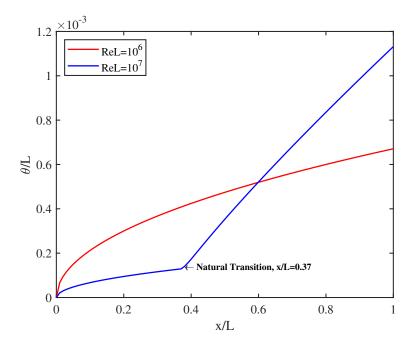


Figure 5:  $\frac{\theta}{L}$  against  $\frac{x}{L}$ , for  $\frac{d(u_e/U)}{d(x/L)} = 0$ 

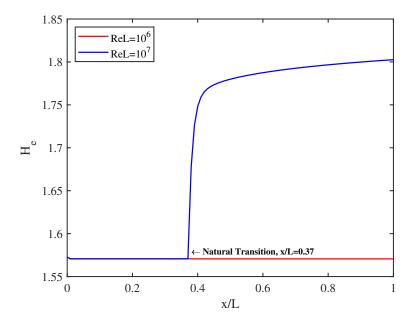


Figure 6:  $H_e$  against  $\frac{x}{L}$ , for  $\frac{d(u_e/U)}{d(x/L)} = 0$ 

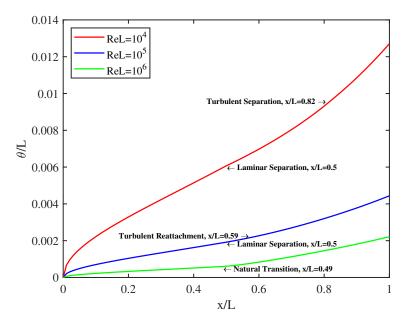


Figure 7:  $\frac{\theta}{L}$  against  $\frac{x}{L},$  for  $\frac{d(u_e/U)}{d(x/L)}=-0.25$ 

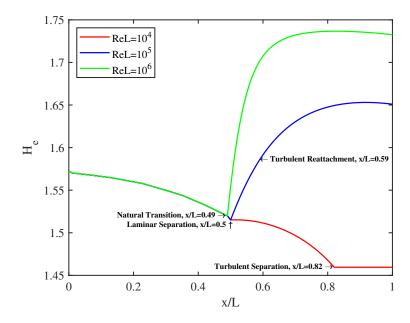


Figure 8:  $H_e$  against  $\frac{x}{L}$ , for  $\frac{d(u_e/U)}{d(x/L)} =$  -0.25

## 6.3 Critical Velocity Gradient

At  $Re_L=10^5, \ \frac{d(u_e/U)}{d(x/L)}=$  -0.38 gives turbulent separation at x=L.