File - C:\Users\geoff\Desktop\FlyOx Concept\Python\API\airfoilAPI.py

```
1 import math
2 import numpy as np
3 # import sys
4 # import os
6 class CLalfa:
8
      def __init__ (self, clalfa,rangeAR,Mach,sweepHalfChord):
          self.clalfa = clalfa
10
          self.rangeAR = rangeAR
11
          self.Mach = Mach
12
          self.sweepHalfChord = sweepHalfChord
13
14
      def ellipticalCLalfa(self):
15
          ellipticalCLalfa = self.clalfa/(1+(self.clalfa/(math.pi*self.rangeAR)))
16
          return ellipticalCLalfa
17
18
      def hemboldCLalfa(self):
19
          hemboldCLalfa = (2*math.pi*self.rangeAR)/(2+math.sqrt(self.rangeAR**2 + 4 ))
20
          return hemboldCLalfa
21
22
      def polhamusCLalfa (self):
          beta = (1-self.Mach**2)**(0.5)
23
24
          k = self.clalfa/(2*math.pi)
25
          #remember to cremove 57.3 later
26
         polhamusCLalfa = (2*math.pi*self.rangeAR)/( 2 + math.sqrt(((self.rangeAR*beta )/k)**2 *(1+((
  math.tan(self.sweepHalfChord/57.3))**2/beta **2))+4))
27
         return polhamusCLalfa
28
29
30 class classCLmax:
31
32
      def init
                 (self, clmaxRoot, yMGC, wingSpan, clmaxTip, sweepQuarterChord):
33
          self.clmaxRoot = clmaxRoot
34
          self.yMGC = yMGC
35
          self.wingSpan = wingSpan
36
          self.clmaxTip = clmaxTip
37
          self.sweepQuarterChord = sweepQuarterChord
38
39
      def rapidClmax(self):
40
          clmax = self.clmaxRoot + (self.yMGC/self.wingSpan)*(self.clmaxTip-self.clmaxRoot)
41
          NACA graphs
42
          CLmaxo = 0.9*clmax
43
          kA = math.cos(self.sweepQuarterChord/57.3)
         rapidClmax = CLmaxo*kA
44
45
         return rapidClmax
46
47
```