State Variable Analysis - Introduction

The procedure of determining the state of a system is called state variable analysis.

State: The state of a system at any time 'to' is the minimum set of numbers $x_1, x_2, x_3 - x_n$ which along with the shpit to the system for time $t \ge t_0$ is sufficient to determine the behaviour of the system for all $t \ge t_0$.

State Variable & State Vector

State Variables: The state of a system cit any time 'to' is the minimum set of numbers $x_1, x_2, x_3 = x_n$ which along with the liput to the system for time $t \ge t_0$ is sufficient to determine the hehaviour of the system for all $t \ge t_0$. This set of variables is called state variables.

State Vector: - 9% we need n variables to completely describe the behaviour of a given system, then there n' state variables may be considered as n component of a vector x. Such a vector is called state vector.

State Equations

State equations:

9n likear time - invariant systems, the general form of state equations are;

$$\dot{x}(t) = Ax(t) + Built$$

$$y(t) = Cx(t) + Dult$$

where $x = n$ -dimensional state vector
$$y = n$$
-dimensional output vector
$$u = input vector (9n - dimensional)$$

$$A = mxn system matrix$$

$$B = mxr control matrix
$$C = mxn output matrix$$$$

State Space Representation

Three Techniques

- 1. For Electrical Network
- 2. Higher Order Differential Equation
- 3. From Transfer Function

State Space Representation for Electrical Network

$$\frac{di}{dt} = -\frac{R_{iL}}{L} - \frac{V_C}{L} + \frac{V}{L} - (1)$$

$$\frac{dV_C}{dt} = -\frac{L}{L}$$

$$\frac{dV_C}{dt} =$$

State Space Representation for Electrical Network

$$\frac{di_{L_{1}}}{dt} = -\frac{1}{2}i_{L_{1}} + \frac{1}{2}V_{c} - \frac{1}{2}V_{1} - (1)$$

$$\frac{di_{L_{2}}}{dt} = -2i_{L_{2}} - V_{2} + V_{c} - (2)$$

$$\frac{di_{L_{2}}}{dt} = \frac{1}{2}i_{L_{1}} + \frac{1}{2}i_{L_{2}} - (3)$$

$$\begin{bmatrix} \dot{u}_1 \\ \dot{u}_2 \\ \dot{v}_c \end{bmatrix} = \begin{bmatrix} -\frac{1}{2} & 0 & \frac{1}{2} \\ 0 & -2 & 1 \\ \frac{1}{2} & \frac{1}{2} & 0 \end{bmatrix} \begin{bmatrix} \dot{u}_1 \\ \dot{u}_2 \\ \dot{v}_c \end{bmatrix} + \begin{bmatrix} -\frac{1}{2} & 0 \\ 0 & -1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \dot{v}_1 \\ \dot{v}_2 \end{bmatrix}$$

Ques: A system is described by the following differential equation. Represent the system en state space. $\frac{d^3x}{dt^3} + 3\frac{d^2x}{dt^2} + 4\frac{dx}{dt} + 4x = 4(t) + 342 + 443$ and outputs are $y_1 = 4\frac{dx}{dt} + 3u_1$ $y_2 = \frac{d^2x}{dt^2} + 4 - 4 + 4 = 4$

$$\frac{d^3x}{dt^3} + 3\frac{d^2x}{dt^2} + 4\frac{dx}{dt} + 4x = 4(t) + 342 + 443$$

Soln: Select state variables as

Let
$$x_1 = x$$
 $x_1' = x' = x_2$
 $x_2' = x' = x_3$

and $x_3' = x'' = x_1 + 3u_2 + 4u_3 - 4x - 4dx - 3dx$

or $x_3' = u_1 + 3u_2 + 4u_3 - 4x_1 - 4x_2 - 3x_3 - 3$

Thus, $\begin{bmatrix} x_1' \\ x_2' \\ x_3' \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -4 & -4 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 1 & 3 & 4 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}$

Above

Now outputs are:
$$y_{1} = 4\frac{dn}{dt} + 3u_{1} = 4x_{1} + 3u_{1}$$

$$y_{2} = \frac{d^{3}x}{dt^{2}} + 4u_{2} + u_{3} = x_{3} + 4u_{2} + u_{3}$$

$$\begin{bmatrix} y_{1} \\ y_{2} \end{bmatrix} = \begin{bmatrix} 0 & 4 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ u_{3} \end{bmatrix} + \begin{bmatrix} 3 & 0 & 0 \\ 0 & 4 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix}$$

Ques: For the given transfer function, obtain the state model.
$$G(s) = \frac{Y(s)}{U(s)} = \frac{K(C_2 S + G_1)}{S^3 + a_3 S^2 + a_2 S + a_1}$$

$$\frac{Y(s)}{U(s)} = \frac{X(s)}{U(s)} \cdot \frac{Y(s)}{X(s)} = \left[\frac{K}{s^2 + a_3 s^2 + a_2 s + a_1} \right] \cdot \left[C_2 s + c_1 \right]$$

$$\frac{Y(s)}{U(s)} = \frac{X(s)}{U(s)} \cdot \frac{Y(s)}{X(s)} = \left[\frac{K}{s^2 + a_3 s^2 + a_2 s + a_1} \right] \cdot \left[C_2 s + c_1 \right]$$

Consider,
$$\frac{X(s)}{U(s)} = \frac{K}{s^3 + q_3 s^2 + a_2 s + a_3}$$

or $s^3 X(s) + q_3 s^2 X(s) + q_2 s X(s) + q_3 X(s) = K U(s)$
taking shverose Laplau,
 $\frac{d^3 x(t)}{dt^3} + a_3 \frac{d^2 x(t)}{dt^2} + a_2 \frac{d x(t)}{dt} + a_4 X(t) = K U(t)$

Het
$$x_1 = x$$

then $\hat{x}_1 = \frac{dx}{dt} = x_2$
 $\hat{x}_2 = \frac{d^2x}{dt^2} = x_3$
 $\hat{x}_3 = \frac{d^3x}{dt^2} = kulty - a_1x - a_2\frac{dx}{dt} - a_3\frac{dx}{dt^2}$
or $\hat{x}_3 = kulty - a_1x_1 - a_2x_2 - a_3x_3$

$$\begin{pmatrix} \hat{x}_1 \\ \hat{x}_2 \\ \hat{x}_1 \end{pmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -a_1 & -a_2 & -a_3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ K \end{bmatrix} \begin{bmatrix} u \\ A \end{bmatrix}$$
At

Now consider
$$\frac{y(s)}{v(s)} = C_2 \beta + C_1$$

or $y(s) = \delta C_2 v(s) + C_1 v(s)$

Taking inverse Laplace;

 $y(t) = C_1 \times (t) + C_2 \xrightarrow{\alpha} (t)$

or $y = C_1 \times (t) + C_2 \times (t)$

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State Model

Solution of State Equation (Homogeneous Equation)

We know that

$$\dot{x}(t) = Ax(t) + Bu(t)$$

$$u(t) = 0 \text{ for unforced response}$$
then
$$\dot{x}(t) = Ax(t)$$

State Transition Matrix

Computed by Three Techniques

- 1. Laplace Inverse Method
- 2. Series Summation
- 3. Modal Matrix

State Transition Matrix (Laplace Inverse Method)

Ques: Compute the STM when
$$A = \begin{bmatrix} -1 & 1 \\ 0 & -2 \end{bmatrix}$$

$$[SI-A]^{-1} = \frac{1}{S^2 + 3s + 2} \begin{bmatrix} S+2 & 1 \\ 0 & S+1 \end{bmatrix}$$

State Transition Matrix (Laplace Inverse Method)

$$\left[SI-A\right]^{-1} = \frac{1}{S^2+3s+2} \left[\begin{array}{cc} S+2 & 1\\ 0 & S+1 \end{array}\right]$$

$$\beta(t) = L^{-1} \left[SI - A \right]^{-1} = L^{-1} \left[\frac{S+2}{(S+1)(S+2)} \right]$$

$$O = \frac{S+1}{(S+1)(S+2)}$$
or
$$\beta(t) = \left[e^{-t} - e^{-2t} \right]$$

$$O = e^{-2t}$$

$$O = e^{-2t}$$

State Transition Matrix (Series Summation Method)

$$\beta(t) = e^{At} = 1 + At + \frac{A^2t^2}{L^2} + \frac{A^3t^3}{L^3} + \dots$$

Evaluate the STM by series summation method

$$A = \begin{bmatrix} 2 & -2 & 3 \\ 1 & 1 & 1 \\ 1 & 3 & -1 \end{bmatrix}$$

$$A^{2} = \begin{bmatrix} 5 & 3 & 1 \\ 4 & 2 & 3 \end{bmatrix}$$

$$A^{2} = \begin{bmatrix} 5 & 3 & 1 \\ 4 & 2 & 3 \end{bmatrix}$$

$$A^{3} = \begin{bmatrix} 14 & -4 & 17 \\ 13 & 3 & 11 \\ 13 & 11 & 3 \end{bmatrix}$$

$$A^{3} = \begin{bmatrix} 14 & -4 & 17 \\ 13 & 3 & 11 \\ 13 & 11 & 3 \end{bmatrix}$$

State Transition Matrix (Series Summation Method)

$$\beta(t) = e^{At} = 1 + At + \frac{A^2t^2}{(2)^2} + \frac{A^3t^3}{(3)^3} + \dots$$

$$= \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} + \begin{bmatrix} 2 & -2 & 3 \\ 1 & 1 & 1 \\ 1 & 3 & -1 \end{bmatrix} + \begin{bmatrix} 5 & 3 & 1 \\ 4 & 2 & 3 \\ 4 & -2 & 7 \end{bmatrix} \frac{t^2}{2}$$

$$+ \begin{bmatrix} 14 & -4 & 17 \\ 13 & 3 & 11 \\ 13 & 11 & 3 \end{bmatrix} \frac{t^3}{6}$$

$$= \begin{bmatrix} 1 + 2t + \frac{5t^2}{2} + \frac{14t^3}{6} - -2t + \frac{3t^2}{2} - \frac{4t^2}{6} + 3t + \frac{t^2}{2} + \frac{17t^3}{6} - 1 + \frac{3t^2}{2} + \frac{17t^3}{6} + -1 + \frac{3t^2}{2} + \frac{17$$

State Transition Matrix (Modal Matrix Method)

where
$$M = Model matrix$$
 $\Lambda = diagonal matrix with eigen values on its mathing diagonal matrix.$
 $M' = ehverse of model matrix.$

Eigen Value

Ques:- Compute som when
$$A = \begin{bmatrix} 0 & 1 \\ -3 & -4 \end{bmatrix}$$

For eigen values;
$$|AI-A| = 0$$

 $A^{2} + 4A + 3 = 0$
or $A_{1} = -1$, $A_{2} = -3$

Eigen Vector

For eigen values;
$$[AI-A] = 0$$

 $A^2 + 4A + 3 = 0$
or $A_1 = -1$, $A_2 = -3$
Hence eigen vactor;
 $P_1 = \begin{bmatrix} 1 \\ A_1 \end{bmatrix} = \begin{bmatrix} -1 \\ -1 \end{bmatrix}$
 $P_2 = \begin{bmatrix} 1 \\ A_2 \end{bmatrix} = \begin{bmatrix} -3 \\ -3 \end{bmatrix}$

Modal Matrix

Modal Matrix
$$[M] = \begin{bmatrix} P_1 & P_2 \end{bmatrix} = \begin{bmatrix} -1 & -3 \end{bmatrix}$$

and $[M^{-1}] = -1 \begin{bmatrix} -3 & -1 \\ 1 & 1 \end{bmatrix}$
or $M^{-1} = \begin{bmatrix} 1.5 & 0.5 \\ -0.5 & 0.5 \end{bmatrix}$

Diagonal Matrix

$$e^{\Lambda t} = \begin{bmatrix} e^{\Lambda_1 t} & 0 \\ 0 & e^{\Lambda_2 t} \end{bmatrix} = \begin{bmatrix} e^{-t} & 0 \\ 0 & e^{-3t} \end{bmatrix}$$

STM using Modal Matrix

$$e^{At} = M^{*}e^{At} M^{-1}$$

$$\begin{bmatrix} 1 & 1 \\ -1 & -3 \end{bmatrix} \begin{bmatrix} e^{-t} & 0 \\ e^{-3t} \end{bmatrix} \begin{bmatrix} 1.5 & 0.5 \\ -0.5 & 0.5 \end{bmatrix}$$

$$\begin{bmatrix} 1.5e^{-t} - 0.5e^{-3t} & 0.5e^{-t} - 0.5e^{-3t} \\ 1.5e^{3t} - 1.5e^{-t} & 1.5e^{-3t} - 0.5e^{-t} \end{bmatrix}$$

Solution of State Equation (Non-Homogeneous Equation)

$$x(t) = \phi(t) x(0) + \int_{0}^{t} \phi(t-\tau) Bu(\tau) d\tau$$

Find the time response of the system described by the equation

$$\dot{x}(t) = \begin{bmatrix} -1 & 1 \\ 0 & -2 \end{bmatrix} x(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t)$$

$$x(0) = \begin{bmatrix} -1 \\ 0 \end{bmatrix}, u(t) = 1, t > 0$$

Solution of State Equation (Non-Homogeneous Equation)

Calculation of
$$\phi(t)$$
:

tion of
$$\phi(t)$$
:

$$\phi(t) = \mathcal{L}^{-1}\phi(s) = \mathcal{L}^{-1}(sI - A)^{-1}$$

$$\phi(s) = [sI - A]^{-1}$$

$$sI - A = \begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{bmatrix} -1 & 1 \\ 0 & -2 \end{bmatrix} = \begin{bmatrix} s+1 & -1 \\ 0 & s+2 \end{bmatrix}$$

$$[sI - A]^{-1} = \begin{bmatrix} \frac{s+2}{s^2 + 3s + 2} & \frac{1}{s^2 + 3s + 2} \\ 0 & \frac{s+1}{s^2 + 3s + 2} \end{bmatrix}$$

$$\phi(t) = \mathcal{L}^{-1}\phi(s) = \begin{bmatrix} \mathcal{L}^{-1} \frac{s+2}{s^2 + 3s + 2} & \mathcal{L}^{-1} \frac{1}{s^2 + 3s + 2} \\ 0 & \mathcal{L}^{-1} \frac{s+1}{s^2 + 3s + 2} \end{bmatrix}$$

$$\phi(t) = \begin{bmatrix} e^{-t} & e^{-t} - e^{-2t} \\ 0 & e^{-2t} \end{bmatrix}$$

Time Response of State Space System

$$x(t) = \phi(t) \ x(0) + \int_{0}^{t} \phi(t-\tau) \ Bu(\tau) \ d\tau$$

$$x(t) = \begin{bmatrix} e^{-t} & e^{-t} - e^{-2t} \\ 0 & e^{-2t} \end{bmatrix} \begin{bmatrix} -1 \\ 0 \end{bmatrix} + \int_{0}^{t} \begin{bmatrix} e^{-(t-\tau)} & e^{-(t-\tau)} - e^{-2(t-\tau)} \\ 0 & e^{-2(t-\tau)} \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} 1 \ d\tau$$

$$= \begin{bmatrix} -e^{-t} \\ 0 \end{bmatrix} + \int_{0}^{t} \begin{bmatrix} e^{-(t-\tau)} - e^{-2(t-\tau)} \\ e^{-2(t-\tau)} \end{bmatrix} d\tau$$

$$x_{1}(t) = -e^{-t} + \int_{0}^{t} e^{-(t-\tau)} - e^{-2(t-\tau)} \ d\tau$$

$$x_{2}(t) = \int_{0}^{t} e^{-2(t-\tau)} \ d\tau$$

$$x_{2}(t) = \frac{1}{2} - \frac{1}{2} e^{-2t}$$

Transfer Function of State Space System

The system equations are given by

$$\dot{x}(t) = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} x(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t)$$
$$y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} x(t)$$

Find the transfer function of the system.

Transfer Function of State Space System

$$A = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}, C = \begin{bmatrix} 1 & 0 \end{bmatrix}$$

$$I = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

$$sI - A = \begin{bmatrix} s & -1 \\ 2 & s+3 \end{bmatrix} \quad \therefore [sI - A]^{-1} = \begin{bmatrix} \frac{s+3}{s^2 + 3s + 2} & \frac{1}{s^2 + 3s + 2} \\ \frac{2}{s^2 + 3s + 2} & \frac{s}{s^2 + 3s + 2} \end{bmatrix}$$

$$C[sI - A]^{-1}.b = [1 \ 0]\begin{bmatrix} \frac{s+3}{s^2 + 3s + 2} & \frac{1}{s^2 + 3s + 2} \\ \frac{-2}{s^2 + 3s + 2} & \frac{s}{s^2 + 3s + 2} \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = \frac{1}{s^2 + 3s + 2}$$

Kalman's Test

A linear time invariant, described by the state equation $\dot{x} = Ax + Bu$ y = Cxis completely controllable if and only if the rank of the controllability matrix is defined as: Re=[B; AB; AB; AB; AB] is equal to size of the matrix [A].

Ques: Consider the following system
$$\begin{bmatrix} \dot{x_i} \\ \dot{z_i} \end{bmatrix} = \begin{bmatrix} -0.5 & 0 \\ 0 & -2 \end{bmatrix} \begin{bmatrix} x_4 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t) \\
y(t) = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_4 \\ x_2 \end{bmatrix}$$
Test for controllability and observability.

Soln: Here
$$A = \begin{bmatrix} -0.5 & 0 \\ 0 & -2 \end{bmatrix}$$
, $B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$

Size of matrix $A = 2 \times 2$

Thus, the size of Q_c and Q_c is 2×2 .

Thus, $Q_c = \begin{bmatrix} B & A B \end{bmatrix}$
 $Q_c = \begin{bmatrix} 0 & 0 \\ 1 & -2 \end{bmatrix}$

Rank of $Q_c = no. of non zero rows = 1$

Size of matrix $A = 2$

Thus, $A = 2$

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Thus, the size of

A system characterised by the transfer function

$$\frac{Y(s)}{u(s)} = \frac{2}{s^3 + 6s^2 + 11s + 6}$$

Find the state and output equation in matrix form and also test the controllability and observability of the system.

A system characterised by the transfer function

$$\frac{Y(s)}{u(s)} = \frac{2}{s^3 + 6s^2 + 11s + 6}$$

Find the state and output equation in matrix form and also test the controllability and observability of the system.

Solution:
$$(s^3 + 6s^2 + 11s + 6) \ Y(s) = 2u(s)$$
.

Taking inverse laplace
 $\dot{Y}(t) + 6\dot{Y}(t) + 11\dot{Y}(t) + 6Y(t) = 2u(t)$.

let

 $Y(t) = x_1$
 $\dot{Y}(t) = \dot{x}_1 = x_2$
 $\dot{Y}(t) = \dot{x}_2 = x_3$
 $\dot{Y}(t) = \dot{x}_3 = x_4$
 $\dot{Y}(t) = x_4 = \dot{x}_3 = 2u(t) - 6x_3 - 11x_2 - 6x_1$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -11 & -6 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 2 \end{bmatrix} u(t)$$

$$y(t) = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} x(t)$$

$$Q = [b:Ab:A^2b]$$

$$A = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -11 & -6 \end{bmatrix}, b = \begin{bmatrix} 0 \\ 0 \\ 2 \end{bmatrix}$$

$$Ab = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -11 & -6 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 2 \end{bmatrix} = \begin{bmatrix} 0 \\ 2 \\ -12 \end{bmatrix}$$

$$A^2b = \begin{bmatrix} 2 \\ -12 \\ 50 \end{bmatrix}$$

$$Q = \begin{bmatrix} 0 & 0 & 2 \\ 0 & 2 & -12 \\ 2 & -12 & 50 \end{bmatrix}$$

$$Q \neq 0$$

system is controllable

Kalman's Test

A linear time invariant, described by the state equation $\dot{x} = Ax + Bu$ y = Cx=> Also, the given system is observable if and only if the rank of the observability matrix is defined as: Qo = [CT; ATCT; (AT)2. CT; --] is equal to size of the metrix [A].

Ques: Consider the following system
$$\begin{bmatrix}
\dot{x_i} \\
\dot{x_i}
\end{bmatrix} = \begin{bmatrix}
-0.5 & 0 \\
0 & -2
\end{bmatrix} \begin{bmatrix} x_4 \\
x_2
\end{bmatrix} + \begin{bmatrix} 0 \\
1
\end{bmatrix} u(t)$$

$$y(t) = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_4 \\
x_2
\end{bmatrix}$$
Test for controllability and observability.

Here
$$A = \begin{bmatrix} -0.5 & 0 \\ 0 & -2 \end{bmatrix}$$
, $B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$

Now $C = \begin{bmatrix} 0 & 1 \\ 1 \end{bmatrix}$, $C^T = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$
 $\therefore Q_0 = \begin{bmatrix} C^T \\ i \end{bmatrix} \in A^T \cdot C^T \end{bmatrix} = \begin{bmatrix} 0 & 0 \\ 1 & -2 \end{bmatrix}$
 $\therefore \text{ rank of } Q_0 \neq \text{ size of matrix } A$

Hence system is unobservable.

Also sikce | Po/ = 0, hence system is unobservable.

A system characterised by the transfer function

$$\frac{Y(s)}{u(s)} = \frac{2}{s^3 + 6s^2 + 11s + 6}$$

Find the state and output equation in matrix form and also test the controllability and observability of the system.

A system characterised by the transfer function

$$\frac{Y(s)}{u(s)} = \frac{2}{s^3 + 6s^2 + 11s + 6}$$

Find the state and output equation in matrix form and also test the controllability and observability of the system.

Solution:
$$(s^3 + 6s^2 + 11s + 6) \ Y(s) = 2u(s)$$
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Taking inverse laplace
 $\dot{Y}(t) + 6\dot{Y}(t) + 11\dot{Y}(t) + 6Y(t) = 2u(t)$.

let

 $Y(t) = x_1$
 $\dot{Y}(t) = \dot{x}_1 = x_2$
 $\dot{Y}(t) = \dot{x}_2 = x_3$
 $\dot{Y}(t) = \dot{x}_3 = x_4$
 $\dot{Y}(t) = x_4 = \dot{x}_3 = 2u(t) - 6x_3 - 11x_2 - 6x_1$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -11 & -6 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} u(t)$$

$$y(t) = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} x(t)$$

$$Q' = \begin{bmatrix} C^T : A^T C^T : A^{T^2} C^T \end{bmatrix}$$

$$C^T = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}, A^T = \begin{bmatrix} 0 & 0 & -6 \\ 1 & 0 & -11 \\ 0 & 1 & -6 \end{bmatrix}$$

$$A^{T}C^{T} = \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix}, \quad A^{T^{2}}C^{T} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$
$$Q' = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Rank of the matrix = 3, Hence the system observable.