

Total ionizing dose results and displacement damage results for candidate spacecraft electronics for NASA


Zoran Kahric

Cite this paper

Downloaded from [Academia.edu](#) 

[Get the citation in MLA, APA, or Chicago styles](#)

Related papers

[Download a PDF Pack](#) of the best related papers 



[Recent total ionizing dose results and displacement damage results for candidate spacecra...](#)

C. Poivey

[Single event effects results for candidate spacecraft electronics for NASA](#)

C. Poivey

[A compendium of recent optocoupler radiation test data](#)

jerry wert

Recent Total Ionizing Dose Results and Displacement Damage Results for Candidate Spacecraft Electronics for NASA

Donna J. Cochran, Scott D. Kniffin,
Raymond L. Ladbury, Martha V. O'Bryan,
Christian F. Poivey, Hak Kim, Tim. L. Irwin
Anthony M. Phan, Martin A. Carts, and
James D. Forney
Muñiz Engineering Inc.
NASA/GSFC Code 561.4
Greenbelt, MD USA
Donna.cochran@gsfc.nasa.gov

Kenneth A. LaBel, Robert A. Reed,
Anthony B. Sanders, Donald K. Hawkins,
Stephen R. Cox
Flight Data Systems and Radiation Effects Branch
NASA/GSFC Code 561.4
Greenbelt, MD USA
Kenneth.A.Label@nasa.gov

James W. Howard, Jr., and Rebecca C. DiBari
Jackson & Tull Chartered Engineers
Washington, DC USA
Jim.howard@gsfc.nasa.gov

Stephen P. Buchner
QSS Group Inc.
NASA/GSFC Code 561.4
Greenbelt, MD USA
Stephen.buchner@gsfc.nasa.gov

Christopher D. Palor
Orbital Sciences Corporation
McLean, VA USA
Scott.D.Kniffin.1@gsfc.nasa.gov

Abstract-- We present data on the vulnerability of a variety of candidate spacecraft electronics to total ionizing dose and displacement damage. Devices tested include optoelectronics, digital, analog, linear bipolar devices, hybrid devices, Analog-to-Digital Converters (ADCs), and Digital-to-Analog Converters (DACs), among others.

I. INTRODUCTION

In order to meet the demands of reduced cost, higher performance and more rapid delivery schedules imposed by the space flight community, commercial and emerging technology devices have assumed a prominent role in meeting these needs. With the skyrocketing increase in the use of such devices, the importance of ground based testing for the effects of total ionizing dose (TID) and proton displacement damage to qualify such devices for flight is paramount. The novel ways in which some of these devices are used also highlights the need for application specific testing to ensure their proper operation and ability to meet mission goals.

The test results presented here were gathered to establish the sensitivity of the devices selected as candidate spacecraft electronics to TID and proton damage. Proton-induced degradation is a mix of ionizing (TID) and non-ionizing damage. This non-ionizing damage is commonly referred to

as displacement damage (DD). This testing serves to determine the limit to which a candidate device may be used in space applications. For single event effects (SEE) results, see a companion paper submitted to the 2005 IEEE NSREC Radiation Effects Data Workshop entitled: "Recent Single Event Effects Results for Candidate Spacecraft Electronics for NASA" by M. O'Bryan, et al. [1]

II. TEST TECHNIQUES AND SETUP

A. Test Facilities - TID

TID testing was performed using a Co-60 source at the Goddard Space Flight Center Radiation Effects Facility (GSFC REF). The source is capable of delivering a dose rate of up to 0.5rads(Si)/s, with dosimetry being performed by an ion chamber probe.

B. Test Facilities – Proton

Proton DD/TID tests were performed at the University of California at Davis Crocker Nuclear Laboratory (UCD-CNL) that has a 76" cyclotron (maximum energy of 63 MeV. Table I lists the proton damage test facility and energies used on the devices.

Table I: Proton Test Facilities

Facility	Proton Energy, (MeV)
University of California at Davis Crocker Nuclear Laboratory (UCD-CNL)	26.6-63

The Authors would like to acknowledge the sponsors of this effort: NASA Electronic Parts and Packaging Program (NEPP), and NASA Flight Projects.

C. Test Methods

Unless otherwise noted, all tests were performed at room temperature and with nominal power supply voltages.

1) TID Testing

TID testing was performed to the MIL-STD-883 1019.6 test method [2].

2) Proton Damage Testing

Proton damage tests were performed on biased devices with functionality and parametrics being measured either continually during irradiation (in-situ) or after step irradiations (for example: every 10krads(Si), or every 1×10^{10} protons).

III. TEST RESULTS OVERVIEW

Abbreviations for principal investigators (PIs) are listed in Table II. Definitions for the categories are listed in Table III. Abbreviations and conventions are listed in Table IV. This paper is a summary of results. Please note that these test results can depend on operational conditions. Complete test reports are available online at <http://radhome.gsfc.nasa.gov> [3].

TABLE II: LIST OF PRINCIPAL INVESTIGATORS

Abbreviation	Principal Investigator (PI)
SB	Steve Buchner
BD	Becky DiBari
SK	Scott Kniffin

TABLE III: LIST OF CATEGORIES

1	Not tested to failure.
2	Degradation at >50krads(Si)
3	Degradation at 20-50krads(Si)
4	Degradation at 5-20krads(Si)
5	Degradation at 5krads(Si) or less
REV	Research Test Vehicle – Please contact the P.I. before utilizing this device for spacecraft applications.

TABLE IV: ABBREVIATIONS AND CONVENTIONS:

ACRONYM	DEFINITION	ACRONYM	DEFINITION
ADC	analog to digital converter	LDC	lot date code
ASIC	application specific integrated circuit	LED	light emitting diode
CCD	charge coupled device	I _{CC}	power supply current
CMOS	complementary metal oxide semiconductor	MeV	mega electron volt
DAC	digital to analog converter	N/A	not applicable
DD	displacement damage	op amp	operational amplifier
DNL	differential non-linearity	p/cm ²	protons/cm ²
FET	field effect transistor	PI	Principal Investigator
GSFC REF	Goddard Space Flight Center Radiation Effects Facility	PT	photo transistor
I _b	bias current	TID	total ionizing dose
I _c	collector current	UCD-CNL	University of California at Davis Crocker Nuclear Laboratory
I _f	forward current	V _{OL}	output saturation voltage
I _{OS}	offset current	V _{out}	output voltage
I _{STDBY}	standby current	V _{ce}	collector emitter voltage

TABLE V: SUMMARY OF TID AND DD TEST RESULTS

Part Number	Manufacturer	LDC	Function	Facility Date/P.I (Co-60 source unless otherwise noted).	Dose rate (rads(Si)/s)	Summary of Results	Degradation Level (krads(Si))	Test Report	Cat.
Data Converters									
MAX529	Maxim	0101, 0126	8-Bit DAC	GSFC04MAY/SK	0.51	All parts passed all tests up to 5.0krads(Si).	>5	G04May_M AX529_TID. pdf	1
AD574	Analog Devices	9245, 9248, 9442	12-Bit ADC	GSFC04SEP/SK	0.17	All parts passed all tests up to 20krads(Si). After 30krads(Si) and higher, all devices exceed the specification limit for INL (greater than 1lsb).	30	G04SEP_AD5 74_TID.pdf	3
AD7545	Analog Devices	0503A	12-Bit ADC	GSFC05JAN/BD	0.15	Functional failure at 5krad (Si), no recovery observed after annealing.	5	G05JAN_AD7 545_TID.pdf	5
AD7846SQ	Analog Devices	Q0408A	16-Bit DAC	GSFC05FEB/BD	0.42	DNL exceeds specification limit at 10krads(Si). Functional failure at 15krads(Si), recovered after 168 hour annealing., parametric degradation continues. Devices were taken to 20krads(Si) and no functional failure was observed. After 25krads(Si), functional failures were again observed.	10	G05FEB_AD7 846_TID.pdf	4
Operational Amplifiers									
OP27	Analog Devices	0402F	Op Amp	GSFCFEB05/BD	0.6	All parts passed all tests up to 20krads(Si).	>20	G05FEB_OP27 _TID.pdf	1
OP27A	Analog Devices	9347, 9407	Op Amp	GSFC04SEP/SK	0.17	All parts passed all tests up to 30krads(Si). Degradation was seen in +I _b and -I _b from 50 to 100krads(Si). After annealing two of the devices return to within specification limits.	50	G04SEP_OP27 A_TID.pdf	3
OP200	Analog Devices	2C0347G	Op Amp	GSFC05JAN/BD	1.13	All parts passed all tests up to 5krads(Si). I _{os} exceeds specification limits after 10krads(Si). After annealing, all devices returned to within specification limits.	10	G05JAN_OP20 0_TID.pef	4
OP42AZ	Analog Devices	2C0345G	Op Amp	GSFC05JAN/BD	1.07	+I _b exceeded specification limits at 10krad(Si).	10	G05JAN_OP42 _TID.pdf	4
OP400	Analog Devices	2B0404F	Op Amp	GSFC05JAN/BD	1.13	All devices pass all tests after 1 krad(Si). +I _b exceeded specification limits after 5krads(Si).	5	G05JAN_OP40 0_TID.pdf	5
OP77	Analog Devices	3B0402F	Op Amp	GSFC05MAR/BD	1.13	+I _b exceeded specification limits after 10krads(Si).	10	G05MAR_OP7 7_TID.pdf	4
AD744	Analog Devices	0000G	BiFET Op Amp	GSFC05FEB/BD	0.71	All parts passed all tests up to 5krads(Si). After 10krads(Si), all devices exceeded the specification limit for +I _b . After annealing, all devices remain above the specification limit.	10	G05FEB_AD7 44_TID.pdf	4

TABLE V: SUMMARY OF TID AND DD TEST RESULTS (CONT.)

Part Number	Manufacturer	LDC	Function	Facility Date/P.I (Co-60 source unless otherwise noted).	Dose rate (rads(Si)/s)	Summary of Results	Degradation Level (krads(Si))	Test Report	Cat.
DC/DC Converters and Related Devices									
AFL2803R3S	Advanced Analog (IR)	0351	3.3V, DC/DC Converter	GSFC04JUN/SK	0.028	All parts passed all tests up to 10krads(Si).	>10	G04JUN_AFL 2803R3S_TID. pdf	1
MAX724ECK	Maxim	0342	DC/DC Regulator	GSFC04NOV/SK	0.23	All parts passed all tests up to 20krads(Si).	>20	G04NOV_MA X724ECK_TID .pdf	1
Logic Devices									
54ACTQ16245	National Semiconductor	0409	16-Bit Transceiver	GSFC04NOV/SK	0.23	Some VOL measurements exceeded specification limits after 10krads(Si), however these parameters were within specification after this step. All 8 ICC measurements exceeded specification limits after 15krads(Si) and 20krads(Si). Significant changes occurred following annealing, see report.	10	G04NOV_54A CTQ16245_TI D.pdf	4
HN58C1001T15	Renesas	0433	EEPROM	GSFC05JAN/SB	0.8	All parts passed all tests up to 100krads(Si).	>100	G05JAN_HN5 8C1001T15_TI D.pdf	1
Other Linear Devices									
AD584	Analog Devices	0348B, 0413D	Voltage Reference	GSFC05JAN/BD	0.7	V _{OUT} (10V) exceeded specification limits after 15krads(Si). No recovery was noted after annealing.	15	G05JAN_AD5 84_TID.pdf	4
AD822	Analog Devices	0029B	Low Power FET	GSFC04MAY/SK	0.04	All parts passed all tests up to 10krads(Si).	>10	G04MAY_AD 822_TID.pdf	1
Optical Devices									
LXA0387	LSI Logic	0414	ASIC/512k SRAM	GSFC04AUG/CP	1.3	Parametric degradation at ~90krads(Si). Functional failure at ~270krads(Si).	~90	G04AUG_LXA 0387.pdf	
Custom C-2 LED (InGaN blue)	Micropac (AXT Opto)	N/A	LED	CNL 04NOV/SK	1×10^{10} p/cm ² to 1×10^{12} p/cm ²	No degradation was seen up to 3×10^{11} p/cm ² . A small decrease in I _c was seen after 5×10^{11} p/cm ² and 1×10^{12} p/cm ² .	5×10^{11} p/cm ²	D04NOV_C2L ED.pdf	2
Custom TCM405 (GaN UV)	Micropac (III-V Components)	N/A	LED	CNL04NOV/SK	1×10^{10} p/cm ² to 1×10^{12} p/cm ²	No degradation was seen up to 1×10^{12} p/cm ² .	$>1 \times 10^{12}$ p/cm ²	D04NOV_C2L ED.pdf	1
62087-301 (LED), 61055-305 (PT)	Micropac	N/A	LED/PT Encoder	CNL04JAN/SK	1×10^{10} p/cm ² to 1×10^{12} p/cm ²	No degradation was seen up to 5×10^{10} p/cm ² . Uniform degradation occurs from 1×10^{11} p/cm ² to 1×10^{12} p/cm ² . Degradation is dependent on LED forward current. See report.	1×10^{11} p/cm ²	D04JAN_6105 5_62087.pdf	1

IV. TEST RESULTS AND DISCUSSION

1) OP27A

The OP27A operational amplifier from Analog Devices was tested to 100krads(Si) with an average dose rate of 0.17krads(Si)/s. The two LDCs tested were 9347 and 9407. The devices were statically biased. For both $+I_b$ and $-I_b$, three devices exceeded the specification limits after 40krads(Si); all devices exceeded specification limits for both parameters after 50krads(Si) and continued to degrade through 100krads(Si). There was significant recovery in these parameters with two devices having readings within specification limits following annealing. Both LDCs of devices behaved similarly in terms of degradation; however, LDC 9407 did perform slightly better overall. See Figures 1 and 2 for comparisons of lot-to-lot I_b degradation.

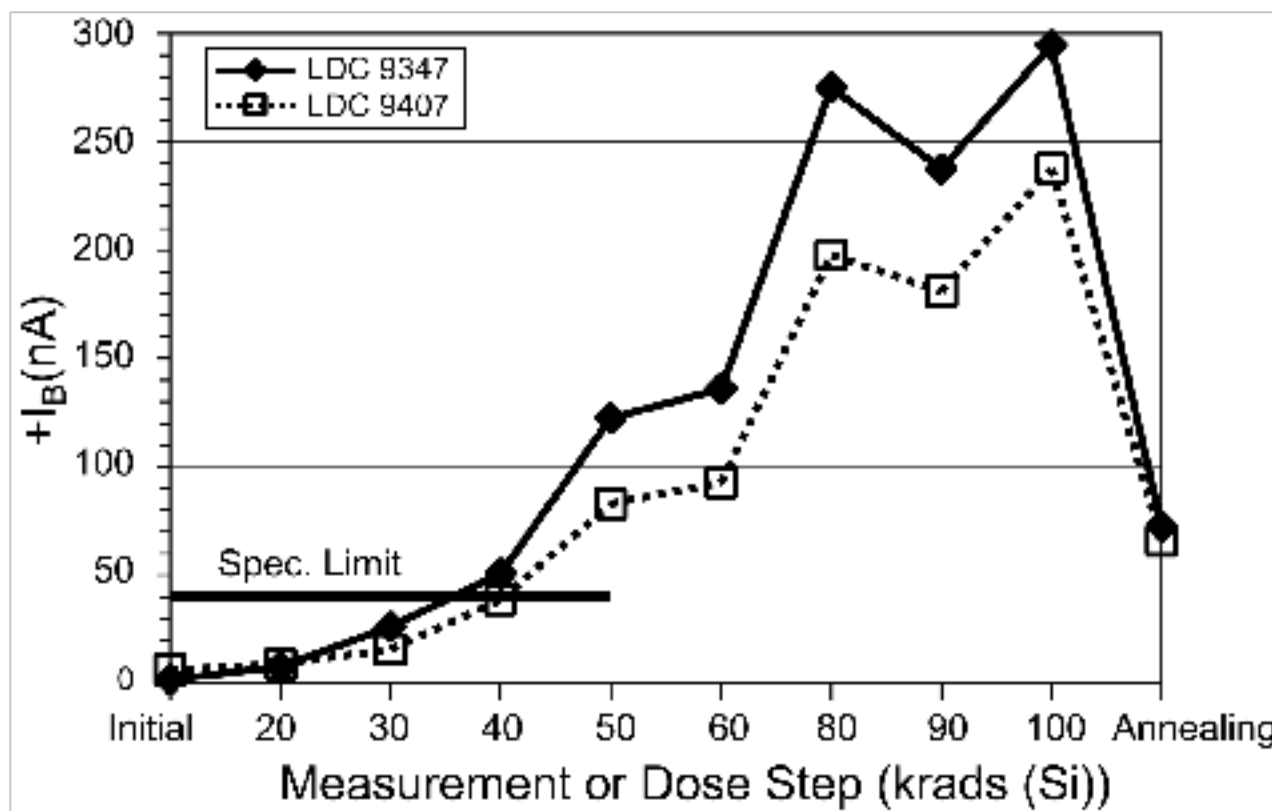


Fig. 1. Analog Devices OP27A $+I_b$ degradation by LDC.

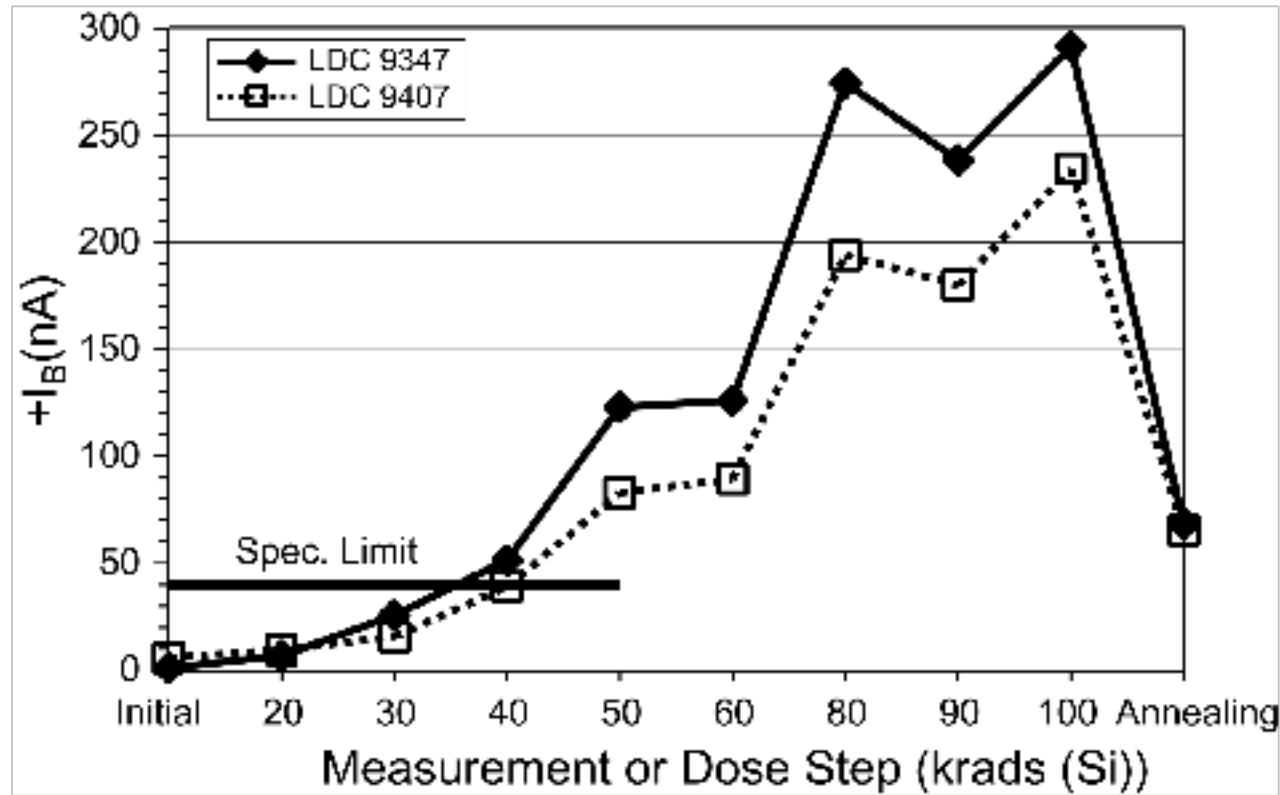


Fig. 2. Analog Devices OP27A $-I_b$ degradation by LDC.

2) 54ACTQ16245

The 54ACTQ16245 16-bit transceiver from National Semiconductor was tested to 20krads(Si) with an average dose rate of 0.23krads(Si)/s. The eight test devices were statically biased. Some V_{OL} measurements exceeded specification limits after 10krads(Si) only to return to within specification limits at higher dose levels and following annealing. Following the 15krads(Si) exposure, all devices go beyond specifications for all 8 I_{CC}

measurements. Two devices had readings in certain I_{CC} measurements that fell below the lower specification limit for those tests while all devices in all other I_{CC} tests exceeded the upper specification limits. After 20krads(Si), all devices had all 8 I_{CC} measurements exceeding specification limits. Following 168 hours of room temperature annealing, the results become more complicated. Five devices exhibited what is interpreted to be a significant secondary effect in the form of long-term charge trap collection. The four I_{CC} -high measurements in these devices went from significantly exceeding the specification limits for I_{CC} to falling significantly below the specification limits for these I_{CC} parameters with three specific exceptions within this subgroup where the measurements remained only slightly higher than specification limits. For these five devices, all of the I_{CC} -low measurements continued to exceed the specification limits for those parameters. All other devices continued to exceed specification limits for all I_{CC} parameters. These results imply that there is a propensity for the devices to collect charge traps over time that cause additional damage. The fact that this is not noted in all of the samples tested indicates that there is an inconsistent electrical margin for the devices within this lot.

3) Blue LED (470nm)

Displacement damage testing was performed on a 470nm blue LED die (InGaN), manufactured by AXT Optoelectronics, custom packaged by Micropac. Five devices were exposed to 63MeV protons at UCD-CNL. The devices were unbiased during each irradiation step. The LEDs were tested in a custom wooden jig to eliminate stray light and enable test repeatability.

The LED response was measured by sweeping the LED forward current from 0.1mA to 20mA in log steps and collecting the light output (I_C) with a photo diode that remained constant and unirradiated during all testing. The devices were measured twice after each exposure to check for charge injection annealing that can result from testing the devices. There was typically a 10nA increase in collector current between the first and second test. This does not significantly change the results in any way and implies nominal charge injection annealing.

The LEDs showed no significant degradation up to 3×10^{11} p/cm² with a photocurrent drop of ~50nA. Slight degradation is seen after 5×10^{11} p/cm² and 1×10^{12} p/cm² with a 0.103 to 0.168μA drop in photocurrent. It should be noted that so long as I_F is greater than 1mA, the devices do perform consistently. Fig. 3 shows typical LED response to increasing fluence for these devices.

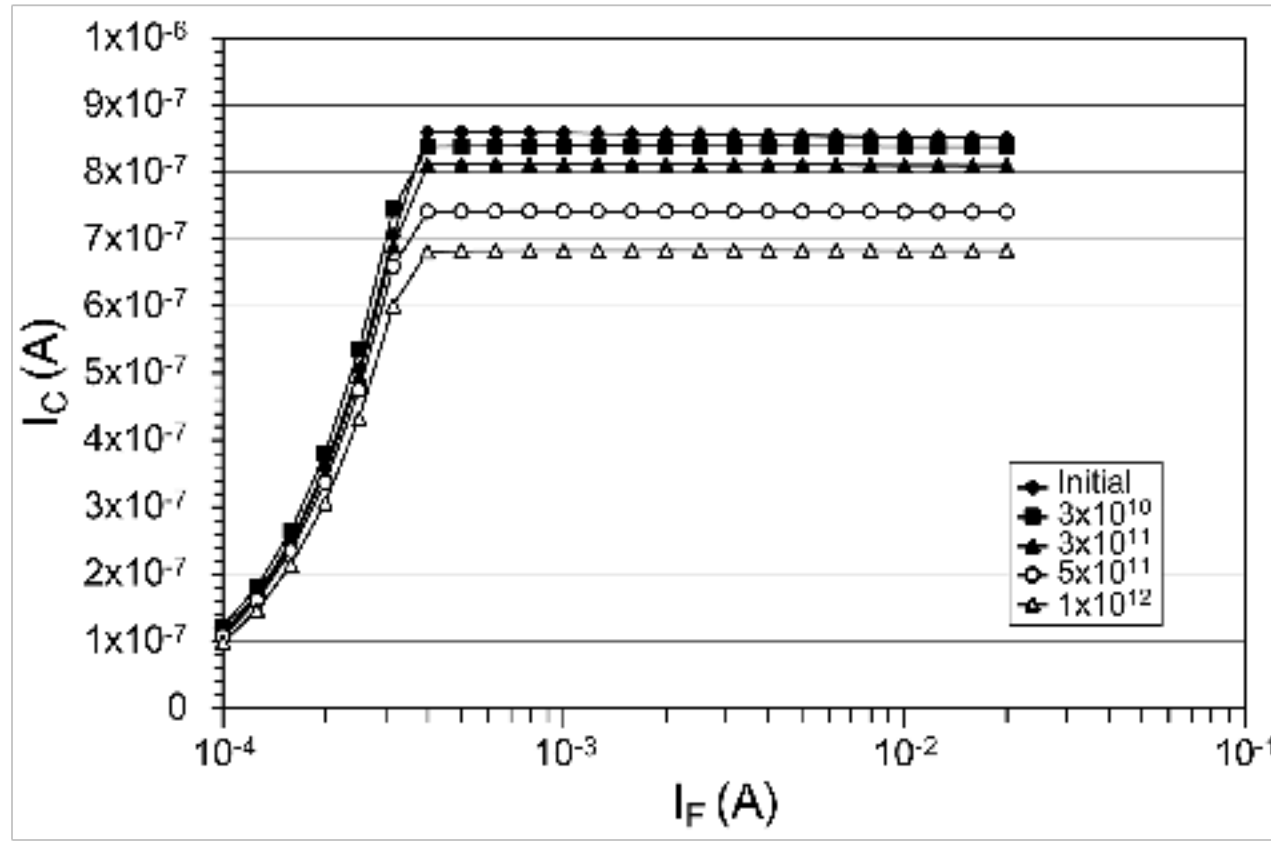


Fig. 3. Micropac (AXT Optoelectronics die) 470nm InGaN blue LED proton displacement damage as a function of total fluence in p/cm^2 .

4) LED/PT Encoder Pair

Displacement damage testing was performed on LED/PT encoder pairs, custom packaged by Micropac. A total of four pairs of devices were exposed to 63MeV protons at UCD CNL. The devices were unbiased during each irradiation step. The device pairs were custom packaged in a single unit that had the full area of the LED exposed, a small air gap, a tall but narrow aperture, and the PT behind the aperture. The aperture was designed to mimic the encoder blade that will pass between the devices in this mission's application. This enabled the devices to be qualified together as a system in a mission-specific flight configuration. The importance of this style of testing was demonstrated by Kniffin, et.al., at the 2003 RADECS Conference [4].

The tests performed on the encoder pairs were conducted as follows. The collector current (I_C) of the PT was swept from $1\mu\text{A}$ to 20mA in log steps and was done by a parametric analyzer while measuring V_{CE} . This was done while the LED forward current (I_F) was held constant from 0 to 20mA in 1mA steps. Figures 4 through 9 show the progression of degradation for a given device pair. Each line of data on the graphs represents each PT I_C sweep with the corresponding LED I_F given in the legend.

No significant degradation was seen up to $5 \times 10^{10} \text{ p}/\text{cm}^2$. Degradation was uniform from this point forward, affecting all devices nearly equally. The devices show degradation in both the amount of I_C that can be delivered before shut off and in the increase in V_{CE} . At the mission required test fluence of $3 \times 10^{11} \text{ p}/\text{cm}^2$, there is nearly an order of magnitude increase in V_{CE} for any given point where the pair is on. The device pairs also show what was effectively a failure for LED $I_F = 1\text{mA}$ at $1 \times 10^{12} \text{ p}/\text{cm}^2$ total fluence.

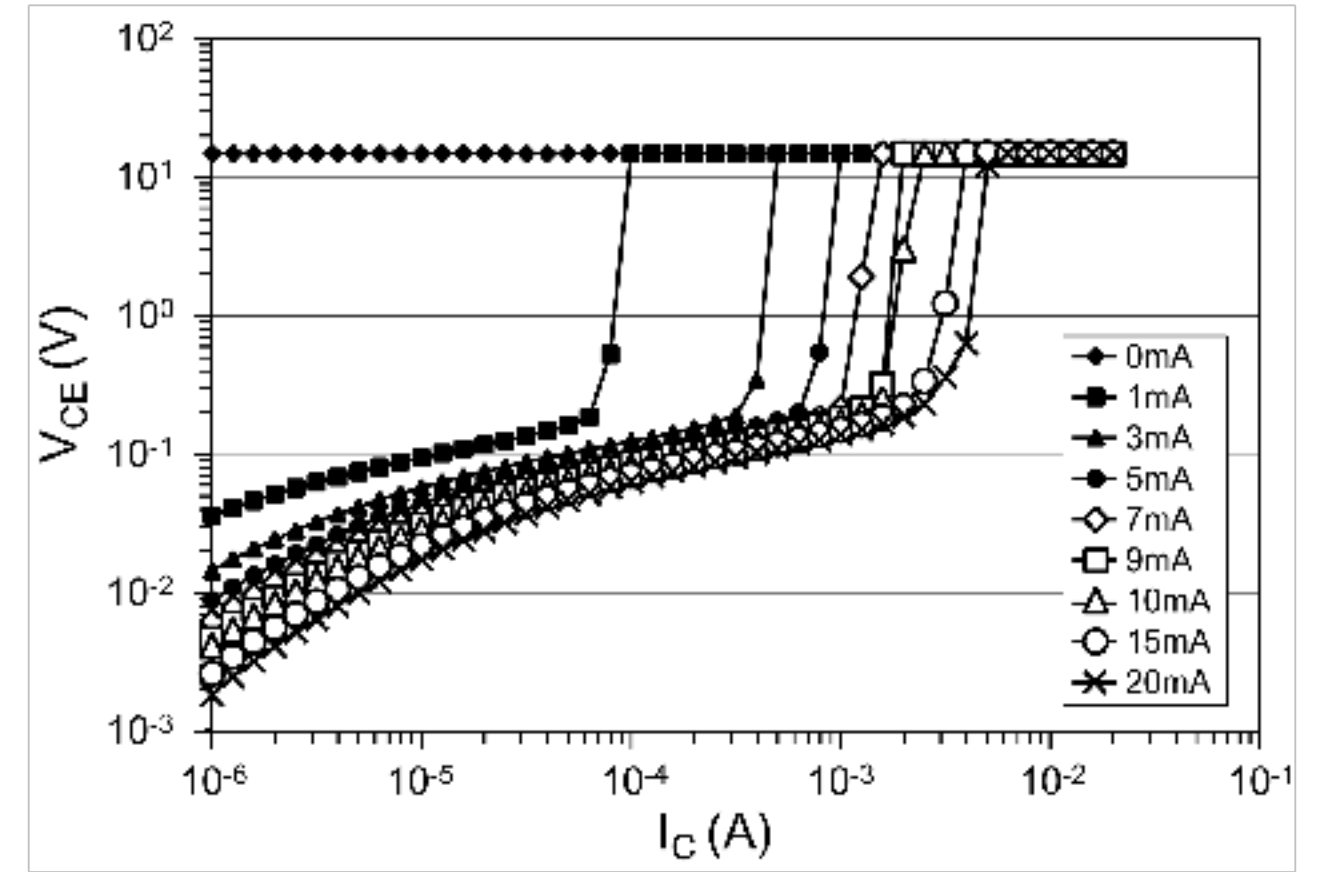


Fig. 4. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F (Pre-Irradiation).

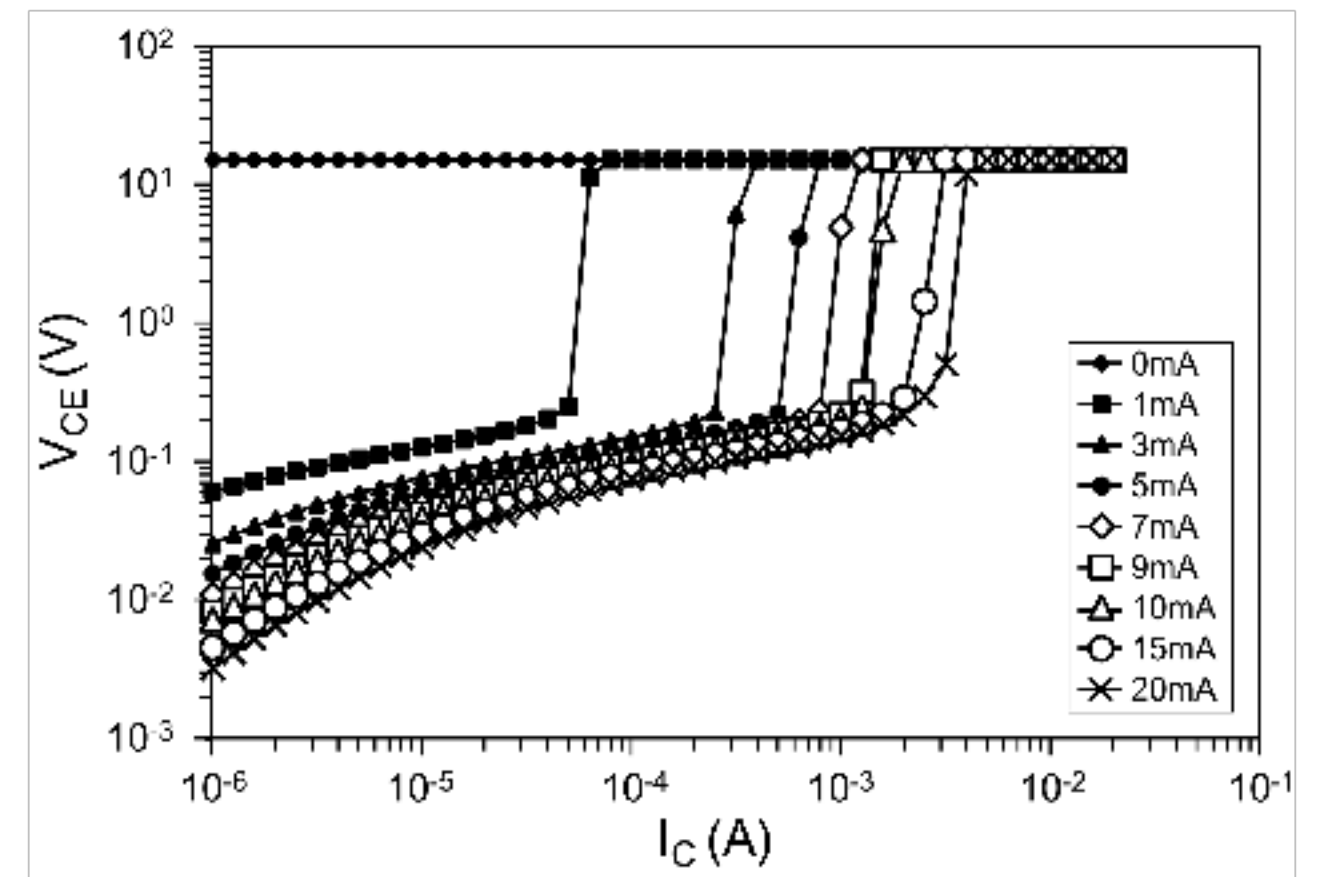


Fig. 5. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F ($5 \times 10^{10} \text{ p}/\text{cm}^2$).

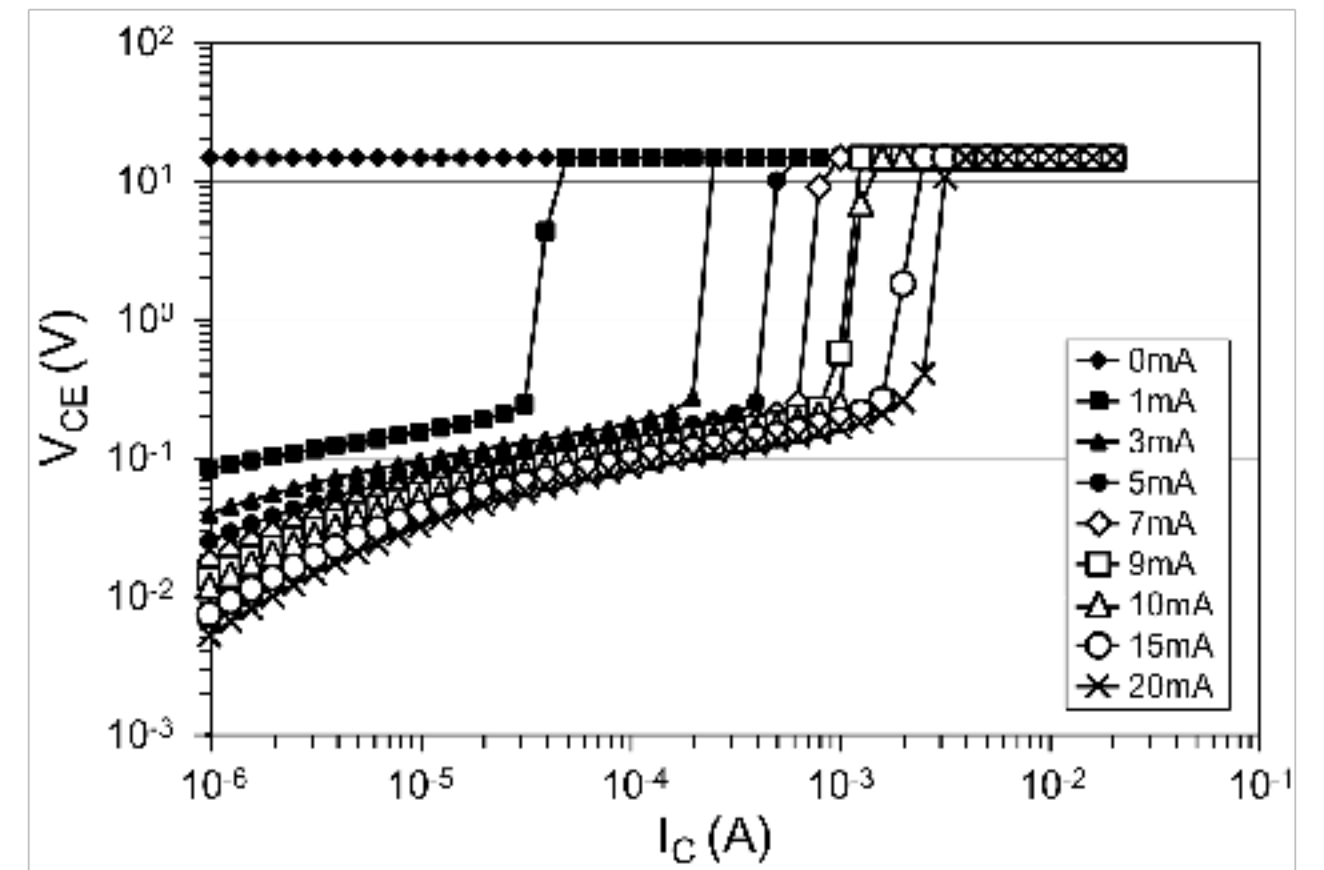


Fig. 6. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F ($1 \times 10^{11} \text{ p}/\text{cm}^2$).

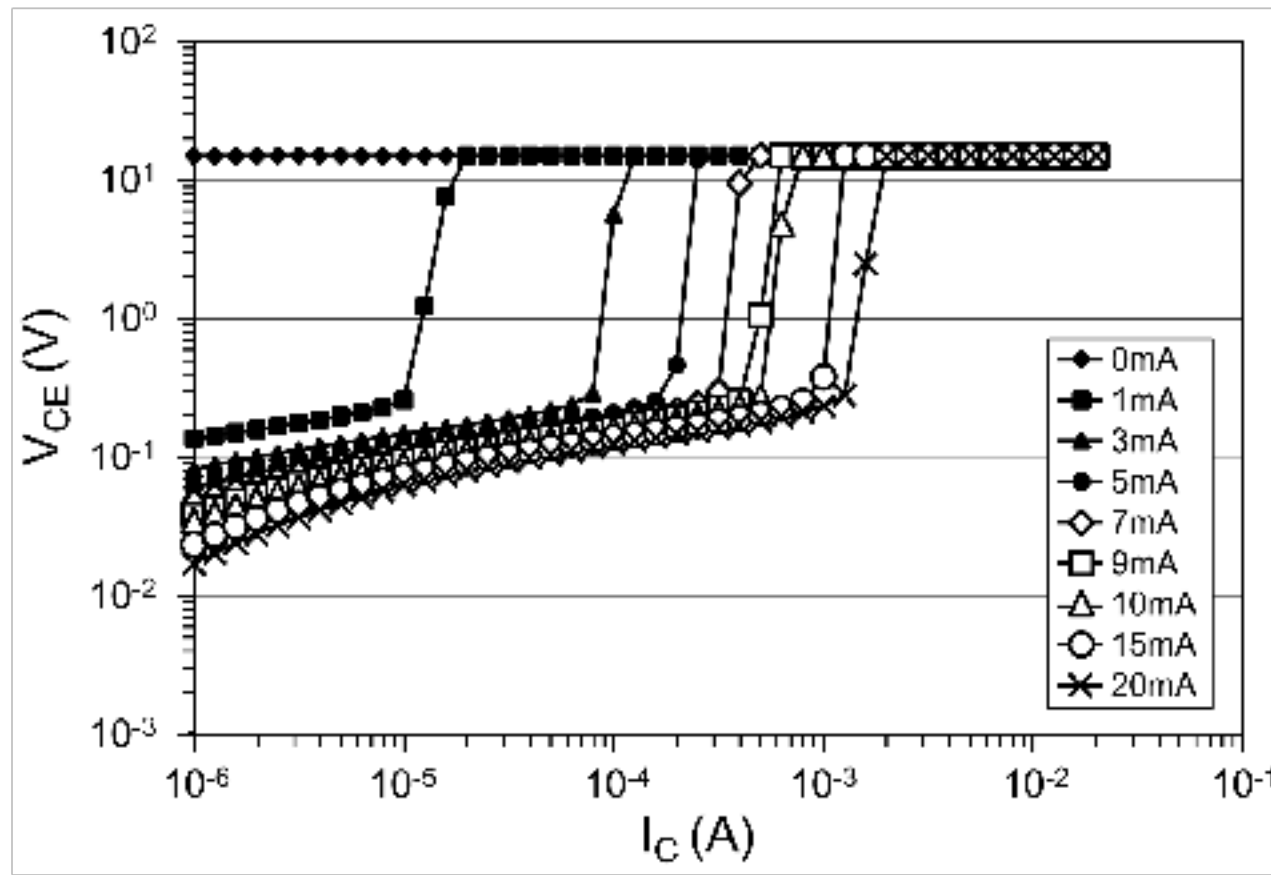


Fig. 7. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F (3×10^{11} p/cm²).

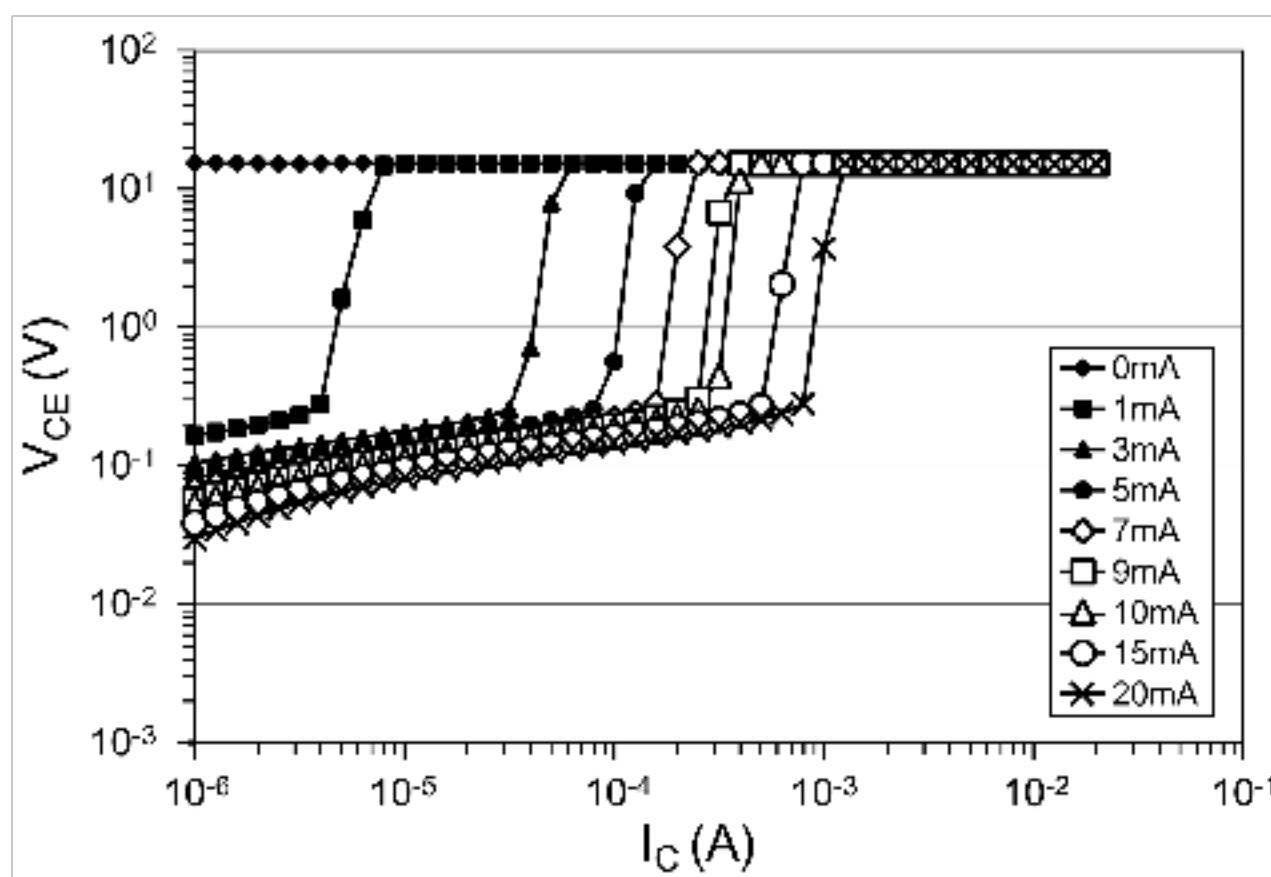


Fig. 8. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F (5×10^{11} p/cm²).

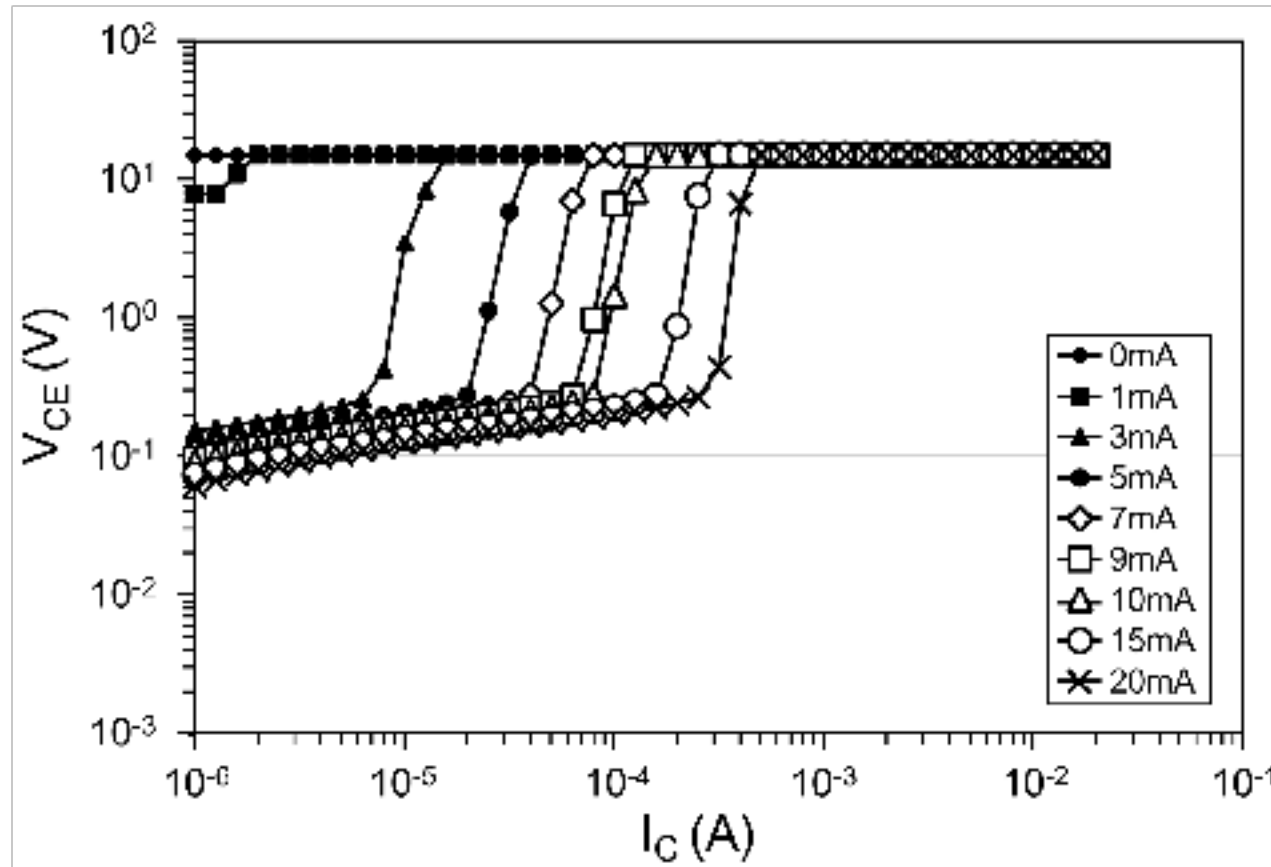


Fig. 9. Micropac custom encoder pair PT V_{CE} measurements as a function of PT I_C at various LED I_F (1×10^{12} p/cm²).

V. SUMMARY

We have presented data from recent TID and proton-induced damage tests on a variety of primarily commercial devices. It is the authors' recommendation that this data be used with caution. We also highly recommend that lot testing be performed on any suspect or commercial device.

VI. ACKNOWLEDGMENT

The authors would like to acknowledge the sponsors of this effort: a portion of the NASA Electronic Parts and Packaging (NEPP) program, and NASA Flight Projects.

VII. REFERENCES

- [1] M. O'Bryan, et al., Recent Single Event Effects Results for Candidate Spacecraft Electronics for NASA" submitted to the 2004 IEEE NSREC Radiation Effects Data Workshop.
- [2] Department of Defense Test Method Microcircuits, MIL-STD-883 Test Method Standard, Microcircuits, MIL-STD-883 Test Method 1019.6, Dated: 07 March 2003, File name: std883not5.pdf, <http://www.dscc.dla.mil/Downloads/MilSpec/Docs/MIL-STD-883/std883not5.pdf>
- [3] NASA/GSFC Radiation Effects and Analysis home page, <http://radhome.gsfc.nasa.gov>
- [4] [MAX529] S. Kniffin, P. Kang, "Radiation Report on MAX529 (LDCs 0101 and 0126)," G04MAY_MAX529_TID.pdf
- [5] [AD574] S. Kniffin, J. Forney, "Radiation Report on AD574 (LDC 9245, 9248 and 9442)," G04SEP_AD574_TID.pdf
- [6] [AD7545] B. DiBari, and A. Pham, "Radiation Report on AD7545AUQ/883B (DC: 0503A)," G05JAN_AD7545_TID.pdf
- [7] [AD7846SQ] B. DiBari, C. Palor, and A. Pham, "Radiation Report on AD7846SQ (DC: Q0408A)," G05JAN_AD7846SQ_TID.pdf
- [8] [OP27] B. DiBari, C. Palor, and A. Pham, "Radiation Report on OP27 (LDC 0402F)," G05FEB_OP27_TID.pdf
- [9] [OP27A] S. Kniffin, and C. Palor, "Radiation Report on OP27A (LDC 9347 and 9407)," G04SEP_OP27A_TID.pdf
- [10] [OP200] B. DiBari, C. Palor, and A. Pham, "Radiation Report on OP200AZMDA (DC: 2C0347G)," G05MAR_OP200_TID.pdf
- [11] [OP42] B. DiBari, C. Palor, and A. Pham, "Radiation Report on OP42AZ/883 (LDC 2C0345G)," G05MAR_OP42_TID.pdf
- [12] [OP400] B. DiBari, C. Palor, and A. Pham, "Radiation Report on OP400AYMDA (DC: 2B0404F)," G05MAR_OP400_TID.pdf
- [13] [OP77] B. DiBari, C. Palor, and A. Pham, "Radiation Report on OP77AZMDA (DC: 3B0402F)," G05MAR_OP77_TID.pdf
- [14] [AD744] B. DiBari, C. Palor, and A. Pham, "Radiation Report on AD744TH (DC: 0000G)," G05MAR_AD744_TID.pdf
- [15] [AFL2803R3S] S. Kniffin, M. Carts, "Radiation Report on AFL2803R3S (IR) for the GLAST Project," G04JUN_AFL2803R3S_TID.pdf
- [16] [MAX724ECK] S. Kniffin, C. Palor, and H. Ngien, "Radiation Report on MAX724ECK (LDC 0342)," G04NOV_MAX724ECK_TID.pdf
- [17] [54ACTQ16245] S. Kniffin, C. Palor, and L. Hua, "Radiation Report on 54ACTQ16245 (LDC 0409)," G04NOV_54ACTQ16245_TID.pdf
- [18] [HN58C1001T15] Stephen Buchner, "Total Ionizing Dose Testing of HN58C1001T15 EEPROM (Renesas)," G05JAN_HN58C1001T15_TID.pdf
- [19] [AD584] B. DiBari, C. Palor, and A. Pham, "Radiation Report on AD584TH/883B (LDC 0348B & 0413D)," G05JAN_AD584_TID.pdf
- [20] [AD822] S. Kniffin, S. Norris, "Radiation Report on AD822 (LDC 0029B)," G04MAY_AD822_TID.pdf
- [21] [Custom C-2 LED (InGaN blue)] S. Kniffin, and H. Kim, "Radiation Report on Blue and Violet calibration LEDs," D04NOV_C2LED.pdf
- [22] [Custom TCM405 (GaN UV)] [TCM405] S. Kniffin, and H. Kim, "Radiation Report on Blue and Violet calibration LEDs," D04NOV_C2LED.pdf
- [23] [61055_62087] S. Kniffin, and H. Kim, "Radiation Report on LED/PT encoder pair," D111604_61055_62087.pdf
- [24] [RADECS02_Kniffin] S.D. Kniffin, R.A. Reed, P.W. Marshall, J.W. Howard, H.S. Kim, and J.P. Schepis, "The Impact of System Configuration on Device Radiation Damage Testing of Optical Components", Proceedings of RADECS 2003, Noordwijk, The Netherlands. ESA SP-536, September 2003: 17-21.