



EMB145 - EMB135

AIRCRAFT
MAINTENANCE MANUAL

EMERGENCY SHUTDOWN - ADJUSTMENT/TEST

EFFECTIVITY: ALL

1. General

- A. This section gives the procedure to do the operational check of the emergency shutdown system.
- B. The operational check of the emergency shutdown system includes the fuel, hydraulic, bleed air, and anti-ice valves.
- C. The procedures in this section are given in the sequence below. The tasks identified with (♦) are part of the Scheduled Maintenance Requirements Document (SMRD).

TASK NUMBER	DESCRIPTION	EFFECTIVITY
76-20-00-700-801-A ♦	EMERGENCY SHUTDOWN - OPERATIONAL CHECK	ALL



EMB145 - EMB135

AIRCRAFT
MAINTENANCE MANUAL

TASK 76-20-00-700-801-A

EFFECTIVITY: ALL

2. EMERGENCY SHUTDOWN - OPERATIONAL CHECK

A. General

- (1) The operational check of the emergency shutdown system is done with the operation of the emergency shutdown handle.
- (2) These procedures are applicable to the LH and RH engines.

B. References

REFERENCE	DESIGNATION
AMM MPP 06-41-01/100	-
AMM MPP 06-43-00/100	- COMPONENT LOCATION
AMM TASK 20-40-01-860-801-A/200	ENERGIZATION OF THE AIRCRAFT WITH AN EXTERNAL POWER SOURCE
AMM TASK 28-41-00-200-801-A/600	-
AMM TASK 32-00-01-910-801-A/200	LG SAFETY PIN - INSTALLATION AND REMOVAL
WM 30-20-50	-
WM 30-20-51	-
WM 36-11-50	-
WM 36-11-51	-

C. Zones and Accesses

ZONE	PANEL/DOOR	LOCATION
192	192AL	Wing stub (For EMB-145ER, EP, EU, MP, MK, MR and EMB-135ER, KE models)
192	192BR	Wing stub (For EMB-145ER, EP, EU, MP, MK, MR and EMB-135ER, KE models)
155	155DZ	Landing gear bay (For EMB-145LR, LU and EMB-135LR, KL models)
155	155EZ	Landing gear bay (For EMB-145LR, LU and EMB-135LR, KL models)
156	156DZ	Landing gear bay (For EMB-145LR, LU and EMB-135LR, KL models)
156	156EZ	Landing gear bay (For EMB-145LR, LU and EMB-135LR, KL models)
193	193JR	Aft lower fairing
193	193FL	Aft lower fairing
414	414DB	LH Pylon
424	424DB	RH Pylon
412	412AT	LH Upper cowling



EMB145 - EMB135

AIRCRAFT
MAINTENANCE MANUAL

(Continued)

ZONE	PANEL/DOOR	LOCATION
422	422AT	RH Upper cowling

D. Tools and Equipment

Not Applicable

E. Auxiliary Items

ITEM	DESCRIPTION	PURPOSE	QTY
Commercially available	Workstand	Access to the work area	1

F. Consumable Materials

Not Applicable

G. Expandable Parts

Not Applicable

H. Persons Recommended

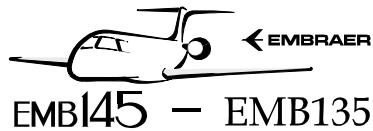
QTY	FUNCTION	PLACE
1	Does the task	Airplane cockpit
1	Monitors the shutoff valves	Aft lower fairing/wing stub or landing gear bay

I. Preparation

SUBTASK 841-002-A

WARNING: BEFORE YOU OPEN THE N2 CIRCUIT BREAKERS AND TO PREVENT INJURY TO PERSONS AND DAMAGE TO THE MATERIAL, MAKE SURE THAT THE SENSORS PITOT 1 - TAT 1/AOA, PITOT 3, AND PITOT 2 - TAT 2/AOA 2, ON THE OVERHEAD PANEL, ARE SET TO OFF.

- (1) On the circuit breaker panel, open these circuit breakers and attach DO-NOT-CLOSE tags to them:
 - FIRE EXTG BTL A1/A2.
 - FIRE EXTG BTL B1/B2.
 - POWERPLANT N2 SIGNAL 1A/2A.
 - POWERPLANT N2 SIGNAL 1B/2B.
- (2) For EMB-145ER, EP, EU, MP, MK, MR and EMB-135ER, KE models:
 - (a) Remove fuel shutoff valve access panels 192AL/192BR (AMM MPP 06-41-01/100).
- (3) For EMB-145LR, LU and EMB-135LR, KL models:



EMB145 – EMB135

AIRCRAFT
MAINTENANCE MANUAL

WARNING: MAKE SURE THAT THE LANDING GEAR SAFETY PINS ARE INSTALLED TO PREVENT INJURIES TO PERSONS AND DAMAGE TO MATERIAL (AMM TASK 32-00-01-910-801-A/200).

- (a) Remove fuel shutoff valve access panels 155DZ/155EZ and/or 156DZ/156EZ (AMM MPP 06-41-01/100).
- (4) Open hydraulic shutoff valve access panels 193JR and/or 193FL (AMM MPP 06-41-01/100).
- (5) Remove engine bleed valve access panels 414DB and/or 424DB (AMM MPP 06-43-00/100).
- (6) Remove the electrical connector from the engine bleed valve, and connect a 28 V DC lamp to pins (A) and (B) of the wire bundle connector (WM 36-11-50/WM 36-11-51).
- (7) Remove engine anti-ice valve access panels 412AT and/or 422AT (AMM MPP 06-43-00/100).
- (8) Remove the electrical connector from the engine anti-ice valve, and connect a 28 V DC lamp to pins (A) and (C) of the wire bundle connector (WM 30-20-50/WM 30-20-51).

J. Operationally Check Emergency Shutdown System ([Figure 501](#)) ([Figure 502](#)) ([Figure 503](#)) ([Figure 504](#)) ([Figure 505](#))

SUBTASK 710-002-A

WARNING: • **MAKE SURE THAT THE CIRCUIT BREAKERS OF THE FIRE EXTINGUISHING BOTTLE ARE OPEN. THIS WILL PREVENT AN ACCIDENTAL BOTTLE DISCHARGE.**
• **BEFORE YOU OPEN THE N2 CIRCUIT BREAKERS AND TO PREVENT INJURY TO PERSONS AND DAMAGE TO THE MATERIAL, MAKE SURE THAT THE SENSORS PITOT 1 - TAT 1/AOA, PITOT 3, AND PITOT 2 - TAT 2/AOA 2, ON THE OVERHEAD PANEL, ARE SET TO OFF.**

CAUTION: MAKE SURE THAT THE CIRCUIT BREAKERS RELATED TO THE POWERPLANT N2 SIGNAL ARE OPEN. THIS WILL REMOVE THE SIGNAL TO CLOSE THE ANTI-ICE VALVE WITH THE ENGINE STOPPED.

NOTE: This check is the same for the LH and RH engines. Set the engine anti-ice and engine bleed valves correctly (LH or RH) to the related system side to be examined.

- (1) Energize the aircraft. Refer to [AMM TASK 20-40-01-860-801-A/200](#).
 - (a) Set the ice detection OVERRIDE switch to the "ENG" position. Push ENG AIR INLET pushbuttons 1 or 2 to send the open signal to the engine anti-icing valve.
Result:
 - 1 The striped bar at the pushbutton goes off.
 - 2 The 28 V DC lamp, on the wire bundle connector of the engine anti-ice valve, goes off. This shows that a signal was sent to open the anti-ice valve.
 - (b) Push BLEED pushbuttons 1 or 2 to send the open signal to the engine bleed valve.

Result:

- 1 The striped bar at the pushbutton goes off.
 - 2 The 28 V DC lamp, on the wire bundle connector of the engine bleed valve, will come on. This shows that a signal was sent to open the bleed valve.
 - (c) Pull the emergency shutdown handle and make sure that:
- Result:
- 1 The advisory messages "E1(2) FUELSOV CLSD" and "E1(2) HYDSOV CLSD" are shown on the EICAS.
 - 2 The 28 V DC lamp, on the wire bundle connector of the engine anti-ice valve, comes on. This shows that a signal was sent to close the anti-ice valve.
 - 3 The 28 V DC lamp, on the wire bundle connector of the engine bleed valve, goes off. This shows that a signal was sent to close the bleed valve.
 - 4 Make sure that the position indicator at the hydraulic shutoff valve and fuel shutoff valve shows "CLOSED" position.
 - (d) Move the emergency shutdown handle back to the usual position and examine that:
- Result:
- 1 The advisory messages "E1(2) FUELSOV CLSD" and "E1(2) HYDSOV CLSD" go out of view from the EICAS display.
 - 2 The 28 V DC lamp, on the wire bundle connector of the engine anti-ice valve, goes off. This shows that a signal was sent to open the anti-ice valve.
 - 3 The 28 V DC lamp, on the wire bundle connector of the engine bleed valve, comes on. This shows that a signal was sent to open the bleed valve.
 - 4 Make sure that the position indicator, at the hydraulic shutoff valve and the fuel shutoff valve, shows "OPEN" position.
- (2) Set the ice detection OVERRIDE switch back to the AUTO position. Push the ENG AIR INLET and BLEED pushbuttons to set them back to off.
 - (3) Deenergize the aircraft ([AMM TASK 20-40-01-860-801-A/200](#)).

K. Follow-on

SUBTASK 842-002-A

- (1) Disconnect the 28 V DC lamp from pins (A) and (C), on the wire bundle connector, and install the electrical connector to the engine anti-ice valve.
- (2) Install engine anti-ice valve access panels 412AT and/or 422AT ([AMM MPP 06-43-00/100](#)).
- (3) Disconnect the 28 V DC lamp from pins (A) and (B), on the wire bundle connector, and install the electrical connector to the engine bleed valve.
- (4) Install engine bleed valve access panels 414DB and/or 424DB ([AMM MPP 06-43-00/100](#)).
- (5) Close hydraulic shutoff valve access panels 193JR and/or 193FL ([AMM MPP 06-41-01/100](#)).



EMB145 - EMB135

AIRCRAFT
MAINTENANCE MANUAL

- (6) Do an inspection on the fuel quantity indication harness (AMM TASK 28-41-00-200-801-A/600).

NOTE: The inspection of fuel quantity indication harness is a part of Critical Design Configuration Control Limitations (CDCCL) in the Airworthiness Limitations of the Maintenance Review Board Report (MRB).

- (7) For EMB-145ER, EP, EU, MP, MK, MR and EMB-135ER, KE models:
(a) Install fuel shutoff valve access panels 192AL and/or 192BR (AMM MPP 06-41-01/100).

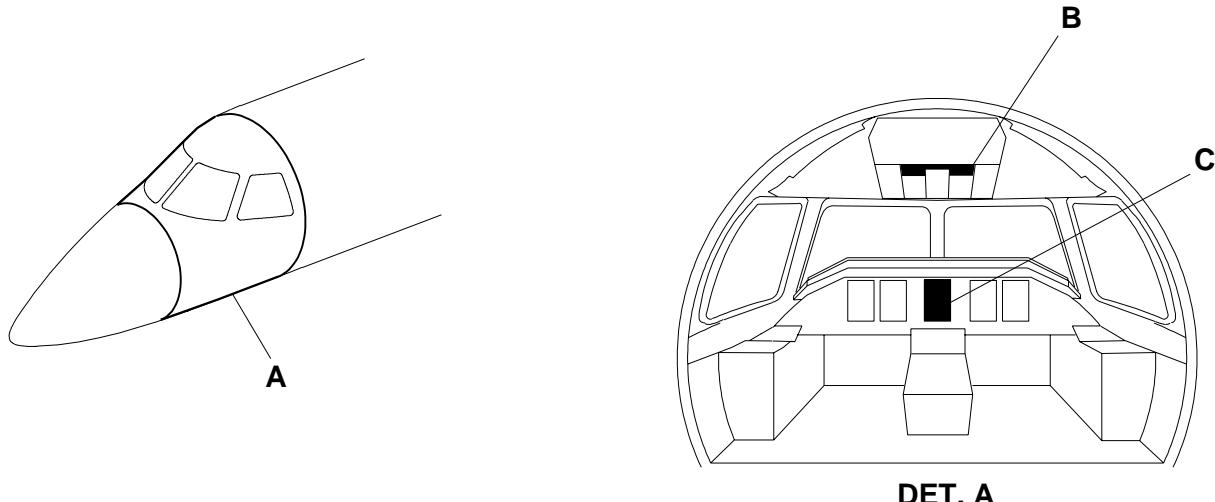
- (8) For EMB-145LR, LU and EMB-135LR, KL models:

WARNING: MAKE SURE THAT THE LANDING GEAR SAFETY PINS ARE INSTALLED TO PREVENT INJURIES TO PERSONS AND DAMAGE TO MATERIAL (AMM TASK 32-00-01-910-801-A/200).

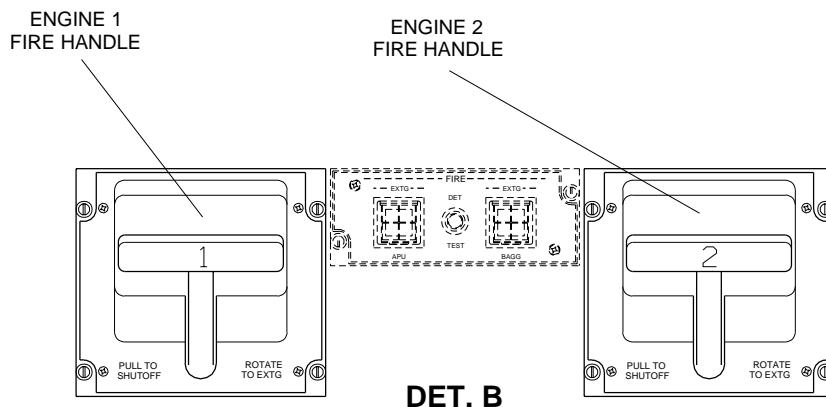
- (a) Install fuel shutoff valve access panels 155DZ/155EZ and/or 156DZ/156EZ (AMM MPP 06-41-01/100).
- (9) On the circuit breaker panel, close the circuit breakers below and remove the DO-NOT-CLOSE tags from them.
- FIRE EXTG BTL A1/A2.
 - FIRE EXTG BTL B1/B2.
 - POWERPLANT N2 SIGNAL 1A/2A.
 - POWERPLANT N2 SIGNAL 1B/2B.

EFFECTIVITY: ALL

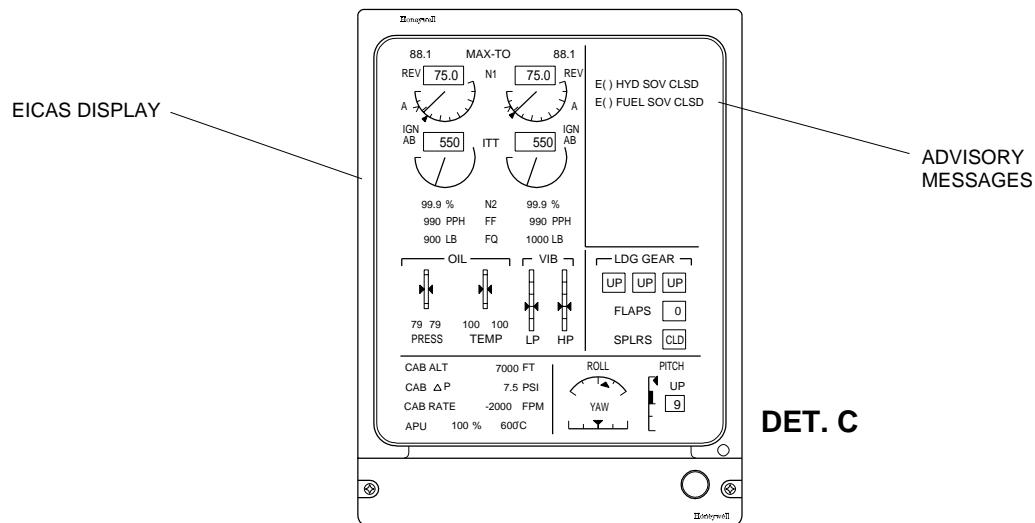
Emergency Shutdown System - Adjustment/Test
Figure 501



DET. A



DET. B

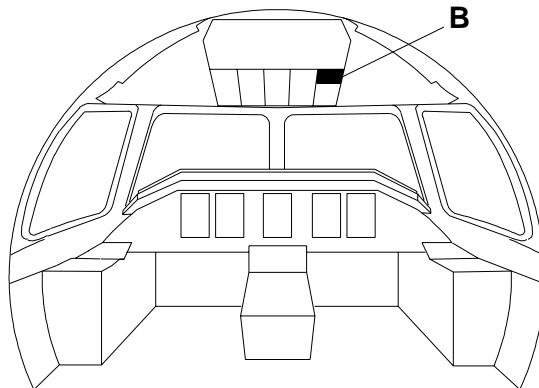
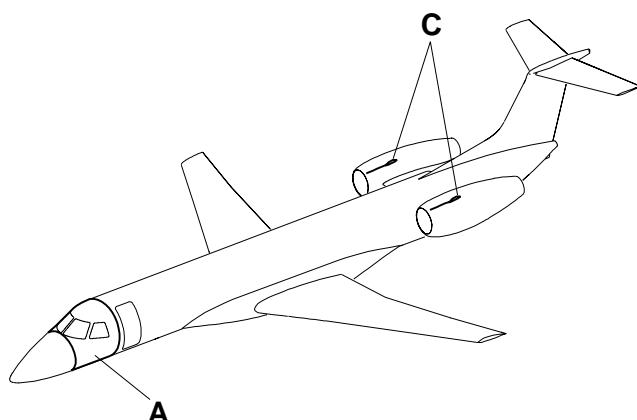


145AMM760034.MCE D

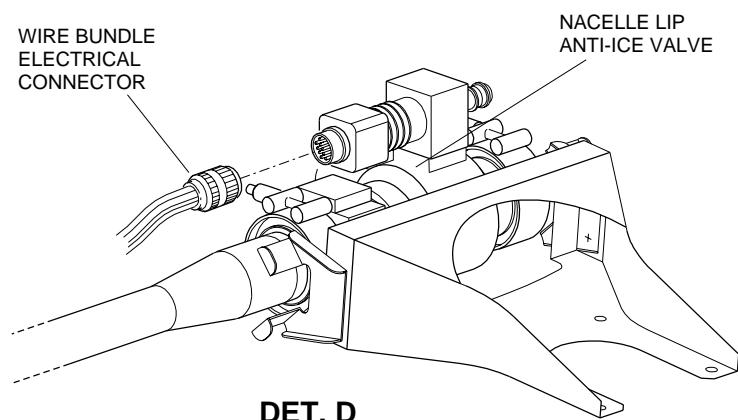
EFFECTIVITY: ALL

Emergency Shutdown System - Adjustment/Test

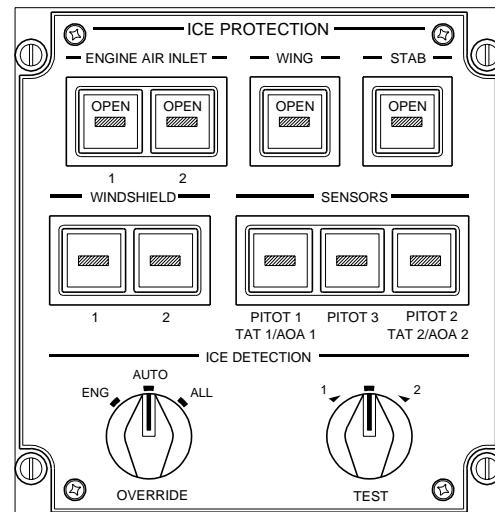
Figure 502 - Sheet 1



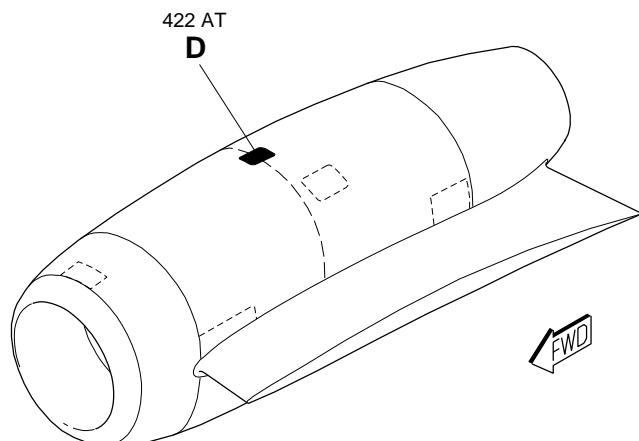
DET. A



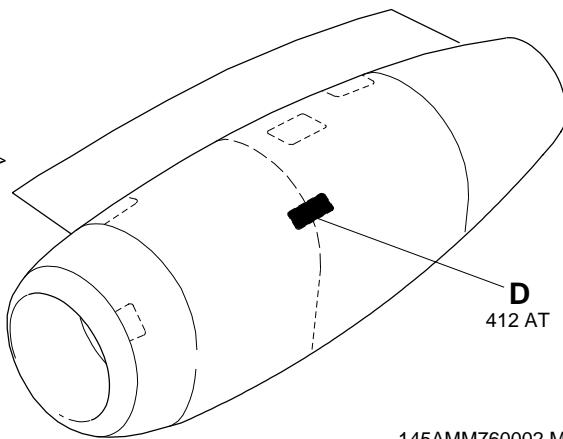
DET. D



DET. B

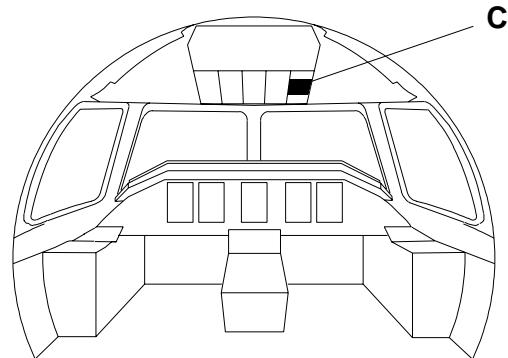
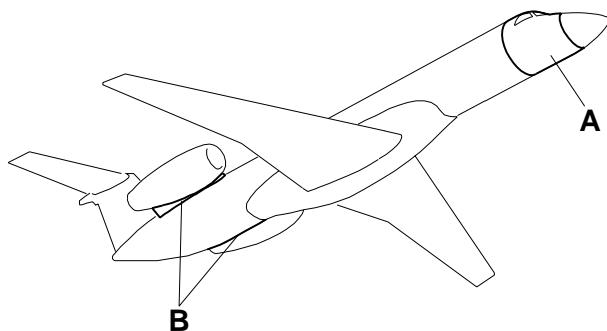
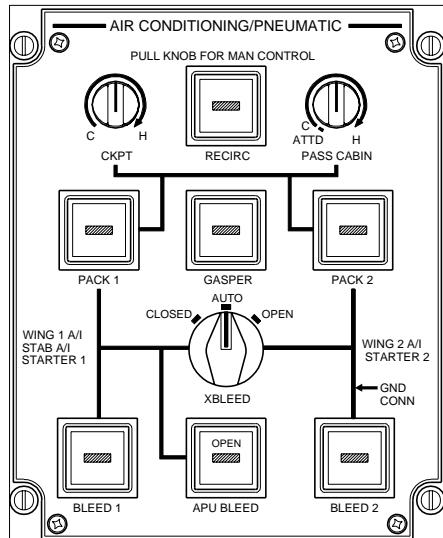
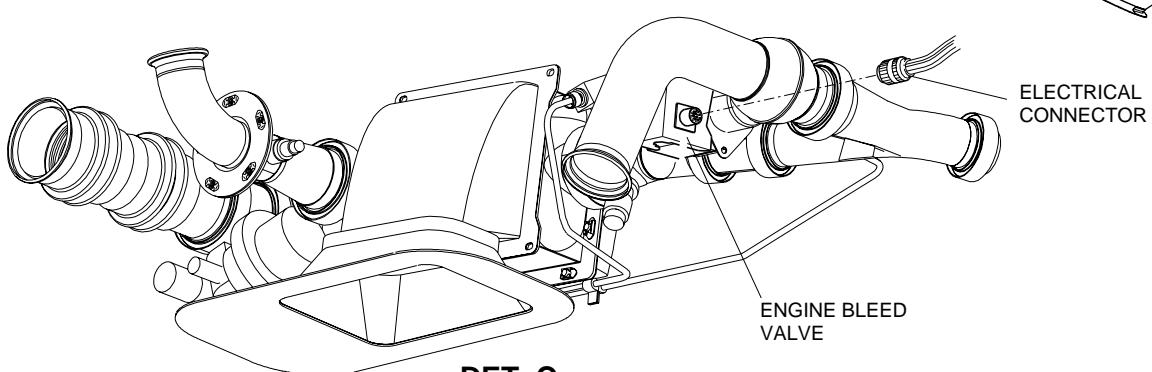
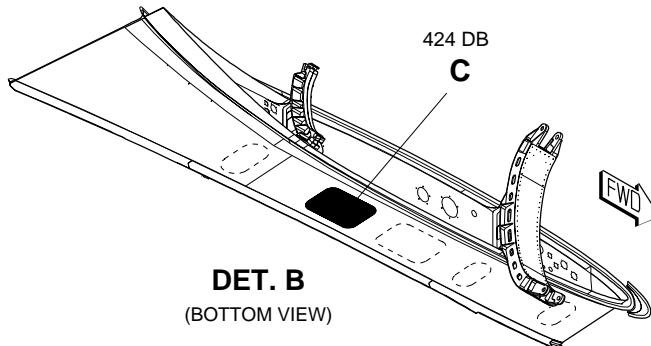
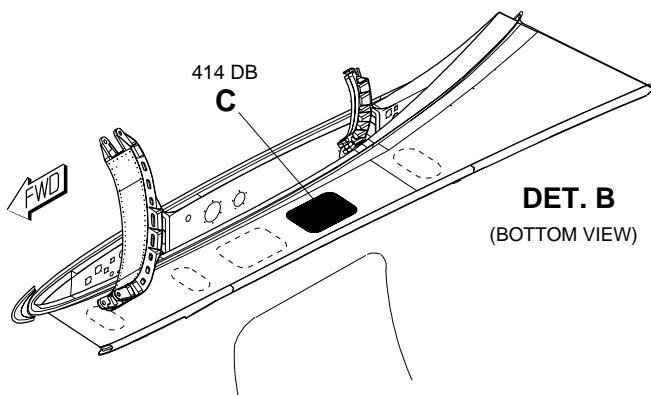


DET. C



145AMM760002.MCE C

EFFECTIVITY: ALL

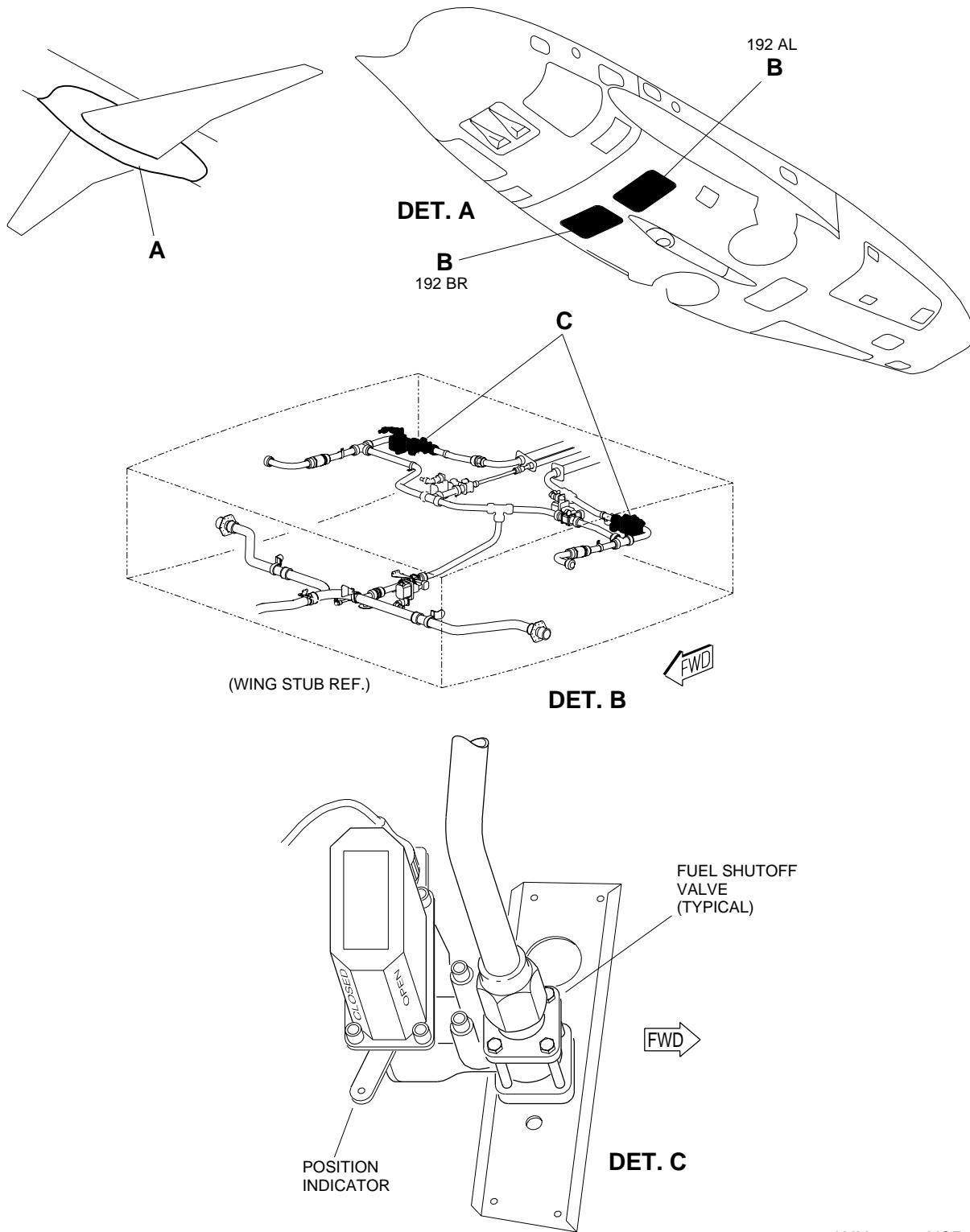
Emergency Shutdown System - Adjustment/Test
Figure 502 - Sheet 2

DET. A

DET. D

DET. C

145AMM760003.MCE B

EFFECTIVITY: EMB-145ER/EP/EU/MP/MR and EMB-135ER/KE MODELS

Emergency Shutdown System - Adjustment/Test

Figure 503

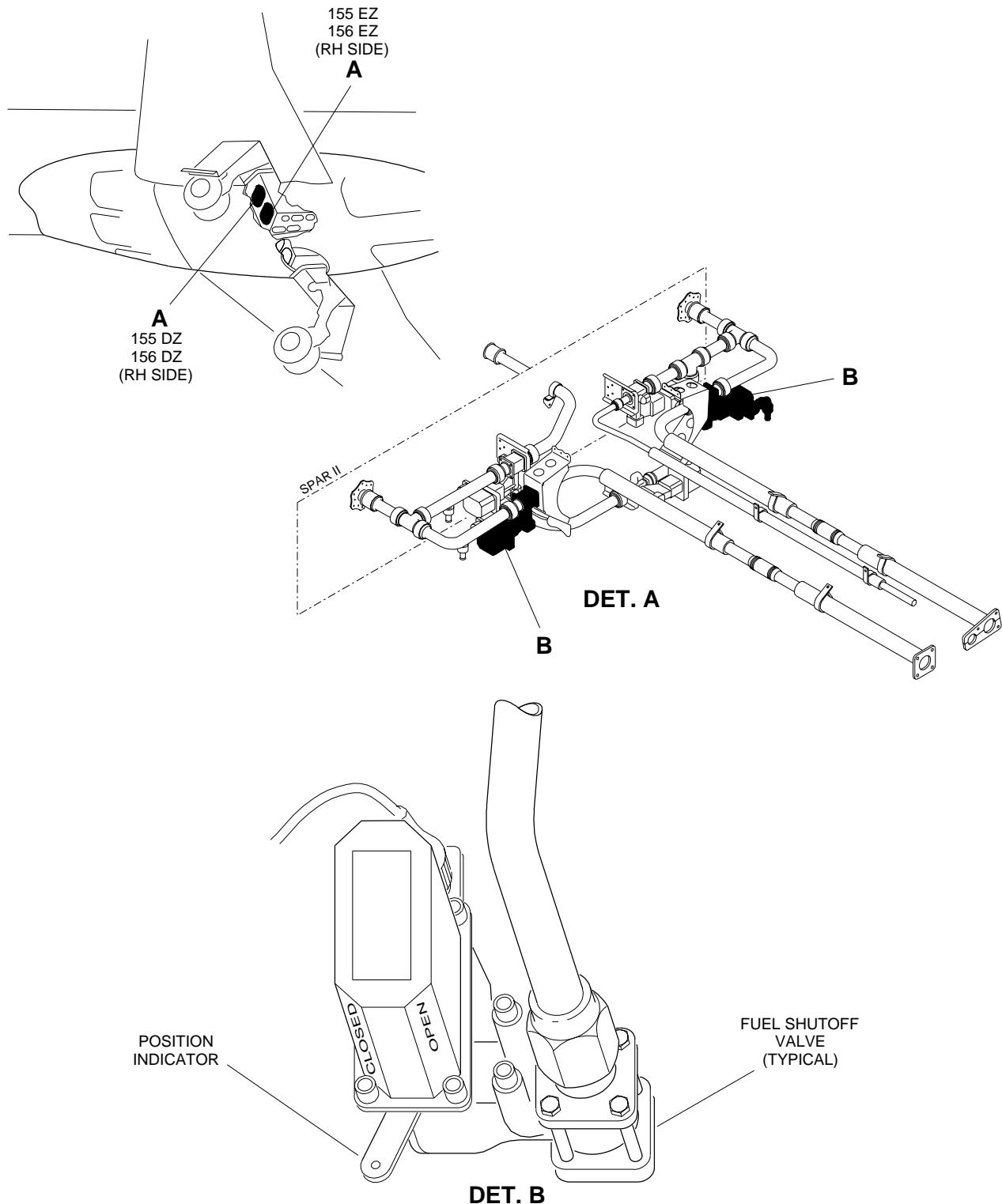


145AMM760004.MCE C

EFFECTIVITY: EMB-145LR/LU and EMB-135LR/KL MODELS

Emergency Shutdown System - Adjustment/Test

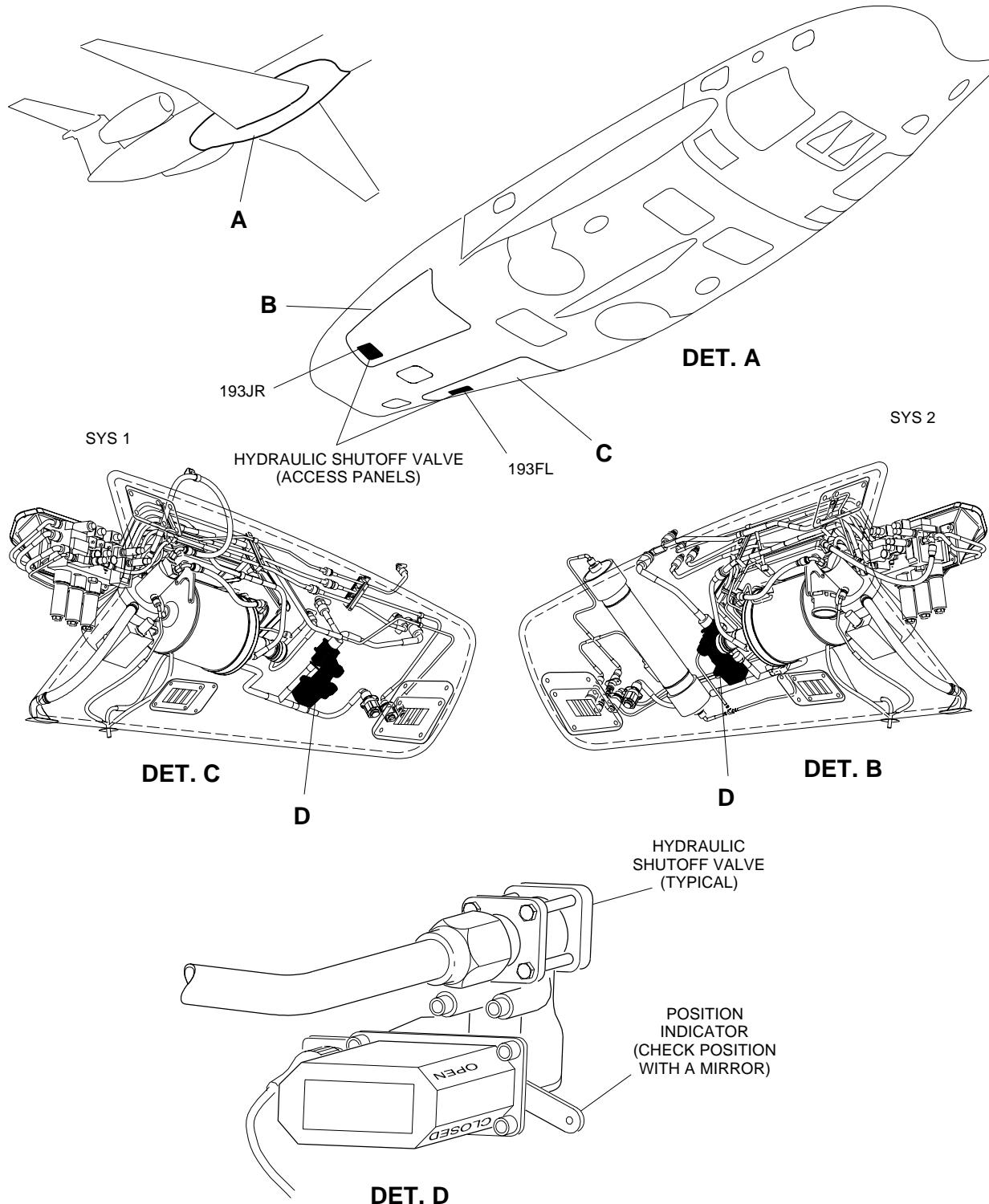
Figure 504



145AMM760035.MCE D

EFFECTIVITY: ALL

Emergency Shutdown System - Adjustment/Test
Figure 505



EM145AMM760060A.DGN