



AIRCRAFT MAINTENANCE MANUAL

REAR WING-TO-FUSELAGE FAIRING - REMOVAL/INSTALLATION

EFFECTIVITY: ACFT MODEL(S) EMB-135

1. General

- A. This section gives the procedures to remove and install the Rear Wing-to-Fuselage Fairing.
- B. The procedures in this section are given in the sequence below. The tasks identified with (♦) are part of the Scheduled Maintenance Requirements Document (SMRD).

TASK NUMBER	DESCRIPTION	EFFECTIVITY
53-04-30-000-801-A	REAR WING-TO-FUSELAGE FAIRING - REMOVAL	ACFT MODEL(S) EMB-135
53-04-30-400-801-A	REAR WING-TO-FUSELAGE FAIRING - INSTALLATION	ACFT MODEL(S) EMB-135



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TASK 53-04-30-000-801-A

EFFECTIVITY: ACFT MODEL(S) EMB-135

2. REAR WING-TO-FUSELAGE FAIRING - REMOVAL

A. General

(1) This procedure gives the instructions to remove the Rear Wing-to-Fuselage Fairing.

B. References

REFERENCE	DESIGNATION
AMM MPP 06-41-01/100	-
SRM 51-20-01-PR	-

C. Zones and Accesses

ZONE	PANEL/DOOR	LOCATION
193	193AL	Rear wing-to-fuselage attachment area
193	193LL	Rear wing-to-fuselage attachment area
193	193BL (LH side)	Rear wing-to-fuselage attachment area
193	193CR (RH side)	Rear wing-to-fuselage attachment area

D. Tools and Equipment

ITEM	DESCRIPTION	PURPOSE	QTY
Commercially available	Acrylic spatula	To remove sealant	

E. Auxiliary Items

ITEM	DESCRIPTION	PURPOSE	QTY
Commercially available	Protective Gloves	For protection of technician's hands	1
Commercially available	Safety Goggles	For protection of technician's eyes	1

F. Consumable Materials

SPECIFICATION (BRAND)	DESCRIPTION	QTY
ASTM-D-740	Methyl Ethyl Ketone - MEK	AR

G. Expandable Parts

Not Applicable

H. Persons Recommended

QTY	FUNCTION	PLACE
3	Do the task	Rear wing-to-fuselage attachment area

I. Preparation

SUBTASK 841-015-B

- (1) Put the aircraft on safe condition for maintenance.

J. Removal (Figure 401)

SUBTASK 020-015-B

- (1) With an acrylic spatula, remove the old aerodynamic sealant in the rear wing-to-fuselage attachment area.
- (2) Refer to (AMM MPP 06-41-01/100) and remove the hydraulic compartment access panels.
 - 193BL (LH side) (4).
 - 193CR (RH side) (1).

WARNING: THE HYDRAULIC SYSTEM CONTAINS PHOSPHATE-ESTER HYDRAULIC FLUID. THE FLUID CAN CAUSE IRRITATION IN YOUR SKIN OR INJURY TO YOUR EYES. USE THE APPLICABLE GOGGLES AND RUBBER GLOVES. IF THE FLUID TOUCHES YOU, FLUSH YOUR SKIN WITH WATER. IF IT GETS IN YOUR EYES, FLUSH THEM WITH WATER AND GET MEDICAL HELP.

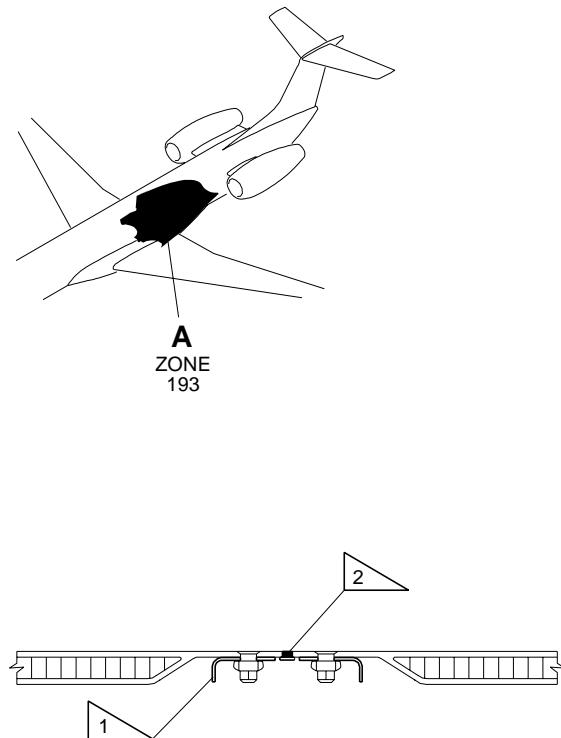
- (3) Remove the clamps (LH and RH sides) which connect the flexible hose to the hydraulic compartment drain.
- (4) Remove the clamps (LH and RH sides) which connect the drain to the air-scoop outlet ducts.
- (5) Remove the 193AL access panel (5). Refer to (AMM MPP 06-41-01/100).
- (6) Remove the 193LL access panel (8). Refer to (AMM MPP 06-41-01/100).
- (7) Remove the clamps which connect the drains to the fan air control thermostat hoses.
- (8) Remove the clamp which connects the drain to the drain valve hose.
- (9) Remove the clamp which connects the flexible hose to the heated water drain.
- (10) Remove the screws (7) that attach the splices (LH and RH sides) (6) to the fairings.
- (11) Remove the splices (6) from the interface between the center and the rear wing-to-fuselage fairings (2).
- (12) Remove the screws (3) that attach the rear wing-to-fuselage fairing (2).
- (13) Remove the rear wing-to-fuselage fairing (2).



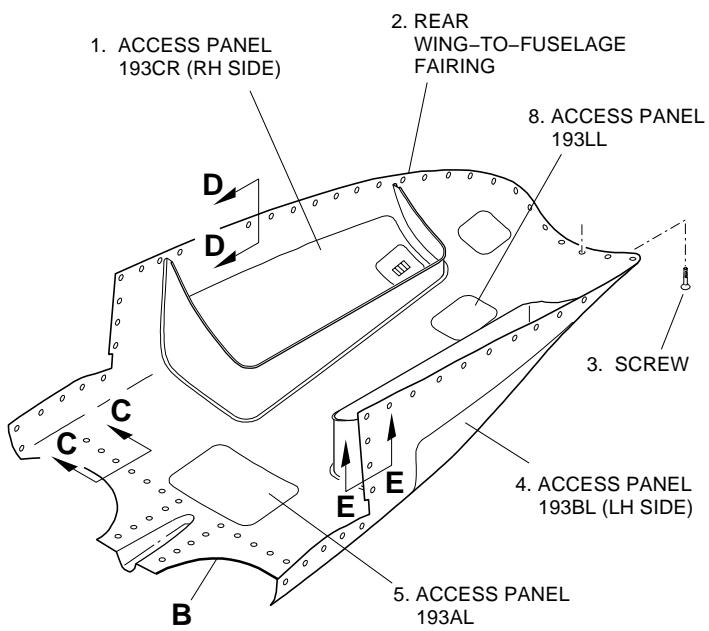
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NOTE: Make sure that the interface sealing is not damaged. If necessary, apply sealant again. Refer to (SRM 51-20-01-PR).

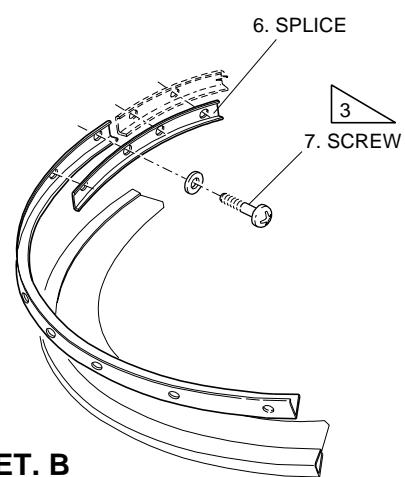
EFFECTIVITY: ACFT MODEL(S) EMB-135
Rear Wing-to-Fuselage Fairing - Removal/Installation
Figure 401



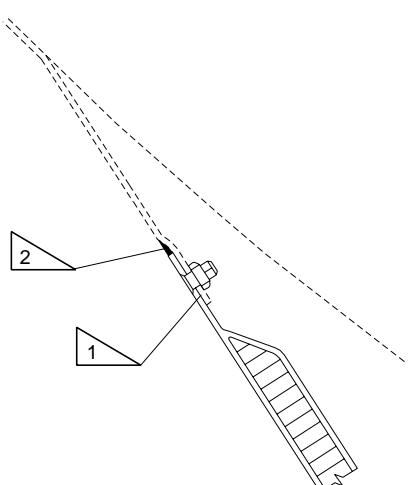
SECTION E-E



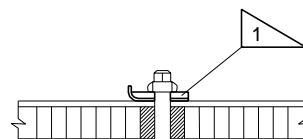
DET. A



DET. B



SECTION D-D

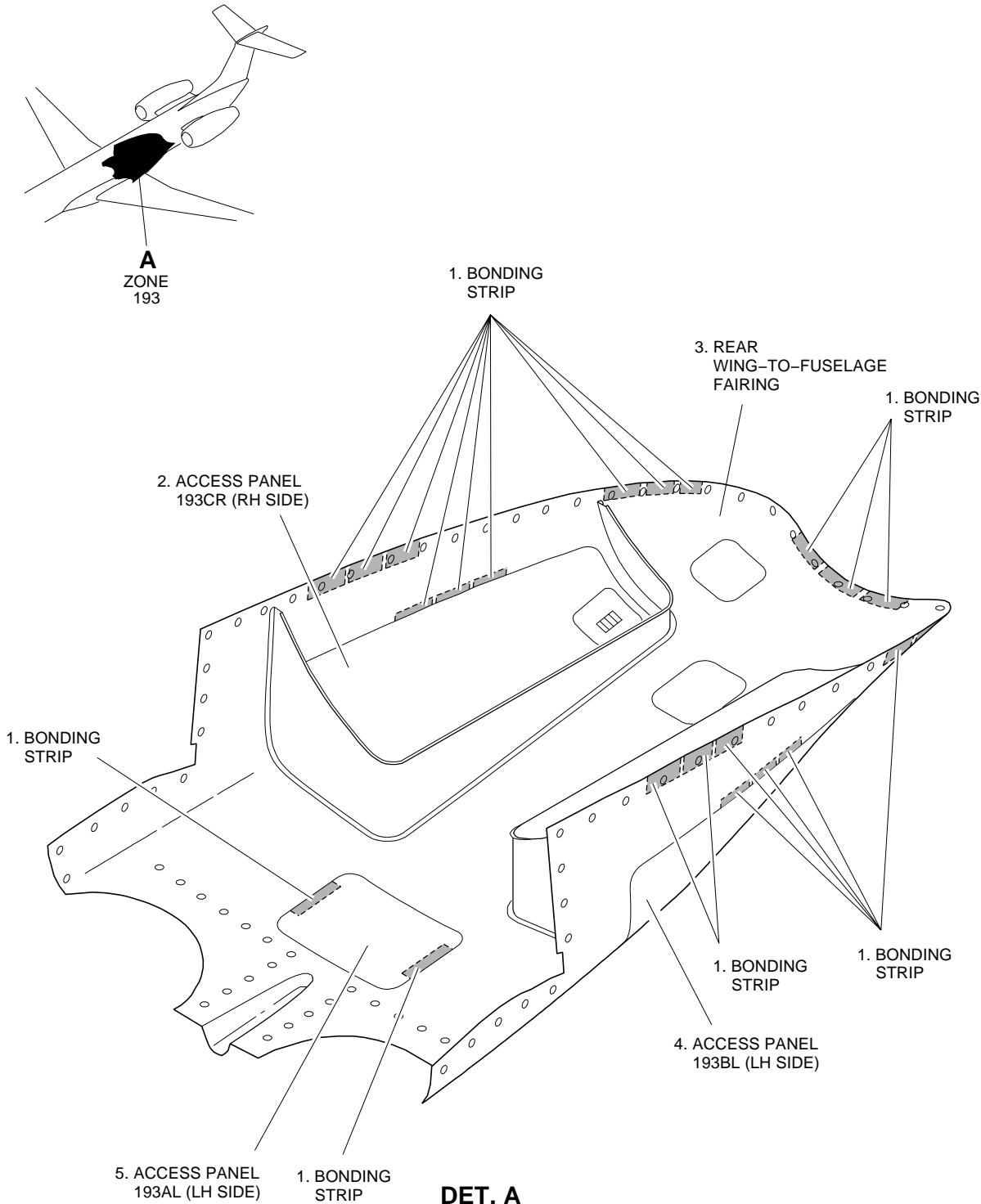


SECTION C-C

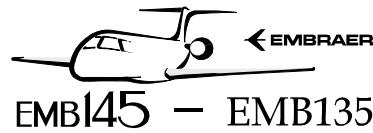
-  1. APPLY COR-BAN 27L.
-  2. AERODYNAMIC SEALING.
-  3. TORQUE: 1.4-1.7 N.m (12-15 lb.in)

EM145AMM530160B.DGN

EFFECTIVITY: ACFT MODEL(S) EMB-135
Bonding Strap
Figure 402



EM145AMM530346B.DGN



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TASK 53-04-30-400-801-A

EFFECTIVITY: ACFT MODEL(S) EMB-135

3. REAR WING-TO-FUSELAGE FAIRING - INSTALLATION

A. General

(1) This procedure gives the instructions to install the rear wing-to-fuselage fairing.

B. References

REFERENCE	DESIGNATION
AMM MPP 06-41-01/100	-
AMM MPP 20-10-01/200	- MAINTENANCE PRACTICES
AMM TASK 20-13-21-700-801-A/200	ELECTRICAL BONDING TEST - STANDARD PROCEDURES
AMM TASK 28-41-00-200-801-A/600	-
IPC 53-04-00	FAIRINGS WING/FUSELAGE
SRM 51-20-01-PR	-

C. Zones and Accesses

ZONE	PANEL/DOOR	LOCATION
193	193AL	Rear wing-to-fuselage attachment area
193	193LL	Rear wing-to-fuselage attachment area
193	193BL (LH side)	Rear wing-to-fuselage attachment area
193	193CR (RH side)	Rear wing-to-fuselage attachment area

D. Tools and Equipment

ITEM	DESCRIPTION	PURPOSE	QTY
Commercially available	Torque wrench	To tighten the screws	

E. Auxiliary Items

ITEM	DESCRIPTION	PURPOSE	QTY
Commercially available	Protective Gloves	For protection of technician's hands	1
Commercially available	Safety Goggles	For protection of technician's eyes	1

F. Consumable Materials

SPECIFICATION (BRAND)	DESCRIPTION	QTY
MEP 09-075	Corrosion-Inhibiting Compound (COR-BAN 27L)	AR
MIL-PRF-81733 TYPE II-2	Sealant, P/S 870 - B 2	AR

(Continued)

<i>SPECIFICATION (BRAND)</i>	<i>DESCRIPTION</i>	<i>QTY</i>
MIL-PRF-81733 TYPE II-1/2	Sealant, P/S 870 - B 1/2	AR
Commercially available	Adhesive Tape	AR
Commercially available	Aluminum Tape (AL Tape 425)	AR
Commercially available	Polyethylene Film	AR
ASTM-D-740	Methyl Ethyl Ketone - MEK	AR
Commercially available	Scotch Brite Sponge	AR

G. Expendable Parts

<i>ITEM</i>	<i>IPC REFERENCE (VENDOR REFERENCE)</i>	<i>QTY</i>
Gasket	IPC 53-04-00	AR

H. Persons Recommended

<i>QTY</i>	<i>FUNCTION</i>	<i>PLACE</i>
3	Do the task	Rear wing-to-fuselage attachment area

I. Installation (Figure 401) (Figure 402)
SUBTASK 420-015-B

CAUTION: THE INSPECTION OF FUEL QUANTITY INDICATION HARNESS IS PART OF CRITICAL DESIGN CONFIGURATION CONTROL LIMITATIONS (CDCCL) IN THE AIRWORTHINESS LIMITATIONS OF THE AIRCRAFT MAINTENANCE PROGRAM.

- (1) Do an inspection on the fuel quantity indication harness, according to AMM TASK 28-41-00-200-801-A/600.

WARNING: COR-BAN 27L IS TOXIC TO SKIN, EYES, AND RESPIRATORY INHALATION. USE PVC GLOVES AND EYE PROTECTION GOGGLES. USE ONLY IN WELL VENTILATED AREAS. OBEY THE MANUFACTURER'S HEALTH AND SAFETY INSTRUCTIONS.

- (2) Apply COR-BAN 27L to the faying surface of the rear wing-to-fuselage (3).

- (3) Install the rear wing-to-fuselage fairing (2).

- (4) Install the screws (3) that attach the rear wing-to-fuselage fairing (2).

NOTE: For torque, refer to ([AMM MPP 20-10-01/200](#)).

- (5) Install the clamps to connect the flexible hose to the hydraulic compartment drain.

- (6) Install the clamps (LH and RH sides) to connect the drain to the air-scoop outlet hose.

- (7) Install the clamp to connect the drain to the drain valve hose.

- (8) Install the clamp to connect the flexible hose to the heated water drain.
- (9) Install the clamp to connect the drains to the fan air-control thermostat hoses.
- (10) Install the splices (6) at the interface between the center and the rear wing-to-fuselage fairings (2).
- (11) Install the screws (7) that attach the splices (LH and RH side) (6) to the fairings.
NOTE: Apply torque: 1.4 - 1.7 N.m (12 - 15 lb.in).
- (12) Install the hydraulic compartment access panels (AMM MPP 06-41-01/100):
 - 193BL (LH side) (4).
 - 193CR (RH side) (1).
- (13) Install the 193AL access panel (5) (AMM MPP 06-41-01/100).
- (14) Install the 193LL access panel (8) (AMM MPP 06-41-01/100).
- (15) There are different methods to do the aerodynamic sealing in the rear wing-to-fuselage attachment area. Use the applicable procedure as necessary.
 - (a) Full sealant-curing time:
 - 1 Apply sealant P/S 870 - B 1/2 or P/S 870 - B 2.
NOTE: Sealant curing time will change according to the environmental conditions. Refer to (SRM 51-20-01-PR).
 - (b) Acceleration of the sealant curing time with heating:
 - 1 Apply sealant P/S 870 - B 1/2 or P/S 870 - B 2.
 - 2 Wait for one hour after the sealant is applied.
CAUTION: THE TEMPERATURE MUST NOT BE HIGHER THAN 55°C (131°F).
 - 3 Heat the area which received the sealant.
NOTE: Sealant curing time will change according to the environmental conditions. Refer to (SRM 51-20-01-PR).
 - (c) Application of aluminum tape after tack-free time.
NOTE: This procedure must only be done when a faster aircraft clearance is necessary.
 - 1 Apply sealant P/S 870 - B 1/2 or P/S 870 - B 2.
 - 2 You can accelerate the sealant curing time as written in paragraph (b).
 - 3 After the sealant is tack-free, apply aluminum tape.
NOTE: You can operate the aircraft immediately after the aluminum tape application.



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4 Remove the aluminum tape after 150 hours.

(d) Application of polyethylene film and aluminum tape.

1 Apply sealant P/S 870 - B 1/2 or P/S 870 - B 2.

2 Apply a polyethylene film over the sealant.

3 Apply the aluminum tape.

NOTE: You can operate the aircraft immediately after the aluminum tape application.

4 Remove the aluminum tape after 10 days.

J. Follow-on

SUBTASK 842-015-B

- (1) Do the electrical-bonding test procedure as given in [AMM TASK 20-13-21-700-801-A/200](#). Refer to (Figure 402).
- (2) Do the procedure to put the aircraft back to initial condition.