Kepler's Law

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Abstract

Kepler's law is a cornerstone of orbital mechanics.

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1 Overview

Kepler's law is a cornerstone of orbital mechanics.

- $1. \, \, {\tt env}$
- 2. ldd

2 Geometry of Kepler's Law

Disagreement with this YouTuber True Anomaly vs. Mean Anomaly

3 Mathematics

3.1 Definitions

Definition 1 (The ellipse). Given $\theta \in [0, 2\pi)$, and parameters $a, b \in \mathbb{R}^+$ with a > b the following parametric form defines an ellipse.

$$\epsilon(\theta) = (a\cos\theta, b\sin\theta) \tag{1}$$

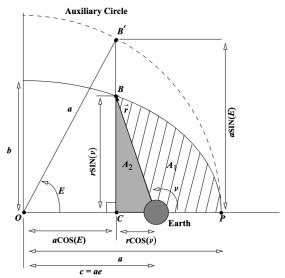


Figure 2-2. Geometry of Kepler's Equation. The eccentric anomaly uses an auxiliary circle as shown. The ultimate goal is to determine the area, A_1 , which allows us to calculate the time.

Figure 1: Vallado's figure 2-2 showing E and ν .

Sec 4.2 TIME-OF-FLIGHT - ECCENTRIC ANOMALY 183

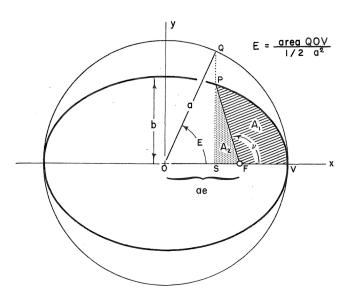


Figure 4.2-2 Eccentric anomaly, E

Figure 2: Figure 4-4 in Bate et~al. showing E and $\nu.$

ACQ will be defined as the eccentric anomaly, E, and it will be shown that the relation between M and E is given by Kepler's equation.

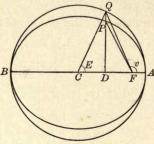


Fig. 28.

From the law of areas and the properties of the auxiliary circle, it follows that

$$\frac{M}{2\pi} = \frac{\text{area } AFP}{\text{area ellipse}} = \frac{\text{area } AFQ}{\text{area circle}}.$$

Area $AFQ = \text{area } ACQ - \text{area } FCQ = \frac{a^2E}{2} - \frac{a}{2} ae \sin E.$

Therefore

$$\frac{M}{2\pi} = \frac{a^2}{2} \frac{(E - e \sin E)}{\pi a^2};$$

 $\begin{cases} M = E - e \sin E, \\ FP = r = \frac{a(1 - e^2)}{1 + e \cos v} = \sqrt{PD^2 + FD^2} = a(1 - e \cos E), \end{cases}$

Figure 3: Moulton's figure 28 showing E and ν .

Definition 2 (Eccentricity of the ellipse). The eccentricity is a scalar parameter $e \in (0,1)$ and can be expressed in terms of fundamental parameters of the ellipse where a > b as

$$e = \frac{c}{a} = \sqrt{1 - \left(\frac{b}{a}\right)^2} \tag{2}$$

Definition 3 (Mean anomaly). Kepler's Law¹ defines the mean anomaly as the angular measure $M(e, E): (0, 1) \times [0, 2\pi) \mapsto [0, 2\pi)$ as

$$M(e, E) = E - e\sin E \tag{3}$$

Theorem 4 (Continuity of the mean anomaly). The mean anomaly as defined in definition 3 is a continuous function.

Proof. To prove continuity show that for any two points p and q in the domain there exists a majorization constant K such that

$$M(p) - M(q) \le K|p - q| \tag{4}$$

Spoiler alert: the majorization constant is 2π .

Observation: given the continuity of the mean anomaly, one may use Newton's method Gautschi 2011, §4.6 to solve the nonlinear equation.

References

Bate, Roger R et al. (2020). Fundamentals of astrodynamics. Courier Dover Publications. Gautschi, Walter (2011). Numerical analysis. Springer Science & Business Media. Kaula, William M (2013). Theory of satellite geodesy: applications of satellites to geodesy. Courier Corporation.

Moulton, Forest Ray (1970). An introduction to celestial mechanics. Dover.

Vallado, D.A. (2022). Fundamentals of Astrodynamics and Applications. 5th ed. Microcosm Press;

¹Bate et al. 2020, eq 4.5 Moulton 1970, p.159 Vallado 2022, §2.2, Kaula 2013, pp. 3–19