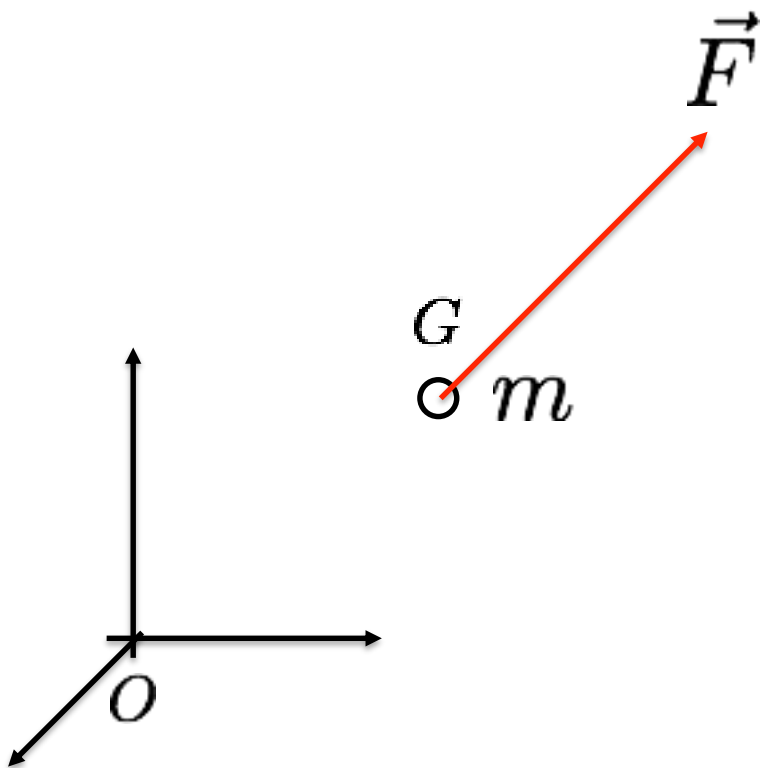


# Robot dynamics for whole body control

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Gabriele Nava  
Stefano Dafarra

# The point mass equations of motion

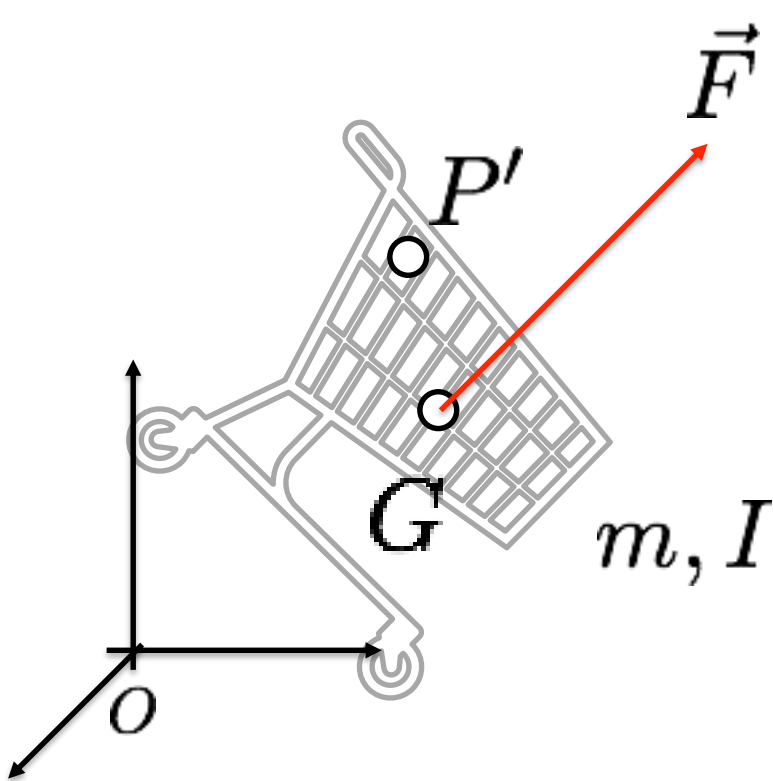


$$m\vec{a} = \vec{F}$$

$$\vec{v} := \frac{d}{dt} \vec{OG}$$

$$\frac{d}{dt}(m\vec{v}) = \vec{F}$$

# The rigid body equations of motion 1/2



$$I = - \int_V \rho S(r)^2 dV$$

$$\vec{r} = P' - G$$

$$\frac{d}{dt} \begin{pmatrix} m\vec{v} \\ I\vec{\omega} \end{pmatrix} = \begin{pmatrix} \vec{F} \\ \vec{M} \end{pmatrix}$$

# The rigid body equations of motion 2/2

$$\frac{d}{dt} \begin{pmatrix} m\vec{v} \\ I\vec{\omega} \end{pmatrix} = \begin{pmatrix} \vec{F} \\ \vec{M} \end{pmatrix} \quad \Rightarrow \quad \frac{d}{dt} \begin{pmatrix} m & 0 \\ 0 & I \end{pmatrix} \begin{pmatrix} \vec{v} \\ \vec{\omega} \end{pmatrix} = \begin{pmatrix} \vec{F} \\ \vec{M} \end{pmatrix}$$

$$\frac{d}{dt} \mathbb{M} \nu = \tau \quad \Rightarrow \quad \mathbb{M} \dot{\nu} + C \nu = \tau$$

$$\mathbb{M} \dot{\nu} + C \nu + g = \tau$$

Forces and torques  
(tau) do not contain  
gravity

# Robot Dynamics

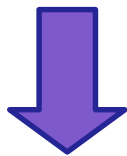


$f = ma$  for the robot?

# Robot Dynamics

$$\frac{d}{dt} \frac{\partial}{\partial \dot{q}} \mathcal{L} - \frac{\partial}{\partial q} \mathcal{L} = \tau$$

- $\mathcal{L} = T - U$ : Lagrangian
- $q, \dot{q}, \ddot{q}$ : joints' positions, velocities, and accelerations
- $\tau$  joint torques



$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) - J^\top F_{ext} = \tau$$

- $M, C, g$ : mass and Coriolis matrices, gravity torques
- $F_{ext}, J$ : **vectorized** external forces and its Jacobian

# Robot Dynamics Terminology

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) - J^\top F_{ext} = \tau$$

## Important Facts

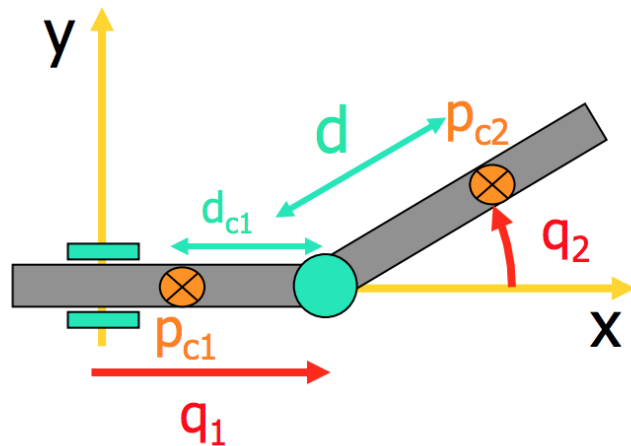
$$\tau = \text{invDyn}(q, \dot{q}, \ddot{q}, F_{ext})$$

Inverse dynamics

$$\ddot{q} = \text{fwDyn}(q, \dot{q}, \tau, F_{ext})$$

Forward dynamics

# Dynamic Model of PR robot

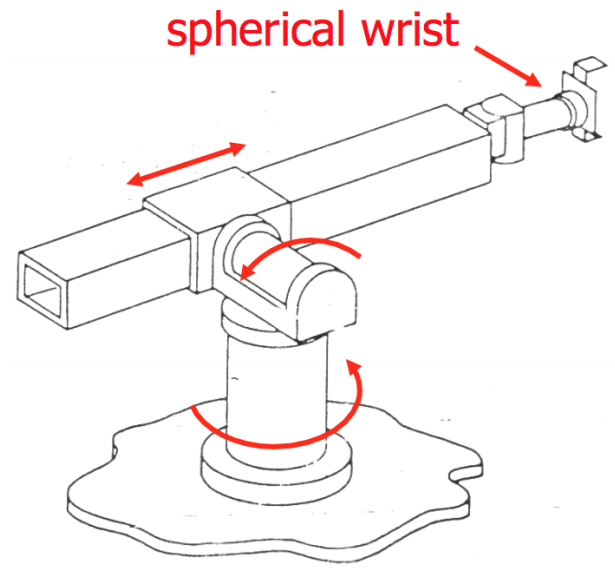


From [http://www.diag.uniroma1.it/~deluca/rob2\\_en/03\\_LagrangianDynamics\\_1.pdf](http://www.diag.uniroma1.it/~deluca/rob2_en/03_LagrangianDynamics_1.pdf), p23

$$\begin{pmatrix} m_1 + m_2 & -m_2 d \sin(q_2) \\ -m_2 d \sin(q_2) & I_{c_2,zz} + m_2 d^2 \end{pmatrix} \begin{pmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{pmatrix} + \begin{pmatrix} -m_2 d \cos(q_2) \dot{q}_2^2 \\ 0 \end{pmatrix} = \begin{pmatrix} \tau_1 \\ \tau_2 \end{pmatrix}$$



# Dynamic Model Complexity 1/2



From

[http://www.diag.uniroma1.it/~deluca/rob2\\_en/04\\_LagrangianDynamics\\_2.pdf](http://www.diag.uniroma1.it/~deluca/rob2_en/04_LagrangianDynamics_2.pdf) p2

$$B_{11} = m_1 k_{122}^2$$

$$\begin{aligned}
 &+ m_2 \left[ k_{211}^2 s^2 \theta_2 + k_{233}^2 c^2 \theta_2 + r_2 (2\ddot{y}_2 + \dot{r}_2) \right] \\
 &+ m_3 \left[ k_{322}^2 s^2 \theta_2 + k_{333}^2 c^2 \theta_2 + r_3 (2\ddot{z}_3 + \dot{r}_3) s^2 \theta_2 + \dot{r}_2^2 \right] \\
 &+ m_4 \left\{ \frac{1}{2} k_{411}^2 \left[ s^2 \theta_2 (2s^2 \theta_4 - 1) + s^2 \theta_4 \right] + \frac{1}{2} k_{422}^2 (1 + c^2 \theta_2 + s^2 \theta_4) \right. \\
 &\quad \left. + \frac{1}{2} k_{433}^2 \left[ s^2 \theta_2 (1 - 2s^2 \theta_4) - s^2 \theta_4 \right] + r_3^2 s^2 \theta_2 + r_2^2 - 2\ddot{y}_4 r_3 s^2 \theta_2 + 2\ddot{z}_4 (r_2 s \theta_4 + r_3 s \theta_2 c \theta_2 c \theta_4) \right\} \\
 &+ m_5 \left\{ \frac{1}{2} (-k_{511}^2 + k_{522}^2 + k_{533}^2) \left[ (s \theta_2 s \theta_5 - c \theta_2 s \theta_4 c \theta_5)^2 + c^2 \theta_4 c^2 \theta_5 \right] \right. \\
 &\quad \left. + \frac{1}{2} (k_{511}^2 - k_{522}^2 + k_{533}^2) (s^2 \theta_4 + c^2 \theta_2 c^2 \theta_4) \right. \\
 &\quad \left. + \frac{1}{2} (k_{511}^2 + k_{522}^2 - k_{533}^2) \left[ (s \theta_2 c \theta_5 + c \theta_2 s \theta_4 s \theta_5)^2 + c^2 \theta_4 s^2 \theta_5 \right] + r_3^2 s^2 \theta_2 + \dot{r}_2^2 \right. \\
 &\quad \left. + 2\ddot{z}_5 \left[ r_3 (s^2 \theta_2 c \theta_5 + s \theta_2 s \theta_4 c \theta_4 s \theta_5) - r_2 c \theta_4 s \theta_5 \right] \right\} \\
 &+ m_6 \left\{ \frac{1}{2} (-k_{611}^2 + k_{622}^2 + k_{633}^2) \left[ (s \theta_2 s \theta_5 c \theta_6 - c \theta_2 s \theta_4 c \theta_5 c \theta_6 - c \theta_2 c \theta_4 s \theta_6)^2 + (c \theta_4 c \theta_5 c \theta_6 - s \theta_4 s \theta_6)^2 \right] \right. \\
 &\quad \left. + \frac{1}{2} (k_{611}^2 - k_{622}^2 + k_{633}^2) \left[ (c \theta_2 s \theta_4 c \theta_5 s \theta_6 - s \theta_2 s \theta_5 s \theta_6 - c \theta_2 c \theta_4 c \theta_6)^2 + (c \theta_4 c \theta_5 s \theta_6 + s \theta_4 c \theta_6)^2 \right] \right. \\
 &\quad \left. + \frac{1}{2} (k_{611}^2 + k_{622}^2 - k_{633}^2) \left[ (c \theta_2 s \theta_4 s \theta_5 + s \theta_2 c \theta_5)^2 + c^2 \theta_4 s^2 \theta_5 \right] \right. \\
 &\quad \left. + \left[ r_3 c \theta_4 s \theta_2 s \theta_6 + (r_3 c \theta_6 + r_2) s \theta_5 \right]^2 + (r_3 c \theta_4 s \theta_5 - r_2)^2 \right\}
 \end{aligned}$$

## Dynamic Model Complexity 2/2

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) = \tau$$

Algorithms to compute dynamic quantities in real time

# Robot Dynamics

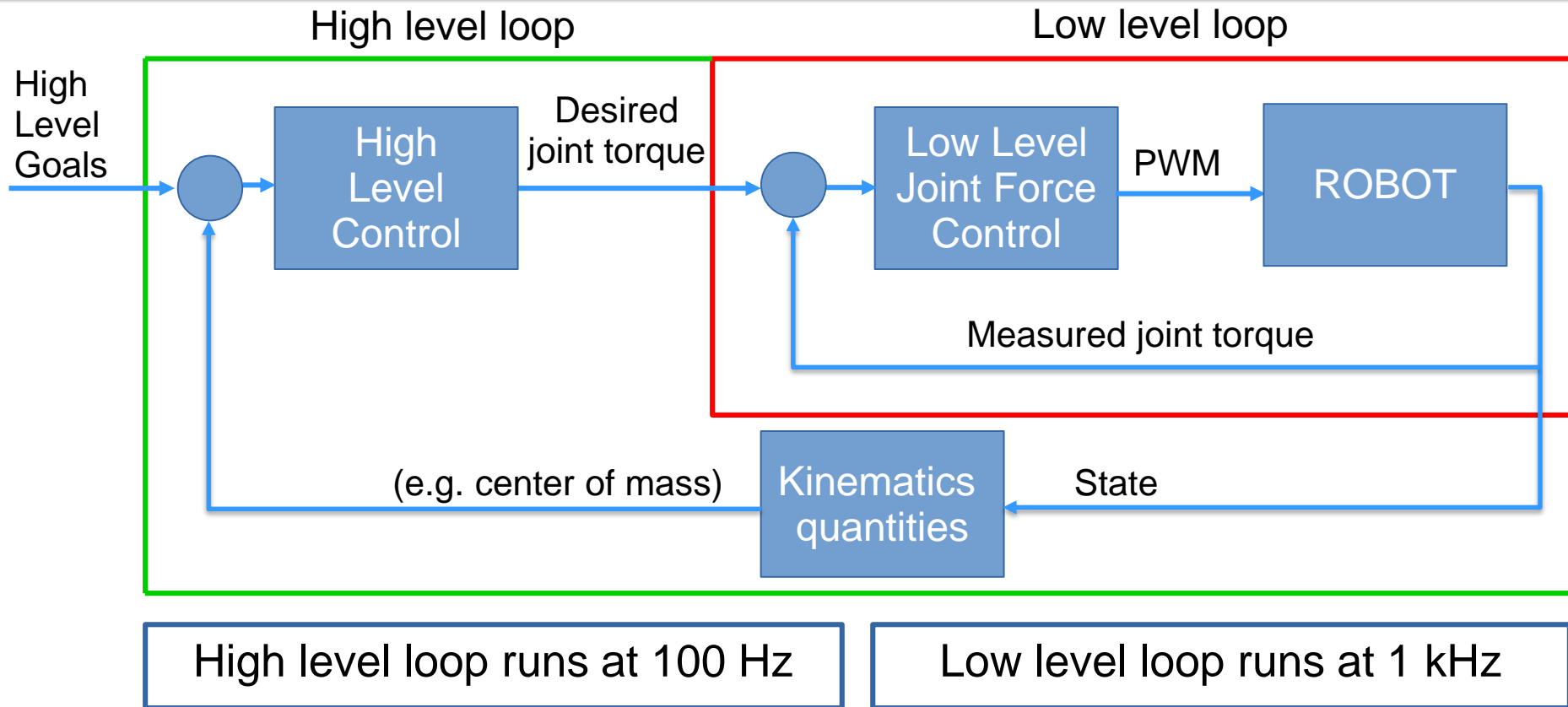
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$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) = \tau$$

Joint torques assumed  $\tau$  as control input

We assume that  $\tau$  can be chosen at will... in reality

# Torque Control Architecture

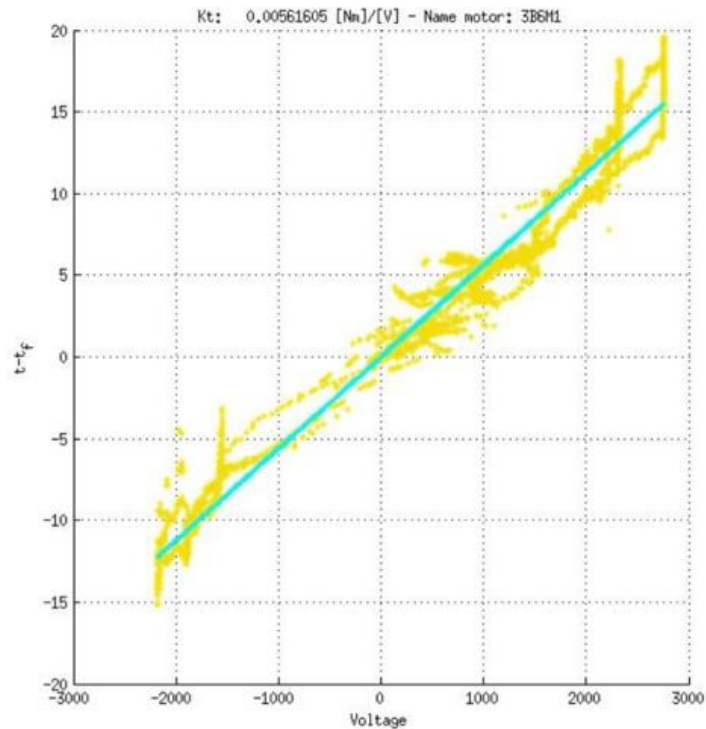
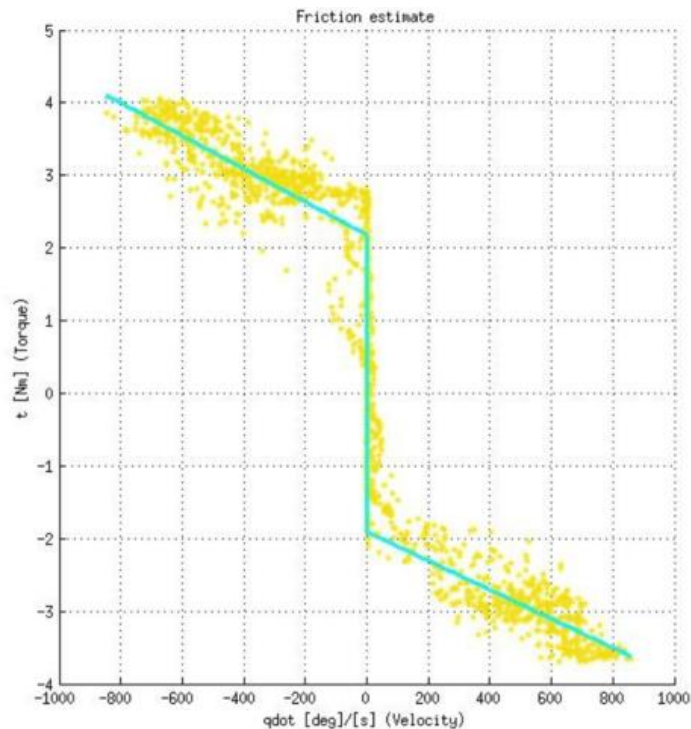


# Torque Control Architecture

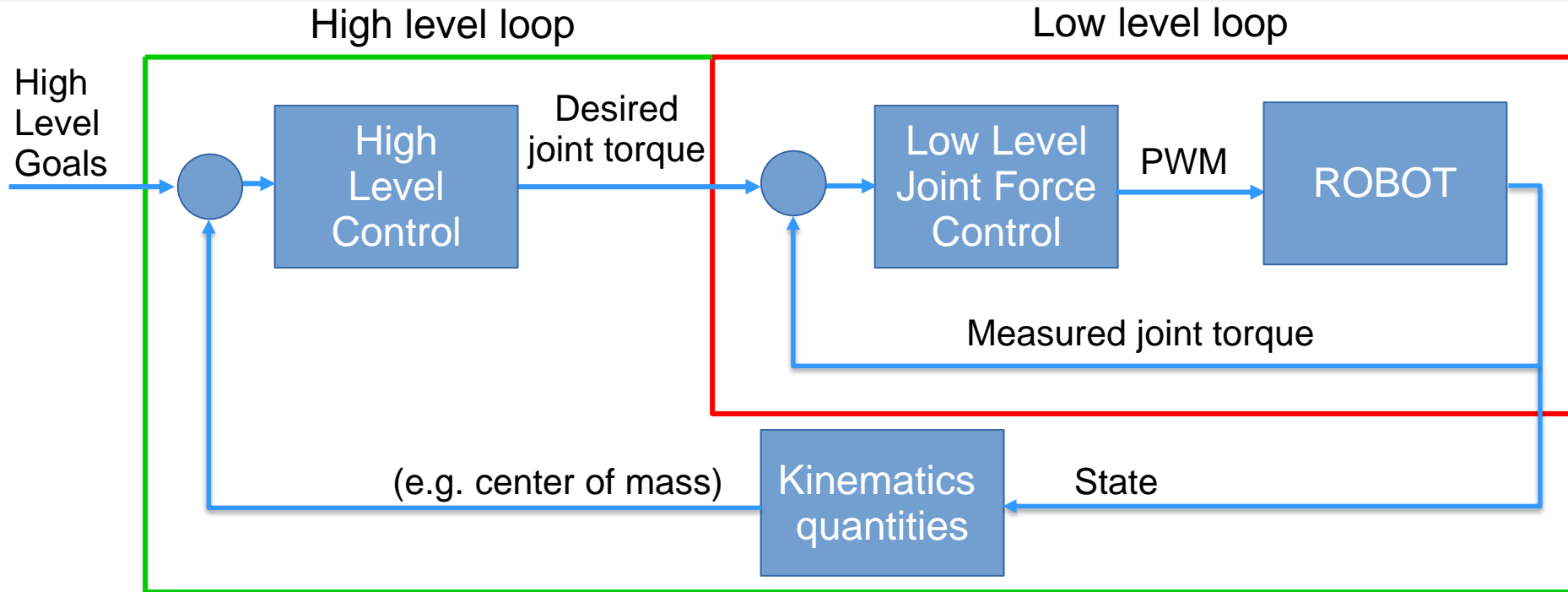
$$\tau = k_{\tau} PWM - k_v \dot{m} - k_c \text{sign}(\dot{m})$$



$$PWM = \bar{k}_t \tau + \bar{k}_v \dot{m} + \bar{k}_c \text{sign}(\dot{m})$$



# Torque Control Architecture



How can we choose the “desired” joint torques?

# Control Objective

---

1) Stabilisation of a desired joint trajectory

$$q_d(t) \in \mathbb{R}^n$$

2) Impose some joint compliance

$$K_p$$

# PD plus gravity compensation control

---

Joint position error:  $q - q_d$

Joint velocity error:  $\dot{q} - \dot{q}_d$

Joint torques ensuring stabilisation of joint trajectory:

$$\tau = M(q)\ddot{q}_d - K_p(q - q_d) - K_d(\dot{q} - \dot{q}_d) + C(q, \dot{q})\dot{q}_d + g(q)$$

Joint stiffness:  $K_p$

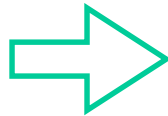


# PD plus gravity compensation control

Fact i):

Stability and convergence of the control law can be proven by using

$$V = \frac{1}{2}(\dot{q} - \dot{q}_d)^\top M(q)(\dot{q} - \dot{q}_d) + \frac{1}{2}(q - q_d)^\top K_p(q - q_d)$$



$$M = M^\top > 0$$

$$\dot{M} - 2C = -(\dot{M} - 2C)^\top$$

$$\dot{V} = -\frac{1}{2}(\dot{q} - \dot{q}_d)^\top K_d(\dot{q} - \dot{q}_d) \leq 0$$

# PD plus gravity compensation control

---

Fact ii):

The control law to stabilise set points, i.e.

$$\dot{q}_d = \ddot{q}_d = 0$$

becomes

$$\tau = g(q) - K_p(q - q_d) - K_d\dot{q}$$

which is simple and does not need Coriolis and mass matrix

# Computed torque control law

---

Joint position error:  $q - q_d$

Joint velocity error:  $\dot{q} - \dot{q}_d$

Joint torques ensuring stabilisation of joint trajectory:

$$\tau = M(q) [\ddot{q}_d - K_p(q - q_d) - K_d(\dot{q} - \dot{q}_d)] + C(q, \dot{q})\dot{q} + g(q)$$

Joint stiffness:  $M(q)K_p$

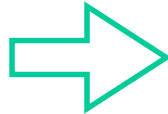
# Computed torque control law

---

Fact i):

Stability and convergence of the control law can be proven by using

$$V = \frac{1}{2}(\dot{q} - \dot{q}_d)^\top (\dot{q} - \dot{q}_d) + \frac{1}{2}(q - q_d)^\top K_p(q - q_d)$$



$$\dot{V} = -\frac{1}{2}(\dot{q} - \dot{q}_d)^\top K_d(\dot{q} - \dot{q}_d) \leq 0$$

# Computed torque control law

Fact ii):

Given the dynamics  $M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) = \tau$

with

$$\tau = M(q) [\ddot{q}_d - K_p(q - q_d) - K_d(\dot{q} - \dot{q}_d)] + C(q, \dot{q})\dot{q} + g(q)$$

gives

$$\ddot{q} = \ddot{q}_d - K_p(q - q_d) - K_d(\dot{q} - \dot{q}_d)$$

Decoupling

# Computed torque control law

---

Fact iii):

The control law to stabilise set points, i.e.

$$\dot{q}_d = \ddot{q}_d = 0$$

becomes

$$\tau = C(q, \dot{q})\dot{q} + g(q) - M(q)K_p(q - q_d) - M(q)K_d\dot{q}$$

which needs Coriolis and mass matrix

# Comparisons of control laws for set points

---

PD plus gravity compensation:  $\tau = g(q) - K_p(q - q_d) - K_d\dot{q}$

## Prons

- needs only gravity
- ensures a constant stiffness
- robust

## Cons

- does not ensure decoupling

---

Computed torque  $\tau = C(q, \dot{q})\dot{q} + g(q) - M(q)K_p(q - q_d) - M(q)K_d\dot{q}$

## Prons

- ensures decoupling

## Cons

- does not ensure constant stiffness
- Requires mass matrix and coriolis terms
- less robust

# What is the difference?

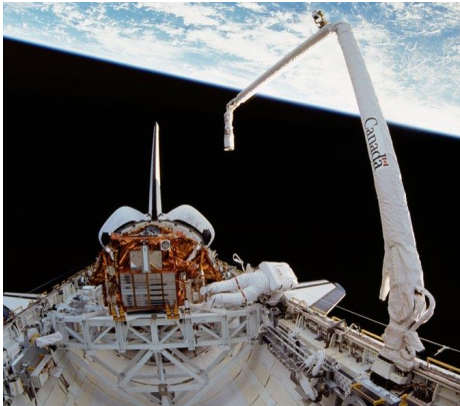




# What is the difference?



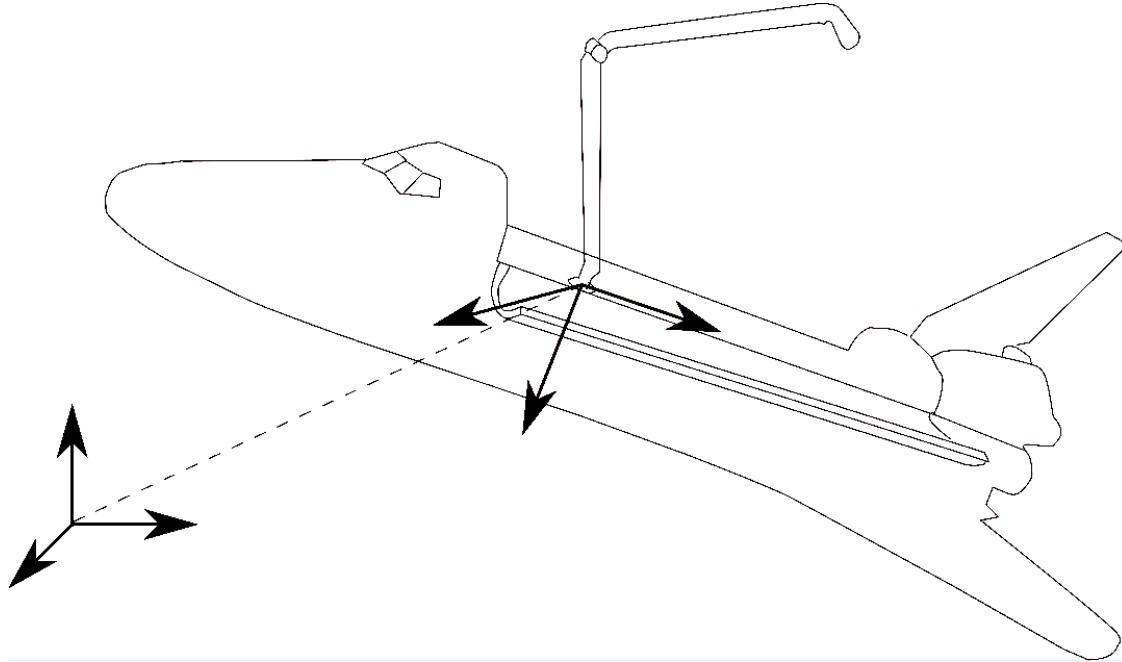
The system is called **fixed-base** if one of the bodies composing it has a constant pose with respect to the inertial frame



The system is called **floating-base** if none of the bodies composing it has a constant pose with respect to the inertial frame

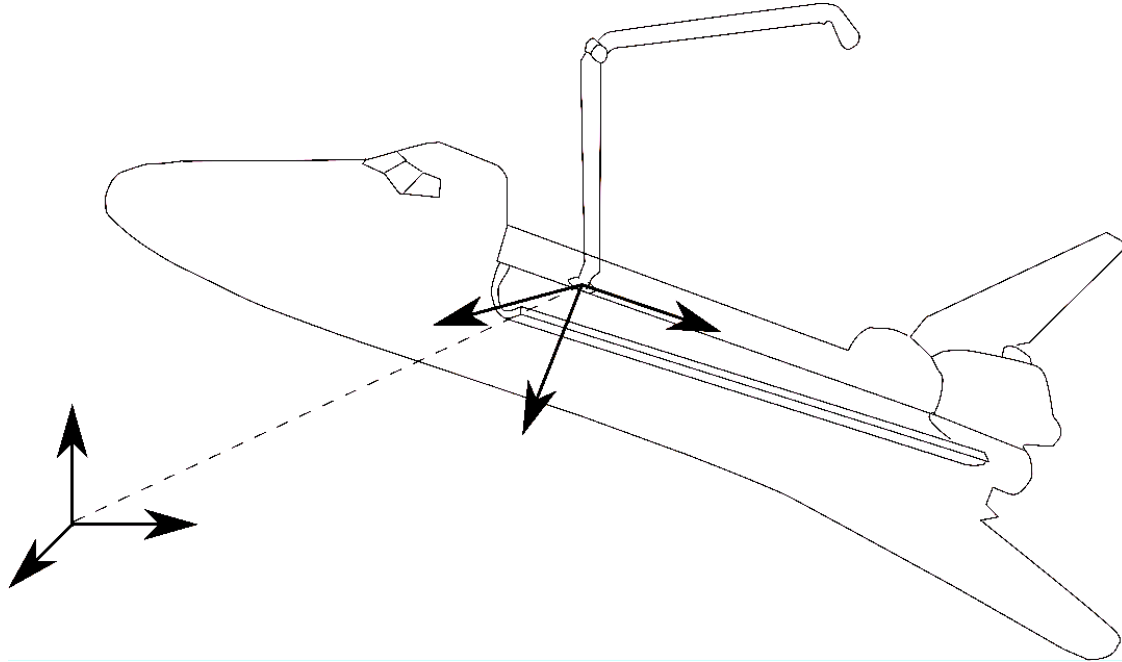
# Extension to floating base systems

Pose of the base frame with respect to the inertial frame must be characterised



# Extension to floating base systems

Additional six degrees of freedom modelled as additional six fictitious joints



# Extension to floating base systems

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) - J^T F_{ext} = \begin{pmatrix} 0_6 \\ \tau \end{pmatrix}$$

- $q = \begin{pmatrix} q_b \\ q_j \end{pmatrix} \quad q_b \in \mathbb{R}^6, \quad q_j \in \mathbb{R}^{DOF}$
- $q_b$ : base's position and orientation
- $q_j$ : joint positions
- $\tau \in \mathbb{R}^{DOF}$
- $J$ : Jacobian,  $F_{ext}$ : vectorized external forces

# Extension to floating base systems

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + g(q) - J^T F_{ext} = \begin{pmatrix} 0_6 \\ \tau \end{pmatrix}$$

Problem:  $q_b \in \mathbb{R}^6$  is the orientation and position of the base frame

$q_b \in \mathbb{R}^6$  is a local representation of  $SE(3) \Rightarrow$

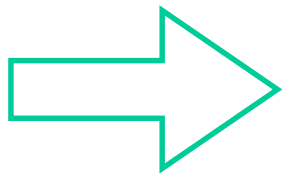
(Forwards) dynamics is not globally defined!

# Extension to floating base systems

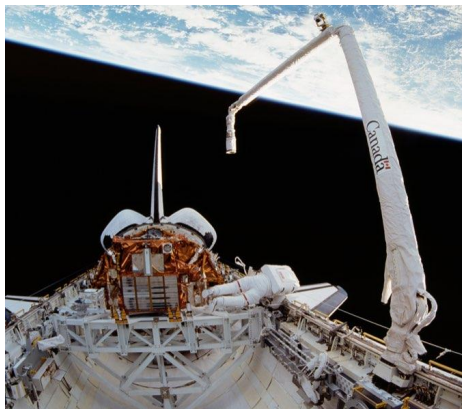


Configuration space:

$$\mathbb{Q} = \mathbb{R}^n$$

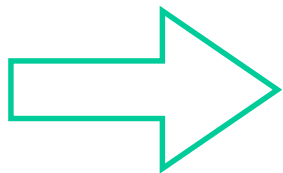


Euler-Lagrange for equations of motion



Configuration space:

$$\mathbb{Q} = SE(3) \times \mathbb{R}^n$$



Euler-Poincaré for equations of motion

# Extension to floating base systems

$$M(q)\dot{\nu} + C(q, \nu)\nu + g(q) - J^T F_{ext} = \begin{pmatrix} 0_6 \\ \tau \end{pmatrix}$$

- $q \in SE(3) \times \mathbb{R}^n$ , e.g.  $q = ({}^wT_b, q_j)$
- $v \in se(3) \times \mathbb{R}^n$ , e.g.  $v = (v_b, \dot{q}_j)$
- ${}^wT_b = (p_b, {}^wR_b)$ : position and rotation matrix of the base
- $q_j$ : joint positions
- $v_b = (\dot{x}_b, \omega_b)$ : linear and angular velocity of base frame