0.1 Exercise 2 - Fluid tutorial group B

A viscous fluid of constant density, ρ , and kinematic viscosity, ν , flows over a flat plate inclined at an angle α and moving with a constant velocity V_w . The flow is stationary and no pressure gradients are applied. The only body force acting on the fluid is due to gravity, g. You can assume zero velocity component in the direction orthogonal to the plate and negligible air resistance at the interface between the viscous fluid and air (y = h).

0.1.1 Determine by means of the Navier-Stokes equation the velocity profile, u(y)

NSE (x,y):

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left[\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right] + f_x \tag{1}$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial y} + \nu \left[\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right] + f_y \tag{2}$$

- $\frac{\partial}{\partial t} = 0$ steady flow
- v = 0 negligible
- $\frac{\partial}{\partial x}$ no variation in flow along plate length

$$0 + 0 + 0 = 0 + \nu \frac{\partial^2 u}{\partial u^2} + g \sin \alpha \tag{3}$$

$$0 + 0 + 0 = -\frac{1}{\rho} \frac{\partial p}{\partial y} - g \cos \alpha \tag{4}$$

$$\frac{\partial^2 u}{\partial u^2} = -\frac{g\sin(\alpha)}{\nu} \tag{5}$$

$$\frac{\partial p}{\partial y} = -\rho g \cos \alpha \tag{6}$$

Integrating equation (5) once:

$$\int \left(\frac{\partial^2 u}{\partial y^2}\right) dy = \int \left(-\frac{g \sin \alpha}{\nu}\right) dy \frac{\partial u}{\partial y} = -\frac{g \sin \alpha}{\nu} y + C \tag{7}$$

Apply the following boundary equation, due to 0 air resistance at y=h boundary:

$$\tau(h) = \mu \frac{\partial u}{\partial y} \bigg|_{y=h} = 0 : 0 = -\frac{g \sin \alpha}{\nu} h + CC = \frac{g \sin \alpha}{\nu} h \tag{8}$$

Integrate equation (8) once more:

$$\int \left(\frac{\partial u}{\partial y}\right) dy = \int \left(-\frac{g \sin \alpha}{\nu} y + \frac{g \sin \alpha}{\nu} h\right) dy \tag{9}$$

$$u(y) = -\frac{g\sin\alpha}{2\nu}y^2 + \frac{g\sin\alpha}{\nu}hy + D \tag{10}$$

$$y(0) = -V_w = D \tag{11}$$

$$u(y) = -\frac{g\sin\alpha}{2\nu}y^2 + \frac{g\sin\alpha}{\nu}hy - V_w \tag{12}$$

$$u(y) = \frac{g \sin \alpha}{\nu} y \left(h - \frac{1}{2} y \right) - V_w \tag{13}$$

0.1.2 If the net flow rate across the fluid height is zero, determine the corresponding plate velocity, V_{wo} , as a function of α

Volume flow rate:

$$\int_{o}^{h} u(y) \, \mathrm{d}y = \int_{o}^{h} \left(\frac{g \sin \alpha}{\nu} \left(hy - \frac{1}{2} y^{2} \right) - V_{w} \right) \mathrm{d}y \tag{14}$$

$$= \left[\frac{g \sin \alpha}{\nu} y \left(\frac{hy^2}{2} - \frac{1}{6} y^3 \right) - V_w y \right]_0^h \tag{15}$$

$$=\frac{g\sin\alpha h^3}{3\nu} - V_w h\tag{16}$$

This is equal to 0 for V_{wo} , hence:

$$V_{wo} = \frac{g \sin \alpha h^2}{3\nu} \tag{17}$$

0.1.3 For the condition found above determine the y coordinate corrsponding to y=0. Sketch three velocity profiles for $V_{wo} < V_w$, $V_{w0} = V_w$ and $V_{wo} > V_w$ and comment them

Combining above equations for u(y), we get:

$$u(y) = \frac{g \sin \alpha}{\nu} y \left(-\frac{1}{2} y^2 + hy - \frac{1}{3} h^2 \right)$$
 (18)

Set this equal to 0 and solve:

$$0 = -\frac{1}{2}y^2 + hy - \frac{1}{3}h^2 \tag{19}$$

$$y = \frac{3 - \sqrt{3}}{3} \cdot h \tag{20}$$

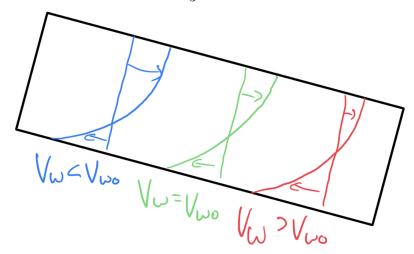


Figure 1: We can see that in the blue one velocity is increased towards the air boundary. As we move along to the point where $V_w = V_{wo}$, we see that the point where u = 0 is $\frac{1}{2}h$. Finally we see the velocity highest close to the plate boundary.

0.1.4 If the flat plate surface is S, determine the force, F, and power, P, required to move it with constant velocity V_{wo}

Using tensor equation:

$$\tau = \left. \mu \frac{\partial u}{\partial y} \right|_{y=0} = \rho g h \sin \alpha \tag{21}$$

Power is simply $F \times A \times V$:

$$P = \tau S V_w = \rho g h \sin \alpha S V_w \tag{22}$$