

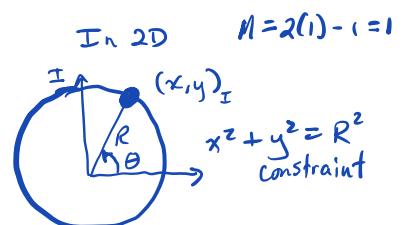
Ch 3 in KF Particle Dynamics

Constraints and Degrees of freedom (DOF)

Collection of particles: The number of DOF is the number of independent coordinates needed to describe the position of every particle.

$$2\text{-D}: M = 2N - K$$

↑ #DOF
 ↑ #particles ↓ #constraints



$$3\text{-D}: M = 3N - K$$

Collection of rigid bodies: The number of DOF is the number of coordinates needed to describe the position and orientation of every body

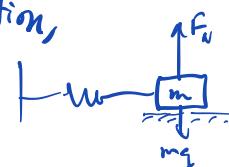
$$2\text{-D}: M = 3N - K$$



$$3\text{-D}: M = 6N - K$$

↑
Multiplier: DOF per particle

A constraint force reduces the number of DOF and often acts orthogonal to the direction of motion, in which case it is called a normal force.



Constraints can often be expressed algebraically.

Ex. Particle on sphere

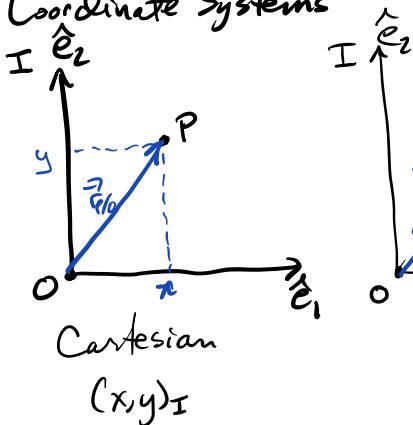


$$\|\vec{r}_{p/I_0}\| = \sqrt{x^2 + y^2 + z^2} = R$$

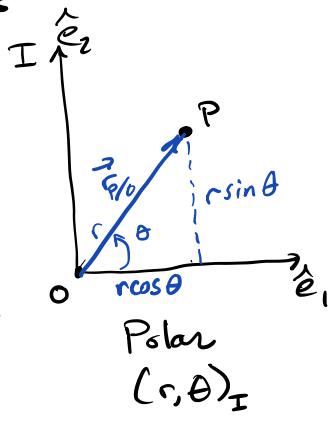
We will re-visit constraints later in the course.

Inertial Particle Kinematics in the Plane

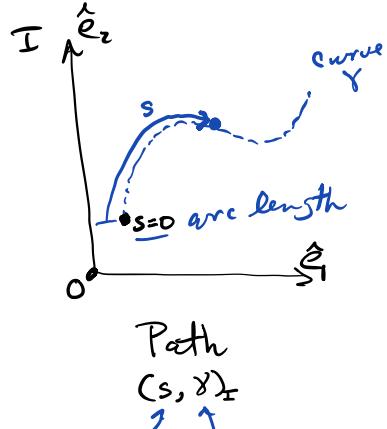
Coordinate Systems



Cartesian
(x, y)_I



Polar
(r, \theta)_I



Path
(s, y)_I

Note:
1 DOF
Not a coordinate,
but the curve Y

The description of a system's motion based on geometry and reference frames only;
Does not include forces.

Solving for the Kinematics of a particle (or rigid body) involves calculating the velocity and acceleration by differentiating the position vector w. r. t. time.

Cartesian coord:

$$\begin{aligned}\vec{r}_{P/I} &= x\hat{e}_1 + y\hat{e}_2 \\ \mathcal{I}\vec{v}_{P/I} &= \frac{d}{dt}(\vec{r}_{P/I}) = \dot{x}\hat{e}_1 + \dot{y}\hat{e}_2 \\ \mathcal{I}\vec{a}_{P/I} &= \frac{d}{dt}(\mathcal{I}\vec{v}_{P/I}) = \ddot{x}\hat{e}_1 + \ddot{y}\hat{e}_2\end{aligned}$$

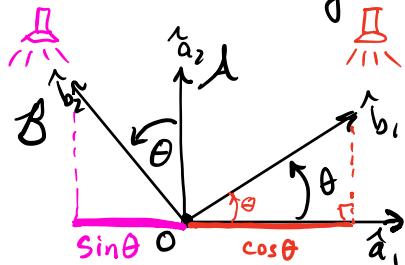
Polar Coord.:

$$\begin{aligned}\vec{r}_{P/I} &= r\cos\theta\hat{e}_1 + r\sin\theta\hat{e}_2 \\ \mathcal{I}\vec{v}_{P/I} &= \frac{d}{dt}(\vec{r}_{P/I}) = (r\cos\theta - r\sin\theta\dot{\theta})\hat{e}_1 + (r\sin\theta + r\cos\theta\dot{\theta})\hat{e}_2 \\ \mathcal{I}\vec{a}_{P/I} &= \frac{d}{dt}(\mathcal{I}\vec{v}_{P/I}) = (r\ddot{\cos\theta} - 2\dot{r}\dot{\theta}\sin\theta - r\ddot{\sin\theta} - r\dot{\theta}^2\cos\theta)\hat{e}_1 \\ &\quad + (r\ddot{\sin\theta} + 2\dot{r}\dot{\theta}\cos\theta + r\ddot{\theta}\cos\theta - r\dot{\theta}^2\sin\theta)\hat{e}_2\end{aligned}$$

Sometimes problems are easily solved with making use of different types of reference frames, including moving and noninertial frames.

But first...

Transforming between reference frames



SOH CAH TOA

$$\begin{aligned} A &= (O, \hat{a}_1, \hat{a}_2, \hat{a}_3) \\ B &= (O, \hat{b}_1, \hat{b}_2, \hat{b}_3) \end{aligned}$$

Transformation Table

	\hat{b}_1	\hat{b}_2
\hat{a}_1	$\langle \hat{a}_1, \hat{b}_1 \rangle$	$\langle \hat{a}_1, \hat{b}_2 \rangle$
\hat{a}_2	$\langle \hat{a}_2, \hat{b}_1 \rangle$	$\langle \hat{a}_2, \hat{b}_2 \rangle$

dot product

$$\hat{a}_1 \cdot \hat{b}_2 \text{ or } \hat{a}_1^T \hat{b}_2$$

The (i, j) th entry represents the dot product between the i th row and j th column

$$\begin{matrix} & \hat{b}_1 & \hat{b}_2 \\ \hat{a}_1 & \boxed{\cos\theta \quad -\sin\theta} \\ \hat{a}_2 & \boxed{\sin\theta \quad \cos\theta} \end{matrix}$$

To $\overset{\text{Frame}}{A}$ $\overset{\text{Frame}}{B}$

$$C = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix}$$

$$\begin{bmatrix} \vec{r}_{P/O} \\ A \end{bmatrix} = C \begin{bmatrix} \vec{r}_{P/O} \\ B \end{bmatrix}$$

This transformation table process results in
Transformation matrices.

Often, but not always, these transformation matrices are also rotation matrices. (If there is a reflection involved, the resulting matrix will not be a rotation matrix.)

An important property for rotation matrices:

$$({}^B C {}^A)^T = ({}^B C {}^A)^{-1} \text{ orthogonal matrices}$$

Now, if I know the ${}^A C {}^B$ is a rotation matrix, I can find ${}^B C {}^A$ easily:

$$[\vec{r}_{P/O}]_A = {}^A C {}^B [\vec{r}_{P/O}]_B$$

$$({}^A C {}^B)^{-1} [\vec{r}_{P/O}]_A = [\vec{r}_{P/O}]_B$$

or

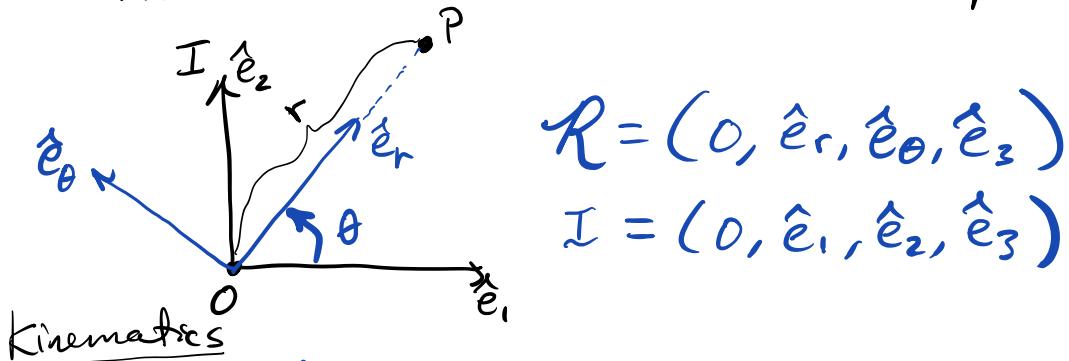
$$[\vec{r}_{P/O}]_B = ({}^A C {}^B)^{-1} [\vec{r}_{P/O}]_A$$

$$[\vec{r}_{P/O}]_B = \underbrace{({}^A C {}^B)^T}_{{}^B C {}^A} [\vec{r}_{P/O}]_A$$

$$\Rightarrow {}^B C {}^A = ({}^A C {}^B)^T$$

Polar Frame

↳ Reference frame that rotates to follow a point



$$R = (0, \hat{e}_r, \hat{e}_\theta, \hat{e}_z)$$

$$I = (0, \hat{e}_1, \hat{e}_2, \hat{e}_3)$$

Kinematics

$$\vec{r}_{P/O} = r \hat{e}_r$$

$${}^I \vec{V}_{P/O} = \dot{r} \hat{e}_r + r \frac{d}{dt} (\hat{e}_r)$$

what is
this?

Option #1

Convert all basis vectors to the I frame, then differentiate, and substitute back to get back to the polar frame

Transformation Table

\hat{e}_1	\hat{e}_r	\hat{e}_θ
\hat{e}_1	$\cos\theta$	$-\sin\theta$
\hat{e}_2	$\sin\theta$	$\cos\theta$

To
↓
 I
From
↓
 R

$$\vec{r}_{P/O} = r \hat{e}_r$$

$$= r(\cos\theta \hat{e}_1 + \sin\theta \hat{e}_2)$$

$${}^I \vec{V}_{P/O} = \dot{r} \cos\theta \hat{e}_1 - r \sin\theta \dot{\theta} \hat{e}_1 + \dot{r} \sin\theta \hat{e}_2 + r \cos\theta \dot{\theta} \hat{e}_2$$

$$= \dot{r} (\cos\theta \hat{e}_1 + \sin\theta \hat{e}_2) + r \dot{\theta} (-\sin\theta \hat{e}_1 + \cos\theta \hat{e}_2)$$

$$= \dot{r} \hat{e}_r + r \dot{\theta} \hat{e}_\theta$$

Compare to

$$= \dot{r} \hat{e}_r + r \frac{d}{dt} (\hat{e}_r)$$

$$\frac{d}{dt} (\hat{e}_r) = \dot{\theta} \hat{e}_\theta$$

Caution: The \hat{e}_3 vectors must align between frames for this result.



Option #2: Differentiate unit vectors using the angular velocity vector

Angular velocity of R with respect to I } $\overset{I}{\vec{\omega}} = \dot{\theta} \hat{e}_3$

$$\begin{aligned}\overset{I}{\frac{d}{dt}}(\hat{e}_r) &= \overset{I}{\frac{d}{dt}}(\cos\theta \hat{e}_1 + \sin\theta \hat{e}_2) = \\ &= \dot{\theta} \underbrace{(-\sin\theta \hat{e}_1 + \cos\theta \hat{e}_2)}_{\hat{e}_\theta} = \dot{\theta} \hat{e}_\theta = \dot{\theta} (\hat{e}_3 \times \hat{e}_r)\end{aligned}$$

$$\overset{I}{\frac{d}{dt}}(\hat{e}_r) = \overset{I}{\vec{\omega}} \times \hat{e}_r = \dot{\theta} \hat{e}_\theta$$

$$\overset{I}{\frac{d}{dt}}(\hat{e}_\theta) = \overset{I}{\vec{\omega}} \times \hat{e}_\theta = -\dot{\theta} \hat{e}_r$$

Inertial kinematics of point P in the Polar Frame

$$\vec{r}_{P/O} = r \hat{e}_r$$

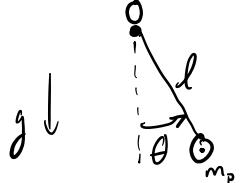
$$\overset{I}{\vec{v}}_{P/O} = \dot{r} \hat{e}_r + r \dot{\theta} \hat{e}_\theta$$

$$\begin{aligned}\overset{I}{\vec{a}}_{P/O} &= \ddot{r} \hat{e}_r + 2\dot{r} \dot{\theta} \hat{e}_\theta + r \ddot{\theta} \hat{e}_\theta - r \dot{\theta}^2 \hat{e}_r \\ &= (\ddot{r} - r \dot{\theta}^2) \hat{e}_r + (2\dot{r} \dot{\theta} + r \ddot{\theta}) \hat{e}_\theta\end{aligned}$$

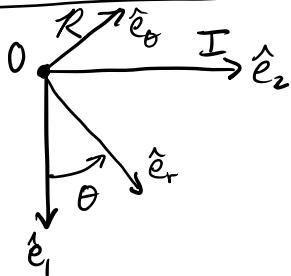
→ This is what you plug into N2L

$$\vec{F}_P = m_P \overset{I}{\vec{a}}_{P/O}$$

Ex. Pendulum



Reference Frames



$$\underline{\text{DOF}}$$

$$M = 2N - K$$

↑ ↑
| |
I I

$$M = 1 \text{ DOF}$$

Kinematics:

$$\vec{r}_{\text{Pc}} = l \hat{e}_r$$

$$\vec{V}_{p0} = \ell \hat{\theta} \hat{e}_\theta$$

$$\boxed{I\vec{a}_{Pb}} = l\ddot{\theta}\hat{e}_\theta - l\dot{\theta}^2\hat{e}_r$$

$$\frac{d}{dt}(\hat{\mathbf{e}}_\theta) = -\dot{\theta} \hat{\mathbf{e}}_r$$

N2L:

$$\vec{F}_p = m_p \vec{a}_{p/b}$$

$$-T\hat{e}_r + m_p g \hat{e}_1 = m(\ddot{\ell}\hat{e}_\theta - \dot{\ell}\dot{\theta}^2 \hat{e}_r)$$

Rewriting \hat{e}_i

To get the scalar eqn of motion, we will try to stay in the polar R frame; Let's convert the \vec{e}_1 vector

$$-T\hat{e}_r + mg(\cos\theta\hat{e}_r - \sin\theta\hat{e}_\theta) = m_p(l\ddot{\theta}\hat{e}_\theta - l\dot{\theta}^2)\hat{e}_r$$

Equate coefficients

$$\hat{e}_r: -T + mg \cos\theta = -ml\dot{\theta}^2$$

$$\vec{e}_\theta: -mg \sin \theta = m l \ddot{\theta}$$

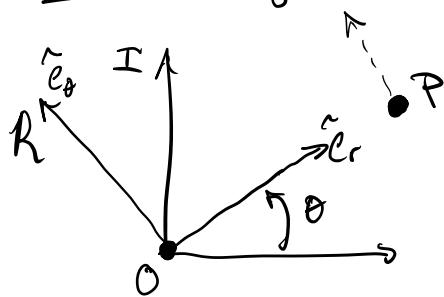
 Shorthand for sine

E.O.M.

$$\ddot{\theta} = -\frac{g}{l} \sin \theta$$

Note: 1 DOF
Since 1 constraint

Ex. Straight line motion using a polar frame



$$\overset{I}{\vec{a}}_{P/O} = (\ddot{r} - r\dot{\theta}^2) \hat{e}_r + (r\ddot{\theta} + 2\dot{r}\dot{\theta}) \hat{e}_\theta$$

Centripetal
acceleration
term

Coriolis
acceleration

$$\overset{I}{\vec{F}}_P = m_p \overset{I}{\vec{a}}_{P/O}$$

$$0 = m_p ((\ddot{r} - r\dot{\theta}^2) \hat{e}_r + (r\ddot{\theta} + 2\dot{r}\dot{\theta}) \hat{e}_\theta)$$

$$\Rightarrow \hat{e}_r : \ddot{r} = r\dot{\theta}^2$$

$$\hat{e}_\theta : \ddot{\theta} = -\frac{2\dot{r}\dot{\theta}}{r}$$

These are
kinematic
terms that
arise due to
the rotating frame.