## Estimate Remaining Useful Life of Turbofan engines



Photographer: Luke Sharrett/Bloomberg

One of the main challenges in the aviation industry is to reduce maintenance costs an downtime of machines by maintaining or improving safety standard. A large per cent costly delays are a result of unplanned maintenance such as when an aircraft has an a behaviour on the previous flight, creating an operational disruption and even requiring change.

Since you don't know when failure will occur, you have to be conservative in your plan especially if you're operating safety-critical equipment like engines. But by scheduling maintenance very early, you're wasting machine life or spare part life that is still usable adds to costs to the owner.

However, if you can predict when machine failure will occur, you can schedule mainte before it. By implementing predictive maintenance we can minimize unnecessarily sol maintenance and production hours lost during maintenance (improve the overall avail equipment) and reduce the cost of spare parts and its consumables during the maintenance.

In order to develop an algorithm that predicts the breakdown of a piece of equipment given time window (typically some number of days), we require enough historical data us to capture information about events leading to failure.

## About Data set:

In this post, we are using the C-MAPSS dataset, which is engine degradation simulatic carried out at NASA by using C-MAPSS simulation software. C-MAPSS has created fou data sets simulated under different combinations of operational conditions and fault i

Data sets are consist of three operational settings and 21 sensor measurements (tem pressure, fan speed, etc.) for several engines and for every cycle of their lifetime.

The engine is operating normally at the start of each time series and develops a fault as during the series. In the training set, the fault grows in magnitude until system failure set, the time series ends sometime prior to system failure.

## Problem statement:

The goal of predictive maintenance is to predict at the time t, using the data up to that whether the equipment will fail in the near future.

This problem is can be formulated in two ways:

Classification: we are aims to predict the probability that the equipment will fail with specified time window.

**Regression:** A regression-based approach to which aims to estimate the remaining till end of the equipment's useful life of an engine.

In this blog, we'll focus on the second dataset (FD002) in which all engines develops with six operating conditions.



### **Data Exploration**

Show Train dataset

The section of the section of

Before we shart EDA, we need to compute a tenget veriable his this data and which is it QUA; Kormaning Orania Life(the length of time a machine is likely to operate before it repoir or replacement.

As mention about the definition,

 $86. \ \circ s$  total nebel of the 115s cycle engine - listent engine 115s cycle

To calculate FUL in our training data set, we just did group by engine number(max multiplicity) and prevent engine life carlie.

For test data, the time auries ends cometime union to system failure, often that we don

Courtgles for consider the wegine one-life cycle dess. In a training data life cycle of win from 1 and end 223. In tool data, the engine runs owner cycle like 120, we deed to fin many cycles still remain.

To concate the SLA in our tent data bet, see that the same thing an training data by adding truth data value to test data.

-

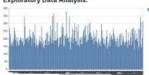
Data with RID calculation

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8.	1.	2	12,861,6814	8.606	340	247,722
2	1.	1.	Y, 810, 6342	9.6218	40	231.966
2	1	4	13.803,9476	0.0400	300	347,000
			W 456 4834	W 45 WH.	-	while warm

Since the degradation is a system will generally remain negligible until after socie per specialism time. The muly wiel lighter IRUs values are protably unmonateable. We can't anything about the IRUs before that point because see house no interroation about the it and how.

Show Statch of the IRU, of Engines

#### **Exploratory Data Analysis**



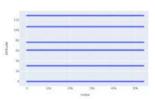
- Engines have different the durations. The overage working time in train data is: with a trivianum of 128 and a maximum of 378.
- With of engines are within IDE (the cycle and 4 % crity engines lived above IDE).

  this engines are consider as an extreme case.

#### Operational Setting:

What column to you must be obeing

Attitude



As plots illustrate, all engines are operated six different abitive conditions with a run, 12000M,) and with 5 different operating speed(Plach number) of (0+0.84), with two to the condition of (0+0.84), with two to the conditions of (0+0.84).

### sensors data

arm is you next to clocky



- surroury data remain some throughout engine's like.
- use understand that services data T2,T2,ET30,T50,P2,P15,F30 highly constate unother (Based on Temperature and pressure graph look like sames).
- samples Will well No above correlated and also With and Will
- As we can see, all the sensors data is highy correlated, even data is highly correlated, even data is highly correlated.

Let's plot PDF individual sensors for any given engines.

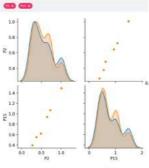
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O Pressure services

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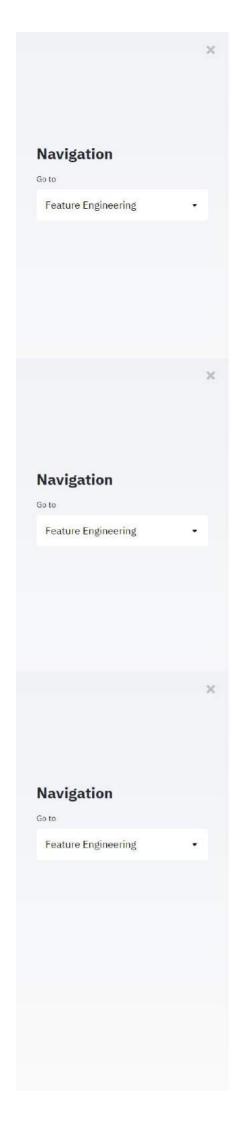
Select horn-Pressure serve



Engine number is  $\pm$  , total number of engine cycle(RRE) in [68]

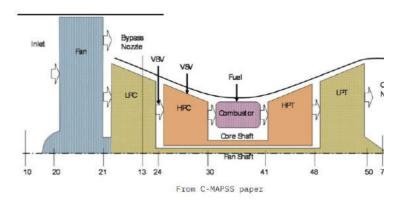
Engine number is 5 , total number of engine cycle(RAL) is  $354\,$ 

when we are comparing this engines sensors data are can't reflict/fecentiate between by booking into PDF plots for this data set.



## Feature engineering

Based on my domain knowledge i come up with 40 new features which can improve n performance. This features are performance parameters of turbofan engines like, Thrust, Propulsion efficiency, Thermal efficiency, Turbine Entry Temperature (TET), etc...



Schematic diagram showing the components, station numbers and sensor data taken C-MAPSS engine. A brief description some of the major modules is summarized below

Exhaust Gas Temperature (EGT)(T90): expressed in Kelvin, is the temperature at the exhaust and a measure of an engine's efficiency in producing its design level thrust; the EGT the more wear and deterioration affect an engine. High EGT can be an indicat degraded engine performance. An exceedance in EGT limits can lead to immediate da engine parts and/or a life reduction of engine parts. With this in mind it then becomes important to keep the EGT as low as possible for as long as possible.

**Turbine Inlet Temperature(T41):** The general trend is that raising the turbine inlet te increases the specific thrust of the engines with a small increase in fuel consumption rate. However, high temperatures can damage the turbine, as the blades are under lark centrifugal stresses and materials are weaker at high temperature.

Thermal efficiency is a prime factor in turbine performance. The three most importan affecting the thermal efficiency are turbine inlet temperature (T41), compression ratio and the component efficiencies of the compressor and turbine. Other factors that affe efficiency are compressor inlet temperature (P21) and combustion efficiency. Maximul efficiency can be obtained by maintaining the highest possible exhaust temperatures life is greatly reduced at high turbine inlet temperatures, the operator should not excee exhaust temperatures specified for continuous operation.

All this feature are created using given sensor data with some assumptions. I am not discuss all the feature hear. Please ref to [1],[2]

More about my Feature engineering GitHub

### Improvement:

- From EDA we understand that data set having different operating conditions, so create clustering and do labeling on top that create onehotencoder but this only FD002 and FD004 datasets
- · If possible we can create feature engineering with autoencoder.
- If we can remove noise from data, we can achieve good results.Let's try noise n
  effectively by using the autoencoder or other noise reduction techniques



## **Model Building**

As mention early in home page we are going to solve both regression and classification

After completing feature engineering, we got total 6:3 feature and applied random for algorithm to select feature and we got 23 feature.

We have 4 different data set with different operating condition. Please Select data as

Please choose a Data set Type

#### Scaling:

Sklearns' MinMaxScaler can be used to create a scaler fitted to our train data. The del settings create a scaler for scale our training features between 0–1. The scaler is then both our x\_train and x\_test set.

Shape of train data: (28531, 26)

Shape of test data: (13096), 263

#### Data processing:

By defauld we are having Train and Test datasets,we keep Train data for train and split into cross, selfdation and test data, we pick the lest sequence data points as test data data points as cross, validation.

we are taken RUL and Label columns as dependent variable, remaining 23 columns as independent variable and remaining one columns is Engine. No column.

For classification approach, generating the label by conself in this way, from 0 to 30 cyllabeled as  $1(\ln \delta)$  and rest (>30) as 0(Mot fail). For regression problem we keep RIX, or without any changes,

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78	4	8.3998	0.0424	0.3690	0.6333	6.2659	6.1996	8.36
1	1	#,3526	+.2121	0.3810	8,7669	8.2794	8.1628	8.41
2	1	0.3705	0.2727	0.2580	0.7863	8.2280	6.1730	8.36
- 12	1	0.3353	0.3392	0.1667	8.8191	8.2941	8-1749	ff. 16
14	1	0.4506	0.2424	0.2560	8.2463	8.2353	8.1242	11.00

After split data we process data for machine learning and deep learing algorithm little which is followed below.

### Choose the algorithm

Inase Chickes a Piccal Type

Shape of Train data after split: (29631, 233

Shape of Train labeling data: (2861)

Shape of Cross validation data after split: (129%, 23)

Shape of Cross validation labeling data: (12996,).

Shape of Test data after split: (188, 23)

Shape of Test labeling data ( (106, )

Show code for ML data processing

Please choose a Model from list

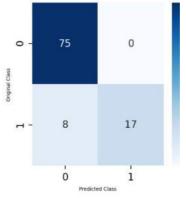
Logistic regression

Let's train Logistic regression on FD001 data set.
The train log loss is: 0.8955228858812181

The cross velidation log loss is:  $\pm .838429358297633476$ 

The test log loss is: 6.2258766212223636

Confusion matrix



Number of mis-classified points 1 8:0

precision: 0.9636544576313253

recall: 1.8

Incore: 0.9493678895875949

Negative class labels: 7% positive class labels: 25

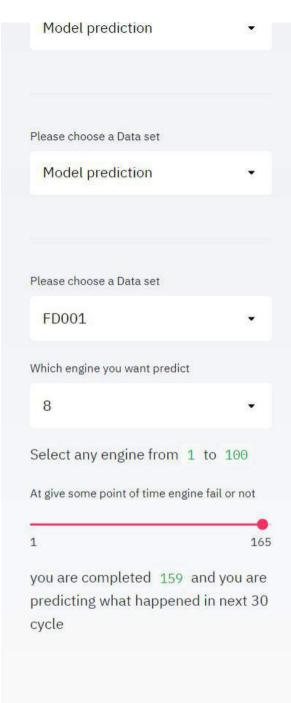
### Conclusion

The task of predicting RUL is certainly interesting But there is still quite a bit of room I improvement with botter training, models, and teature organizing.

Our model give good results,but that's not good enough for aviation standards we nee improve model performance as follows:

- Geep earning modes are not performing as we expected maybe deep learning performs well if have large data sets.
- As we know that data generated by sensors, it means that the data set is contail
  with sensor noise, we can improve model performances by applying denoising?

You can find the full code on my github page bere



# **Model Prediction**

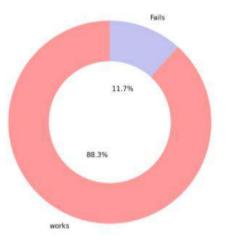
Let's predict how models are performing by giveing unseen data.

Selected raw data file from

https://raw.githubusercontent.com/gethgle/predictive-maintenance/main/data/test

Probability of Engine 8 failure with next 30 life cycle:

Remaining useful life of Engine 8:



Predicted cycles remaining: 89

Made with Streamlit