2_planetary_motion

December 19, 2023

1 Planetary Motion

```
[9]: import numpy as np
     from scipy.integrate import odeint
     import matplotlib.pyplot as plt
     from matplotlib import animation
     %matplotlib inline
     %config InlineBackend.figure_format = 'retina'
     # default values for plotting
     plt.rcParams.update({'font.size': 12,
                           'axes.titlesize': 18,
                           'axes.labelsize': 16,
                           'axes.labelpad': 14,
                           'lines.linewidth': 1,
                           'lines.markersize': 10,
                           'xtick.labelsize' : 16,
                           'ytick.labelsize' : 16,
                           'xtick.top' : True,
                           'xtick.direction' : 'in',
                           'ytick.right' : True,
                           'ytick.direction' : 'in',})
```

[9]: <IPython.core.display.HTML object>

1.1 Physical Model

From the above defined equation of motion for the spring pendulum, it is only a small step to simulate planetary motion, which you should know well from you mechanics lectures. The equations of motion in angular and radial direction can be obtained very similarly. Here, however, there is no force in the tangential direction as we deal with the central symmetric gravitational potential. The equations of motion read:

$$\ddot{r} = r\dot{\theta}^2 - \frac{GM}{r^2} \tag{1}$$

$$\ddot{\theta} = -\frac{1}{r}2\dot{r}\dot{\theta} \tag{2}$$

We know the resulting trajectory of this motion

$$r(\theta) = \frac{p}{1 + \epsilon \cos(\theta)}$$

with

$$p = \frac{L^2}{GMm^2} \tag{3}$$

$$\epsilon = \sqrt{1 + \frac{2\frac{E}{m}\frac{L^2}{m^2}}{G^2M^2}}\tag{4}$$

The trajectory is therefore determined by p and the excentricity ϵ . For $0 < \epsilon < 1 (E < 0)$ there is a closed orbit with an elliptical shape. For $\epsilon = 0$ the orbit is circular.

```
[16]: def planetary_motion(state, time ):
    g0 = state[1]
    g1 = state[0]*state[3]**2 - G*M/(state[0]**2)
    g2 = state[3]
    g3 = -2.0*state[1]*state[3]/state[0]
    return np.array([g0, g1, g2, g3])
```

1.2 Numerical Solution

1.2.1 Initial Parameters: Planets

```
state[0]=r_o
state[1]=v_o
state[2] = theta_o
state[3] = omega_o

time = np.linspace(0, 3, N)
```

1.2.2 Solution: Planets

```
[22]: answer = odeint ( planetary_motion , state , time )

xdata = answer[:,0]*np.cos(answer[:,2])
ydata = answer[:,0]*np.sin(answer[:,2])
```

```
[23]: # ellipse parameters

L=m*r_o**2*omega_o # angular momentum
E=0.5*m*(v_o**2+r_o**2*omega_o**2)-G*M*m/r_o

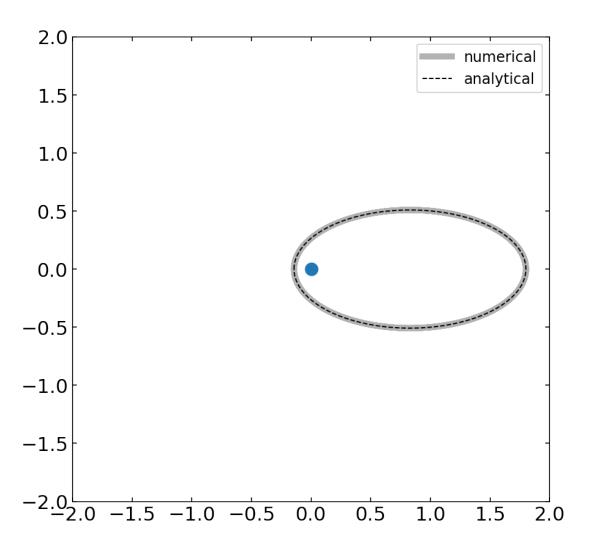
p=(L/m)**2/(G*M)
e=np.sqrt(1+(2*E*L**2/(m**3))/(G*G*M*M))
```

1.2.3 Plotting: Planets

Trajectory

```
[25]: # analytical solution
    theta=np.linspace(0,2*np.pi,1000)
    r=p/(1+e*np.cos(theta))

fig=plt.figure(1, figsize = (7,7))
    plt.plot(xdata,ydata,'k-',lw=5,alpha=0.3,label='numerical')
    plt.plot(-r*np.cos(theta),r*np.sin(theta),'k--',lw=1,label='analytical')
    plt.plot(0,0,'o')
    plt.xlim(-2,2)
    plt.ylim(-2,2)
    plt.legend()
    plt.show()
```



```
Energy
```

```
[7]: Etot=0.5*m*(answer[:,1]**2+answer[:,0]**2*answer[:,3]**2)-G*M*m/answer[:,0]
    Ekin=0.5*m*(answer[:,1]**2+answer[:,0]**2*answer[:,3]**2)
    Epot=-G*M*m/answer[:,0]

[27]: plt.plot(time,Ekin)
    plt.plot(time,Epot)
    plt.plot(time,Etot)
    plt.ylabel('time')
    plt.ylabel('energy')
    plt.ylabel('energy')
    plt.ylim(-100,100)
    plt.show()
```

