HIGH-STAGE PRESSURE SWITCH - ADJUSTMENT/TEST

EFFECTIVITY: ALL

1. General

   A. This section gives the procedure to do the functional test of the high stage pressure switch of the LH and RH bleed systems.

   B. The procedures in this section are given in the sequence below. The tasks identified with (♦) are part of the Scheduled Maintenance Requirements Document (SMRD).

TASK NUMBER	*DESCRIPTION*	*EFFECTIVITY*
36-11-07-700-801-A	HIGH STAGE PRESSURE SWITCH - FUNCTIONAL TEST	ALL

TASK 36-11-07-700-801-A
EFFECTIVITY: ALL

2. HIGH STAGE PRESSURE SWITCH - FUNCTIONAL TEST

A. General

(1) This task gives the procedure to do the functional test of the high stage pressure switch of the LH and RH bleed systems.

B. References

REFERENCE	*DESIGNATION*
AMM MPP 06-43-00/100	- COMPONENT LOCATION
AMM MPP 71-00-00/200	- MAINTENANCE PRACTICES
AMM TASK 20-40-01-860-801-A/200	ENERGIZATION OF THE AIRCRAFT WITH AN EXTERNAL POWER SOURCE
AMM TASK 49-10-00-910-803-A/200	APU - SHUTDOWN
AMM TASK 49-13-00-910-803-A/200	APU - SHUTDOWN
AMM TASK 71-00-01-910-801-A/200	ENGINE START PROCEDURE (NORMAL)
AMM TASK 71-00-01-910-804-A/200	ENGINE STOP PROCEDURE
SB145-36-0045	-

C. Zones and Accesses

ZONE	*PANEL/DOOR*	*LOCATION*
412	412BT - Upper Cowling	LH Engine
422	422BT - Upper Cowling	RH Engine

D. Tools and Equipment

Not Applicable

E. Auxiliary Items

ITEM	*DESCRIPTION*	*PURPOSE*	*QTY*
Commercially available	Ladder	To get access to the work area	1

F. Consumable Materials

Not Applicable

G. Expandable Parts

Not Applicable

H. Persons Recommended

QTY	*FUNCTION*	*PLACE*
1	Does the task	LH and RH engines and in the cockpit

I. Preparation

SUBTASK 841-002-A

(1) Remove the upper cowling access panel 412BT (LH Engine) or 422BT (RH Engine) (AMM MPP 06-43-00/100).

(2) Disconnect the electrical connector (1) from the LH/RH High Stage Valve (2), as shown on Figure 501 for PRE-MOD SB145-36-0045 or Figure 502 for POST-MOD SB145-36-0045.

(a) Energize the aircraft (AMM TASK 20-40-01-860-801-A/200).

Result:

1 The HS VLV 1/2 FAIL caution message must come into view on EICAS display.

WARNING: USE PRECAUTIONS AND OBEY OPERATION LIMITATIONS FOR THE ENGINE GROUND OPERATION DURING THE AIRCRAFT MAINTENANCE (AMM MPP 71-00-00/200).

(3) Do the engine start procedure (AMM TASK 71-00-01-910-801-A/200).

(4) Make sure that the MFD is on and set the ECS page.

(5) Make sure that the thrust rating mode messages ALT T/O-1 show on the EICAS display.

(6) On the AIR CONDITIONING/PNEUMATIC control panel, release the APU BLEED pushbutton (light goes off).

(a) On the AIR CONDITIONING/PNEUMATIC control panel, set the XBLEED selector knob to the AUTO position.

(b) Make sure that the PACK 1 and PACK 2 pushbuttons, on the AIR CONDITIONING/PNEUMATIC control panel, on the overhead panel, are released (pushbutton light on).

(c) Make sure that the BLEED 1 and BLEED 2 pushbuttons, on the AIR CONDITIONING/PNEUMATIC control panel, on the overhead panel, are released (pushbutton light on).

(d) Make sure that the thrust levers are in the IDLE position.

(7) Make sure that these pushbuttons and knob on the ICE PROTECTION control panel, on the overhead panel, are set as follows:

(a) WING pushbutton - OFF.

(b) STAB pushbutton - OFF.

(c) OVERRIDE knob - AUTO.

J. High Stage Pressure Switch - Functional Test

SUBTASK 710-002-A

(1) Do the functional test of the High Stage Pressure Switch as follows:

(a) On the AIR CONDITIONING/PNEUMATIC control panel, on the overhead panel, push the BLEED 1 and PACK 1 pushbuttons.

Result:

1 The BLEED 1 pushbutton light goes off.

2 The PACK 1 pushbutton light goes off.

3 The HS VLV 1 FAIL message goes out of view.

(b) Set the LH thrust lever (LH engine) to ± 75% N2 (54°).

Result:

1 The HS VLV 1 FAIL caution message comes into view.

(c) Set the LH thrust lever (LH engine) to idle position.

Result:

1 The HS VLV 1 FAIL caution message goes out of view.

(d) If the HS VLV 1 FAIL message goes out of view, it shows that the High Stage Pressure Switch is functional in the system.

(e) On the AIR CONDITIONING/PNEUMATIC control panel on the overhead panel, release the BLEED 1 pushbutton.

Result:

1 The BLEED 1 pushbutton light comes on.

(f) On the AIR CONDITIONING/PNEUMATIC control panel on the overhead panel, release the PACK 1 pushbutton.

Result:

1 The PACK 1 pushbutton light comes on.

(g) On the AIR CONDITIONING/PNEUMATIC control panel, on the overhead panel, push the BLEED 2 and PACK 2 pushbuttons.

Result:

1 The BLEED 2 pushbutton light goes off.

2 The PACK 2 pushbutton light goes off.

3 The HS VLV 2 FAIL message goes out of view.

(h) Set the RH thrust lever (RH engine) to ± 75% N2 (54°).

Result:

1 The HS VLV 2 FAIL caution message comes into view.

(i) Set the RH thrust lever (RH engine) to idle position.

Result:

1 The HS VLV 2 FAIL caution message goes out of view.

(j) If the HS VLV 2 FAIL message goes out of view, the High Stage Pressure Switch is functional in the system.

(k) On the AIR CONDITIONING/PNEUMATIC control panel, on the overhead panel, release the PACK 2 and BLEED 2 pushbuttons.

Result:

1 The PACK 2 and BLEED 2 pushbutton lights come on.

(l) On the AIR CONDITIONING/PNEUMATIC control panel, release the APU BLEED pushbutton.

Result:

1 The APU BLEED pushbutton light goes off.

K. Follow-on

SUBTASK 842-002-A

(1) Do the engine stop procedure (AMM TASK 71-00-01-910-804-A/200).

(2) On the AIR CONDITIONING/PNEUMATIC control panel, set the XBLEED selector knob to the CLOSED position.

(3) On the AIR CONDITIONING/PNEUMATIC control panel, release the APU BLEED pushbutton (light goes on).

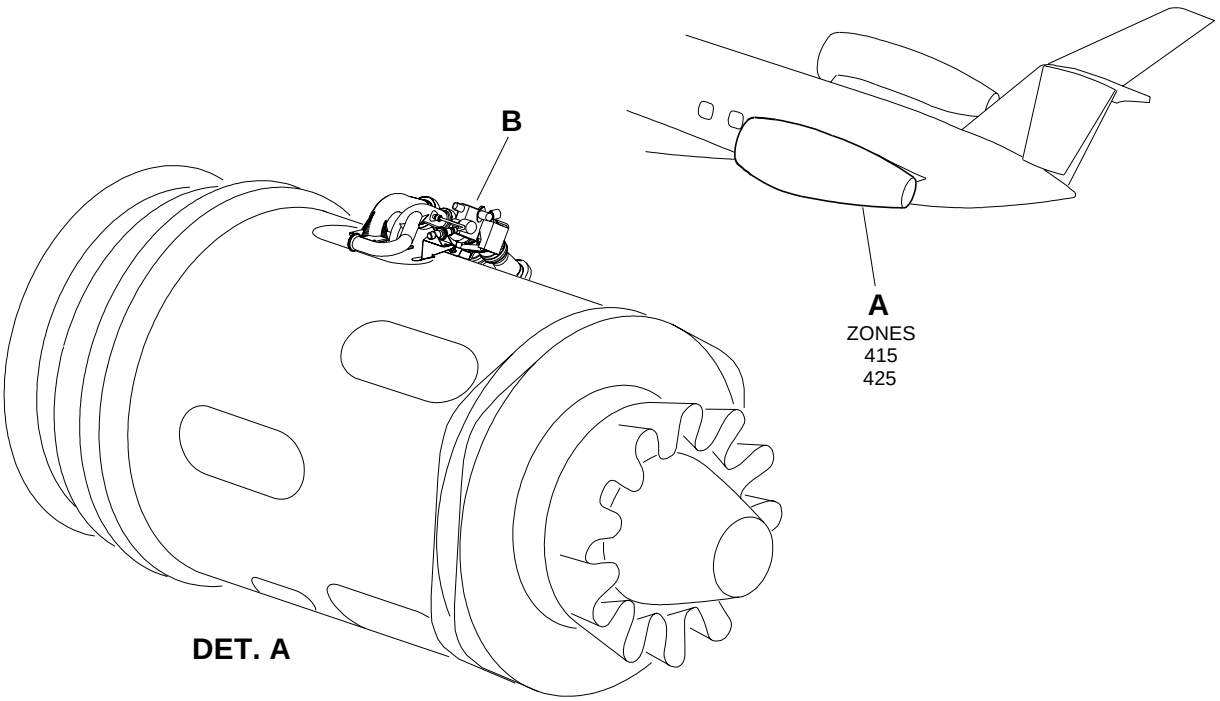
(4) Stop the APU (AMM TASK 49-10-00-910-803-A/200 for APU T-62T-40C11 or AMM TASK 49-13-00-910-803-A/200 for APU T-62T-40C14).

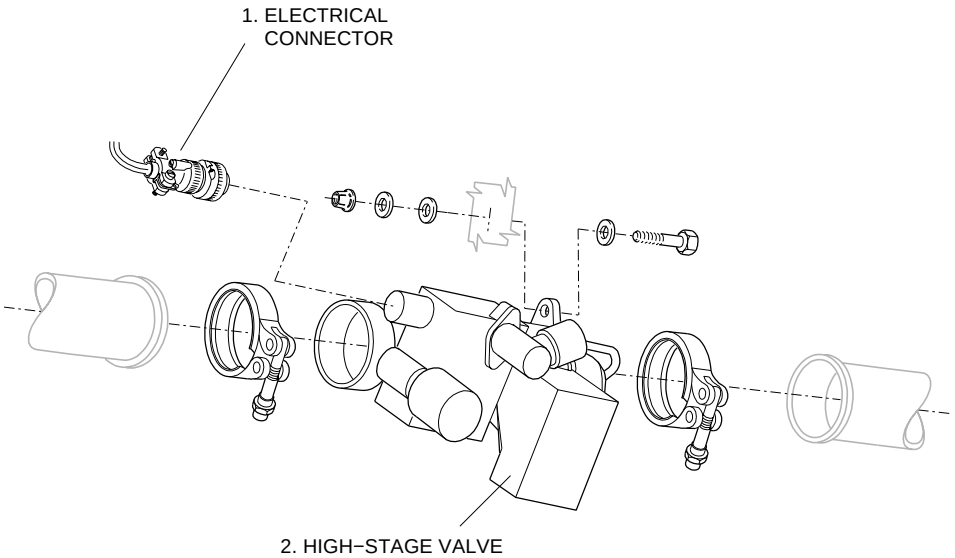
(5) De-energize the aircraft (AMM TASK 20-40-01-860-801-A/200).

(6) Connect the electrical connector (1) to the LH/RH High Stage Valve (2). Refer to Figure 501 for PRE-MOD SB145-36-0045 or Figure 502 for POST-MOD SB145-36-0045.

(7) Close the upper cowling access panel 412BT (LH Engine) or 422BT (RH Engine) (AMM MPP 06-43-00/100).

EFFECTIVITY: PRE-MOD SB145-36-0045
Electrical Connector - High Stage Valve
Figure 501





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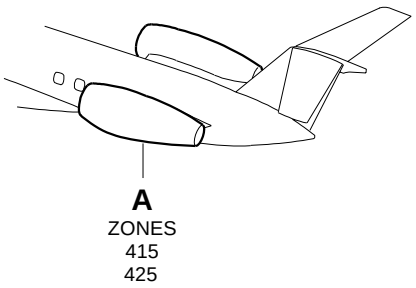
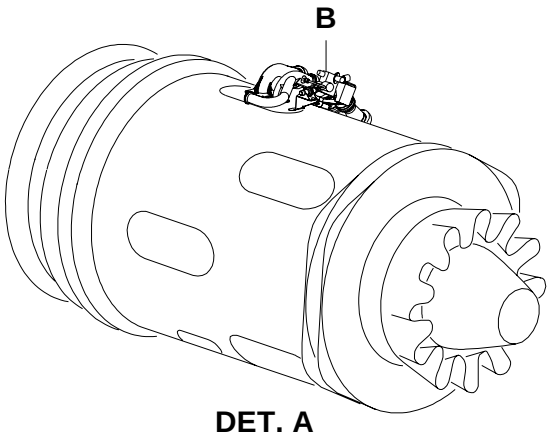
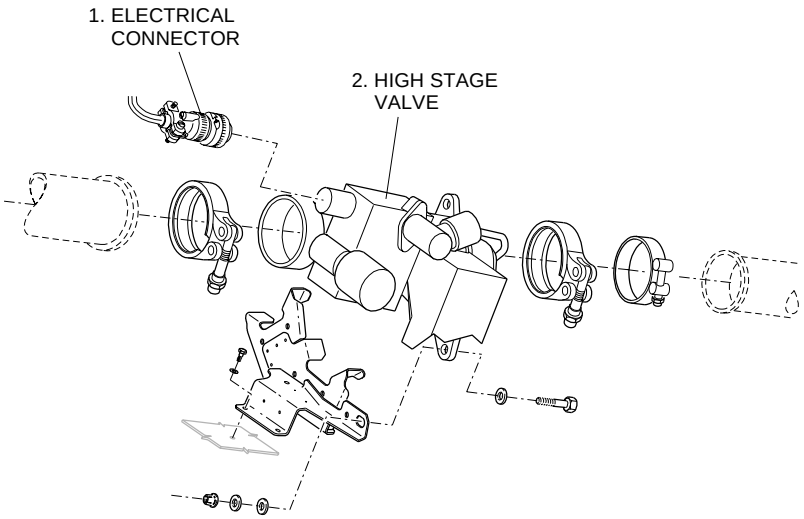
EFFECTIVITY: POST-MOD SB145-36-0045
Electrical Connector - High Stage Valve
Figure 502

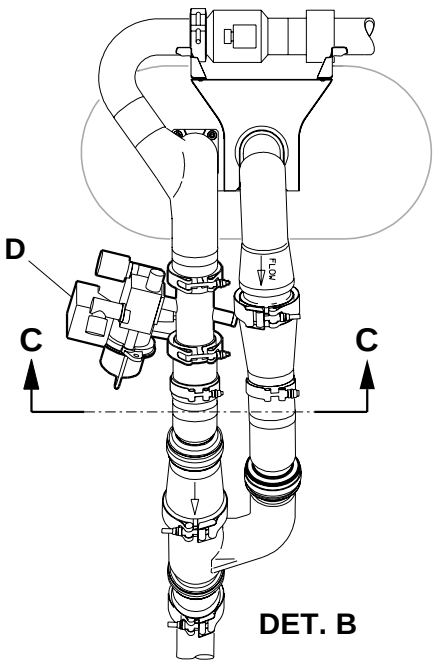




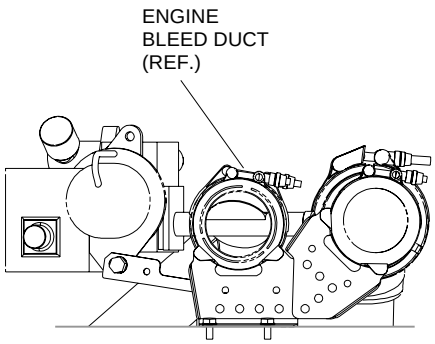
DET. A



DET. D



DET. B



SECTION C–C

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