



## AIRCRAFT MAINTENANCE MANUAL

### AILERON - ADJUSTMENT/TEST

*EFFECTIVITY: ALL*

#### 1. General

- A. This section gives the procedures to measure the maximum values of deflections of the ailerons.
- B. The procedures in this section are given in the sequence below. The tasks identified with (♦) are part of the Scheduled Maintenance Requirements Document (SMRD).

| <i>TASK NUMBER</i> | <i>DESCRIPTION</i>           | <i>EFFECTIVITY</i> |
|--------------------|------------------------------|--------------------|
| 27-10-00-700-801-A | AILERON ADJUSTMENT AND CHECK | ALL                |



EMB145 – EMB135

AIRCRAFT  
MAINTENANCE MANUAL

TASK 27-10-00-700-801-A

EFFECTIVITY: ALL

2. AILERON ADJUSTMENT AND CHECK

A. General

- (1) This task gives the procedures to measure the maximum values of deflections of the ailerons. [Figure 501](#) shows the deflections of the ailerons.

B. References

| REFERENCE                       | DESIGNATION                                |
|---------------------------------|--|
| AMM TASK 29-10-00-860-801-A/200 | HYDRAULIC SYSTEM - PRESSURIZATION WITH HTS |

C. Zones and Accesses

Not Applicable

D. Tools and Equipment

| ITEM    | DESCRIPTION        | PURPOSE                                    | QTY |
|---------|--------------------|--|-----|
| GSE 070 | Digital Protractor | To measure the deflections of the ailerons |     |

E. Auxiliary Items

Not Applicable

F. Consumable Materials

Not Applicable

G. Expandable Parts

Not Applicable

H. Persons Recommended

| QTY | FUNCTION      | PLACE    |
|-----|---------------|----------|
| 1   | Does the task | Cockpit  |
| 1   | Does the task | Ailerons |

I. Preparation ([Figure 501](#))

SUBTASK 841-002-A

- (1) Make sure that the aircraft is safe for maintenance.
- (2) Do not do other tasks on the aileron system.
- (3) Pressurize the hydraulic system ( [AMM TASK 29-10-00-860-801-A/200](#)).

J. Aileron Adjustment and Check ([Figure 501](#))

SUBTASK 720-002-A

- (1) Adjust the digital protractor to zero position.

- (a) Install the Rig Pin in the right wing sector DET. B..
- (b) With the aileron system pressurized (system 1 or 2), the aileron will be in the neutral position.
- (c) Adjust the digital protractor to zero position and remove the rig pin.

**NOTE:** The neutral position of the aileron is identified by the alignment of the aileron and wing trailing edge.

- (2) Do a check of the right aileron for deflection.

Table 501 - RIGHT AILERON DEFLECTIONS

| <b>AILERON</b>   |                  |
|------------------|------------------|
| <b>UP</b>        | <b>DOWN</b>      |
| $25 \pm 1^\circ$ | $15 \pm 1^\circ$ |

- (a) Set the aileron to the UP position.

Result:

1 The deflection must be  $25 \pm 1^\circ$ .

- (b) Set the aileron to the DOWN position.

Result:

1 The deflection must be  $15 \pm 1^\circ$ .

- (3) Adjust the digital protractor to zero position.

- (a) Install the Rig Pin in the left wing sector DET. B.

- (b) With the aileron system pressurized (system 1 or 2), the aileron will be in the neutral position.

- (c) Adjust the digital protractor to zero position and remove the rig pin.

**NOTE:** The neutral position of the aileron is identified by the alignment of the aileron and wing trailing edge.

- (4) Do a check of the left aileron deflection.

Table 502 - LEFT AILERON DEFLECTIONS

| <b>AILERON</b>   |                  |
|------------------|------------------|
| <b>UP</b>        | <b>DOWN</b>      |
| $25 \pm 1^\circ$ | $15 \pm 1^\circ$ |

- (a) Set the aileron to the UP position.

Result:

1 The deflection must be  $25 \pm 1^\circ$ .

- (b) Set the aileron to the DOWN position.

Result:

1 The deflection must be  $15 \pm 1^\circ$ .



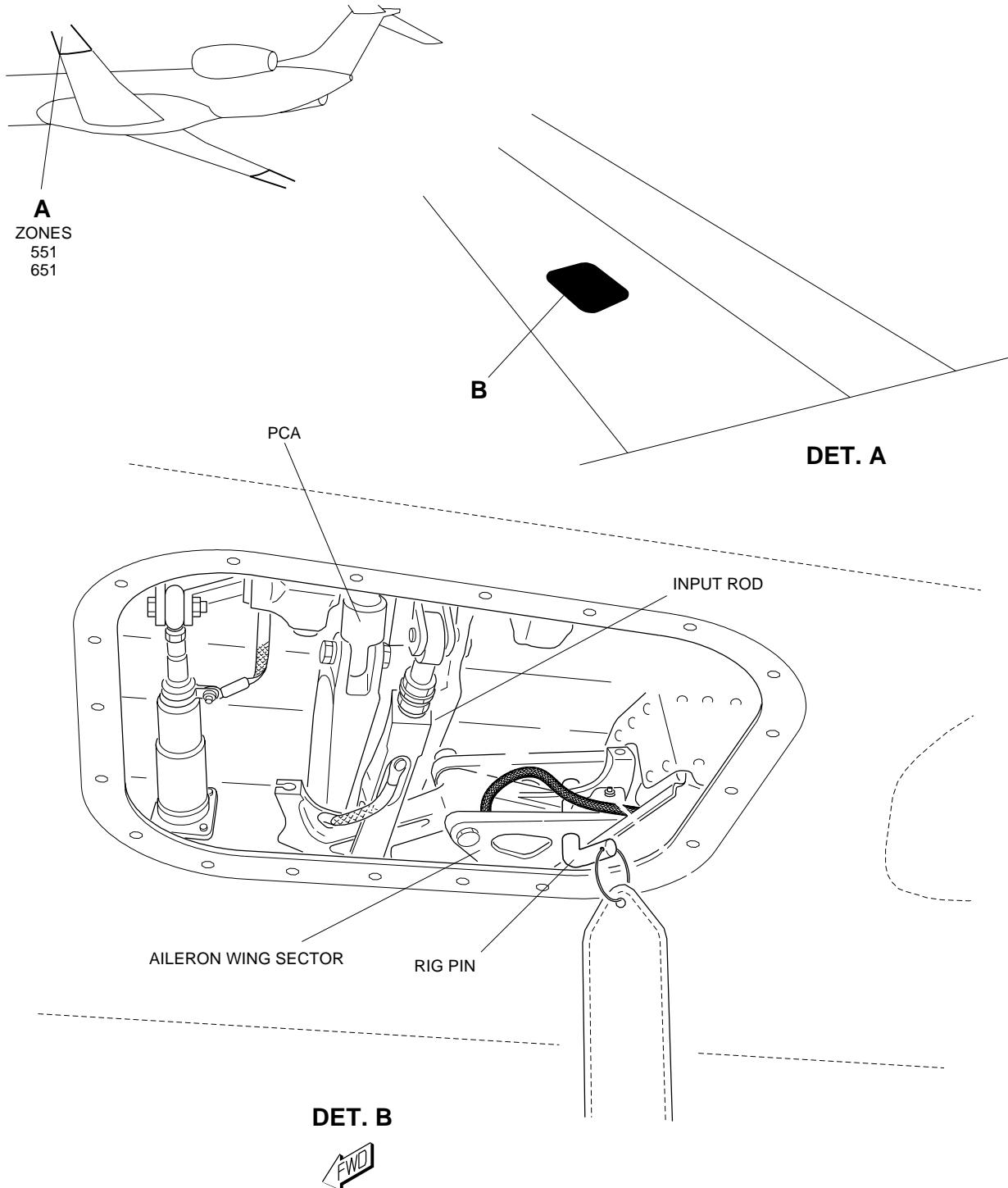
## AIRCRAFT MAINTENANCE MANUAL

### K. Follow-on

#### SUBTASK 842-002-A

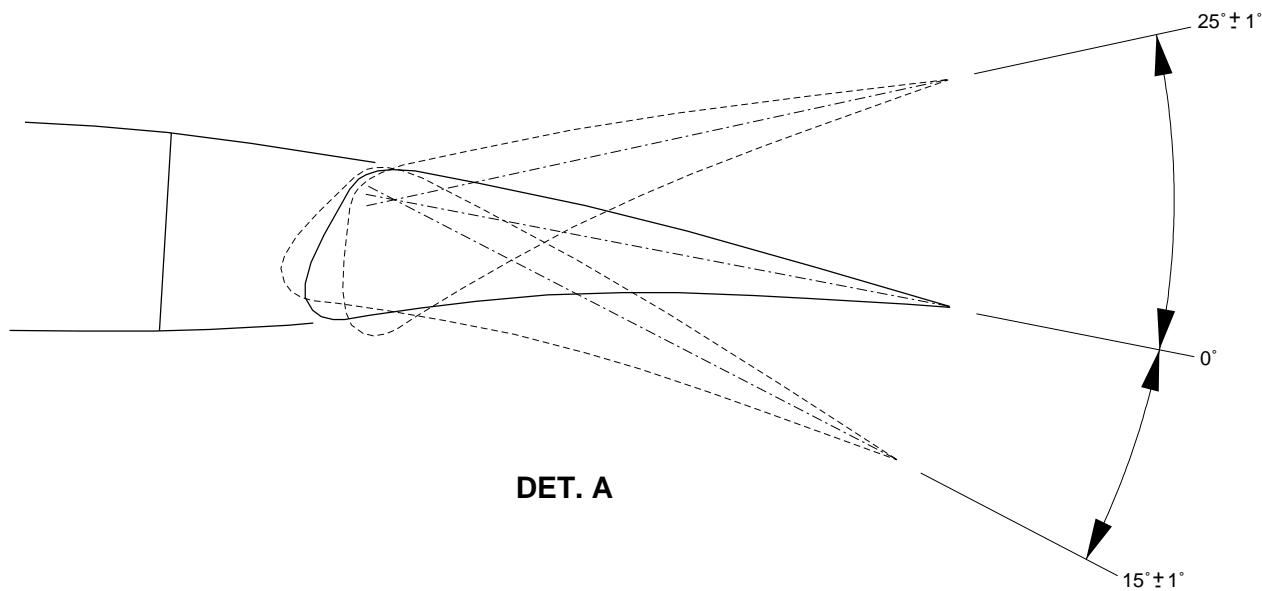
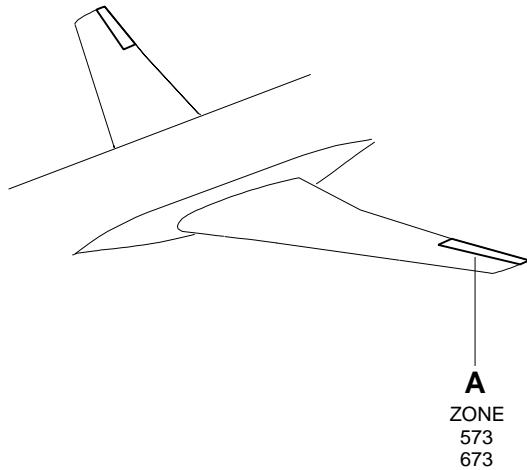
- (1) Set the ailerons to the neutral position.
- (2) Release the pressure of the hydraulic system ([AMM TASK 29-10-00-860-801-A/200](#)).

**EFFECTIVITY: ALL**  
Deflections of the Ailerons  
Figure 501 - Sheet 1



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**EFFECTIVITY: ALL**  
Deflections of the Ailerons  
Figure 501 - Sheet 2



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