

## ELECTROMECHANICAL ACTUATOR - ADJUSTMENT/TEST

*EFFECTIVITY: AIRCRAFT WITH ELECTROMECHANICAL GUST LOCK*

### 1. General

- A. This section gives the procedures to do the functional check of the Electromechanical Gust Lock Actuator.
- B. The procedures in this section are given in the sequence below. The tasks identified with (◆) are part of the Scheduled Maintenance Requirements Document (SMRD).

TASK NUMBER	DESCRIPTION	EFFECTIVITY
27-71-01-700-801-A	ELECTROMECHANICAL ACTUATOR - FUNCTIONAL CHECK	AIRCRAFT WITH ELECTROMECHANICAL GUST LOCK

TASK 27-71-01-700-801-A

*EFFECTIVITY: AIRCRAFT WITH ELECTROMECHANICAL GUST LOCK*

## 2. ELECTROMECHANICAL ACTUATOR - FUNCTIONAL CHECK

### A. General

- (1) This task gives the procedures to do a check of the load on the Electromechanical Gust Lock Actuator due to friction forces in the Gust Lock mechanism.

### B. References

REFERENCE	DESIGNATION
<a href="#">AMM TASK 20-13-02-000-801-A/400</a>	RELAYS - REMOVAL (TYPICAL)
<a href="#">AMM TASK 20-13-02-400-801-A/400</a>	RELAYS - INSTALLATION (TYPICAL)
<a href="#">AMM TASK 20-40-01-860-801-A/200</a>	ENERGIZATION OF THE AIRCRAFT WITH AN EXTERNAL POWER SOURCE
<a href="#">AMM TASK 55-36-00-000-801-A/400</a>	TAIL BOOM - REMOVAL
<a href="#">AMM TASK 55-36-00-400-801-A/400</a>	TAIL BOOM - INSTALLATION
FIM TASK 27-70-00-810-808-A	-
<a href="#">ITEM GSE 050</a>	MULTIMETER - DIGITAL

### C. Zones and Accesses

ZONE	PANEL/DOOR	LOCATION
321	321	Tail boom

### D. Tools and Equipment

ITEM	DESCRIPTION	PURPOSE	QTY
<a href="#">GSE 419</a>	Gust Lock Actuator Test Harness	To adapt the multimeter to the Gust Lock Connectors	
<a href="#">GSE 050</a>	Multimeter - digital	To do the check	
<a href="#">GSE 044</a>	Head Set	For communications	

### E. Auxiliary Items

Not Applicable

### F. Consumable Materials

Not Applicable

### G. Expandable Parts

Not Applicable

### H. Persons Recommended

QTY	FUNCTION	PLACE
1	Does the task	Horizontal stabilizer
1	Helps the other technician	Cockpit

I. Preparation

SUBTASK 841-002-A

**WARNING: MAKE SURE THAT THERE ARE NO PERSONS OR EQUIPMENT IN THE ELEVATOR TRAVEL AREA.**

- (1) Make sure that the aircraft is safe for maintenance.
- (2) Do not do other tasks in the horizontal stabilizer, elevators and rudders at this time.
- (3) Energize the aircraft with the External DC Power Supply ( [AMM TASK 20-40-01-860-801-A/200](#)).
- (4) (WITH GSE-419) Remove rear fairings 321 of the horizontal stabilizer. Refer to [AMM TASK 55-36-00-000-801-A/400](#).
- (5) Make sure that the electromechanical gust-lock system is in the locked position.
- (6) On the Circuit Breaker Panel, open the GUST-LOCK circuit breaker and attach a DO-NOT-CLOSE tag to it.
- (7) (WITHOUT GSE-419) Remove the relay K0848 ([AMM TASK 20-13-02-000-801-A/400](#)).
- (8) (WITH GSE-419) Disconnect the electrical connector of the Electromechanical Actuator from the aircraft harness.

J. Electromechanical Actuator - Functional Check ([Figure 501](#)) ([Figure 502](#))

SUBTASK 020-002-A

**WARNING: MAKE SURE THAT THE RUDDER, THE HORIZONTAL STABILIZER AND THE ELEVATOR CANNOT BE OPERATED ACCIDENTALLY. AN ACCIDENTAL OPERATION MAY CAUSE INJURY TO PERSONS.**

- (1) (WITH GSE-419) Do the subsequent procedure to check the friction forces in the Electromechanical Gust-Lock Actuator ([Figure 501](#)).
  - (a) Connect J3014 connector of GSE-419 to the electrical connector of the Electromechanical Actuator and P3014 connector of GSE-419 to the aircraft harness.
  - (b) On the Circuit Breaker Panel, close the GUST-LOCK circuit breaker and remove the DO-NOT-CLOSE tag from it.
  - (c) Install GSE-050 to the bornes of GSE-419.
  - (d) Do these steps three times.

**CAUTION: DO NOT APPLY TOO MUCH FORCE TO HOLD THE CONTROL COLUMN AGAINST THE BACKSTOPS, IN THE FULLY FORWARD POSITION, DURING THE UNLOCKING TRAVEL.**

- 1 Move the control column to the fully forward position (nose down) and keep it thus during the unlocking travel.
          - a Lift the handle and command the gust-lock lever from the locked position to the intermediate position and, with the aid of the digital

multimeter (GSE-050), measure the peak current in the actuator during the extension travel of the actuator.

**NOTE:** Do not use functions that record the maximum and minimum values of current.

- 1) The gust lock light comes on for approximately eight seconds after the gust lock lever is at the intermediate position.
- 2) The gust lock actuator extends and the spring cartridges are retracted.

- 2 Write down the current values, measured in each unlocking travel, in [Table 501](#).

Table 501 - CURRENT LOADS ON ELECTROMECHANICAL ACTUATOR DURING THE UNLOCKING TRAVELS OF THE SPRING CARTRIDGES

Current (mA)	U1	U2	U3
Values measured with GSE 419			
Unlocking Average	Xu =		

- 3 Move the control column to the fully rearward position (nose up) and keep it thus during the locking travel.

- a Lift the handle and command the gust lock lever to the locked position and, with the aid of the digital multimeter (GSE-050), measure the peak current in the actuator during the retraction travel of the actuator.

**NOTE:** Do not use functions that record the maximum and minimum values of current.

- 1) The gust lock actuator retracts and the spring cartridges are extended.

- 4 Write down the current values, measured in each locking travel, in [Table 502](#).

Table 502 - CURRENT LOADS ON ELECTROMECHANICAL ACTUATOR DURING THE LOCKING TRAVELS OF THE SPRING CARTRIDGES

Current (mA)	L1	L2	L3
Values measured with GSE-419			
Locking Average	XI =		

- 5 Move the control column back to the fully forward position (nose down) and make sure that it is locked.

- (e) Find the average value Xu, between U1, U2 and U3, and write it down in [Table 501](#).

- 1  $Xu = (U1+U2+U3)/3$

- (f) Find the average value  $X_I$ , between  $L_1$ ,  $L_2$  and  $L_3$ , and write it down in [Table 502](#).

$$1 \quad X_I = (L_1 + L_2 + L_3) / 3$$

- (g) The best average value for the peak current in the gust lock actuator on the unlocking travel ( $X_u$ ) is 130mA. A maximum of 180mA is acceptable.

**NOTE:** If the peak current of the actuator, on the unlocking travel ( $X_u$ ), is more than 180mA, do the troubleshooting FIM TASK 27-70-00-810-808-A.

- (h) The best average value for the peak current in the gust lock actuator on the locking travel ( $X_I$ ) is 250mA. A maximum of 300mA is acceptable.

**NOTE:** If the peak current of the actuator, on the locking travel ( $X_I$ ), is more than 300mA, do the troubleshooting FIM TASK 27-70-00-810-808-A.

- (2) (WITHOUT GSE-419) Do the subsequent procedure to check the friction forces in the Electromechanical Gust-Lock Actuator ([Figure 502](#)).

- (a) Connect a jumper between pins D1 and D2 of socket XK0848.
- (b) Connect the digital multimeter ([ITEM GSE 050](#)) between pins B1 and B2 of socket XK0848 to measure the current.
- (c) Do these step three times.

**CAUTION:** DO NOT APPLY TOO MUCH FORCE TO HOLD THE CONTROL COLUMN AGAINST THE BACKSTOPS, IN THE FULLY FORWARD POSITION, DURING THE UNLOCKING TRAVEL.

- 1 Move the control column to the fully forward position (nose down) and keep it thus during the unlocking travel.

- a Lift the handle and command the gust-lock lever from the locked position to the intermediate position and, with the aid of the digital multimeter (GSE-050), measure the peak current in the actuator during the extension travel of the actuator.

- 1) The gust lock light comes on for approximately eight seconds after the gust lock lever is at the intermediate position.
  - 2) The gust lock actuator extends and the spring cartridges are retracted.

- 2 Write down the current values, measured in each unlocking travel, in [Table 503](#).

Table 503 - CURRENT LOADS ON ELECTROMECHANICAL ACTUATOR DURING THE UNLOCKING TRAVELS OF THE SPRING CARTRIDGES

Current (mA)	U1	U2	U3
Values measured with GSE-050			
Unlocking Average	$X_u =$		

- 3 Move the control column to the fully rearward position (nose up) and keep it thus during the locking travel.
  - a Lift the handle and command the gust lock lever to the locked position and, with the aid of the digital multimeter (GSE-050), measure the peak current in the actuator during the retraction travel of the actuator.
    - 1) The gust lock actuator retracts and the spring cartridges are extended.

- 4 Write down the current values, measured in each locking travel, in [Table 504](#).

Table 504 - CURRENT LOADS ON ELECTROMECHANICAL ACTUATOR DURING THE LOCKING TRAVELS OF THE SPRING CARTRIDGES

Current (mA)	L1	L2	L3
Values measured with GSE-050			
Locking Average	XI =		

- 5 Move the control column back to the fully forward position (nose down) and make sure that it is locked.
- (d) Find the average value  $X_u$ , between U1, U2 and U3, and write it down in [Table 503](#).
    - 1  $X_u = (U1+U2+U3)/3$
  - (e) Find the average value  $X_I$ , between L1, L2 and L3, and write it down in [Table 504](#).
    - 1  $X_I = (L1+L2+L3)/3$
  - (f) The best average values for the peak current in the gust lock actuator on the unlocking travel ( $X_u$ ) is 130mA with the maximum of 180mA.
 

**NOTE:** If the peak current of the actuator, on the unlocking travel ( $X_u$ ), is higher than 180mA, do the troubleshooting FIM TASK 27-70-00-810-808-A.
  - (g) The best average values for the peak current in the gust lock actuator on the locking travel ( $X_I$ ) is 250mA with the maximum of 300mA.
 

**NOTE:** If the peak current of the actuator, on the locking travel ( $X_I$ ), is higher than 300mA, do the troubleshooting FIM TASK 27-70-00-810-808-A.

## K. Follow-on

### SUBTASK 842-002-A

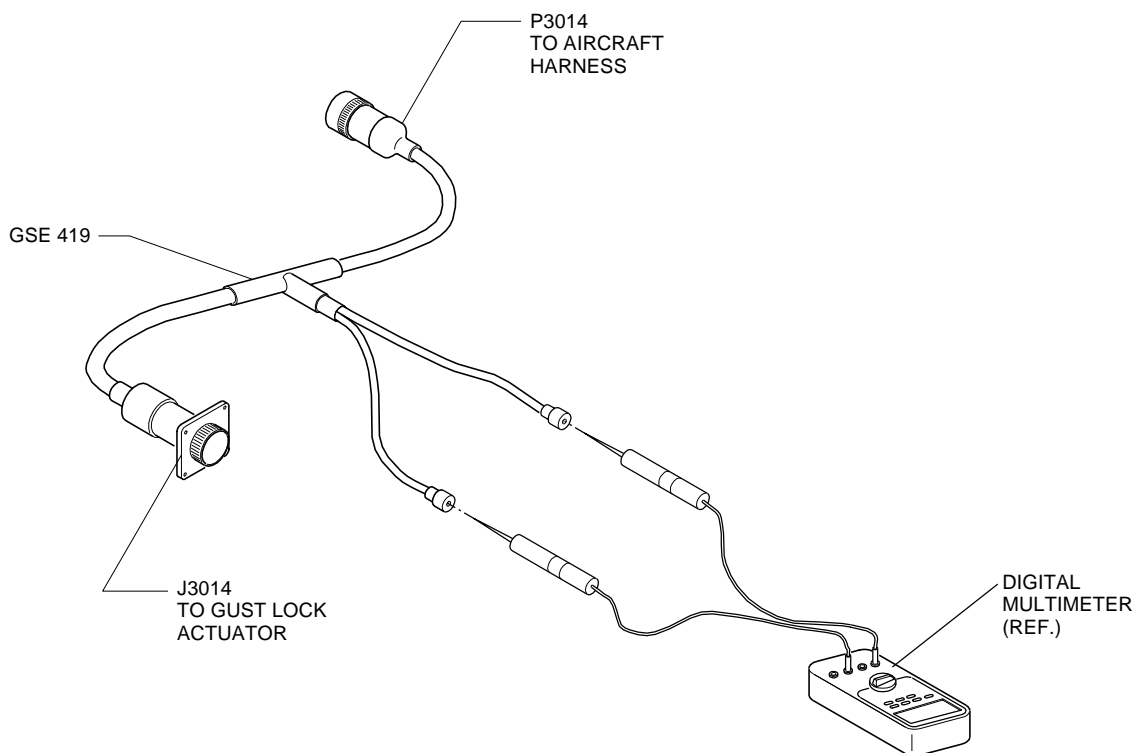
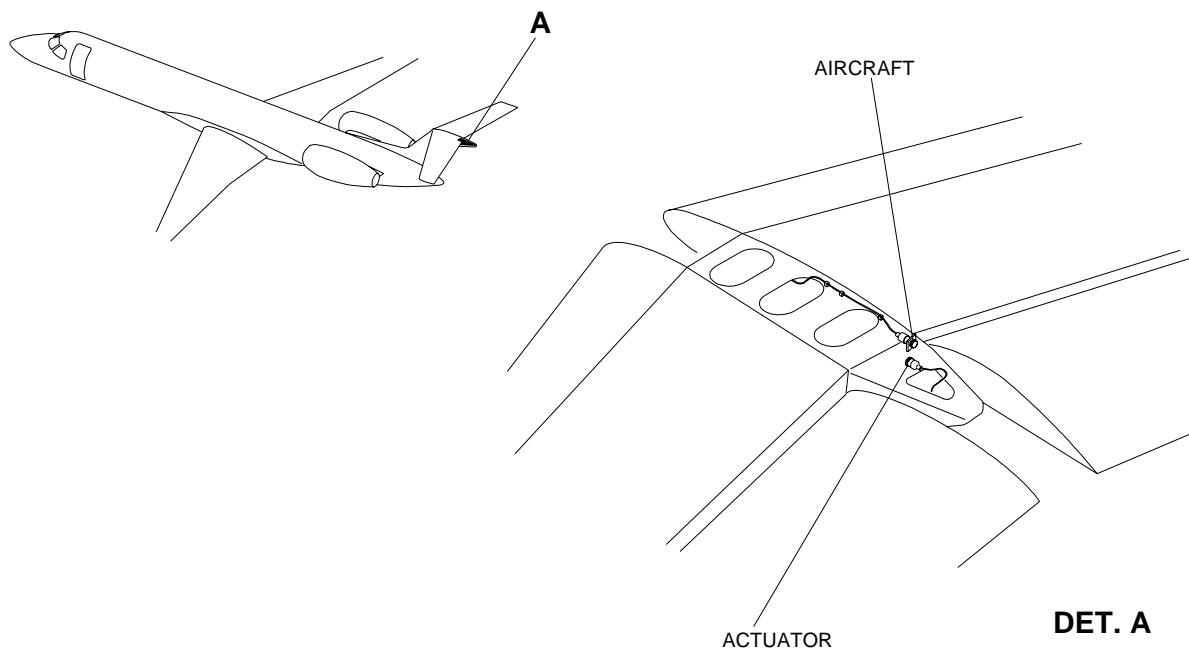
- (1) On the Circuit Breaker Panel, open the GUST-LOCK circuit breaker and attach a DO-NOT-CLOSE tag to it.
- (2) (WITH GSE-419) Disconnect J3014 connector of GSE-419 from the electrical connector of the Electromechanical Actuator and P3014 connector of GSE-419 from the aircraft harness.

- (3) (WITH GSE-419) Connect the electrical connector of the Electromechanical Actuator to the aircraft harness.
- (4) (WITH GSE-419) Install rear fairings 321 to the horizontal stabilizer. Refer to [AMM TASK 55-36-00-400-801-A/400](#).
- (5) (WITHOUT GSE-419) Reinstall the relay K0848 ([AMM TASK 20-13-02-400-801-A/400](#)).
- (6) On the Circuit Breaker Panel, close the GUST-LOCK circuit breaker and remove the DO-NOT-CLOSE tag from it.
- (7) De-energize the aircraft [AMM TASK 20-40-01-860-801-A/200](#).

**EFFECTIVITY: WITH GSE-419**

Electromechanical Actuator - Measure of Load

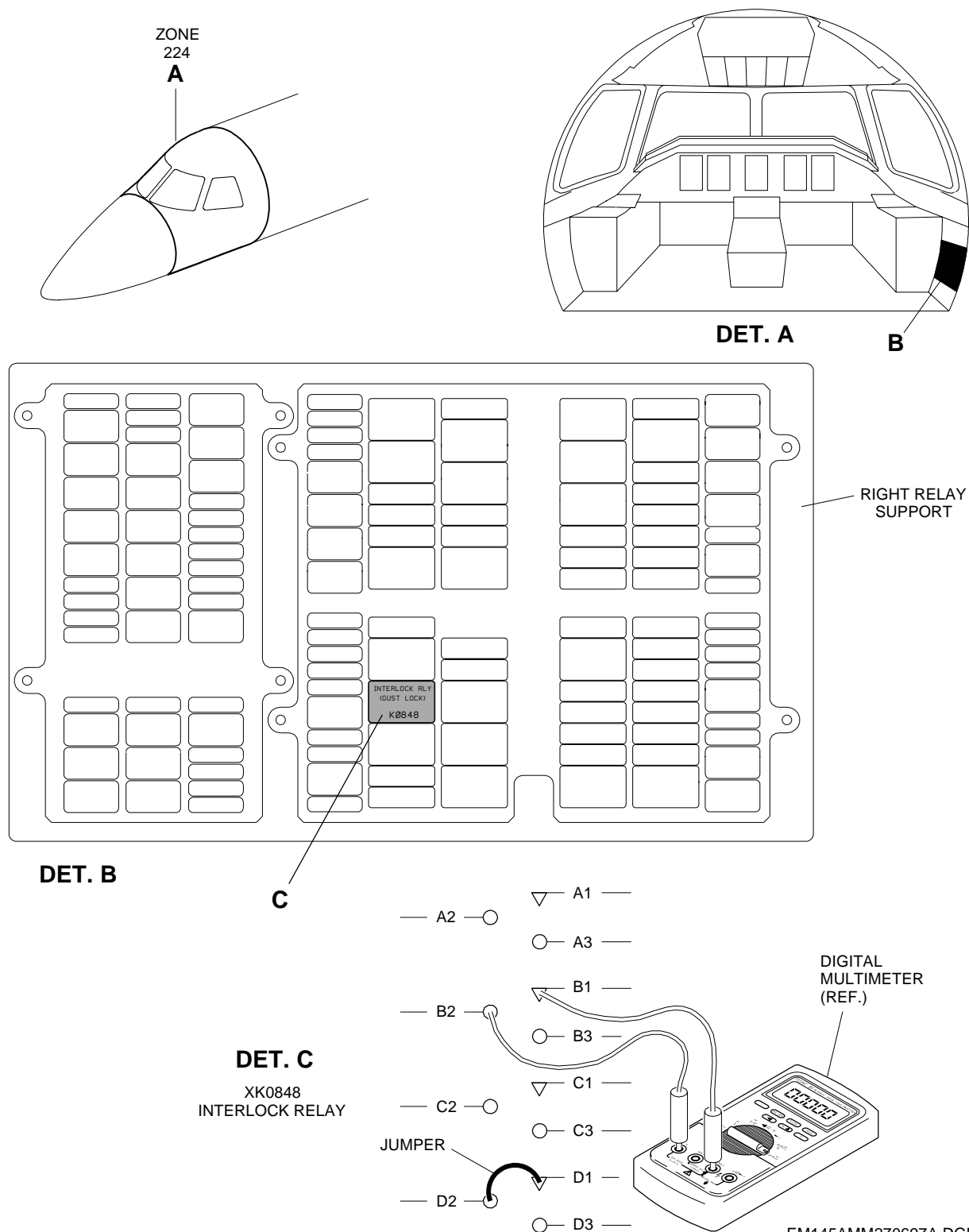
Figure 501



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EFFECTIVITY: WITHOUT GSE-419  
Electromechanical Actuator - Measure of Load  
Figure 502



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