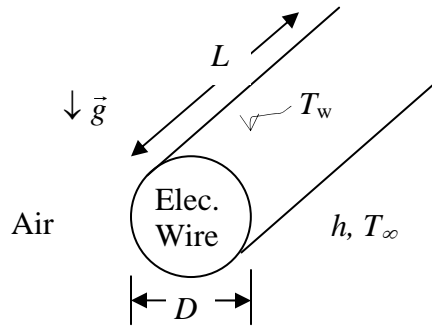


## Solutions - Problem Set # 9

## Problem 1:



**Given:**  $T_\infty = 27^\circ\text{C}$ ; Cable heat dissipation rate:

$$q_{\text{loss}} / L = 30 \text{ W/m}; D = 25 \text{ mm}$$

**Assumptions:** Steady-state free convection heat transfer; Radiation negligible

**NOTE:**

At this stage we do not have information about the wall temperature to calculate the film temperature. Thus, we guess some value for  $T_w$  for example be  $50^\circ\text{C}$  warmer than  $T_\infty$  (i.e., guess value  $T_w = 77^\circ\text{C}$ ) to evaluate the thermophysical properties and the  $Ra_D$  number. We will check this later and improve the results if necessary once we get the actual wall temperature by recalculating the properties at new film temperature.

E-balance on the electrical cable:

$$q_{\text{loss}} = hA(T_w - T_\infty)$$

$$Gr_D = \frac{g\beta[T_w - T_\infty]D^3}{\nu^2}; \quad Ra_D = Pr Gr_D$$

$$T_{\text{film}} = (T_w + T_\infty) / 2$$

Thus, at  $T_{\text{film}} = (T_w + T_\infty) / 2 = (27 + 77) / 2 = 52^\circ\text{C} = 325.15\text{K}$ :

$$T_{\text{film}} = 325.15 \text{ K}; \rho_f = 1.087 \text{ kg/m}^3; c_{p,f} = 1007 \text{ J/kg}\cdot^\circ\text{C}; \mu_f = 19.6 \times 10^{-6} \text{ kg/m}\cdot\text{s}; k_f = 0.028 \text{ W/m}\cdot^\circ\text{C}$$

$$\beta = 1/T = 1/325.15 = 3.0755 \times 10^{-3} \text{ K}^{-1}$$

$$Gr_D = \frac{g\beta[T_w - T_\infty]D^3}{\nu^2} = \frac{9.81 \times 3.0755 \times 10^{-3} \times 50 \times (25 \times 10^{-3})^3}{(19.6 \times 10^{-6} / 1.087)^2} = 72497.3$$

$$Pr_f = \frac{\mu_f c_{p,f}}{k_f} = \frac{19.6 \times 10^{-6} \times 1007}{0.028} = 0.7049$$

$$Ra_D = Pr Gr_D = 72497.3 \times 0.7049 = 51103.347$$

You can use either the Morgan correlation or the Churchill-Chu correlation [Property data at

$$T_{\text{film}} = (T_w + T_\infty) / 2]$$

I have used the Churchill-Chu correlation

$$Nu_{av} \triangleq \left( \frac{h_{av} D}{k_{fluid}} \right) = \left\{ 0.60 + \frac{0.387 Ra_D^{1/6}}{\left[ 1 + (0.559 / Pr)^{9/16} \right]^{8/27}} \right\}^2 \quad 10^{-5} < Ra_D < 10^{12}$$

$$Nu_{av} \triangleq \left( \frac{h_{av} D}{k_{fluid}} \right) = \left\{ 0.60 + \frac{0.387 \times (51103.347)^{1/6}}{\left[ 1 + (0.559 / 0.7049)^{9/16} \right]^{8/27}} \right\}^2 = 6.533$$

$$\frac{h_{av} D}{k_{fluid}} = 7.21695 \Rightarrow h_{av} = \frac{0.028}{25 \times 10^{-3}} \times 6.533 = 7.317 \text{ W/m}^2 \cdot ^\circ\text{C}$$

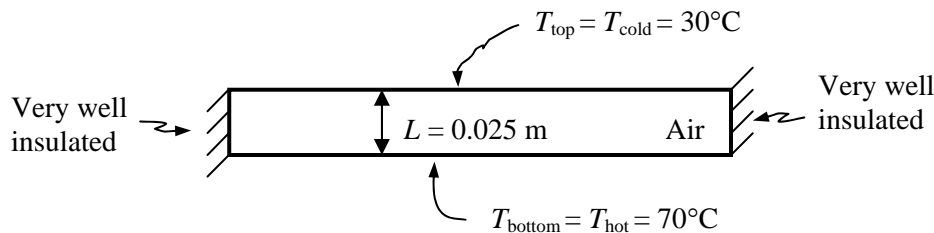
As a result:

$$q_{loss} = hA(T_w - T_\infty) \Rightarrow T_w = T_\infty + \frac{q_{loss}}{hA} = T_\infty + \frac{q_{loss} / L}{h\pi D}$$

$$T_w = 27 + \frac{30}{7.317 \times \pi \times 25 \times 10^{-3}} = 27 + 52.2 = 79.2^\circ\text{C}$$

Thus, in this case, my guess was very close; so, it is not needed to recalculate everything, and we can accept this result. However, if the temperature obtained for the wall was much different than that initially guessed, a new film temperature and  $Ra_D$  number must be calculated. Then a new  $Nu$  and  $h$  value, and from the energy balance equation a new wall temperature should be get. This process may be continued until getting a reasonably acceptable results. Generally speaking, one or two iterations provide enough precision for engineering practices.

### Problem 2:



Air properties at 1 atm, and  $T_{mean} = 50^\circ\text{C}$ :

$$\rho_f = 1.087 \text{ kg/m}^3; c_{p,f} = 1007 \text{ J/kg} \cdot ^\circ\text{C};$$

$$\mu_f = 19.6 \times 10^{-6} \text{ kg/m} \cdot \text{s}; k_f = 0.028 \text{ W/m} \cdot ^\circ\text{C}$$

$$\beta = 1/T = 1/(273.15 + 50) = 3.095 \times 10^{-3} \text{ K}^{-1}$$

**Assumptions:** Steady-state free convection heat transfer; Air behaves as perfect gas;  $\mu$ ,  $k$ ,  $c_p$ , not changing appreciably with pressure (for the range of pressure studied here); neglect radiation

a)

$$Gr_L = \frac{g\beta[T_{hot} - T_{cold}]L^3}{\nu^2} = \frac{9.81 \times 3.095 \times 10^{-3} \times 40 \times (25 \times 10^{-3})^3}{(19.6 \times 10^{-6} / 1.087)^2} = 58365.55$$

$$Pr_f = \frac{\mu_f c_{p,f}}{k_f} = \frac{19.6 \times 10^{-6} \times 1007}{0.028} = 0.7049$$

$$Ra_L = Pr Gr_L = 58365.55 \times 0.7049 = 41141.88 > 1708$$

$$\text{Using the correlation provided in handout \#9: } Nu_{av} \triangleq \left( \frac{h_{av} L}{k_{fluid}} \right) = 0.212 (Ra_L)^{1/4} = 3.019$$

$$h_{av} = \frac{0.028}{25 \times 10^{-3}} \times 3.019 = 3.38 \text{ W/m}^2 \cdot ^\circ\text{C}$$

$$q_{av} = h_{av} [T_h - T_c] = 3.38 \times [70 - 30] = 135.2 \text{ W/m}^2$$

b) In order to eliminate the natural convection (have only heat conduction) in the air gap,

$Ra_L = Pr Gr_L \leq 1708$ , The only property of the air that change with pressure is the mass density, thus,

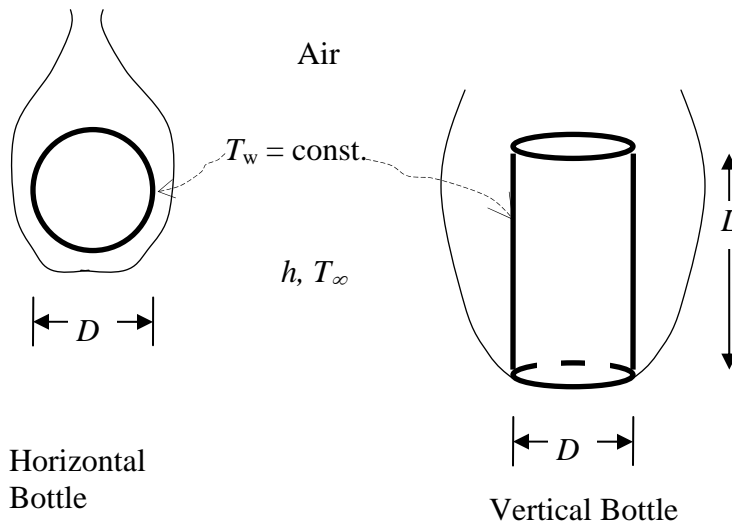
$$Ra_L = Pr Gr_L = Pr \frac{g \beta [T_{hot} - T_{cold}] L^3}{(\mu / \rho)^2} = 0.7049 \times \frac{9.81 \times 3.095 \times 10^{-3} \times 40 \times (25 \times 10^{-3})^3}{(19.6 \times 10^{-6})^2} \rho^2$$

Thus,  $Ra_L = 34819.6 \times \rho^2 = 1708$ ;  $\rho = \left( \frac{1708}{34819.6} \right)^{0.5} = 0.221 \text{ kg/m}^3$ . And from perfect gas

relation at constant temperature for any two states, 1, and 2:  $P_2 / P_1 = \rho_2 / \rho_1$

$$P_2 = P_1 \times \rho_2 / \rho_1 = 1_{atm} \times \frac{0.221}{1.087} = 0.203 \text{ atm}$$

### Problem 3:



The general form of Nusselt number for Natural convection is:  $Nu_{av} \triangleq \left( \frac{h_{av} L_c}{k_{fluid}} \right) = C (Ra_{L_c})^n$

Where  $C$  is a constant and  $n$  is generally a number less than unity. And,  $L_c$  is the characteristic length, depending on geometry and orientation of the object with respect to the gravitational field.

Neglecting the heat transfer from the circular ends and estimating the  $h$  values for both positions considering isothermal vertical walls, we can roughly evaluate the heat transfer from the bottle in both horizontal and vertical positions. For the horizontal bottle, using the symmetry we approximate the  $h$  value with vertical wall of height  $\pi D/2$  ( $L_c = \pi D/2$ ), and for the vertical position we use  $L_c = L$ . Thus,

$$Nu_{av} \triangleq \left( \frac{h_{av} L_c}{k_{fluid}} \right) = C (Ra_{L_c})^n; Gr_{L_c} = \frac{g \beta [T_w - T_\infty] L_c^3}{\nu^2}; Ra_{L_c} = Pr Gr_{L_c}$$

$$Nu_{av} \triangleq \left( \frac{h_{av} L_c}{k_{fluid}} \right) = C \left( \frac{g \beta [T_w - T_\infty] L_c^3}{\nu^2} \frac{\nu}{\alpha} \right)^n$$

$$Nu_{av} \triangleq \left( \frac{h_{av} L_c}{k_{fluid}} \right) = C \left( \frac{g \beta [T_w - T_\infty]}{\nu \alpha} \right)^n L_c^{3n} \Rightarrow h_{av} = \underbrace{C k_{fluid} \left( \frac{g \beta [T_w - T_\infty]}{\nu \alpha} \right)^n}_{\text{constant here } (\xi)} L_c^{3n-1}$$

$$h_{av} = \xi L_c^{3n-1}$$

Horizontal:

$$h_{av} = \xi L_c^{3n-1}$$

$$h_{av, horizontal} = \xi (\pi D / 2)^{3n-1}; h_{av, vertical} = \xi (L)^{3n-1}$$

$$\frac{h_{av, horizontal}}{h_{av, vertical}} = \frac{\xi (\pi D / 2)^{3n-1}}{\xi (L)^{3n-1}} = \left( \frac{\pi D / 2}{L} \right)^{3n-1}$$

$$\text{Also, } L = 4D \Rightarrow \frac{h_{av, horizontal}}{h_{av, vertical}} = \left( \frac{\pi}{8} \right)^{3n-1}$$

$$n = 1/4$$

$$\frac{h_{av, horizontal}}{h_{av, vertical}} \approx \left( \frac{\pi}{8} \right)^{3/4-1} = 1.263$$

**NOTE:**

For air as fluid and assuming bottle wall temperature of  $T_w = 22^\circ\text{C}$  and air temperature of  $T_\infty = 4^\circ\text{C}$ , we can estimate the  $Ra$  number.

$$Ra_L = \left( \frac{g \beta [T_w - T_\infty]}{\nu \alpha} L^3 \right). \text{ With the bottle height of } L = 20 \text{ cm,}$$

and

$$T_f = 13^\circ\text{C}, \rightarrow \nu \approx 15.6 \times 10^{-6} \text{ m}^2/\text{s}; \alpha \approx 22.16 \times 10^{-6} \text{ m}^2/\text{s}$$

$$Ra_L = \left( \frac{g \beta [T_w - T_\infty]}{\nu \alpha} L^3 \right)$$

$$Ra_L = \frac{9.81 \frac{1}{(286)} [22 - 4]}{15.6 \times 10^{-6} \times 22.16 \times 10^{-6}} (0.2)^3 \approx 1.488 \times 10^7$$

For vertical wall as given in the handout #9:

$$Nu_{av} \triangleq \left( \frac{h_{av} L}{k_{fluid}} \right) = 0.59 (Ra_L)^{1/4} \quad 10^4 \leq Ra_L \leq 10^9$$

Thus, horizontal position enhances 26.3% the heat transfer. Please note that for short cylinders, e.g., beer can ( $L/D = 2$ ) the enhancement may be insignificant.