

Danmarks Tekniske Universitet / Technical University of Denmark

Written examination:

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Course title: Wind Turbine Aeroelasticity

Course number: 46310

Aids allowed: All

Exam duration: 4 hours

Weighting: Indicated at the questions

**Q#1:** (30%) Use your script for the 5DOF system including controller and set wind speed to 8 m/s (turbulent). Specify the initial rotational speed to  $\omega=0.7$  rad/s and the pitch angle as  $\theta_p=0$  degrees. Assume that the generator torque is lost at  $t=100$ s so that the generator torque suddenly becomes  $M_G=0$  Nm. For a maximum allowable pitch angle of 45 degrees plot as function of time the rotational speed, the pitch angle, the thrust and the tip deflection in both  $y$  and  $z$  direction. Also make a PSD plot of the time history of the tip deflection in the  $z$ -direction after it has settled to something periodic. Explain from the graphs what is happening.

**Q#2:** (30%) Now assume that the pitch actuator is malfunctioning so that the pitch remains at  $\theta_p=0$  degrees. Simulate again the sudden loss of the generator at  $t=100$  s. Plot  $\omega(t)$ , thrust  $T(t)$ , flapwise bending moment  $M_y(t)$  and tip deflection in  $z$  -direction. Compare with the results from Q#1.

**Q#3:** (20%) A simple 2 DOF mechanical system for a wind turbine blade is shown in Figure 1. The first part of the blade is stiff and rigidly clamped. At  $x=L$  and  $x=2L$  there is a torsional spring of stiffness  $k_1$  and  $k_2$ , respectively. All the mass in the beam is concentrated at  $x=1.5L$  and  $x=2.5L$ . Define the DOFs and derive the mass and stiffness matrix.

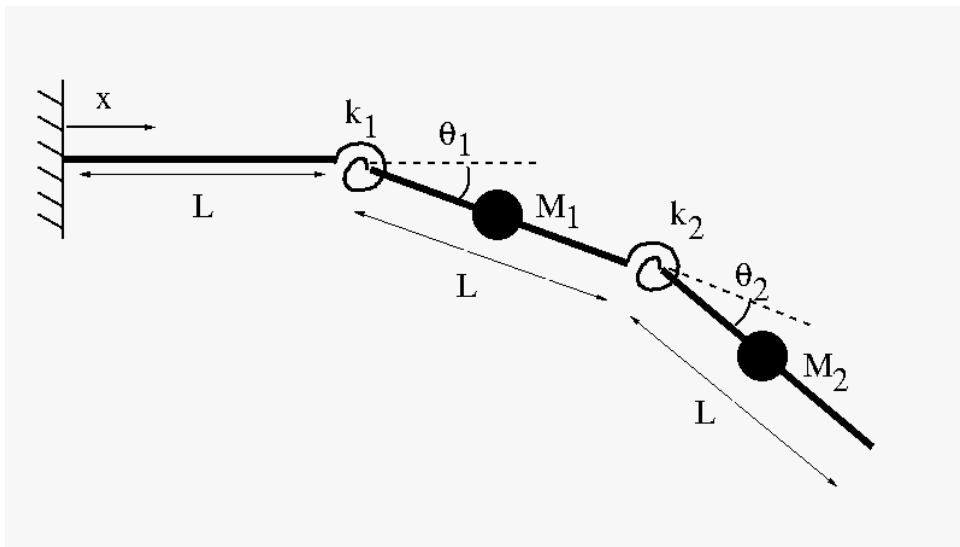


Figure 1: A simple 2-DOF mechanical system simulating a cantilever beam

**Q#4:** (20%) To experimentally simulate the dynamics and aerodynamic damping of edgewise blade vibrations an airfoil is put in a wind tunnel and suspended in a spring that allows the airfoil to vibrate in the  $x$ -direction only, see Figure 2. The airfoil is mounted at a geometrical attack of  $\alpha_g$  and has a mass of  $M$ , a chord of  $c$  and a width of  $b$ . The airfoil data are known as  $C_l(\alpha)$  and  $C_d(\alpha)$ . Set up the equation of motion and explain how it can numerically be solved.

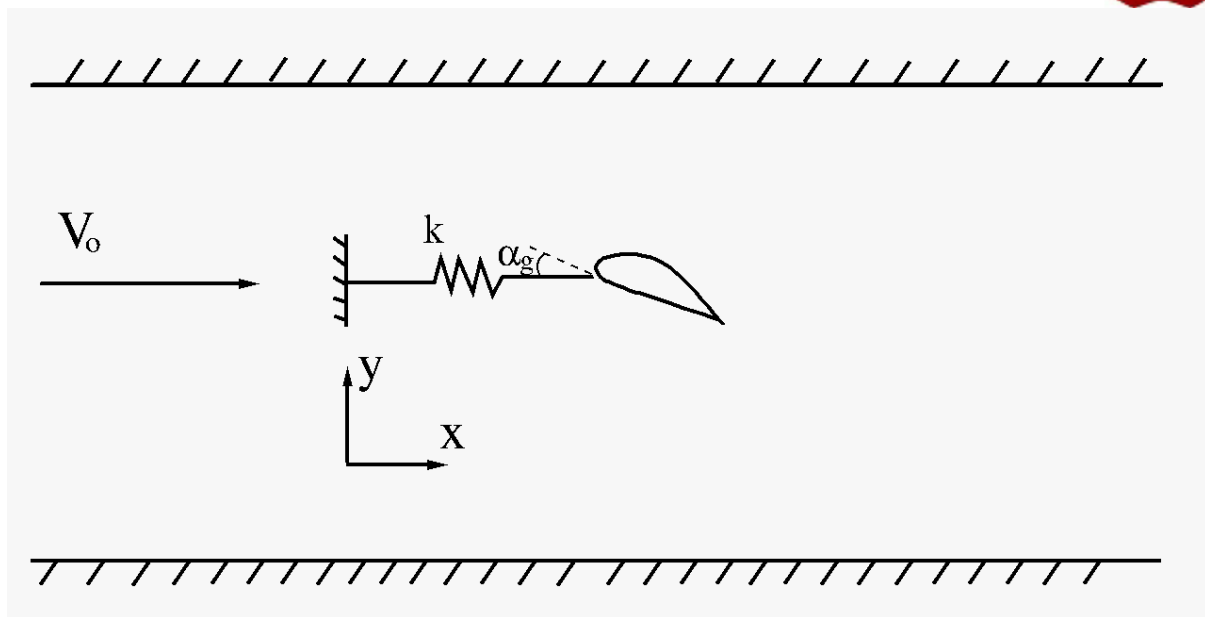


Figure 2: An airfoil with known airfoil data  $C_l(\alpha)$  and  $C_d(\alpha)$  of chord  $c$  and  $a$  with  $w=b$  is suspended in a spring in a wind tunnel and free to vibrate in the  $x$ -direction.

The wind velocity is  $\begin{pmatrix} V_{o,x} \\ V_{o,y} \end{pmatrix} = \begin{pmatrix} V_o \\ 0 \end{pmatrix}$