

AE - 2022

AI24BTECH11015 - Harshvardhan Patidar

- 1) Consider a conventional subsonic fixed-wing airplane. e is the Oswald efficiency factor and AR is the aspect ratio. Corresponding to the minimum $\left(\frac{C_D}{C_L^{3/2}}\right)$, which of the following relations is true?
- $\frac{C_D}{C_L^2} = \frac{1}{\pi e AR}$
 - $\frac{C_D}{C_L^2} = \frac{4}{\pi e AR}$
 - $\frac{C_D}{C_L} = \frac{1}{\pi e AR}$
 - $\frac{C_D}{\sqrt{C_L}} = \frac{1}{\sqrt{\pi e AR}}$
- 2) A horizontal load F is applied at point R on a two-member truss, as shown in the figure. Both the members are prismatic with cross-sectional area, A_0 , and made of the same material with Young's modulus E . The horizontal displacement of point R is:

Fig. 2.1

- 0
 - $\frac{Fl}{EA_0}$
 - $\sqrt{2} \frac{Fl}{EA_0}$
 - $2 \frac{Fl}{EA_0}$
- 3) Which of the following is **NOT** always true for a combustion process taking place in a closed system?
- Total number of atoms is conserved
 - Total number of molecules is conserved
 - Total number of atoms of each element is conserved
 - Total mass is conserved
- 4) The real function $y = \sin^2(|x|)$ is
- continuous for all x
 - differentiable for all x
 - not continuous at $x = 0$
 - not differentiable at $x = 0$
- 5) A convergent nozzle fed from a constant pressure, constant temperature reservoir, is discharging air to atmosphere at 1 bar (absolute) with choked flow at the exit (marked as Q)
- Flow through the nozzle can be assumed to be isentropic.
- If the exit area of the nozzle is increased while all the reservoir parameters and ambient conditions remain the same, then at steady state

Fig. 5.1

- a) the nozzle will remain choked
 b) the nozzle will be un-choked
 c) the Mach number at section P will increase
 d) the Mach number at section P will decrease
- 6) For a conventional airplane in straight, level, constant velocity flight condition, which of the following condition(s) is/are possible on Euler angles (ϕ, θ, ψ) , angle of attack (α) and the sideslip angle (β) ?
- a) $\phi = 0^\circ, \theta = 2^\circ, \psi = 0^\circ, \alpha = 2^\circ, \beta = 0^\circ$
 b) $\phi = 5^\circ, \theta = 0^\circ, \psi = 0^\circ, \alpha = 2^\circ, \beta = 0^\circ$
 c) $\phi = 0^\circ, \theta = 3^\circ, \psi = 0^\circ, \alpha = 3^\circ, \beta = 5^\circ$
 d) $\phi = 0^\circ, \theta = 5^\circ, \psi = 0^\circ, \alpha = 2^\circ, \beta = 5^\circ$
- 7) Consider a high Earth-orbiting satellite of angular momentum per unit mass (h) and eccentricity e .
 The mass of the Earth is M and G is the universal gravitational constant.
 The distance between satellite's center of mass and the Earth's center of mass is r , the true anomaly is θ , and the phase angle is zero.
 Which of the following statements is/are true?
- a) The trajectory equation is $r = r(\theta) = \frac{|h|}{GM(1+e \cos \theta)}$
 b) The trajectory equation is $r = r(\theta) = \frac{|h|^2}{GM(1+e \cos \theta)}$
 c) (h) is conserved
 d) The sum of potential energy and kinetic energy of the satellite is conserved
- 8) A rocket operates at an absolute chamber pressure of 20 bar to produce thrust, F_1 .
 The hot exhaust is optimally expanded to 1 bar (absolute pressure) using a convergent-divergent nozzle with exit to throat area ratio $\left(\frac{A_e}{A_t}\right)$ of 3.5 and thrus coefficient, $C_{F,1} = 1.42$.
 The same rocket when operated at an absolute chamber pressure of 50 bar produces thrust F_2 and the thrust coefficient is $C_{F,2}$.
 Which of the following statements(s) is/are correct?
- a) $\frac{F_2}{F_1} = 2.5$
 b) $\frac{F_2}{F_1} > 2.5$
 c) $\frac{C_{F,2}}{C_{F,1}} = 1$
 d) $\frac{C_{F,2}}{C_{F,1}} > 1$
- 9) $\mathbf{v} = x^3 \hat{i} + y^3 \hat{j} + z^3 \hat{k}$ is a vector field where $\hat{i}, \hat{j}, \hat{k}$ are the base vectors of a cartesian coordinate system.
 Using the Gauss divergence theorem, the value of the outward flux of the vector field over the surface of a sphere of unit radius centered at the origin is _____ (rounded off to one decimal place).