

BIO MECHANICS - I:

① Deformable bodies  $\rightarrow$  Stress  $\rightarrow$  Strain  $\rightarrow$  Elastic deformations

① STRESS  $\rightarrow$  NORMAL  
                          SHEAR

② LINEAR RELATIONSHIPS  $\rightarrow$    
 ↓  
 → YOUNG'S MODULUS  
 → POISSON'S RATIO  
 → SHEAR MODULUS  
 → HOOKE'S LAW

③ STRESS VS STRAIN

④ EQUALITY OF CROSS-SHEARS

⑤ SIGN CONVENTION

⑥ IN-CLASS PROBLEMS

⑦ 2D STRESS FIELDS

Will be discussed  
on 02/04/2026

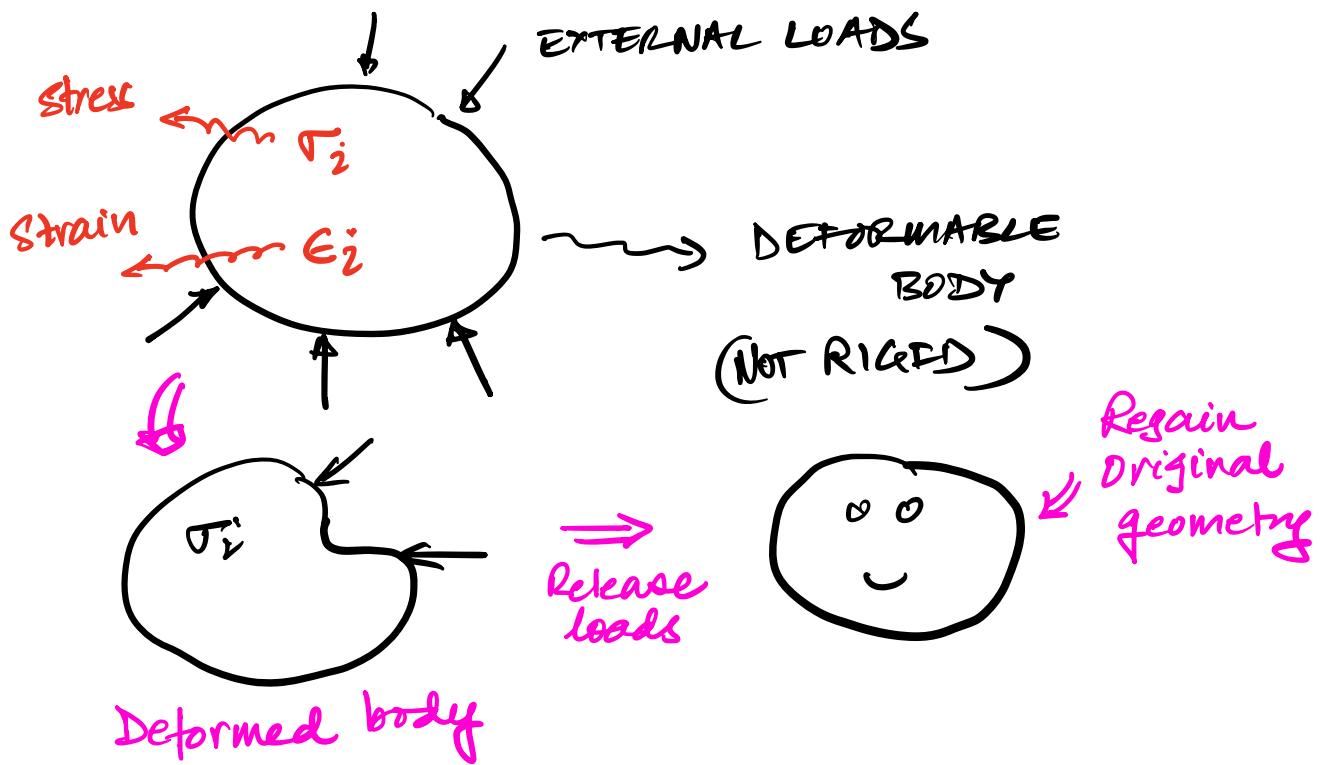
⑧ PRINCIPAL STRESS

$$\sigma_{\text{in}, \text{max}, \text{min}} = \left( \frac{\sigma_{xx} + \sigma_{yy}}{2} \right) \pm \sqrt{\left( \frac{\sigma_{xy} - \sigma_{yy}}{2} \right)^2 + \tau_{xy}^2}$$

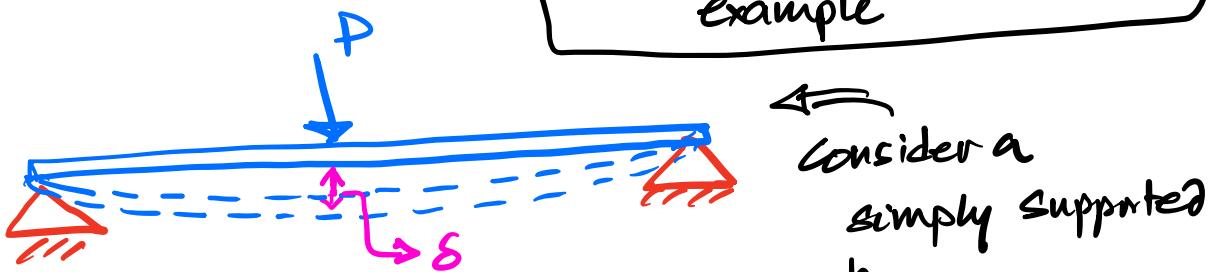
$$\tau_{\text{max}} = \sqrt{\left( \frac{\sigma_{xy} - \sigma_{yy}}{2} \right)^2 + \tau_{xy}^2}$$

## KEY IDEA - 1

Deformable bodies  $\rightarrow$  Stress ( $\sigma$ )  
 $\rightarrow$  Strain ( $\epsilon$ )

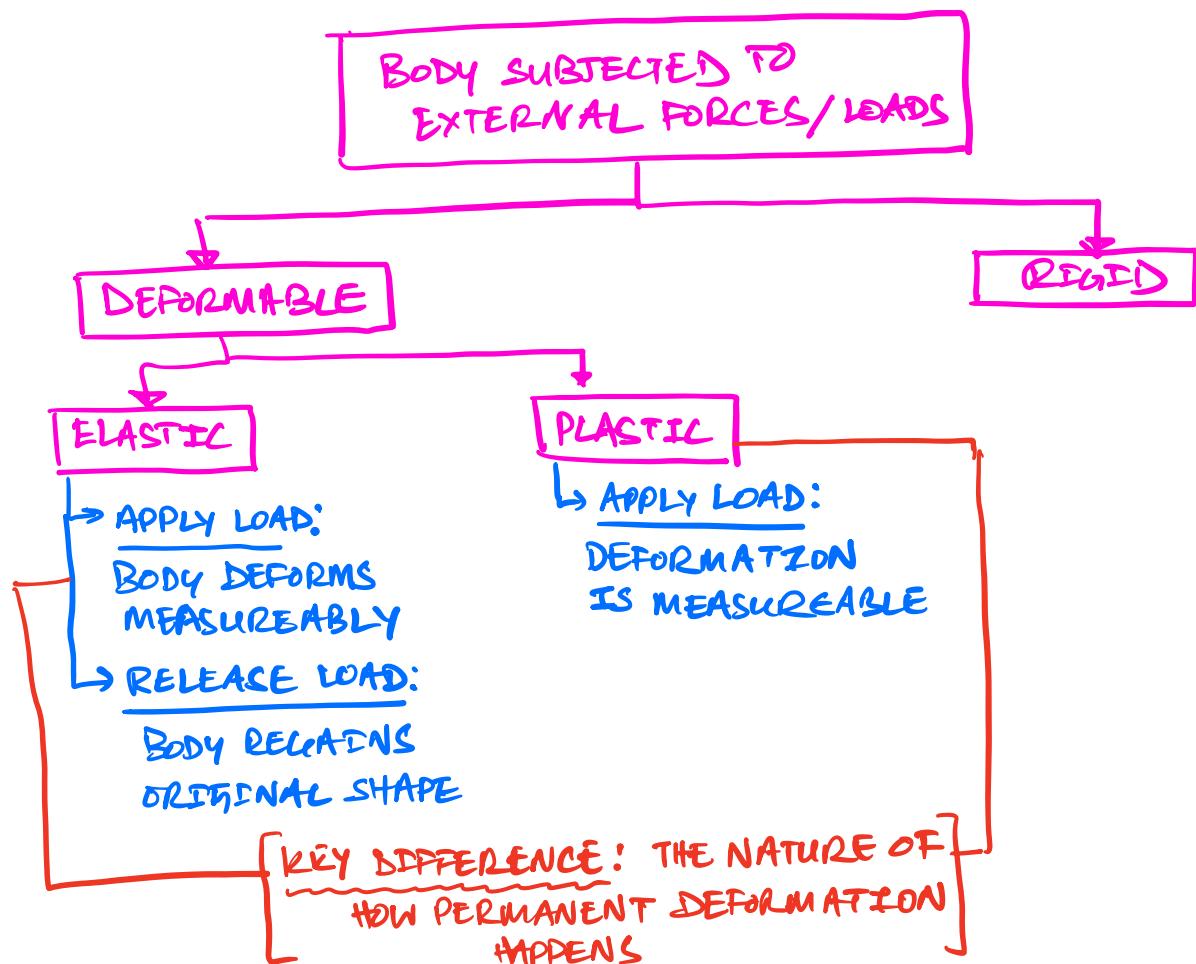


### Civil/Structural Engineering example



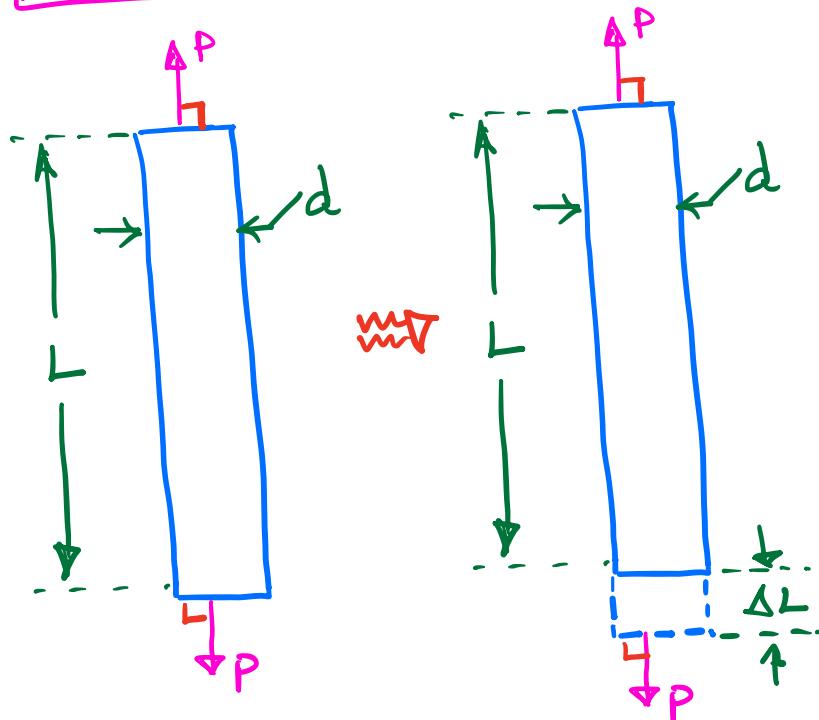
- $\rightarrow$  Apply a load ( $P$ )
- $\rightarrow$  Beam goes through some measurable deformation
- $\rightarrow$  Internal force intensities responding to  $P$   $\rightarrow$  STRESS ( $\sigma$ )
- $\rightarrow$  External changes in geometry/length-scales ( $\epsilon$ )

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We will focus our attention on deformable bodies, only!

**KEY IDEA-2** Elastic Deformations:



Consider a prismatic bar:

- ↳ Cross-sectional area of bar is  $A$
- ↳ External force applied ( $P$ )
- ↳ Stress ( $\sigma$ ) develops inside the body due to the action of external forces.
- ↳ Strain ( $\epsilon$ ) is the measure of deformation

④

$$\sigma = \frac{F}{A}$$

External force axially along the bar  
Area of cross-section

**NORMAL STRESS**

Stress develops internally over the entire cross-section, uniformly, and along the same directional sense as the external force

Measureable change in length

$$E = \frac{\Delta L}{L}$$

Initial length      Original length      Gauge length

**NORMAL STRAIN**

Simply, the ratio of change in length to original length.

Alternately, TRUE STRAIN is associated with actual or deformed length

### KEY IDEA - 3

Relationship between  $\sigma$  &  $E$  under elastic deformations: (Hooke's Law & Young's Modulus)

$$\sigma \propto E$$

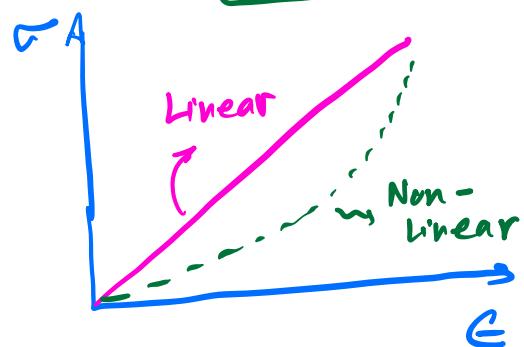
Stress, strain proportionality ( $\sigma$ ) ( $E$ )

How can we define this proportionality?

$$\sigma = K \cdot E$$

Constant  $\rightarrow$  Linear proportionality  
Function (quadratic, cubic, etc.)  $\rightarrow$  Non-Linear proportionality

proportionality constant



Let's focus only on the linear relationship between  $\sigma$  &  $E$ .

**HOOKES LAW**  $\Rightarrow \sigma = E \epsilon$  **Young's Modulus**

Linear proportionality constant

## Assumptions in Linear Relationship

between  $\sigma$  &  $\epsilon$ :

↳ Material is homogenous

composition of material same throughout its geometry

↳ Material is elastic

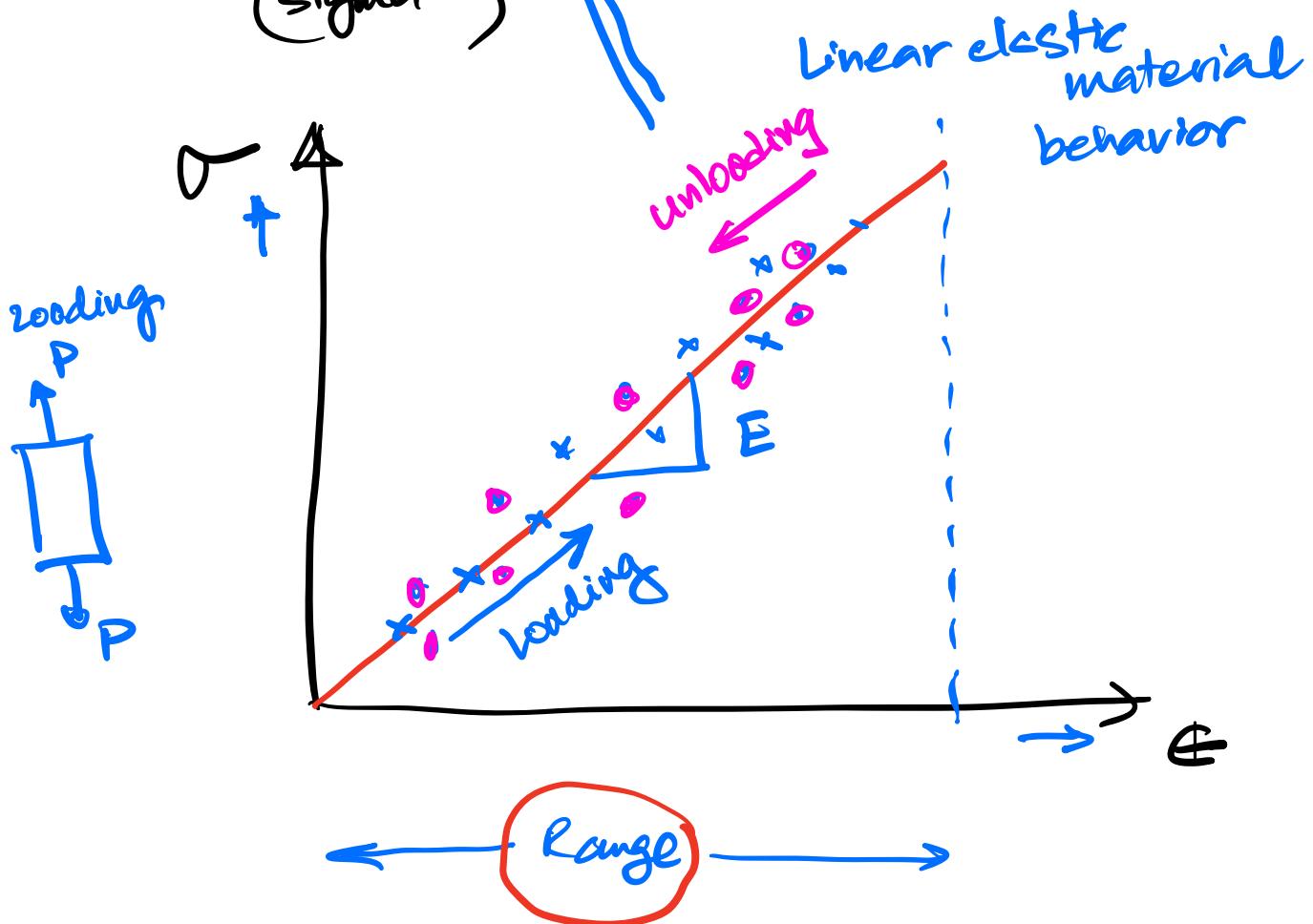
Measurable deformation and ability to regain its original shape

↳ Material is isotropic

Material property such as Young's Modulus don't change when direction of stresses change

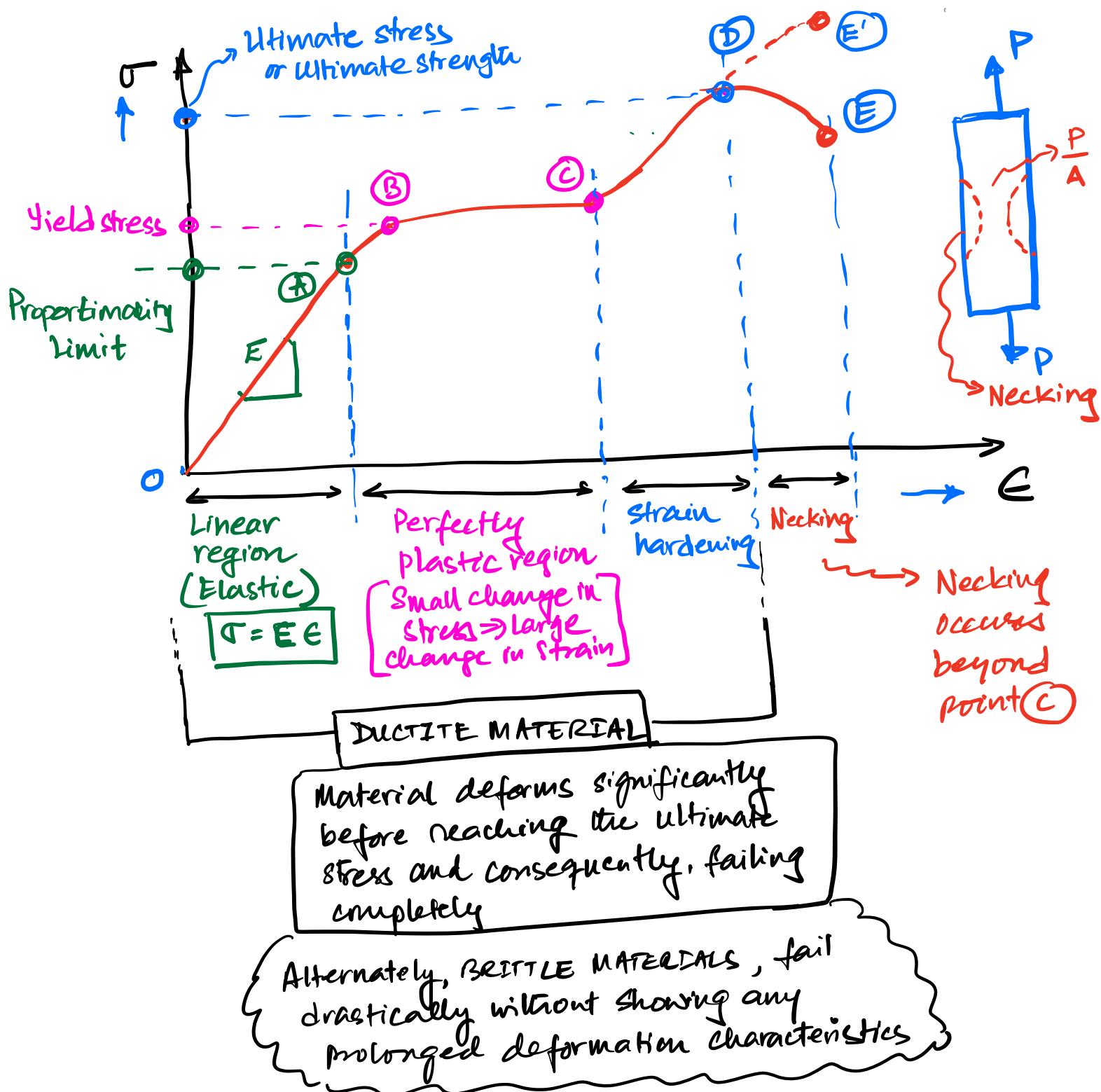
$$\sigma = E \epsilon$$

(Sigma)  $\uparrow$  (epsilon)

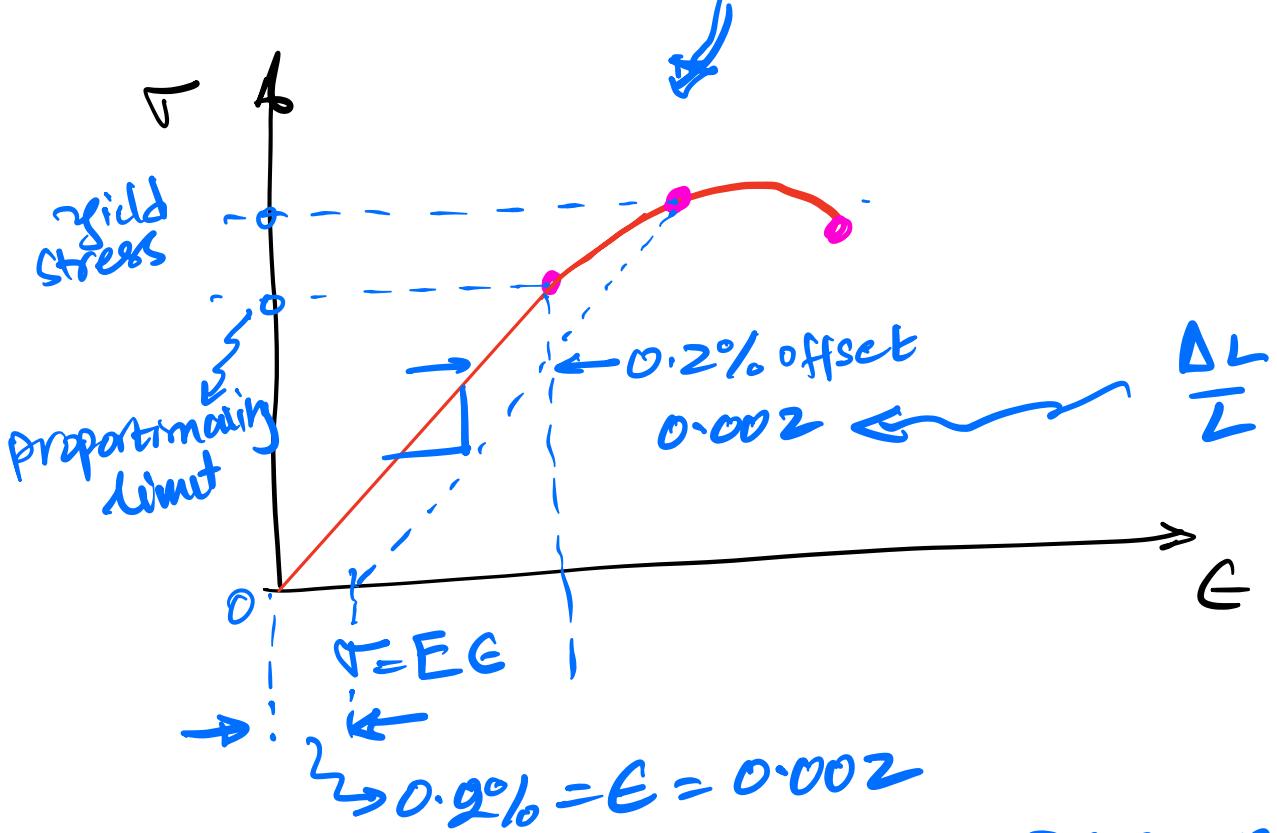
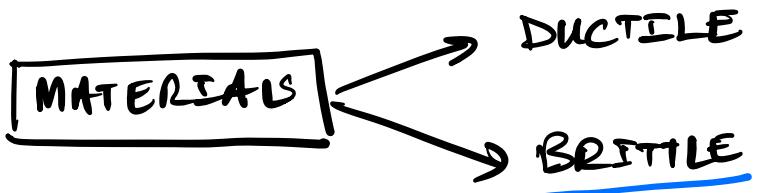


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## STRESS VS STRAIN : (EXAMPLE: STEEL)

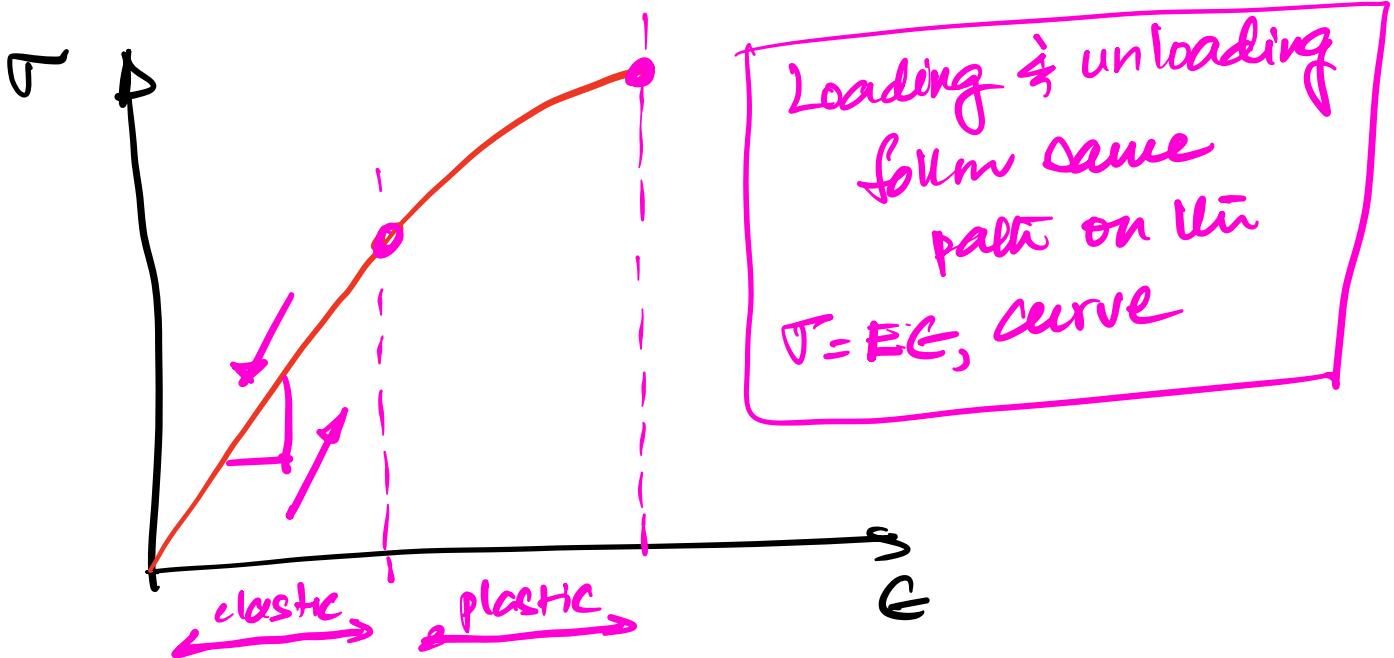


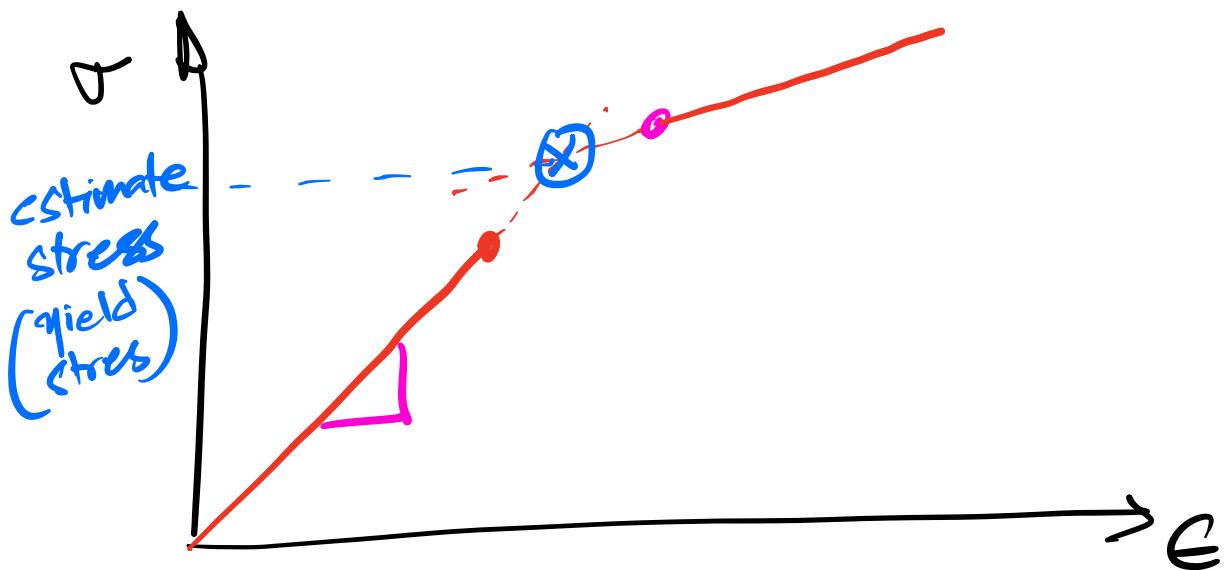
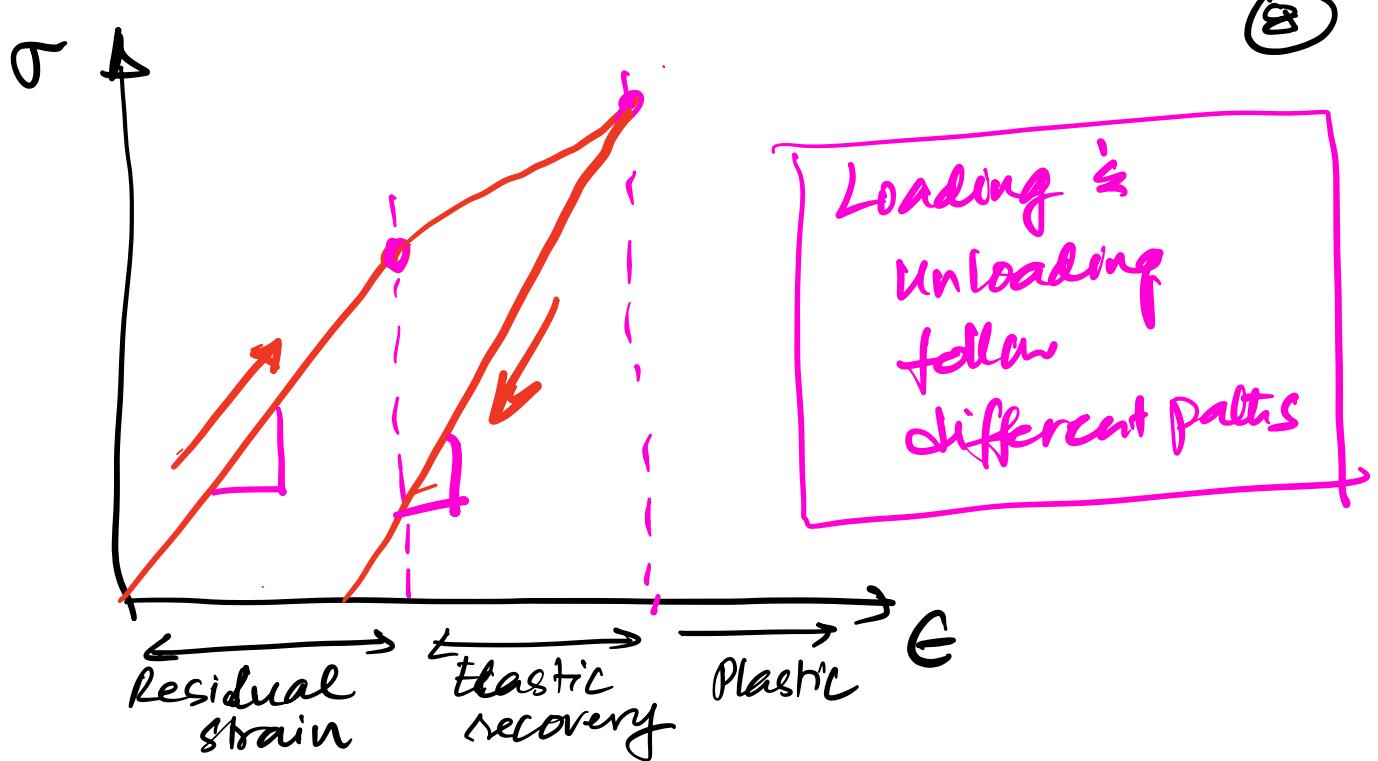
Generally, universal testing machines or uniaxial testing machines (UTMs) are used to provide forces to generate this type of data & curves



# Draw a line parallel to  $\sigma = E\epsilon$  curve

### Other case scenarios:





KEY TAKEAWAY :

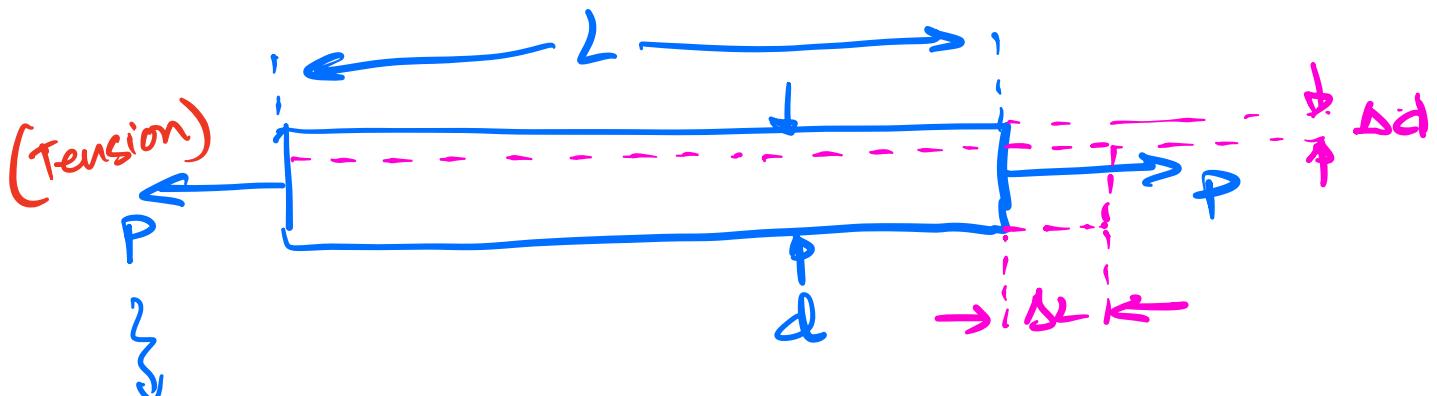
Depending on stress-strain characteristics  
Material can be ductile or brittle.

## KEY IDEA - 4

## Lateral Strain & Poisson's Ratio:

(8)

Consider (again) a prismatic bar:



→ acts uniformly over  
the entire section

→ homogenous

→ elastic

Assumptions

Recall:

$$\text{Axial strain: } \epsilon_A = \frac{\Delta L}{L}$$

$$\text{Lateral strain: } \epsilon_L = \frac{\Delta d}{d}$$

Defn:

$$\text{Poisson's ratio: } \gamma = \frac{\text{Lateral strain}}{\text{Axial strain}}$$

(+) We can  
quantify  
strain in  
the orthogonal  
direction

w.r.t  
applied  
force

[So far we have two material properties viz.,]

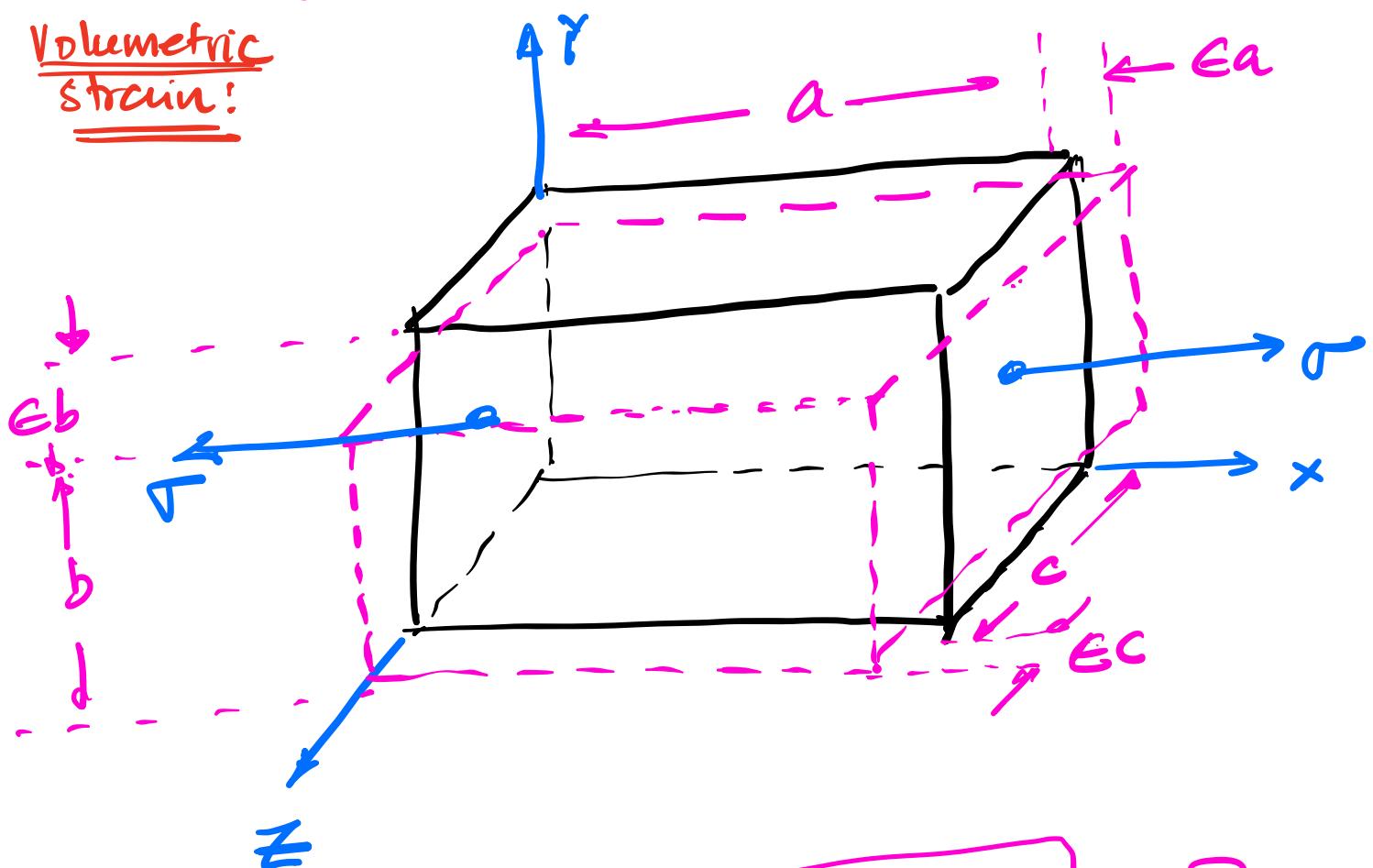
1. Young's Modulus ( $E$ )
2. Poisson's Ratio ( $\gamma$ )

## KEY IDEA - 5

Volume change under uniaxial tension: (5)

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Volumetric strain:



Initial Volume:

$$abc = V_{in} \rightarrow ①$$

Final Volume:  $V_{fin} =$

$$a(1+\epsilon) \cdot b(1-\gamma\epsilon) \cdot c(1-\gamma\epsilon)$$

$$= abc (1+\epsilon)(1-\gamma\epsilon)^2$$

$$= abc (1+\epsilon)(1 - 2\gamma\epsilon + \gamma^2\epsilon^2)$$

Assume:  $\epsilon \ll 1 \therefore \epsilon^2 \rightarrow \text{negligible}$

$$V_{fin} = abc(1 + \epsilon - 2\gamma\epsilon)$$

(11)

$\hookrightarrow z$

$$\Delta V = V_{fin} - V_{in}$$

$$= \{ (abc)(1 + \epsilon - 2\gamma\epsilon) \} - (abc)$$

$$\boxed{\Delta V = abc \epsilon (1 - 2\gamma)} \quad \xrightarrow{\text{Poisson's ratio}} \quad \xrightarrow{\text{3}}$$

}

Axial strain

Volumetric Strain:

$$e = \frac{\Delta V}{V_{in}}$$

$$e = \frac{abc \epsilon (1 - 2\gamma)}{abc}$$

$$e = \epsilon (1 - 2\gamma)$$

$$\sigma = E \epsilon$$

Stress

$$e = \frac{\sigma}{E} (1 - 2\gamma)$$

Volumetric Strain

Poisson's ratio

Young's Modulus

## Summarizing so far:

① Stress  $\rightarrow$  Normal stress ( $\sigma$ )

② Strain  $\rightarrow$  Axial strain ( $\epsilon_A$ )

$\rightarrow$  Lateral strain ( $\epsilon_L$ )

$\rightarrow$  Volumetric strain ( $\epsilon$ )

③ Hooke's law:  $\sigma = E \epsilon$   $\rightarrow$  Axial, lateral or volumetric strain

$\rightarrow$  Young's modulus

④ Material Properties:

$E \rightarrow$  Young's Modulus

$$E = \frac{\sigma}{\epsilon}$$

$\nu \rightarrow$  Poisson's Ratio

$$\nu = \epsilon_L / \epsilon_A$$

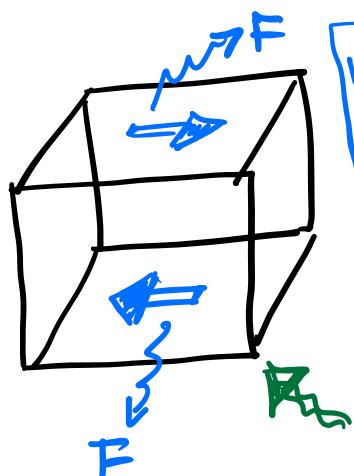
### KEY IDEA-6

### Shear Stress:

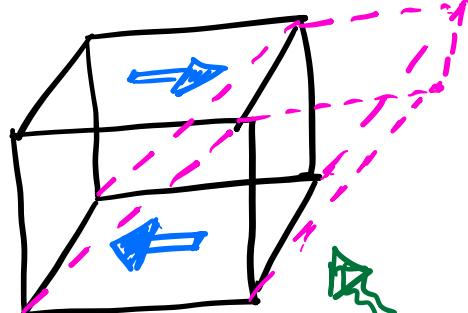
$\rightarrow$  So far we considered normal stress and strains(s) under uniaxial tension.

$\rightarrow$  Now we consider the stress developed along two planes parallel to each other under uniaxial tension.

### SHEAR STRESS



In consider the volume shown  
In Two forces are acting on the opposite surfaces  
In Along the surface  
In Over its entire surface area

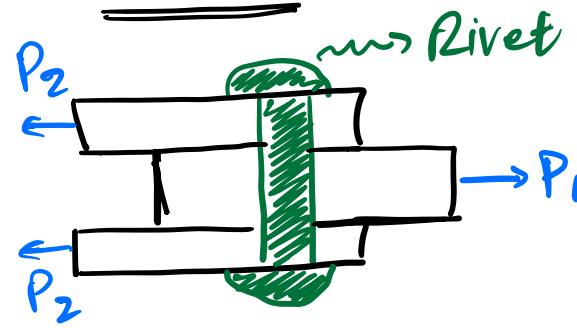


See how it deforms as shown with dotted lines

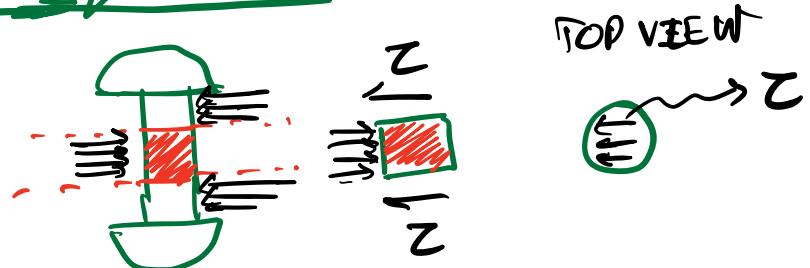
$$\text{Long, shear} = \frac{F}{A}$$

EXAMPLE: Three plates connected by a rivet

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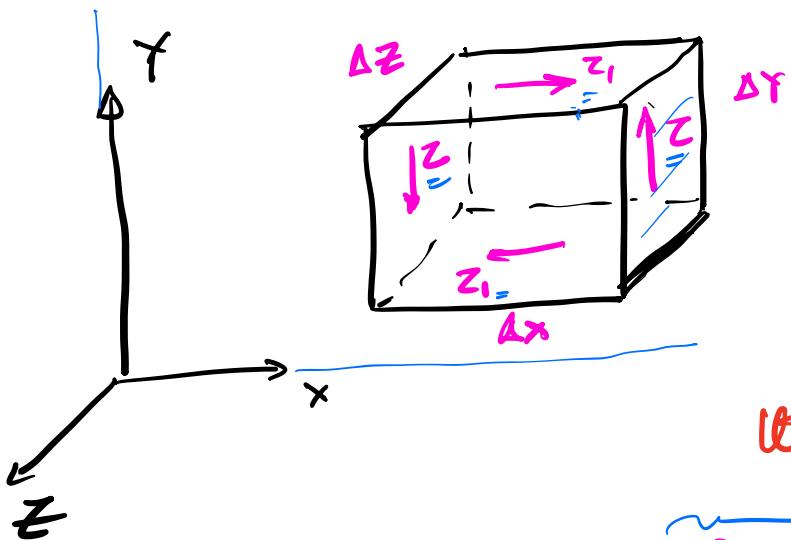


FBD of a section:



**KEY IDEA - F**

Concept of the equality of cross-shears



Shear stress ( $\tau$ )  
separated by  $\Delta x$   
forms a couple!

We therefore calculate  
the moment:

$$M = \underbrace{(\tau)}_{\text{Force}} (\underbrace{\Delta y}_{\text{Surface area}}) (\underbrace{\Delta z}_{\text{distance}})$$

$$M = (\tau) (\Delta y) (\Delta z) (\Delta x)$$

$$M = \tau (\Delta x) (\Delta y) (\Delta z)$$

$$\text{Hence } M_1 = \tau_1 (\Delta x) (\Delta y) (\Delta z)$$

Moment balance  $\sum M = 0 \Rightarrow M_1 = M$

$$\tau = \tau_1$$

x plane

y plane

Equality of  
cross-shear

No Rotation!

## KEY IDEA-8

Sign convention & Stress notation:

T4

( $\sigma$ )

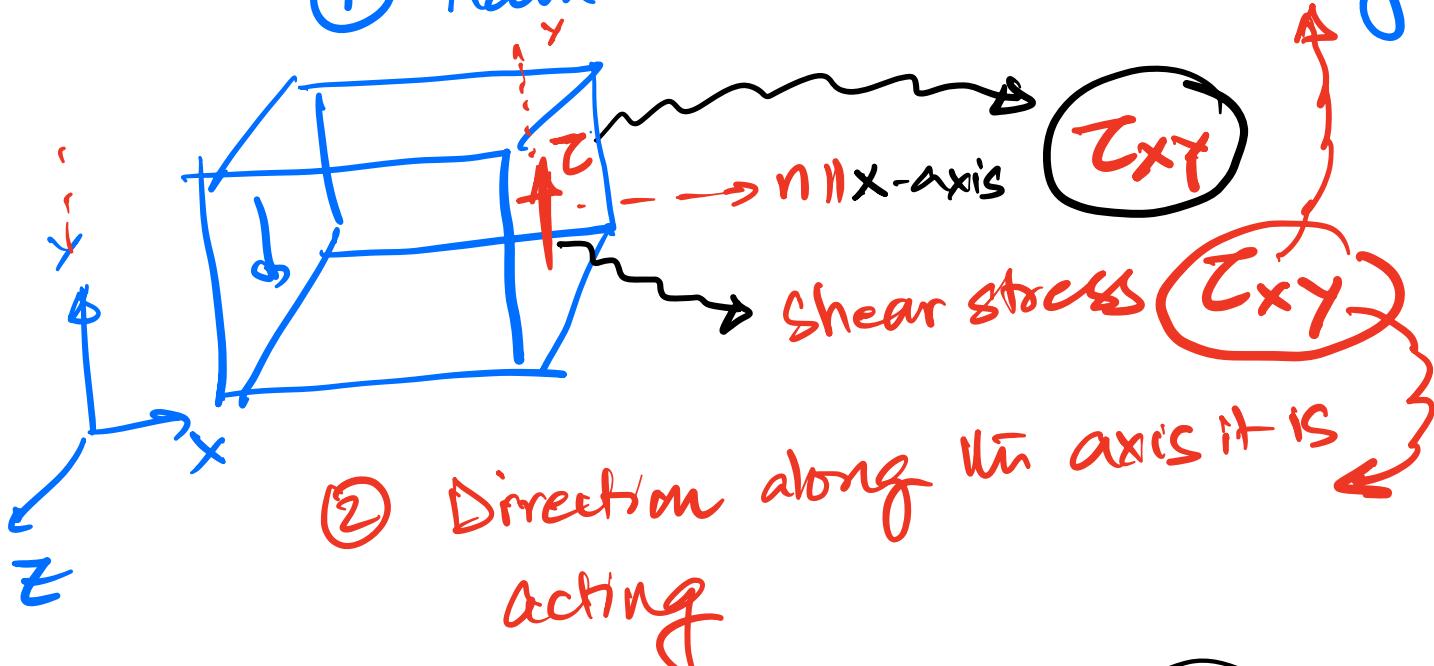
Axial stress  
(Sign convention)

Tension  $\rightarrow$  +ve

Compression  $\rightarrow$  -ve

Shear stress notation ( $\tau$ )

① Plain on which it is acting



② Direction along the axis it is acting

Axial or Normal stress notation

$\sigma_{xx}$

$\sigma_{yy}$

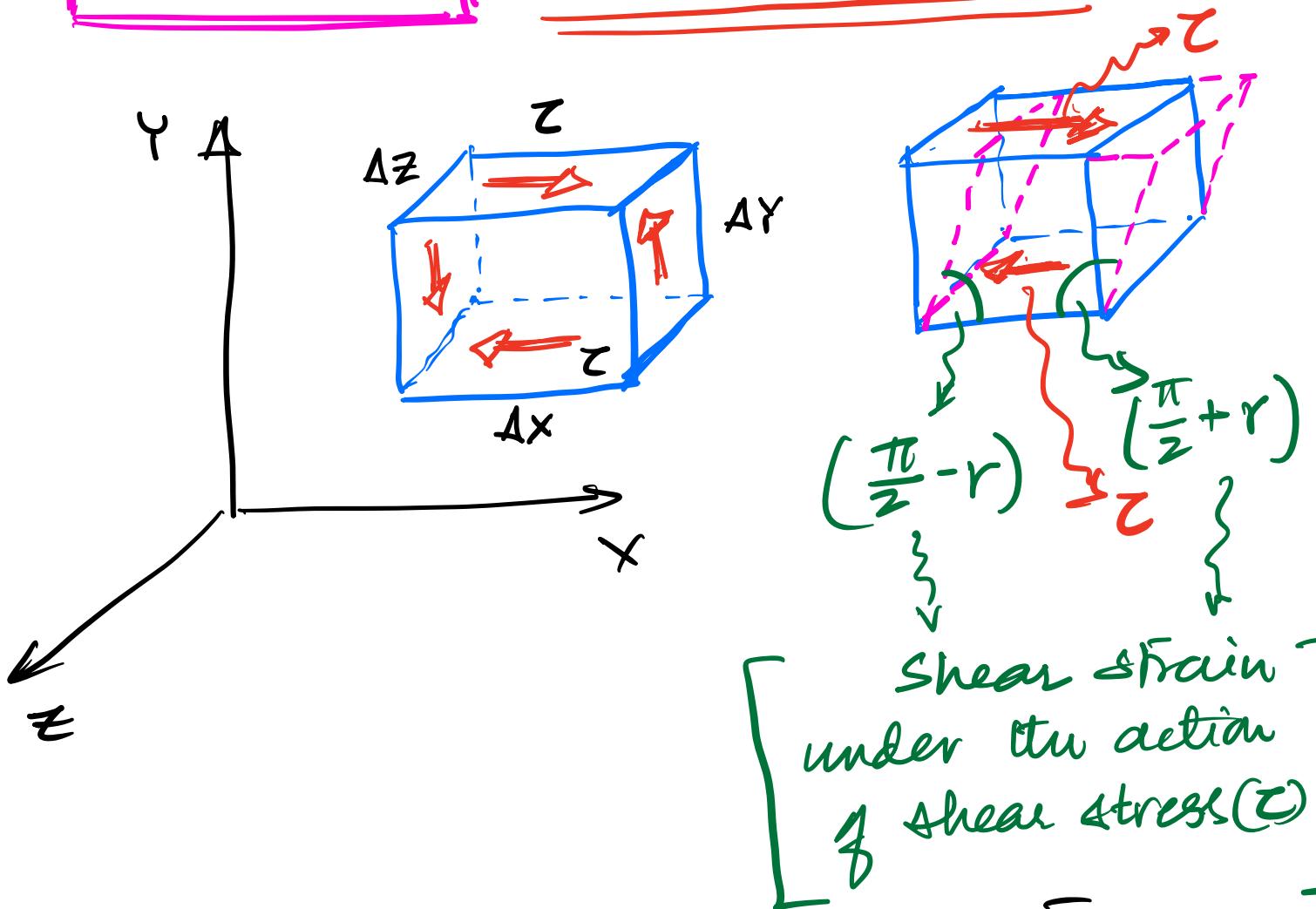
Plane on which it is acting

Direction along the axis it is acting

## KEY IDEA - 9

## Hooke's Law (revisited):

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Recall: Hooke's Law for axial stress

$$T = E \in$$

Axial  
or Normal  
stress

Young's modulus  
or  
elastic modulus

$$T = G_r u$$

# Shear stress

## Shear modulus

## Summarizing so far:

① Stress  $\rightsquigarrow$  Normal stress ( $\sigma$ )  
 $\rightsquigarrow$  Shear stress ( $\tau$ )

② Strain  $\rightsquigarrow$  Axial strain ( $\epsilon_A$ )  
 $\rightsquigarrow$  Lateral strain ( $\epsilon_L$ )  
 $\rightsquigarrow$  Volumetric strain ( $\epsilon$ )  
 $\rightsquigarrow$  Shear strain ( $\gamma$ )

③ Hooke's law:  $\sigma = E \epsilon$   $\rightsquigarrow$  Axial, lateral or volumetric strain  
 $\downarrow$   $\rightsquigarrow$  Normal stress  $\rightarrow$  Young's modulus

$$\tau = G \gamma$$

$\downarrow$

Shear stress      Shear modulus

## ④ Material Properties:

$$E \rightarrow \text{Young's Modulus} \quad E = \frac{\sigma}{\epsilon}$$

$$\gamma \rightarrow \text{Poisson's Ratio}$$

$$\gamma = \epsilon_L / \epsilon_A$$

$$G \rightarrow \text{Shear modulus}$$

$$G = \frac{E}{2(1+\gamma)}$$

$$\gamma \rightarrow \text{Poisson's ratio}$$

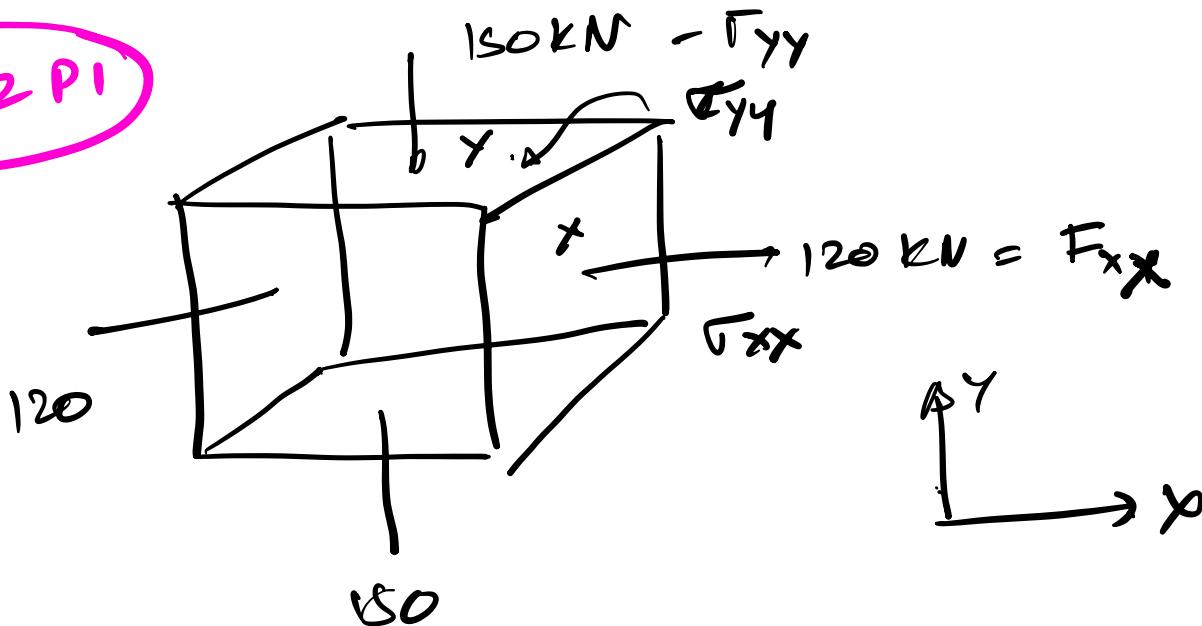
Young's modulus

# HINTS FOR IN CLASS PROBLEM SET - 2

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**NOTE:** ①  $\tau_{xx}$  &  $\tau_{yy}$  are presented as normal & shear stress in the written notes  
 ② They are same as  $\tau_{xx} \neq \tau_{xy}$  in text book or in-class problem set

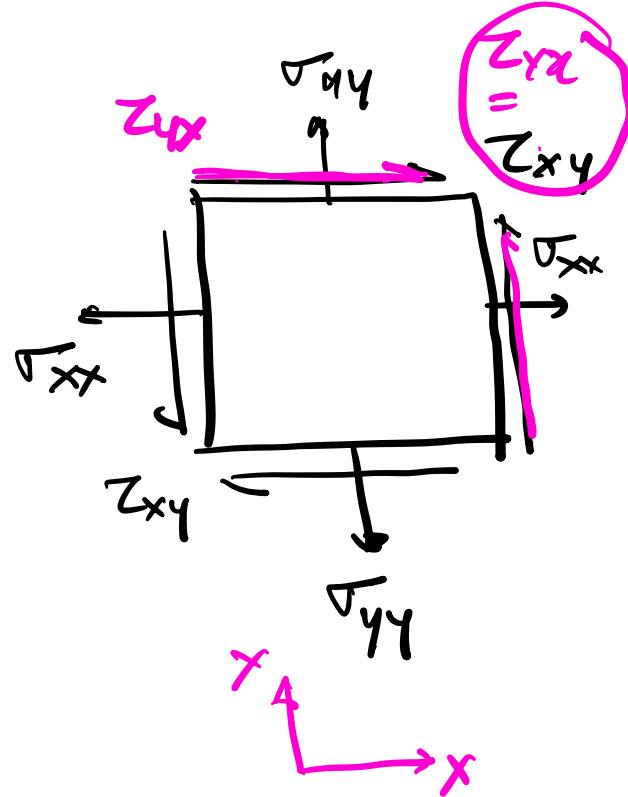
CW2 P1



$$120 \text{ kPa} = \tau_{xx} = \frac{F_{xx}}{A}$$

$$150 \text{ kPa} = \tau_{yy} = \frac{F_{yy}}{A}$$

$$\boxed{\begin{aligned} \tau_{xy} &= \tau_{yx} = 0 \\ z_{xy} &= z_{yx} = 0 \end{aligned}}$$



(N2 P3)

(RB)

$$\sigma_{xx}' = \sigma_{xx} \cos^2 \alpha + 2\sigma_{xy} \sin \alpha \cos \alpha + \sigma_{yy} \sin^2 \alpha$$

$$\sigma_{yy}' = \sigma_{xx} \sin^2 \alpha - 2\sigma_{xy} \sin \alpha \cos \alpha + \sigma_{yy} \cos^2 \alpha$$

$$\sigma_{xy}' = 2 \sin \alpha \cos \alpha \left( \frac{\sigma_{yy} - \sigma_{xx}}{2} \right) + (\cos^2 \alpha - \sin^2 \alpha) \sigma_{xy}$$

(CWQ  
P3)

$$\alpha_p = \frac{1}{2} \tan^{-1} \left( \frac{\sigma_{xy}}{(\sigma_{xx} - \sigma_{yy})/2} \right)$$

$$\alpha_s = \frac{1}{2} \tan^{-1} \left( \frac{\sigma_{yy} - \sigma_{xx}}{2\sigma_{xy}} \right)$$