

## 0.1 Thin-walled cylinders

**Thin walled cylinders** have a wall thickness  $t$  much smaller than the cylinder radius  $R$  (at least one twentieth).

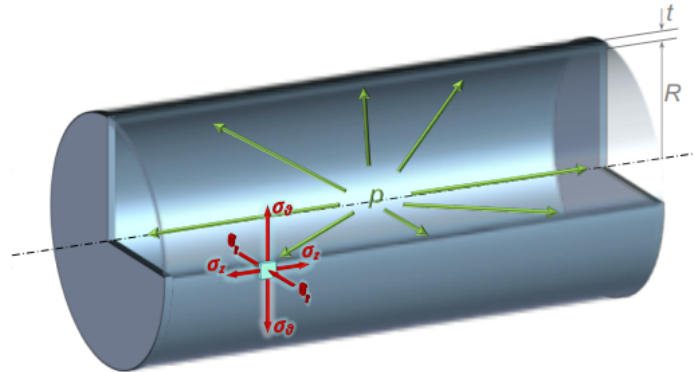


Figure 1:

The **hoop stresses**  $\sigma_\theta$  can be considered **constant** across the wall thickness  $t$ . The **radial stresses**  $\sigma_r$  is **negligible** in comparison to the hoop stresses  $\sigma_\theta$ . Under these conditions, the state of stress at each point of the cylinder can be estimated to good accuracy by simple equilibrium considerations.

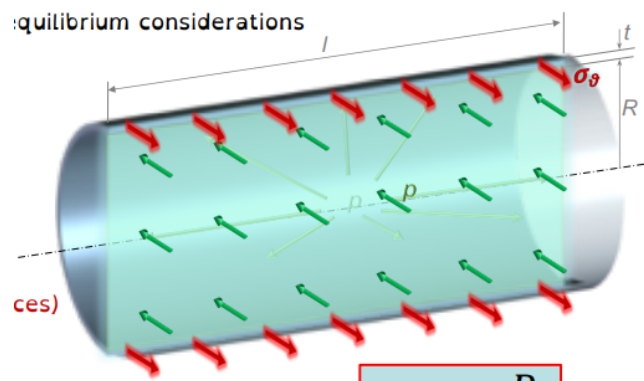


Figure 2: Hoop stress - equilibrium of transversal forces.

$$p \cdot l \cdot 2R = 2(\sigma_\theta t l) \sigma_\theta = \frac{pR}{t} \quad (1)$$

Under these conditions, the state of stress at each point of the cylinder can be estimated to good accuracy by simple equilibrium considerations.

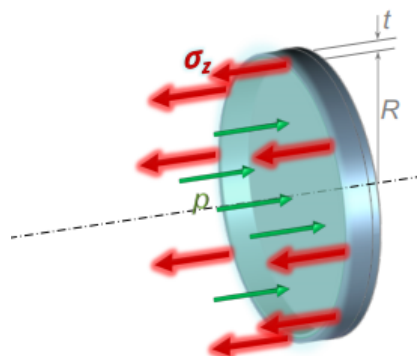


Figure 3: Longitudinal stress - equilibrium of axial forces.

$$p \cdot \pi R^2 = \sigma_z \cdot 2\pi R \cdot t \quad (2)$$

$$\sigma_z = \frac{pR}{2t} \quad (3)$$

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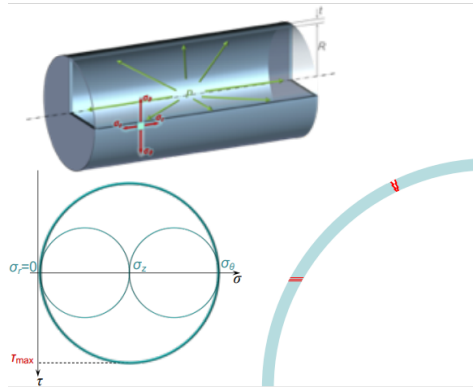


Figure 4:

$$\sigma_\theta = \frac{pR}{t} \quad (4)$$

$$\sigma_z = \frac{pR}{2t} = \frac{\sigma_\theta}{2} \quad (5)$$

## 0.2 Thick-walled cylinders - radial and hoop stresses

### 0.2.1 Failure of thick wall cylinders



Figure 5:

When thickness increases with respect to radius, the radial variation of hoop stress becomes considerable and the radial stress is no longer available.

## 0.2.2 Radial and hoop stresses & strains

For the determination of radial and circumferential stresses and strains, equilibrium equations on their own are not sufficient.

## 0.2.3 Solid mechanics equations

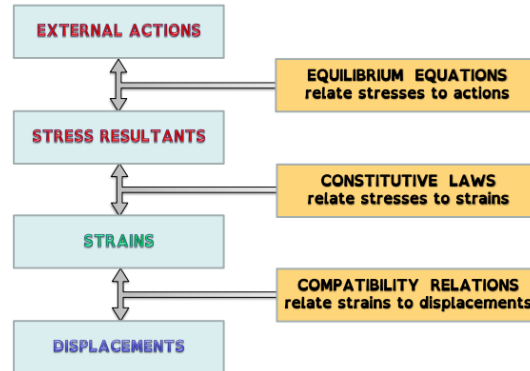


Figure 6:

## 0.2.4 Assumptions

For long cylinder, far from the ends, we can assume that the deformation of the cylinder is symmetrical respect to the axis:

- Cross-sections remain plane when subjected to pressure: longitudinal strain  $\epsilon_z$  is independent of radius. i.e.  $\epsilon_z = \text{constant}$
- Displacements in each cross-section are purely radial
- Displacement is constant with circumferential co-ordinate and varies only with radial co-ordinate i.e.  $u = f(r)$
- $\sigma_r$ ,  $\sigma_\theta$  and  $\sigma_z$  are principal stress
- Cross-sections remain plane when subjected to pressure: longitudinal strain  $\epsilon_z$  is independent of radius. i.e.  $\epsilon_z = \text{constant}$

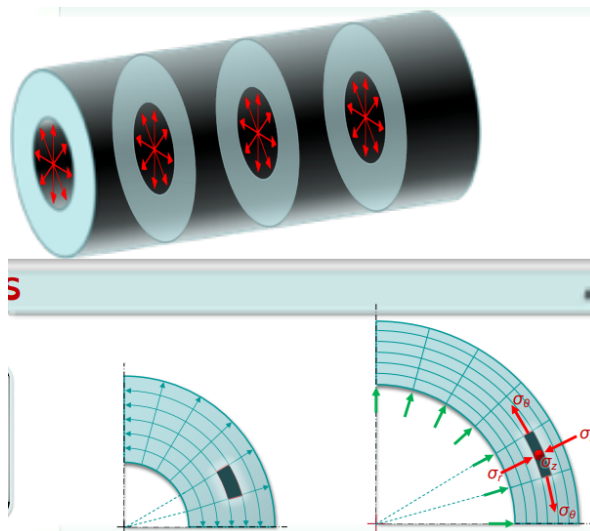


Figure 7:

### 0.2.5 Equilibrium equations

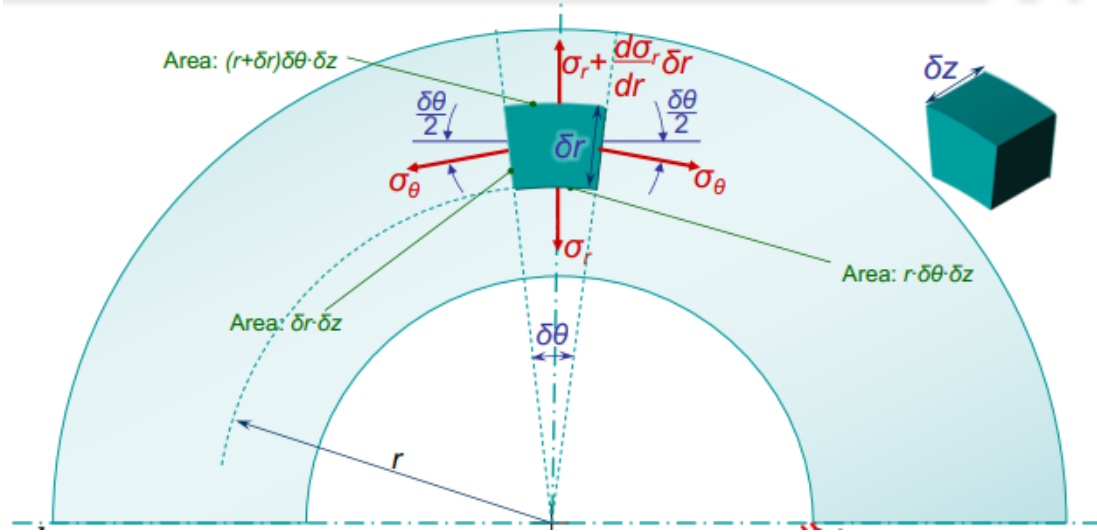


Figure 8:

The equilibrium of radial forces is given by the following equation:

$$-\sigma_r (r \cdot \delta\theta \cdot \delta z) + \left( \sigma_r + \frac{d\sigma_r}{dr} \delta r \right) [(r + \delta r) \cdot \delta\theta \cdot \delta z] - 2\sigma_\theta (\delta r \cdot \delta z) \cdot \sin \frac{\delta\theta}{2} = 0 \quad (6)$$

$$\rightarrow -\cancel{\sigma_r r} + \cancel{\sigma_r r} + \sigma_r \delta r + r \frac{d\sigma_r}{dr} \delta r + \cancel{\frac{d\sigma_r}{dr} \delta r^2} - \sigma_\theta \delta r = 0 \quad (7)$$

$$\rightarrow \sigma_r + r \frac{d\sigma_r}{dr} - \sigma_\theta = 0 \quad (8)$$

### 0.2.6 Compatibility relations

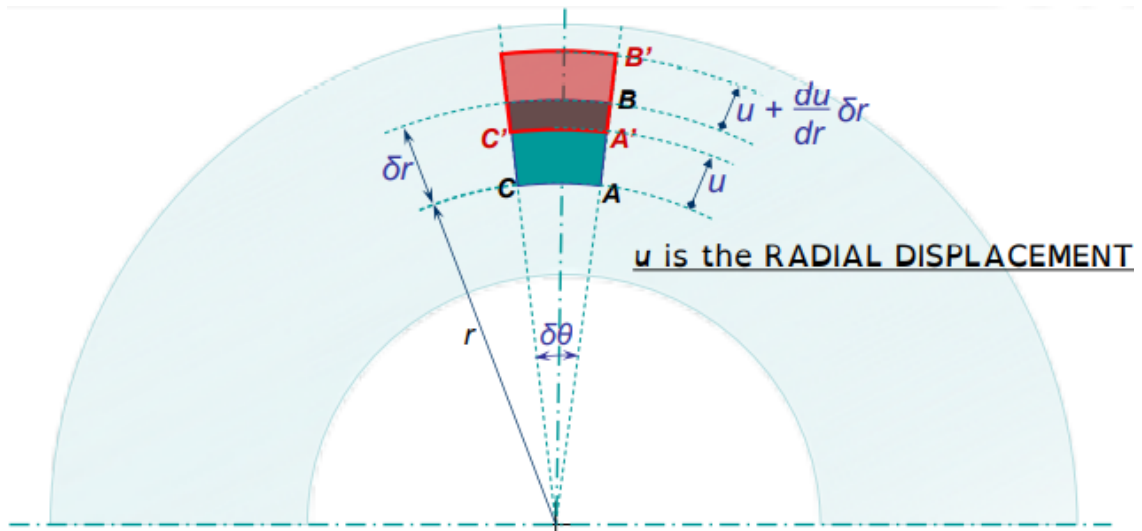


Figure 9:

The radial strain is given by the following equation:

$$\epsilon_r = \frac{\overline{A'B'} - \overline{AB}}{\overline{AB}} \quad (9)$$

with:

$$\overline{AB} = \delta r \quad (10)$$

and

$$\overline{A'B'} = \overline{AB} + \overline{BB'} - \overline{AA'} = \delta r + \left( \mathcal{K} + \frac{du}{dr} \delta r \right) - \mathcal{K} = \delta r + \frac{du}{dr} \delta r \quad (11)$$

Leading to:

$$\epsilon_r = \frac{\left( \cancel{\delta r} + \frac{du}{dr} \cancel{\delta r} \right) - \cancel{\delta r}}{\cancel{\delta r}} \quad (12)$$

$$\epsilon_\theta = \frac{u}{r} \quad (13)$$

### 0.2.7 Constitutive laws

Uniaxial:

$$\begin{aligned} \epsilon_1 &= \frac{\sigma_1}{E} & \sigma_1 &= E\epsilon_1 \\ \epsilon_2 &= -\nu\epsilon_1 & \sigma_2 &= 0 \\ \epsilon_3 &= -\nu\epsilon_1 & \sigma_3 &= 0 \end{aligned} \quad (14)$$

Biaxial:

$$\begin{aligned} \epsilon_1 &= \frac{1}{E} (\sigma_1 - \nu\sigma_2) & \sigma_1 &= \frac{E}{1-\nu^2} (\epsilon_1 + \nu\epsilon_2) \\ \epsilon_2 &= \frac{1}{E} (\sigma_2 - \nu\sigma_1) & \sigma_2 &= \frac{E}{1-\nu^2} (\epsilon_2 + \nu\epsilon_1) \\ \epsilon_3 &= -\frac{\nu}{E} (\sigma_1 + \sigma_2) & \sigma_3 &= 0 \end{aligned} \quad (15)$$

Triaxial:

$$\begin{aligned} \epsilon_1 &= \frac{1}{E} [\sigma_1 - \nu(\sigma_2 + \sigma_3)] & \sigma_1 &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_1 + \nu(\epsilon_2 + \epsilon_3)] \\ \epsilon_2 &= \frac{1}{E} [\sigma_2 - \nu(\sigma_3 + \sigma_1)] & \sigma_2 &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_2 + \nu(\epsilon_1 + \epsilon_3)] \\ \epsilon_3 &= \frac{1}{E} [\sigma_3 - \nu(\sigma_1 + \sigma_2)] & \sigma_3 &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_3 + \nu(\epsilon_1 + \epsilon_2)] \end{aligned} \quad (16)$$

The analytical solution of the stress distribution is possible, in simple form, only in the case of *biaxial state of stress*.

### Cylinders with internal pressure

A pressure difference can be maintained inside the cylinder in one of the following methods:

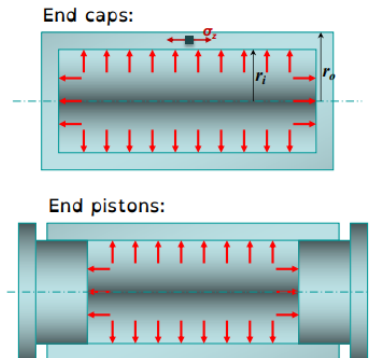


Figure 10:

Only in the case of end pistons, can we find an analytical solution for the stress distributions in thick cylinders: no axial force acts on the cylinder and the state of stress is plane.

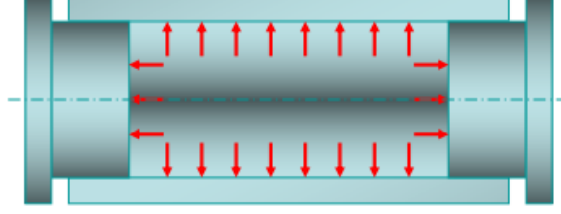


Figure 11:

### Plane stress state

Biaxial:

$$\begin{aligned} \varepsilon_1 &= \frac{1}{E} (\sigma_1 - \nu \sigma_2) & \sigma_1 &= \frac{E}{1-\nu^2} (\varepsilon_1 + \nu \varepsilon_2) \\ \varepsilon_2 &= \frac{1}{E} (\sigma_2 - \nu \sigma_1) & \sigma_2 &= \frac{E}{1-\nu^2} (\varepsilon_2 + \nu \varepsilon_1) \\ \varepsilon_3 &= -\frac{\nu}{E} (\sigma_1 + \sigma_2) & \sigma_3 &= 0 \end{aligned} \quad (17)$$

### Cylindrical

In the biaxial case, in a cylindrical coordinate system  $r, \theta, z$ .

$$\begin{aligned} \varepsilon_r &= \frac{1}{E} (\sigma_r - \nu \sigma_\theta) & \sigma_r &= \frac{E}{1-\nu^2} (\varepsilon_r + \nu \varepsilon_\theta) \\ \varepsilon_\theta &= \frac{1}{E} (\sigma_\theta - \nu \sigma_r) & \sigma_\theta &= \frac{E}{1-\nu^2} (\varepsilon_\theta + \nu \varepsilon_r) \\ \varepsilon_z &= -\frac{\nu}{E} (\sigma_r + \sigma_\theta) & \sigma_z &= 0 \end{aligned} \quad (18)$$

## 0.2.8 Solid mechanics equations

Equilibrium equations relate stresses to actions:

$$\frac{d\sigma_r}{dr} + \frac{\sigma_r - \sigma_\theta}{r} = 0 \quad (19)$$

Constitutive laws relate stresses to strains:

$$\sigma_r = \frac{E}{1-\nu^2} (\varepsilon_r + \nu \varepsilon_\theta) \quad (20)$$

$$\sigma_\theta = \frac{E}{1-\nu^2} (\varepsilon_\theta + \nu \varepsilon_r) \quad (21)$$

Compatibility relations relate strains to displacements:

$$\varepsilon_r = \frac{du}{dr} \quad (22)$$

$$\varepsilon_\theta = \frac{u}{r} \quad (23)$$

Combining the equations together, we arrive at Euler's differential equation:

$$\frac{d^2 u}{dr^2} + \frac{1}{r} \cdot \frac{du}{dr} - \frac{u}{r^2} = 0 \quad (24)$$

$$r^2 u'' + r u' - u = 0 \quad (25)$$

Euler's differential equation is characterised by the fact that its coefficients depend on the variable  $r$ . The general solution to Euler's differential equation is:

$$u = Ar + \frac{B}{r}u' = A - \frac{B}{r^2} \quad (26)$$

The constitutive laws hence become:

$$\sigma_r = \frac{E}{1-v^2} \left( \frac{du}{dr} + v \frac{u}{r} \right) \quad (27)$$

$$\sigma_\theta = \frac{E}{1-v^2} \left( \frac{u}{r} + v \frac{du}{dr} \right) \quad (28)$$

$$\sigma_r = \frac{E}{1-v^2} \left[ A - \frac{B}{r^2} + v \left( A + \frac{B}{r^2} \right) \right] = C - \frac{D}{r^2} \quad (29)$$

$$\sigma_\theta = \frac{E}{1-v^2} \left[ A + \frac{B}{r^2} + v \left( A - \frac{B}{r^2} \right) \right] = C + \frac{D}{r^2} \quad (30)$$

Note that:

$$\sigma_r + \sigma_\theta = 2C = \text{const} \quad (31)$$

## Boundary conditions

$$\sigma_r = C - \frac{D}{r^2} = \frac{p_i r_i^2 - p_0 r_0^2}{r_0^2 - r_i^2} - \frac{(p_i - p_0) r_i^2 r_0^2}{(r_0^2 - r_i^2) r^2} \quad (32)$$

$$\sigma_\theta = C + \frac{D}{r^2} = \frac{p_i r_i^2 - p_0 r_0^2}{r_0^2 - r_i^2} + \frac{(p_i - p_0) r_i^2 r_0^2}{(r_0^2 - r_i^2) r^2} \quad (33)$$

If we have an internal pressure ' $p_i$ ' and an external pressure is ' $p_0$ ':

$$\sigma_r(r_i) = -p_i = C - \frac{D}{r_i^2} \quad (34)$$

$$C = \frac{p_i r_i^2 - p_0 r_0^2}{r_0^2 - r_i^2} \quad (35)$$

$$\sigma_r(r_i) = -p_0 = C - \frac{D}{r_0^2} \quad (36)$$

$$D = \frac{(p_i - p_0) r_i^2 r_0^2}{(r_0^2 - r_i^2) r^2} \quad (37)$$

## Radial and hoop stresses

Lame's equations

$$\sigma_r = \frac{p_i r_i^2 - p_0 r_0^2}{r_0^2 - r_i^2} - \frac{(p_i - p_0) r_i^2 r_0^2}{(r_0^2 - r_i^2) r^2} \quad (38)$$

$$\sigma_\theta = \frac{p_i r_i^2 - p_0 r_0^2}{r_0^2 - r_i^2} + \frac{(p_i - p_0) r_i^2 r_0^2}{(r_0^2 - r_i^2) r^2} \quad (39)$$

## 0.2.9 Stress variation

General case

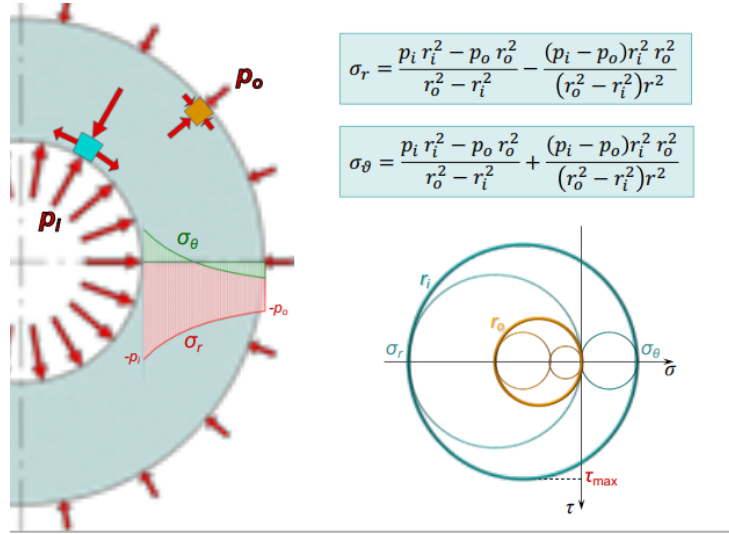


Figure 12:

$p_i$  only

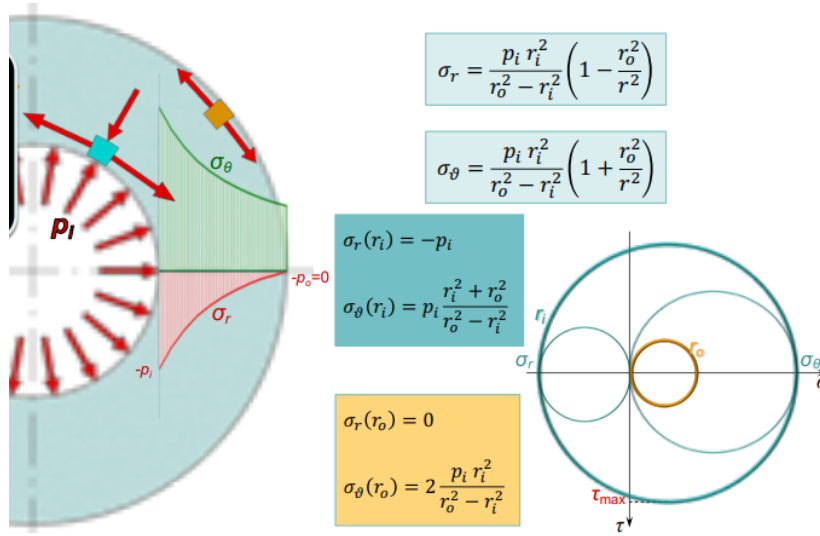


Figure 13:



$p_0$  only

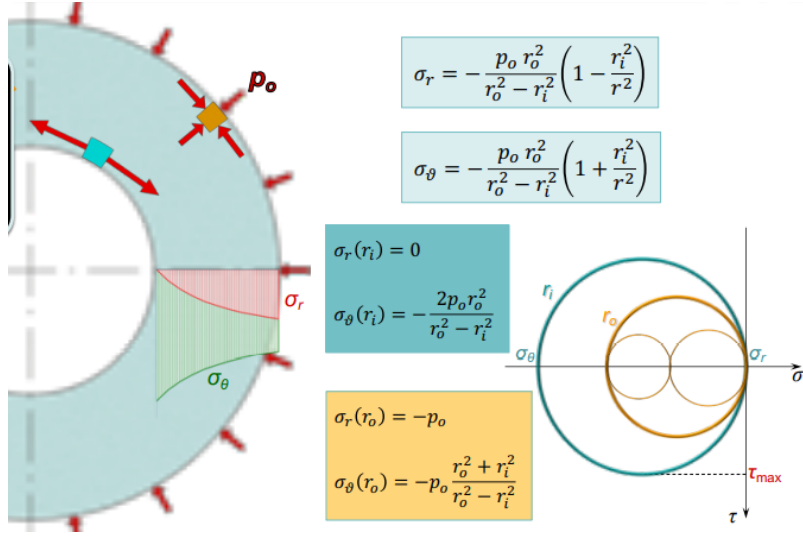


Figure 14:

### 0.3 Thick-walled cylinders - longitudinal stress

How do we deal with the case of end caps pistons? (Triaxial stress state). We know:

$$\sigma_r = C - \frac{D}{r^2} \quad (40)$$

$$\sigma_\theta = C + \frac{D}{r^2} \quad (41)$$

$$\sigma_r + \sigma_\theta = 2C = \text{const} \quad (42)$$

Triaxial stress state:

$$\begin{aligned} \varepsilon_r &= \frac{1}{E} [\sigma_r - \nu (\sigma_\theta + \sigma_z)] & \sigma_r &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu) \varepsilon_r + \nu (\varepsilon_\theta + \varepsilon_z)] \\ \varepsilon_\theta &= \frac{1}{E} [\sigma_\theta - \nu (\sigma_z + \sigma_r)] & \sigma_\theta &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu) \varepsilon_\theta + \nu (\varepsilon_r + \varepsilon_z)] \\ \varepsilon_z &= \frac{1}{E} [\sigma_z - \nu (\sigma_r + \sigma_\theta)] & \sigma_z &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu) \varepsilon_z + \nu (\varepsilon_\theta + \varepsilon_r)] \end{aligned} \quad (43)$$

We know that  $\varepsilon_z$ ,  $\frac{1}{E}$  and  $\nu (\sigma_r + \sigma_\theta)$  are constant, therefore  $\sigma_z$  is also constant. Longitudinal stresses can be estimated by simple equilibrium considerations.

$$p \cdot \pi r_i^2 = \sigma_z (\pi r_o^2 - \pi r_i^2) \quad (44)$$

$$\sigma_z = p \frac{r_i^2}{r_o^2 - r_i^2} \quad (45)$$

with  $p = p_i - p_o$ . In the case of end pistons, the longitudinal component of the stress can be superimposed.

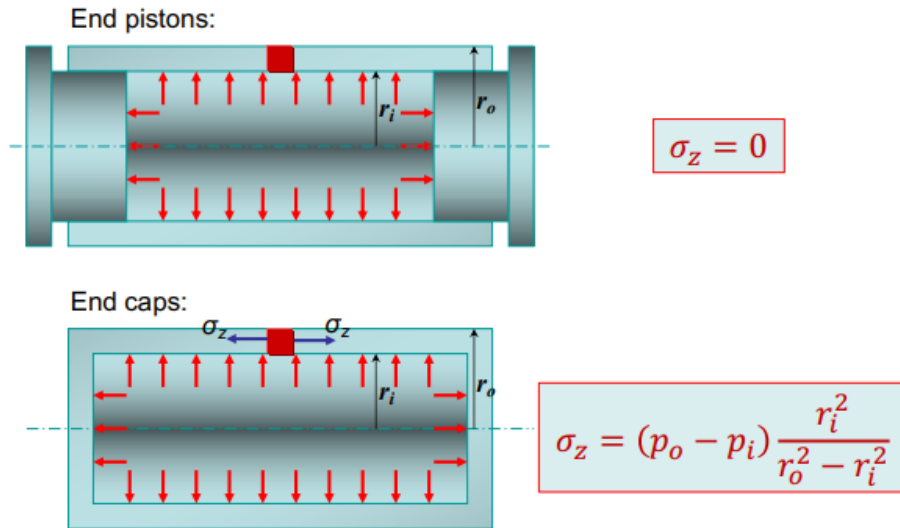


Figure 15:

### 0.3.1 Thick vs thin cylinders

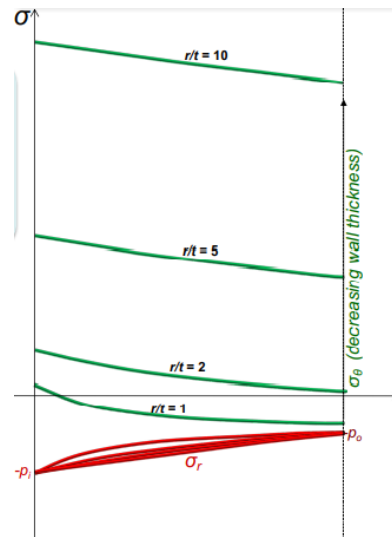


Figure 16:

Decreasing wall thickness with respect to radius, the hoop stress increases dramatically, but the radial stress does not change significantly. Assumptions of thin wall theories become more accurate:

- constant hoop stresses
- negligible radial stresses