Homework7

March 22, 2023

1 Weible Homework 7

Imports and preferences

```
[89]: # import necessary modules
      # numpy provides powerful multi-dimensional arrays to hold and manipulate data
      import numpy as np
      from numpy.linalg import norm
      # matplotlib provides powerful functions for plotting figures
      import matplotlib.pyplot as plt
      import matplotlib as mpl
      %matplotlib inline
      plt.rcParams['figure.dpi'] = 1200
      plt.rcParams['savefig.dpi'] = 1200
      # Computer Modern with TeX
      mpl.rc('font', **{'family': 'serif', 'serif': ['Computer Modern']})
      mpl.rc('text', usetex=True)
      # astropy provides unit system and constants for astronomical calculations
      import astropy.units as u
      import astropy.constants as const
      # import Latex module so we can display the results with symbols
      from IPython.display import Latex
      # **** import CenterOfMass to determine the COM pos/vel of M33
      from CenterOfMass import CenterOfMass
      # **** import the GalaxyMass to determine the mass of M31 for each component
      from GalaxyMass import ComponentMass
```

2 M33AnalyticOrbit

```
[92]: class M33AnalyticOrbit:
          """ Calculate the analytical orbit of M33 around M31 """
          def __init__(self, filename):
               Initializes our AnalyticOrbit object, storing the filename to write to,
               the M31 and M33 COM positions and velocities (incl. relative positions \Box
       \hookrightarrow and
              velocities), along with M31 halo, disk, and bulge masses and scale_{\sqcup}
       \hookrightarrow lengths.
              Inputs:
                   filename : 'str'
                       string filename to write our integrated orbit to.
              Returns:
                   none
              ### get the gravitational constant (the value is 4.498502151575286e-06)
              self.G = const.G.to(u.kpc**3/u.Msun/u.Gyr**2).value
              ### store the output file name
              self.filename = filename
              ### get the current pos/vel of M33
              # create an instance of the CenterOfMass class for M33
              # 2 = disk particles
              M33_com = CenterOfMass('M33_000.txt', 2)
              # store the position VECTOR of the M33 COM (.value to get rid of units)
              # tolerance of 0.1 kpc, reduce radius by factor of 4 for stripped M33
              M33_{com_p} = M33_{com.COM_P(0.1, 4)} # kpc
              self.M33_com_p = M33_com_p.value # kpc
              # store the velocity VECTOR of the M33 COM (.value to get rid of units)
              # need units on position when calling COM_V, so use M33_com_p
              M33_{com_v} = M33_{com.COM_v(*M33_{com_p})} # km/s
              self.M33\_com\_v = M33\_com\_v.value # km/s
              ### get the current pos/vel of M31
              # create an instance of the CenterOfMass class for M31
              M31_com = CenterOfMass('M31_000.txt', 2)
              # store the position VECTOR of the M31 COM (.value to get rid of units)
```

```
# tolerance of 0.1 kpc, reduce radius by factor of 2 for less-stripped
⊶M31
       M31_{com_p} = M31_{com.COM_P(0.1, 2)} # kpc
       self.M31_com_p = M31_com_p.value # kpc
       # **** store the velocity VECTOR of the M31 COM (.value to get rid of ...
\rightarrow units)
       # need units on position when calling COM_V, so use M31_com_p
       M31\_com\_v = M31\_com.COM\_V(*M31\_com\_p) # km/s
       self.M31_com_v = M31_com_v.value# km/s
       ### store the DIFFERENCE between the vectors posM33 - posM31
       # create two VECTORs self.r0 and self.v0 and have them be the
       # relative position and velocity VECTORS of M33
       self.r0 = self.M33_com_p - self.M31_com_p
       self.v0 = self.M33_com_v - self.M31_com_v
       ### get the mass of each component in M31
       ### disk
       # self.rdisk = scale length (no units)
       self.rdisk = 5 # kpc
       \# self.Mdisk set with ComponentMass function. *1e12 to get the right_{\sqcup}
\rightarrow units.
       # ptype=2 for disk particles
       self.Mdisk = ComponentMass('M31_000.txt', 2).value * 1e12
       ### bulge
       # self.rbulge = set scale length (no units)
       self.rbulge = 1 # kpc
       \# self.Moulge set with ComponentMass function. *1e12 to get the right
\neg units.
       # ptype=3 for bulge particles
       self.Mbulge = ComponentMass('M31 000.txt', 3).value * 1e12
       # Halo
       # self.rhalo = set scale length from HW5 (no units)
       self.rhalo = 62 # kpc
       # self.Mhalo set with ComponentMass function. *1e12 to get the right_{\sqcup}
\rightarrow units.
       # ptype=1 for halo
       self.Mhalo = ComponentMass('M31_000.txt', 1).value * 1e12
  def HernquistAccel(self, M, r_a, r):
```

```
Computes the acceleration vector from a Hernquist potential,
       which we'll use for our bulges and halos. Takes a Hernquist mass, scale _{\sqcup}
\hookrightarrow length,
       and the position w.r.t. the potential for the acceleration to be \Box
\hookrightarrow calcualted at.
       Inputs:
           M : 'float'
                Halo/bulge mass in MSun for the Hernquist potential
           r_a : 'float'
                Hernquist profile scale length in kpc (from mass profile fit)
           r : 'numpy array'
                Vector of floats, relative position to calculate the ⊔
\hookrightarrow acceleration at
                (kpc)
       Returns:
           acc : 'numpy array'
                Vector of floats, acceleration vector due to the Hernquist_{\sqcup}
\hookrightarrow potential
                (kpc/Gyr^2 ~ km/s/Gyr)
       ### Store the magnitude of the position vector
       rmag = norm(r)
       ### Store the Acceleration vector
       Hern = -self.G * M * r / (rmag * (r_a + rmag)**2)
       return Hern
  def MiyamotoNagaiAccel(self, M, r_d, r):
       Computes the acceleration vector from a Miyamoto-Nagai profile,
       while we'll use for the disk. Takens a disk mass, scale radius, and \Box
\hookrightarrow position
       to caclulate the acceleration at.
       Inputs:
            M : 'float'
                Disk mass in MSun
           r_d : 'float'
                Miyamoto-Nagai disk scale length in kpc
           r : 'numpy array'
```

```
Vector of floats, relative position to calculate the ⊔
\hookrightarrow acceleration at
               (kpc)
        Returns:
           acc : 'numpy array'
               Vector of floats, acceleration vector due to the Miyamoto-Nagai
               potential (kpc/Gyr^2 ~ km/s/Gyr)
       11 11 11
       # Decompose the position vector
      x = r[0]
      y = r[1]
      z = r[2]
       # Miyamoto-Nagai 1975 profile parameters
      z_d = r_d / 5.0
      R = np.sqrt(x**2 + y**2)
      B = r_d + np.sqrt(z**2 + z_d**2)
       # prefactor
      prefac = -self.G * M / (R**2 + B**2)**(1.5)
       # put it all together
      acc = prefac * r * np.array([1, 1, B / np.sqrt(z**2 + z_d**2)])
      return acc
  def M31Accel(self, r):
       Computes the total acceleration vector due to all three components of \Box
→M31
       combined, at position vector r. Takes only the relative position w.r.t.
⇔M31.
       Inputs:
           r : 'numpy array'
               Vector of floats, relative position to calculate the ⊔
\hookrightarrow acceleration at
               (kpc)
       Returns:
           acc_sum : 'numpy array'
               Vector of floats, acceleration vector due to all three∟
⇔components of
               M31, at position r (kpc/Gyr^2 ~ km/s/Gyr)
```

```
11 11 11
       # Call the previous functions for the halo, bulge and disk
       acc_disk = self.MiyamotoNagaiAccel(self.Mdisk, self.rdisk, r)
       acc_halo = self.HernquistAccel(self.Mhalo, self.rhalo, r)
       acc_bulge = self.HernquistAccel(self.Mbulge, self.rbulge, r)
       # return the SUM of the output of the acceleration functions
       # this will return a VECTOR
       acc_sum = acc_disk + acc_halo + acc_bulge
      return acc sum
  def LeapFrog(self, dt, r, v):
       Integrates our galaxy COM equation of motion (d^2r/dt^2 = a) using a
       leapfrog integrator (second order, symplectic, etc. etc.). Takes a time\sqcup
\hookrightarrowstep,
       relative position, and relative velocity, and returning the next relative
      posion and velocity.
       Inputs:
           dt : 'float'
              Integration time step (Gyr)
           r : 'numpy array'
               Vector of floats, relative position of M33 to M31 (kpc)
           v : 'numpy array'
               Vector of floats, relative velocity of M33 to M31 (km/s ~ kpc/
\hookrightarrow Gyr)
       Returns:
           rnew: 'numpy array'
               Vector of floats, updated relative position of M33 (kpc)
           vnew : 'numpy array'
               Vector of floats, update velocity of M33 (km/s ~ kpc/Gyr)
       11 11 11
       # predict the position at the next half timestep
      rhalf = r + v * (dt/2)
       # predict the final velocity at the next timestep using the
→acceleration field at the rhalf position
       vnew = v + self.M31Accel(rhalf) * dt
```

```
# predict the final position using the average of the current velocity \Box
\hookrightarrow and the
       # final velocity. This accounts for the fact that we don't know how the
\hookrightarrowspeed
       # changes from the current timestep to the next, so we approximate it_{\sqcup}
\hookrightarrow using
       # the average expected speed over the time interval dt.
       rnew = rhalf + vnew * (dt/2)
       # return the new position and velocity vectors
       return rnew, vnew
  def OrbitIntegration(self, t0, dt, tmax):
       Loops over the LeapFroq method in order to integrate the COM equation \Box
\hookrightarrow of motion
       for M33 around M31 from time tO until time tmax with time step dt.
       Inputs:
           t0 : 'float'
                Start time for orbit integration (Gyr)
           dt : 'float'
                Time step (Gyr)
           tmax : 'float'
                End time for orbit integration (inclusive) (Gyr)
       HHHH
       # initialize the time to the input starting time
       t = t0
       # initial position and velocity
       r = self.r0
       v = self.v0
       # initialize an empty array of size : rows int(tmax/dt)+2 , columns 7
       orbit = np.zeros((int(tmax/dt)+3, 7))
       # initialize the first row of the orbit
       orbit[0] = t0, *tuple(self.r0), *tuple(self.v0)
       # this above is equivalent to
       \# \ orbit[0] = t0, \ self.r0[0], \ self.r0[1], \ self.r0[2], \ self.v0[0], \ self.
90[1], self.00[2]
       # initialize a counter for the orbit.
       i = 1 # since we already set the Oth values, we start the counter at 1
```

```
# start the integration (advancing in time steps and computing LeapFroq.
→at each step)
       while t < tmax: # as long as t has not exceeded the maximal time
           # advance the time by one timestep, dt
           t += dt
           # store the new time in the first column of the ith row
           orbit[i, 0] = t
           # advance the position and velocity using the LeapFrog scheme
           rnew, vnew = self.LeapFrog(dt, r, v)
           # store the new position vector into the columns with indexes 1,2,3_{\sqcup}
⇔of the ith row of orbit
           orbit[i, 1:4] = rnew
           # store the new velocity vector into the columns with indexes 4,5,6_{\sqcup}
⇔of the ith row of orbit
           orbit[i, 4:] = vnew
           \# update counter i , where i is keeping track of the number of rows
           # (i.e. the number of time steps)
           i += 1
           # Update position and velocity for the next run of the loop.
           r = rnew
           v = vnew
       # write the data to a file
       np.savetxt(self.filename, orbit, fmt = "%13.4f"*7, comments='#',
                  header="{:>10s}{:>11s}{:>11s}{:>11s}{:>11s}{:>11s}{:>11s}{:>11s}{:>11s}{:>11s}
                  .format('t', 'x', 'y', 'z', 'vx', 'vy', 'vz'))
```

3 5 Analysis

3.1 1.

Initialization and integration:

```
[112]: # Integration interval w/time step
t0 = 0.0 # Gyr
tmax = 10.0 # Gyr
dt = 0.00001 # Gyr
# construct a filename with the time interval
```

```
filename = 'dt_'+str(dt)+'.txt'

# Integrate the orbit

M33_orbit = M33AnalyticOrbit(filename)
M33_orbit.OrbitIntegration(t0, dt, tmax)
```

Read data and compute:

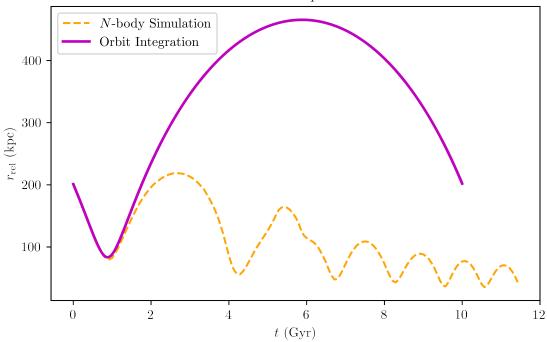
```
[]: # Read in orbit integration data
     data = np.genfromtxt(filename, dtype=None, names=True)
     # assign to vars and create vectors
     t = data['t'] # orbit integration time (Gyr)
     # relative position components (kpc)
     x, y, z = data['x'], data['y'], data['z']
     r = np.array([x, y, z]) # to a vector (kpc)
     rmag = norm(r, axis=0) # magnitude (kpc)
     # velocity/speed (km/s ~ kpc/Gyr):
     vx, vy, vz = data['vx'], data['vy'], data['vz'] # components
     v = np.array([vx, vy, vz]) # to a vector
     vmag = norm(v, axis=0) # magnitude
     # Read in the N-body file for M31 from Homework 6
     M31 = np.genfromtxt('Orbit_M31.txt', dtype=None, names=True)
     # assign to vars and create vectors:
     t_N = M31['t'] \# N-body time (Gyr)
     # kpc
     x_M31, y_M31, z_M31 = M31['x'], M31['y'], M31['z'] # position components
     r_M31 = np.array([x_M31, y_M31, z_M31]) # to a vector
     # M31 velocity components (km/s ~ kpc/Gyr)
     vx_M31, vy_M31, vz_M31 = M31['vx'], M31['vy'], M31['vz']
     v_M31 = np.array([vx_M31, vy_M31, vz_M31]) # to a vector
     # Read in the N-body file for M33 from Homework 6
     M33 = np.genfromtxt('Orbit_M33.txt', dtype=None, names=True)
     # assign to vars and create vectors (kpc)
     x_M33, y_M33, z_M33 = M33['x'], M33['y'], M33['z'] # position components
     r_M33 = np.array([x_M33, y_M33, z_M33]) # to a vector
     # M31 velocity components (km/s ~ kpc/Gyr)
     vx_M33, vy_M33, vz_M33 = M33['vx'], M33['vy'], M33['vz']
     v_M33 = np.array([vx_M33, vy_M33, vz_M33]) # to a vector
```

```
# magnitude of differences (relative displacement and speed)
N_rel_rmag = norm(r_M33 - r_M31, axis=0) # kpc
N_rel_vmag = norm(v_M33 - v_M31, axis=0) # km/s ~ kpc/Gyr
```

3.1.1 Plot:

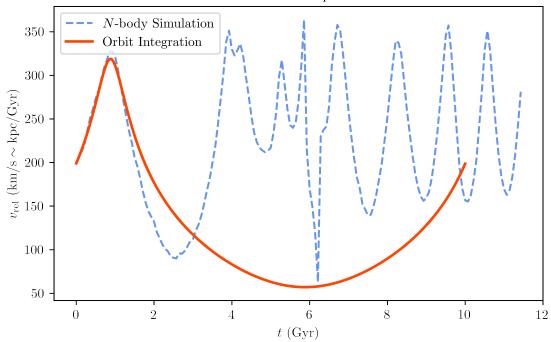
Relative separation

M31-M33 Relative Separation v. Time



Relative Speed

M31-M33 Relative Speed v. Time



3.2 2.

The two methods of orbit simulation agree well for the first ~ 1.5 Gyr or so, before diverging. The semi-analytic orbit integration predicts a single pericenter-apocenter pair in the next 10 Gyr for the M31-M33 system, while the N-body simulation gives ~ 5 such pairs in the next 10 Gyr. The semi-analytic integration gives a much larger apocenter at ~ 6 Gyr than any of those predicted by the N-body simulation. The semi-analytic integration shows M33 being flung out further from M31 into a longer-period orbit.

3.3 3.

We do not have any dynamical friction present in the semi-analytic orbit integration, as we do not actually model dark matter particles and their motions, just a constant Hernquist profile for M31. This would help to explain the damping present in the N-body simulation's relative position. I think that this is the main bit of physics that would serve to bring M33 closer and onto a shorter- and shortenting-period orbit were it accounted for in the semi-analytic approach. Somewhere energy is being removed from M33, and it makes sense for this to be from dynamical friction.

3.4 4.

We could model the MW similarly to M31, with idealized bulge, disk, and halo potentials, along with its initial conditions in the M31 COM frame from the N-body simulation. This is then a (restricted) 3-body problem and we should account for the acceleration of M33 due to the MW, as well as the acceleration of the MW due to M31 (i.e., stay in the M31 COM frame, and don't model

the force on the MW from M33, though we could if we wanted to). I imagine that there would be issues here during the merger though, where we may get infinities from the COM positions of the MW and M31 getting very close. Maybe a softening length could be added, and/or once the two big galaxies come close enough (but not too close), the remnant could then be modeled as a single more-or-less elliptical galaxy with associated idealized potential(s). I'm not really sure I like this idea though, because the more we rely on the results of a single N-body simulation derived from a single choice of initial conditions compatible with observations the less informative the semi-analytic integration is for exploring the range of compatible initial conditions. On the other hand, we might still be able to learn something about the subset of compatible initial conditions that do lead to a merger by doing this.

[]: