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************************ Question 1 *********************
Q1 a) Equations of motion of the Non-Linear system:
cart acceleration, ddx =
-(11*m1*sin(theta1)*dtheta1^2 + 12*m2*sin(theta2)*dtheta2^2 - F + g*m1*cos(theta1) ✓
*sin(theta1) - g*m2*cos(theta2)*sin(theta2))/(m1*sin(theta1)^2 + m2*sin(theta2)^2 + <math>\checkmark
pendulum 1 acceleration, ddtheta1 =
-(g*sin(theta1) + (cos(theta1)*(l1*m1*sin(theta1)*dtheta1^2 + 12*m2*sin(theta2) 
*dtheta2^2 - F + g*m1*cos(theta1)*sin(theta1) - <math>g*m2*cos(theta2)*sin(theta2))) / \checkmark
(m1*sin(theta1)^2 + m2*sin(theta2)^2 + M))/11
pendulum 2 acceleration, ddtheta2 =
-(g*sin(theta2) + (cos(theta2)*(l1*m1*sin(theta1)*dtheta1^2 + l2*m2*sin(theta2) 
*dtheta2^2 - F + g*m1*cos(theta1) *sin(theta1) - g*m2*cos(theta2) *sin(theta2))) / \boldsymbol{\iota}
(m1*sin(theta1)^2 + m2*sin(theta2)^2 + M))/12
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Q1 b) State Space representation of our Linearized system:
A Matrix:
[0, 1,
                    0, 0,
                                        0,01
[0, 0,
            -(g*m1)/M, 0,
                                  (q*m2)/M, 01
                    0, 1,
                                        0,0]
[0, 0, -(g + (g*m1)/M)/11, 0,
                             (q*m2)/(M*11), 0]
[0, 0,
                    0, 0,
                                        0, 1]
        -(g*m1)/(M*12), 0, -(g - (g*m2)/M)/12, 0]
[0, 0,
B Matrix:
    1/M
1/(M*11)
1/(M*12)
C Matrix:
    1
         \cap
             0
                  0
    0
         1
             0
                   0
                        0
    0
         0
             1
                  0
                        0
    0
         0
             0
                   1
                        0
        0
                  0
    0
             0
                        1
    0
        0
             0
                   0
                        0
D Matrix:
    0
```

\_\_\_\_\_ Q1 c) Checking conditions on controllability:  $-(g^6*11^2 - 2*g^6*11*12 + g^6*12^2)/(M^6*11^6*12^6)$ \_\_\_\_\_\_ \_\_\_\_\_\_ Q1 d) Substitute the values in the matrices, obtain LQR controller, simulate response  $\checkmark$ and certify local/global stability: Substituting M = 1000 Kg, m1=m2=100 Kg, 11=20m, 12=10m, g=9.8m/s A matrix: 0 1.0000 0 0 0 -0.9800 0 0.9800 0 0 0 0 1.0000 0 -0.5390 0 0.0490 0 1.0000 0 0 0 0 0 0 -0.8820 -0.0980 0 B matrix: 1.0e-03 \* 0 1.0000 0.0500 0.1000 Rank of the controllability matrix : It is full rank matrix, so the system at equilibrium is controllable. The Open loop poles are: 0.0000 + 0.9313i0.0000 - 0.9313i

-0.0000 + 0.7441i

-0.0000 - 0.7441i

0.0000 + 0.0000i

0.0000 + 0.0000i

Desgining Linear Quadratic Regulator...

Q matrix :

•					
2000	0	0	0	0	0
0	10	0	0	0	0
0	0	1500000	0	0	0
0	0	0	10	0	0
0	0	0	0	2000000	0
0	Ο	0	0	0	1.0

R matrix :

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1.0000e-04
LQR optimal gain matrix :
  1.0e+05 *
   0.0447
          0.1426
                   1.1610 0.5137 0.9684 -1.0504
Closed Loop System Matrix (Ac):
Closed Loop poles:
 -2.7678 + 2.8972i
 -2.7678 - 2.8972i
 -0.1197 + 0.7599i
 -0.1197 - 0.7599i
 -0.2757 + 0.3873i
 -0.2757 - 0.3873i
all poles have real negative components, closed loop the system is stable by the arksigma
Lyapunov's indirect method.
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****************** end of Question 1 *****************
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*********************** Question 2 ********************
Q2 e) Check observability conditions:
Checking Observability condition ...
Rank with only x(t) 6
Rank with theta1, theta2: 4
Rank with x, theta2: 6
Rank with x, theta1, theta2: 6
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Qn f) Designing Luenberger Observer..
##### For the Linearized System :
The real part of leftmost pole in the LQR closed loop system is
  -2.7678
Lets choose the desired observer poles to be lesser than that
  -3.8500 -4.7500 -3.5000 -3.4000 -3.1500 -2.9000
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Qn g) Designing LQG Controller..
Kalman Observer Gain:
   0.1448
   0.0105
   0.0063
   0.0008
  -0.0066
  -0.0011
Warning: Simulation will start at a nonzero initial time.
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> In DynamicSystem.checkLsimInputs (line 92)

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In DynamicSystem/lsim (line 67)
In controller (line 305)
****End****
>>
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