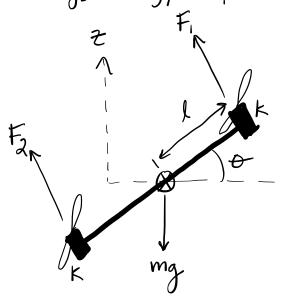
- 1. Continuous nonlinear dynamics
- 2. Controller design
- 3. Measurements

Consider a simplified planar model of a drone: (no drag, no gyroscopic forces, no nonlinear aero effects)



States: 0,0, x, x, Z, Z, Z, k
Controls: U,, U2: motor speed

parameters: m: mass (known)

K: F= KU, (unknown)

1: leigth of arm (known)

Tyy: moment of inestin (known)

Reunte:

$$F = F_1 + F_2$$

$$96^{\circ} + \theta$$

$$T = l(F_1 - F_2)$$

$$mg$$

Dynamics:

(to make simplify) Rewrite contros: j, = W, - W2 j2 = 4,+42

Continuous State Space

write in form:
$$\frac{\partial}{\partial x} = f(\overline{x}, \overline{u})$$

$$\vec{\chi} = \frac{d}{dt} \begin{bmatrix} \vec{\theta} \\ \vec{\lambda} \\ \vec{\chi} \\ \vec{z} \end{bmatrix} = \begin{bmatrix} \vec{\phi} \\ \chi \\ \vec{\chi} \\ -Fsinp/m \\ \vec{z} \\ (Fcos\theta - mg)/m \end{bmatrix} = \begin{bmatrix} \vec{\phi} \\ lk/I_{yy} j, \\ \vec{\chi} \\ -KsinP/m j_{\alpha} \\ \vec{z} \\ (kj_{\alpha}cos\theta - mg)/m \end{bmatrix}$$

write in control affine form:

$$\dot{\vec{\chi}} = \frac{d}{dt} \begin{bmatrix} \dot{\theta} \\ \dot{\theta} \\ \dot{\chi} \\ \dot{\chi} \\ \dot{z} \end{bmatrix} = \begin{bmatrix} \dot{\theta} \\ \dot{\chi} \\ \dot{\chi} \\ \dot{z} \\ \dot{z} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} \dot{j}_{1} + \begin{bmatrix} 0 \\ 0 \\ -K \sin \theta / m \end{bmatrix} \dot{j}_{\lambda}$$

add the unknown parameter k as a static state:

$$\frac{1}{x} = \frac{d}{dt} \begin{bmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial$$

Allows us to write dyamics as:

$$\dot{\vec{\chi}} = f(\vec{\chi}, \vec{u}) = f_{\bullet}(\vec{\chi}) + f_{\bullet}(\vec{\chi}) j_{\bullet} + f_{\bullet}(\vec{\chi}) j_{\bullet}$$

Measurements:

Consider 3 options:



is 3 is are not states, need to rewrite as states, based on dyamics egns.

seperale controls:
$$\vec{x} = f(\vec{x}) + f(\vec{x})u_1 + \dots + f(\vec{x})u_n$$

$$\frac{1}{x} = \frac{d}{dt} \begin{bmatrix} \frac{\partial}{\partial t} \\ \frac{\partial}{\partial t} \\ \frac{\partial}{\partial t} \end{bmatrix} = \begin{bmatrix} \frac{\partial}{\partial t} \\ 0 \\ \frac{\partial}{\partial t} \\ \frac{\partial}{\partial t} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ -6in\theta/m \\ 0 \\ \cos\theta/m \end{bmatrix} + \begin{bmatrix} 0 \\ T_{\gamma\gamma} \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} \gamma_{\gamma}$$

$$\frac{1}{5} \quad f_{1} \quad f_{2}$$

$$\frac{\dot{\vec{x}}}{\vec{x}} = \int_{o} (\vec{x}) + \int_{i} (\vec{x}) F + \int_{z} (\vec{x}) \gamma_{i}$$