

# **Van's Aircraft RV-7**

## **Pilot's Operating Handbook**

***Registration: N732DH***

***S/N: 74261***

***Builder: Dave Hock***

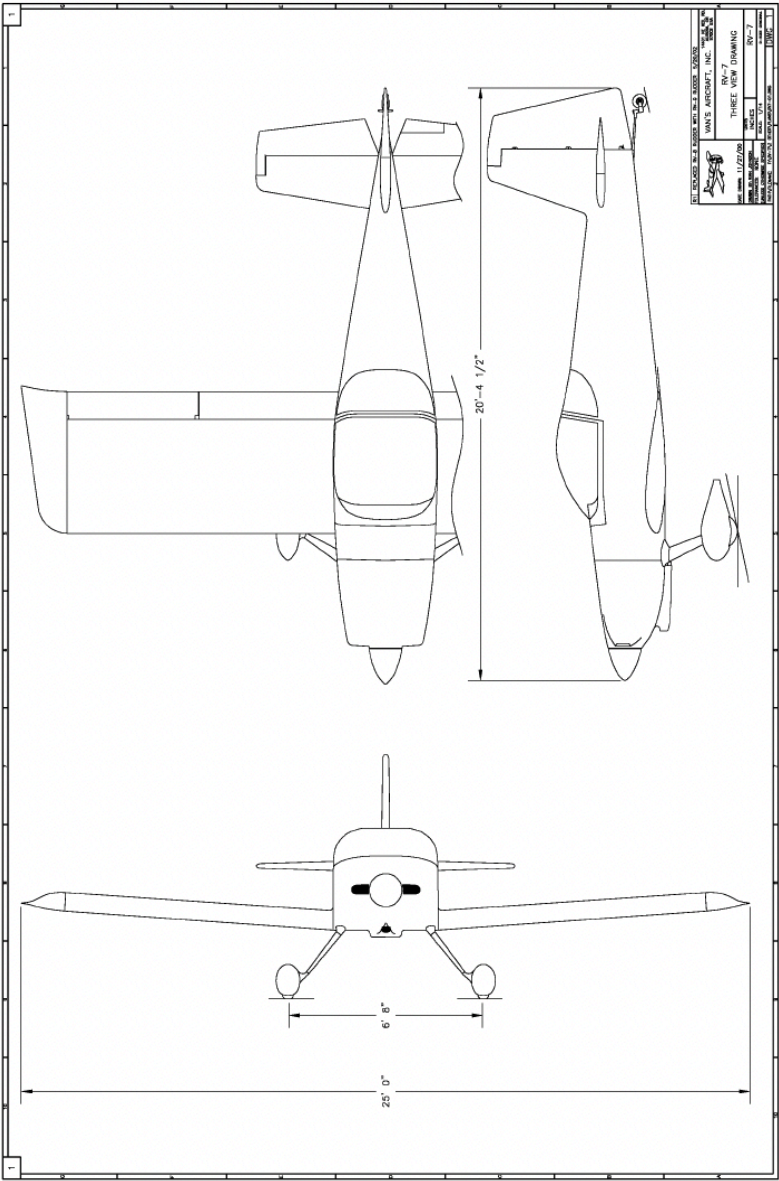


Table of Contents:

	Section
General.....	1
Limitations.....	2
Emergency Procedures.....	3
Normal Procedures.....	4
Performance.....	5
Aerobatic Information.....	6
Weight and Balance.....	7
Equipment List.....	8
Engine Operation.....	9
Miscellaneous .....	10

# Section 1 - General:

## 3 VIEW:



# Section 1 - General:

## Engine Information:

Model: ..... Lycoming YIO-360-EXP139 Fuel Injected  
HP: ..... 180 @ 2700  
Fuel: ..... UL 91/96 or 100LL 100/130 octane minimum  
Ignition: ..... Dual P-Mag P114  
Spark Plugs:.... NGK BR8ES 2.5mm center electrode,  
Stock #3961 solid post  
Oil Filter: ..... Champion CH48110-2

## Lycoming Oil Recommendations:

Avg Ambient	MIL-L-6082 or	MIL-L-22851 or
Air Temperature	SAEJ1966 SPEC	SAEJ1899 SPEC
	Mineral Grades	Ashless Dispersant

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All Temperatures	-----	SAE15W50 or 20W50
Above 80F	SAE 60	SAE 60
Above 60F	SAE 50	SAE 40 or 50
30 – 90F	SAE 40	SAE 40
0-70F	SAE 30	SAE 30,40 or 20W40
Below 10F		SAE 20 SAE 30 or 20W30

Oil Sump Capacity:	8 U.S. Quarts
Minimum Safe Quantity:	4 U.S. Quarts

## Operating Conditions:

Oil Inlet Temp:	245 F max	180 F desired	
Oil Pressure:	95 psi max	55 psi min	25 psi idle
Fuel Pressure:	45 psi max	14 psi min	30 psi desired
Cyl. Head Temp	435 F max	350-400 F desired	

Max oil consumption:  $0.006 \times \text{BHP} \times 4/7.4 = \text{Qt./Hr.}$   
(Lycoming Formula)  
Result = .58 qts/hr. at cruise

## Section 2 - Limitations:

### AIRSPEED LIMITATIONS

	<b>SPEED</b>	<b>IAS</b>	<b>REMARKS</b>
<b>Vne</b>	Never Exceed Speed	<b>200 KTS</b>	Do not exceed this speed in any operations. <b>*KTAS or KIAS</b>
<b>Vno</b>	Maximum Structural Cruising Speed	<b>168 KTS</b>	Exceed this speed only in smooth air.
<b>Va</b>	Maneuvering Speed	<b>123 KTS</b>	Do not make full control movements above this speed.
<b>Vfe</b>	Maximum Flap Extend Speed	<b>86 KTS</b>	Full Flaps -- Do not exceed this speed with flaps down
<b>Vy</b>	Best Rate of Climb	<b>90 KTS</b>	
<b>Vx</b>	Best Angle of Climb	<b>70 KTS</b>	
<b>Vs</b>	Stall Speed Clean	<b>55 KTS</b>	
<b>Vso</b>	Stall Speed Landing Configuration	<b>50 KTS</b>	Full Flaps
<b>Vr</b>	Rotation Speed	<b>65 KTS</b>	Initial liftoff speed
<b>Vref</b>	Short Final Landing Speed	<b>69 KTS</b>	Threshold crossing airspeed
<b>Vcc</b>	Cruise Climb	<b>115-120 KTS</b>	Best climb speed for cooling and over the nose visibility

### AIRSPEED INDICATOR MARKINGS

<b>MARKING</b>	<b>IAS VALUE OR RANGE</b>	<b>SIGNIFICANCE</b>
<b>White Arc</b>	<b>50-86 KTS</b>	Full Flap Operating Range. Lower limit is Vso. Upper limit is maximum speed with flaps fully extended
<b>Green Arc</b>	<b>56-168 KTS</b>	Normal Operating Range. Lower limit is Vs. Upper limit is maximum structural cruising speed
<b>Yellow Arc</b>	<b>168-200 KTS</b>	Operations must be conducted with caution and only in smooth air.
<b>Red Line</b>	<b>200 KTS</b>	Maximum speed for all operations <b>*KTAS or KIAS</b>

## Section 2 - Limitations:

### Vne TRUE AIRSPEED ALTITUDE CONSIDERATIONS

<b>RV7</b> <b>Vne (Redline) = 230 MPH or TAS = 200 KTAS</b>			
<b>Altitude (FT)</b>	<b>TAS (MPH/KTS)</b>	<b>IAS (MPH)</b>	<b>IAS (KTS)</b>
<b>Sea Level</b>	<b>230 / 200</b>	<b>230</b>	<b>200</b>
<b>2,500</b>	<b>230 / 200</b>	<b>222</b>	<b>193</b>
<b>5,000</b>	<b>230 / 200</b>	<b>214</b>	<b>186</b>
<b>7,500</b>	<b>230 / 200</b>	<b>206</b>	<b>179</b>
<b>10,000</b>	<b>230 / 200</b>	<b>198</b>	<b>172</b>
<b>12,500</b>	<b>230 / 200</b>	<b>191</b>	<b>166</b>
<b>15,000</b>	<b>230 / 200</b>	<b>183</b>	<b>159</b>
<b>17,500</b>	<b>230 / 200</b>	<b>176</b>	<b>153</b>
<b>20,000</b>	<b>230 / 200</b>	<b>169</b>	<b>147</b>

\* Assumes standard temperature and pressure

## Section 2 - Limitations:

### ENGINE MARKINGS

INSTRUMENT and RANGE	WHITE ARC	YELLOW ARC	GREEN ARC	YELLOW ARC	RED LINE
	NOTES:	Minimum Limit	Normal Operating	Caution Range	Maximum
Tachometer (RPM) 0-2800	< 900 Min speed for P-Mag self powering	-	900 - 2700	-	2700
Manifold (IN HG) 0-32	29.92 - 32.0	-	10.0 - 29.92	-	-
Oil Temp (F) 100 - 260	-	140 - 160	160 - 220	220 - 245	245>
Oil Pressure (PSI) 10 - 130	-	25 - 45	45 - 95	95 - 115	115
Fuel Pressure (PSI) 4 - 55	-	<14	14 - 45	45>	-
CHT (F) 100 - 450	-	250 - 320	320 - 420	420 - 435* *Max cruise	435 >
EGT (F) 1000 - 1800	-	-	1000 - 1400	1400 - 1650	1650* - 1800 *Red line

- Never exceed the maximum red line cylinder head temperature limit of 500°F (260°C).
- For maximum service life, cylinder head temperatures should be maintained below 435°F (224°C) during high performance cruise operation and below 400°F (205°C) for economy cruise powers.
- Per Lycoming, "oversquare" operations are permitted, see Section 9 for performance data.

### FUEL MARKINGS

INSTRUMENT and RANGE	NOTES:	RED ARC	YELLOW ARC	GREEN ARC
		Minimum	Caution	Normal
LEFT FUEL: 0-21 GAL MAX	Unusable: 1 GAL	0-1 GAL	1-6 GAL	6-21 GAL
RIGHT FUEL: 0-21 GAL MAX	Unusable: 1 GAL	0-1 GAL	1-6 GAL	6-21 GAL

- NOTE: Flight plan for no more than **30 gal** to ensure 10 usable for reserve.

### BATTERY MARKINGS

INSTRUMENT and RANGE	NOTES:	Transition Voltage	RED ARC	YELLOW ARC	GREEN ARC	YELLOW ARC	RED ARC
MAIN BATTERY 10.5 - 16.0	*EFIS Low Voltage	10.5-11.5	10.5-11.5	11.5-13.6 13.0*	13.6-14.5	14.5-15.5	15.5-16.0
IBBS BKUP 10.5 - 16.0			< 10.5	10.5-11.5	11.5-14.7	14.7-15.5	15.5-16.0

- IBBS SUPPORT: "...You will see a peak charging voltage of 14.7 nominally and when the system is running under load approx. 12.8 volts and the system idle and not under load and not charging about 13.6 volts..."

## Section 3 - Emergency Procedures:

### AIRSPEEDS FOR EMERGENCY OPERATIONS

Engine Failure After Takeoff:

Wing Flaps Up ..... 78 KTS

Wing Flaps Down ..... 70 KTS

Maneuvering Speed (Va) ..... 123 KTS

Maximum Glide ..... 75 KTS

### ELECTRICAL / ALTERNATOR FAILURE

1. Avionics – OFF
2. Master Switch – OFF
3. Alt Field -- OFF
4. Master Switch – ON

IF ALTERNATOR IS STILL OFF-LINE:

1. Master Switch – ON
2. Electrical Switches – OFF
3. Alternator Field – OFF
4. Avionics – ON as required
5. Electrical Equipment – ON, as required
6. Flight – TERMINATE as soon as practical, aircraft is on battery reserves only.



# ENGINE FAILURES

## ENGINE FAILURE DURING TAKEOFF RUN

1. Throttle – IDLE
2. Brakes – APPLY
3. Wing Flaps – RETRACT
4. Mixture – IDLE CUT-OFF
5. Ignition Switch – OFF
6. Master Switch – OFF

## ENGINE FAILURE IMMEDIATELY AFTER TAKEOFF

1. Airspeed – 70 KTS
2. Mixture – IDLE CUT-OFF
3. Fuel Selector Valve – OFF
4. Ignition Switch – OFF
5. Wing Flaps – AS REQUIRED
6. Master Switch – OFF

## ENGINE FAILURE DURING FLIGHT

1. Airspeed – 78 KTS
2. Boost Pump – ON
3. Fuel Selector – SWITCH TANKS
4. Mixture – RICH
5. Ignition Switch – BOTH, LEFT, RIGHT
6. Transponder – 7700

# **FIRES**

## **DURING START ON GROUND**

1. Cranking – CONTINUE, to get a start which would suck the flames and accumulated fuel through the carb and into the engine.

### **If engine starts:**

2. Power – 1700 RPM for a few minutes
3. Engine – SHUTDOWN and inspect for damage

### **If engine fails to start:**

4. Throttle – FULL OPEN
5. Mixture – IDLE CUT-OFF
6. Cranking – CONTINUE
7. Fire Extinguisher – OBTAIN
8. Engine – SECURE

## **ENGINE FIRE IN FLIGHT**

1. Mixture – IDLE CUT-OFF
2. Fuel Selector Valve – OFF
3. Master Switch – OFF
4. Cabin Heat and Air – OFF

## **ELECTRICAL FIRE IN FLIGHT**

1. Master Switch – OFF
2. Avionics – OFF
3. All Other Switches (except ignition) – OFF
4. Vents/ Cabin Air/ Heat – CLOSED
5. Fire Extinguisher – ACTIVATE (if available)

## **CABIN FIRE**

1. Master Switch – OFF
2. Vents/ Cabin Heat – CLOSED
3. Fire Extinguisher – ACTIVATE

## **WING FIRE**

1. Nav & Strobe Lights – OFF
2. Landing Light – OFF

# Section 4 - Normal Procedures:

## PREFLIGHT INSPECTION

### CABIN:

1. Documentation – Available In Airplane
2. Aeronautical Charts – CURRENT AND APPROPRIATE TO FLIGHT
3. Seat Belt Securing Control Stick – RELEASE
4. Ignition Switches – OFF
5. Master Switch – OFF
6. AVI Battery – OFF
7. Avionics – OFF
8. Engine gages – OFF
9. Flaps – DOWN
10. Alternate Air – CLOSED

### NOSE:

1. Engine Oil Level – CHECK, do not operate with less than 6 quarts
2. Propeller and Spinner – CHECK for nicks and security
3. Cowl Hinge Pins – CHECK for security
4. Air Intakes – CHECK for restrictions
5. Air Inlet – CHECK for restrictions
6. Inspect Engine Compartment - Secure Oil Door
7. Fuel Tank Vents – CHECK for blockage

### RIGHT WING:

1. Aileron – CHECK freedom of movement and security
2. Flap – CHECK security
3. Nav and Strobe – CHECK condition
4. Right Landing Light – CHECK condition
5. Wing Tie-Down – DISCONNECT
6. Main Wheel Tire – **CHECK** for proper inflation (35-38 psi)
7. Chock – REMOVE
8. Right Wing Tank – SUMP
9. Fuel Quantity – CHECK VISUALLY
10. Fuel Filler Cap – SECURE

# PREFLIGHT INSPECTION

## EMPENNAGE:

1. Control Surfaces – CHECK freedom of movement and security
2. Static Sources (both sides of fuselage) – CHECK for blockage
3. Tail and Strobe – CHECK condition
4. Inspect Tailwheel and Springs for Security

## LEFT WING:

1. Aileron – CHECK freedom of movement and security
2. Flap – CHECK security
3. Nav and Strobe – CHECK condition
4. Right Landing Light – CHECK condition
5. Wing Tie-Down – DISCONNECT
6. **REMOVE PITOT COVER**
7. Main Wheel Tire – **CHECK** for proper inflation (35-38 psi)
8. Chock – REMOVE
9. Right Wing Tank – SUMP
10. Fuel Quantity – CHECK VISUALLY
11. Fuel Filler Cap – SECURE

# STARTING PROCEDURES

## BEFORE STARTING ENGINE:

1. Exterior Preflight – COMPLETE / REMOVE CHOCKS
2. Seat Belts and Shoulder Harnesses – ADJUSTED and SECURE
3. Fuel Selector Valve – FULLEST TANK
4. AVI Battery switch – ON (G3X EFIS ON)
5. MSTR Battery – ON
6. Brakes – TEST and HOLD
7. Ignition Circuit Breakers – CHECK IN

## STARTING ENGINE (COLD):

1. Alternate Air – CLOSED
2. Ignition Circuit Breakers - IN
3. Master and AVI Battery Switches – ON
4. Fuel Boost Pump – **ON** until pressure rises then **OFF**
5. Mixture – **FULL RICH**
6. Crack Throttle to 1/4"
7. Propeller – HIGH RPM
8. Propeller Area – CLEAR
9. Ignition Switch – START (3-4 blades to start)
10. Oil Pressure – CHECK min 25 psi at idle
11. Mixture – **MAXIMUM LEAN** for GROUND OPS
12. Alternator – ON
13. Flaps – UP
14. Nav Lights – ON

## STARTING ENGINE (HOT START):

1. Throttle **FULL** Open
2. Mixture – **IDLE CUT OFF**
3. Fuel Boost Pump – **OFF**
4. Propeller – HIGH RPM
5. Propeller Area – CLEAR
6. Ignition Switch – START (3-4 blades to start)
7. THEN PROMPTLY:
  - > Mixture – Feed in gradually
  - > Throttle - Retard quickly to idle
8. THEN STEPS 10+ ABOVE

# BEFORE TAXI

## TAXI:

1. Nav/Strobe Lights – ON
2. Taxi Lights – AS NEEDED
3. Ramp Area – CLEAR
4. Brake – TEST and HOLD
5. Flight Controls – FREE and CORRECT
6. ATIS & Taxi Instructions / Clearance

## BEFORE TAKE OFF:

1. Flight Controls – FREE and CORRECT
2. Elevator Trim – TAKEOFF POSITION
3. Fuel Selector Valve – FULLEST TANK
4. Brakes – HOLD
5. Mixture – RICH (below 5000'). Lean for best power as needed
6. Throttle – 1,700 RPM (at least 100 degrees oil temp)
7. P-Mag Functional Check:
  - a. *> Switch to Left then Right P-Mag, as long as smooth RPM drop is not critical (<100 RPM) confirms that P-Mag is functional, **then***
  - b. *> Remove bus PWR to respective P-Mag, if engine does not cut out P-Mag internal alternator is functioning correctly*
8. Propeller – Cycle 2-3 times, CHECK operation
9. Engine Instruments – Correct Range in GREEN
10. Minimum Oil Temp for Take OFF - 120 degrees
11. Bus Voltage – 13.6 to 14.4 Volts
12. Throttle – IDLE (800 RPM)
13. Flight Instruments and Radios – SET
  - a. *> Altimeter – CORRECT PRESSURE*
  - b. *> BUG DEPARTURE ALTITUDE*
  - c. *> ADHARS – G5/G3X ALIGNED*
  - d. *> GPS – ON Check Signal & WAAS Availability*
  - e. *> Transponder – Check ALTITUDE*
14. Fuel Boost Pump – ON
15. Canopy – Main Latch – SECURE
16. Passenger – Briefed and seatbelt secure

# TAKEOFF PROCEDURES

## NORMAL TAKEOFF:

1. WING FLAPS - UP
2. MIXTURE LEAN ABOVE 5000 FT.
3. ALTERNATE AIR - CLOSED
4. ALIGN ACFT ON RUNWAY **CENTERLINE**
5. AILERON **INTO WIND**
6. SMOOTHLY APPLY FULL POWER
7. LIFT TAILWHEEL WHEN CONTROLS BECOME FIRM
8. ROTATE AT 65 KTS - LIFT NOSE TO CLIMB ATTITUDE
9. CLIMB SPEED -  $V_y$  - 90 KTS
10. CLIMB POWER 25" MP 2,500 RPM
11. TRIM

## SHORT / SOFT FIELD TAKEOFF:

1. WING FLAPS - 10 DEGREES (IN TRAIL WITH AILERON)
2. TAIL UP AS SOON AS POSSIBLE
3. CLIMB AT  $V_x$  - 70 KTS UNTIL CLEAR THEN  $V_y$  - 90 KTS

## NORMAL CLIMB:

1. POWER TO 23" MP 2,300 RPM
2. AIRSPEED - 115-120 KTS
3. MIXTURE - RICH - LEAN ABOVE 5000 FT FOR BEST RPM
4. FUEL PUMP - OFF

## CRUISE:

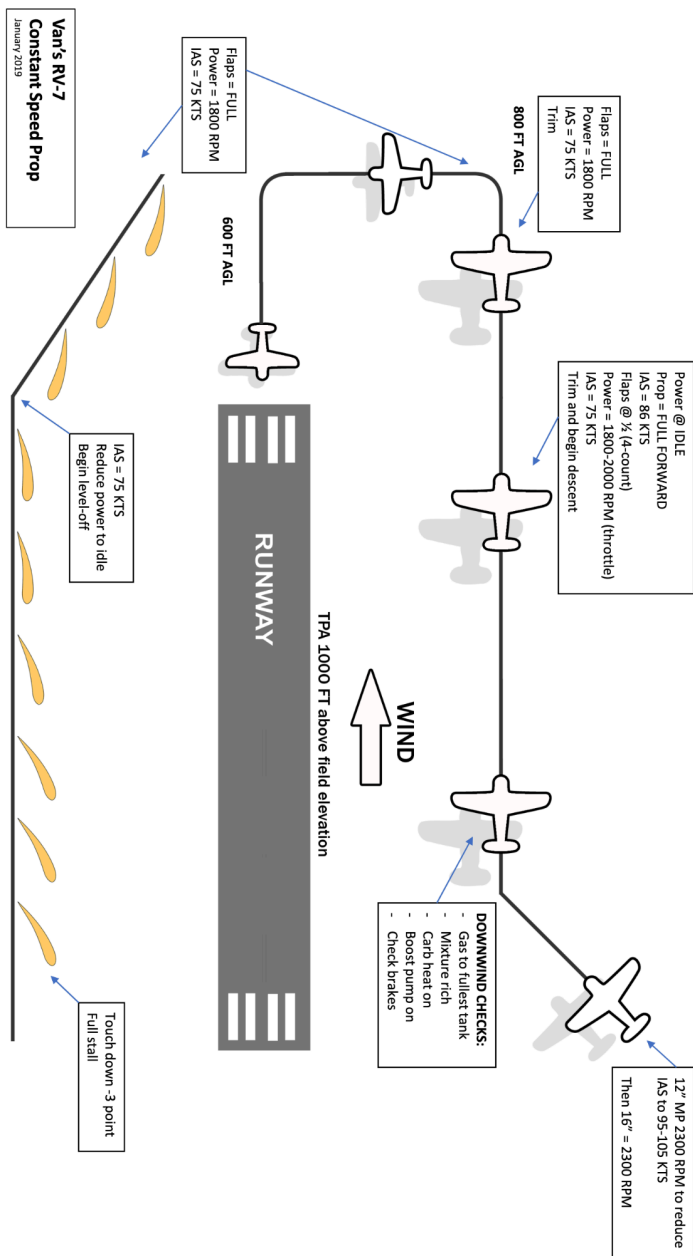
1. LOWER NOSE TO LEVEL ATTITUDE
2. TRIM TO HANDS OFF
3. HIGH SPEED: POWER WOT AND 2,300 RPM = 162 KTAS
4. ECON: 21" AND 2,300 RPM
5. LEAN MIXTURE TO ROP BELOW 65%
6. LEAN TO ENGINE STUMBLES THEN ENRICHEN  $\frac{1}{2}$ -1 TURN IN
7. FUEL FLOW BETWEEN 7 AND 8 GPH AT FULL THROTTLE
8. MONITOR ENGINE:
  - > CHT, BELOW 380 DEGREES CRUISING
  - > TARGET EGT ~1325 DEGREES
  - > TARGET OIL TEMP 180 DEGREES

## **RV POWER SETTINGS**

- T/O POWER AT SEA LEVEL = 30" MP @ 2,700 RPM
- CLIMB POWER = 25" MP @ 2,500 RPM
- CRUISE POWER = 23"MP @ 2,300 RPM
- LOW CRUISE POWER = 16"- 20" MP @ 2,300 RPM
- BEFORE PATTERN POWER = 10" MP @ 2,300 RPM
- PATTERN POWER = 16" MP @ 2,300 RPM
- APPROACH POWER = 1,800 - 2,000 RPM (THROTTLE MOVEMENT) PROP IS ALL THE WAY IN



# LANDING PATTERN SPEEDS AND SETTINGS



# LANDING PROCEDURES

## BEFORE LANDING @ 10 MILES:

1. ATIS, ALTIMETER SET, PLAN APPROACH AND RUNWAY
2. SLOW FROM CRUISE SPEED
3. FUEL - FULLEST TANK
4. MIXTURE - RICH AS NEEDED
5. BOOST PUMP - ON
6. LANDING LIGHTS ON
7. BRAKES - TEST

## APPROACH TO LANDING @ 5 MILES:

1. ALTITUDE - MAINTAIN 1,000' AGL UNTIL 3 MILES
2. SLOW TO 85 KTS
3. IF STRAIGHT IN BEGIN DESCENT AT 3 MILES @ 75 KTS

## LANDING:

1. ABEAM TOUCHDOWN - REDUCE POWER TO 1,800 - 2,000
2. PROPELLOR - FULL FORWARD
3. WING FLAPS - 1/2 BELOW 86 KTS
4. AIRSPEED - TRIM FOR 75 KTS
5. POWER - 1,800 - 2,000 RPM
6. TRIM ELEVATOR FOR HANDS OFF FLIGHT
7. FINAL LEG - FLAPS FULL @ 500' - AIRSPEED = **75 KTS**
8. SHORT FINAL = **70 KTS**, OVER NUMBERS **68-70 KTS**
9. TOUCHDOWN 3 POINT ATTITUDE - FULL STALL
10. LANDING ROLL - BRAKES AS REQUIRED

## BALKED LANDING:

1. THROTTLE - FULL OPEN
2. PITCH - ESTABLISH CLIMB ATTITUDE
3. WING FLAPS - UP
4. AIRSPEED - 90 KTS

## AFTER LANDING:

1. WING FLAPS - UP
2. BOOST PUMP - OFF
3. LIGHTS (Landing and Strobe) - OFF
4. MIXTURE - AGGRESSIVELY LEAN

# AFTER LANDING PROCEDURES

## ENGINE SHUTDOWN:

1. PROPELLER – FULL FORWARD
2. MIXTURE - ENRICHEN FOR RUNUP
3. THROTTLE – INCREASE TO 1,500 RPM FOR 30 SEC
4. TEMPS STABILIZED THEN REDUCE TO IDLE
5. MIXTURE – IDLE CUT-OFF **and** THROTTLE CLOSED
6. AFTER ENGINE STOPS - CRACK MIXTURE
7. FLAPS – DOWN
8. SWITCHES – OFF
9. AVIONICS AND INSTRUMENTS – OFF
10. P-MAGS – OFF
11. MASTER – OFF
12. AVIONICS BATTERY – OFF
13. FUEL SELECTOR - OFF
14. REMOVE PLACE KEY ON GLARE SHIELD!

## SECURING AIRCRAFT:

1. VENT ENGINE COMPARTMENT - OPEN OIL DOOR
2. WHEEL CHOCKS - AS NEEDED
3. WING AND TAIL TIE DOWNS - AS NEEDED
4. CONTROL LOCKS SECURED - AS NEEDED
5. ENGINE AND AIR INTAKES PLUGS IN PLACE
6. PITOT COVERED
7. CANOPY COVER AS NEEDED
8. KEY – REMOVED

# Section 5 - Performance:

## PERFORMANCE – SPECIFICATIONS

SPAN:	25' 0"
LENGTH:	20' 4"
HEIGHT:	7' 10"

### PERFORMANCE @ Gross Weight 1800 lbs:

SPEED:	
Maximum at Sea Level	181 KTS
Cruise, 75% Power at 8,000 Ft	173 KTS
Cruise, 55% Power at 8,000 Ft	155 KTS

RANGE (includes 3 gal. for taxi, takeoff & climb):	
75% @ 8000', no reserve	775 sm / 673 nm
55% @ 8000', no reserve	950 sm / 825 nm
75% @ 8000', one hour (10 gal) reserve	500 sm / 434 nm
55% @ 8000', one hour (10 gal) reserve	680 sm / 590 nm

RATE OF CLIMB AT SEA LEVEL	1,650 fpm
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SERVICE CEILING	20,500 ft
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TAKEOFF PERFORMANCE:	575 ft
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LANDING PERFORMANCE:	500 ft
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STALL SPEED (CAS):	
Flaps Up, Power Off	54 KTS
Flaps Down, Power Off	50 KTS

MAXIMUM WEIGHT (Normal Category):	1,800 Lbs
EMPTY WEIGHT	1,115 Lbs
MAXIMUM USEFUL LOAD:	685 Lbs
BAGGAGE ALLOWANCE	100 Lbs
WING LOADING (Pounds/ Sq. Ft)	14.8 Lbs
POWER LOADING (Pounds/ HP)	10 Lbs
FUEL:	

Capacity:	42 Gal Total
Unusable:	2 Gal
Type:	100 LL

ENGINE: Lycoming Thunderbolt

YIO-360-EXP139 = I(O)360-M1B  
S/N: EL-38733-51E  
180 BHP @ 2700 RPM

OIL CAPACITY

8 Qts Max  
6 Qts Min Normal Qty  
4 Qts Min Safe

PROPELLER: Hartzell

HC-M2YR-1BFPX/F7497-2X

## Section 6 - Aerobatic Information:

Weight Limitation – **1,600** lbs

Recommended Entry Speeds:

Loops, Horizontal Eights	121-165 KTS
Immelmann Turns	130-165 KTS
Aileron Rolls, Barrel Rolls	104-165 KTS
Vertical Rolls	156-165 KTS
Split-S	87-95 KTS

# Section 7 - Weight & Balance:

## WEIGHT AND BALANCE DATA

Make: Van's Aircraft  
Model: RV-7  
Serial Number: 74261  
Registration: N732DH

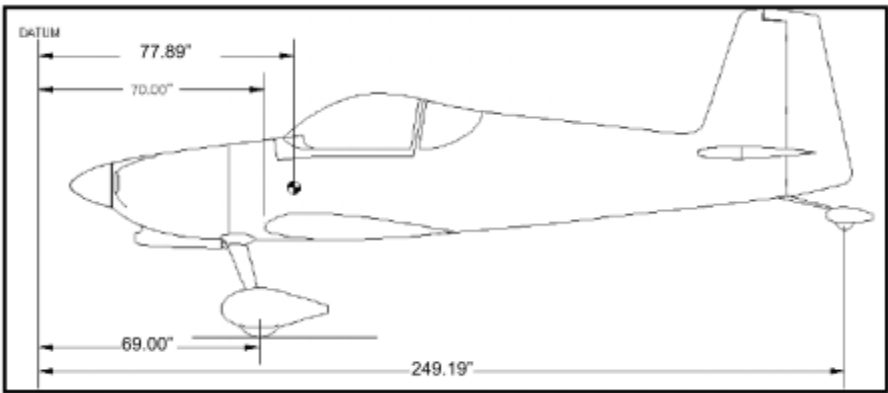
### Maximum Weights:

Aerobatic Category ..... 1,600 Lbs  
Normal Category ..... 1,800 Lbs

Datum = ..... 70 inches forward of the wing leading edge (L.E.)  
DESIGN CG Range = 15% to 29% of wing chord or 8.7" to 16.82" from L.E. or 78.7" to 86.82" aft of Datum

**AEROBATIC CG Range = 15% to 25% of wing chord or 8.7" to 14.5" from L.E. or 78.7" to 84.5" aft of Datum**

Wing L.E. = ..... 70 inches aft of datum  
Fuel = ..... 80" aft of datum  
Pilot and Passenger = 97.48" aft of datum  
Baggage = ..... 126.78" aft of datum



Main wheel right = ..... 69.00" aft of datum  
Main wheel left = ..... 69.00" aft of datum  
Tail Wheel = ..... 249.19" aft of datum

## WEIGHT AND BALANCE DATA

Aircraft weighed empty in level flight attitude including 6 qts. of oil, no fuel.

**Make: Van's**  
**Model: RV-7**

**Serial: 74261**  
**Reg: N732DH**

**Weighed on October 18, 2020**

<b>R-Main</b>	<b>69.00 Aft of datum</b>
<b>L-Main</b>	<b>69.00 Aft of datum</b>
<b>Tail</b>	<b>249.19 Aft of datum</b>

### **WEIGHTS:**

<b>L-Main</b>	<b>R-Main</b>	<b>Tail</b>	<b>Total</b>
<b>532</b>	<b>528</b>	<b>55</b>	<b>1,115.00</b>

<b>COMPONENT:</b>	<b>Weight</b>	<b>Arm</b>	<b>Moment</b>
L-Main	532	69.00	36,708.00
R-Main	528	69.00	36,432.00
Tail	55	249.19	13,705.45

<b>Subtotal:</b>	<b>1,115.00</b>	<b>86,845.45</b>
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<b>CG:</b>	<b>77.89 Aft of datum for <u>empty</u> weight CG</b>
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<b>STD</b>	<b>FWD:</b>	<b>78.70</b>	<b>AFT:</b>	<b>86.82</b>
<b>AEROBATIC</b>	<b>FWD:</b>	<b>78.70</b>	<b>AFT:</b>	<b>84.50</b>



# WEIGHT AND BALANCE SAMPLES

SAMPLE SITUATION 1: GROSS WEIGHT CG				
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	42	252	80.00	20,160.00
Pilot		200	97.48	19,496.00
Co-Pilot		140	97.48	13,647.20
Baggage		93	126.78	11,790.54
<b>Total</b>		<b>Weight: 1,800</b>		<b>151,939.19</b>
		<b>CG: 84.41</b>		

SAMPLE SITUATION 2: FURTHEST AFT CG (Gross Weight, min. fuel)				
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	5	30	80.00	2,400.00
Pilot		200	97.48	19,496.00
Co-Pilot		140	97.48	13,647.20
Baggage		93	126.78	11,790.54
<b>Total</b>		<b>Weight: 1,578</b>		<b>134,179.19</b>
		<b>CG: 85.03</b>		

SAMPLE SITUATION 3: MOST FORWARD CG W/ 200 LBS PILOT				
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	42	252	80.00	20,160.00
Pilot		200	97.48	19,496.00
Co-Pilot		0	97.48	0.00
Baggage		0	126.78	0.00
<b>Total</b>		<b>Weight: 1,567</b>		<b>126,501.45</b>
		<b>CG: 80.73</b>		

## WEIGHT AND BALANCE SAMPLES

SAMPLE SITUATION 4:		MOST FORWARD CG W/ 200 LBS PILOT		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	42	252	80.00	20,160.00
Pilot		200	97.48	19,496.00
Co-Pilot		0	97.48	0.00
Baggage		0	126.78	0.00
<b>Total</b>		<b>Weight: 1,567</b>		<b>126,501.45</b>
		<b>CG: 80.73</b>		

SAMPLE SITUATION 5:		GROSS WEIGHT W/HEAVY PILOT/PSG AND REDUCED FUEL		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	22	132	80.00	10,560.00
Pilot		230	97.48	22,420.40
Co-Pilot		230	97.48	22,420.40
Baggage		93	126.78	11,790.54
<b>Total</b>		<b>Weight: 1,800</b>		<b>154,036.79</b>
		<b>CG: 85.58</b>		

SAMPLE SITUATION 6:		AS ABOVE, BUT WITH MINIMUM FUEL		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	5	30	80.00	2,400.00
Pilot		230	97.48	22,420.40
Co-Pilot		230	97.48	22,420.40
Baggage		94	126.78	11,917.32
<b>Total</b>		<b>Weight: 1,699</b>		<b>146,003.57</b>
		<b>CG: 85.94</b>		

# WEIGHT AND BALANCE SAMPLES

SAMPLE SITUATION 7:		AS ABOVE, BUT WITH REDUCED BAGGAGE		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	5	30	80.00	2,400.00
Pilot		230	97.48	22,420.40
Co-Pilot		230	97.48	22,420.40
Baggage		0	126.78	0.00
<b>Total</b>		<b>Weight: 1,605</b>		<b>134,086.25</b>
		<b>CG: 83.54</b>		

SAMPLE SITUATION 8:		NO PSGR, NO BAGGAGE, MINIMUM FUEL		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	5	30	80.00	2,400.00
Pilot		200	97.48	19,496.00
Co-Pilot		0	97.48	0.00
Baggage		0	126.78	0.00
<b>Total</b>		<b>Weight: 1,345</b>		<b>108,741.45</b>
		<b>CG: 80.85</b>		

# WEIGHT AND BALANCE SAMPLES

SAMPLE SITUATION 9:		AEROBATIC - MAX GROSS = 1,600 Lbs		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	20	120	80.00	9,600.00
Pilot		200	97.48	19,496.00
Co-Pilot		165	97.48	16,084.20
Baggage		0	126.78	0.00
Total		Weight: 1,600		132,025.65
		CG: 82.52		

SAMPLE SITUATION 10:		AEROBATIC - MAX GROSS = 1,600 Lbs		
Component	Gal.:	Weight	Arm	Moment
Aircraft		1,115	77.89	86,845.45
Fuel	42	252	80.00	20,160.00
Pilot		200	97.48	19,496.00
Co-Pilot		0	97.48	0.00
Baggage		0	126.78	0.00
Total		Weight: 1,567		126,501.45
		CG: 80.73		

# Section 8 - Equipment List:

N732DH EQUIPMENT LIST:

Updated: January 1, 2024

DESCRIPTION	MFG	MODEL	SERIAL #	REVISION	AFT DATUM LOCATION	COMMENTS
<b>PROPELLER:</b>						
Constant Speed	Hartzell	HC-M2YR-1BFPX	B37303B	N/A	N/A	Extended Hub for James Cowl
Blade 1	Hartzell	F7497-2X	M10271	N/A	N/A	
Blade 2	Hartzell	F7497-2X	M10273	N/A	N/A	
<b>ENGINE:</b>						
Thunderbolt	Lycoming	Y10-360-EXP139	EL-38733-51E	N/A	N/A	180 HP
Oil Filter	Champion	CH48110-1	N/A	N/A	N/A	or Tempest AA48110-2
Oil Cooler	Aero-Classic	8001802	4655567	N/A	N/A	High capacity
Left P-Mag	E-Mag	P-114-L4	7100	VL-184117	N/A	
Right P-Mag		P-114-L4	7022	VL-184117	N/A	
Spark Plugs	NGK	BR8ES	N/A	N/A	N/A	Stock #3961 solid post
Fuel Servo	Avstar	AVX3015002-1	AV53190736	AVX-NNSS8	N/A	Avstar Servo Kit - AVX360-1
Flow Divider		AVX3015004-1	AV53190987	3015002	N/A	
Nozzles		AV2524864-2	N/A			
Prop Governor	MT	P-860-4	21G054-K	N/A	N/A	
Starter	Sky-Tec	149-12HT	H-X112397	N/A	N/A	High torque
Fuel Pump	Lycoming	62B26931	4123L0084	N/A	N/A	Engine Driven
Fuel Flow	Electronics International	FT-60	228565	N/A	Under Plenum	TSO-C44c
Auxiliary Pump & Filter	Airflow Performance	2090255 / 1090079	N/A	N/A	N/A	Forward of Selector Valve
Alternator	B&C	LX60	2BB10D9M1	N/A	N/A	14V/60A
Regulator	B&C	LR3D-14	N/A	N/A	64"	Sub panel
Alternator Belt	Gates	7360	N/A	N/A	N/A	
<b>AVIONICS:</b>						
G3X EFIS	Garmin	GDU 460	350010844	SW 9.31	Panel	ID: 60002C9E8F5D6
Backup Attitude	Garmin	G5	4JQ063696	SW 8.41	Panel	
GPS/ADS-B	Garmin	GNX-375	5GJ004668	Main 3.22 Trans 2.80 Touch 2.00 GPS 8.20	Panel	ID: A310060FF984C
COMM 1	Garmin	GTR-200	5JC004450	SW 3.50 BB 2.40	Panel	
COMM 2	Garmin	GTR-20	5JE004310		64"	Sub-panel aft FW
Audio	Garmin	GMA-245	3YL003456	ASW 2.20 BT 2.00 SW 3.10 BB 2.30	Panel	
Autopilot	Garmin	GMC-507	5H1008389	SW 5.00 BB 2.30	Panel	
Roll Servo	Garmin	GSA-28	6N7005112	SW 5.60	Right Wing	At Aileron Bellcrank
Pitch Servo	Garmin	GSA-28	2PG007451	SW 5.60	130"	Behind Baggage Compartment
Engine Monitor	Garmin	GEA 24	2J2021304	SW 3.90 BB 2.20	64"	Sub-panel
MAG	Garmin	GMU 11	56J023476	SW 2.20 BB 2.10	160"	Behind Baggage Compartment
ARINC	Garmin	GAD 29C	6V9001122	SW 2.10 BB 2.00	64"	Sub-panel
ADHARS	Garmin	GSU 25C	5Q2101793	SW 2.90 BB 2.10	Panel	Attached to G3X
Pitot	Garmin	GAP 26	34F600819	-	Left Wing	Heated, un-regulated
USB PWR	Garmin	GSB 15	63M250617	-	Panel	USB Type A & C
CO Detector	Aithre	EX-2.0	2OE7	-	Panel	Co-Pilot side
ELT	ACK	E-04	36420	N/A	150" Behind Baggage Compartment	HEX: 2DC8A 41C86 FFBF, Registered NOAA, Exp 1-24-24

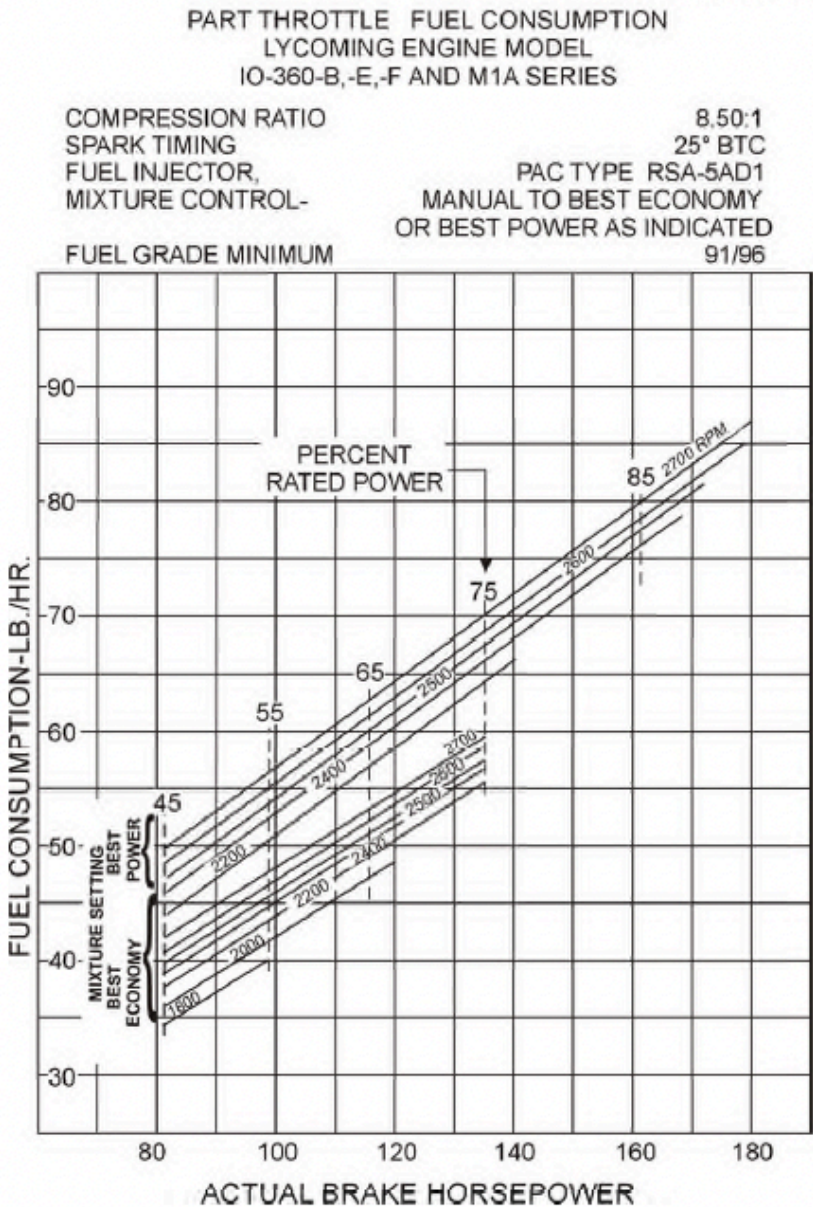
# Section 8 - Equipment List:

N732DH EQUIPMENT LIST:

Updated: January 1, 2024

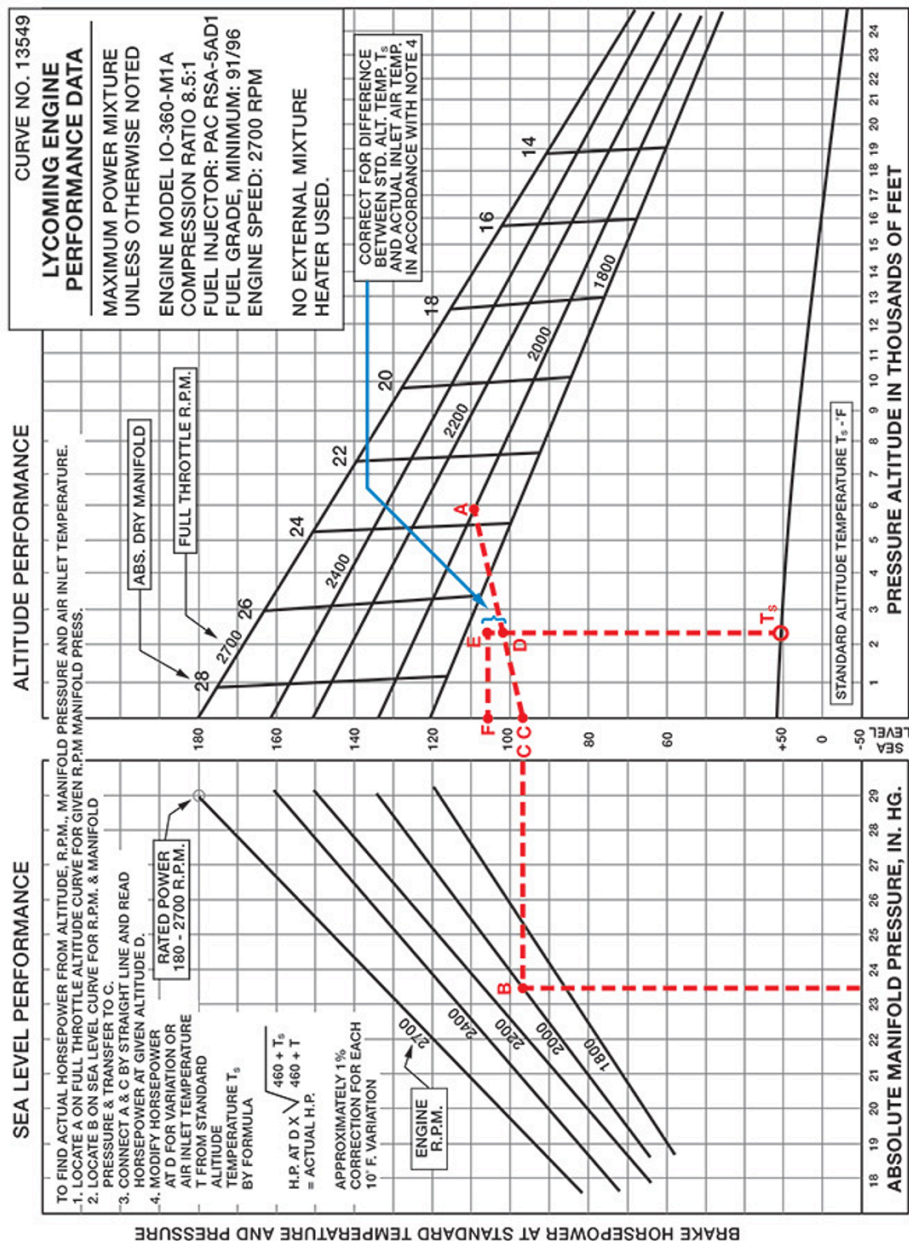
DESCRIPTION	MFG	MODEL	SERIAL #	REVISION	AFT DATUM LOCATION	COMMENTS
<b>ELECTRICAL:</b>						
PWR Distribution	Vertical Pwr	VPX-Pro	3558	1.06 Rev D	64"	Between Sub-panel and Firewall
Battery	Odyssey	PC680	-	-	74"	w/attached trickle charge lead
AVI Bkup Battery	IBBS	IBBS-12V-6AH	20219319 H	-	64"	Above VPX - Remove with VPX
<b>ANTENNAS:</b>						
COMM 1/2	RAMI	RV-17	-	-	N/A	COMM 1 L, COMM 2 R
ELT/ADS-B Trans	Mono-pole	104-17	-	-	N/A	Belly near FW
PFD GPS	Garmin	GA 26C	-	-	N/A	On glare shield
WAAS GPS	Garmin	GA 35	-	-	N/A	Under cowl
<b>LIGHTS:</b>						
Wing Nav/Strobe	AeroLED	Pulsar NS	M22982-002 M22982-012	-	N/A	Left/Right
Tail Nav/Strobe	AeroLED	Suntail	-	-	N/A	Rudder
LNDG Light	AeroLED	Aerosun VX	-	-	N/A	Left/Right w/WigWag via VPX

Section 9 - Engine Operations:  
Lycoming Data: PN 60297-12



## Section 9 - Engine Operations:

### Lycoming Data: PN 60297-12





## Section 10 - Miscellaneous:

### FAR 91.125 - ATC light signals.

ATC light signals have the meaning shown in the following table:

Color and type of signal	Meaning with respect to aircraft on the surface	Meaning with respect to aircraft in flight
Steady <b>Green</b>	Cleared for takeoff	Cleared to land
Flashing <b>Green</b>	Cleared to taxi	Return for landing
Steady <b>Red</b>	Stop	Give way to other aircraft and continue circling.
Flashing <b>Red</b>	Taxi clear of runway in use	Airport unsafe—do not use
Flashing <b>White</b>	Return to starting point on airport	N/A
Alternating <b>Red</b> and <b>Green</b>	Exercise extreme caution	Exercise extreme caution

### Compass Headings, VFR under 18,000ft

Course	Altitude
0 – 179 degrees	Odd thousand +500
180 – 360 degrees	Even thousand + 500

### TEMPERATURE Conversion table:

Celc.	Fahr.	Celc.	Fahr.	Celc.	Fahr.	Celc.	Fahr.	Celc.	Fahr.
10	50	60	140	110	230	160	320	210	410
16	60	66	150	116	240	166	330	216	420
21	70	71	160	121	250	171	340	221	430
27	80	77	170	127	260	177	350	227	440
32	90	82	180	132	270	182	360	232	450
38	100	88	190	138	280	188	370	238	460
43	110	93	200	143	290	193	380	243	470
49	120	99	210	149	300	199	390	249	480
54	130	104	220	154	310	204	400	254	490