# MECH 325 Gear Stress

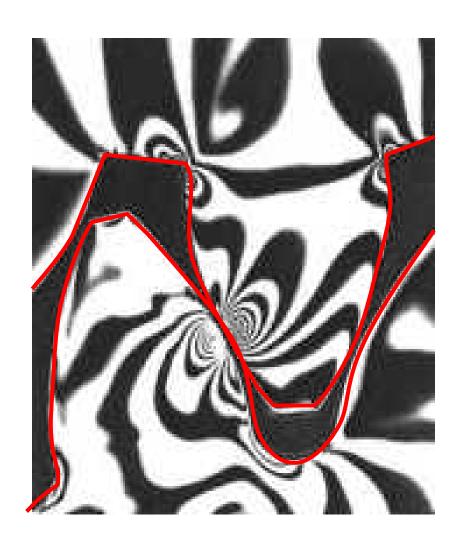


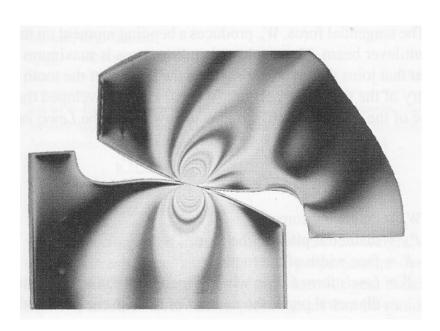
#### **Objectives**

By the end of this section, you should be able to:

- Analyze bending and contact stress failure of spur gears using AGMA standards
- Describe the operation of power screws and cite typical applications
- Describe thread geometry for common types of power screws

#### **Stress in Gears**

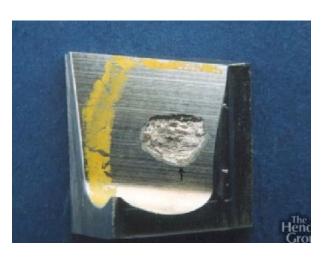




#### **Stress Failures in Gears**



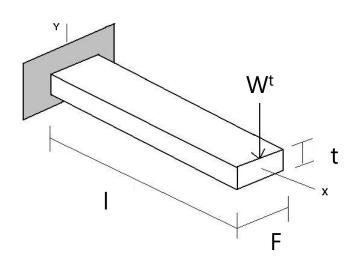




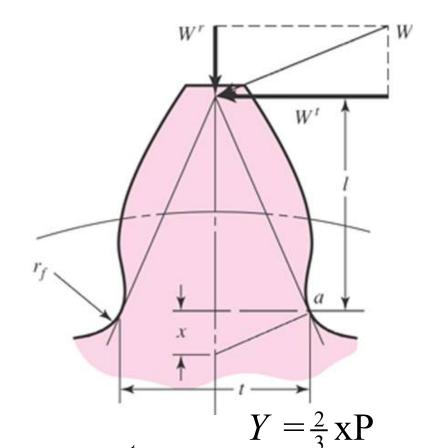


#### Lewis Bending Equation

Cantilever Beam Model



$$\sigma = \frac{6W^t l}{Ft^2}$$



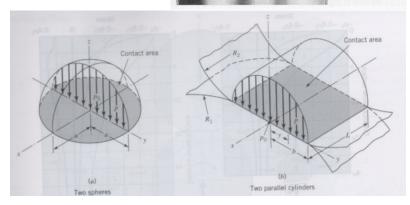
$$\sigma = \frac{W^t P}{FY}$$

Table 14-1

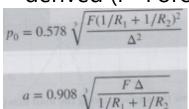
#### **Hertz Contact Stress**

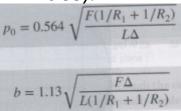
- When curved elastic bodies are in contact, finite contact areas are developed due to deflections in the materials.
- Compressive stresses are developed as a result of these deflections.
- The stress equations are credited to Heinrich Hertz (1881)





 The following stress equations are derived (F=Force, L= Face):



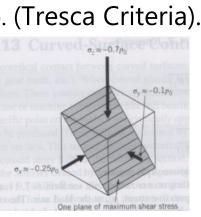


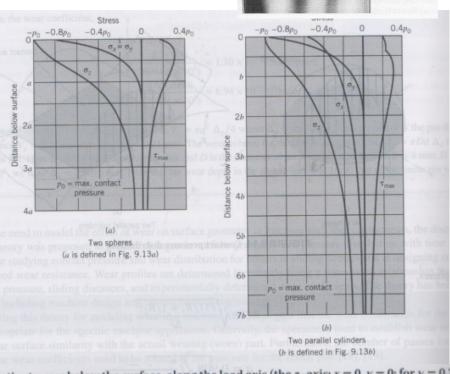
• The contact modulus,  $\Delta$ , defined as;

$$\Delta = \frac{1 - v_1^2}{E_1} + \frac{1 - v_2^2}{E_2}$$

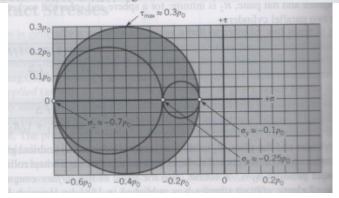
#### **Hertz Contact Stress**

- Vertical contact stresses,  $\sigma_{7}$ , generate transverse compressive stresses,  $\sigma_x$  and  $\sigma_v$  due to Poissons ratio, υ.
- The three compressive stresses are the principal stresses for a stress element and can be plotted as a 3D Mohr Circle.
- Maximum shear stresses occur below the contact surface at a depth of approx. 0.5 the contact width, b. (Tresca Criteria).



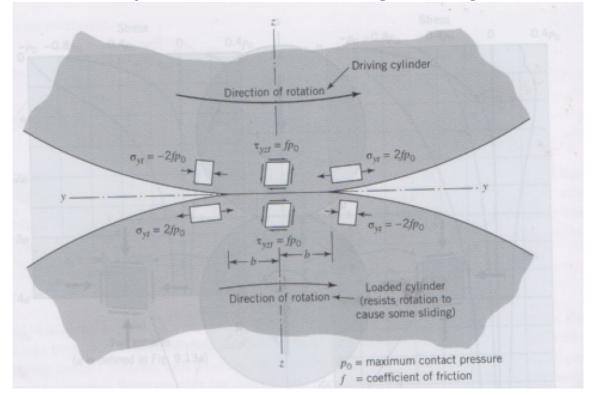


Elastic stresses below the surface, along the load axis (the z-axis; x = 0, y = 0; for v = 0.3).



#### Rolling Friction Stress

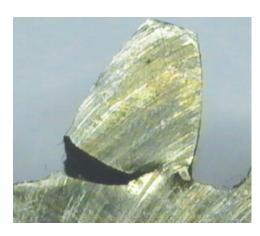
- The Hertz contact stresses are for the static load condtion. With the addition of motion along with friction and sliding, the Hertz contact stresses are not the only stress that must be modelled.
- The stresses are cyclic in nature leading to fatigue failure.



#### AGMA Bending Stress Equations

$$\sigma = W^t K_o K_v K_s \frac{P_d}{F} \frac{K_m K_b}{J} \qquad \sigma_{all} = \frac{S_t}{S_E} \frac{Y_N}{K_T K_R}$$

$$\sigma_{all} = \frac{S_t}{S_F} \frac{Y_N}{K_T K_R}$$

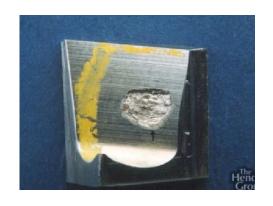


Bending stress safety factor

$$S_F = \frac{S_t}{\sigma} \frac{Y_N}{K_T K_R}$$

#### **AGMA Contact Stress Equations**

$$\sigma_c = C_P \sqrt{W^t K_o K_v K_s \frac{K_m}{d_p F} \frac{C_f}{I}} \qquad \sigma_{c,all} = \frac{S_C}{S_H} \frac{Z_N C_H}{K_T K_R}$$



Contact stress safety factor

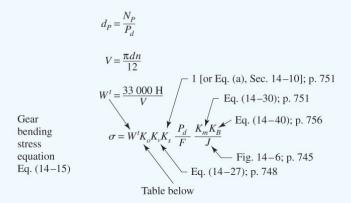
$$S_H = \frac{S_C}{\sigma_C} \frac{Z_N C_H}{K_T K_R}$$

#### **AGMA Empirical Factors**

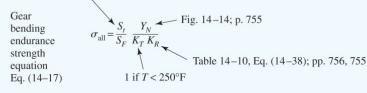
- Not all factors used in every case
- Refer to text for complete details
- C<sub>f</sub> = surface condition factor
- C<sub>H</sub> = hardness factor
- $C_p$  = elastic coefficient
- I = geometry factor
- J = geometry factor
- $K_B = rim thickness factor$
- $K_o$  = overload factor

- $K_m = load distribution factor$
- $K_R$  = reliability factor
- K<sub>s</sub> = size factor
- $K_T$  = temperature factor
- $K_v = dynamic factor$
- $Y_N$  = stress cycle factor

#### SPUR GEAR BENDING Based on ANSI/AGMA 2001-D04 (U.S. customary units)



 $_{0.99}(S_t)_{10}$  Tables 14–3, 14–4; pp. 740, 741



Bending factor of safety  $S_F = \frac{S_t Y_N / (K_T K_R)}{\sigma}$  Eq. (14–41)

Remember to compare  $S_F$  with  $S_H^2$  when deciding whether bending or wear is the threat to function. For crowned gears compare  $S_F$  with  $S_H^3$ .

Table of Overload Factors,  $K_o$ 

Driven Machine				
Power source	Uniform	Moderate shock	Heavy shock	
Uniform	1.00	1.25	1.75	
Light shock	1.25	1.50	2.00	
Medium shock	1.50	1.75	2.25	

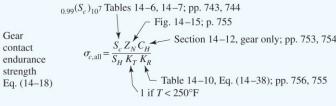


$$d_{p} = \frac{N_{p}}{P_{d}}$$

$$V = \frac{\pi dn}{12}$$

$$W' = \frac{33\ 000\ \text{H}}{V}$$

$$\sigma_{c} = C_{p} \left(W'K_{o}K_{o}K_{o}K_{s}\frac{K_{m}}{d_{p}F}\frac{C_{f}}{I}\right)^{1/2}$$
Eq. (14–23); p. 747
$$Eq. (14–27); p. 748$$
Table below



Wear	✓	r only
factor of	$S_H = \frac{S_c Z_N C_H' / (K_T K_R)}{T_R}$	
safety Eq. (14–42)	$\sigma_{\!_{C}}$	

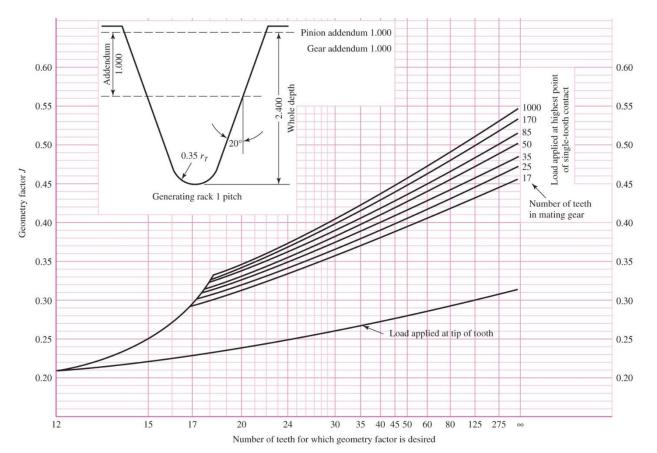
Remember to compare  $S_F$  with  $S_H^2$  when deciding whether bending or wear is the threat to function. For crowned gears compare  $S_F$  with  $S_H^3$ .

Table of Overload Factors,  $K_0$ 

Driven Machine			
Power source	Uniform	Moderate shock	Heavy shock
Uniform	1.00	1.25	1.75
Light shock	1.25	1.50	2.00
Medium shock	1.50	1.75	2.25

### Geometry Factor J

- Accounts for shape of tooth in bending stress equation. It is related to the Lewis Equation. A nominal bending stress is developed.
- The value of J increases with number of teeth as the load is shared.



 $\sigma = \overline{FY}$ 

J replaces Y

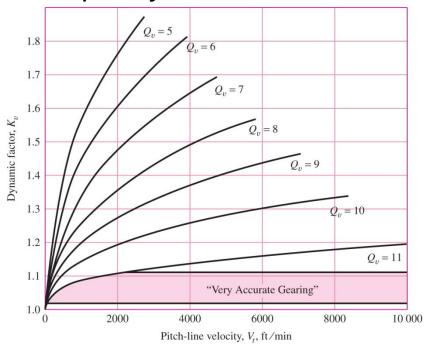
# Overload (Service) Factor K<sub>O</sub>

- To account for likelihood of increase in nominal tangential load due to particular application.
- Recommended values,

Table of Overload Factors, $K_o$				
Driven Machine				
Power source	Uniform	Moderate shock	Heavy shock	
Uniform	1.00	1.25	1.75	
Light shock	1.25	1.50	2.00	
Medium shock	1.50	1.75	2.25	

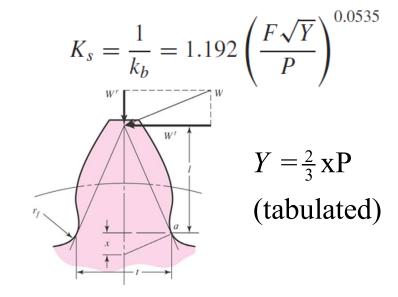
# Dynamic & Size Factor $K_{\nu}$ & $K_{s}$

• Dynamic factor  $K_v$  is a function of tangential speed,  $\omega r$ , and gear quality Q.



- Size factor K<sub>s</sub>
- Typically K<sub>s</sub> = 1

Or use



#### Load-Distribution Factor $K_m$

- Accounts for non-uniform distribution of load across the line of contact.
- Depends on mounting and face width.

$$K_m = C_{mf} = 1 + C_{mc}(C_{pf}C_{pm} + C_{ma}C_e)$$
 (14–30)
$$C_{mc} = \begin{cases} 1 & \text{for uncrowned teeth} \\ 0.8 & \text{for crowned teeth} \end{cases}$$
 (14–31)
$$C_{pf} = \begin{cases} \frac{F}{10d_p} - 0.025 & F \leq 1 \text{ in} \\ \frac{F}{10d_p} - 0.0375 + 0.0125F & 1 < F \leq 17 \text{ in} \end{cases}$$
 (14–32)
$$\frac{F}{10d_p} - 0.1109 + 0.0207F - 0.000228F^2 & 17 < F \leq 40 \text{ in} \end{cases}$$

$$C_e = \begin{cases} 0.8 & \text{for gearing adjusted at assembly, or compatibility} \\ & \text{is improved by lapping, or both} \end{cases}$$
 (14–35)

### Load-Distribution Factor $K_m$

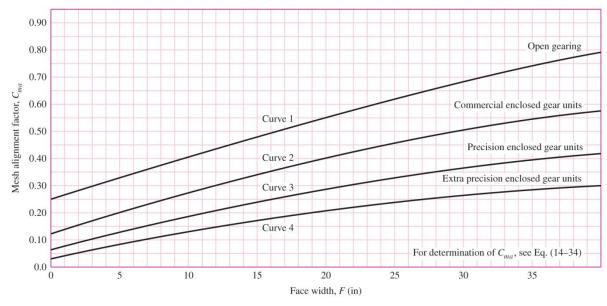
$$K_m = C_{mf} = 1 + C_{mc}(C_{pf}C_{pm} + C_{ma}C_e)$$

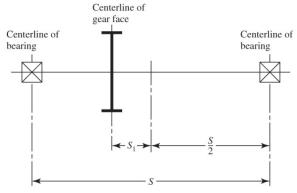
(14 - 30)

$$C_{pm} = \begin{cases} 1\\ 1.1 \end{cases}$$

 $C_{pm} = \begin{cases} 1 & \text{for straddle-mounted pinion with } S_1/S < 0.175 \\ 1.1 & \text{for straddle-mounted pinion with } S_1/S \ge 0.175 \end{cases}$ 

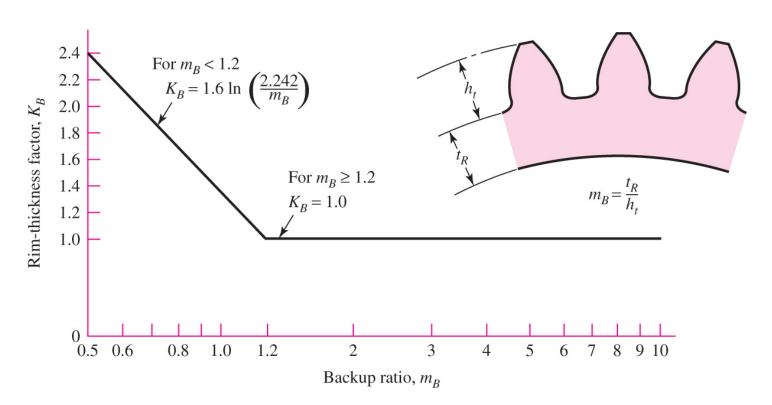






# Rim-Thickness Factor $K_B$

 Accounts for bending of rim on a gear that is not solid



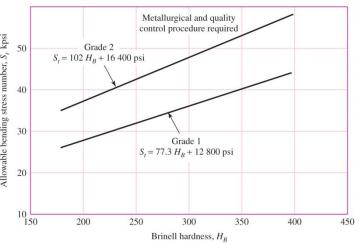
### AGMA Allowable Strengths, S<sub>t</sub>

- AGMA uses allowable stress numbers rather than strengths.
- The gear strength values are only for use with the AGMA stress values, and should not be compared with other true material strengths.

#### **Table 14-3**

Repeatedly Applied Bending Strength  $S_t$  at  $10^7$  Cycles and 0.99 Reliability for Steel Gears Source: ANSI/AGMA 2001-D04.

Material	Heat	Minimum Surface	Allowable Bending Stress	
Designation Designation	Treatment	Hardness <sup>1</sup>	Grade 1	Grade 2
Steel <sup>3</sup>	Through-hardened Flame <sup>4</sup> or induction hardened <sup>4</sup> with type A pattern <sup>5</sup>	See Fig. 14–2 See Table 8*	See Fig. 14–2 45 000	See Fig. 14–2 55 000
	Flame <sup>4</sup> or induction hardened <sup>4</sup> with type B pattern <sup>5</sup>	See Table 8*	22 000	22 000
	Carburized and hardened	See Table 9*	55 000	65 000 or 70 000 <sup>6</sup>
	Nitrided <sup>4,7</sup> (through- hardened steels)	83.5 HR15N	See Fig. 14–3	See Fig. 14–3
Nitralloy 135M, Nitralloy N, and 2.5% chrome (no aluminum)	Nitrided <sup>4,7</sup>	87.5 HR15N	See Fig. 14–4	See Fig. 14–4

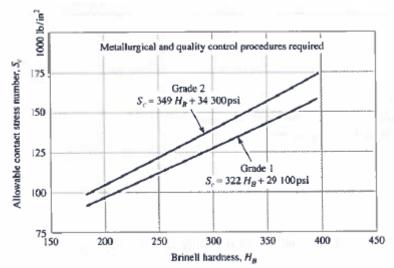


# AGMA Allowable Strengths, S<sub>t</sub>

• The AGMA *allowable stress numbers* are different between bending stress and contact stress.

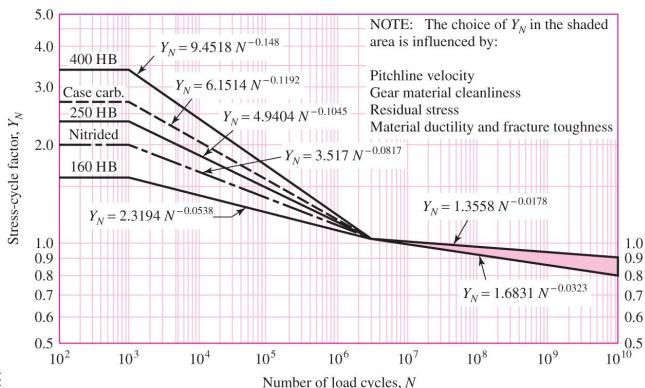
ľ	Table 14-6
ı	Repeatedly Applied Contact Strength S <sub>c</sub> at 10 <sup>7</sup> Cycles and 0.99 Reliability for Steel Gears
L	Source: ANSI/AGMA 2001-D04,

Material	Heat	Minimum Surface	Allowable Contact Stress Number, 2 Sc,		
Designation	Treatment	Hardness <sup>1</sup>	Grade 1	Grade 2 。	Grade 3
Steel <sup>3</sup>	Through hardened <sup>4</sup>	See Fig. 14-5	See Fig. 14-5	See Fig. 14-5	_
	Flame <sup>5</sup> or induction	50 HRC	170 000	190 000	_
	hardened <sup>5</sup>	54 HRC	175 000	195 000	_
	Carburized and hardened <sup>5</sup>	See Table 9*	180 000	225 000	275 000
	Nitrided <sup>5</sup> (through	83.5 HR15N	150 000	163 000	175 000
	hardened steels)	84.5 HR15N	155 000	168 000	180 000
2.5% chrome (no aluminum)	Nitrided <sup>5</sup>	87.5 HR15N	155 000	172 000	189 000
Nitralloy 135M	Nitrided <sup>5</sup>	90.0 HR15N	170 000	183 000	195 000
Nitralloy N	Nitrided <sup>5</sup>	90.0 HR15N	172 000	188 000	205 000
2.5% chrome (no aluminum)	Nitrided <sup>5</sup>	90.0 HR15N	176 000	196 000	216 000



# Stress-Cycle Factors $Y_N$ and $Z_N$

- AGMA strengths are for 10<sup>7</sup> cycles
- Stress-cycle factors account for endurance of the gear train
- Fig. 14–14 gives  $Y_N$  for bending
- Fig. 14–15 gives  $Z_N$  for contact stress



# Reliability Factor $K_R(Y_Z)$

- Accounts for statistical distributions of material fatigue failures
- Does not account for load variation
- Use Table 14–10 (Normally 99% is used  $K_R = 1.00$ )

Reliability	$K_R(Y_Z)$
0.9999	1.50
0.999	1.25
0.99	1.00
0.90	0.85
0.50	0.70

# Temperature Factor $K_T(Y_{\theta})$

- AGMA has not established values for this factor.
- For temperatures up to 250°F (120°C),  $K_T = 1$  is acceptable.

# Safety Factors $S_F$ and $S_H$

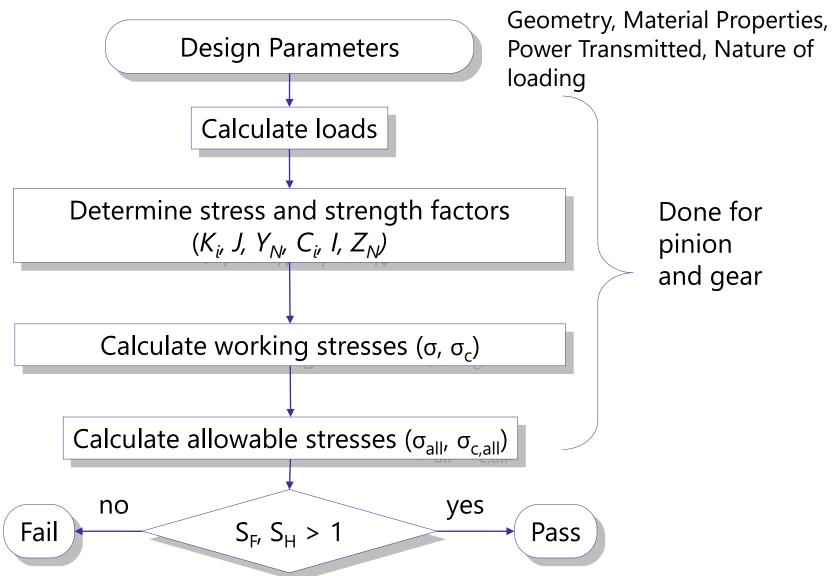
- Included as design factors in the strength equations
- Can be solved for and used as factor of safety

$$S_F = \frac{S_t Y_N / (K_T K_R)}{\sigma} = \frac{\text{fully corrected bending strength}}{\text{bending stress}}$$

$$S_H = \frac{S_c Z_N C_H / (K_T K_R)}{\sigma_c} = \frac{\text{fully corrected contact strength}}{\text{contact stress}}$$
(14-41)

• Or, can set equal to unity, and solve for traditional factor of safety as  $n = \sigma_{alv}/\sigma$ 

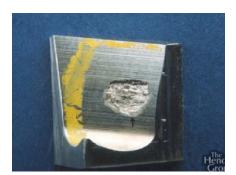
#### Gear Stress Analysis Procedure



# Comparing $S_F$ and $S_H$

- Bending stress is linear with transmitted load.
- Contact stress is not linear with transmitted load
- To compare the factors of safety between the different failure modes, to determine which is critical,
  - Compare  $S_F$  with  $S_H^2$  for linear or helical contact
  - Compare  $S_F$  with  $S_H^3$  for spherical contact





#### Design of a Gear Mesh

1. A priori decisions (established in advance) 2. Choose a diametral pitch, Pd. Examine implications on face width, pitch diameters, and material properties. 3. Choose pinion material. examine hardness requirements 4. Choose gear material. examine hardness requirements Done

#### Design of a Gear Mesh - Details

- 1. A priori decisions (established in advanced)
  - Function: load, speed, reliability, life, overload factor
  - Overall design safety factor (n<sub>d</sub>)
  - Tooth system: φ, ψ, addendum, dedendum, root fillet radius
  - Gear ratio ( $m_G = N_G/N_P$ )
  - Quality number (Q<sub>v</sub>)
- 2. Choose a diametral pitch
  - Initially select median face width  $4\pi/P$
  - Find range of necessary ultimate strengths

#### Design of a Gear Mesh - Details

- 3. Choose pinion material and core hardness
  - Find face width to meet safety factor in bending
  - Choose face width
  - Check safety factor in bending
  - Find necessary contact fatigue strength, S<sub>c</sub>
  - Choose a case hardness
  - Check safety factor in wear
- 4. Choose a gear material and core hardness
  - Check safety factor in bending
  - Choose a case hardness
  - Check safety factor in wear

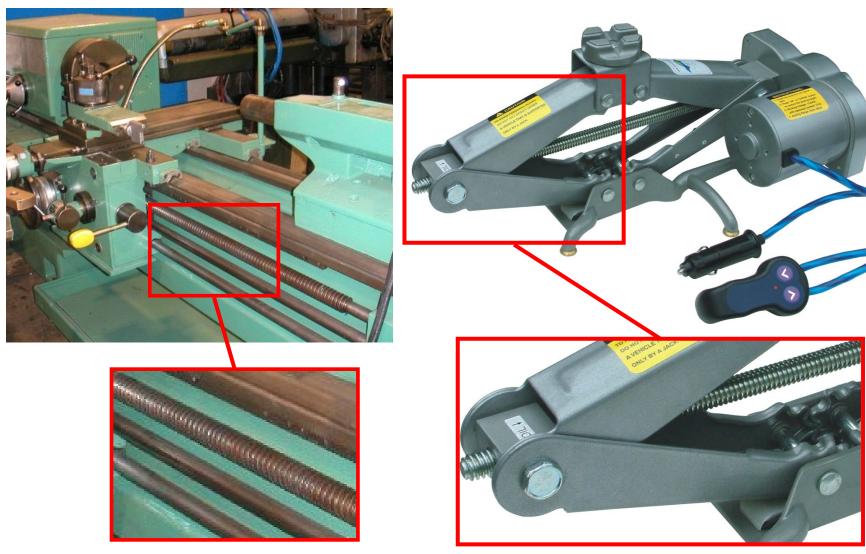
#### Design of a Gear Mesh - Notes

- Iterate at each stage until no decisions are changed
- If you cannot find a reasonable solution at one stage, you need to move back to the previous stage
- This is not a rigid process; you can work in any sequence
- The process is the similar for all gear types

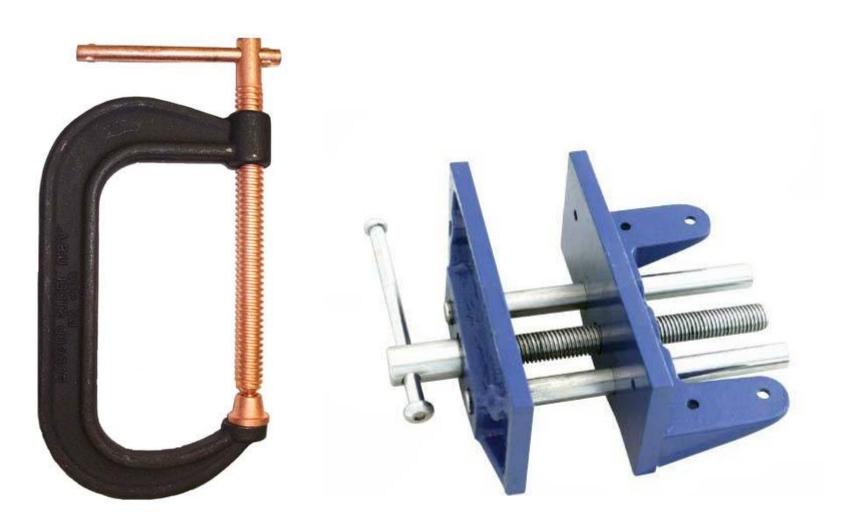
# MECH 325 Power Screw Introduction



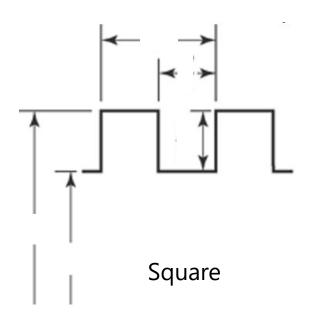
#### **Power Screw Examples**



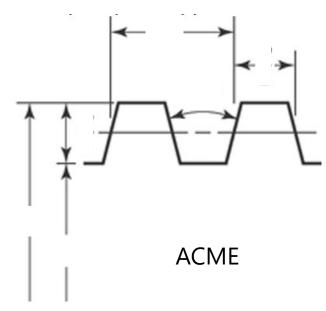
#### **Power Screw Examples**



#### **Thread Types**





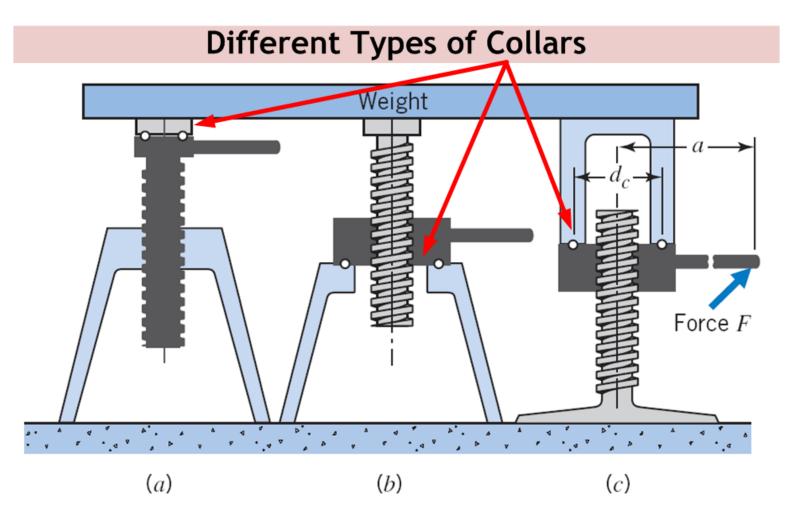




#### Multiple Start Threads

1 Start 2 Starts 01234 4 Starts Pitch Lead

#### Power Screw Configurations



#### Power Screw Configurations

Identify preferred configurations

