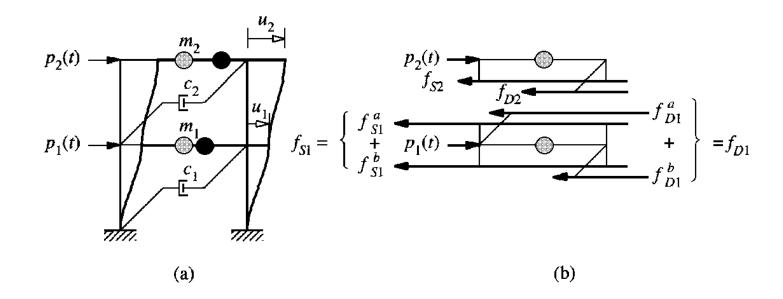
ÇOK SERBESTLİK DERECELİ SİSTEMLER



İki katlı kayma çerçevesi ve kütlelere etkiyen kuvvetler

Newton'un II. Hareket Yasası ve Hareketin denklemi

Her bir kütle için Newton'un hareket yasası yazılırsa;

$$p_j - f_{Sj} - f_{Dj} = m_j \ddot{u}_j$$
 or $m_j \ddot{u}_j + f_{Dj} + f_{Sj} = p_j(t)$ (9.1.1) $j = 1$ and 2,

$$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{Bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{Bmatrix} + \begin{Bmatrix} f_{D1} \\ f_{D2} \end{Bmatrix} + \begin{Bmatrix} f_{S1} \\ f_{S2} \end{Bmatrix} = \begin{Bmatrix} p_1(t) \\ p_2(t) \end{Bmatrix}$$
(9.1.2)

Hareketin denklemleri matris-vektör formunda gösterilebilir.

$$\mathbf{m\ddot{u}} + \mathbf{f}_D + \mathbf{f}_S = \mathbf{p}(t) \tag{9.1.3}$$

$$\mathbf{u} = \left\{ \begin{array}{c} u_1 \\ u_2 \end{array} \right\} \qquad \mathbf{m} = \left[\begin{array}{cc} m_1 & 0 \\ 0 & m_2 \end{array} \right] \qquad \mathbf{f}_D = \left\{ \begin{array}{c} f_{D1} \\ f_{D2} \end{array} \right\} \qquad \mathbf{f}_S = \left\{ \begin{array}{c} f_{S1} \\ f_{S2} \end{array} \right\} \qquad \mathbf{p} = \left\{ \begin{array}{c} p_1 \\ p_2 \end{array} \right\}$$

u: deplasman vektörü (1. ve 2. katın döşemelerinin deplasmanları)

f_D: Sönüm kuvveti vektörü

m: Kütle matrisi

f_S: Elastik kuvvet vektörü (Lineer bir davranış için)

p: Dış yük vektörü

Katlar arası deplasman farkı (göreli kat ötelemesi)

$$\Delta_j = u_j - u_{j-1}.$$

Kat kesme kuvveti

$$V_j = k_j \, \Delta_j \tag{9.1.4}$$

k_i: j. Kat rijitliği

İki ucu ankastre bir kolonun yanal rijitliği

$$k_j = \sum_{\text{columns}} \frac{12EI_c}{h^3} \tag{9.1.5}$$

Herhangi bir kattaki elastik kuvvet o katın altı ve üstündeki elastik kuvvetlerin toplamından oluşur.

$$f_{S1} = f_{S1}^b + f_{S1}^a$$

$$\Delta_1 = u_1$$
 and $\Delta_2 = u_2 - u_1$

$$f_{S1} = k_1 u_1 + k_2 (u_1 - u_2)$$
 (9.1.6a)

$$f_{S2} = k_2(u_2 - u_1) \tag{9.1.6b}$$

$$\begin{cases}
f_{S1} \\
f_{S2}
\end{cases} = \begin{bmatrix}
k_1 + k_2 & -k_2 \\
-k_2 & k_2
\end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} \quad \text{or} \quad \mathbf{f}_S = \mathbf{k}\mathbf{u} \tag{9.1.7}$$

k: rijitlik matrisi

Sönüm kuvveti, j. kat için

$$V_j = c_j \,\dot{\Delta}_j$$

$$f_{D1} = c_1 \dot{u}_1 + c_2 (\dot{u}_1 - \dot{u}_2)$$
 $f_{D2} = c_2 (\dot{u}_2 - \dot{u}_1)$ (9.1.9)

$$\begin{cases}
f_{D1} \\
f_{D2}
\end{cases} = \begin{bmatrix}
c_1 + c_2 & -c_2 \\
-c_2 & c_2
\end{bmatrix} \begin{Bmatrix} \dot{u}_1 \\ \dot{u}_2
\end{cases} \quad \text{or} \quad \mathbf{f}_D = \mathbf{c}\dot{\mathbf{u}} \tag{9.1.10}$$

c: Sönüm matrisi

Hareketin denklemi matris-vektör formunda

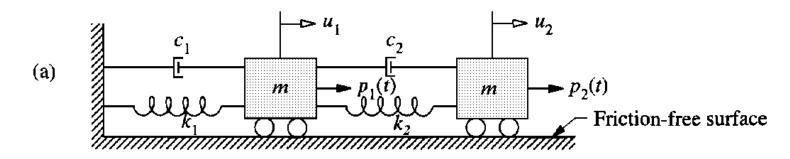
$$\mathbf{m\ddot{u}} + \mathbf{c\dot{u}} + \mathbf{ku} = \mathbf{p}(t) \tag{9.1.11}$$

Dinamik denge (D'Alembert Prensibi)

$$p_{2}(t) = -\frac{f_{12}}{f_{S2}} - \frac{f_{D2}}{f_{D2}} - \frac{f_{I1}}{f_{D1}} - \frac{f_{I1}}{f_{S1}} - f_{D1}$$

Kütle-Yay-Sönümleyici Sistemi

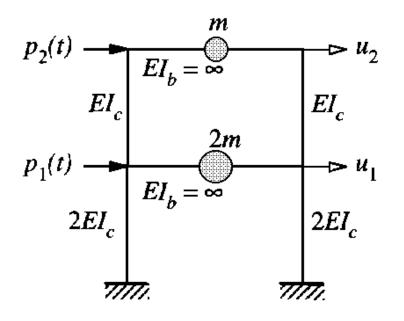
İki serbestli dereceli sistem ve serbest cisim diyagramları



(b)
$$c_{1}\dot{u}_{1} - \frac{m_{1}\ddot{u}_{1}}{m_{1}\ddot{u}_{1}} - p_{1}(t) - p_{2}(t)$$

$$k_{1}u_{1} - \frac{m_{1}\ddot{u}_{1}}{m_{2}\ddot{u}_{2}} - p_{2}(t)$$

ÖRNEK



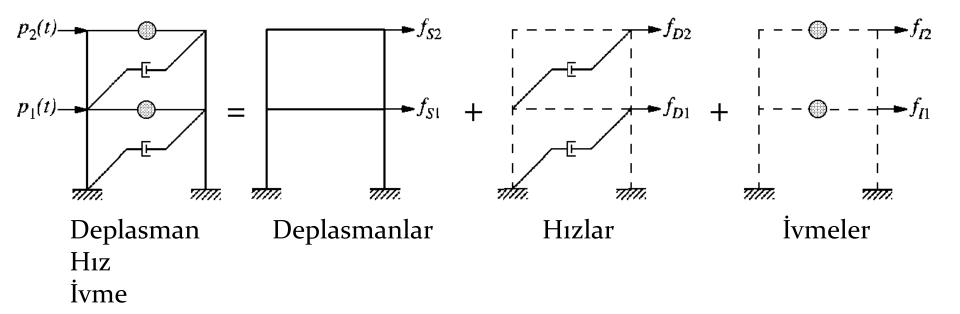
$$m_{1} = 2m m_{2} = m$$

$$k_{1} = 2\frac{12(2EI_{c})}{h^{3}} = \frac{48EI_{c}}{h^{3}} k_{2} = 2\frac{12(EI_{c})}{h^{3}} = \frac{24EI_{c}}{h^{3}}$$

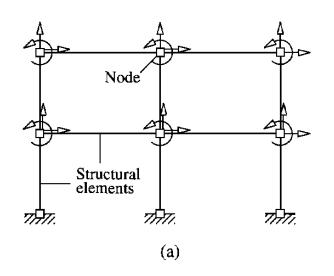
$$\mathbf{m} = m\begin{bmatrix} 2 & 0\\ 0 & 1 \end{bmatrix} \mathbf{k} = \frac{24EI_{c}}{h^{3}} \begin{bmatrix} 3 & -1\\ -1 & 1 \end{bmatrix}$$

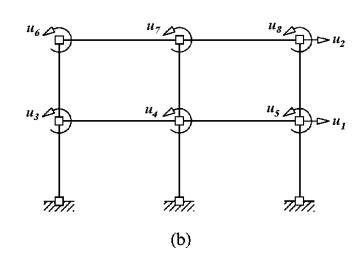
$$m\begin{bmatrix}2 & 0\\0 & 1\end{bmatrix}\begin{bmatrix}\ddot{u}_1\\\ddot{u}_2\end{bmatrix} + 24\frac{EI_c}{h^3}\begin{bmatrix}3 & -1\\-1 & 1\end{bmatrix}\begin{bmatrix}u_1\\u_2\end{bmatrix} = \begin{bmatrix}p_1(t)\\p_2(t)\end{bmatrix}$$

Rijitlik , Sönüm ve Kütle kuvvetlerinin bileşenleri



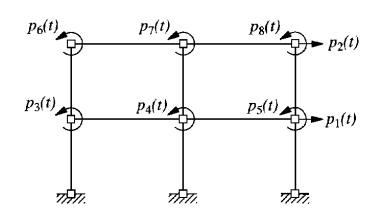
Lineer Sistemlere Genel Yaklaşım





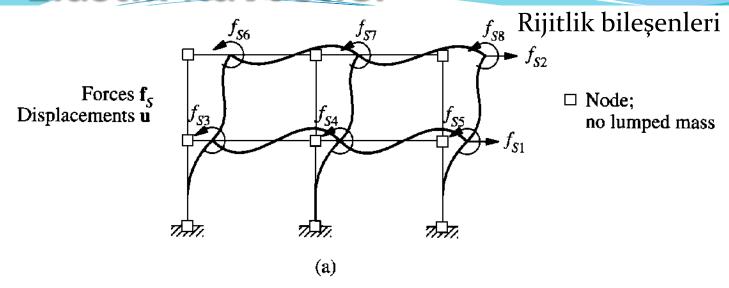
N serbestlik dereceli sistem

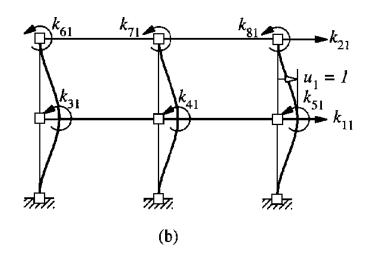
Eksenel deformasyonlar ihmal edildiğinde

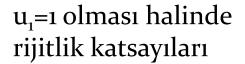


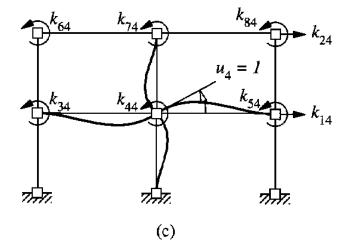
Dış dinamik kuvvetler

Elastik Kuvvetler









u₄=1 olması halinde rijitlik katsayıları

$$f_{Si} = k_{i1}u_1 + k_{i2}u_2 + \dots + k_{ij}u_j + \dots + k_{iN}u_N$$
 (9.2.1)

$$i=1$$
 to N .

$$\begin{bmatrix} f_{S1} \\ f_{S2} \\ \vdots \\ f_{SN} \end{bmatrix} = \begin{bmatrix} k_{11} & k_{12} & \cdots & k_{1j} & \cdots & k_{1N} \\ k_{21} & k_{22} & \cdots & k_{2j} & \cdots & k_{2N} \\ \vdots & \vdots & & \vdots & & \vdots \\ k_{N1} & k_{N2} & \cdots & k_{Nj} & \cdots & k_{NN} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_N \end{bmatrix}$$
(9.2.2)

$$\mathbf{f}_S = \mathbf{k}\mathbf{u} \tag{9.2.3}$$

Sönüm Kuvvetleri

$$f_{Di} = c_{i1}\dot{u}_{1} + c_{i2}\dot{u}_{2} + \dots + c_{ij}\dot{u}_{j} + \dots + c_{iN}\dot{u}_{N}$$

$$i = 1 \text{ to } N$$

$$\begin{bmatrix} f_{D1} \\ f_{D2} \\ \vdots \\ f_{DN} \end{bmatrix} = \begin{bmatrix} c_{11} & c_{12} & \cdots & c_{1j} & \cdots & c_{1N} \\ c_{21} & c_{22} & \cdots & c_{2j} & \cdots & c_{2N} \\ \vdots & \vdots & & \vdots & & \vdots \\ c_{N1} & c_{N2} & \cdots & c_{Nj} & \cdots & c_{NN} \end{bmatrix} \begin{bmatrix} \dot{u}_{1} \\ \dot{u}_{2} \\ \vdots \\ \dot{u}_{N} \end{bmatrix}$$

$$(9.2.5)$$

$$\mathbf{f}_D = \mathbf{c}\dot{\mathbf{u}} \tag{9.2.6}$$

Sönüm Kuvvetleri

Forces \mathbf{f}_D Velocities $\dot{\mathbf{u}}$

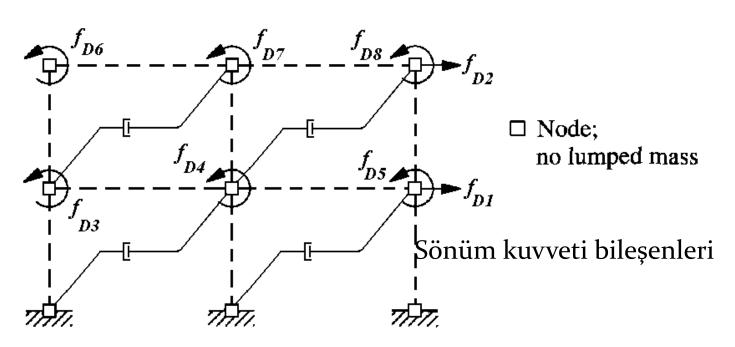
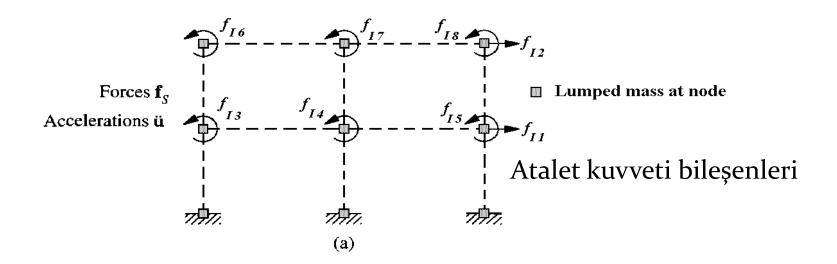
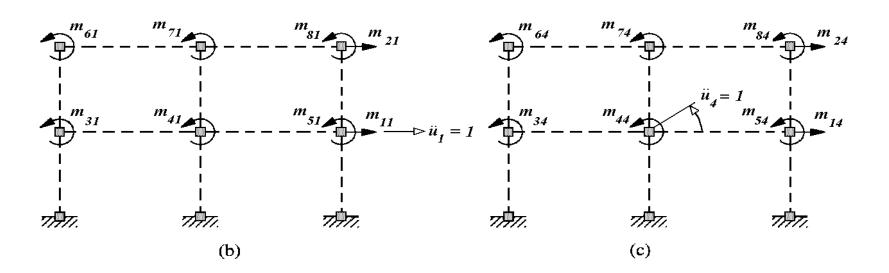


Figure 9.2.4 Damping component of frame.

Atalet Kuvvetleri





Atalet Kuvvetleri

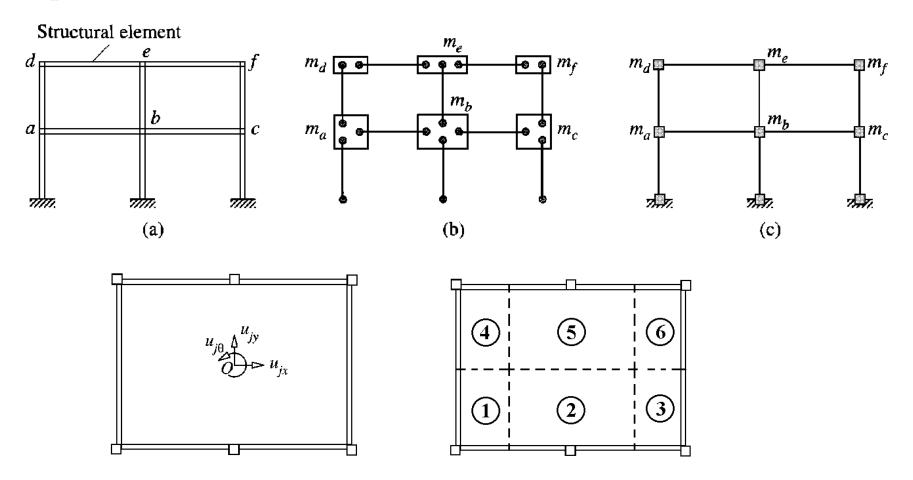
$$f_{Ii} = m_{i1}\ddot{u}_1 + m_{i2}\ddot{u}_2 + \dots + m_{ij}\ddot{u}_j + \dots + m_{iN}\ddot{u}_N$$
 (9.2.7)

i=1 to N.

$$\begin{bmatrix} f_{I1} \\ f_{I2} \\ \vdots \\ f_{IN} \end{bmatrix} = \begin{bmatrix} m_{11} & m_{12} & \cdots & m_{1j} & \cdots & m_{1N} \\ m_{21} & m_{22} & \cdots & m_{2j} & \cdots & m_{2N} \\ \vdots & \vdots & & \vdots & & \vdots \\ m_{N1} & m_{N2} & \cdots & m_{Nj} & \cdots & m_{NN} \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \\ \vdots \\ \ddot{u}_N \end{bmatrix}$$
(9.2.8)

$$\mathbf{f}_I = \mathbf{m}\ddot{\mathbf{u}} \tag{9.2.9}$$

Kütlelerin Düğümlerde Toplanması Toplu Kütleli Sistemler



Genelde, kütle matrisi diyagonaldir.

$$m_{ij} = 0$$
 $i \neq j$ $m_{jj} = m_j$ or 0 (9.2.10)

Örnek: Aşağıdaki sistemde uı ve u2 deplasmanlarına göre hareketin denklemini çıkarınız.

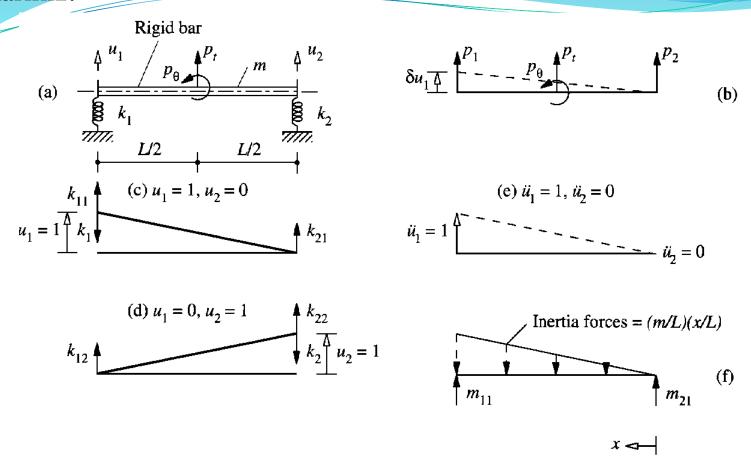


Figure E9.2

Kuvvetlerin Hesabı

virtual displacement δu_1 along DOF 1.

$$\delta W = p_t \frac{\delta u_1}{2} - p_\theta \frac{\delta u_1}{L} \tag{a}$$

$$\delta W = p_1 \delta u_1 + p_2(0) \tag{b}$$

$$p_1 = \frac{p_t}{2} - \frac{p_\theta}{L} \tag{c}$$

virtual displacement δu_2

$$p_2 = \frac{p_t}{2} + \frac{p_\theta}{L} \tag{d}$$

Rijitlik Matrisinin Hesabı

$$u_1 = 1 \text{ with } u_2 = 0$$

 $u_1 = 1$ with $u_2 = 0$ k_{11} and k_{21} $k_{11} = k_1$ and $k_{21} = 0$.

$$u_1 = 1$$
 k_{11} k_{21} k_{21} k_{21}

$$u_2 = 1$$
 $u_1 = 0$

$$k_{12} = 0$$
 and $k_{22} = k_2$.

(d)
$$u_1 = 0, u_2 = 1$$

$$k_{22}$$

$$k_2 \downarrow u_2 = 1$$

$$k = \begin{bmatrix} k_1 & 0 \\ 0 & k_2 \end{bmatrix}$$

Kütle Matrisinin Hesabı

Birim ivme

$$\ddot{u}_1 = 1$$
 with $\ddot{u}_2 = 0$,

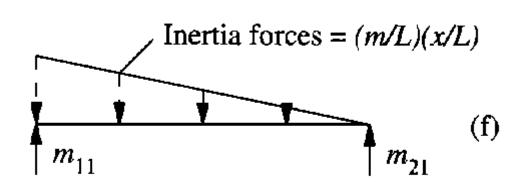
$$m_{11} = m/3$$
 and $m_{21} = m/6$ $\ddot{u}_1 = 1$

(e) $\ddot{u}_1 = 1$, $\ddot{u}_2 = 0$

$$\ddot{u}_2 = 1$$
 with $\ddot{u}_1 = 0$,

$$m_{12} = m/6$$
 and $m_{22} = m/3$.

$$\mathbf{m} = \frac{m}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$



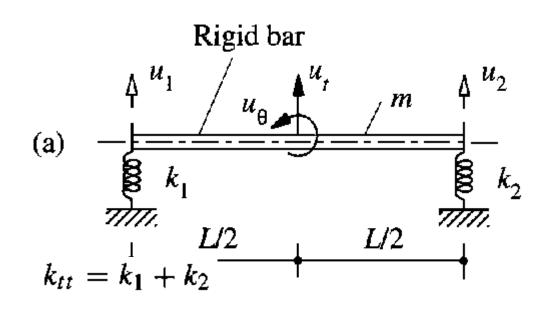
$$x \triangleleft$$

Hareketin Denklemi (Kütle girişimli iki diferansiyel denklemin matris formu)

$$\frac{m}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{bmatrix} + \begin{bmatrix} k_1 & 0 \\ 0 & k_2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} (p_t/2) - (p_\theta/L) \\ (p_t/2) + (p_\theta/L) \end{bmatrix}$$

İki dif. Denklem girişimlidir. Çubuğun yayılı kütlesi serbestliklere yığılı olmadığından köşegen dışı terimlerin işaret ettiği gibi kütle matrisi girişimli olmasından kaynaklıdır.

Örnek: Aşağıdaki şekilde hareketin denklemini u_t ve u_θ deplasmanlarına göre çıkartınız.

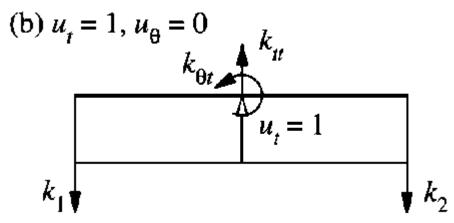


Rijitlik matrisinin hesabı

$$k_{tt}=k_1+k_2$$

$$k_{\theta t} = (k_2 - k_1)L/2.$$

$$u_t = 1$$
 with $u_\theta = 0$



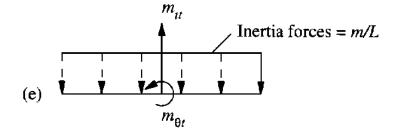
Rijitlik matrisinin hesabı

$$u_{\theta} = 1 \text{ with } u_{t} = 0$$
 (c) $u_{t} = 0, u_{\theta} = 1$
 $k_{t\theta} = (k_{2} - k_{1})L/2$
 $k_{\theta\theta} = (k_{1} + k_{2})L^{2}/4.$ $k_{1}L/2$

$$\bar{\mathbf{k}} = \begin{bmatrix} k_1 + k_2 & (k_2 + k_1)L/2 \\ (k_2 - k_1)L/2 & (k_1 + k_2)L^2/4 \end{bmatrix}$$

Kütle matrisinin hesabı

$$\ddot{u}_t = 1$$
 with $\ddot{u}_{\theta} = 0$, $m_{tt} = m$ and $m_{\theta t} = 0$.



Kütle matrisinin hesabı

$$\ddot{u}_{\theta} = 1$$
 with $\ddot{u}_t = 0$,

$$m_{\theta\theta} = mL^2/12.$$

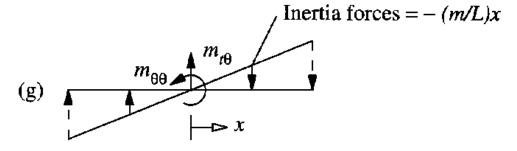
$$m_{\theta\theta} = I_O$$

$$m_{t\theta}=0$$

$$\bar{\mathbf{m}} = \begin{bmatrix} m & 0 \\ 0 & mL^2/12 \end{bmatrix}$$

(f)
$$\ddot{u}_{t} = 0$$
, $\ddot{u}_{\theta} = 1$

$$- \sqrt{\ddot{u}_{\theta}} = 1$$



Hareketin denklemi (Rijitlik matrisi girişimli)

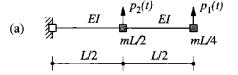
$$\mathbf{u} = \langle u_t \quad u_\theta \rangle^T, \, \mathbf{p} = \langle p_t \quad p_\theta \rangle^T$$

$$\begin{bmatrix} m & 0 \\ 0 & mL^2/12 \end{bmatrix} \begin{Bmatrix} \ddot{u}_t \\ \ddot{u}_{\theta} \end{Bmatrix} + \begin{bmatrix} k_1 + k_2 & (k_2 - k_1)L/2 \\ (k_2 - k_1)L/2 & (k_1 + k_2)L^2/4 \end{bmatrix} \begin{Bmatrix} u_t \\ u_{\theta} \end{Bmatrix} = \begin{Bmatrix} p_t \\ p_{\theta} \end{Bmatrix}$$

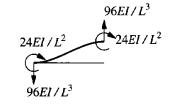
Örnek: Şekildeki L uzunluğundaki kütlesiz EI rijitliğindeki çubukta kütleler görüldüğü gibi iki noktada toplanmıştır. Bu noktalara uygulanan kuvvetler Pı(t) ve P2(t) olduğuna göre hareketin denklemini kurunuz. Eksenel ve kayma deformasyonları ihmal edilmektedir.

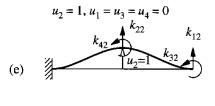
Kütle matrisi

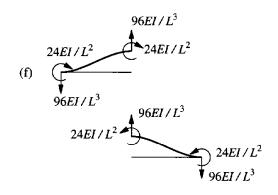
$$\mathbf{m} = \begin{bmatrix} mL/4 & & & \\ & mL/2 & & \\ & & 0 & \\ & & & 0 \end{bmatrix}$$

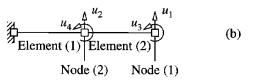


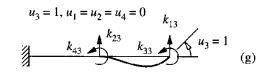
$$u_1 = 1, u_2 = u_3 = u_4 = 0$$
 k_{11}
 k_{21}
 k_{31}
 $u_1 = 1$

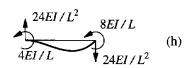


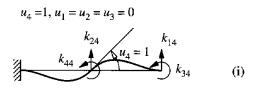


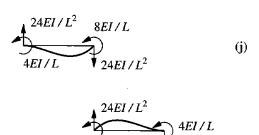












Rijitlik matrisi

$$\mathbf{k} = \frac{8EI}{L^3} \begin{bmatrix} 12 & -12 & -3L & -3L \\ -12 & 24 & 3L & 0 \\ -3L & 3L & L^2 & L^2/2 \\ -3L & 0 & L^2/2 & 2L^2 \end{bmatrix}$$

$$\mathbf{m\ddot{u}} + \mathbf{ku} = \mathbf{p}(t)$$

$$\mathbf{u} = \langle u_1 \quad u_2 \quad u_3 \quad u_4 \rangle^T$$

$$\mathbf{p}(t) = \langle p_1(t) \quad p_2(t) \quad 0 \quad 0 \rangle^T.$$

Örnek: Hareketin denklemini sadece uı ve u2 serbestlik derecelerine göre çıkartınız. (Rijitlik katsayıları yerine fleksibilite katsayılarını kullanarak) Fleksibilite matrisinin diyogonal olmayan elemanları eşittir. Tersi alınarak k rijitlik matrisi bulunur.

Fleksibilite matrisi

(a)
$$EI \qquad P_{2}(t) \qquad p_{1}(t)$$

$$mL/2 \qquad mL/4$$

$$L/2 \qquad L/2 \qquad L/2$$

$$f_{S1} = 1, f_{S2} = 0$$

$$\hat{f}_{21}$$

$$\hat{f}_{11}$$
(c)

$$f_{S2} = 1, f_{S1} = 0$$

$$f_{S2} = 1$$

$$f_{12} \text{ (d)}$$

$$\hat{f}_{22}$$

$$\hat{\mathbf{f}} = \frac{L^3}{48EI} \begin{bmatrix} 16 & 5\\ 5 & 2 \end{bmatrix}$$

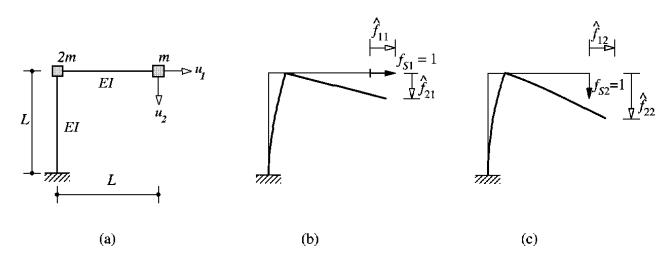
$$\mathbf{k} = \frac{48EI}{7L^3} \begin{bmatrix} 2 & -5 \\ -5 & 16 \end{bmatrix}$$

$$\mathbf{m} = \begin{bmatrix} mL/4 & \\ & mL/2 \end{bmatrix}$$

Hareketin Denklemi

$$\begin{bmatrix} mL/4 & \\ & mL/2 \end{bmatrix} \begin{bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{bmatrix} + \frac{48EI}{7L^3} \begin{bmatrix} 2 & -5 \\ -5 & 16 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} p_1(t) \\ p_2(t) \end{bmatrix}$$

Örnek: Şekildeki çerçevenin hareket denklemini kurunuz

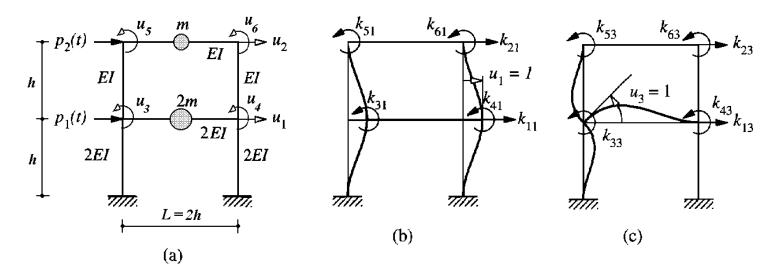


$$\ddot{u}_1 = 1 \text{ is } 2m + m = 3m$$

$$\mathbf{m} = \begin{bmatrix} 3m \\ m \end{bmatrix} \qquad \hat{\mathbf{f}} = \frac{L^3}{6EI} \begin{bmatrix} 2 & 3 \\ 3 & 8 \end{bmatrix} \qquad \mathbf{k} = \frac{6EI}{7L^3} \begin{bmatrix} 8 & -3 \\ -3 & 2 \end{bmatrix}$$

$$\begin{bmatrix} 3m & \\ & u_1 \end{bmatrix} \begin{Bmatrix} \ddot{u}_1 \\ \ddot{u}_2 \end{Bmatrix} + \frac{6EI}{7L^3} \begin{bmatrix} 8 & -3 \\ -3 & 2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

Örnek: Şekildeki çerçevenin hareket denklemini kurunuz



$$\mathbf{u} = \begin{pmatrix} u_1 & u_2 & u_3 & u_4 & u_5 & u_6 \end{pmatrix}^T$$

$$\mathbf{m} = m \begin{bmatrix} 2 & & & & \\ & 1 & & & \\ & & 0 & & \\ & & & 0 & \\ & & & 0 & \\ & & & 0 & \\ & & & 0 & \\ \end{bmatrix} \qquad \mathbf{k} = \frac{EI}{h^3} \begin{bmatrix} 72 & -24 & 6h & 6h & -6h & -6h \\ -24 & 24 & 6h & 6h & 6h & 6h \\ 6h & 6h & 16h^2 & 2h^2 & 2h^2 & 0 \\ 6h & 6h & 2h^2 & 16h^2 & 0 & 2h^2 \\ -6h & 6h & 2h^2 & 0 & 6h^2 & h^2 \\ -6h & 6h & 0 & 2h^2 & h^2 & 6h^2 \end{bmatrix}$$

$$\mathbf{p}(t) = \langle p_1(t) \quad p_2(t) \quad 0 \quad 0 \quad 0 \quad 0 \rangle^T \qquad \mathbf{m}\ddot{\mathbf{u}} + \mathbf{k}\mathbf{u} = \mathbf{p}(t)$$

Doğal Titreşim Frekansları ve Modlar

Sönümsüz Serbest Titreşim

(a)

$$\mathbf{m\ddot{u}} + \mathbf{ku} = 0$$

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(d)

Sönümsüz sistemin rastgele başlangıç deplasmanları sonrası serbest titreşimleri

$$T_n = \frac{2\pi}{\omega_n}$$
 $f_n = \frac{1}{T_n}$ $(n = 1, 2)$ $\phi_n = \langle \dot{\phi}_{1n} \quad \phi_{2n} \rangle^{\ddot{T}}$.

Doğal açısal $\omega_1 < \omega_2$ frekanslar

$$\mathbf{u}(t) = q_n(t)\boldsymbol{\phi}_n$$

: ϕ_n does not vary with time.

$$q_n(t) = A_n \cos \omega_n t + B_n \sin \omega_n t$$
 simple harmonic function

$$\mathbf{u}(t) = \phi_n \left(A_n \cos \omega_n t + B_n \sin \omega_n t \right)$$

 ω_n and ϕ_n are unknown.

$$\left[-\omega_n^2 \mathbf{m} \phi_n + \mathbf{k} \phi_n\right] q_n(t) = \mathbf{0}$$

$$\mathbf{k}\boldsymbol{\phi}_n = \omega_n^2 \mathbf{m} \boldsymbol{\phi}_n$$

Özdeğer - Özvektör problemi

$$\left[\mathbf{k} - \omega_n^2 \mathbf{m}\right] \phi_n = \mathbf{0}$$

$$\det\left[\mathbf{k} - \omega_n^2 \mathbf{m}\right] = 0$$

$$\omega_n \quad (n=1, 2)$$

TRANSFORMATION OF ${
m k}\phi=\omega^2{
m m}\phi$ TO THE STANDARD FORM

$$\mathbf{k}\boldsymbol{\phi} = \lambda \mathbf{m}\boldsymbol{\phi}$$

$$\lambda_n \equiv \omega_n^2$$

$$p(\lambda) = \det(\mathbf{k} - \lambda \mathbf{m}) = 0$$

$$\mathbf{A}\mathbf{y} = \lambda \mathbf{y}$$

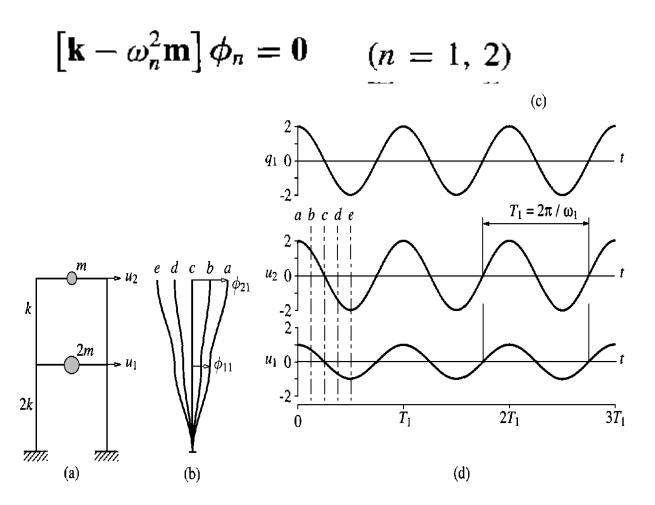
$$\mathbf{k}\boldsymbol{\phi} = \omega^2 \mathbf{m}\boldsymbol{\phi}$$

Standart özdeğer $\mathbf{A}\phi = \lambda \phi$ özvektör problemi

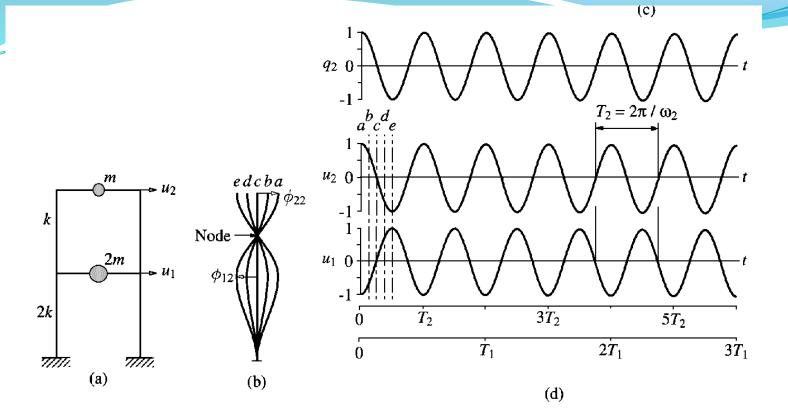
$$\mathbf{A} = \mathbf{m}^{-1}\mathbf{k} \qquad \lambda = \omega^2$$

Determinant alınınca ortaya çıkan denkleme karakteristik denklem denir. Denklemin köklerinden özdeğerlere ve oradan doğal açısal frekanslara ulaşılır.

Her bir frekans, Özdeğer-özvektör Denkleminde yerine yazılarak, onlara karşı gelen mod vektörleri bulunur.



İki katlı kayma çerçevesi, titreşimin ilk modu, a,b,c,d,e zamanlarında deplasman şekli, modal koordinat deplasmanının değişimi



İki katlı kayma çerçevesi, titreşimin ikinci modu, a,b,c,d,e zamanlarında deplasman şekli, modal koordinat deplasmanının değişimi

Mod ve Spektrum Matrisleri

$$\mathbf{\Phi} = [\phi_{jn}] = \begin{bmatrix} \phi_{11} & \phi_{12} & \cdots & \phi_{1N} \\ \phi_{21} & \phi_{22} & \cdots & \phi_{2N} \\ \vdots & \vdots & \ddots & \vdots \\ \phi_{N1} & \phi_{N2} & \cdots & \phi_{NN} \end{bmatrix}$$

$$\mathbf{\Omega}^2 = \begin{bmatrix} \omega_1^2 & & & \\ & \omega_2^2 & & \\ & & \ddots & \\ & & & \omega_N^2 \end{bmatrix}$$

$$\mathbf{k}\boldsymbol{\phi}_n = \mathbf{m}\boldsymbol{\phi}_n\omega_n^2$$

$$\mathbf{k}\mathbf{\Phi} = \mathbf{m}\mathbf{\Phi}\mathbf{\Omega}^2$$

Bu denklem özdeğer ve özvektörler arasındaki ilişkilerin bir özet gösterimidir.

Modların Dikliği

$$\omega_n \neq \omega_r$$
,

$$\phi_n^T \mathbf{k} \phi_r = 0 \qquad \phi_n^T \mathbf{m} \phi_r = 0$$

$$\phi_r^T \mathbf{k} \phi_r = \omega_n^2 \phi_r^T \mathbf{m} \phi_n$$

veya

$$\boldsymbol{\phi}_n^T \mathbf{k} \boldsymbol{\phi}_r = \boldsymbol{\omega}_r^2 \boldsymbol{\phi}_n^T \mathbf{m} \boldsymbol{\phi}_r$$

Bu denklemin sol tarafının transpozu sağ tarafın transpozuna eşittir.

$$\boldsymbol{\phi}_n^T \mathbf{k} \boldsymbol{\phi}_r = \omega_n^2 \boldsymbol{\phi}_n^T \mathbf{m} \boldsymbol{\phi}_r^T$$

Kütle ve rijitlik matrislerinin simetri özelliğinden

$$(\omega_n^2 - \omega_r^2) \boldsymbol{\phi}_n^T \mathbf{m} \boldsymbol{\phi}_r = 0$$

 $\omega_n^2 \neq \omega_r^2$ olması halinde yukarıda verilen ikinci denklem geçerlidir. Onun sıfır olması halinde birinci denklemde geçerlidir.

Doğal modların dikliği aşağıdaki kare matrislerin köşegen olmasına sebep olur.

$$\mathbf{K} \equiv \mathbf{\Phi}^T \mathbf{k} \mathbf{\Phi} \qquad \mathbf{M} \equiv \mathbf{\Phi}^T \mathbf{m} \mathbf{\Phi}$$

Diyagonal elemanlar

$$K_n = \boldsymbol{\phi}_n^T \mathbf{k} \boldsymbol{\phi}_n \qquad \qquad M_n = \boldsymbol{\phi}_n^T \mathbf{m} \boldsymbol{\phi}_n$$

m and k are positive definite,

K and **M** Diyagonal elemanları pozitif

Pozitif tanımlı

Simetrik pozitif tanımlı matris: Simetrik bir A matrisi ($A=A^T$) ile elemanlarının en az biri sıfırdan farklı olan, bunun dışında tamamen keyfi bir x \neq 0 kolon vektörü verilmiş olsun. $P = x^T A x$ çarpımı sabit bir sayı olur. Eğer P>0 ise A pozitif tanımlıdır (positive definite)

$$K_n = \omega_n^2 M_n$$
İspatı
$$K_n = \phi_n^T (\omega_n^2 \mathbf{m} \phi_n) = \omega_n^2 (\phi_n^T \mathbf{m} \phi_n) = \omega_n^2 M_n$$

Modların Normalizasyonu

Modları mod vektörünün içindeki bir değere göre normalize etmek, mod vektörünün elemanlarını standardize etmek için kullanılır. Bir tanesi birim değer olarak seçilir, diğerleri onun oranına göre bulunur. Bu durumda; genellikle her Mn değeri 1 olacak şekilde bir ölçeklendirme yapılır.

$$M_n = \phi_n^T \mathbf{m} \phi_n = 1$$
 $\Phi^T \mathbf{m} \Phi = \mathbf{I}$ \mathbf{I} Birim matris

Yukarıdaki denkleme göre modlar hem diktir, hem de \mathbf{m} 'ye göre boyutlandırılmışlardır. Bu durum da K_n değeri açısal frekansın karesine eşit olurken, \mathbf{K} matrisi aşağıdaki gibi açısal frekansların karelerinden oluşan bir diyagonal matris olur.

$$K_n = \phi_n^T \mathbf{k} \phi_n = \omega_n^2 M_n = \omega_n^2 \qquad \mathbf{K} = \mathbf{\Phi}^T \mathbf{k} \mathbf{\Phi} = \mathbf{\Omega}^2$$

Örnek: Şekildeki sistemin her iki durum için, frekanslarını hesaplayınız ve mod şekillerini çiziniz

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 &$$

$$\mathbf{k} - \omega_n^2 \mathbf{m} = \begin{bmatrix} k - m\omega_n^2/3 & -m\omega_n^2/6 \\ -m\omega_n^2/6 & 2k - m\omega_n^2/3 \end{bmatrix}$$
$$m^2 \omega_n^4 - 12km\omega_n^2 + 24k^2 = 0$$

$$\omega_1^2 = (6 - 2\sqrt{3}) \frac{k}{m} = 2.536 \frac{k}{m}$$
 $\omega_2^2 = (6 + 2\sqrt{3}) \frac{k}{m} = 9.464 \frac{k}{m}$

$$\omega_n^2 = \omega_1^2 =$$

$$k \begin{bmatrix} 0.155 & -0.423 \\ -0.423 & 1.165 \end{bmatrix} \begin{Bmatrix} \phi_{11} \\ \phi_{21} \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

Şimdi herhangi bir değeri bilinmeven olarak şeçelim. Bildiğimize **1** değeri verelim $\phi_{11} = 1$.

Birinci veya ikinci denklemden $\phi_{21} = 0.366$. buluruz.

$$\omega_n^2 = \omega_2^2$$
- 155 -15771 (dua)

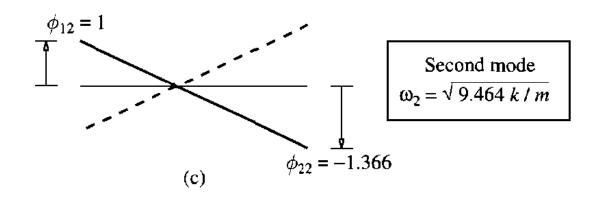
$$k \begin{bmatrix} -2.155 & -1.577 \\ -1.577 & -1.155 \end{bmatrix} \begin{Bmatrix} \phi_{12} \\ \phi_{22} \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$\phi_{12} = 1, \phi_{22} = -1.366.$$

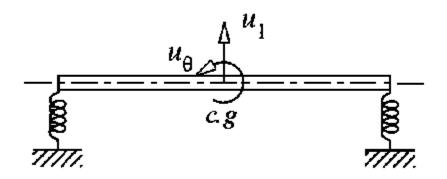
$$\phi_1 = \left\{ \begin{array}{l} 1 \\ 0.366 \end{array} \right\} \qquad \phi_2 = \left\{ \begin{array}{l} 1 \\ -1.366 \end{array} \right\}$$

$$\frac{\phi_{11} = 1}{2}$$

$$\frac{\phi_{21} = 0.366}{4}$$
First mode
$$\omega_1 = \sqrt{2.536 \, k / m}$$



(b)



$$\mathbf{m} = \begin{bmatrix} m & 0 \\ 0 & mL^2/12 \end{bmatrix} \qquad \mathbf{k} = \begin{bmatrix} 3k & kL/2 \\ kL/2 & 3kL^2/4 \end{bmatrix}$$

$$\mathbf{m} = \begin{bmatrix} m & 0 \\ 0 & mL^2/12 \end{bmatrix} \qquad \mathbf{k} = \begin{bmatrix} 3k & kL/2 \\ kL/2 & 3kL^2/4 \end{bmatrix}$$

$$\mathbf{k} - \omega_n^2 \mathbf{m} = \begin{bmatrix} 3k - m\omega_n^2 & kL/2 \\ kL/2 & (9k - m\omega_n^2)L^2/12 \end{bmatrix}$$

$$m^2\omega_n^4 - 12km\omega_n^2 + 24k^2 = 0$$

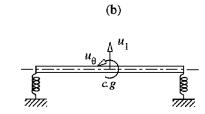
$$\omega_1^2 = 2.536k/m$$
 and $\omega_2^2 = 9.464k/m$

$$\left[\mathbf{k} - \omega_n^2 \mathbf{m}\right] \phi_n = \mathbf{0}$$

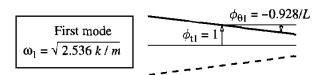
$$(3k - m\omega_n^2) \phi_{tn} + \frac{kL}{2} \phi_{\theta n} = 0$$
 or $\phi_{\theta n} = -\frac{3k - m\omega_n^2}{kL/2} \phi_{tn}$

$$\frac{L}{2}\phi_{\theta 1} = -0.464\phi_{t1} \qquad \frac{L}{2}\phi_{\theta 2} = 6.464\phi_{t2}$$

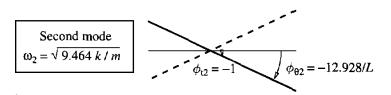
$$\phi_{t1} = 1$$
, then $\phi_{\theta 1} = -0.928/L$,



$$\phi_{t2} = -1$$
, then $\phi_{\theta 2} = -12.928/L$.



$$\phi_1 = \left\{ \begin{array}{c} 1 \\ -0.928/L \end{array} \right\} \qquad \phi_2 = \left\{ \begin{array}{c} -1 \\ -12.928/L \end{array} \right\}$$



Örnek 2: Şekildeki sistemin frekanslarını hesaplayıp mod şekillerini bulunuz. Modların dikliğini gösteriniz.

(a)
$$EI \qquad \downarrow^{u_2} \qquad \downarrow^{u_1}$$

$$mL/2 \qquad mL/4$$

$$L/2 \qquad \downarrow$$

$$\mathbf{m} = \begin{bmatrix} mL/4 & \\ & mL/2 \end{bmatrix} \qquad \mathbf{k} = \frac{48EI}{7L^3} \begin{bmatrix} 2 & -5 \\ -5 & 16 \end{bmatrix}$$

$$\mathbf{k} - \omega^2 \mathbf{m} = \frac{48EI}{7L^3} \begin{bmatrix} 2 - \lambda & -5 \\ -5 & 16 - 2\lambda \end{bmatrix}$$

$$\lambda = \frac{7mL^4}{192EI}\omega^2$$

$$2\lambda^2 - 20\lambda + 7 = 0$$

 $\lambda_1 = 0.36319$ and $\lambda_2 = 9.6368$.

$$\omega_1 = 3.15623 \sqrt{\frac{EI}{mL^4}}$$
 $\omega_2 = 16.2580 \sqrt{\frac{EI}{mL^4}}$

$$\phi_1 = \left\{ \begin{array}{l} 1 \\ 0.3274 \end{array} \right\} \qquad \phi_2 = \left\{ \begin{array}{l} 1 \\ -1.5274 \end{array} \right\}$$

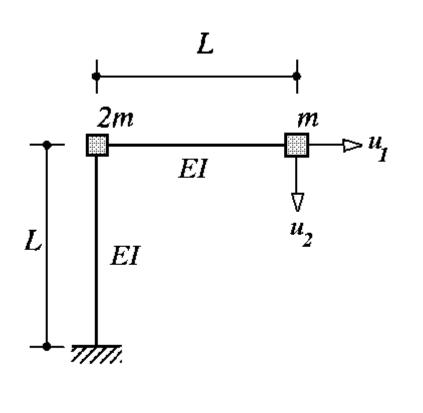
(b)
$$\omega_1 = 3.15623 \sqrt{\frac{EI}{mL^4}}$$

(c)
$$\omega_2 = 16.2580 \sqrt{\frac{EI}{mL^4}}$$

$$\phi_1^T \mathbf{m} \phi_2 = \frac{mL}{4} \langle 1 \quad 0.3274 \rangle \begin{bmatrix} 1 \\ 2 \end{bmatrix} \begin{Bmatrix} 1 \\ -1.5274 \end{Bmatrix} = 0$$

$$\phi_1^T \mathbf{k} \phi_2 = \frac{48EI}{7L^3} \langle 1 \quad 0.3274 \rangle \begin{bmatrix} 2 & -5 \\ -5 & 16 \end{bmatrix} \begin{Bmatrix} 1 \\ -1.5274 \end{Bmatrix} = 0$$

Örnek 3: Şekildeki sistemin frekanlarını hesaplayıp mod şekillerini çiziniz. Mod vektörlerini uç deplasmanın 1 birim olması haline göre gösteriniz.



$$\mathbf{m} = \begin{bmatrix} 3m & \\ & m \end{bmatrix} \qquad \mathbf{k} = \frac{6EI}{7L^3} \begin{bmatrix} 8 & -3 \\ -3 & 2 \end{bmatrix}$$

$$\lambda = \frac{7mL^3}{6EI}\omega^2$$

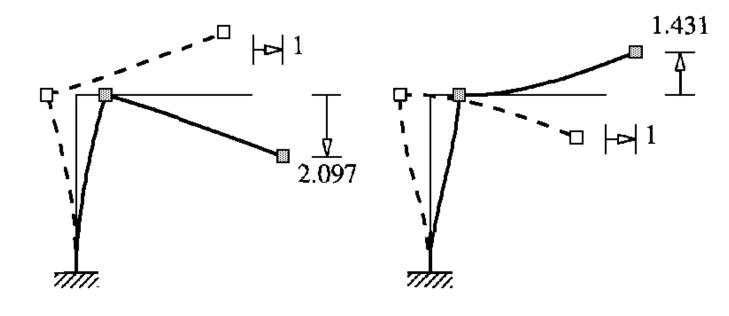
$$3\lambda^2 - 14\lambda + 7 = 0$$

$$\lambda_1 = 0.5695$$
 and $\lambda_2 = 4.0972$.

(a)

$$\omega_1 = 0.6987 \sqrt{\frac{EI}{mL^3}}$$
 $\omega_2 = 1.874 \sqrt{\frac{EI}{mL^3}}$

$$\phi_1 = \left\{ \frac{1}{2.097} \right\} \qquad \phi_2 = \left\{ \frac{1}{-1.431} \right\}$$

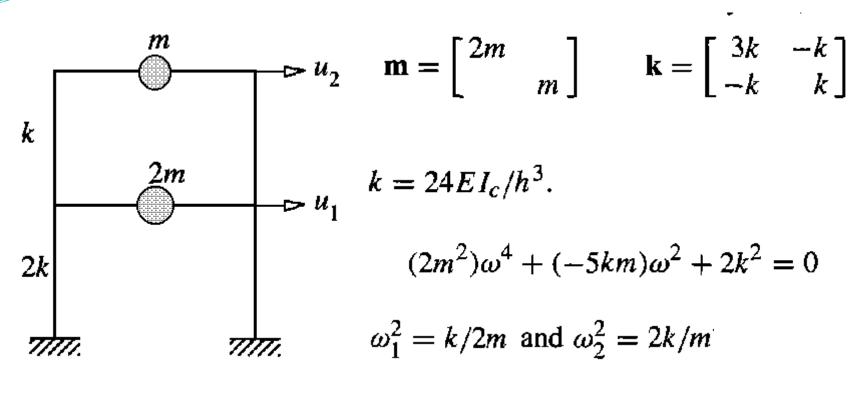


(b)
$$\omega_1 = 0.6987 \sqrt{\frac{EI}{mL^3}}$$
 (c) $\omega_2 = 1.874 \sqrt{\frac{EI}{mL^3}}$

(c)
$$\omega_2 = 1.874 \sqrt{\frac{EI}{mL^3}}$$

$$\phi_1 = \left\{ \begin{array}{l} 0.4769 \\ 1 \end{array} \right\} \qquad \phi_2 = \left\{ \begin{array}{l} -0.6988 \\ 1 \end{array} \right\}$$

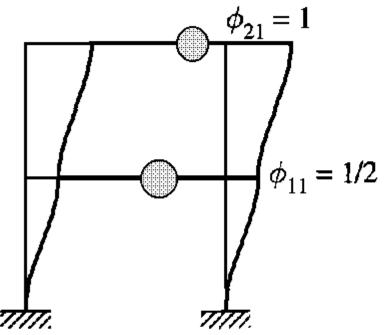
Örnek 4: Şekildeki iki katlı kayma çerçevesinin frekanslarını hesaplayarak mod şekillerini bulunuz.

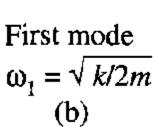


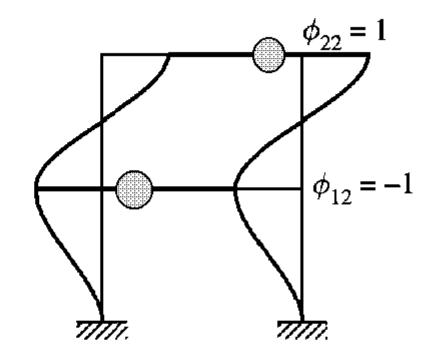
$$\omega_1 = \sqrt{\frac{k}{2m}} \qquad \omega_2 = \sqrt{\frac{2k}{m}}$$

$$\omega_1 = 3.464 \sqrt{\frac{EI_c}{mh^3}}$$
 $\omega_2 = 6.928 \sqrt{\frac{EI_c}{mh^3}}$

$$\phi_1 = \left\{ \begin{array}{c} \frac{1}{2} \\ 1 \end{array} \right\} \qquad \phi_2 = \left\{ \begin{array}{c} -1 \\ 1 \end{array} \right\}$$







Second mode
$$\omega_2 = \sqrt{2k/m}$$
 (c)

Kaynaklar

- 1) ANIL K. CHOPRA, Dynamics of Structures, Theory, and Aplications to Earthquake Engineering, Second Edition, Prentice Hall.
- 2) ANIL K. CHOPRA, Çeviren: HİLMİ LUŞ, Yapı Dinamiği, Teori ve deprem Mühendisliği Uygulamaları, Dördüncü Baskıdan Çeviri, Palme Yayıncılık.