

Reactive Flow Simulations in the Hyshot II Scramjet: Validation and Quantification of the Margin to Unstart

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1 Introduction & Motivation **GI/DS/JL**

- scramjet and hyshot
- validation challenges
- unstart by thermal choking
- simulations RANS/LES
- validation & uncertainty quantification
- QMU definitions and objectives
- the PSAAP program at Stanford

2 The Hyshot II Scramjet **JL/ME/FP**

- the Hyshot configuration
- previous computational & experimental studies
- DLR experiment
- defying the unstart bound (Exp, LES, theory)
- metrics for detecting unstart

3 Physical Modeling and Computational Tools

- The governing equations
 - RANS equations and turbulence models **ME/CG**
 - Combustion modeling **JU/NK/JL**
 - Mixing modeling **CG/ME**
- Computational tools
 - the JOE code **FP**
 - flamelet solver **NK/JU**

4 Verification and Validation **FP/ME/JL**

- Grid convergence study
- Error estimation of numerical errors using adjoints
- Comparisons to experiments

5 Uncertainty Quantification **PC/ME/JU**

- Aleatory & Epistemic
- Sources of uncertainty
 - Inflow uncertainties
 - Kinetic rates uncertainties
 - Turbulence model uncertainties
- Subsystem decomposition

6 Quantification of Margins and Uncertainties **PC/JL/GI**

- Uncertainty propagation methodology
 - Plan \mathcal{A}
 - Plan \mathcal{B}
- Estimation of the margins

7 Conclusions **GI/DS/JL**

- Summary and main achievements
- Limitations of the present work
- Perspective on QMU