

# Detailed feasibility study of a gamma ray detector system for nanosatellites using GEANT4 simulations

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## Introduction

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The main aim of this thesis is to simulate the detection of  $\gamma$ -rays from GRBs for the CAMELOT (Cubesats Applied for MEasuring and LOcalising Transients) mission. This CubeSat (miniaturized satellite) is being developed by C3S LLC, in Budapest. The Geant4 simulation described in this paper also predicts the background signal originating from solar and cosmic electrons.

The simulation was first validated by comparing the spectrum of an  $^{241}\text{Am}$  source measured in laboratory with the spectrum obtained with the simulation. Secondly the photon yield dependence of the position of the source above the scintillator detector was determined by the simulation. The results agreed with the experimental data.

The satellite consists of seven modules, each with a specific material composition. Therefore the CAD model of the satellite had to be imported into the simulation in order to understand how the satellite itself would affect the measurements. The validated scintillator afterwards was placed on the side of the satellite in the simulation.

The detector and the satellite was radiated with a parallel  $\gamma$ -rays. The energy spectrum of these particles was chosen to be the same as GRB 9900123. In order to understand how the satellite would absorb the  $\gamma$ -rays several simulations were carried out by chaning the incident angle of the beam around the longitudinal axis of the satellite.

Secondly the effect of solar and cosmic electrons was investigated. The satellite was radiated by electrons with an energy distribution obtained from the SPENVIS information system. This distribution represents the energy spectrum of electrons in polar orbit 500 km above the surface of Earth.

## Gamma-ray bursts

Gamma-ray bursts (GRBs) [1, 2, 3, 4] are one of the most investigated and yet least understood astrophysical objects. They have been studied for more than forty years as one of the most extreme explosive events in the Universe. Their origin is still not yet fully understood. GRBs were discovered in 1967–73 by military satellites. The scientific community only got to know their existence in the early seventies [5]. A GRB event is brighter than any other object in the sky during its appearance that can last from a millisecond to minutes. Every day a new GRB is observed by the satellites investigating them. The main part of the energy spectrum of these events in the keV to the MeV ranges.

After a decade of investigation it was found that these events can be categorized into two groups by their length [6, 7, 8]. The first group is called long GRBs with softer spectrum and with prompt  $\gamma$ -ray emission of  $\geq 2$  s. These GRBs were linked to the gravitational collapses of massive stars. These objects are associated with type Ic core-collapse supernovae. The second group of GRBs are called short GRBs as their prompt  $\gamma$ -ray emission lasts  $\leq 2$  s. These objects also possess a harder spectrum. They were linked to the merge of two very compact objects, such as neutron star - neutron star (NS-NS) and neutron star - black hole (NS-BH) mergers [9].

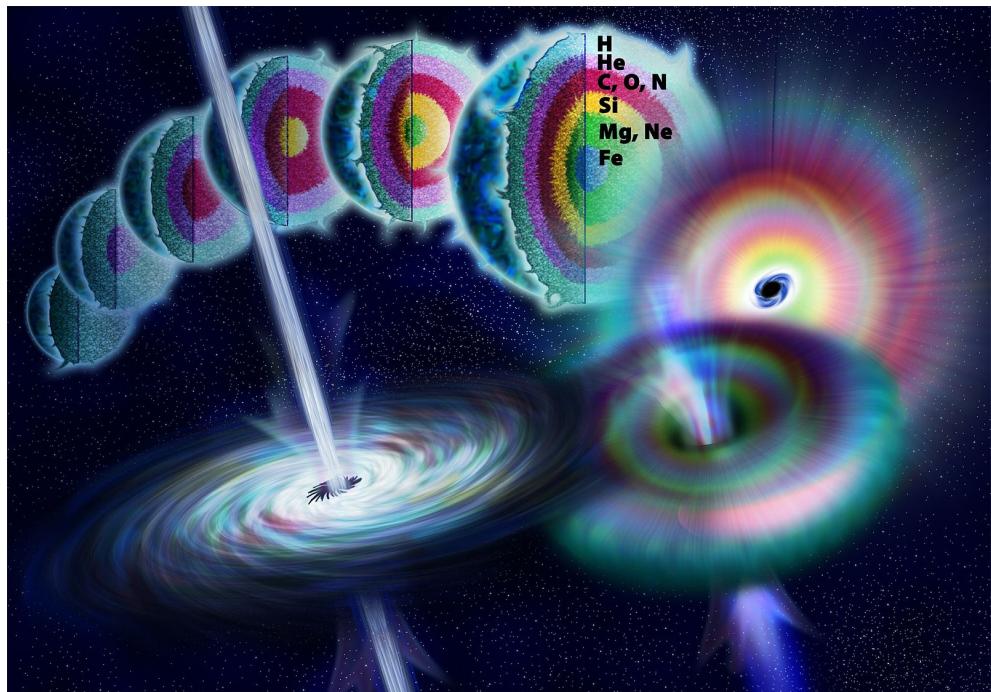


Figure 1.1: The life cycle of a massive star. When fusion stops the collapsing star releases energy in the form of jets observed as a gamma-ray burst

One of the most accepted theories describe GRBs as collisions of highly

relativistic outflow of the accelerated jets [10]. For most of the observed GRBs the prompt  $\gamma$ -ray emission was followed by a longer-lasting afterglow in soft X-ray, optical or radio waves [11]. This is due to the propagation of a relativistic shockwave through the medium that surrounds the burst. Observations proved that GRBs are at cosmological distances [12]. The farthest GRB observed was at a distance of  $z=9.4$ . The average short GRBs lie at a distance of  $z=0.5$ . The long GRBs are even farther away, on average at  $z=2$ . The energy that is released in such an event is tremendous, about 10 erg. Long GRBs have been linked to the brightest regions of galaxies those contain the the most massive stars [13].

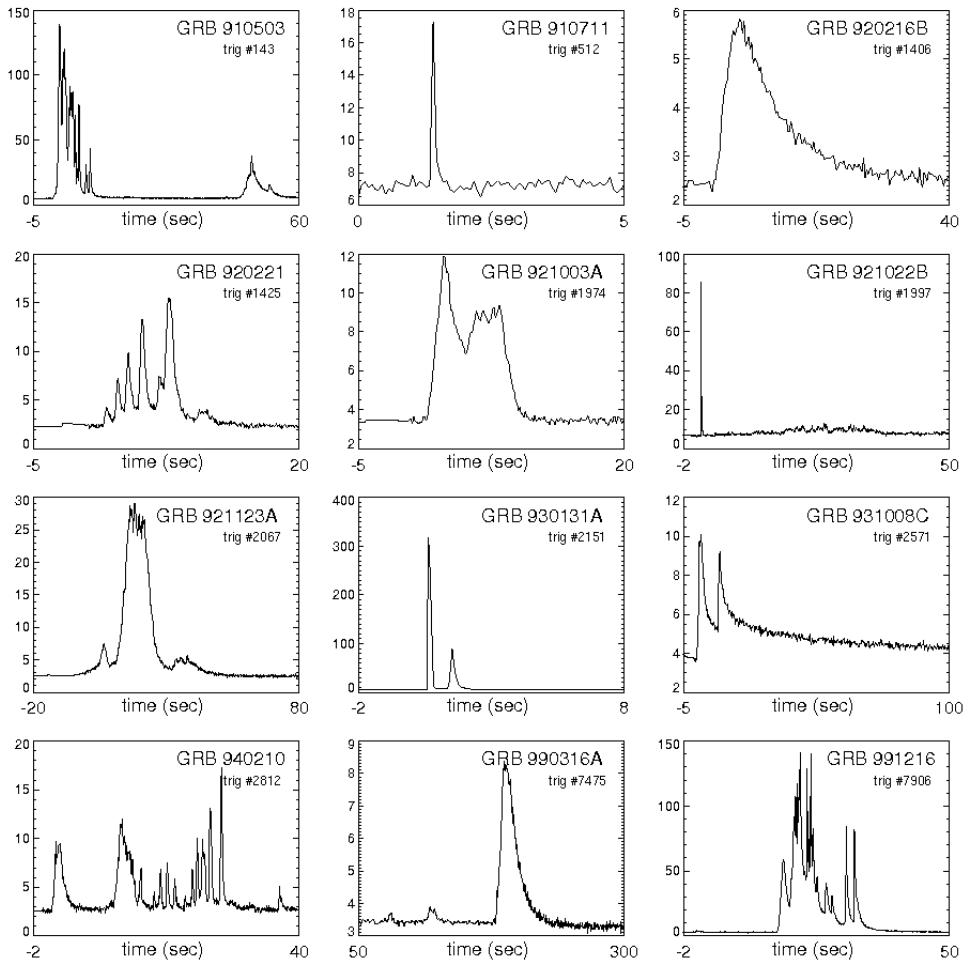


Figure 1.2: The light curves of 12 gamma-ray bursts observed by BATSE. The diversity of the curves is extraordinary. Events with a length from millisecond to a second exist either with smooth or highly variable curves.

On the contrary, short GRBs are thought to originate in old populations. The theory that they originate from the merging of two neutron stars has been recently confirmed by detecting the gravitational waves of such events. This was detected by the LIGO/Virgo collaboration in 2017 [14]. The electromagnetic counterpart of this gravitational wave event was detected as a short GRB by the

Fermi/Gamma-ray Burst Monitor (GBM) and INTEGRAL instruments followed by the observation of the GRB's afterglow and its host galaxy [15]. Short GRBs emit electromagnetic radiation due to the radioactive decay of heavy r-process nuclei that are produced and ejected almost isotropically during the merger process [16].

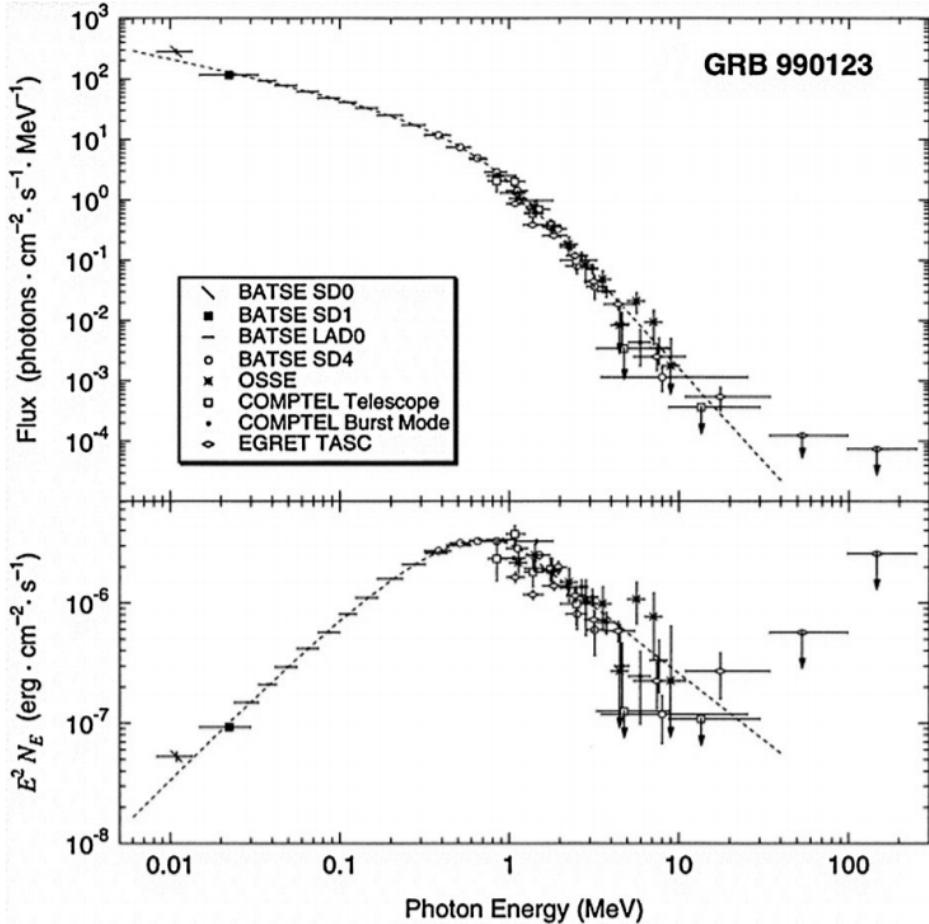


Figure 1.3: The band function of GRB 990123 [45]

## Open questions on gamma-ray bursts

Despite the fact that GRBs have been studied for more than fourty years a lot of questions are still not understood about their physics[17]. In the following the main open questions are listed:

- Classification: some of the observed light curves can not be classified into short and long GRBs, such as the ultra-long GRBs.
- Central engine: in most cases it is thought that a black hole is formed after the merging of neutron stars and core collapse of massive stars. Although there are models that predict the birth of a magnetar. The most exotic theories predict the birth of a quark star that has never yet been observed.

- Acceleration mechanism of the jets: How particles are accelerated to a velocity of a Lorentz-factor of 300 in the shock waves is not yet understood.
- GRB jet composition: there are two possible driving forces behind the acceleration of jet material. The first one is magnetic field dominated and the other one is dominated by baryons. As the acceleration of particles is not yet fully understood, it is not known whether protons can be accelerated to ultra-high energies or not. These could produce Ultra-High Energy Cosmic Rays.
- Radiation mechanism: there are several physical processes that create the  $\gamma$ -ray spectrum of GRBs, that is called the "Band" function. The following three processes are though to be the main contributors to the spectrum: synchrotron radiation, synchrotron self-compton radiation and Compton up-scattering of thermal photons.
- First stars: different population of stars on avarage have different masses. Models predict that population III stars end their life with a black hole that has a mass about ten times bigger than population I and II stars. The progenitors of GRBs can be classified into populations by their redshifts [18].

## Search for the electromagnetic counterparts of gravitational wave events

Multi-messenger astronomy is based on the investigation of astronomical objects with disparate "messengers". It opened a possibility gather information in different ways to understand astrophysical objects. Until 2017 the three messenger were electromagnetic radiation, cosmic rays and neutrinos. 2017 opened a new era in multi-messenger astronomy as the electromagnetic counterpart of GW signal GW170817 was detected by the LIGO/Virgo collaboration. The source of this signal was the short GRB 170817A [19], a binary neutron star merger [15].

This historical detection occurred on August 17, 2017. The Fermi satellite detected the (short) GRB 170817A, approximately 1.7 s after the LIGO-Virgo detector network observed a GW signal GW170817. The origin of this source was a neutron-neutron star merger. The source was localized in a sky region of  $28 \text{ deg}^2$  (with 90 % confidence).

After the detection of the GRB signal, several space and earth based observatories in a wide range of wavelengths started the investigation of the source. The discovery of a bright optical transient linked to the GRB was discovered

in NGC 4993. Observatories involved in the investigation included Chandra X-ray Observatory [21], INTEGRAL [20], Swift and Nuclear Spectroscopic Telescope ARray (NuSTAR). A blue kilonova associated with this event was detected [22]. It is important to emphasize that not only neutron-neutron star mergers but neutron star-black hole and black hole-black hole mergers can emit electromagnetic signal [23]. Therefore all short GRBs can be investigated with multi-messenger astronomy in the future. As LIGO is constantly upgrading its detectors, gravitational waves will be detected more often. An even more sensitive gravitational wave detector is under construction, called Kamioka Gravitational Wave Detector (KAGRA). First scientific measurements will be taken in the 2020s [24] with this device.

## Particle detectors in space

Almost all types of detectors starting from semiconductor based detectors to gas-filled detectors have been used in space. In the case of search for GRBs the detection of x-rays is the most relevant. Several types of detectors can be used for detecting  $\gamma$  photons[27], for example the semiconductor based German Position Sensitive Proportional Counters (PSPC) of the ROSAT X-ray telescope. For our satellite the size of the detector is limited, therefore the best option was to choose scintillators. Mostly used scintillators in space include inorganic crystal scintillators, for example NaI(Tl) or CsI(Tl), and plastic scintillators. CsI(Tl) was chosen as the detector of the Camelot CubeSat due to its high light yield.

## Current missions observing gamma-ray bursts

This subsection provides an overview of the existing missions that are aimed to detect GRBs. Field of view (FOV) and localization accuracy are chosen to compare them with the Camelot CubeSat. The proposed design of the Camelot satellite has the advantage of a large field of view and a high localization accuracy. The main idea is that each CubeSat will detect GRBs at the same time. By measuring the time difference between the triggering of each satellite, the GRB source can be triangulated. By using more satellites, the field of view becomes full-sky and localization accuracy increases.

The planned localization accuracy is more precise than any other existing GRB mission, that might reach  $\sim 10'$ . While providing the highest localization accuracy, the field of view would also be the highest, even larger than the field of view of the INTEGRAL/SPI-ACS and Fermi.

The size and mass of the Camelot CubeSat is less than 1% of the Fermi

telescope. Therefore from a fraction of the cost several satellites could be built and deployed in orbit. Having several satellites also provide a high redundancy.

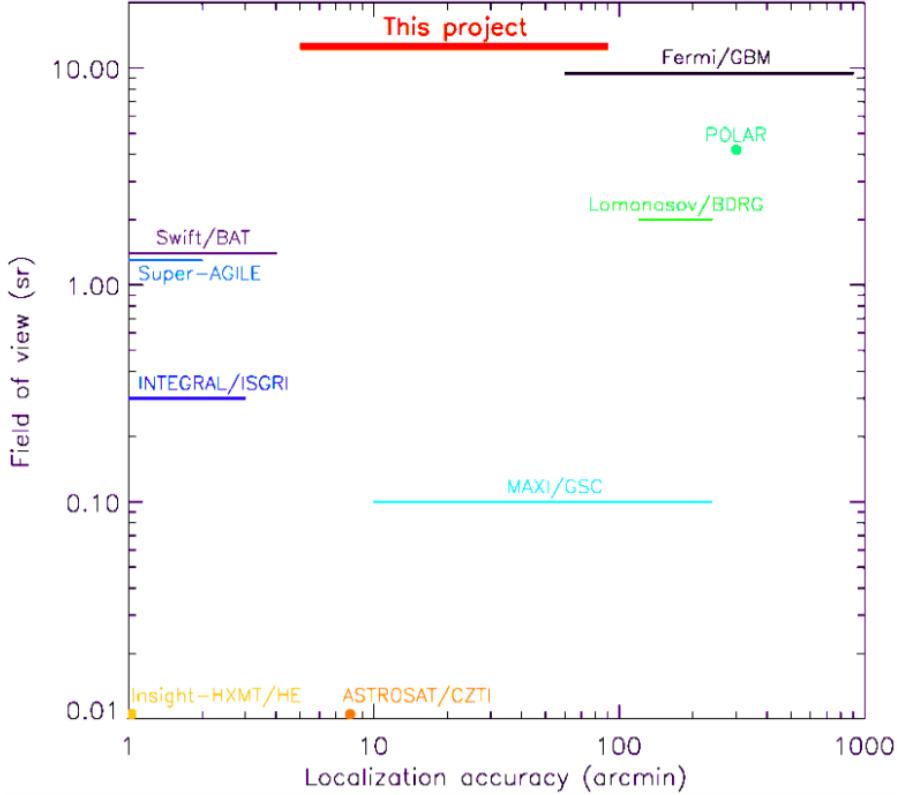


Figure 1.4: The field of view and localization accuracy of several missions including the CAMELOT CubeSat.

The following list includes the most relevant, currently ongoing missions looking for GRBs:

- Fermi 7 [26] was launched in 2008 and its main instruments are the Gamma-ray Burst Monitor and the Large Area Telescope (LAT).
- Neil Gehrels Swift Observatory (Swift) 8 [28] was launched in 2005 with its main instruments, those are the Burst Alert Telescope (BAT), the X-ray Telescope (XRT), and the UV/Optical Telescope (UVOT).
- INTERnational Gamma-Ray Astrophysics Laboratory (INTEGRAL) 9 [29] was launched in 2002. The most succesfull instruments on board include are the Anti-Coincidence Shield of the SPectrometer of INTEGRAL (SPI-ACS) and the Integral Soft Gamma-Ray Imager (ISGRI) detector layer of the "Imager on Board the Integral Satellite (IBIS)" detector.
- Astro-Rivelatore Gamma a Immagini Leggero (AGILE) 11 [30] was launched in 2007 and its Hard X-ray Imaging Detector (Super-AGILE) detected

several GRBs.

- CALorimetric Electron Telescope (CALET) 12 [31] / Gamma-Ray Burst Monitor (CGBM), was launched in 2015 and it is placed on the International Space Station (ISS).
- Monitor of All-sky X-ray Image (MAXI) 13 [32], started nominal observation in 2009. Its instrument Gas Slit Camera (GSC) detects GRBs. This instrument is placed on ISS.
- Gamma-ray Burst Polarimeter POLAR 14 [33], was launched in 2016 and is dedicated to the measurement of  $\gamma$ -rayGRB polarization.
- Lomonosov 15 /BDRG [34], was launched in 2016. The BDRG instrument on board detected several GRBs.
- The Hard X-ray Modulation Telescope (Insight-HXMT) 16 [35] was launched in 2017. It has the high energy X-ray telescope (HE), the medium energy X-ray telescope, and the low energy X-ray telescope.
- ASTROSAT 17 [36] was launched in 2015 and its Cadmium Zinc Telluride Imager (CZTI) has already detected many GRBs.
- Reuven Ramaty High Energy Solar Spectroscopic Imager (RHESSI) 18 [37], was launched in 2002. It is designed to study hard x-ray and gamma-ray emission from solar flares, however it is also efficient instrument to detect non-solar gamma-ray events like GRBs.

## The CAMELOT satellite

The CAMELOT satellite platform is being developed by C3S LLC, in Budapest. There would most likely be four scintillators on the CAMELOT satellite. On two neighbouring sides of the satellite have two scintillators each. The scintillators will have a case either from aluminium or carbon fibre.

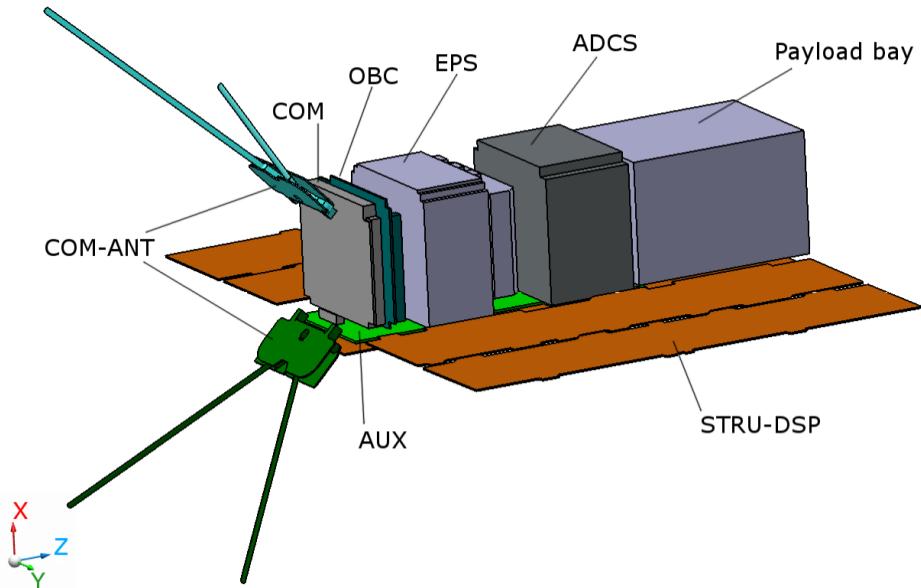


Figure 1.5: The modules of the satellite (Courtesy of C3S LLC) [38]

The CAMELOT satellite will consist of the following modules:

- On-Board Computer [OBC] is responsible for the control of all the satellite's sub-systems. OBC performs housekeeping and collects payload data and archives it in its internal storage.
- Electrical Power System [EPS] is responsible for managing primary and secondary power sources and distributing power at appropriate voltage levels to on-board subsystems.
- COM UHF Transceiver [COM] provides an RF communication link between the Ground Segment and the satellite. The COM codes are received messages from the OBC and are transmitted to the Ground Segment. Furthermore, the COM decodes the received RF messages from the Ground Segment and transmits them to the OBC.
- Attitude Determination and Control System [ADCS] determines and controls the altitude of the satellite.
- Auxiliary Electronics [AUX] subsystems control the deploying mechanisms and monitor the solar arrays. The Aux subsystem also contains the rigid backplane of the satellite; therefore, it also connects the subsystems.
- Structure [STRU]
- Ground Support Equipment [EGSE]

## The detection of GRBs

In the case of our satellite, the detection of  $\gamma$  photons from GRBs will be carried out by a Caesium Iodide scintillator doped with Tellurium. This is the most popular alkali halide scintillation crystal apart from NaI(Tl). The  $\gamma$  photons create scintillation, thus thousands of optical photons in the scintillator. These photons will be detected by Silicon Photomultiplier device called Multi Pixel Photon Counter, MPPC for short. This is by far the best and most compact solution as the size of cubesat standard is limited to be  $\sim 300 \times 75 \times 5$  mm<sup>3</sup>. The first tests were carried out by a single MPPC readout, which was followed by a design with two MPPCs to maximize the light yield and minimize the noise. The size of the Hamamatsu MPPC is 6x6 mm<sup>2</sup>. The scintillators will be placed onto the surface of the satellite as can be seen in fig. 1.6. While fig. ?? shows the experimental setup used for the tests.

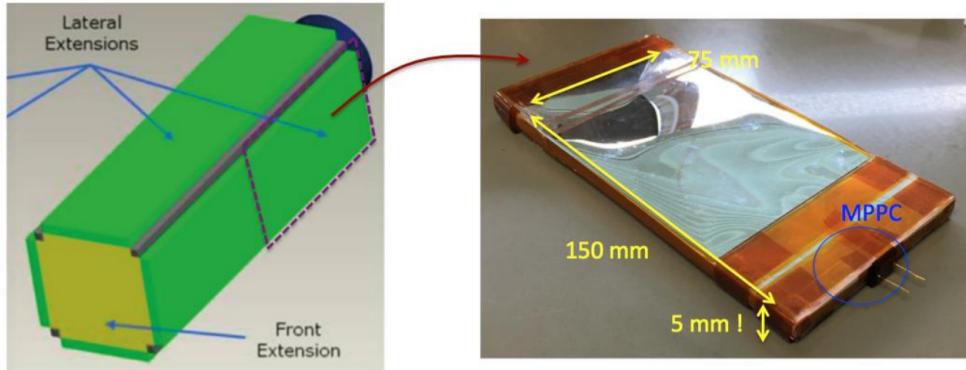


Figure 1.6: Side extensions including the CAMELOT satellite (left, Courtesy of C3S LLC) where scintillators are placed and a wrapped scintillator with the MPPC read out (right) [39]

The large area and small thickness of the scintillator makes it challenging for the read out as optical photons have to be reflected several times to reach the MPPC. Two factors play role here, the self absorption of the reflectivity of the tape that is used to cover the scintillator. The length of the self absorption is in the order of a meter and the reflectivity of the ESR tape, which is above 99 % can be seen in fig. 1.10. The idea that we came up is using two MPPC-s with coincidence. This results in a smaller leakage current (noise) as the noise has a poissonian statistic. Therefore the noise is "random" and the chance of a trigger from both of the MPPC has a very low chance. The other advantage is the increased photon yield due to the double of the photon collection area. The difference between one and two channel readout can be seen in fig. 1.7.

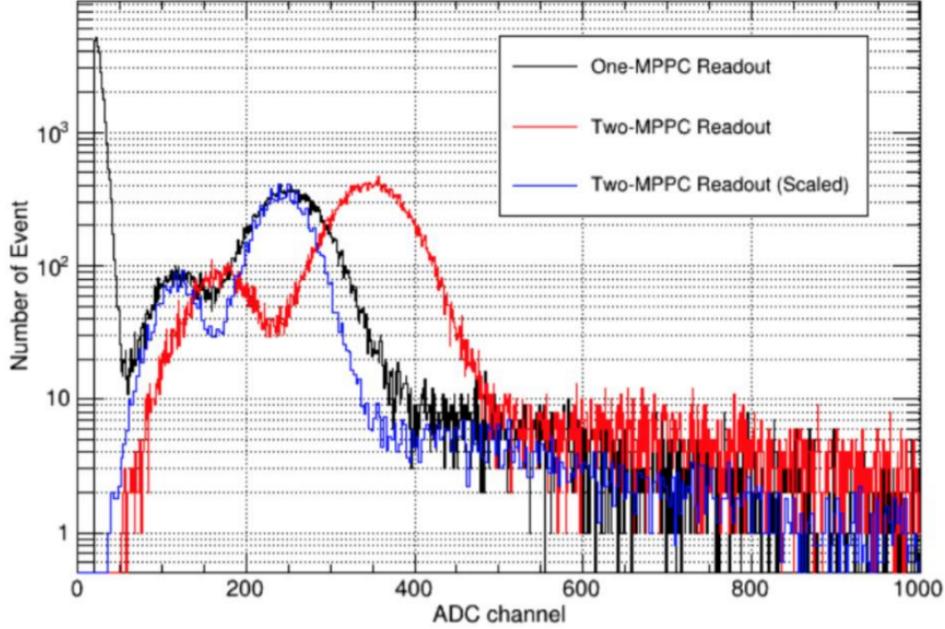


Figure 1.7: Energy spectra obtained by collecting the signal of 1 million  $\gamma$  with the one channel and the two channel read out [39]

## The Geant4 platform

The most used software in particle physics are MCNPX, FLUKA and Geant4. For this thesis Geant4 was chosen as it is the best for simulation of several types of particles at the same time and building up the geometry of detectors is most probably is the most convinient in this software.

Geant4 [46, 47, 48] (for GEometry ANd Tracking) is a platform for "the simulation of the passage of particles through matter," using the Monte Carlo method. It has been developed at CERN for more than three decades by dozens of physicist and programmers. The cross sections for all interactions between matter and particles are included in this software. It is an object orientated programming platform in C++. Its development, maintenance and user support are taken care by the international Geant4 Collaboration. Areas in science that has used Geant4 are vary widely from the design of complex medical machines to accelerator, nuclear physics, high energy physics and space applications. Several thousand people use it all around the globe.

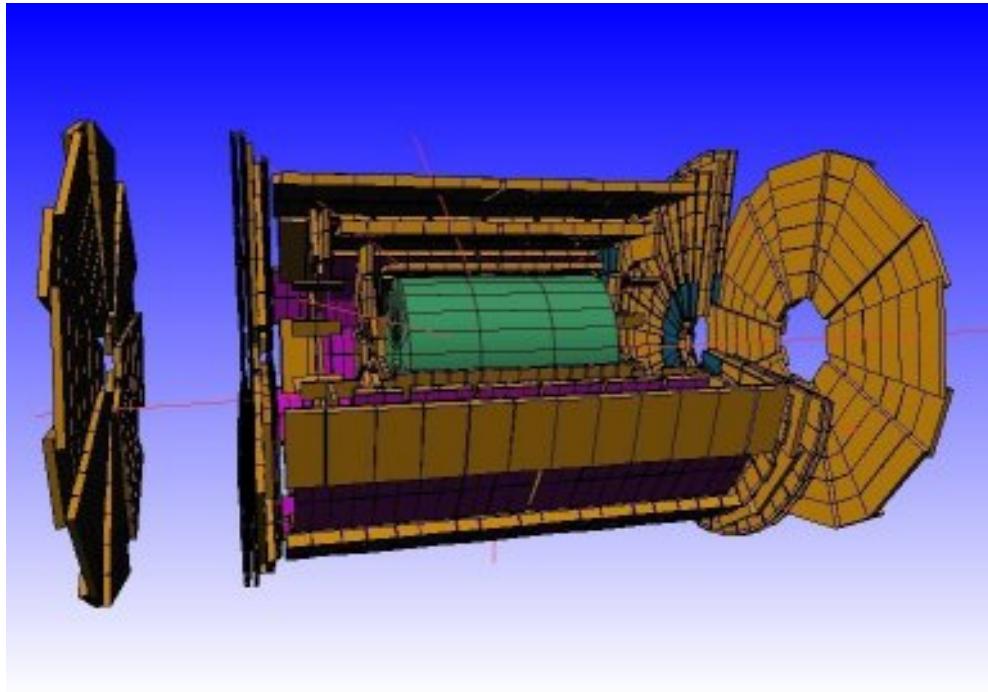


Figure 1.8: The Geant4 simulation of the whole ATLAS experiment

The Geant4 platform includes all the important features required in the full simulation of an experiment from handling the geometry of the detector and readout, tracking of particles in the detectors for visualization.

## The experimental setup

In order to understand and characterize the behaviour of the large-area CsI(Tl) scintillator detector and the MPPC readout; an experimental setup was built in Hiroshima, Japan. The experimental setup provided vital information for the simulation, mostly for the position dependence of the light yield. Different  $\gamma$  source were used, mostly an  $^{241}\text{Am}$  source.

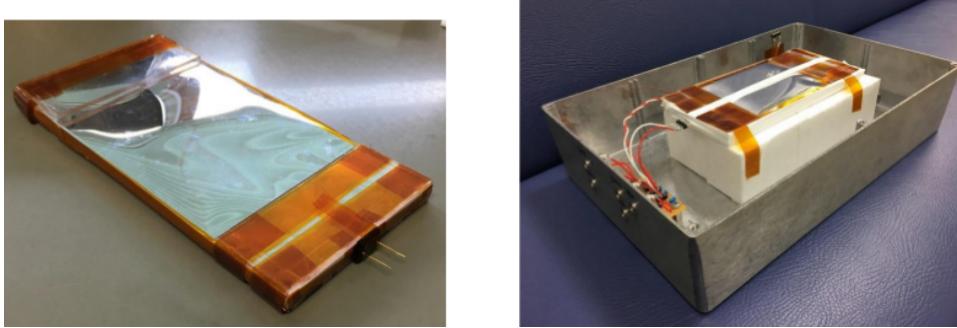


Figure 1.9: The scintillator on the left and the cage holding the setup on the right [39]

The detector was wrapped in a so called ESR foil that is designed to reflect as much light as possible. In fig. 1.10 the reflectivity of this tape can be seen

depending on the wavelength of the photons, exceeding 99 % for most of the relevant spectrum. It is important as the optical photons have to be reflected several times in order to be read out with the small effective area of the MPPC. This reflectivity was included in the simulation.

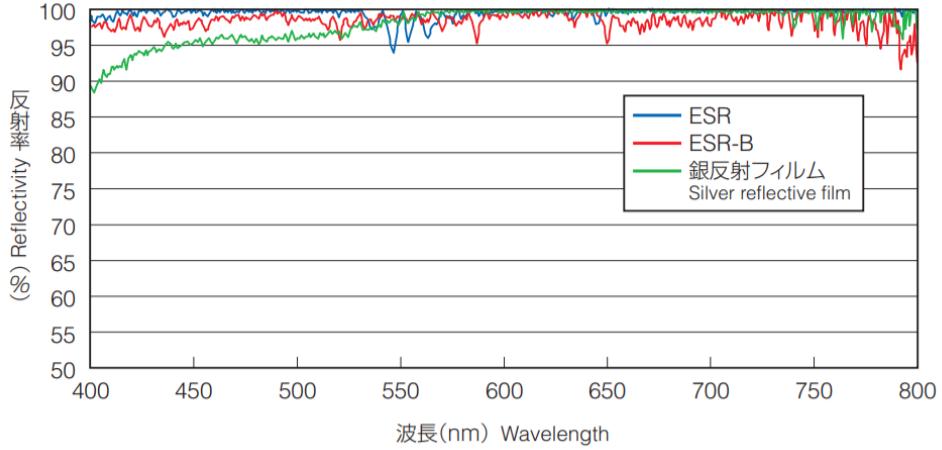


Figure 1.10: The reflectivity [52] of the ESR tape that was used to wrap the scintillator in order to increase the light yield. This data was included in the simulation.

The CsI(Tl) scintillator used is produced by AmCrys [52]. The dimension of this detector are 150 mm x 75 mm x 5 mm, the largest possible size that can fit onto the surface of the satellite. The spectra of the optical photons produced in the scintillation process can be seen in fig. 1.11. An aluminium case of 2mm is used, the same as planned for the mission in space. The time constant for the scintillation of the detector was 200 ns.

Energy [eV]	3.54	3.10	2.76	2.48	2.36	2.25	2.07	1.91
Relative emittance	0.02	0.1	0.3	0.6	0.9	1.0	0.7	0.4
Refractive index	1.59	1.57	1.54	1.54	1.54	1.54	1.54	1.54
Absorption length [cm]	50	50	50	50	50	50	50	50

Table 1: The parameters of the CsI(Tl) scintillator used in our experiment

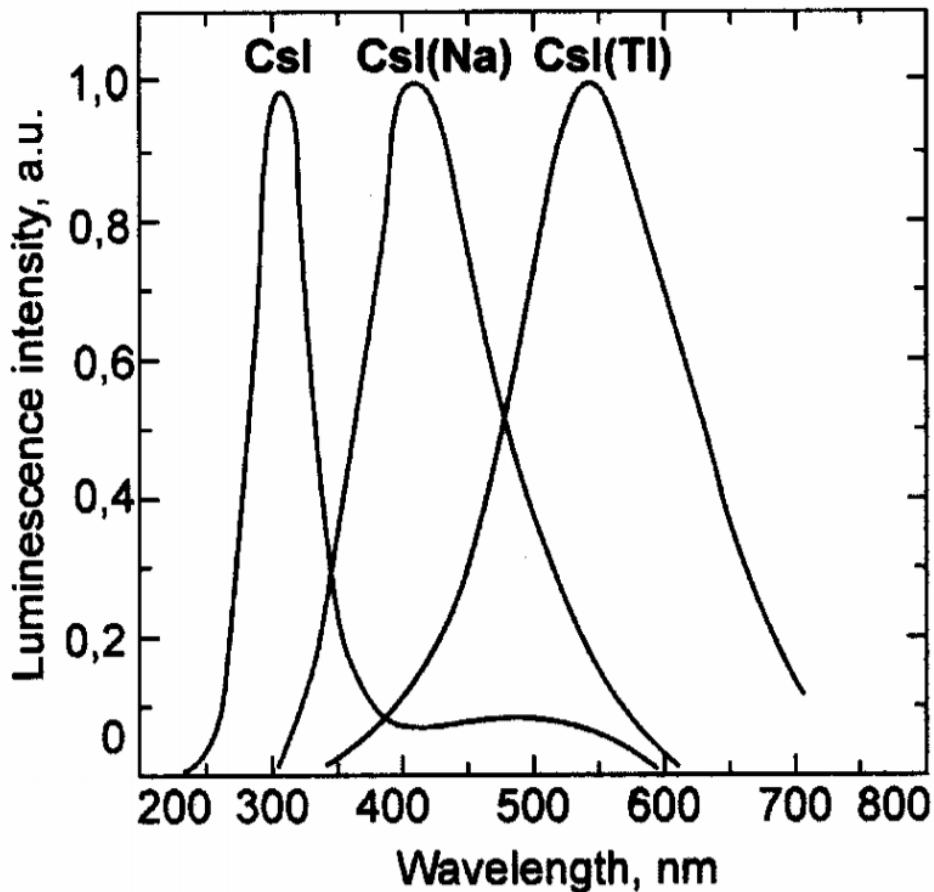


Figure 1.11: The emitted scintillation light spectra of three types of scintillators. In our experiment CsI(Tl) was used

The MPPC chosen for this satellite is one of the latest models (HAMAMATSU S13360-6050CS) [53]. It has an effective area of 6mm x 6mm. The photon detection efficiency of this specific model can be seen in fig. 1.12.

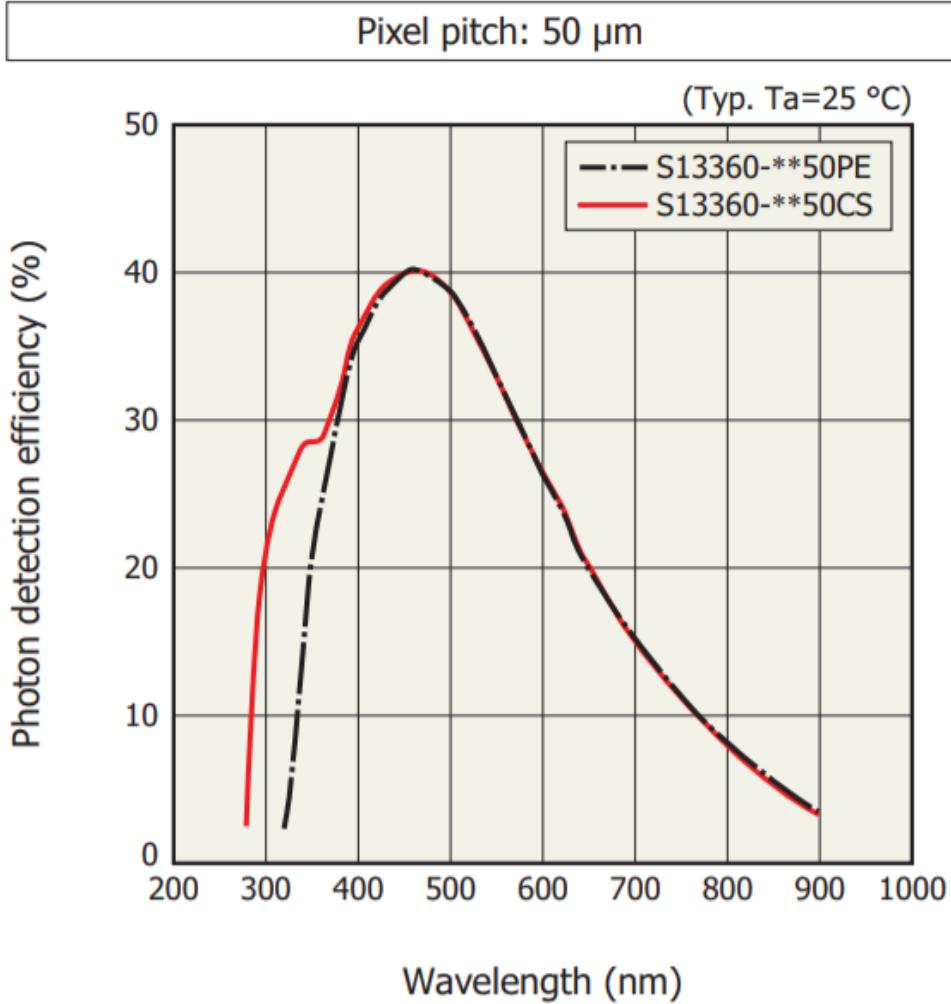


Figure 1.12: The photon detection efficiency [53] of the MPPCs used for the readout of the scintillators. This data was included in the simulation.

## The readout of the detector

The response of the MPPC detector used depends on temperature. Therefore the experimental setup was placed in a thermostatically controlled chamber (ESPEC LU-113) keeping the temperature at 25°. Two setups were investigated. The first one with a single MPPC readout placed in the middle of one side of the scintillator. In the other case two MPPCs were utilized symmetrically placed on one side of the detector.

For the single MPPC readout setup a commercial charge-sensitive preamplifier (CLEAR-PULSE 5028), a shaping amplifier (EG&G ORTEC 571), and an ADC (AMPTEC MCA800A) were used to obtain an electric signal read out by a PC. The shaping time was set to 1  $\mu$ s.

For the case of the two-channel read out an FADC board was utilized that contains all the necessary electronics to read out the signal of the MPPC with an additional coincidence unit. This board was developed for the astrophysical

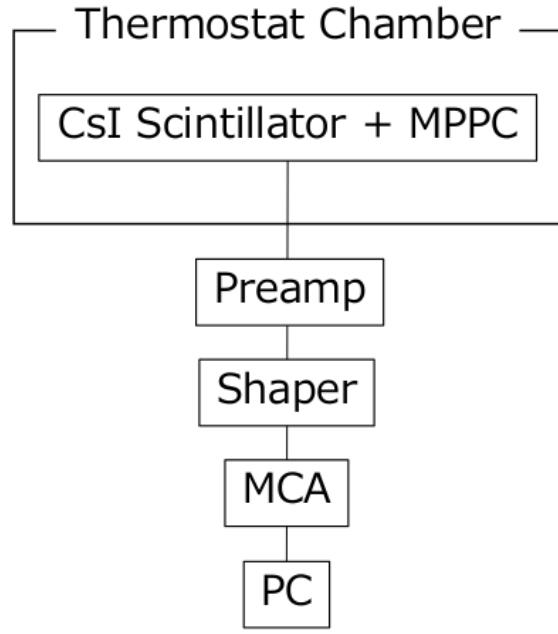


Figure 1.13: The schematic design of the readout of the MPPC for the 1 channel configuration [39]

polarimetry mission PoGO+. The shaping time was set to  $2.2 \mu\text{s}$  in this case. The breakdown voltages of the two-MPPCs are almost the same— $50.82$  and  $50.87$  V—at  $25$  C. We used a KEITHLEY 2400 to supply the operational voltage of  $53.4$  V.

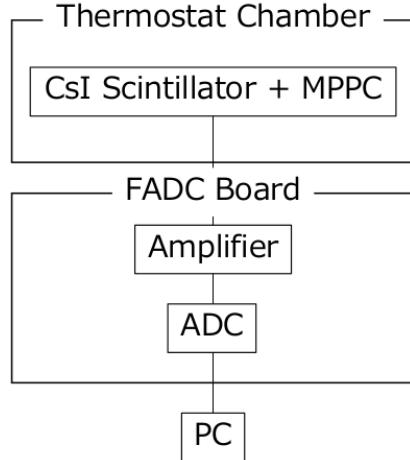


Figure 1.14: The schematic design of the readout electronics of the 2 channel setup [39]

## 1 channel setup

The  $\gamma$  source that was used to test the experimental setup was chosen to be an  $^{241}Am$ . The reason for this is that the peak of energy spectrum of the  $\gamma$  photons from GRBs is at 50 keV which is very close to the peak of the  $^{241}Am$  that is 59.5 keV. The activity of the source at the time of the experiment was 471 kBq. The source was collimated to a beam that hit the surface of the scintillator on a circle with a radius of 10 mm. The distance of the source from the scintillator was 13.5 mm.

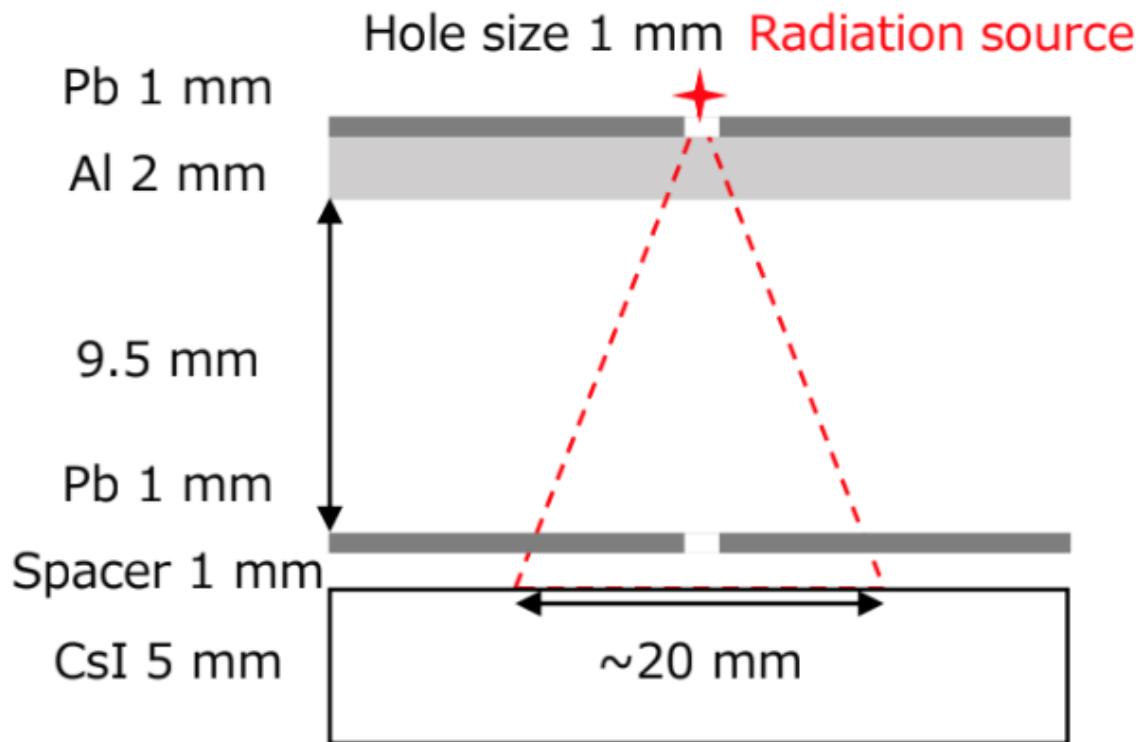


Figure 1.15: The  $^{241}Am$  *gamma* source was placed in a collimator system as can be seen on the image [39]

Therefore the solid angle of the part of the whole sphere where  $\gamma$ s could be detected was

$$\Theta = 2\pi(1 - \cos(\theta)) = 1.234\text{sr}$$

## 2 channel read out setup

In order to maximize the light yield, thus the signal of the detector and minimize the noise with coincidence a 2 channel setup was tested. In fig. 1.16 this setup can be seen with the MPPCs placed symmetrically to the scintillator.

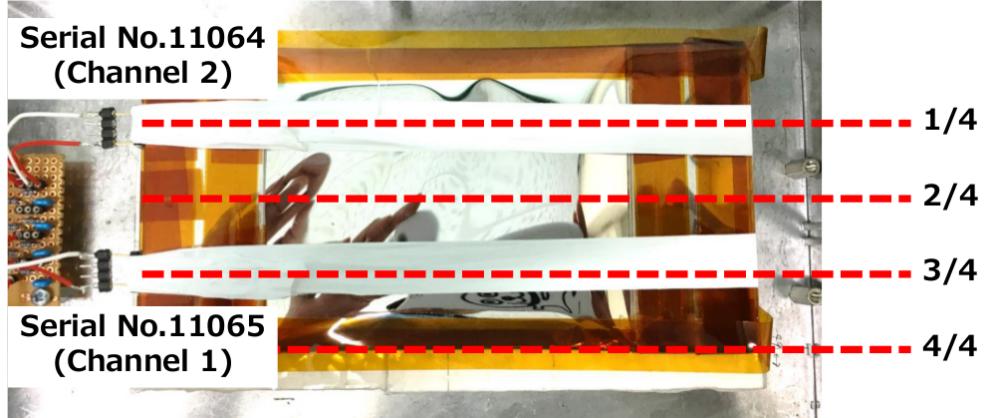


Figure 1.16: The MPPCs for the 2 channel readout are positioned symmetrically from the center of the scintillator [39]

In order to simulate the satellite with the two channel read out setup, first this configuration was tested. The simulation of this setup can be seen in fig. 1.17. The  $\gamma$  source was collimated with a lead sheet that was drilled. The optical photons that were detected are shown only. The two read volumes correspond to the MPPCs.

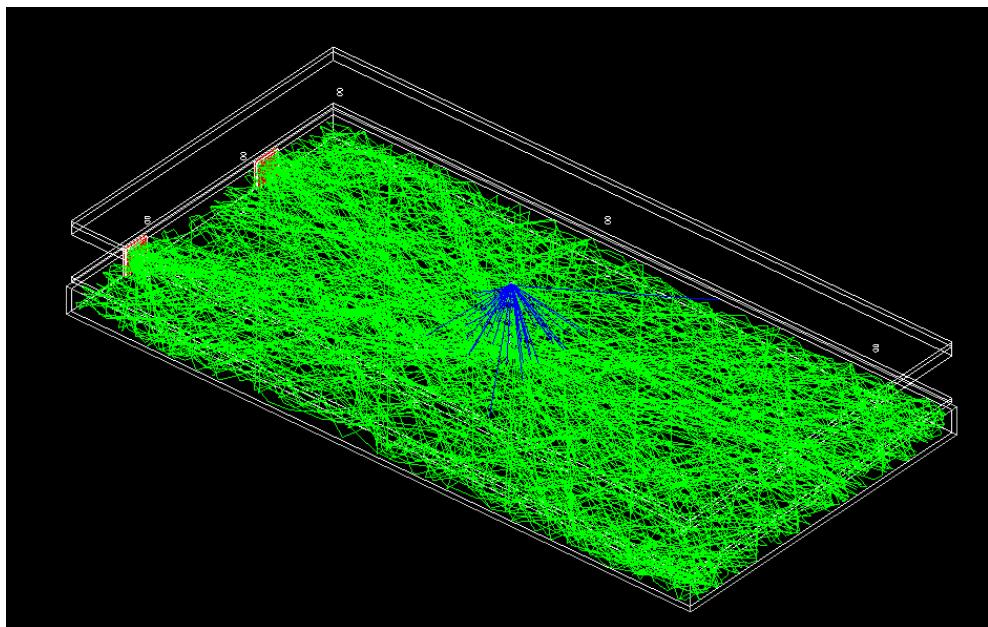


Figure 1.17: Simulation of 100  $\gamma$ s emitted from the source. The blue lines represent the track of the  $\gamma$ s and the green lines represent the track of the optical photons created by scintillation. Only the photons that were detected are drawn.

## The satellite

The direction of GRBs is isotropic in the sky as they are extragalactic objects. As the positioning of the scintillators is not decided yet it is important to understand which configuration would be the best for detecting as many GRBs as possible. Therefore it is vital to understand how the material of the satellite itself absorbs  $\gamma$ s. The satellite consists of several modules with specific shapes and material contents. Therefore building up the whole geometry and material composition was not possible in Geant4, thus the model of the satellite had to be imported into Geant4.

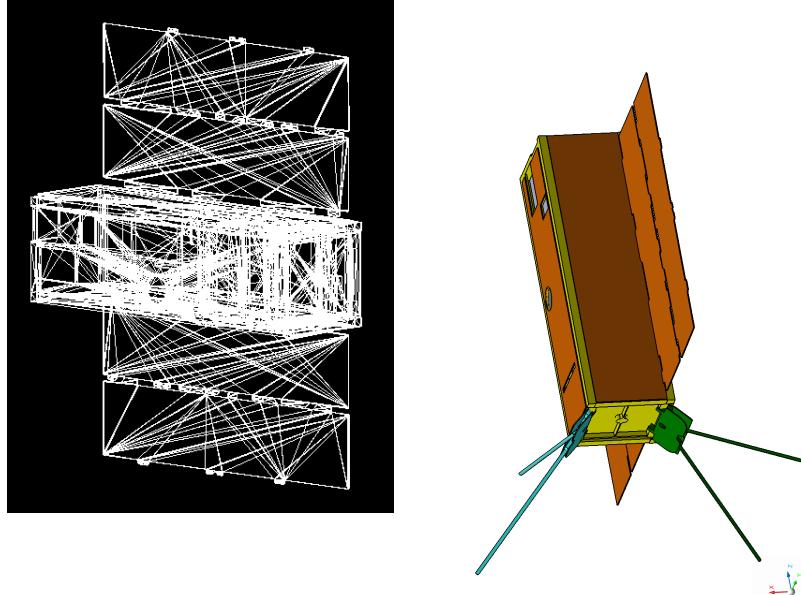


Figure 1.18: The CAD model of the satellite [38] (on the right, Courtesy of C3S LLC) and the view of the imported model in Geant (on the left)

Importing predefined CAD models into GEANT4 is not always possible or requires intermediate file format conversion to Geometry Description Markup Language (GDML) using commercial or third party software. CADMesh [49] is a direct CAD model import interface for GEANT4 leveraging ASSIMP [50] for reading the CAD files. In this work the CAD model of the satellite was exported to STL files. These files were imported by the CADMesh software to Geant4. The tessellation was done by the TETGEN [51] software.

Name of module	mass [g]	Type of material	Mass ratio [%]
ADCS	710	Aluminum 6061-T6	50
		Copper Electric	25
		Glass Borosilicate	25
COM	90	Stainless Steel	2
		Brass Generic	25
		Aluminum 7075-T73	40
		FR4 Glass-Epoxy sheet	33
EPS	750	FR4 Glass-Epoxy sheet	25
		Aluminum 6061-T6	75
OBC	50	FR4 Glass-Epoxy sheet	100
STRU	980	Aluminum 6061-T6	100
SP	570	Solar Panel	100
Payload	100	Aluminum 7075-T73	100

Table 2: The mass ratio of materials that are used for the satellite (Courtesy of C3S LLC) [38]

The material composition of the satellite can not be exported to Geant4 from CAD files directly. Therefore an average material composition was calculated for all the modules. In table 2 the material composition of the 7 modules are listed. These materials, mostly alloys consist of several elements. The Geant4 cross section database is built up for elements. Therefore these materials had to be broken down to elements.

Material name											
Aluminum 6061-T6	Al	96.90	Mg	1.20	Si	0.80	Fe	0.70	Cu	0.40	
Aluminum 7075-T73	Al	88.60	Zn	6.10	Mg	2.90	Cu	2.00	Si	0.40	
Stainless Steel	Fe	66.50	Cr	20.00	Ni	10.50	Mn	2.00	Si	1.00	
Copper Electric	Cu	100.00									
Glass Borosilicate	Si	42.10	O	54.80	B	3.10					
FR4 Glass-Epoxy	Si	23.39	O	36.02	C	37.04	H	3.55			
Brass Generic	Cu	85.00	Zn	15.00							
Solar Panel	Ge	38.00	Si	24.00	O	20.00	C	13.00	H	4.00	B
											1.00

Table 3: The chemical composition of materials in mass fraction that are used for the satellite (Courtesy of C3S LLC) [38]

In fig. 1.13 the simulation of the satellite and the two channel read out setup can be seen. 100  $\gamma$  particles were used to shoot the satellite with a beam parallel to the plane of the scintillator. A scattered  $\gamma$  particle can be seen being scattered into the scintillator and being detected.

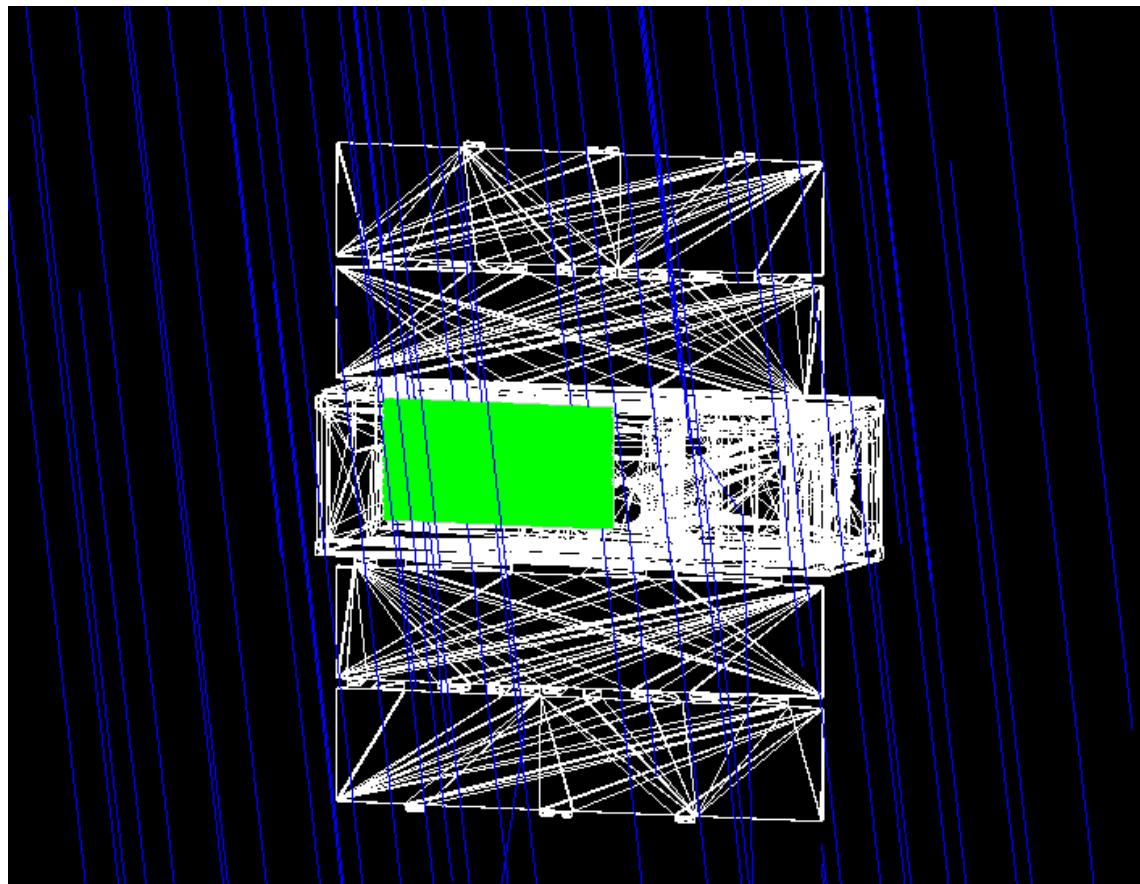


Figure 1.19: The satellite radiated by a parallel beam of 100  $\gamma$  particles. The scattered  $\gamma$ s induce signal in the detector.

## Results of the simulation

---

### X-ray fluorescence

The measured spectra of the scintillation light created by  $\gamma$ -rays (fig. ??) has a main peak that corresponds to the 60 keV  $\gamma$ s from the  $^{241}\text{Am}$  source. There is an additional peak with an amplitude of roughly 20 % of the main peak. The physical process that is behind this is the X-ray fluorescence. By default this process is not included in the Geant4 simulations, thus I had to turn it on. A histogram with and without x-ray fluorescence can be seen in fig. 2.1

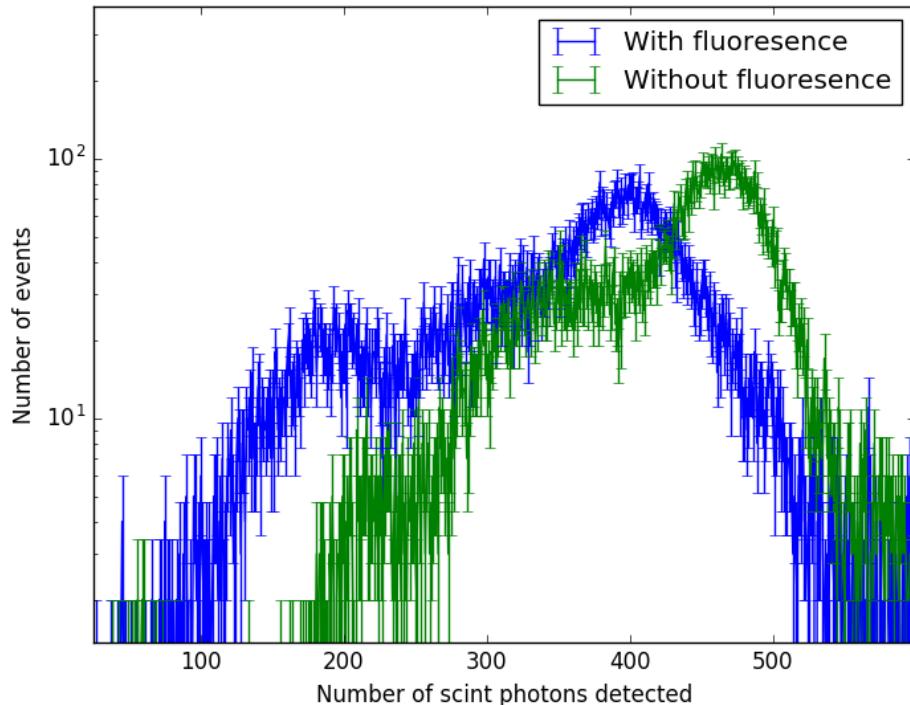


Figure 2.1: The obtained energy spectra from simulation with and without x-ray fluorescence being turned on. One can see that, the first peak, which can be seen in the experiments only appear when x-ray fluorescence was turned on.

## Calibration of the position dependence

The main aim of this thesis is to predict what signal the satellite would measure in space. In order to fine tune the parameters of the simulation experimental data was used. The light yield from different parts of the detector is different as scintillation photons have to travel a different distance and they need to be reflected back from the ESR tape that is wrapped around the detector.

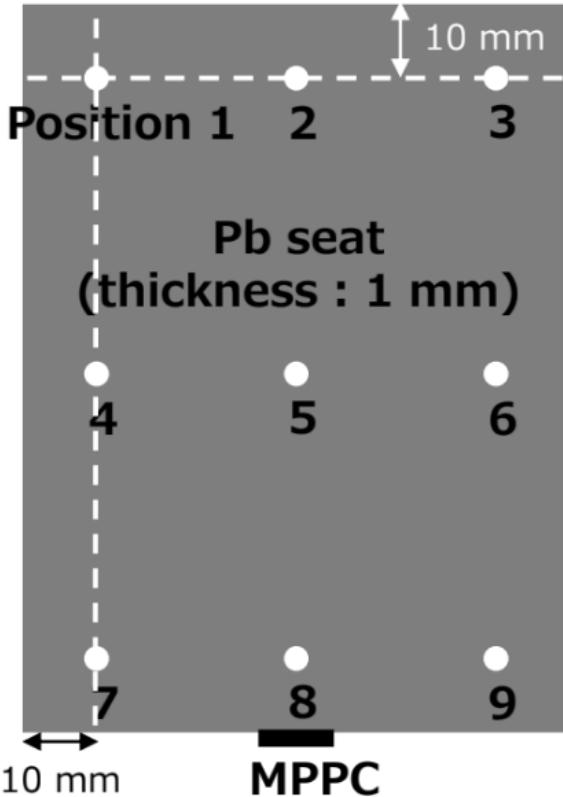


Figure 2.2: The positions of irradiation for the investigation of the response of the scintillator [39]

A lead sheet was drilled with nine holes to collimate the  $\gamma$  source to certain parts of the scintillator. These measurements were carried out by Kento Torigoe at the University of Hiroshima. This configuration with a lead sheet was built up in Geant4 (fig. 1.15 and 3.6 million  $\gamma$ s were simulated at each position).

Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak	0.642	0.664		70.7	0.743		0.598		

Table 4: The position dependence of the photon yield measured at the University of Hiroshima by Kento Torigoe

The three most relevant parameters of the setup in this case are the scintillation photon yield of the scintillator, the reflection rate of the ESR tape and the

self-absorption of the scintillator. The photon yield was fixed to  $540 \text{ keV}^{-1}$  taken from the data sheet of the scintillator [52]. The self-absorption and the reflection rate was increased until the position dependence was the same as in the experiment (table. 5).

Refl:	0.995	Abs.: 50 cm							
Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak	0.4013	0.3917	0.3981	0.4045	0.4172	0.4140	0.2580	1	0.2739
Refl:	0.997	Abs.: 60 cm							
Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak	0.4588	0.4587	0.4697	0.4734	0.4700	0.4737	0.3388	1	0.3365
Refl:	0.999	Abs.: 80 cm							
Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak								1	
Refl:	0.9996	Abs.: 80 cm							
Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak								1	
Refl:	0.9999	Abs.: 105 cm							
Pos. of source	1	2	3	4	5	6	7	8	9
Pos. of main peak	0.6911	0.6865	0.6956	0.7002	0.6956	0.7002	0.6041	1	0.5955

Table 5: The position of the Gaussian peak fitted to the 60 keV peak of the Am source for several irradiated positions. The last set of parameters fit the experimental data.

### Calibration of the photon yield

In the simulation 3.6 million  $\gamma$ s were simulated. The measurement took 240 s. Therefore the number of photons emitted by the source was  $\sim 11.0$  million. The amplitude of the peaks in the energy spectrum of signal obtained by the MPPC depends mostly on three parameters. The absorption length of the scintillation light in the scintillator, the reflectivity of the surface and the scintillation photon yield of the scintillator. In order to obtain the histogram that we would measure by the MPPC, the histogram obtained by the simulation had to be scaled up both in amplitude and energy. (The introduction of an offset was not needed as the origin point of the histogram obtained in the measurements was shifted to zero energy.)

The energy bins had to be scaled up by a factor of 2.4 to match the experimental results. The amplitude had to be scaled up by a factor of 26.2.

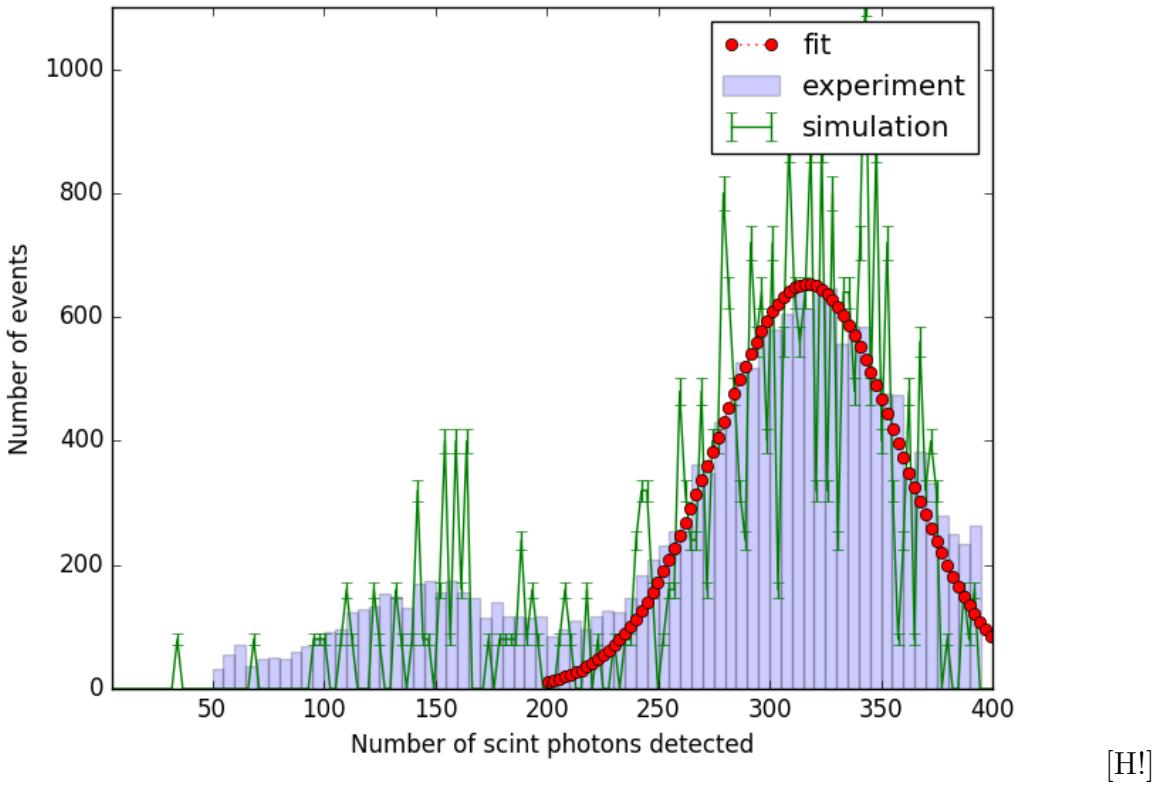


Figure 2.3: The spectra of 11 million photons taken experimentally [39] (blue histogram) matched up with the results of the simulation (green histogram) and the Gaussian fitted to the simulated data

### Simulation of the signal induced by GRBs in the satellite

One of the key questions is what would the satellite measure in space when the  $\gamma$  photons of the GRB hit it. It is important to understand how the material of the satellite would absorb the  $\gamma$  photons.

In order to quantify the detection, the number of particles detected was used as a measure. 10,000  $\gamma$  particles were shot at the detector parallelly on an area of 60cm x 60cm from several angles around the longitudinal axis of the satellite. This way the self absorption of the satellite was quantified.

The energy spectra of  $\gamma$  photons (called band function) from GRBs is fitted by the following equation [55]:

$$f_{BAND}(E) = A \begin{cases} \left( \frac{E}{100 \text{ keV}} \right)^\alpha \exp \left[ -\frac{(\alpha+2)E}{E_{peak}} \right] & E \geq \frac{(\alpha-\beta) E_{peak}}{\alpha+2} \\ \left( \frac{E}{100 \text{ keV}} \right)^\beta \exp(\beta - \alpha) \left[ \frac{(\alpha-\beta)E_{peak}}{100 \text{ keV} (\alpha+2)} \right]^{\alpha-\beta} & E < \frac{(\alpha-\beta) E_{peak}}{\alpha+2} \end{cases}$$

where the parameters of the fit are following for GRB 990123:

$$\alpha = -0.87 \pm 0.01$$

$$\beta = -2.9 \pm 0.11$$

$$E_{peak} = (617 \pm 7.09)keV$$

$$A = (0.0262 \pm 0.0000993)s^{-1}cm^{-1}keV^{-1}$$

Therefore the total flux from GRB 990123 at peak intensity was 9,866 photons for every  $cm^2$  each second. The illuminated area in the simulation was 3600  $cm^2$ . Thus the simulated 10,000 photons correspond to 282  $\mu s$ . Therefore the number of particles detected in the simulation had to be multiplied by 3551.76 to achieve the countnumber for a second.

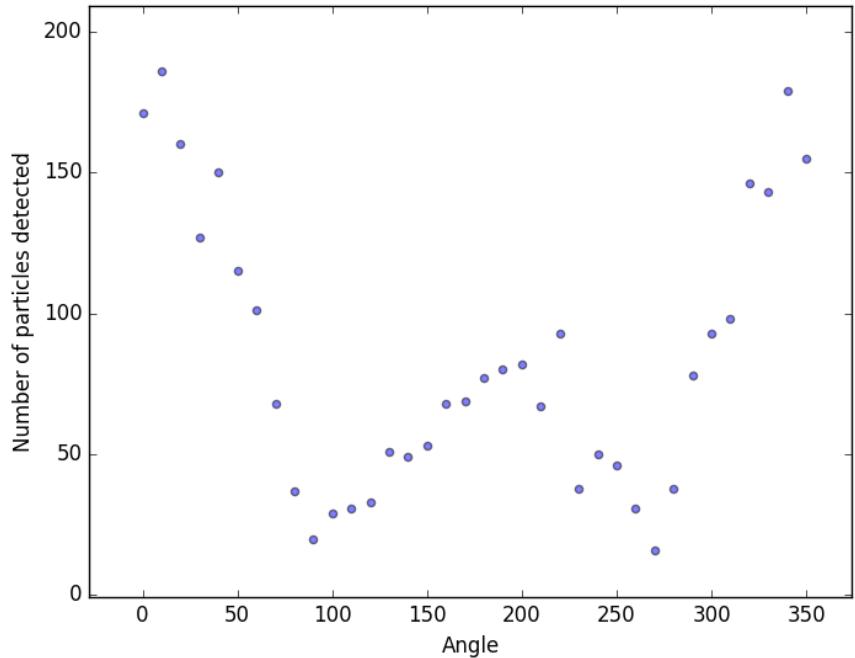


Figure 2.4: The number of detections from the 10,000  $\gamma$  shoot in the simulation depending on the position of the GRB with respect to the longitudinal axis of the satellite.

As one expects the most  $\gamma$ s ( $\sim 200$ ) are detected when these particles hit the scintillator directly, with a perpendicular direction to the surface of the satellite. The effective area of the scintillator is the largest in this case and the  $\gamma$ s are not absorbed by the material of the satellite. The detection rate would be  $6,393,168 \frac{Hz}{cm^2}$  at peak flux.

The worst case scenario is then the  $\gamma$ s hit the satellite with a direction parallel with the plane of the scintillator. Although some detections ( $\sim 20$ ) occur even in this case because  $\gamma$ s that scatter in the material of the satellite can be detected. In

fig. 2.4 one can see the number of detections from 10,000  $\gamma$ s shot at the detector for a given position of the GRB.

In fig. 2.5 one can see the comparison between the energy spectrum measured when the  $\gamma$ s hit the scintillator with a perpendicular direction and when they hit the satellite from behind. The difference between these two scenarios is that  $\gamma$ s had to travel through the material of the satellite in the second case. About 60 % of gammas are lost do the absorption of the satellite.

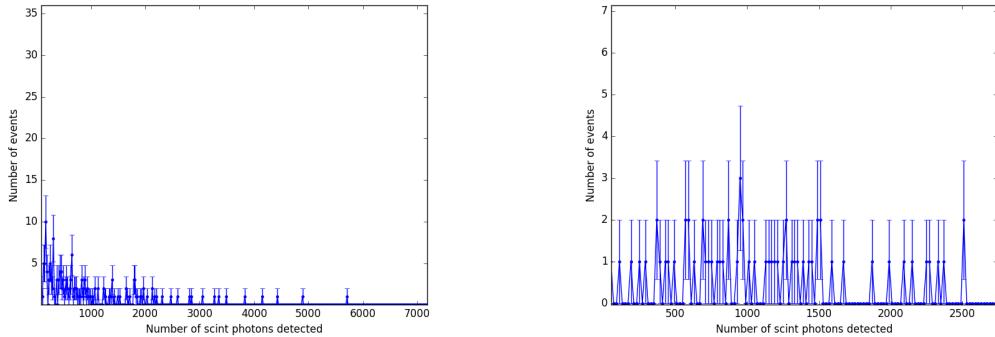


Figure 2.5: The measured spectrum when the  $\gamma$ s had to pass through the material of the satellite (right) and when they hit the scintillator directly (left)

## Simulation of the cosmic background in space

In order to understand what would give the measured background of the satellites, cosmic and solar electrons were simulated in the same way as the  $\gamma$  particles were in sec. 2.1. The flux of and energy distribution of protons and electrons vary widely in the lower orbits around Earth due to its magnetic field. The SPENVIS [56] information system was used to obtain the proton and electron energy spectra.

SPENVIS is ESA's SPace ENVironment Information System, a WWW interface to models of the space environment and its effects; including cosmic rays, natural radiation belts, solar energetic particles, plasmas, gases, and "micro-particles".

## Electron background

The AE-8 model was used in SPENVIS to obtain the spatially averaged electron energy spectra for an altitude of 500 km both for solar minimum and maximum. The inclination of the orbit was chosen to be  $89^\circ$ .

The integrated and differential spectra of electrons can be seen in fig. 2.6. For the simulation the worst case scenario, the spectrum at solar maximum was

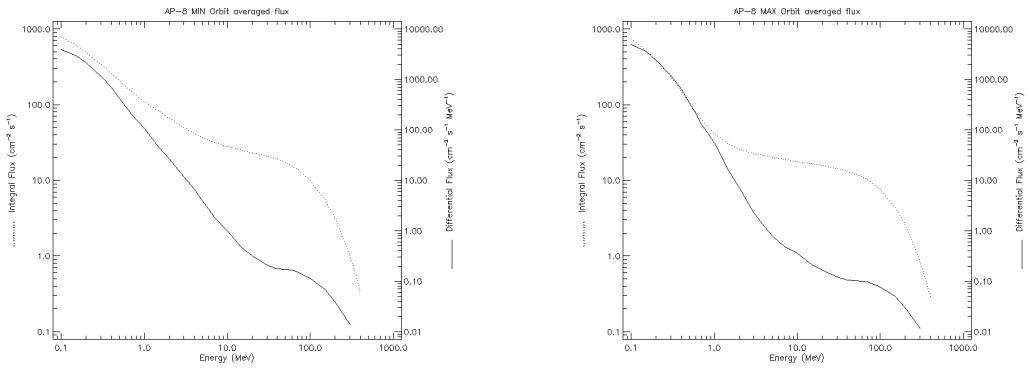


Figure 2.6: The spatially avaraged band function (differential flux for a given energy band, solid line) and the integrated band function (dotted line) of protons at an altitude of 500 km for solar minimum (left) and solar maximum (right)

utilized. Fig. 2.7 shows the energy spectrum for the simulation of  $0^\circ$  when electrons hit the scintillator with a perpendicular direction.

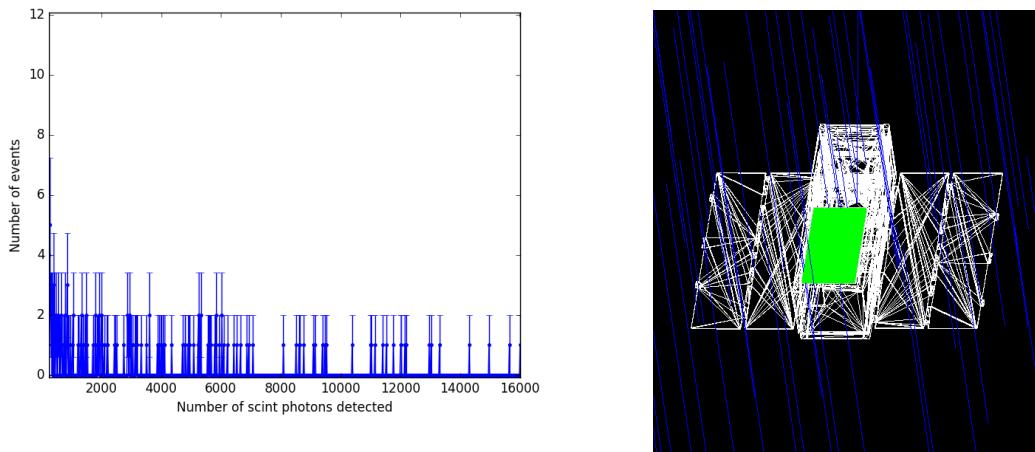


Figure 2.7: The spectrum measured when the electrons were shot perpendicular to the scintillator (called angle of  $0^\circ$ )

The total flux from the spectrum taken from SPENVIS (fig. 2.6) for 1 s and a  $\text{cm}^2$  was 1,226,000. Therefore in order to obtain the count rate, the values obtained for the simulation of 1,000,000 electron with different directions have to be averaged and multiplied by  $1.226 \cdot 3600$ , which yields 230,478 counts each second.

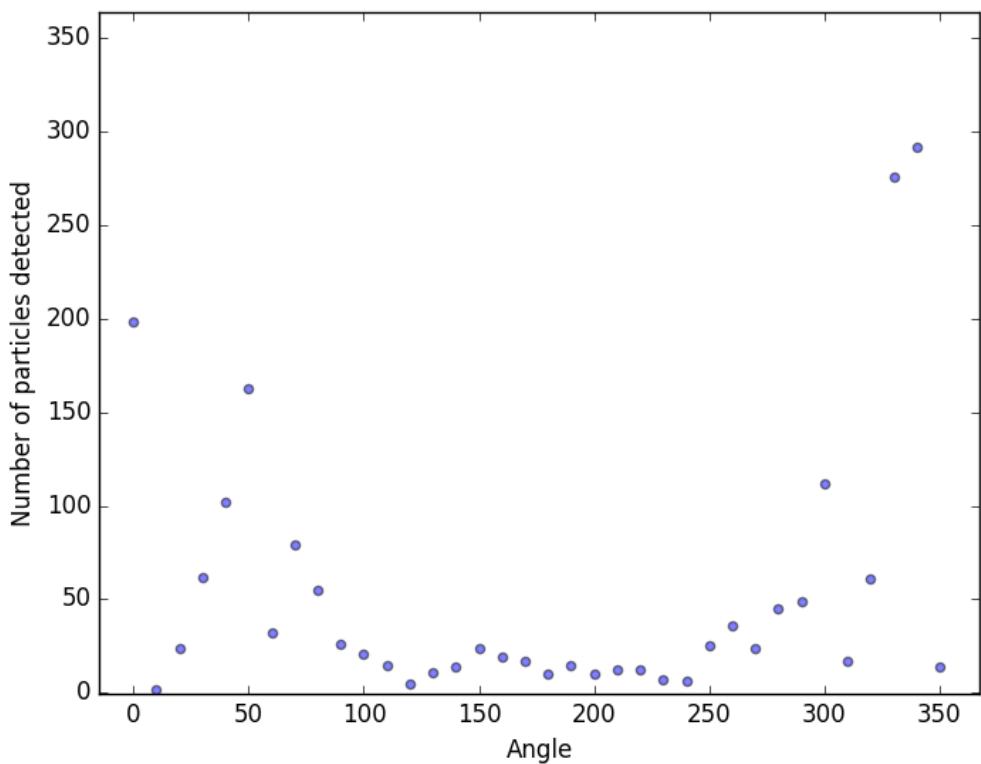


Figure 2.8: The amount of electrons detected from 1,000,000 shoot in the simulation

## Conclusion

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A Geant4 Monte-Carlo based simulation was developed in order to understand how the CAMELOT CubeSat would detect  $\gamma$ -rays originating from short GRBs. The effect of solar and cosmic electrons on the satellite was also investigated with the simulation.

As the first step, the experimental setup that was used to test the CsI(Tl) scintillator – the particle detector of the satellite – was implemented in Geant4. The optical parameters of the simulation were fine tuned by comparing the light yield of the MPPCs in the simulation with experimental data. After the calibration, the results of the simulation was capable of predicting the signal that would be measured in space.

Secondly, the complex CAD model of the satellite –consisting of 7 modules, each with a given avarage material composition– was exported to Geant4. The scintillator and its read out was placed on the side of the satellite.

Thirdly, the  $\gamma$  absorption of the satellite was invastigated. Parallel  $\gamma$ -rays with the energy distribution of GRB 9900123 (fig. 1.3) was simulated. 36 positions were investigated by varying the incident angle of the  $\gamma$ s around the major axis of the satellite.

Fourthly, the effect of cosmic radiation was invastigated by utilizing the energy spectra of cosmic electrons at an altitude of 500km. The satellite was radiated by these particles in Geant4 in the same positions that were used for the  $\gamma$ -rays of GRB 9900123. The induced signal of these particles was determined, which is about one order of magnitude lower than the signal of a GRB. This signal of electrons needs to be minimalized as it "competes" with the signal of gamma particles from GRBs.

The results of the simulation shows that the satellite is clearly capable of detecting a GRB with a high signal to noise ratio. The effect of absorption can reduce the measured light yield by 60 %. The effect of the electron background was also quantified. It is an order of magnitude lower to the signal induced by the  $\gamma$ -rays of a GRB.

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## **NYILATKOZAT**

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**NEPTUN azonosító:**

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A **diplomamunka** szerzőjeként fegyelmi felelősségem tudatában kijelentem, hogy a dolgozatom önálló munkám eredménye, saját szellemi termékem, abban a hivatkozások és idézések standard szabályait következetesen alkalmaztam, mások által írt részeket a megfelelő idézés nélkül nem használtam fel.

Budapest, 20

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*a hallgató aláírása*