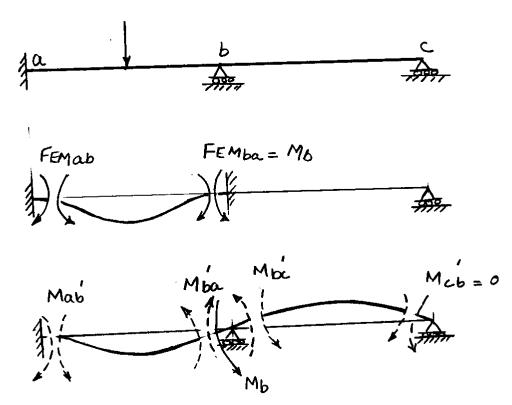
When in a structure the exterior span is simply supported at the exterior support point, it is possible to introduce certain modifications that significantly cut down the amount of computations needed to analyze the structure.



Consider the structure shown about that has a exterior span having a roller at exterior end.

Exterior span having a roller at exterior end.

If joint "b" is initially clamped, Fixed end moments will femba = Mb will develop there.

Will FEMba = Mb will develop there.

As the joint "b" is released, moments Mba and Mba will be induced alongwith carry over moment Mcb at end "c".

However Mcb = 0 as it is a roller end.

$$\Rightarrow \boxed{\Theta c = -\frac{\theta b}{2}}$$

$$Mbc' = 2E K_{bc} (20b + 0c) \qquad ---- 3$$

From 2 and 3 we have

$$Mbc' = 2E \, Kbc \left(20b - \frac{Ob}{2}\right)$$

$$Mbc' = 3EKbc \theta b = 4EKbc \theta b$$

where

 $Kbc' = Modified Relative = \frac{3}{4}Kbc$ 

Stiffness of member be

From Equilibrium considerations we have:

From Slope-Deflection Equations, Egn (4) and (5) we have:

$$4E\theta b\left(K_{ba}+K_{bc}^{m}\right)=-Mb$$

Solving for "06" we have:
$$0b = -\frac{Mb}{4E(Kba + Kbc)}$$

Substituting value of "Bb" from Egn (7) into Egn (8) and Egn (4) we have

$$Mba' = -\frac{Kba}{Kba + Kbc}Mb$$
and
$$Mbc' = -\frac{Kbc}{Kba + Kbc}Mb$$

Note that in about Distribution Factors, modified Relative Stiffness of member "bc" has been used.

General Form of Egn (1) is

Modified Relative

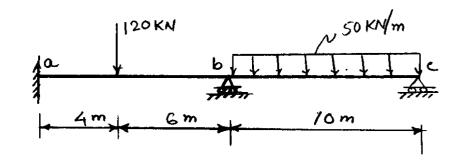
Shiffness for Entorior stons m

with Roller on Exterior End = Knf = 1/2 Knf. — 11

#### MODIFIED RELATIVE STIFFNESS FOR SELECT CASES

Loading Case	Stiffness Mnf/On	Knf	Description of Condition
Normal	4 E Knf		Mnf on f
Sumple Support	3EKnf = 4E(Km)	2 Knf	Mnf $\theta f = -\theta \eta_2$
Symmetric	2E Knf = 4E(Knf)	1/2 Knf	$\frac{Mnf}{n} = -\theta n$
Antisymmetric	6E Knf = 4E(Knf)	<u>3</u> Knf	Must on $Of = On$ Mfn = -Must $Of = Mns$ Mfn = Mns

### Nomerical Enample using Modified Stiffness



Solve the above structure using modified moment distribution method

# Stiffness and Relative Stiffness

$$Kab = Kba = \overline{L} = \overline{I_0} = K$$

$$Kbc = Kcb = \overline{L} = \overline{I_0} = K$$

$$Kbc = Kcb = \frac{3}{4}Kbc = \frac{3}{4}K$$

### Distribution Factors

$$Dba = \frac{Kab}{Kab + Kbc} = \frac{K}{K + \frac{3}{4}K} = 0.571$$

$$Dbc = \frac{Kbc}{Kba + Kbc} = \frac{\frac{3}{4}K}{K + \frac{3}{4}K} = 0.429$$

$$Dcb = \frac{Kcb}{Kcb} = \frac{K}{K} = 1.0$$

### Fixed End Moments

Fixed End Moments

FEMab = 
$$\frac{Pab^2}{L^2}$$
 =  $-\frac{120 \times 4 \times 6^2}{10^2}$  =  $-172.8 \text{ KN-m}$ 

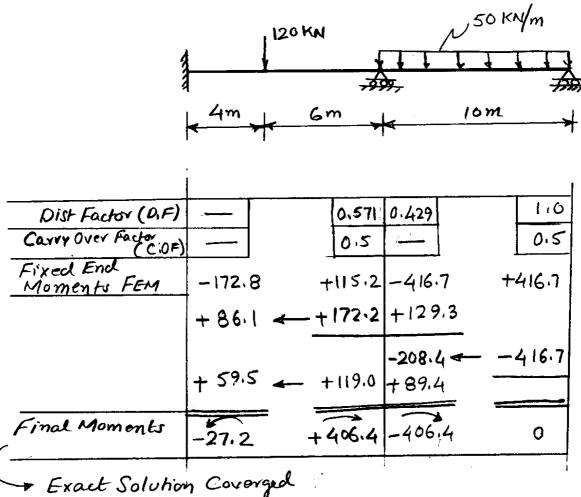
FEMba =  $\frac{Pa^2b}{L}$  =  $-\frac{120 \times 4^2 \times 6}{10^2}$  =  $+115.2$  4

FEMbc =  $-\frac{\omega L^2}{12}$  =  $-\frac{50 \times 10^2}{12}$  =  $-416.7$  4

FEMcb =  $+\frac{\omega L^2}{12}$  =  $+416.7$  4

#### Moment Distribution

### Numerical Example using modified Stiffness.



to in 2 cycles.

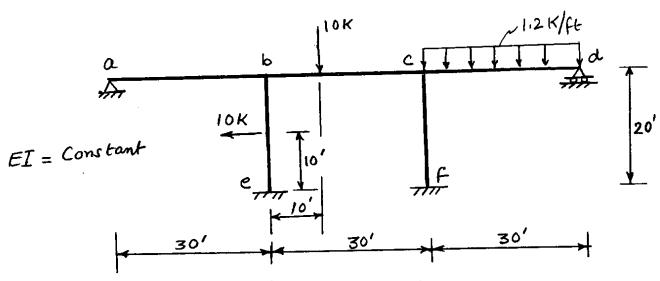
### ALTERNATE DISTRIBUTION

D.F			0.571	0.429		1,0
C.0			0.5			0.5
FEM	-172.	B +	115.2	-416. -208.		+416.7
	+145	6 🚣 🛨	291,2	+218	.7	
Final Moments	-27.2		406.4	-406	4	O

> Exact Solution Converged to in 1 cycle

# Example Problem

Find the end moments for all the members of the frame shown below. The frame is restrained against sway at pt a.



# Stiffnesses & Relative Stiffnesses

$$Kab = Kba = Kbc = Kcb = Kcd = Kdc = \frac{I}{30} = K$$

$$Kbe = Keb = Kcf = Kfc = \frac{I}{20} = 1.5K$$

# Modified Stiffnesses

$$K_{ba}^{m} = \frac{3}{4} K_{ba} = 0.75 K$$
 $K_{cd}^{m} = \frac{3}{4} K_{cd} = 0.75 K$ 

## Distribution Factors

$$Dbi = \frac{Kbi}{\sum Kbi}$$

$$Dab = \frac{K}{K} = 1.0$$

$$Dba = \frac{0.75K}{0.75K + K + 1.5K} = 0.231$$

$$Dbc = \frac{K}{(0.75 + 1 + 1.5)K} = 0.307$$

$$Dbe = \frac{1.5 K}{(0.75 + 1 + 1.5)K} = 0.462$$

## Example Problem

#### Fixed End Moments

$$FEMbc = -\frac{\rho_0 b^2}{\ell^2} = -\frac{10 \times 10 \times 20^2}{30^2} = -44.4 \text{ K-ft}$$

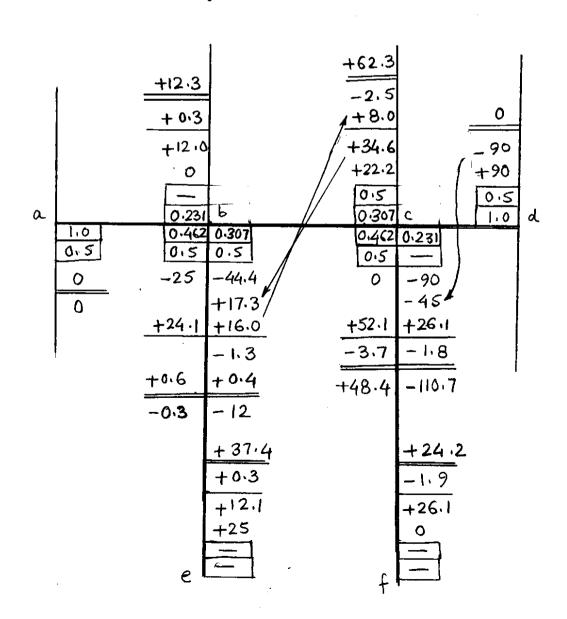
$$FEMcb = -\frac{\rho \alpha^2 b}{\ell^2} = \frac{10 \times 10^2 \times 20}{30^2} = +22.2 \text{ K-ft}$$

$$FEMcd = -\frac{\omega \ell^2}{12} = -\frac{112 \times 30^2}{12} = -90$$

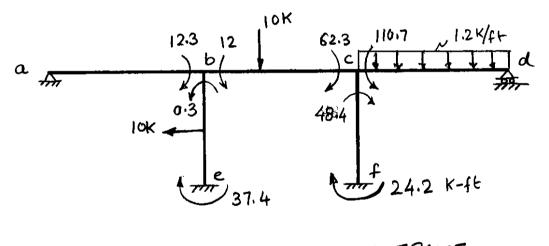
$$FEMdc = \frac{\omega \ell^2}{12} = \frac{-10 \times 10 \times 10}{20^2} = -25$$

$$FEMbe = -\frac{\rho \alpha b^2}{\ell^2} = \frac{-10 \times 10 \times 10}{20^2} = -25$$

$$FEMeb = -\frac{\rho \alpha^2 b}{\ell^2} = \frac{-10 \times 10 \times 10}{20^2} = +25$$



### Example Problem



END MOMENTS IN THE FRAME