

## Turbofan Problem 2

```

import math
import matplotlib.pyplot as plt
import numpy as np
import xlwt
import sys

book = xlwt.workbook()
sheet1 = book.add_sheet('Thermal Efficiency at B=10', cell_overwrite_ok=True)

# Conditions given in the problem
Pa = 26.5
Ta = 223.252
To4_max = 1500
gamma1 = 1.4
gamma2 = 1.35
R = 287
Cp1 = (gamma1/(gamma1-1))*R/1000
Cp2 = (gamma2/(gamma2-1))*R/1000
M=0.8
Fst=0.06
hc=43000
B=1

#Efficiencies
nd=0.94
nc=0.87
#rc=24
nb=0.98
rb=0.97
nt=0.85
ncn=0.97
nf=0.92
nfn=0.98
#rf=2.0

rclist = np.linspace(20.0, 28.0, num=30, endpoint=True)
rflist = np.linspace( 1.5,  2.2, num=30, endpoint=True)

#Flow Conditions
Toa = Ta*(1 + ((gamma1-1)/2)*M**2)
print 'Toa '
print Toa
Poa = Pa*(1 + ((gamma1-1)/2)*M**2)**(gamma1/(gamma1-1))
print 'Poa '
print Poa
u = M*math.sqrt(gamma1*R*Ta)
print 'u '
print u

#Inlet/Diffuser
To2=Toa
To2s=nd*(To2-Ta)+Ta
print 'To2s '
print To2s
Po2=Pa*(To2s/Ta)**(gamma1/(gamma1-1))
print 'Po2 '
print Po2

i = 0
for rc in rclist:
    j=0
    #Compressor
    To3s=To2*rc**((gamma1-1)/gamma1)

```

## Turbofan Problem 2

```

Po3=rc*Po2
To3 = ((To3s-To2)/nc)+To2
wc_in = Cp1*(To3-To2)

#Combustor
To4=To4_max
Fb=((To4/To3)-1)/((nb*hc/(Cp2*To3))-(To4/To3)))
if Fb >= Fst:
    Fb = Fst
    To4 = (Fb*nb*hc/(Cp2)+Toa)/(1+Fb)
Po4=rb*Po3

for rf in rflist:
    #Fan
    Po8=rf*Po2
    To8s=To2*(rf**((gamma1-1)/gamma1))
    To8 = ((To8s-To2)/nf)+To2
    wf_in = B*Cp1*(To8-To2)

    #Turbine
    wt_out=wc_in+wf_in
    To5=To4-(wt_out/(Cp2*(1+Fb)))
    To5s=To4-((To4-To5)/nt)
    Po5=Po4*(To5s/To4)**(gamma2/(gamma2-1))

    #Core Nozzle
    To6=To5
    To7=To6
    Po6=Po5
    P7=Pa
    T7as=(To6/((Po6/P7)**((gamma2-1)/gamma2)))
    T7=To6-ncn*(To6-T7as)
    M7=math.sqrt(((To7/T7)-1)*(2/(gamma2-1)))
    u7 = M7*math.sqrt(gamma2*R*T7)

    #Fan Nozzle
    To8=Toa
    To9=To8
    T9as=(To8/((Po8/Pa)**((gamma1-1)/gamma1)))
    T9=To8-nfn*(To8-T9as)
    M9=math.sqrt(((To9/T9)-1)*(2/(gamma1-1)))
    u9 = M9*math.sqrt(gamma1*R*T9)

    I = B*(u9-u)+((1+Fb)*u7-u)
    TSFC = Fb/I
    Pav=((1+Fb)*(u7**2)/2 + B*(u9**2)/2 - (B+1)*(u**2)/2)
    Pin=Fb*hc*1000
    wp=I*u
    nth=Pav/Pin
    np=wp/Pin
    no=nth*np

    print '-----'
    print rc
    print rf
    print no
    sheet1.write(i+1,j+1,no)
    sheet1.write(i+1,0,rc)
    sheet1.write(0,j+1,rf)
    j = j+1
i = i+1

```

book.save('Overall Efficiency Matrix.xls')

```
Turbofan Problem 2
raw_input('Press any Key to Exit')
sys.exit(-1)
```