## Machine Design Homework 1

June 9, 2022

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```
[1]: # Notebook Preamble
import sympy as sp
import matplotlib.pyplot as plt
from IPython.display import display, Markdown

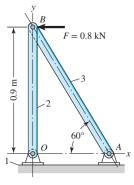
plt.style.use('maroon_ipynb.mplstyle')
```

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### 1.1 Given



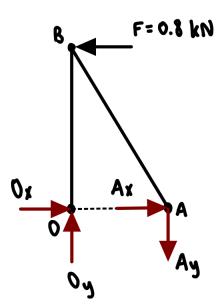
Problem 3-3

### 1.2 Find

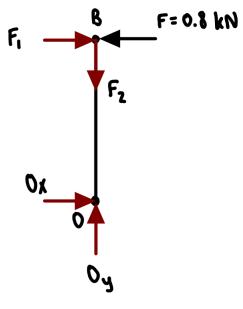
Sketch a free-body diagram of each element in the figure above. Compute the magnitude and direction of each force using an algebraic or vector method.

### 1.3 Solution

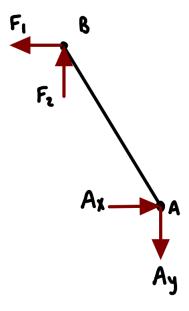
The overall structure has a free body diagram like so,



Separating the structure into individual elements would yield more equations to solve the system,



 ${\rm Link}\ 2$ 



Link 3

The 800 N force is a point load, which means it can only be applied to one link. This is why the load is shown in one FBD and not the other. Changing the link the point load is applied to will change the values for  $F_1$  and  $F_2$ , but it will not change the values of the reactions at O and A.

```
[2]: v = Ox, Oy, Ax, Ay, F1, F2 = sp.symbols('O_x O_y A_x A_y F_1 F_2')
y, x = sp.S(0.9), sp.S(0.9)/sp.tan(60*sp.pi/180) # Vertical and horizontal_
distance of the structure

# Overall structure equations
eq1 = sp.Eq(Ox + Ax, 800) # Sum in x
eq2 = sp.Eq(Oy - Ay, 0) # Sum in y
```

```
eq3 = sp.Eq(-Ay*x + 800*y, 0) # Sum the moments about 0
# Link 2 equations
eq4 = sp.Eq(0x + F1, 800) # Sum in x
eq5 = sp.Eq(0y, F2) # Sum in y
eq6 = sp.Eq(F1*y, 800*y) # Sum the moments about O
# Link 3 equations (don't need to use and still get the same answer)
\# eq4 = sp.Eq(Ax, F1)
\# eq5 = sp.Eq(Ay, F2)
\# eq6 = sp.Eq(F2*x, F1*y)
eqs = [eval(f'eq{i}') for i in range(1, 7)]
display(Markdown('**Equations:**'))
for eq in eqs:
    display(eq)
display(Markdown('---'))
A, b = sp.linear_eq_to_matrix(eqs, v)
matrix_eqn = sp.Eq(sp.MatMul(A, sp.Matrix(v)), b)
display(matrix_eqn)
display(Markdown('---'))
sol = sp.solve(matrix_eqn)
display(Markdown('**Solution:**'))
for val in v:
    display(sp.Eq(val, sol[val]))
```

### **Equations:**

$$\begin{split} A_x + O_x &= 800 \\ -A_y + O_y &= 0 \\ -0.3\sqrt{3}A_y + 720.0 &= 0 \\ F_1 + O_x &= 800 \\ O_y &= F_2 \\ 0.9F_1 &= 720.0 \end{split}$$

$$\begin{bmatrix} 1 & 0 & 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & -0.3\sqrt{3} & 0 & 0 \\ 1 & 0 & 0 & 0 & 1 & 0 \\ 0 & 1 & 0 & 0 & 0 & -1 \\ 0 & 0 & 0 & 0 & 0.9 & 0 \end{bmatrix} \begin{bmatrix} O_x \\ O_y \\ A_x \\ A_y \\ F_1 \\ F_2 \end{bmatrix} = \begin{bmatrix} 800 \\ 0 \\ -720.0 \\ 800 \\ 0 \\ 720.0 \end{bmatrix}$$

### Solution:

$$O_{x} = 0.0$$

$$O_y = 1385.6406460551$$

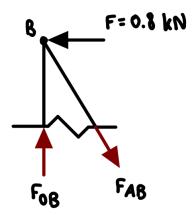
$$A_x = 800.0$$

$$A_y = 1385.6406460551$$

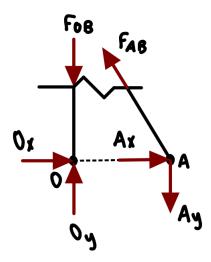
$$F_1 = 800.0$$

$$F_2 = 1385.6406460551 \,$$

The internal stresses of each link may be found by using the method of sections.



Upper Section Cut



Lower Section Cut

```
[3]: F_OB, F_AB = sp.symbols(r'F_{OB} F_{AB}')

# Upper section equations
eq7 = sp.Eq(F_AB*sp.cos(sp.pi/3), 800) # Sum in x
```

```
eq8 = sp.Eq(F_OB, F_AB*sp.sin(sp.pi/3)) # Sum in y

display(eq7)
display(eq8)
display(Markdown('---'))

sol = sp.solve([eq7, eq8])

for val in [F_OB, F_AB]:
    display(sp.Eq(val, sol[val].n()))
```

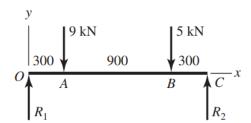
$$\frac{F_{AB}}{2} = 800$$
 
$$F_{OB} = \frac{\sqrt{3}F_{AB}}{2}$$

 $F_{OB} = 1385.6406460551$ 

 $F_{AB} = 1600.0$ 

Link 2 is in compression and link 3 is in tension.

### 2.1 Given



Problem 3–5
Dimensions in millimeters

### 2.2 Find

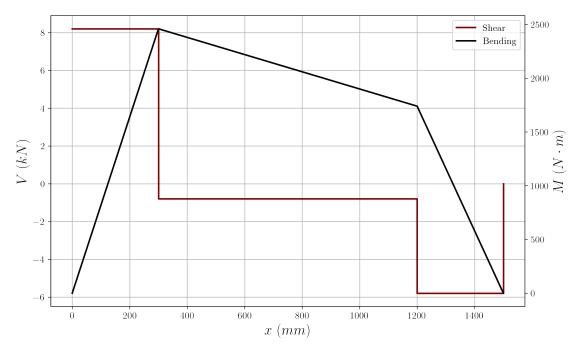
Find the reactions at the supports and plot the shear and bending moment diagrams.

### 2.3 Solution

Due to the large amount of practice, this problem is quite simple and can be plotted instantly without showing calculation.

```
[4]: 0, A, B, C = 0, 300, 300 + 900, 300 + 900 + 300
     V_0, V_A, V_B, V_C = 8.2, 8.2 - 9, 8.2 - 9 - 5, 0 # Values shortly after cross_
     ⇔the point
    M_0, M_A, M_B, M_C = 0, V_0*300, V_0*300 + V_A*900, V_0*300 + V_A*900 + V_B*300
     x = [0, A, A, B, B, C, C]
     V = [V_0, V_0, V_A, V_A, V_B, V_B, V_C]
     x_bending = [0, A, B, C]
     M = [M_O, M_A, M_B, M_C]
     fig, shear = plt.subplots()
     bend = shear.twinx()
     shear.plot(x, V, label='Shear')
     shear.set_xlabel('$x$ ($mm$)')
     shear.set_ylabel('$V$ ($kN$)')
     bend.plot(x_bending, M, label='Bending', color='black')
     bend.set_ylabel(r'$M$ ($N\cdot m$)')
     bend.grid(visible=False)
```

```
shear.legend(handles=[shear.lines[0], bend.lines[0]], labels=['Shear',__
 ⇔'Bending'])
plt.show()
```



Notice that the plot above has a dual y-axis.

The values at the points are (shortly after for the shear values),

- $$\begin{split} \bullet & \ V_O = 8.2 \ kN, \ M_O = 0 \ N \cdot m \\ \bullet & \ V_A = -0.8 \ kN, \ M_A = 2460 \ N \cdot m \\ \bullet & \ V_B = -5.8 \ kN, \ M_B = 1740 \ N \cdot m \\ \bullet & \ V_C = 0 \ kN, \ M_C = 0 \ N \cdot m \end{split}$$

### 3.1 Given

Plane stress values,

a. 
$$\sigma_x=20~kpsi,~\sigma_y=-10~kpsi,~{\rm and}~\tau_{xy}=8~kpsi~cw$$
b.  $\sigma_x=-12~kpsi,~\sigma_y=22~kpsi,~{\rm and}~\tau_{xy}=12~kpsi~cw$ 

### **3.2** Find

Draw a Mohr's circle diagram. Find the principal normal and shear stresses, and determine the angle from the x-axis to  $\sigma_1$ . Draw the stress elements as in Figure 3-11c and d.

### 3.3 Solution

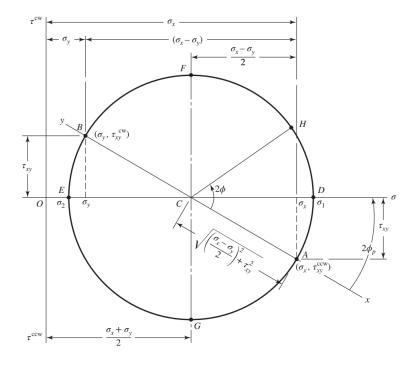
Mohr's circle is a circle that has the following characteristics:

- The x-axis is  $\sigma$  and the y-axis is  $\tau$ .
- The center of the circle is along the x-axis at the average of  $\sigma_x$  and  $\sigma_y$ .
- The radius of the circle is  $R = \sqrt{\left(\frac{\sigma_x \sigma_y}{2}\right)^2 + \tau_{xy}^2}$ .

The principle stresses and max angles may be calculated using the following,

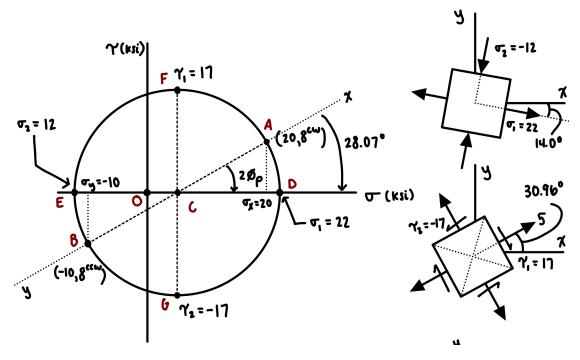
$$\begin{split} \sigma_1, \sigma_2 &= \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \\ \tau_1, \tau_2 &= \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \\ \tan 2\phi_p &= \frac{2\tau_{xy}}{\sigma_x - \sigma_y} \\ \tan 2\phi_s &= -\frac{\sigma_x - \sigma_y}{2\tau_{xy}} \end{split}$$

Here is a breakdown of Mohr's circle,



### 3.3.1 Part A

Using the information from the figure above,



Center and Radius:

$$C = \frac{\sigma_x}{2} + \frac{\sigma_y}{2} = 5$$

$$R = \sqrt{\tau_{xy}^2 + \left(\frac{\sigma_x}{2} - \frac{\sigma_y}{2}\right)^2} = 17.0$$

### **Principle Stresses:**

$$\sigma_1 = C + R = 22.0$$

$$\sigma_2 = C - R = \boxed{-12.0}$$

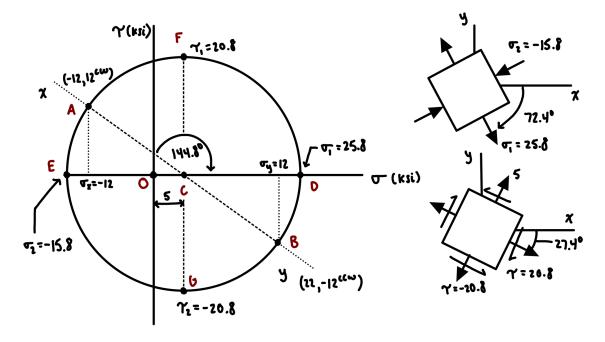
$$\tau_1 = R = 17.0$$

$$\tau_2 = -R = -17.0$$

### Angle of Occurrence:

$$2\phi_p = \operatorname{atan}\left(\frac{2\tau_{xy}}{\sigma_x - \sigma_y}\right) = 28.072486935853$$

### 3.3.2 Part D



### Center and Radius:

$$C = \frac{\sigma_x}{2} + \frac{\sigma_y}{2} = 5.0$$

$$R = \sqrt{\tau_{xy}^2 + \left(\frac{\sigma_x}{2} - \frac{\sigma_y}{2}\right)^2} = 20.8086520466848$$

### **Principle Stresses:**

$$\sigma_1 = C + R = 25.8086520466848$$

$$\tau_1 = R = 20.8086520466848$$

$$\tau_2 = -R = \boxed{-20.8086520466848}$$

## Angle of Occurrence:

$$2\phi_p = \tan\left(\frac{2\tau_{xy}}{\sigma_x - \sigma_y}\right) = \boxed{144.782407031807}$$

### 4.1 Given

Element stresses in ksi:

- $\sigma_x = -6$
- $\sigma_{u} = 18$
- $\sigma_z = -12$
- $\bullet \quad \tau_{xy}=9$
- $\tau_{yz} = 6$
- $\tau_{zx} = -15$

### 4.2 Find

Determine the principal stresses and draw a complete Mohr's three-circle diagram, labeling all points of interest. Report the maximum shear stress for this case.

#### 4.3 Solution

### 4.3.1 Principal Stresses and Max Shear Stress

The principal stresses may be found by finding the roots of the following expression,

$$\sigma^3 - \sigma^2 \left(\sigma_x + \sigma_y + \sigma_z\right) + \sigma \left(\sigma_x \sigma_y + \sigma_x \sigma_z + \sigma_y \sigma_z - \tau_{xy}^2 - \tau_{yz}^2 - \tau_{zx}^2\right) - \sigma_x \sigma_y \sigma_z + \sigma_x \tau_{yz}^2 + \sigma_y \tau_{zx}^2 + \sigma_z \tau_{xy}^2 - 2\tau_{xy} \tau_{yz} \tau_{zx} = 0$$

```
[7]: sig_x, sig_y, sig_z, tau_xy, tau_zx, tau_yz = sp.symbols(r'\sigma_x \sigma_yu\sigma_z \tau_{xy} \tau_{zx} \tau_{yz}')
sig = sp.Symbol(r'\sigma')
sig1, sig2, sig3 = sp.symbols(r'\sigma_1 \sigma_2 \sigma_3')

poly = sig**3 - (sig_x + sig_y + sig_z)*sig**2 + (sig_x*sig_y + sig_x*sig_z + sig_y*sig_z - tau_xy**2 - tau_yz**2 - tau_zx**2)*sig - (sig_x*sig_y*sig_z + sig_y*sig_z - tau_xy**2 - sig_x*tau_yz**2 - sig_y*tau_zx**2 - sig_x*tau_xy**2)

display(sp.Eq(poly.simplify(), 0))
poly_ = poly.subs([(sig_x, -6), (sig_y, 18), (sig_z, -12), (tau_xy, 9), sig_y))
sig(tau_yz, 6), (tau_zx, -15)])
roots = [sp.re(root.n()) for root in sp.roots(poly_, sig)]
roots_ = sorted(list(roots), reverse=True)
s1, s2, s3 = roots_
for i, j in zip((sig1, sig2, sig3), roots_):
    display(sp.Eq(i, j))
```

$$\sigma^3 - \sigma^2 \left(\sigma_x + \sigma_y + \sigma_z\right) + \sigma \left(\sigma_x \sigma_y + \sigma_x \sigma_z + \sigma_y \sigma_z - \tau_{xy}^2 - \tau_{yz}^2 - \tau_{zx}^2\right) - \sigma_x \sigma_y \sigma_z + \sigma_x \tau_{yz}^2 + \sigma_y \tau_{zx}^2 + \sigma_z \tau_{xy}^2 - 2\tau_{xy}\tau_{yz}\tau_{zx} = 0$$

 $\sigma_1 = 21.0369191756785$ 

 $\sigma_2 = 5.67061153118965$ 

 $\sigma_3 = -26.7075307068682$ 

The max shear stress is,

$$\tau_{max} = \frac{\sigma_1 - \sigma_3}{2}$$

- [8]: (s1 s3)/2
- [8]: 23.8722249412734

### 4.3.2 Von Mises Stress

The Von Mises Stress is,

[9]: 42.2137418384109

The Von Mises Stress may also be computed without using the principal stresses,

```
[10]: sig_x, sig_y, sig_z, tau_xy, tau_yz, tau_zx = -6, 18, -12, 9, 6, -15
(1/sp.sqrt(2)*sp.sqrt((sig_x - sig_y)**2 + (sig_y - sig_z)**2 + (sig_z -
sig_x)**2 + 6*(tau_xy**2 + tau_yz**2 + tau_zx**2))).n()
```

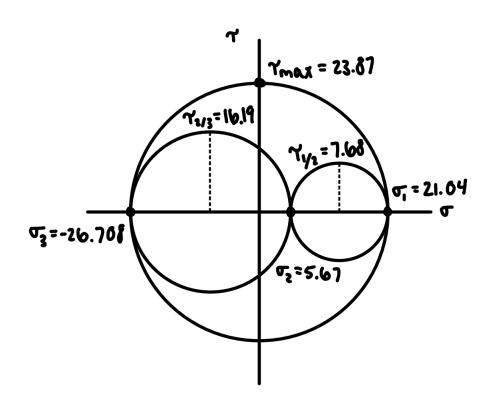
[10]: 42.2137418384109

### 4.3.3 Mohr's Circle Diagram

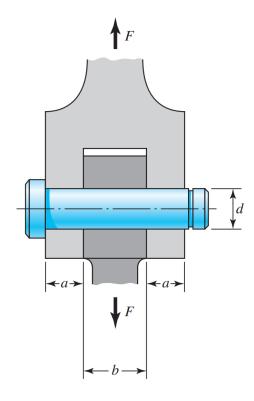
```
[11]: # Calculating additional shear stress values
tau_1_2, tau_2_3 = sp.symbols(r'\tau_{1/2} \tau_{2/3}')
display(sp.Eq(tau_1_2, (s1 - s2)/2))
display(sp.Eq(tau_2_3, (s2 - s3)/2))
```

 $\tau_{1/2} = 7.68315382224444$ 

 $\tau_{2/3} = 16.1890711190289$ 



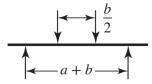
### 5.1 Given



 $F = 4 \ kN$ ,  $a = 6 \ mm$ ,  $b = 18 \ mm$ , and  $d = 12 \ mm$ 

### **5.2** Find

Find the maximum bending and shear stress if modeled like so,



### 5.3 Solution

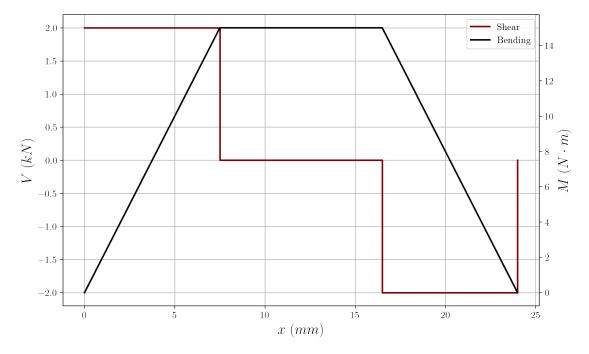
Start with a shear and moment diagram.

```
fig, ax1 = plt.subplots()
ax2 = ax1.twinx()

ax1.plot(x_shear, V, label='Shear')
ax1.set_xlabel('$x$ ($mm$)')
ax1.set_ylabel('$v$ ($kN$)')

ax2.plot(x, M, label='Bending', color='black')
ax2.set_ylabel(r'$M$ ($N\cdot m$)')
ax2.grid(visible=False)

ax1.legend(handles=[ax1.lines[0], ax2.lines[0]], labels=['Shear', 'Bending'])
plt.show()
```



The maximum bending moment is 15  $N \cdot m$ , and the maximum shear stress is  $\tau_{max} = 2000 \ N$ . The normal and shear stress due to bending is,

$$\sigma_{max} = \frac{Mc}{I}$$

$$I = \frac{1}{4}\pi r^4$$

and for shear, the maximum stress for circular cross-sections is,

$$\tau_{max} = \frac{4}{3} \frac{V}{A}$$

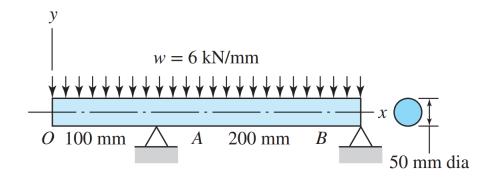
[13]: M, c = sp.S(15), sp.S(0.006) I = sp.S(0.25)\*sp.pi\*sp.S(0.006)\*\*4 sig\_max = M\*c/I sig\_max.n() # in Pa

# [13]: 88419412.8288307

[14]: V, A = 2000, sp.pi\*sp.S(0.006)\*\*2
tau\_max = 4/3\*V/A
tau\_max.n() # in Pa

# [14]: 23578510.0876882

### 6.1 Given

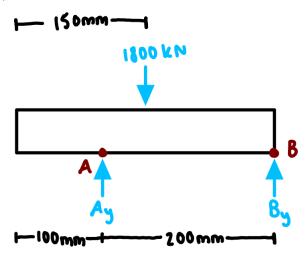


### **6.2** Find

For the beam above, find the maximum tensile stress due to M and the maximum shear stress due to V.

### 6.3 Solution

The free body diagram is,



```
[15]: # Getting reaction forces
Ay, By = sp.symbols('A_y B_y')
eq1 = sp.Eq(Ay + By, 1800)
eq2 = sp.Eq(200*Ay, 150*1800)

sol = sp.solve([eq1, eq2], dict=True)[0]
[display(eq) for eq in [eq1, eq2]]
```

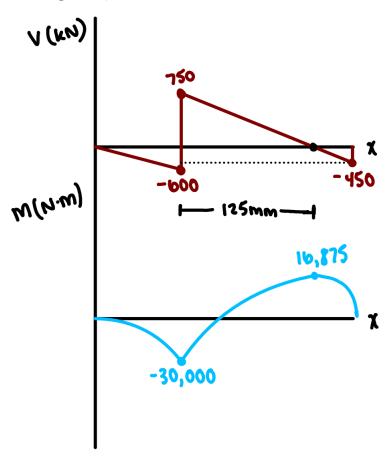
```
display(Markdown('---'))
for key, value in sol.items():
    display(sp.Eq(key, value))
```

$$A_y + B_y = 1800$$
  
 $200A_y = 270000$ 

 $A_y = 1350$ 

$$B_y = 450$$

The shear and moment diagram is,



The maximum shear and tensile stress occur at  $x = 100 \ mm$ .

```
[16]: # Calculating stress due to bending
M, c = 30_000, sp.S(0.025)
(M*c/(sp.pi/4*c**4)).n() # in Pa
```

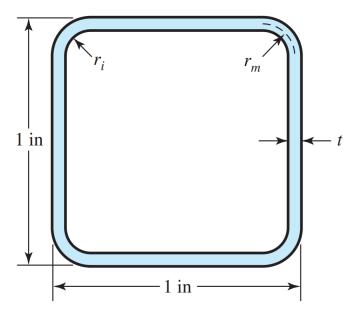
[16]: 2444619925.89151

```
[17]: # Calculating the maximum shear stress
V = 750_000
(sp.Rational(4, 3)*V/(sp.pi*c**2)).n() # in Pa
```

[17]: 509295817.894065

## 7 Problem 3-62 Part A

### 7.1 Given



The tube is 36 in long and  $r_i=r_m=0$ . The thickness t is  $\frac{1}{16}$ ".

### **7.2** Find

The maximum torque that can be applied and the corresponding angle of twist of the tube.

### 7.3 Solution

For thin-walled tubes,

$$\tau = \frac{T}{2A_m t}$$
 
$$\theta_1 = \frac{TL_m}{4GA_m^2 t}$$

See p. 129 for additional details of the above formulas.

```
[18]: # Calculating the maximum torque
    tau_max = 12_000  # lbf in
    t = sp.Rational(1, 16)  # thickness in inches
    Am = (1 - t)**2
    T = tau_max*2*Am*t
    T.n()  # lbf in
```

### [18]: 1318.359375

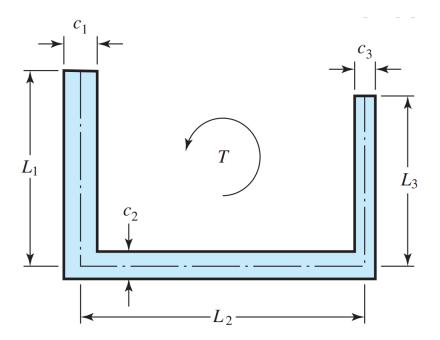
From table A-5, the modulus of rigidity is  $11.5 \ Mpsi$ .

```
[19]: G = 11.5e6
Lm = (1 - t)*4  # total length
L = 36  # inches
phi_1 = T*Lm/(4*G*Am**2*t)*L
(phi_1*180/sp.pi).n()  # in degrees
```

# [19]: 4.59163394776145

The expression gets multiplied by L because  $\theta_1$  is the angle of twist per unit length.

### 8.1 Given



 $c_1=2~mm,\,L_1=20~mm,\,c_2=3~mm,\,L_2=30~mm,\,c_3=0~mm,\,{\rm and}\,\,L_3=0~mm.$  The material is steel and the maximum shear is  $\tau_{allow}=12~ksi.$  The angle of twist is the same for each section.

### 8.2 Find

- a. Determine the torque transmitted by each leg and the torque transmitted by the entire section.
- b. Determine the angle of twist per unit length.

### 8.3 Solution

The relationship for open looped geometry is,

$$T_i = \frac{\theta_i G L_i c_i^3}{3}$$
 
$$\tau_{max} = G \theta_i c_{max}$$

From Table A-5,  $G_{steel}=79.3\ GPa.$  I will find Part B first because it is required to answer Part A.

### 8.3.1 Part B

```
[20]: tau_max = sp.S(82.7371e6)  # shear stress in Pa
G = sp.S(79.3e9)  # modulus of rigidity in Pa
c = [sp.S(c_) for c_ in (0.002, 0.003, 0)]  # in m
L = [sp.S(L_) for L_ in (0.02, 0.03, 0)]  # in m
c_max = max(c)

phi_i = tau_max/(G*c_max)
phi_i  # in rad per m
```

[20]: 0.347781000420345

### 8.3.2 Part A

```
[21]: T = []
    for i in range(len(c)):
        T_i = phi_i*G*L[i]*c[i]**3/3
        display(sp.Eq(sp.Symbol(f'T_{i + 1}'), T_i))
        T.append(T_i)
T = sum(T)
    display(sp.Eq(sp.Symbol('T'), T)) # torques in N m
```

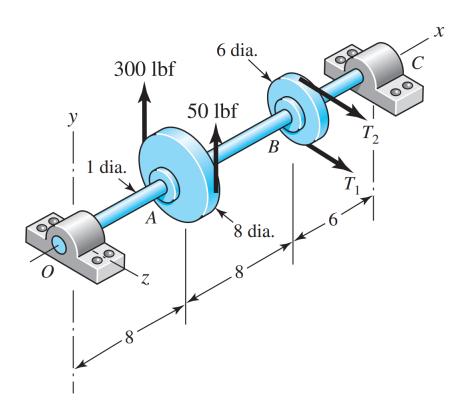
```
T_1 = 1.47088177777778
```

 $T_2 = 7.446339$ 

 $T_3 = 0$ 

T = 8.91722077777778

### 9.1 Given



A counter shaft carrying two V-belt pulleys is shown in the figure. Pulley A receives power from a motor through a belt with the belt tensions shown. The power is transmitted through the shaft and delivered to the belt on pulley B. Assume the belt tension on the loose side at B is 15 percent of the tension on the tight side.

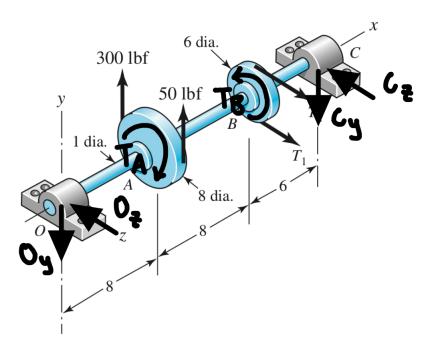
### 9.2 Find

- a. Determine the tensions in the belt on pulley B, assuming the shaft is running at a constant speed.
- b. Find the magnitudes of the bearing reaction forces, assuming the bearings act as simple supports.
- c. Draw shear-force and bending-moment diagrams for the shaft. If needed, make one set for the horizontal plane and another set for the vertical plane.
- d. At the point of maximum bending moment, determine the bending stress and the torsional shear stress.
- e. At the point of maximum bending moment, determine the principal stresses and the maximum shear stress.

### 9.3 Solution

### 9.3.1 Part A

The directions of the torques about A and B are,



Since the shaft has no angular acceleration,  $T_A = T_B$  (with directions shown above). It should also be noted that  $T_1$  must be greater than  $T_2$  because the torque shows that the pulley is more tensile at the bottom.

```
[22]: # Solving for T1 and T2

T1, T2 = sp.symbols('T_1 T_2')

T_A = 4*(sp.S(300) - 50)
eq1 = sp.Eq(3*(T1 - T2), T_A)
eq2 = sp.Eq(T2, sp.S(0.15)*T1)

[display(eq) for eq in [eq1, eq2, Markdown('----')]]

sol = sp.solve([eq1, eq2], dict=True)[0]
_ = [display(sp.Eq(key, value)) for key, value in sol.items()]

3T_1 - 3T_2 = 1000
T_2 = 0.15T_1
```

 $T_1 = 392.156862745098$ 

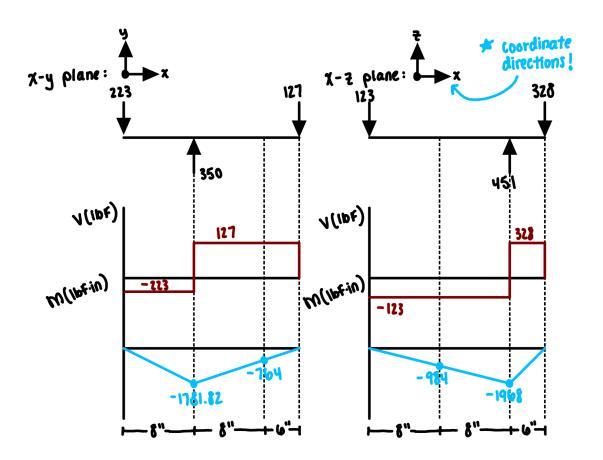
 $T_2 = 58.8235294117647$ 

### 9.3.2 Part B

```
[23]: # Solving for the reactions
      Oy, Oz, Cy, Cz = sp.symbols('O_y O_z C_y C_z')
      eq1 = sp.Eq(300 + 50 - 0y - Cy, 0) # Forces in y direction
      eq2 = sp.Eq(sol[T1] + sol[T2] - Oz - Cz, 0) # Forces in z direction
      eq3 = sp.Eq(8*sp.S(350) - Cy*22, 0) # Moments about z-axis
      eq4 = sp.Eq(-16*(sol[T1] + sol[T2]) + Cz*22, 0) # Moments about the y-axis
      sol2 = sp.solve([eq1, eq2, eq3, eq4], dict=True)[0]
      [display(eq) for eq in [eq1, eq2, eq3, eq4, Markdown('---')]]
      _ = [display(sp.Eq(key, value)) for key, value in sol2.items()]
     -C_u - O_u + 350 = 0
     -C_z - O_z + 450.980392156863 = 0
     2800 - 22C_u = 0
     22C_z - 7215.6862745098 = 0
     C_y = 127.272727272727
     O_y = 222.727272727273
     C_z = 327.985739750445
     O_z = 122.994652406418
```

### 9.3.3 Part C

The shear and moment diagram for the two planes is,



### 9.3.4 Part D

```
[24]: # Getting max bending moment
# At A,

M_A = sp.sqrt(1781.8181818181818**2 + 983.9572195**2)
M_B = sp.sqrt(763.6363636363**2 + 1967.914439**2)
sp.Matrix([M_A, M_B])
```

[24]: [2035.44782366535] 2110.88316471859]

The maximum bending moment occurs at point B with a value of  $2110.88316471859\ lbf\ in$ .

```
[25]: # Getting the bending stress
c = sp.S(0.5)
sig_x = (M_B*c/(sp.pi/4*c**4)).n()
sig_x # psi
```

[25]: 21501.2793570833

```
[26]: # Getting the torsional stress
t_xz = (1000*c/(sp.pi/2*c**4)).n()
t_xz # in psi
```

[26]: 5092.95817894065

### 9.3.5 Part E

### Center and Radius:

$$C = \frac{\sigma_x}{2} + \frac{\sigma_y}{2} = 10750.6396785417$$

$$R = \sqrt{\tau_{xy}^2 + \left(\frac{\sigma_x}{2} - \frac{\sigma_y}{2}\right)^2} = 11895.9857309209$$

### **Principle Stresses:**

$$\sigma_1 = C + R = 22646.6254094625$$

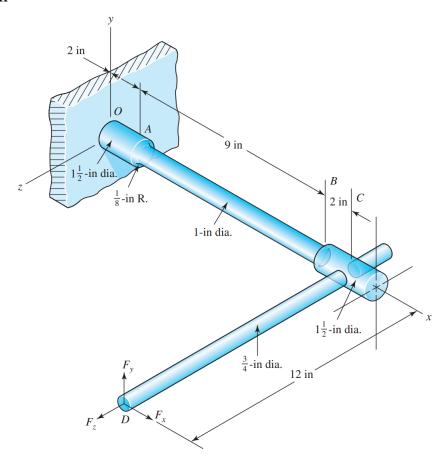
$$\tau_1 = R = 11895.9857309209$$

$$\tau_2 = -R = -11895.9857309209$$

### Angle of Occurrence:

$$2\phi_p = \tan\left(\frac{2\tau_{xy}}{\sigma_x - \sigma_y}\right) = 25.348569568567$$

### 10.1 Given



The cantilevered bar in the figure is made from a ductile material and is statically loaded with  $F_y = 250$  lbf and  $F_x = F_z = 0$ . Analyze the stress situation in the small diameter at the shoulder at A.

### 10.2 Find

- a. Determine the precise location of the critical stress element at the cross section at A.
- b. Sketch the critical stress element and determine the magnitudes and directions for all stresses acting on it.
- c. For the critical stress element, determine the principle stresses and the maximum shear stress.

### 10.3 Solution

### 10.3.1 Part A

The critical stress element will be at the top or bottom  $(y = \pm 0.5 in)$  because both the bending stress and shear stress are maximized at the farthest distance away from the neutral axis.

### 10.3.2 Part B

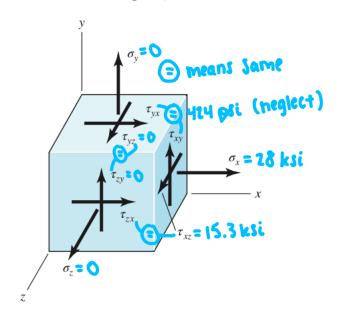
The series of calculations are,

[28]: 15278.874536822

[29]: 28011.2699841736

[30]: 424.413181578388

The transverse shear  $(\tau_{xy})$  may be neglected because it is an order of magnitude smaller than the other values. Here is the stress element diagram,



### 10.3.3 Part C

Center and Radius:

$$C = \frac{\sigma_x}{2} + \frac{\sigma_y}{2} = 14005.6349920868$$

$$R = \sqrt{\tau_{xy}^2 + \left(\frac{\sigma_x}{2} - \frac{\sigma_y}{2}\right)^2} = 20726.8381246034$$

## Principle Stresses:

$$\sigma_1 = C + R = \boxed{34732.4731166902}$$

$$\sigma_2 = C - R = \frac{-6721.20313251659}{}$$

$$\tau_1 = R = 20726.8381246034$$

$$\tau_2 = -R = -20726.8381246034$$

## Angle of Occurrence:

$$2\phi_p = \mathrm{atan}\left(\frac{2\tau_{xy}}{\sigma_x - \sigma_y}\right) = 47.4895529219991$$

### 11.1 Given

Hole and Shaft Dimensions:

• Fit Designation: (2 in)H7/s6

• Basic Size: 2 in

• Hole  $D_{max}$ : 2.0010

• Hole  $D_{min}$ : 2.0000

• Shaft  $d_{max}$ : 2.0023

• Shaft  $d_{min}$ : 2.0017

Use a collar diameter of 4 inches. The material is the same hot-rolled steel.

### 11.2 Find

Find the maximum and minimum values of the radial interference and the corresponding interface pressure.

### 11.3 Solution

The maximum interference occurs with the smallest hole size and the largest shaft size. The opposite is true for the smallest amount of interference. The elastic modulus is 30 Mpsi according to Table A-5.

```
[32]: D_max, D_min = 2.001, 2
d_max, d_min = 2.0023, 2.0017
delta_max = (d_max - D_min)/sp.S(2)
delta_max.n(6) # in inches
```

[32]: 0.00115

```
[33]: delta_min = (d_min - D_max)/sp.S(2) delta_min.n(6)
```

[33]: 0.00035

The outer radius is given  $(r_o = 2 in)$ . The inner radius is  $r_i = 0$ , since the shaft is not hollow.

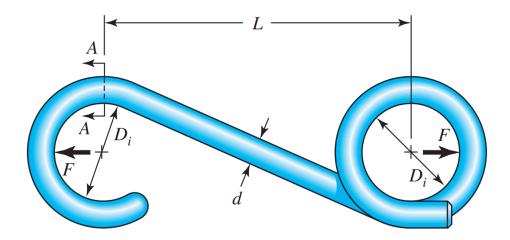
```
[34]: ro, ri = 2, 0
R = 1
E = 30e6

# Eq 3-57
A = E/(2*R**3)*((ro**2 - R**2)*(R**2 - ri**2)/(ro**2 - ri**2))
p_max = A*delta_max
p_max.n(6) # in psi
```

```
[34]:
12937.5
```

[35]:

### 12.1 Given



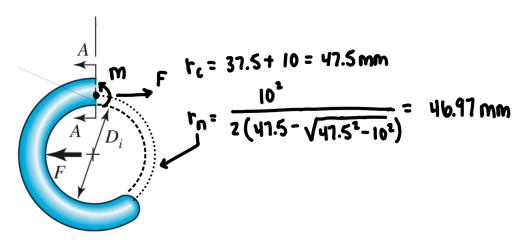
The utility hook above was formed from a round rod of diameter d = 20 mm.

### 12.2 Find

What are the stresses at the inner and outer surfaces at section A-A if F=4~kN,~L=250~mm, and  $D_i=75~mm?$ 

### 12.3 Solution

The internal stress at section A-A is,



The stress at the top and bottom may be found using Eq. 3-64.

```
M = rc*4000 # in N m
ci = (rn - ri)
co = (ro - rn)
e = (rc - rn)
A = sp.pi*(10/1000)**2

sig_i = M*ci/(A*e*ri) + 4000/A
sig_i.n(6)/10**6 # in MPa
```

### [36]: 299.597

[37]: # Getting sig outer

# We subtract the moment portion of the expression because it is in compression

sig\_o = -M\*co/(A\*e\*ro) + 4000/A

sig\_o.n(6)/10\*\*6 # in MPa

[37]: -195.389