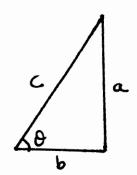
Formula Sheet



$$V_{x}^{2} = V_{0x}^{2} + \lambda \alpha_{x} \Delta X$$

$$\Delta X = \frac{1}{\lambda} (V_{0x} + V_{x}) \Delta t$$

Quadratic equation
$$ay^2 + by + C = 0$$

$$y = -\frac{b \pm \int b^2 - 4ac^7}{\lambda a}$$

$$V = \frac{dx}{dx} = \frac{dx}{dx}$$

$$\frac{qx}{7}(x_{\nu})=v\times_{\nu-1}$$

$$\frac{d}{dx}(x^n) = n \times \frac{1}{n-1} + C = constant of integration$$

Newton's 1st "aw": object @ rest or in motion with a constant relocity stays in that state of motion unless acted upon by a non-ten net force

Newton's 2" "lau": EF = ma

Newton's 3rd "law": Action-reaction

fs, max = usn fs & usn fr= uxn us = welficient of state friction un= coefficiental terretic friction

Projectile motion

x-componen2	y-component
X _o =	Y =
X =	Y=
\(\lambda_k =	Voy =
0x =	مر=
√×=	√ _y =

centripetal acceleration $\alpha_{\nu} = \frac{V^{2}}{\Gamma} = \omega^{2} \Gamma$ apple - Jat + az at = tanden tral acceleration

$$V=\omega\Gamma \qquad \omega=2\pi\Gamma = \frac{2\pi}{T^2}$$

T = Period 5 = frequency W = any low velocity or any frequency

For projectile motion, when appropriate

$$\Delta y = (\tan \theta_0) \delta x - \frac{9(0 \times)^2}{2(v_0 \cos \theta_0)^2}$$
 complimentary angles: $\theta_1 + \theta_2 = 90^\circ$

Of=

Gravitational force For = GMM where "g" = GM R2

For satellites in circular orbits $V = \int \frac{GM}{\Gamma} \qquad TI = \lambda T \int \frac{\Gamma^3}{GM} \qquad \text{where } \Gamma = (R+h)$

$$\vec{t} = \vec{r} \times \vec{F} = r F_{SW} \phi \qquad \hat{c} \cdot \hat{c} = \hat{c} \cdot \hat{c} + \hat{c} \cdot \hat{c} = 1$$

$$\vec{c} \times \hat{c} = \hat{c} \qquad \hat{c} \times \hat{c} = \hat{c} \qquad \hat{c} \times \hat{c} = \hat{c} \times \hat{c} = \hat{c} \times \hat{c} = 0$$

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