A robust state-space approach to stable predictive control strategies

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Abstract: A means of extending the computational and numerical advantages of the stable GPC approach to predictive strategies employing state-space models is presented.

1 Introduction

In recent years, researchers into predictive control have concentrated on algorithms with guaranteed stability. Many early results were based on a *dead-beat* form of terminal constraint (e.g. [1], [2], [3], [5], [4]); later work deployed less restrictive terminal constraints by allowing either or both the input and output predictions to have an infinite settling time, e.g. [6], [9], [8]. It has been shown [7] that Stable Generalised Predictive Control (SGPC) [4] has significant computational and numerical advantages. Here the SGPC philsophy is extended to state-space models with significant improvements in numerical conditioning for the strategies of [5], [6] and [9]. ¹

2 Background and notation

Let the m^{th} order model be

$$\mathbf{x}_{k+1} = A\mathbf{x}_k + B\mathbf{u}_k \qquad \mathbf{y}_k = C\mathbf{x}_k \tag{1}$$

 $\mathbf{x} \in \mathbb{R}^m$, $\mathbf{u} \in \mathbb{R}^l$, $\mathbf{y} \in \mathbb{R}^p$. Define three types of terminal constraint (from [5], [6], [9]):

$$\mathbf{y}_{i} = 0, \quad i > N \qquad \mathbf{u}_{i} = 0, \quad i \geq N \qquad (2)$$

$$\mathbf{y}_{i} \to 0, \quad i \to \infty \qquad \mathbf{u}_{i} = 0, \quad i \geq N \qquad (3)$$

$$\mathbf{y}_{i} \to 0, \quad i \to \infty \qquad \mathbf{u}_{i} = -K_{q}\mathbf{x}_{i}, \quad i \geq N \qquad (4)$$

$$\mathbf{v}_i \to 0, \quad i \to \infty \qquad \mathbf{u}_i = 0, \quad i \ge N$$
 (3)

$$\mathbf{y}_i \to 0, \quad i \to \infty \qquad \quad \mathbf{u}_i = -K_q \mathbf{x}_i, \quad i \ge N \quad (4)$$

where K_q is an LQ optimal controller. The key element in the SGPC philosophy is to satisfy eqns. (2-4) and also retain an explicit handle on the degrees of freedom within $\mathbf{x}_k, \mathbf{u}_k$. Furthermore replacing open-loop by closed-loop prediction equations avoids the ill-conditioning problems associated with open-loop predictions of unstable systems. Define a nominal performance index as:

$$J = \sum_{i=1}^{\infty} \{ \|\mathbf{y}_i\|_2^2 + \lambda \|\mathbf{u}_{i-1}\|_2^2 \}$$
 (5)

A predictive control is defined by minimizing J with respect to \mathbf{u}_k , k=0,1,... subject to appropriate terminal constraints (e.g. 2-4). The terminal constraints of (2-4) must be satisfied exactly; numerical inaccuracies will remove the guarantee of stability.

3 Input/output predictions

The prediction equations are derived by simulating model (1) forward in time:

$$\begin{bmatrix} \mathbf{Y}_{1} \\ \mathbf{Y}_{2} \end{bmatrix} = \begin{bmatrix} P_{1} \\ P_{2} \end{bmatrix} \mathbf{x}_{0} + \begin{bmatrix} H_{1} \\ H_{2} \end{bmatrix} \mathbf{U}; \quad \mathbf{U} = \begin{bmatrix} \mathbf{u}_{0} \\ \vdots \\ \mathbf{u}_{N-1} \end{bmatrix}$$

$$\begin{bmatrix} \mathbf{Y}_{1} \\ \vdots \\ \mathbf{Y}_{N+1} \end{bmatrix}; \quad \begin{bmatrix} P_{1} \\ P_{2} \end{bmatrix} = \begin{bmatrix} CA \\ \vdots \\ CA^{N+1} \end{bmatrix}$$

$$\begin{bmatrix} CA^{N+1} \\ \vdots \\ CA^{N+m} \end{bmatrix}$$

$$\begin{bmatrix} CB & 0 & \dots & 0 \\ \vdots & \vdots & \vdots \\ CA^{N+m} \end{bmatrix}$$

$$\begin{bmatrix} CA^{N+m} \\ \vdots & \vdots & \vdots \\ CA^{N+m-1}B & CA^{N+m-2}B & \dots & CA^{m}B \end{bmatrix}$$

$$\vdots & \vdots & \vdots & \vdots \\ CA^{N+m-1}B & CA^{N+m-2}B & \dots & CA^{m}B \end{bmatrix}$$

$$(6)$$

The optimal control laws from minimisation of J subject to (2,3,4) are given, without proof.

Terminal constraints (2): These reduce to

$$P_2 \mathbf{x}_0 + H_2 \mathbf{U} = 0 \tag{7}$$

The use of an m-vector of lagrange multipliers μ gives an additional equation

$$0 = L_1 \mathbf{U} + H_1^T P_1 \mathbf{x}_0 + H_2^T \boldsymbol{\mu}; \quad L_1 = H_1^T H_1 + \lambda I$$
 (8)

Terminal constraints (3): These reduce to

$$F^{T}\mathbf{x}_{N} = 0; \quad \mathbf{x}_{N} = [P_{3}\mathbf{x}_{0} + H_{3}\mathbf{U}] H_{3} = [A^{N-1}B, A^{N-2}B, \dots, B]; \quad P_{3} = A^{N}$$
(9)

for an appropriate choice of F that forces \mathbf{x}_N into the stable manifold only ([6]). Then it is easy to to find S_2 such that

$$J = \|P_1 \mathbf{x}_0 + H_1 \mathbf{U}\|_2^2 + \lambda \mathbf{U}^T \mathbf{U} + [P_3 \mathbf{x}_0 + H_3 \mathbf{U}]^T S_2 [P_3 \mathbf{x}_0 + H_3 \mathbf{U}]$$
(10)

The control law is given by eqn.(9) and

$$\begin{array}{rcl} 0 & = & L_2 \mathbf{U} + [H_1^T P_1 + H_3^T S_2 P_3] \mathbf{x}_0 + H_3^T F^T \boldsymbol{\mu} \\ L_2 & = & [H_1^T H_1 + \lambda I + H_3^T S_2 H_3] \end{array} \tag{11}$$

Terminal constraints (4): Y_1 is as above, but

$$\mathbf{x}_{i+1} = [A - BK_a]\mathbf{x}_i; \quad \mathbf{u}_i = -K_a\mathbf{x}_i; \quad i \ge N$$
 (12)

¹Saturation constraints and non-zero set-points are omitted here for brevity, but the results presented apply equally well

where $\mathbf{y}_i = C\mathbf{x}_i$ and K_q is the optimal LQ controller minimising J of eqn. (5). Hence it is easy to compute [9] S_3 such that

$$J = \|P_1 \mathbf{x}_0 + H_1 \mathbf{U}\|_2^2 + \lambda \mathbf{U}^T \mathbf{U} + [P_3 \mathbf{x}_0^T + H_3 \mathbf{U}]^T S_3 [P_3 \mathbf{x}_0^T + H_3 \mathbf{U}]$$
(13)

The control law is given as:

$$\mathbf{U} = -M\mathbf{x}_{0}
L_{3} = [H_{1}^{T}H_{1} + \lambda I + H_{3}S_{3}H_{3}]
M = L_{3}^{-1}[H_{1}^{T}P_{1} + H_{3}^{T}S_{3}P_{3}]$$
(14)

Weakness of simple approaches: For unstable A the numerical conditioning of (7,8) and (11,9) and (14) can be poor because $H_i, P_i, L_i, i=1,2,3$ all involve $A^j, j=0,1,...,N+m$; this can lead to numerical inaccuracies.

4 SGPC approach

SGPC uses prestabilisation of the model so that matrices A^i with large elements do not arise.

Theorem 1: Consider inputs of the form

$$\mathbf{u}_{i} = -K_{d}\mathbf{x}_{i} + \mathbf{t}_{i}$$
 $i = 0, ..., N - m - 1$
 $\mathbf{u}_{i} = -K_{d}\mathbf{x}_{i}$ $i > N - m$ (15)

where K_d is a dead-beat state feedback. Terminal constraints (2) are satisfied for any $\mathbf{t}_i, i = 0, ..., N - m - 1$. **Proof:** $[A - BK_d]^m = 0$, hence $\mathbf{x}_{N+i} = 0$. The implied prediction equations arising from using (15) where $\bar{\Phi} = A - BK_d$ are

$$\begin{bmatrix} \mathbf{y}_1 \\ \vdots \\ \mathbf{y}_N \end{bmatrix} = P_4 \mathbf{x}_0 + H_4 \mathbf{T}; \mathbf{T} = \begin{bmatrix} \mathbf{t}_0 \\ \vdots \\ \mathbf{t}_{N-m-1} \end{bmatrix}; P_4 = \begin{bmatrix} C\Phi \\ \vdots \\ C\Phi^N \end{bmatrix}$$

$$H_4 = \begin{bmatrix} CB & 0 & \dots & 0 \\ \vdots & \vdots & \vdots \\ C\Phi^{N-1}B & C\Phi^{N-2}B & \dots & C\Phi^{m-1}B \end{bmatrix}$$

$$(16)$$

$$\mathbf{U} = M_4 \mathbf{x}_0 + N_4 \mathbf{T}; \quad M_4 = \begin{bmatrix} -K_d \\ \vdots \\ -K_d \Phi^{N-1} \end{bmatrix}$$

$$N_4 = \begin{bmatrix} I & 0 & \dots & 0 \\ -K_d B & I & \dots & 0 \\ \vdots & \vdots & \vdots \\ -K_d \Phi^{N-2} B & -K_d \Phi^{N-3} B & \dots & -K \Phi^{m-2} B \end{bmatrix}$$

The performance index J is minimised by

$$\mathbf{T} = -[H_4^T H_4 + \lambda N_4^T N_4]^{-1} [H_4^T P_4 + \lambda N_4^T M_4] \mathbf{x}_0 \quad (18)$$

Combining this with eqn.(15) gives **U**.

Remark 1. The prediction equations for terminal constraints (3,4)) are similar to (16,17) except that $\Phi = A - BK_q$ and the vector **T** has N terms.

Remark 2. T constitute the degrees of freedom. It is easy to show (space does not permit) that there is a one-to-one relationship between U and T.

Remark 3. Selecting the inputs to be of the form

$$\mathbf{u}_{i} = -K_{q}\mathbf{x}_{i} + \mathbf{t}_{i} \qquad i = 0, ..., N - 1 \\ \mathbf{u}_{i} = -K_{2}\mathbf{x}_{i} \qquad i \geq N$$

$$(19)$$

is equivalent to terminal constraints (3, 4) where K_2 is zero or K_q respectively.

5 Examples

The optimal control laws of eqns.(8, 7), (14) (for brevity we omit (9, 11)) are computed as $\mathbf{u}_0 = -\hat{K}\mathbf{x}_0$ for various N. Both approaches should give identical \hat{K} . Let the example be

$$A = \begin{bmatrix} 2.6000 & -0.0500 & -0.5000 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \\ C = \begin{bmatrix} 1 & -2.2 & 1.12 \end{bmatrix}; \quad B = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}$$
(20)

Note that as $N \to \infty$, $\hat{K} \to K_q$

$$K_q = \begin{bmatrix} 3.2630 & -0.3788 & -0.6226 & 2.1279 \end{bmatrix}$$
 (21)

	Section 3		Section 4		
N	14	15	14	15	100
	-0.5108	591.6965	3.4001	3.3755	3.2630
\hat{K}	0.0008	-59.2203	-0.3903	-0.3882	-0.3788
	0.1305	-118.3106	-0.6517	-0.6464	-0.6226
	-0.4129	394.3967	2.1944	2.1824	2.1279

Table 1. \hat{K} for terminal constraints (2)

	Section 3			Section 4	
N	8	13	17	8	17
	3.2630	3.2619	22.0189	3.2630	3.2630
\hat{K}	-0.3788	-0.3786	-2.8117	-0.3788	-0.3788
	-0.6226	-0.6230	-5.5989	-0.6226	-0.6226
	2.1279	2.1279	15.5815	2.1279	2.1279

Table 2. \hat{K} for terminal constraints (3) The algorithm of section 4 has no problems and computes \hat{K} correctly even for large N whereas the approach of section 3 quickly goes numerically ill-conditionned and produces spurious results for N > 10.

5.1 Conclusions: Many formulations for algorithms with guaranteed stability are not numerically stable for large prediction horizons. An alternative numerically robust algorithmic formulation for computing the identical implied state-space control laws has been derived. The key idea is prestabilisation of the plant before the predictions used in a predictive control law are computed.

6 References

[1] D.W. Clarke and R. Scattolini, Proc. IEE, Pt. D, 138,
4, pp347-354, 1991
[2] S.A. Heise and J.M. Maciejowski, Proc. 33rd CDC,

Vol. 4, pp3573-3578, 1994
[3] S.S. Keerthi and E.G. Gilbert, 1988, JOTA, 57, 2, pp265-293

[4] B. Kouvaritakis, J.A. Rossiter and A.O.T. Chang,

1992, Proc. IEE, 139,4, 349-262 [5] E. Mosca, and J. Zhang, 1992, Automatica, 28, 6, pp1229-1233

[6] J.B. Rawlings and K.R. Muske, 1993, IEEE Trans. AC, 38, 10, pp1512-1516

[7] J.A. Rossiter and B. Kouvaritakis, 1994, Proc. IEE Pt. D., 141, 3, pp154-162
[8] J.A.Rossiter, J.R. Gossner and B.Kouvaritakis, 1996, IEEE Trans. AC, 41, 10, 1522-27

[9] P.O.M.Scokaert and J. B. Rawlings, Proc. IFAC96, 1996, Vol. M, pp109-114